BVS	2691
DATE	20 March 1992
REV	•
ORIGINATOR	J. Scilipoti

F12

OLS #12 BEARING RETROFIT

ACCEPTANCE TEST REPORT
VOLUME I OF III
SUMMARY AND SPECIFICATION REQUIREMENTS

(CDRL 006A1)

Contract F04701-90-C-0028

Prepared For

UNITED STATES AIR FORCE Headquarters, Space Division Los Angeles, California

Prepared By

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#### 1.0 <u>INTRODUCTION</u>

The OLS #12 Acceptance Test Report contains the technical data pertinent to the OLS #12 AVE system. This document is intended to present the Acceptance Test data in terms of the requirements of the Prime Item Development Specification (DMSS-OLS-300) and Interface Specification (IS-YD-810A) for testing associated with bearing retrofit (BVS 2579). A copy of the signed-off BVS 2579 is included in this report as Appendix A. During this same period, BVS 2600 "RDS Rework and Retest Procedure" was performed. Testing for BVS 2600 was done at the functional level only. Therefore, no test results are included as part of this ATR. However, a copy of the signed-off BVS 2600 is included in this report as Appendix B. A special test of T channel stability was also performed during retrofit. Results of this testing are compiled in BVS 2698, "OLS 12 Stability Testing".

Test results and data have been reviewed and verified by Westinghouse Electric Corporation and USAF representatives. System performance data, test histories, data summaries and system analyses are included in this report. In addition, a complete set of system log books are on file at the contractor's facility and are available for review. The Test History is in log books K41477-- and K40502--.

It is intended that this report provide a complete summary of OLS 12 performance relative to all requirements. Therefore, data showing performance for requirements not verified as part of bearing retrofit are also provided. When data from previous tests are provided it will be so noted.

This Acceptance Test Report consists of 3 volumes as follows:

BVS 2691 OLS #12 Summary and Specification Requirements

BVS 2692 OLS #12 Acceptance Vibration Report

BVS 2693 OLS #12 Alignment & Synchronization Curves

### 1.1 Summary of System - Specific Parameters

OLS software Program = OLSP02J.FS

Gain Constants and Sensor Switch Points

P(0) = 9.375 dB

P(1) = 51.75 dB

P(2) = 0 dB

P(3) = 29.75 dB

S(1) = 59.875 dB

S(2) = 22 dB

S(3) = 33.75 dB

(These may change during Early Orbit Calibration.)

PMT HV EST (A532) =  $3.634 \text{ volts } \pm .250V$ 

Cone Cooler S/N 024 with T detector S/N K-5

T Cold Patch EST (A549) curve - see Table next page.

T Cold Patch EST Voltage =  $2.211V \pm .200V$ 

TGAIN Left = 4

Right = 4

Both = 4

TLEVEL vs M1 temperature range - see second page following for table

VDGA constant for PMTCAL =  $(0440)_8$ 

Encoder Simulator Bias Constant = Prim -22 Redun -23

Encoder Simulator Separation Constant = Prime -7 Redund -6

# T COLD PATCH TEMP VS EST VOLTS CONE COOLER S/N 024 T DETECTOR S/N K-5

<u>T (deg k)</u>		EST (Volts)
95		5.655
96		5.248
97	2 11 11	4.874
98		4.529
99		4.212
100		3.920
101		3.651
102		3.403
103		3.174
104		2.963
105		2.768
106		2.588
107		2.422
108		2.268
109		2.125
110		1.993
111		1.871
112		1.757
113		1.651
114		1.553
115		1.462
116		1.377
117		1.298
118		1.225
119		1.156
120		1.092
121		1.032
122		0.976
123		0.924
124		0.875
125		0.829

OLS #12
TLEVEL <u>VS</u> M1 TEMPERATURE RANGE
T DETECTOR S/N K-5

TL	M1_TE	MP(°C)
1111	-26.019°	to -21.069°
1110	-21.069°	-16.120°
1101	-16.120°	-11.170°
1100	-11.170*	-6.221°
1011	-6.221°	-1.271°
1010	-1.271°	3.678°
1001	3.678°	8.628°
1000	8.628°	13.577°
0111	13.577°	18.527°
0110	18.527°	23.476°
0101	23.476°	28.426°
0100	28.426°	33.375°
0011	33.375°	38.325°
0010	38.325°	43.274°
0001	43.274°	48.224°
0000	48.224°	53.173°

TLEVEL command changes should be uplinked to the OLS as a function of M1 temperature to maximize T Channel output accuracy.

#### 1.2 Specification Pass-Fail Summary

The following sections of this Acceptance Test Report contain the test results as they pertain to the Development Specification requirements. Each Test Report paragraph heading is followed by the corresponding Segment Spec paragraph number in parentheses.

The table on the following page summarizes the OLS #12 pass-fail status vs. Development Spec. paragraph number.

DE	/ELOPMENT SPEC. PARAGRAPH NUMBER	PASS	FAIL
3.2.1.1.1.1	Infrared Spectrum	×	
3.2.1.1.1.2	Vis-Day Spectrum	×	
3.2.1.1.1.3	Vis-Night Spectrum		X
3.2.1.1.2.1	Fine Geometric Resolution - HRD	, <b>x</b>	
3.2.1.1.2.1	Fine Geometric Resolution - T	x	•
3.2.1.1.2.2	Smooth Geometric Resolution - HRD	×	
3.2.1.1.2.2	Smooth Geometric Resolution - T	x	
3.2.1.1.2.2	Smooth Geometric Resolution - PMT	x	
3.2.1.1.2.3	Data Sampling	x	
3.2.1.1.3.1	Along Track Geometric Accuracy	x	i
3.2.1.1.3.2/3	Along Scan Geometric Accuracy	x	
3.2.1.1.4.1.a	T Channel Radiometric Accuracy Repeatability	×	
3.2.1.1.4.1b	T Channel Radiometric Accuracy - Stability	х	
3.2.1.1.4.1c	T Channel Radiometric Accuracy - Fixed	x	
3.2.1.1.4.2	Daytime Radiometric Accuracy	х	
3.2.1.1.4.3	Nighttime Radiometric Accuracy	х	-
3.2.1.1.4.5.1	Terminator Location	х	<u>,</u>
3.2.1.1.4.5.2	Gain Change Rate	x ç	
3.2.1.1.4.5.3	Maximum Gain Settings	X	
3.2.1.1.4.5.4	Commandable T-Channel Gain	x	
3.2.1.1.4.5.5	Commandable T-Channel Level	X	
3.2.1.1.4.6.2/3	A/D Conversions & Algorithms	х	
3.2.1.1.5	Radiometric Resolution	x	
3.2.1.1.6.1	T Channel Noise	х	
3.2.1.1.6.2	L Channel Noise (Day)	x	
3.2.1.1.6.3	L Channel Noise (Night)	x	, 

DE	VELOPMENT SPEC. PARAGRAPH NUMBER	PASS	FAIL
3.2.1.1.6.4	Dark Current	x	
3.2.1.1.6.5	Stability 11 12 11 11 11 11 11 11 11 11 11 11 11	X	
3.2.1.1.6.6	Along-Track Noise Integration	х	
3.2.1.1.6.7	Glare Suppression	x	
3.2.1.1.7	Survivability	x	
3.2.1.1.8	Scan Angle	x	
3.2.1.1.9	Data Collection Rate	x	
3.2.1.2	Data Management	x	
3.2.1.3.1	28V Power Total Land Market Control of the Control	x	5-
3.2.1.3.2	5V Power	x	
3.2.2.1	Total Mass	x	
3.2.2.2	Component Mass	x	
3.2.2.3	Cable Harness Mass	x	
3.2.2.4	Dimensional Limits	x	
3.3	Design Features	x	
4.1/2	Environment	x	
5.1	Shipping & Storage	x	
INTERFACE SPEC P	ARAGRAPH NUMBER		
3.1.3	Alignment	x	

#### 1.3 Summary of OLS #12 Testing

- 03-25-91 Began RDS testing per BVS 2600
- 04-26-91 OSU(x) and SPS(x) vibration
- 05-20-91 Began Bearing Retrofit testing per BVS 2579
- 05-23-91 SSS(x,y,z) vibration
- 05-29-91 Testing stopped to use OLS 12 units with BTM SSS for special BTM T channel testing per BVS 2654
- 06-09-91 PSU(x) vibration per BVS 2657 (repair of broken wire)
- 06-26-91 Resumed Bearing Retrofit testing
- 07-01-91 Began Thermal Vac testing
- 07-20-91 Break vacuum due to PMT failure
- 08-06-91 OLS 12 to Blue Room
- 08-07-91 Replaced PMT with PMT from OLS14
- 08-09-91 Replaced EST/LMD with unit from OLS14
- 08-13-91 SSS sine and random vibration
- 08-16-91 Thermal Vacuum ambient testing
- 08-19-91 Restarted Thermal Vacuum testing
- 09-19-91 Thermal Vacuum testing complete
- 10-27-91 Final Blue Room testing complete

#### 1.4 <u>Configuration and Serialized Assemblies</u>

The configuration listing on the following pages includes the current configuration of the OLS #12 as of 12-03-86.

1-9

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
Key Drawing	536R500G01	F	5007
SSS Assembly	640R800G08	AM	5007
OSC Assy	623R765G06	AC	5007
HRD Assy	623R754G04	AB	0006
PWR Bd	623R758G03	R	0006
Pre Amp Bd	623R506G03	U	0006
<u>T-Chan</u>	633R049G04	R	0006
T-Chan Bd	633R178G03	AD	0006
Module	623R727G01	В	5009
Module	623R727G01	В	5010
VDGA/Lin Log	644R150G03	F	5007
Lin Log	644R127G03	Р	5007
VDGA	644R152G03	P	5007
VDGA	644R153G03	N	5007
Enc. OPT	688R705H01	C	009
PMT	644R909G04	P	0007
EMR Bd	644R905G03	D	0007
Switch Bd	644R903G04	M	0007
Doubler Bd	644R907G02	F	0007
Regulator Bd	644R807G03	Н	0006
Pre Amp Bd	644R935G03	J -	5008
HRD Post Amp	644R220G04	G	5007
Post Amp Bd	644R228G04	АВ	5007
EST/LMD	644R219G03	D	0007
EST/LMD Bd	758R142G02	E	0007
<u>Heater Cont</u>	633R053G09	J	5015
Elect Assy	633R052G03	V	5015

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
Heat Cont	633R053G10	J J	5016
Elect Assy	633R052G03	V	5016
Heat Cont	633R053G11	J <sub>m</sub> <sub>M</sub>	5017
Elect Assy	633R052G03	V	5017
Heater Cont	633R053G12	J	5018
Elect Assy	633R052G03	V	5018
Rel Mech I	640R701G02	<b>5. F</b> <sub>1000</sub>	5007
Rel Mech II	640R753G02	Н	5007
Rel Mech_III	640R381G02	H H	5007
T-Clamp	623R821G01	G	, T
<u>T-Cal</u>	623R920G01	_ B	
Aux Encd	640R846G04	G	5007
Bd Assy	640R825G04	F	5006
Bd Assy	640R844G04	J	5006
Wire Dia	682R239G03	L LEDE KOOL	- J. Ha
Wire Tab	318R708	В	:: "not a life
Wire Tab	315R386	C mm	nulli fasi
Wire Tab	318R709	(-)	30. II 190.
Motor Assy	623R894G01	В	73L0993
IMC/M3	623R858G02	D	5007
Cover, Cooler	640R320G01	(-)	5007
Cone Cooler	9RA5216H01	J	024
<u>ENPA</u>	682R215G03	M	5007
Al Bd	682R167G03	H	5008
A2 Bd	682R110G03	11 <u> </u>	5007
A3 Bd	682R112G03	Р	5007
Aux Encd B/U	682R300G03	<b>C</b>	5007
	9.4		

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DESCRIPTION	ASSEMBLY NO.	REV.	S/N
Al Bd	682R149G03	THE THE	5007
A2 Bd	682R151G03	E	5007
<u>BB1</u>	KG43		026
<u>BB2</u>	KG43		027
<u>BB3</u>	KG43		028
Ther. Blk. Kit	661R564G03	J	5007
GSSA/DOC	640R790G03	H	5007
GSSB	633R906G01	A III	5007
PR1	688R461H01	E	040
PR2	688R461H01	E	042
PR3	688R461H01	■ <sup>®</sup> E	043
PR4	688R461H01	E	041
Cable Assy	9RA5255H09	■ ¼□ T	006
Cable Assy	9RA5255H02	T I	006
Cable Assy	9RA5255H03	T	007
Cable Assy	9RA5255H04	Т	006
Cable Assy	9RA5255H10	Т	003A
Cable Assy	9RA5255H07	T	006
Cable Assy	9RA5255H06	T m	501
Cable Assy	9RA8118G01	F F	-
Coax Assy	644R327G01	В	-=
Coax Assy	644R327G02	В	-
Coax Assy	644R327G03	В	-
Coax Assy	644R328G01	С	-
Coax Assy	644R328G02	С	-
Coax Assy	644R328G03	С	-
Coax Assy	644R328G04	С	8 -

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
Coax Assy	644R328G05	C	aa
Coax Cable	644R328G06	C	-
Coax Assy	644R329G01	C	-
Coax Assy	644R329G02	C_A	-
Coax Assy	644R329G03	С	-
Coax Assy	644R329G04	C	-
Coax Assy	644R329G05	C	
Coax Assy	644R329G06	C	-
Coax Assy	644R329G07	_ C_	-
Coax Assy	644R329G08	g	-
<u>SPS</u>	651R390G01	AC	5007
Matrix	651R342G03	AV	-
<u>R/B</u> .	644R665G04	AE	5012
Matrix	644R081G03	reconstant.	***
Al Bd	640R618G03	" <u>H F</u>	5014
A2 Bd	640R518G02	P	5013
A3 Bd	640R520G03	The Page	5013
R/B	644R665G04	AE AE	5013
Matrix	644R081G03	. III 86 III L	-
Al Bd	640R618G03	ugu <u>mp</u> F	5014
A2 Bd	640R518G03	P	5014
A3 Bd	640R520G03	P <sub>e</sub>	5014
CU 1	640R612G02	J	5013
CU 1	640R612G02	o January J	5014
CU2	640R614G02	J	5012
CU 2	640R614G02	K	5013
AU 1	640R608G02	D	5013
5		(4	

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DESCRIPTION	ASSEMBLY NO.	REV.	S/N
AU 1	640R608G02	D	5014
AU 2	640R610G02	D	5012
AU 2	640R610G02	D	5013
MC1X	640R560G03	L	5014
MC1X	640R560G03	L L	5015
MC2X	640R562G03	U	5014
MC2X	640R562G03	U	5015
ROM	640R530G03	T T	5012
ROM	640R530G03	T	5013
Core	644R910H03	K	013
Core	644R910H03	K	014
SDS2	640R442G03	The neal T	5012
SDS2	640R442G03	N N	5013
SDS3	640R444G03	Sam Trans	5012
SDS3	640R444G03	N	5013
SDS4	640R446G03	T	5012
SDS4	640R446G03	T	5013
SDS5	640R498G03	P	5012
SDS5	640R498G03	Wall Mark	5013
CLSD	640R458G03	AD	5012
CLSD	640R458G03	AD	5013
SDS1X	640R660G04	AP	5012
SDS1X	640R660G04	AP	5013
FC-1	640R450G03	AA	5012
FC-1	640R450G03	AA	5013
FC-2	640R454G03	V	5012
FC-2	640R454G03	v	5013

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
FC-3	640R456G03	Υ	5012
FC-3	640R456G03	Υ,	5013
SDF-1	640R474G03	AH	5012
SDF-1	640R474G03	AH	5013
SDF-2	640R476G03	AH	5012
SDF-2	640R476G03	AH	5013
SDF-3X	640R540G03	H	5012
SDF-3X	640R540G03	Н	5013
SDF-4X	640R542G03	TT WITH BE	5012
SDF-4X	640R542G03	Н	5013
SDF-5X	640R544G03	N	5012
SDF-5X	640R544G03	N	5013
SDS-6	640R538G03	U	5012
SDS-6	640R538G03	mr ac (Salama	5013
SDS-7	640R546G03	Budino i edileri	5012
SDS-7	640R546G03	P	5013
4B	640R412G03	P	5012
4B	640R412G03	P	5013
7A	640R414G03	AB	5012
7A	640R414G03	AB	5013
7B	640R416G04	AR	5012
7B	640R416G04	AR	5013
1A	640R400G03	AK	5014
1A	640R400G03	AK	5015
1B	640R402G03	AD	5012
1B	640R402G03	AD	5013
FBC	640R448G03	N	5012

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DESCRIPTION	ASSEMBLY NO.	REV.	S/N
FBC	640R448G03	N	5013
RAM	640R558G03	L	5025
RAM	640R558G03	L	5026
RAM	640R558G03	ī.	5027
RAM	640R558G03	- E	5028
2A	640R488G03	Υ	5012
2A	640R488G03	Y	5013
2B	640R410G03	M	5012
2B	640R410G03	W	5013
3A	640R404G03	Υ	5012
3A	640R404G03	Y	5013
10X	640R572G03	J	5012
10X	640R572G03	J	5013
CLCL	640R406G03	AD	5012
CLCL	640R406G03	AD	5013
WF-1X	640R566G03	Р	5012
WF-1X	640R566G03	Р	5013
WF-2	640R432G03	Y	5012
WF-2	640R432G03	11111 MEH Y	5013
WF-3	640R622G02	E	5012
WF-3	640R622G02	E	5013
WF-4	640R436G04	L	5012
WF-4	640R436G04	L	5013
WF-5	640R438G03	W W	5012
WF-5	640R438G03	W	5013
9A	640R420G03	AE	5013
9A	640R420G03	AE	5014

<u>DESCRIPTION</u>	ASSEMBLY NO.	REV.	S/N
9BX	640R586G04	F .	5013
9BX	640R586G04	00 <u>F</u> A	5014
9CX	640R570G03	N	5013
9CX	640R570G03	N N	5014
WF-6	640R568G03	H_	5013
WF-6	640R568G03	Fares of H	5014
<u>OSU</u>	640R960G03	Y	5007
Matrix	522R783G02	egrum an Yaqı	5007
Al III	640R522G03	ma n T <sub>PA</sub>	5007
A2	640R524G03	N	5007
Bottom	644R047G03	and talke	5007
Тор	644R046G02	P	5007
<u>SPU</u>	758R040G01	( Level III	5007
Matrix	640R927G02	III AFERRAVIA	- 2
SSP-8	640R552G03	E E	5014
SSP-8	640R552G03	E_	5015
RTD-1	640R508G03	AH	5012
RTD-1	640R508G03	AH	5013
RTD-2	640R510G03	AP.	5012
RTD-2	640R510G03	AP	5013
RTD-3	640R512G03	K	5012
RTD-3	640R512G03	K	5013
RTD-4	640R526G03	N .	5013
RTD-4	640R526G03	N	5014
RTD-5	640R514G03	R	5012
RTD-5	640R514G03	R	5013
SSP-1X	640R550G03	som and J	5012

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
SSP-1X	640R550G03	J. III	5013
SSP-2	640R462G03	V	5012
SSP-2	640R462G03	٧	5013
SSP-3	640R464G03	υ	5012
SSP-3	640R464G03	υ	5013
SSP-4	640R466G03	M	5012
SSP-4	640R466G03	M	5013
SSP-5	640R468G03	P	5012
SSP-5	640R468G03	P	5013
SSP-6	640R470G03	R	5012
SSP-6	640R470G03	R	5013
SSP-7	640R472G03	V	5012
SSP-7	640R472G03	V € TOTAL	5013
SSP-9	640R554G03	J	5013
SSP-9	640R554G03	J	5014
<u>PSU</u>	758R050G02	Y	5007
Matrix	640R620G01	F	0004
RFI Plate	690R891G01	A	5007
Reg Assy	682R089G03	To a	5004
Misc Bd	644R302G03	R	5007
T-Chan CG	688R483G03	G	5007
T-Left	688R485G03	F 55 9	5007
T-Rgt	688R487G03	G	5007
T-Chan BU	688R489G03	F	5007
T-Ana Fil	688R491G03	н	5012
T-Ana Fil	688R491G03	H	5013
L-Ana Fil	688R493G03	G A	5012

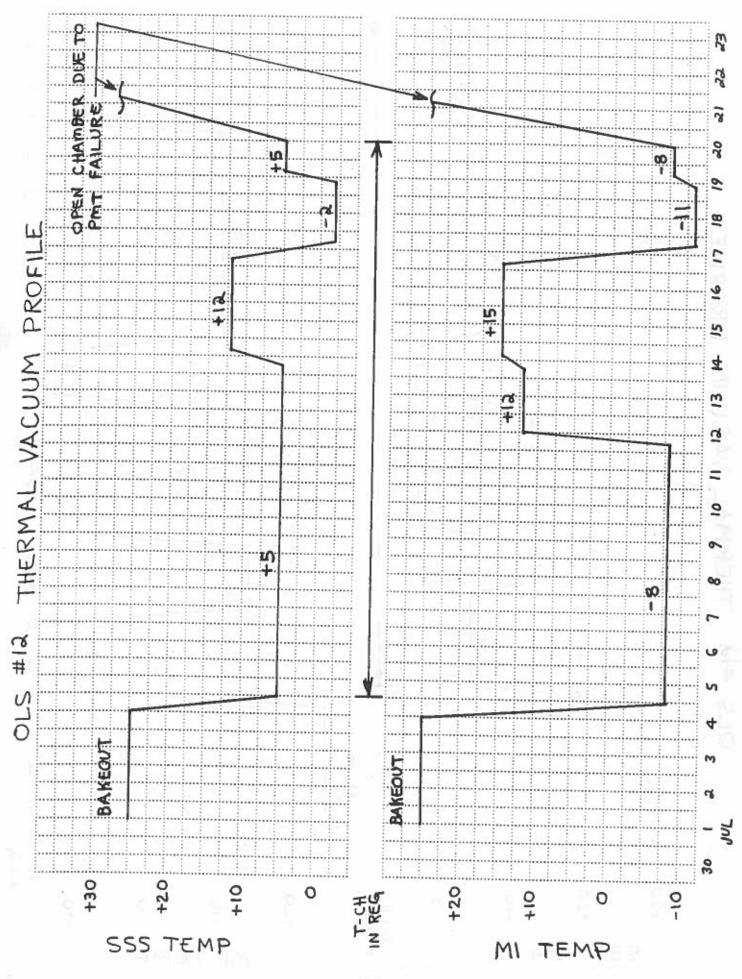
<u>DESCRIPTION</u>	ASSEMBLY NO.	REV.	S/N
L-Ana Fil	688R493G03	G	5013
PSU TRA BLK	640R998G03	н	5013
PSU TRA BLK	640R998G03	Н	5014
DME	688R481G03	G	5013
DME	688R481G03	G	5014
IMC	644R864G03	E	5007
Relay-1	688R501G03	E	5007
+5V	644R078G03	P	5007
Relay-2	688R502G03	D	5007
+12VDA	688R499G03	D	5013
+12VDA	688R499G03	D	5014
Dual ENPA	640R616G02	J	5007
Relay-3	688R503G03	С	5007
-12 <b>V</b>	644R069G03	_N	5007
Relay-5	688R505G03	С	5007
Relay-4	688R504G03	C	5007
+12V Vm	688R500G03	С	5007
MC	688R495G03	<sub>E</sub> F	5013
MC	688R495G03	F	5014
СРН	688R497G03	D	5007
Wire Tab	318R249	N	-
Enable	682R381G03	D	5007

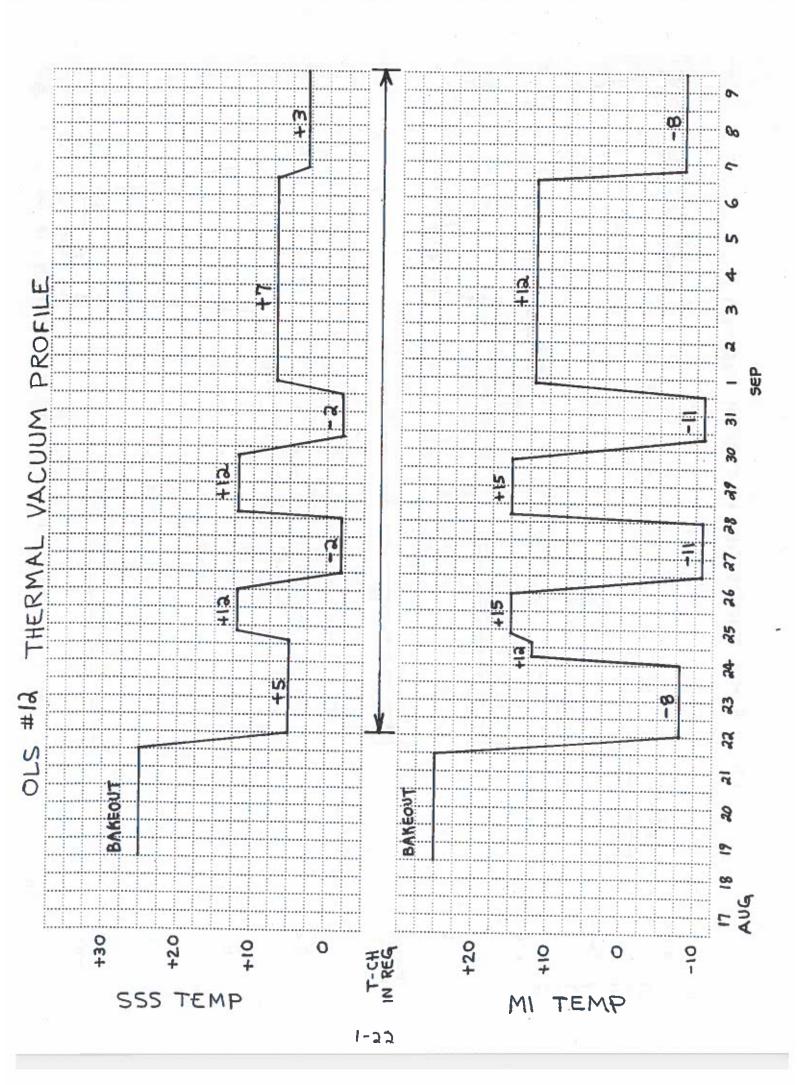
#### 1.5 Thermal Vacuum Profiles

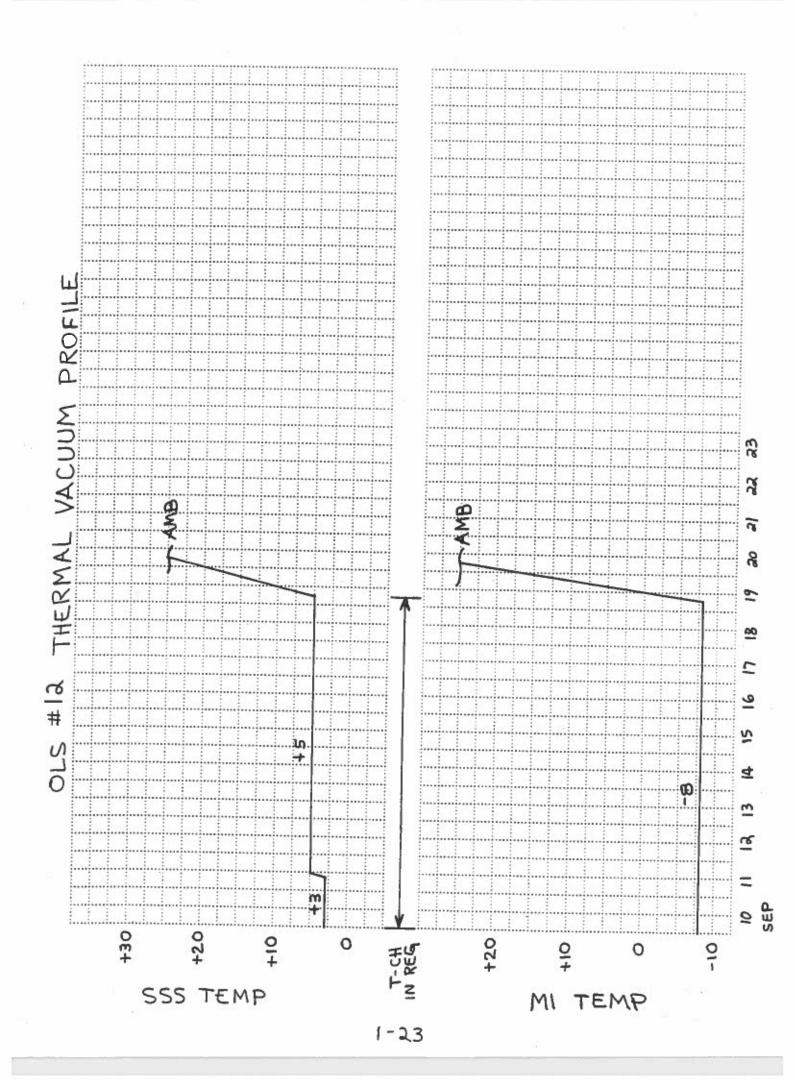
The OLS #12 AVE underwent a series of Thermal Vacuum Tests.

The profiles on the next pages represent the history of pumpdowns,

SSS temperature and M1 temperatures experienced by the OLS #12 AVE.







## 1.6 <u>Test History Calendar</u>

The test history calendar is a capsule look at the day-to-day progress of the OLS #12 AVE throughout its testing period at WEC.

16		_ =			
DATF MAR '9,				-	
NA		31	4.		3
Ļ	7	. 6	9	N N	200
TA(		<u> </u>		1	Stopped because of lack of government funding
<u></u>				4	Stopped Stopped because lack of government funding
		<b>(0)</b>	5)	<u>Z</u>	क्रिंद्र व इक्रद्म
		= 1			1.19
		<b>E</b>	4		065#16 Testing
≿				12	20 O
-0F			3		1,3 6
HISTORY		9	<u> </u>	92	22 015#16 Testing
					1 , 4
ST		33			
TEST		<b>V</b>	2	61	Amb s System
	2	4			lest oo b. Tests
8	17	20		E 10	25 ROS TES+ BVS 2600 Amb sub- System Tests
2#7	10.0	4	=	<b>(0)</b>	85 F
JNIT 015 #12	1-			0 2	
	m	<u>m</u>			*
<b>Z</b> '	Tall .	WII -	9	[1]	<b>X</b>

	=	 <del></del>			
DATE APR '91	9	13 12A102 failure BUS 2646	হ	Amb sub- system tests	
DAT	<u></u>	BVS 2600 T/s RDS Amb sub- system tests anomally	61	4 050 Vib	
	4	11) Restarted BVS 2600 Amb sub- system test	81	Amb sub System tests	
HISTORY	m	0	L)	Restarted Amb sub. 5 BVS 2600 System tests Testing	
TEST P	7	6	9)	23 12A102 failure BVS 2646	30
UNIT OLS#12		<b>©</b>	5)	<b>2</b> 7	sub n tests ned ny pair
LIN	=	7	<del>6</del> /	72	Amb subsystem tes removed 2A204 for repair

JNIT OLS#12		TEST F	HISTORY	100	DATE MAY '91	AY '91
				2A204 reinstalled	3 Amb Sub- 4 Heat Cycle #1  System Tests To cold cycle #1  To Hot Cycle #1 Cold Cycle #1  To Hot Cycle #1 To Hot Cycle #2  Hot Cycle #2	4 Het Cycle #1  To Cold Cycle #1  Cold Cycle #1  Cold Cycle #1  To Hot Cycle #2  Hot Cycle #2
To Cold #2 16 @ 0320 16 @ 0320 10 #3@ 0900 TO 1315 Cold #3@1525 16 #3@1525 16 #4 #2 133		7 To Cold #5 @ 1335 Cold #5 @ 1810	8 To Hot #6 @ 1930 HOT Cycle #6 @ 2330	9 75 Cold & 6 Cold & 0700 To Hot of 1145 Hot & 1700 To Cold #7 & 2230	10 Co 16 Cycle #7 11 To Co 16 #8 To Hot #8@0900 Co 16 Cyc 10 #8 HoT Cycle # 8 @ 0 915	6 2200 50500 60500 60500 60500 60500
Cold #8	Cycle #8	14 Cold Cycle #8 To that (2) 1200	15 70 Co/d (2) 00/0	16 Cold to Amb @ 2215	17 Units 18 moved to Blue Rm	
1 3 2 1 2 2 1 2 2 1 2 1 2 1 2 1 2 1 2 1	Beyan Retrol. 15 2579 25 Fing	11 6-2k 6-3-1 14011PT	22 AHSF B 11 PT 625-1 6-325 AHSF 11 6-7-2 67-1 6-7-2 6-3-2	23 APC11PT 7-8 555 U.B	24 7.8 25 Funct Tests	
92	ETJ 6121 6131 61612 MHA707 AHSF3PTT AHSF3PTT AHSF7PT Funct TESTS	28 Fait Tests APC. 107 APC. 107 60. 27 67.72 MPA. 107 67.43 67.43 67.43	29 AHSFB 9PT Removed 12A 102 12A 103, 12A 10Y For BUS 2653	30 Installed BTM 12 A102. 12 A103, 12 A104	m	

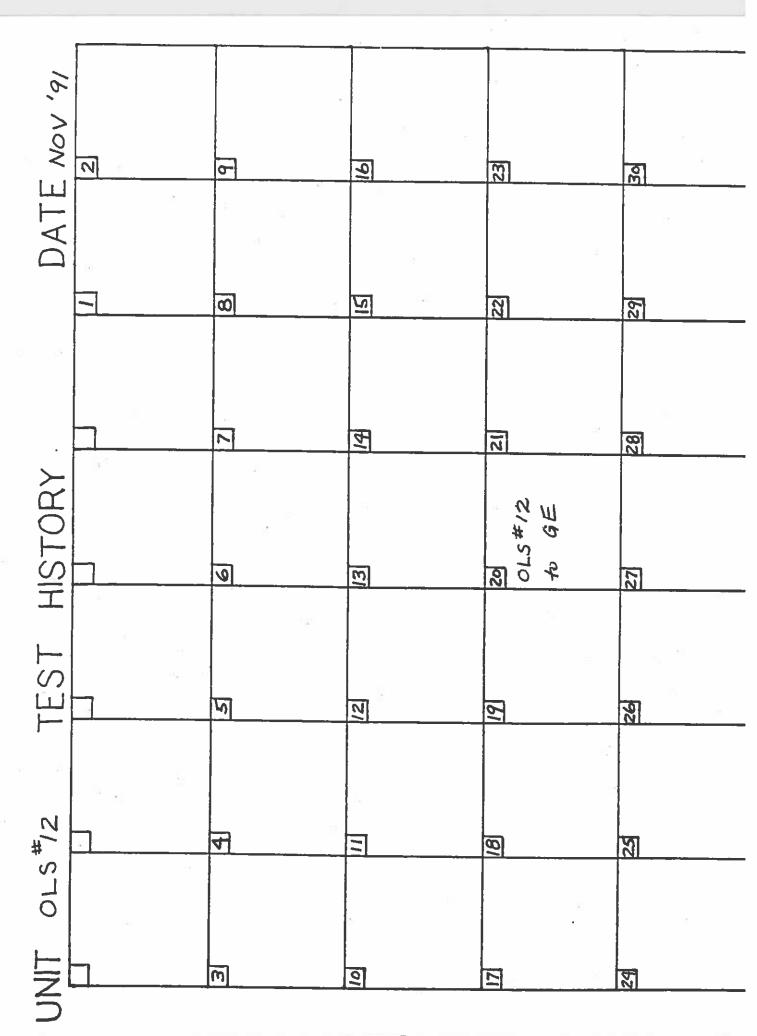
DATE JUNE '91			\frac{1}{2}	22	Fund Tests 62324 629
DATE		2	₹	72	28 MHA707 2 6-5-1 6-6-1 6-6-3 6-6-3 6-7-1 Funct Tests
		9	<u>E)</u>	22	27 6+3+1 6+3+1 Funct Tests AHSF70T
HISTORY		<u>ত্</u> য	[2]	6/	26 Restanted BVS 2579 Testing Funct Tests
TEST		4	11	8/	BTM BVS 2654
#12		3 OLS#12 Digital Boxes used to support BTM T-Ch Test BUS2654	0/	[7]	<b>5</b> 2
JNIT 065#12	30 SIMFLT 7×9 7×10 Pumpdown @ 2000	7	6	9/	

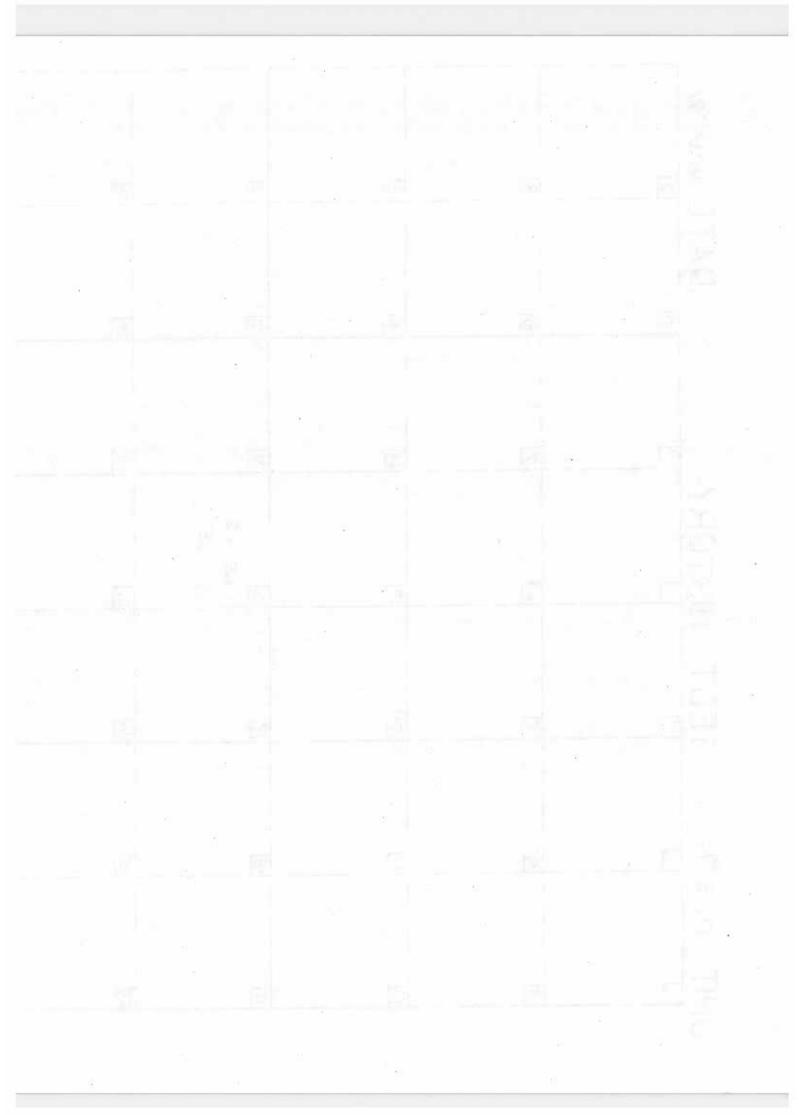
DATF JULY '91	6 ASV 210 Funt Test 77237229 AHCII PT 7725722 6+2+2 6+2 6	13 7/23 625-1 625-1 7/2572 ASV 270 MICIPE TOCKM3 10-12/+15	20 TSTABILITY 22 -25 Chamber Warm-up # Vent	22	
DAT	AHSF3PTT 6+2+3A TDCRM29 4+9+1	12 6-2-3A ASV 210 310 AHSF 7PT APC 7 PT 7 121 7231H WHC 7PT	1976 +51-8 7248 +51-8 @ 0545 7-57881217	92	
	4 7×7 Finct Tests	11 TSTABILITY 12 6-2-3A Runs 8-13 ASV 210 310 MI to +12 D1820 AHSF 70T MI=+12 @2400 APC 7 PT MHC7PT	18 Funct Tests 1970 +57-8 TIZITZZI 8 75-48 6-2-2 PMT Prublem TIABILITY 14-21	No System Tests	
HISTORY		BSIMFLT TSTABILITY Runs 1-7	17 cold Soak #1 60 0915 Funct Tests SIM FLT	MSPTST	No System Tests
TEST H	2 72 Hour Buke - Out	9 AHSFB90T 6x5x2 7x12 6x3x4 9x1x1 6x3x4 9x1x1	16 Fund Tests To cold Souk @ 1915 9+1+3	STOTLMTST	020
	12 Hour Bake-Out	BTIIGTZZOA TDCRM3A ATCIIPT APCIIPT 6x9	M	22 T/s PMT Problem SSS removed from chamber	62
UNIT OLS #12		17/29 7223 A AHSF3 PT T T 13/ 7221 A ASV 210 310 WTC 110 T Funct Tests ASV CROSS 6574 6554	+12/+15 60300 6×2×2 121 T2318	7/5 PMT Problem AM120-29	88

DATE AUG '91	8	10 6,6,2 MPA 11PT 6,2,1 6,3,1	17 Funct Tests 61571 61571 61571 61571	29 MI - 4-12 - 4-5/4-12 6 0636 ARC7 PT 64253A TIZITZ31H TDCR M3K TO 16T SOAK D 15-45 HOT 90/7-12 HOT 90/7-12	31 SIMFLT Funct Tests 10 Hot Limits (6) 1330 17/+12(8) 2100
DAT	73	Asy now on 4x5x1 4x5x1 4x5x1	16 Moved System to TV Funct Tests 4,11x1	23 757 ABILITY 26-38	30 Cold Soak #2 © 0700 61272 T/2/72318 Funct Tests
	No System Tests	Funct Tests	MPR7PT MPR7PT Funct Tests 7x8		29 SiMFLT STS Software Problem Fract Tests To 6016 Soak#2 (2) 1800
HISTORY		System System OLS#12 PMT Assy now on OLS#12	AHSF7PT AHSF3PTI 6x2x/ 6x2x/ 6x3x/ 6x3x/ 6x3x/ 6x3x/	72 Hour Bake-Out	28 To the Seak 10T 508K #2 60000415 60002 60002 Funct Tests
TEST		6 Digital Units moved to Blue Rm	AHCIIPT SSS Vib Funct Tests	72 Hour Buke - Out	Funct Tests
		No No System Tests	APCIIPT	19 72 Hour Bake-Out	26 To Cold Soak 9-1+3 9-1+3 Cold SOAK #1 6+2+2 7121 72318
JNIT OLS #12		4	# 6+7×1 6+7×2 APC11 PT 7+8 6+6+1 6×3+2	No. 95	ES SIMFLT Funct Tests TIZITZ31B

HISTORY. DATF SEPT '91	4 6-11-2 5 AHSFB9PT 6 6-11-2 7 6 6-11-2 7 6 6-11-2 7 6 6-11-2 7 6-11-2 7 6-11-2 7 6-11-2 7 6-11-2 7 7-4 A ASF ATS7PT 51 18-12 75-4 A ASF ATS7PT 51 18-12 75-8 1900 6-1	11 +5/-8 @ 0010 12 Reins 47-51 7/25 7227 C TSTABILITY FUNCT TESTS APC7PT MPA7PT PRINCE 39-46 6222 APC7PT MPA7PT PRINCE TESTS APC7PT MPA7PT ASV 210 210 6224 6234 CH37 TISOTZZZ C CH37 TISOTZZZ C TISOTZZ C TISOTZZ C TISOTZZ C TISOTZZ C TOC RM3C	18 TSTABILITY Chamber 7/5 RTD Problem M Began chamber 7/5 RTD Problem M Warm-up @ 1830	7 7/5 RTO 7/5 RTD No Testing	
TEST H	3 AHC7PT 6224 6225 6333 6261 MPA7PT 6262 6263 62355	10 SIMFLT 6+11+ 6+3+4 Funct Tests 7-1-8 @ 2020	SIMFLT GEPIG	MSPTEST 7	
	2 T 119 T 2318 6x 2x 2 AHSF 3 P T I 6x 3x / Funct Tests TDC RM38	9 9-1-1 6-11-1 6-3-5 PRY Problem	16 TSTABILITY Runs 58-63 GEP16	SPTEST	T/s RTD
UNIT OLS#12		## PER PRINTER TEST AND TEST OF THE PER POTENTIAL TO TEST OF T	15 AHSF89PT 6-1121 6-1124 94121 757AB1217 Runs 52-57	ASPTEST	No. Testing

			33		3 -20
DATE Oct '91	<b>15 P</b> <sup>1</sup> 0	2	61	First Tests 6x4+3A 75E7 6x2x1	
DAT	MSPTEST System removed from TU chamber	23	<u> </u>	52	
	MSPTEST	0	<u>L</u>	<b>2</b>	8
HISTORY	2 7/5 RTD	6	9]	<b>8</b> 2	30 RDS Encryption Tests
TEST	Support for OLS#13 BB	0	<u>S</u>	252	Ed Funct Tests
0LS #12		7	<u>4</u>	12	Funct Tests 7×8
OLS					SET SFIIPT 7.5 FTESTS
LING	<b>1</b>	9	<u>m</u>	8	AHS Franch





#### 2.0 <u>DEVELOPMENT SPECIFICATION REQUIREMENTS</u>

### 2.1 Spectra (3.2.1.1.1.1, 3.1.1.1.2, 3.1.1.1.3)

The original OLS #12 Channel spectral responses were calculated by computer programs (GAINSET for L Channel and TGAIN for T channel) utilizing vendor spectral response data for the detectors, mirror and lens transmissivities/ reflectivities, and solar & lunar spectral radiance.

Since the original publication of OLS #12 spectral response, more current data has become available for HRD and PMT spectral response. The HRD total spectral response has been recalculated using average telescope data for OLS #13 thru OLS #16, measured data for the OLS 16 relay optics and a typical HRD detector response.

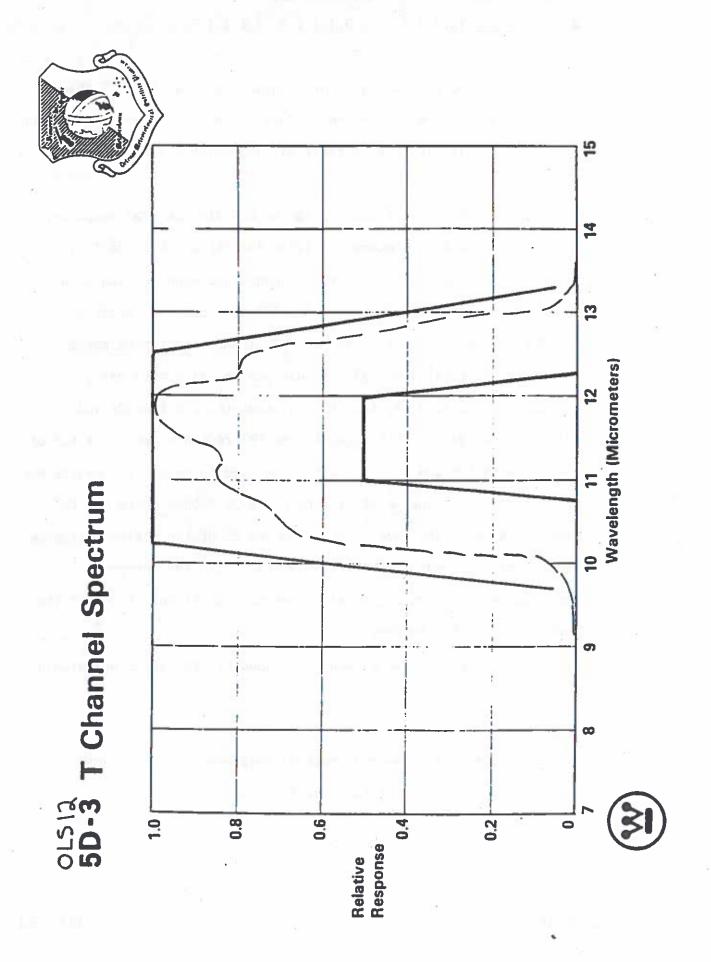
The PMT total spectral response was re-calculated using replacement PMT (S/N 16) from OLS 14, measured OLS #16 ORA and OLS #16 telescope data. This total system PMT response, which is out of spec between 500 and 530 nm, is the one used to report the degree of spec compliance. The worst case point is at 520nm, where the PMT response exceeds the specified maximum by 8% of the maximum response. The out-of-spec region is relatively small, and the increased response in the 500nm region will have no significant effect on the night time visible imagery.

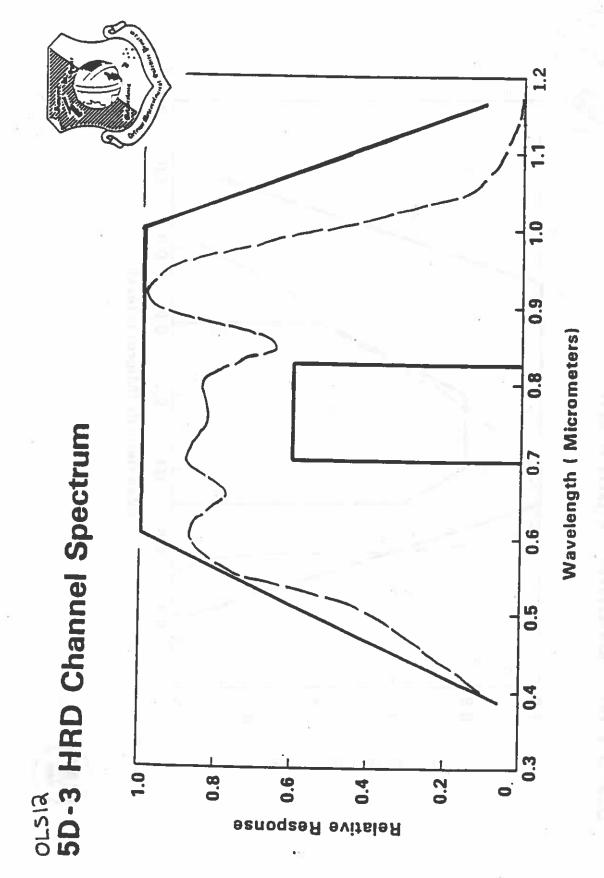
The OLS #12 T channel and L Day spectral responses are within specification.

ATTACHMENTS: OLS #12 HRD Channel Spectral Response.

OLS #12 PMT Channel Spectral Response.

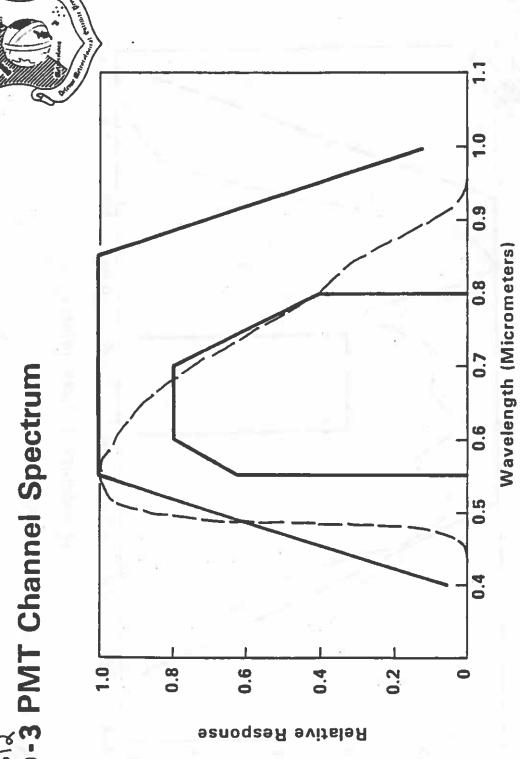
OLS #12 T Channel Spectral Response.







oLSIA 5D-3 PMT Channel Spectrum





- 2.2 GEOMETRIC RESOLUTION
- 2.2.1 Fine Geometric Resolution Infrared (3.2.1.1.2.1)
- 2.2.1.1 Baseline (Orbit Nominal)

The TF Surface Resolution Parameter (SRP) is within the development specification limits.

The VAX Computer programs calculate and plot the Fine Primary SRP, and the T Right & Left Fallback modes. In addition, all Specification required modes are tabulated and presented. The designations on the graphs are defined as follows:

TFP T Fine Primary Electronics

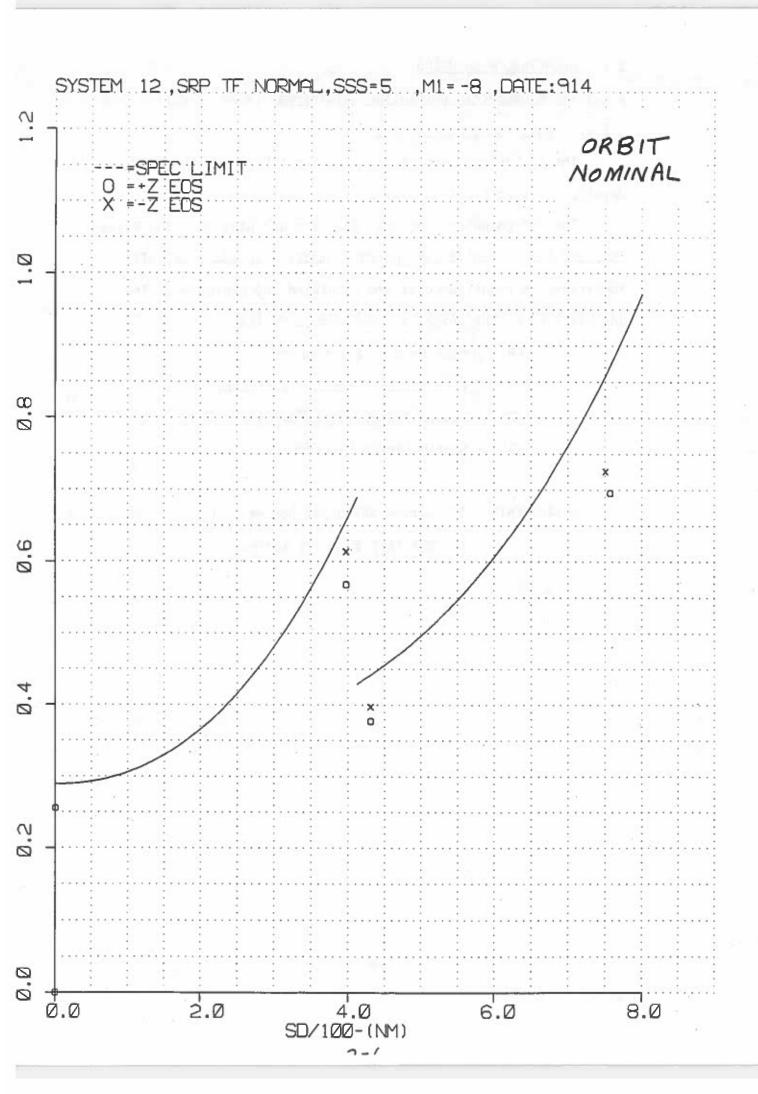
TFB T Fine Backup (Redundant) Electronics

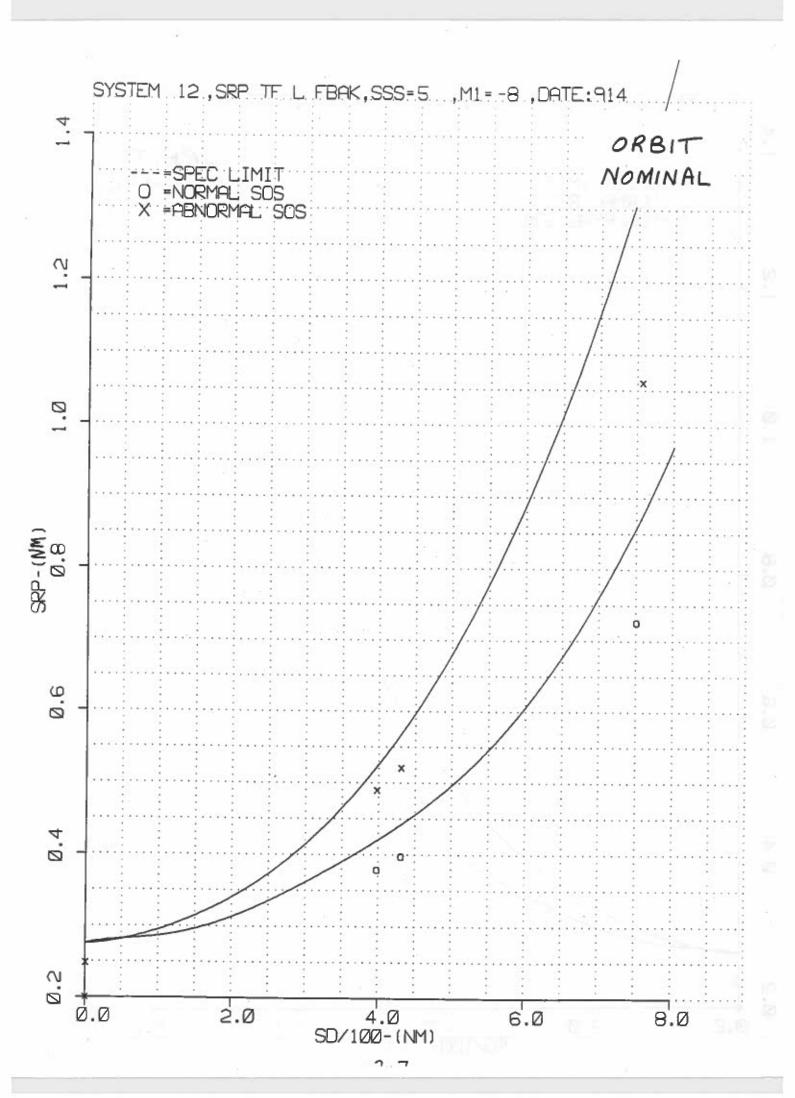
TSP T Smooth Primary Electronics

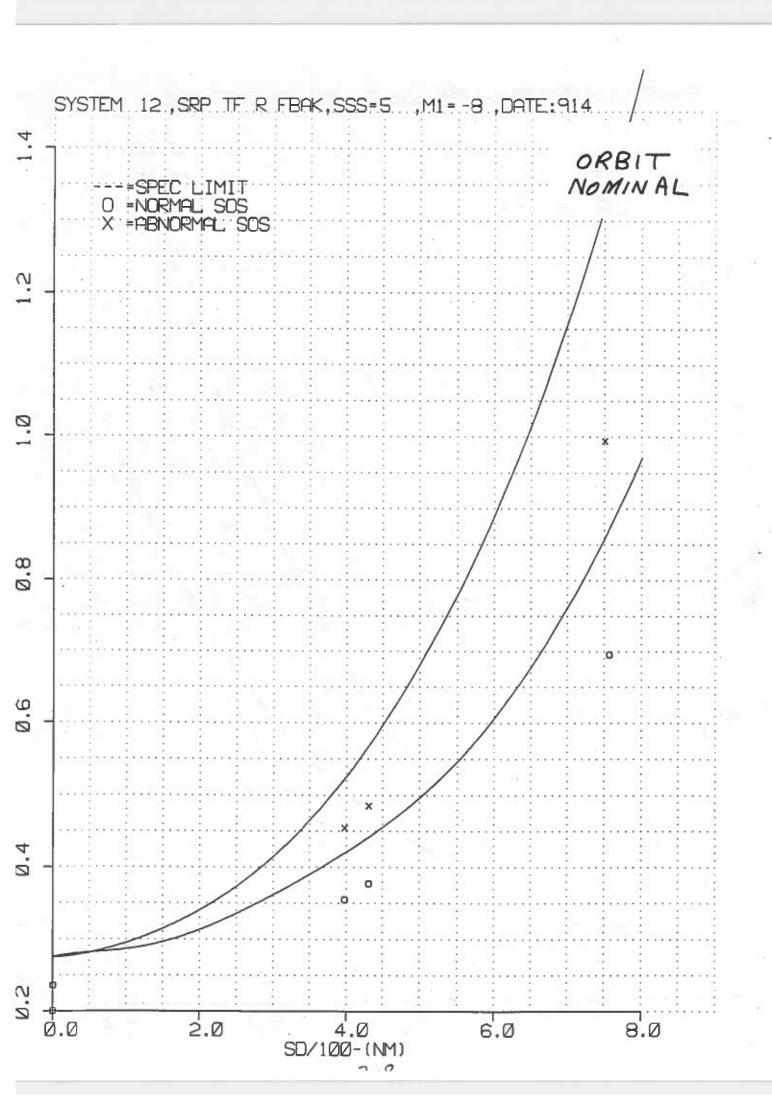
TSB T Smooth Backup Electronics

ATTACHMENTS: TF Curves SRP Orbit Nominal

TF SRP Tables Orbit Nominal







#### T, COMPLETE, SRP(NM)

	FLT. NO. = 1	D ENU =	4 SSS=	5DEGC M1=	-BDEGC DATE:	914
	: E1. NO 1	E 5144	7 333-	SDEGC 111-	ODESC DATE.	714
SEG	SUR. DIST.	TFP	TFB	TSP	TSB	
	(NM)					
LFT	<b>−750.</b>	0. 725	0. 722	1. 732	1. 675	
MID	<b>−750</b> .	1. 253	0.000	1.828	1. 782	
RGT	<b>−750</b> .	0. 995	0. 989	1.766	1.713	
LFT	0.	0.000	0.000	0. 000	0.000	
MID	0.	0.000	0. 000	0. 000	0.000	
RGT -	0.	0. 000	0.000	0.000	0.000	
LFT	-431.	0. 397	0. 391	1. 461	1. 410	
MID 💍	-431.	0. 631	0.000	1. 488	1. 439	
RGT	-431.	0. 485	0. 479	1. 461	1. 410	
LFT	-398.	0. 379	0. 373	1.408	1. 359	
MID	-398.	0. 614	0.611	1. 429	1. 382	
RGT	-398.	0. 454	0. 447	1. 406	1.357	
LFT	0.	0. 000	0. 000	0. 000	0. 000	
MID	0.	0.000	0.000	0. 000	0.000	
RGT	0.	0. 000	0.000	0.000	0.000	
LFT	0.	0. 248	0. 245	0. 968	0. 935	
MID	0.	0. 255	0. 252	0. 968	0. 935	
RGT	0.	0. 236	0. 234	0. 966	0. 932	
LFT	0.	0.000	0.000	0. 000	0.000	
MID	0.	0.000	0. 000	0. 000	0. 000	
RGT	0.	0. 000	0.000	0. 000	0. 000	
LFT	398.	0. 489	0. 484	1. 411	1. 363	
MID	398.	0. 568	0. 565	1. 420	1. 372	
RGT	398.	0. 355	0. 351	1. 397	1.348	
LFT	431.	0. 521	0. 515	1. 466	1. 415	
MID	431.	0. 573	0. 000	1. 479	1. 430	
RGT	431.	0. 377	0. 373	1. 455	1. 405	
LFT	0.	0. 000	0.000	0. 000	0. 000	
MID	0.	0. 000	0. 000	0. 000	0. 000	
RGT	0.	0.000	0. 000	0. 000	0. 000	
LFT	<b>757</b> .	1.061	1. 054	1. 785	1. 735	
MID	<b>757</b> .	1. 449	0. 000	1. 900	1.860	
RGT	757.	0. 695	0. 692	1. 715	1. 658	

## T, COMPLETE, SRP RATIO

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	-750.	0. 845	0.841	0. 770	0. 745
MID	-7 <b>5</b> 0.	0.000	0. 000	0.813	0.793
RGT	<b>−750</b> .	0. 753	0. 749	0. 785	0.762
LFT	0.	0. 000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	Ο.	0.000	0.000	0.000	0.000
LFT	-431.	0. 900	0. 885	0. 912	0.881
MID	-431.	0. 000	0.000	0. 929	0. 899
RGT	-431.	0. 854	0.843	0. 912	0.881
LFT	-398.	0. 903	0.888	0.918	0.886
MID	-398.	0. 934	0. 930	0. 932	0. 901
RGT	-398.	0.869	0.855	0.916	0. 885
LFT	0.	0.000	0.000	0.000	0. 000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	0.	0. 902	0.892	0. 922	0. 890
MID	0.	0. 882	0.871	0. 922	0. 890
RGT	0.	0. 857	0. 852	0. 920	0.888
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0. 000	0. 000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	398.	0. 936	0. 925	0. 920	0.889
MID	398.	0. 845	0.860	0. 926	0. 894
RGT	378.	0. 846	0.838	0. 911	0. 879
LFT	431.	0. 916	0. 906	0. 915	0.884
MID	431.	0. 000	0. 000	0. 924	0. 893
RGT	431.	0. 854	0.844	0. 909	Q. 877
LFT	0.	0.000	0.000	0.000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0.000	0.000	0. 000	0. 000
LFT	757.	0, 790	0. 785	0. 790	0. 768
MID	757.	0.000	0.000	0. 840	0. 823
RGT	757.	0. 797	0. 793	0.759	0. 734

#### TF, LEFT, PRIMARY

FLT. NO. = 12	ENV. =	4 SSS= 5	DEGC M1= -8DEGC	DATE: 914
SUR. DIST. (NM)	SRP	ACTUAL (NM)	SRP RATIO	
<b>-750</b> .		0. 725	0. 845	
0.		0.000	0. 000	
-431.		0. 397	0. 900	
-378.		0. 379	0. 903	•
0.		0.000	0.000	
0.		0. 248	0. 902	
0.		0.000	0. 000	
39B.		0. 489	0. 936	
431.		0. 521	0. 916	
0.		0.000	0. 000	
<b>757</b> .		1.061	0. 790	

#### TF, LEFT, BACKUP

F	LT. NO. = 12	ENV. =	4 555=	5DEGC	M1= -BDEGC	DATE:	914
9	UR. DIST. (NM)	SRP	ACTUAL (N	M) S	RP RATIO		
	<b>-750</b> .		0. 722		0. 841		
	0.		0.000		0. 000		
	-431.		0. 391		0. 885		
	-398.		0. 373		0.888		
	0.		0.000		0.000		
	0.		0. 245		0. 892		
	0.		0.000		0.000		
	378.		0. 484		0. 925		
	431.	- 2	0. 515		0. 906		
	0.		0.000		0. 000		
	757		1 054		0.785		

## TF, RIGHT, PRIMARY

FLT. NO. = 12	ENV. =	4 SSS=	5DEGC	M1= -8DEGC	DATE:	914
SUR. DIST. (NM)	SRP	ACTUAL (N	M) S	RP RATIO		
<b>−750</b> .		0. 995		0. 753	100	
0.		0. 000		0. 000		
-431.		0. 485		0. 854		
-3 <b>78</b> .		0. 454		0.869		
0.		0.000		0.000		
0.		0. 236		0. 857		
0.		0.000		0. 000		
398.		0. 355		O. 846		
431.		0. 377		0. 854		
0.		0.000		0. 000		
757.		0. 695		0. 797		

## TF RIGHT, BACKUP

MET. NU. = 12	ENV. = 4	355= 5DI	EGC MI= -BDEGC	DATE:	914
SUR. DIST. (NM)	SRP A	CTUAL (NM)	SRP RATIO		
<b>-750</b> .	0.	. 989	0. 749		
0.	0.	000	0. 000		
-431.	0.	479	0. 843		
-398.	0.	447	0. 855		
0.	٥.	000	0. 000		
0.	0.	234	0. 852		
0.	0.	000	0. 000		
398.	0.	351	0. 838	10	- 1
431.	0.	373	0. 844		
0.	0.	000	0. 000		
757.	0.	692	0. 793		

## 2.2.1.2 <u>Acceptance - Vibration</u>

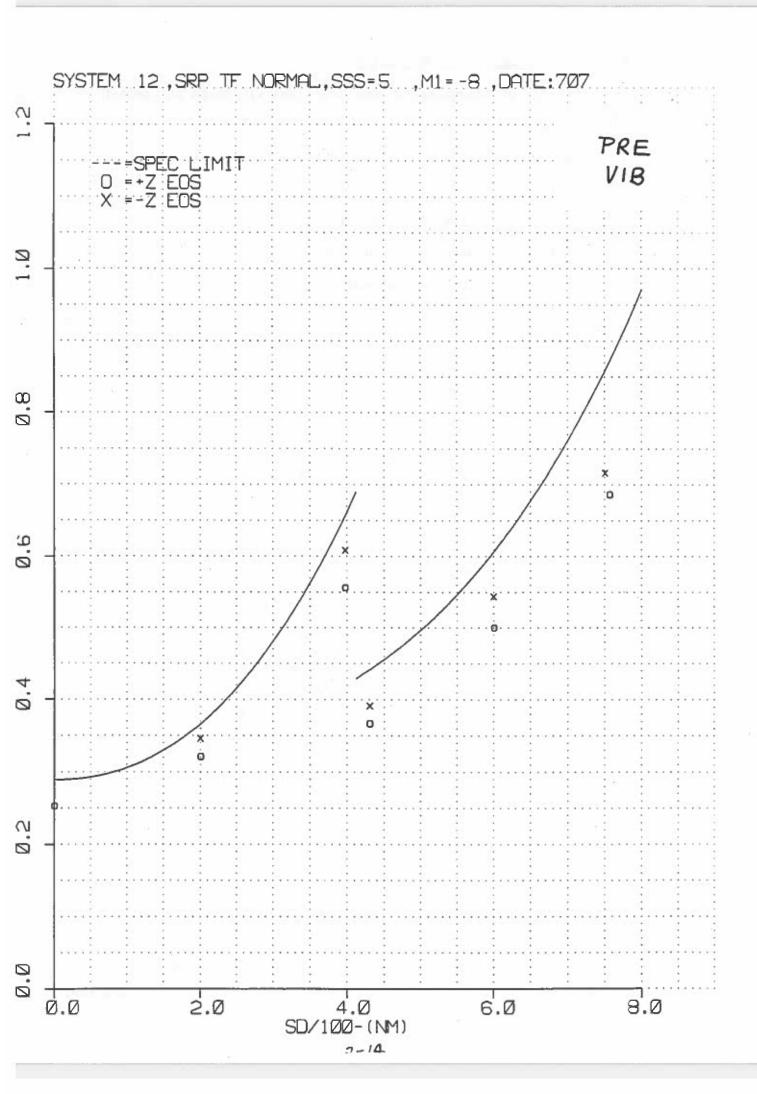
OLS #12 underwent acceptance level SSS vibration per DMSS-OLS-300 with cone cooler S/N 024 on May 23, 1991. The pre-to-post vibration SRP performance is shown on the attached curves and tables.

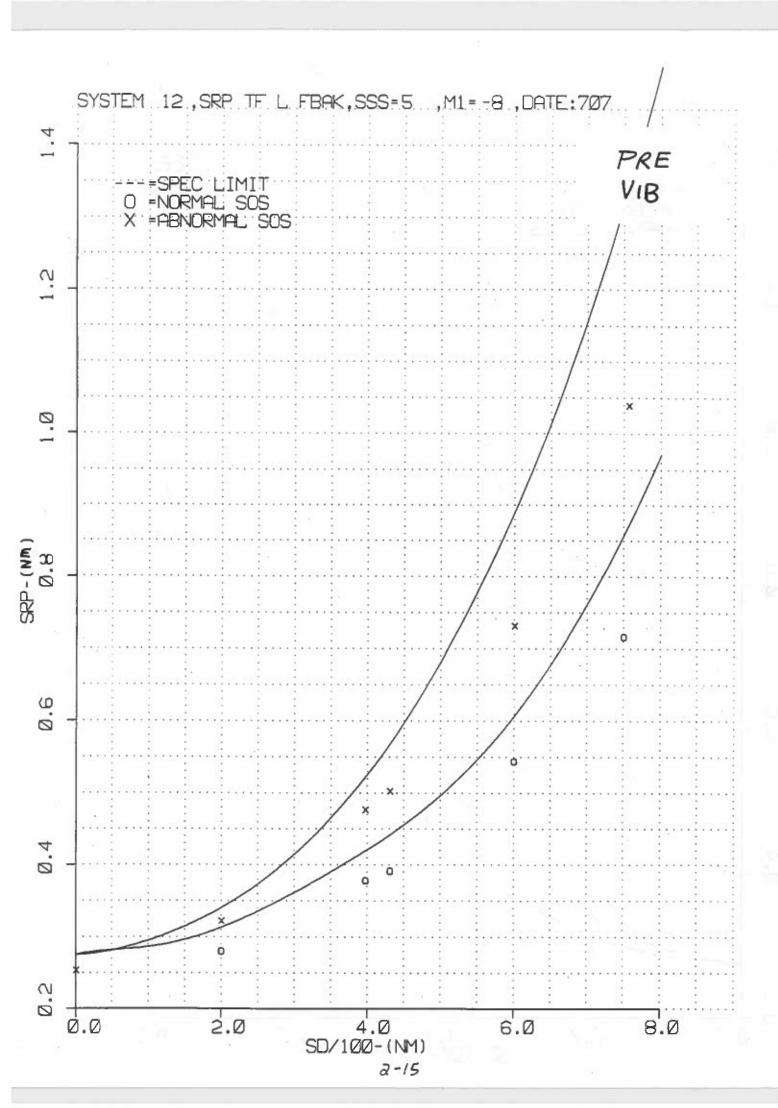
ATTACHMENTS: TF SRP Curves Previbration.

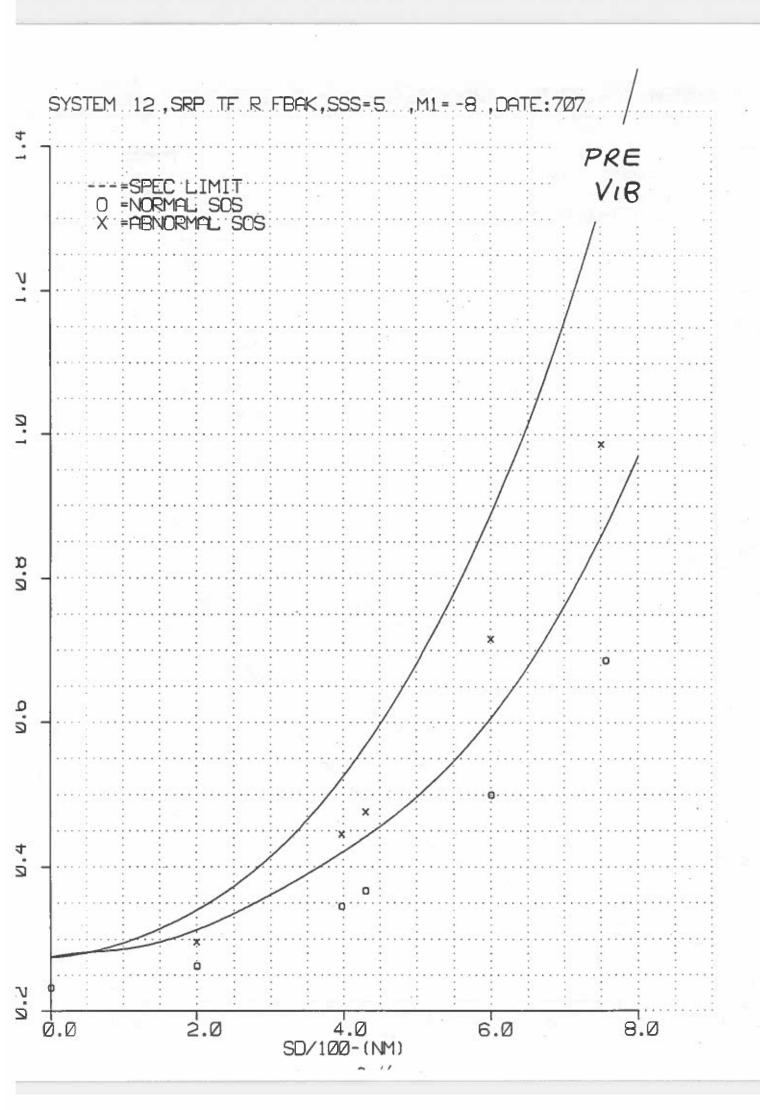
TF SRP Tables Previbration.

TF SRP Curves Postvibration.

TF SRP Tables Postvibration.







## T, COMPLETE, SRP (NM)

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9	FLT. NO. =	12 ENV. = 4	SSS=	5DEGC M1= -8	DEGC DATE:	707
SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB	
LFT	-750.	0.716	0.708	1. 730	1. 665	
MID	<b>−750</b> .	1. 299	0.000	1. 846	1. 793	
RGT	<b>−750</b> .	0. 987	0. 983	1.764	1. 703	
LFT	-600.	0. 544	0. 534	1.715	1. 648	
MID	-600.	1.008	0.000	1. 786	1. 726	
RGT	-600.	0.716	0.708	1. 725	1.660	
LFT	-431.	0. 391	0. 387	1. 459	1. 403	
MID	-431.	0. 621	0.000	1. 489	1. 433	
RGT	<b>-431</b> .	0. 476	0. 469		1. 407	
LFT	-378.	0.378	0. 374	1. 406	1. 351	
MID	-378.	0. 608	0. 602	1. 429	1. 375	
RGT	-398.	0. 445	0.438	1. 410	1. 356	
LFT	<b>-200</b> .	0. 280	0. 278	1. 100	1.057	
MID	-200.	0. 346	0. 341	1. 105	1. 063	
RGT	-200.	0. 296	0. 293	1. 102	1. 059	
LFT	Ο.	0. 253	0. 251		0. 933	
MID	0.	0. 253	0. 251	0. 971	0. 934	
RGT	0.	0. 232	0. 232	0. 968	0. 931	
LFT	200.	0. 322	0. 317		1.061	
MID	200.	0. 321	0. 316	1. 106	1.063	
RGT	200.	0. 263	0. 263	1. 096	1. 054	
LFT	398.	0. 477	0. 469		1. 354	
MID	378.	0. 554	0. 549	1. 419	1. 365	
RGT	378.	0. 345	0. 344	1. 398	1. 344	
LFT	431.	0. 503	0. 495	1. 465	1. 408	
MID	431.	0. 561	0. 000		1. 426	
RGT	431.	0. 367	0. 345	1. 456	1. 399	
LFT	601.	0. 732	0. 724		1. 661	
MID	601.	0. 924	0. 000		1. 704	
RGT		0. 500	0. 492	1. 704	1. 639	
LFT	<b>757</b> .	1. 039	1. 035	1. 780	1. 722	
MID	<b>757</b> .	1. 424	0.000		1.845	
RGT	757.	0. 686	0. 678	1. 717	1. 653	

T. COMPLETE, SRP RATIO

SEG	SUR. DIST. (NM)	TFP		TFB	f-r	TSP	TSB
LFT	<b>-750</b> .	0. 834		0. 825		0. 769	0. 740
MID	<b>-750</b> .	0.000		0.000		0. 821	0.797
RGT	<b>−750</b> .	0. 747		0. 744		0. 784	0. 757
LFT	-600.	0. 897		0.882		0. 871	0. 837
MID	-600.	0.000		0.000		0. 907	0.876
RGT	-600.	0. 805		0.797		0. 876	0.843
LFT	-431.	0.886		0.877		0. 911	0.876
MID	-431.	0.000		0.000		0. 930	0. 895
RGT	-431.	0. 838		0. 826		0. 914	0. 879
LFT	-378.	0. 901		0.891		0. 917	0. 881
MID	-398.	0. 926		0. 917		0. 932	0.897
RGT	-378.	0.851		0. 838		0. 920	0. 884
LFT	-200.	0. 894		0.889		0. 917	0.882
MID	-200.	0. 946		0, 931		0. 922	0.886
RGT	-200.	0. 870	11 - 1	0.862		0. 919	0. 883
LFT	0.	0. 921		0. 914		0. 925	0. 889
MID	0.	0. 873		0.866		0. 925	0. 889
RGT	0.	0. 845		0.844		0. 922	0. 886
LFT	200.	0. 946		0. 932		0. 921	0. 885
MID	200.	0. 877		0.864		0. 922	0. 884
RGT	200.	0. 839		0. 838		0. 914	0. 878
LFT	378.	0. 912		0.898		0. 918	0. 883
MID	378.	0. 847		0. 837		0. 925	0. 890
RGT	378.	0. 823		0.820		0. 912	0. 876
LFT	431.	0.884		0.871		0. 915	0. 879
MID	431.	0.000		0. 000		0. 926	0. 891
RGT	431.	0.831		0. 827		0. 909	0. 874
LFT	601.	0. 821		0. 813		0. 876	0. 843
MID	601.	0.000		0.000		0. 896	0. 865
RGT	601.	0. 823		0.811		0. 865	0. 831
LFT	<b>757</b> .	0. 774		0, 771		0. 788	0. 762
MID	757.	0. 000		0.000		0. 838	0.816
RCT	<b>757</b> .	0. 787		0. 777		0. 760	0. 731

## TF, LEFT, PRIMARY

FLT. NO. = 12	ENV. =	4 SSS= 5DE	GC M1= -8DEGC DATE:	707
SUR. DIST. (NM)	SRP	ACTUAL (NM)	SRP RATIO	
<b>−750</b> .		0. 716	0. 834	
-600.		0. 544	0. 897	
<b>-431</b> .		0. 391	0. 886	
-398.		0. 378	0. 901	
-200.	2.0	0. 280	0. 894	
0.		0. 253	0. 921	
200.		0. 322	0. 946	
398.		0. 477	0. 912	
431.		0. 503	0. 884	
<b>601</b> .		0. 732	0. 821	
<b>757</b> .		1. 039	0. 774	

## TF, LEFT, BACKUP

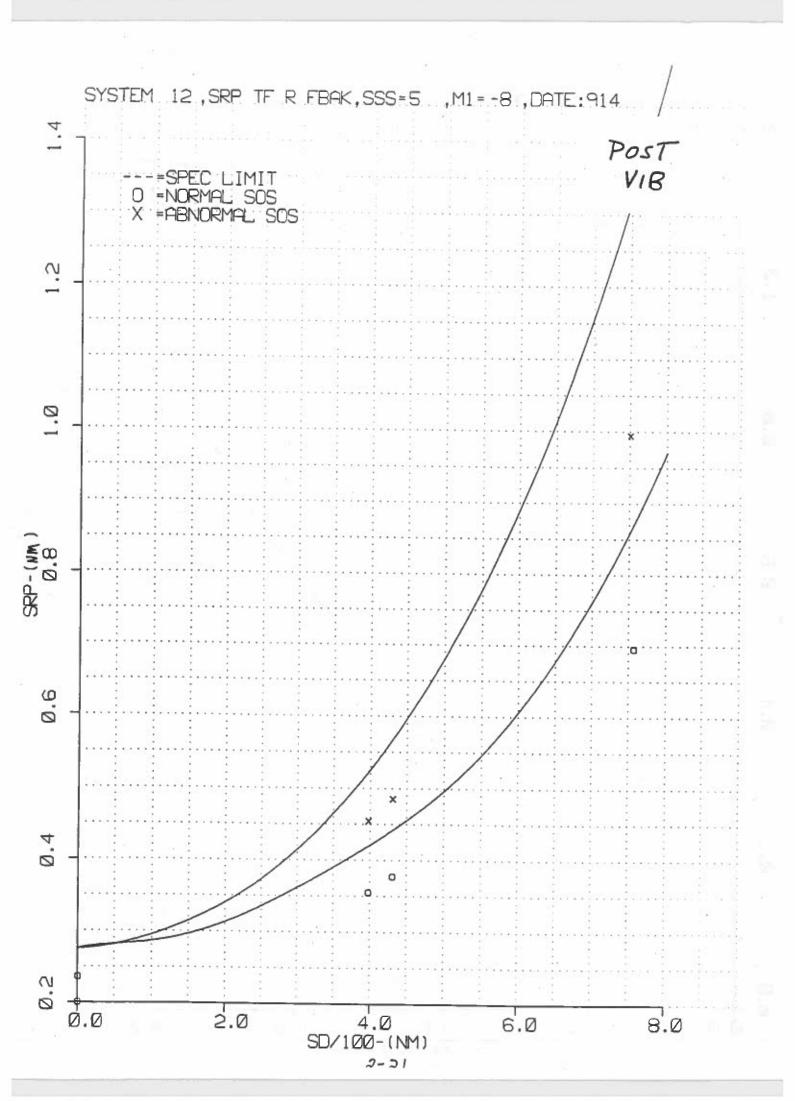
FLT. NO. = 12	ENV. = 4 SSS = 50	EGC M1= -BDEGC	DATE:	707
SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO		
<b>-750</b> .	0. 708	0. 825		
<b>-600</b> .	0. 534	Ö. 882	- +	
-431.	0. 387	0. 877		
<b>−398</b> .	0. 374	0. 891		
-200.	0. 278	0. 889		
0.	0. 251	0. 914		
200.	0. 317	0. 932		
398.	0. 469	0. 878		
431.	0. 495	0. 871		
601.	0. 724	0. 813		
<b>757</b> .	1. 035	0. 771		

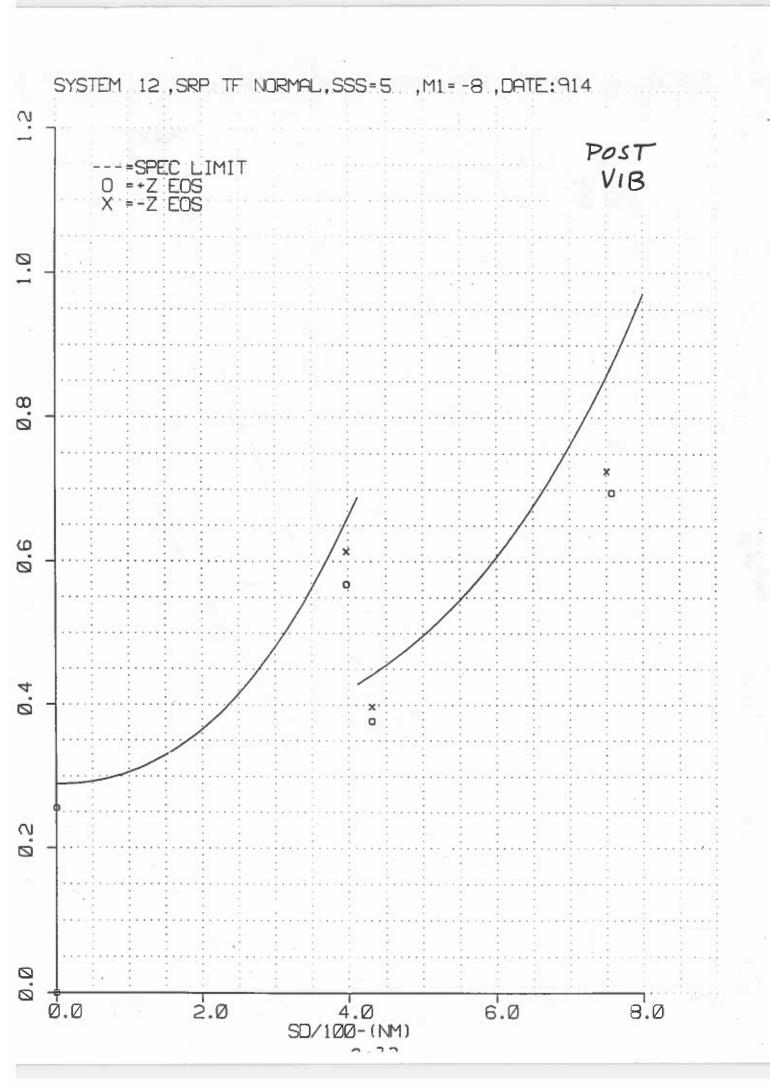
## TF, RIGHT, PRIMARY

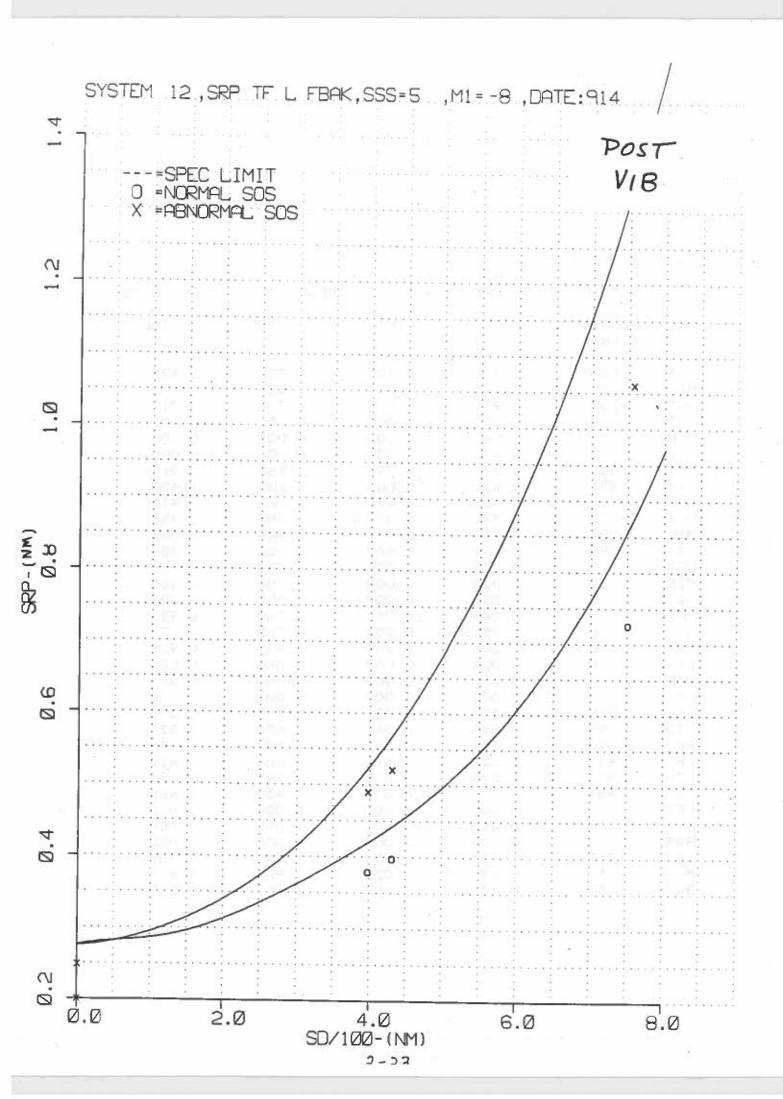
FLT. NO. = 12	ENV. =	4 555=	5DEG0	: M1	= -BDEGC	DATE:	7	'07
SUR. DIST. (NM)	SRP	ACTUAL (N	M)	SRP	RATIO			
<b>-750</b> .		0. 987		0.	747			
-600.		0.716		0.	805			
-431.		0. 476		0.	838			
-398.		0. 445		0.	851			
-200.		0. 296		0.	870			
0.		0. 232		0.	845			
200.		0. 263		0.	837			
398.		0. 345		0.	823			
431.		0.367		0.	831			
601.		0. 500		Ο.	823			
757.		0. 686		0.	787			

## TF RIGHT, BACKUP

FLT. NO. = 12	ENV. = 4 SSS = 5	DEGC M1= -BDEGC	DATE:	707
SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO		
<b>-750</b> .	0. 983	0. 744	5)	
-600.	0. 708	0. 7 <del>9</del> 7		
-431.	0. 469	0. 826		
-398.	0. 438	0. 838		
-200.	0. 293	0.862		
0.	0. 232	O. 844		
200.	0. 263	0. 838		
398.	0. 344	0. 820		
431.	0. 365	0. 827		
601.	0. 492	0.811		
757.	0. 678	0. 777		







## T, COMPLETE, SRP (NM)

	FLT. NO. =	12 ENV. =	4 SSS=	5DEGC M1=	-8DEGC DATE:	914
SEG	SUR. DIST.	TFP	TFB	TSP	TSB	
LFT	<b>-750</b> .	0. 725	0. 722	1. 732	1. 675	
MID	<b>−750</b> .	1. 253	0. 000	1. 828	1. 782	
RGT ·	<b>-750</b> .	0. 995	0. 989	1.766	1,713	
LFT	Ο.	0.000	0.000	0.000	0.000	
MID	0.	0.000	0.000	0.000	0.000	
RGT	0.	0.000	0.000	0.000	0.000	
LFT	-431.	0. 397	0. 391	1. 461	1.410	
MID	-431.	0. 631	0. 000	1.488	1. 439	
RGT	-431.	0. 485	0. 479	1. 461	1.410	
LFT	-398.	0. 379	0. 373	1.408	1. 359	
MID	-398.	0.614	0. 611	1. 429	1. 382	
RGT	-398.	0. 454	0. 447	1. 406	1. 357	
LFT	0.	0.000	0. 000	0. 000	0. 000	
MID	0.	0.000	0. 000	0.000	0.000	
RGT	0.	0. 000	0. 000	0. 000	0. 000	
LFT	0.	0. 248	0. 245	0. 968	0. 935	
MID	Q.	0. 255	0. 252	0. 968	0. 935	
RGT	Ο.	0. 236	Q. 234	0. 966	0. 932	
LFT	0.	0.000	0. 000	0.000	0. 000	
MID	Ο,	0.000	0.000	0.000	0.000	
RGT	Ο.	0. 000	0.000	0.000	0. 000	
LFT	378.	0. 489	0. 484	1. 411	1. 363	
MID	398.	0. 568	0. 565	1. 420	1. 372	
RGT	378.	0. 355	0. 351	1. 397	1. 348	
LFT	431.	0. 521	0. 515	1. 466	1. 415	
MID	431.	0. 573	0.000	1. 479	1. 430	
RGT	431.	0. 377	0. 373	1. 455	1. 405	
LFT	0.	0.000	0.000	0, 000	0.000	
MID	٥.	0.000	0. 000	0. 000	0. 000	
RGT	0.	0.000	0. 000	0. 000	0. 000	
LFT	757.	1.061	1. 054	1. 785	1. 735	
MID	757.	1. 449	0. 000	1. 900	1.860	
RGT	757.	0. 695	0. 692	1.715	1. 658	

T, COMPLETE, SRP RATIO

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	<b>75</b> 0.	0.845	0.841	0. 770	0. 745
MID	<b>−75</b> 0.	0.000	0.000	0.813	0. 793
RGT	<b>−750</b> .	0. 753	0. 749	0. 785	0. 762
LFT	<b>O</b> .	0. 000	0.000	0. 000	0.000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT LFT	0. -431	0. 000	0. 000	0. 000	0. 000
MID	-431. -431.	0. 900 0. 000	0.885	0.912	0.881
RGT	-431. -431.	0. 854	0. 000 0. 843	0. 929 0. 912	0. 899
LFT	-398.	0. 903	0. 888	0. 918	0. 881 0. 884
MID	-398.	0. 734	0. 930	0. 932	0.901
RGT	-378.	0. 869	0. 855	0. 716	0. 885
LFT	0.	0. 000	0.000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0.000	0.000	0. 000
LFT	0.	0. 902	0. 892	0. 722	0. 890
MID	0.	0. 882	0.871	0. 922	0. 890
RGT	0.	0. 857	0. 852	0. 920	0.888
LFT	O.	0. 000	0.000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0.000	0. 000	0. 000	0. 000
LFT MID	398.	0. 936	0. 925	0. 920	0. 889
RGT	378. 378.	0. 845 0. 846	0.860	0. 926	0. 894
LFT	431.	0. 916	0. 838 0. 906	0. 911 0. 915	0. 879 0. 884
MID	431.	0. 000	0. 000	0. 713	0. 893
RGT	431.	0. 854	O. 844	0. 709	0. 877
LFT	0.	0.000	0.000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0.000	0. 000	0. 000	0. 000
LFT	757.	0. 790	0. 785	0. 790	0. 768
MID	757.	0.000	0.000	0. 840	0. 823
RGT	757.	0. 797	0. 793	0. 759	0. 734

## TF, LEFT, PRIMARY

FLT. NO. = 12 ENV. =	4 SSS= 5DEGC M1=	-BDEGC DATE: 914
SUR. DIST. (NM) SR	P ACTUAL (NM) SRP RA	TIO
<b>-750</b> .	0. 725 0. 84	5
0.	0.000 0.00	0
-431.	0. 397 0. 90	0
<b>-378</b> .	0. 379 0. 90	3
0.	0. 000 0. 00	0
0.	0. 248 0. 90	2
0.	0.000 0.00	O
398.	0. 489 0. 93	6
431.	0. 521 0. 91	6
0.	0.000	0
<b>757</b> .	1. 061 0. 79	0

# TF, LEFT, BACKUP

8

FLT. NO. = 12	ENV. =	4 SSS=	5DEGC M1= -8DEGC	DATE:	914
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
<b>-750</b> .		0. 722	0. 841		
0.		0. 000	0. 000		
-431.		0. 391	0. 885		
-378.		0. 373	0.888		
0.		0. 000	0. 000		
0.		0. 245	0. 892		
0.		0.000	0. 000		
398.		0. 484	0. 925		
431.		0. 515	0. 906		
0.		0.000	0. 000		
757.		1, 054	0. 785		

# TF, RIGHT, PRIMARY

FE1. NO. = 12	ENV. =	4 555=	5DEGC	M1= -8DEGC	DATE:	914
SUR. DIST. (NM)	SRP	ACTUAL (N	M) Si	RP RATIO		
-7 <b>5</b> 0.		0. 995 0. 000		0. 753 0. 000		
-431.		0. 485		0. 854		
-378. 0.		0. 454 0. 000		0. 869 0. 000		
0.		0. 236		0. 857		
0.		0. 000		0. 000		
378.		0. 355		0. 846		
431.		0. 377		0. 854		
0.		0.000		0. 000		
<b>757</b> .		0. 695		0. 797		

#### TF RIGHT, BACKUP

FLT. NO. = 12	ENV. =	4 SSS= 5D	EGC M1= -8DEGC	DATE:	914
SUR. DIST. (NM)	SRP	ACTUAL (NM)	SRP RATIO		
<b>−750</b> .		0. 989	0. 749		
- O.		0.000	0. 000		
-431.		0. 479	0. 843		
-398.		0. 447	0. 855		
Ο.		0.000	0. 000		
0.		0. 234	0. 852		
© O.		0.000	0. 000		
398.		0. 351	0. 838		
431.		0. 373	0.844		
Ο.		0. 000	0. 000		
フラフ		A 407	A 700		

- 2.2 Geometric Resolution (Cont'd)
- 2.2.1 Fine Geometric Resolution, Infrared (Cont'd) (3.2.1.1.2.1)

## 2.2.1.3 Acceptance - Thermal Vacuum

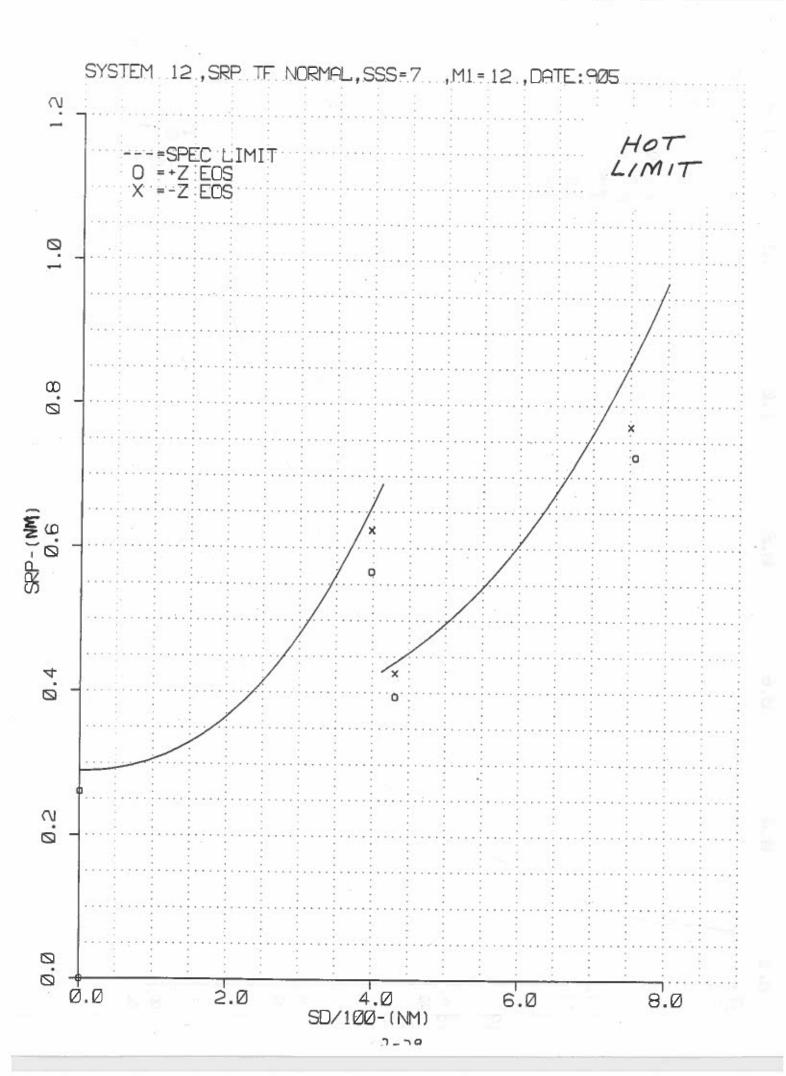
The attached TF SRP curves and tables demonstrate in-spec performance at the thermal vacuum test limits. The Orbit Nominal SRP curves are contained in paragraph 2.2.1.1 and are not included here.

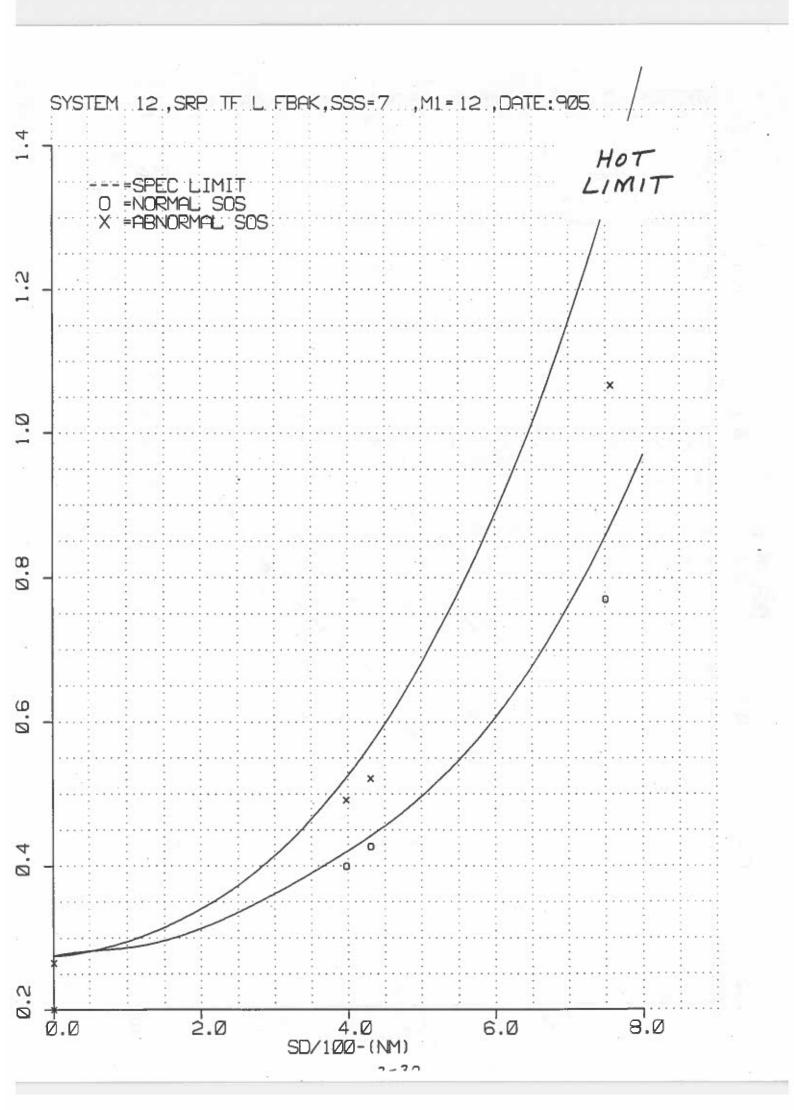
ATTACHMENTS: TF SRP Curves Hot Limits

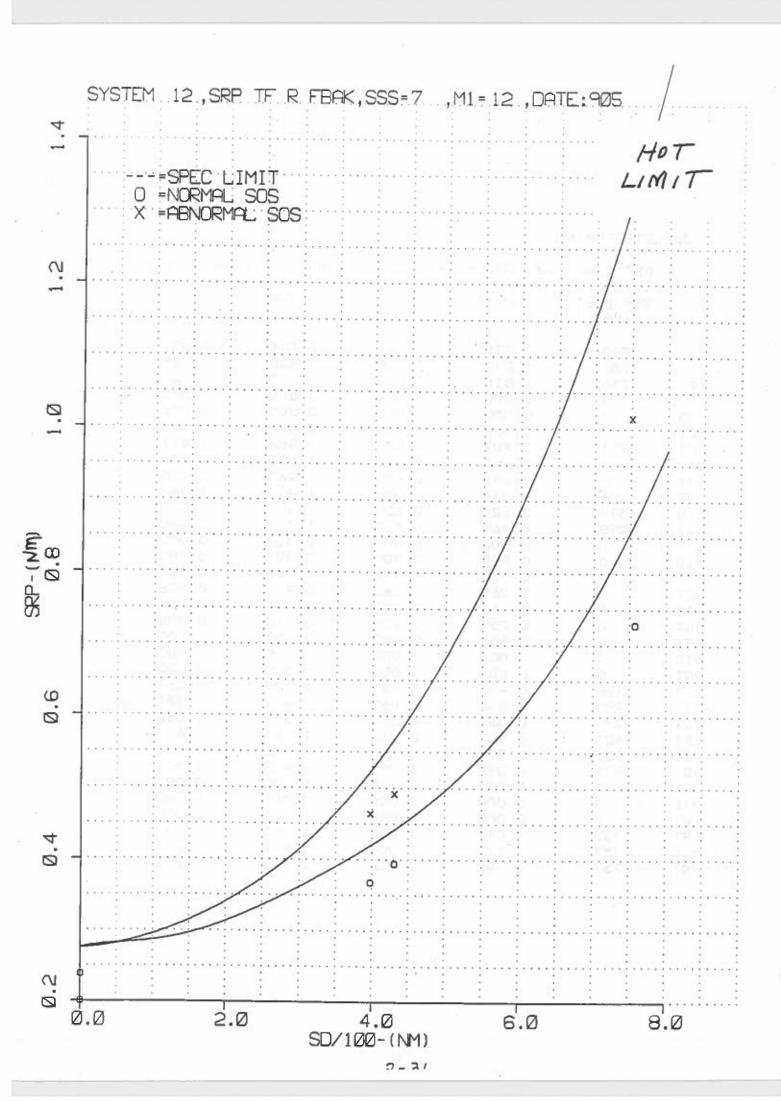
TF SRP Tables Hot Limits

TF SRP Curves Cold Limits

TF SRP Tables Cold Limits







## T, COMPLETE, SRP (NM)

	EL T. NO	12 ENV. =	4 SSS=	7DEGC M1=	12DEGC DATE:	905
	FLT. NO. =	IE ENV	7 355			
SEG	SUR. DIST.	TFP	TFB	TSP	TSB	
	(NM)				,	
	750	0. 771	0. 767	1. 751	1. 697	
LFT	-750.	1. 252	0. 000	1. 833	1. 789	
MID	<b>-750</b> .	1.015	1.016	1. 777	1.727	
RGT	<b>−750</b> .	0. 000	0. 000	0. 000	0.000	
LFT	0.	0. 000	0. 000	0.000	0.000	
MID	0.	0. 000	0. 000	0. 000	0. 000	
RGT	0.	0. 427	0.424	1. 466	1. 417	
LFT	-431.	0. 427	0. 000	1.493	1.446	
MID	-431.	0. 491	0. 489		1.420	
RCT	-431.	0. 399	0. 396	1. 413	1.366	
LFT	-378.	0. 624	0. 622	1.433	1.388	
MID	-398. -398.	0. 464	0. 462	1. 413	1.365	
RGT	-376. O.	0. 000	0. 000	0. 000	0.000	
LFT	0.	0.000	0. 000	0. 000	0.000	
MID RGT	0.	0.000	0.000	0.000	0, 000	
LFT	0.	Q. 265	0. 263	0. 971	0. 938	
MID	0.	0. 260	0. 258	0. 971	0. 938	
RGT	0.	0. 238	0. 236	0. 968	0. 935	
LFT	0.	0.000	0.000	0.000	0.000	
MID	0.	0.000	0. 000	0. 000	0.000	
RGT	o.	0.000	0.000	0.000	0.000	
LFT	398.	0. 491	0. 489	1. 409	1.362	
MID	398.	0. 567	0. 563	1. 421	1. 374	
RGT	398.	0. 367	0. 363	1, 399	1. 352	
LFT	431.	0. 521	0.518	1. 468	1. 419	
MID	431.	0. 579	0.000	1. 483	1. 435	
RGT	431.	0. 394	0. 390	1. 458	1. 409	
LFT	0.	0.000	0. 000	0.000	0. 000	
MID	<b>.</b> 0.	0.000	0.000	0. 000	0.000	
RGT	0.	0.000	0.000	0.000	0.000	
LFT	757.	1.067	1.069		1. 744	
MID	757.	1. 475	0. 000	1. 916	1. 876	
RGT	757.	0. 728	0. 724	1. 728	1. 673	

## T, COMPLETE, SRP RATIO

SEG	SUR. DIST. (NM)	TFP	TFB	TSP
LFT	<b>−750</b> .	0.897	0. 873	0. 779 0. 755
MID	<b>−750</b> .	0.000	0. 000	0. 815 0. 795
RGT	<b>−750</b> .	0. 768	0. 769	0. 790 0. 768
LFT	Ο.	0.000	0.000	0. 000 0. 000
MID	Ο.	0.000	0.000	0. 000 0. 000
RGT	0.	0.000	0.000	0. 000 0. 000
LFT	<b>-431</b> .	0. 966	0. 960	0. 916 0. 885
MID	-431.	0. 000	0.000	0. 932 0. 903
RGT	<b>-431</b> .	0. 865	0.861	0. 918 0. 887
LFT	-398.	0. 952	0. 944	0. 922 0. 890
MID	-398.	0. 950	0. 946	0. 934 0. 905
RGT	-398.	0.887	0.883	0. 921 0. 890
LFT	0.	0. 000	0.000	0. 000 0. 000
MID	0.	0. 000	0. 000	0. 000 0. 000
RGT	Ο.	0. 000	0. 000	0. 000 0. 000
LFT	0.	0. 945	0. 956	0. 924 0. 893
MID	0.	0. 899	0. 890	0. 925 0. 894
RGT	Ο.	0. 865	0. 858	0. 922 0. 891
LFT	Ο.	0.000	0. 000	0.000 0.000
MID	0.	0.000	0.000	0.000 0.000
RGT	0.	0. 000	0. 000	0.000 0.000
LFT	378.	0. 940	0. 935	0. 919 0. 888
MID	378.	0. 863	0. 858	0. 927 0. 896
RGT	398.	0. 876	0. 866	0.912 0.882
LFT	431.	0. 917	0. 911	0. 917 0. 886
MID	431.	0. 000	0. 000	0. 926 0. 896
RGT	431.	0.892	0.883	0. 910 0. 880
LFT	0.	0. 000	0. 000	0.000 0.000
MID	0.	0. 000	0. 000	0.000 0.000
RGT	0.	0.000	0. 000	0.000 0.000
LFT	757.	0. 795	0. 796	0. 793 0. 772
MID	757.	0.000	0.000	0. 848 0. 830
RGT	<b>757</b> .	0. 835	0. 830	0. 764 0. 740

#### "F. LEFT, PRIMARY

FLT. NO. = 12	ENV. =	SSS=	7DEGC	M1= 12DEGC	DATE:	905
SUR. DIST. (NM)	SRP A	ACTUAL (N	lM) S	RP RATIO		
<b>−75</b> 0.		0. 771		0. 897		
<b>Q</b> .	(	0. 000		0. 000		
<b>-431</b> .	(	). 427		0. 966		
-398.	(	). 399		0. 952		
0.	(	0.000		0. 000		
Ο.		). 265		0. 965		
0.		0.000		0. 000		
398.		0. 491		0. 940		
431.		D. 521		0. 917		
0.	(	0. 000		0. 000		
757.		1.067		0. 795		

#### TF, LEFT, BACKUP

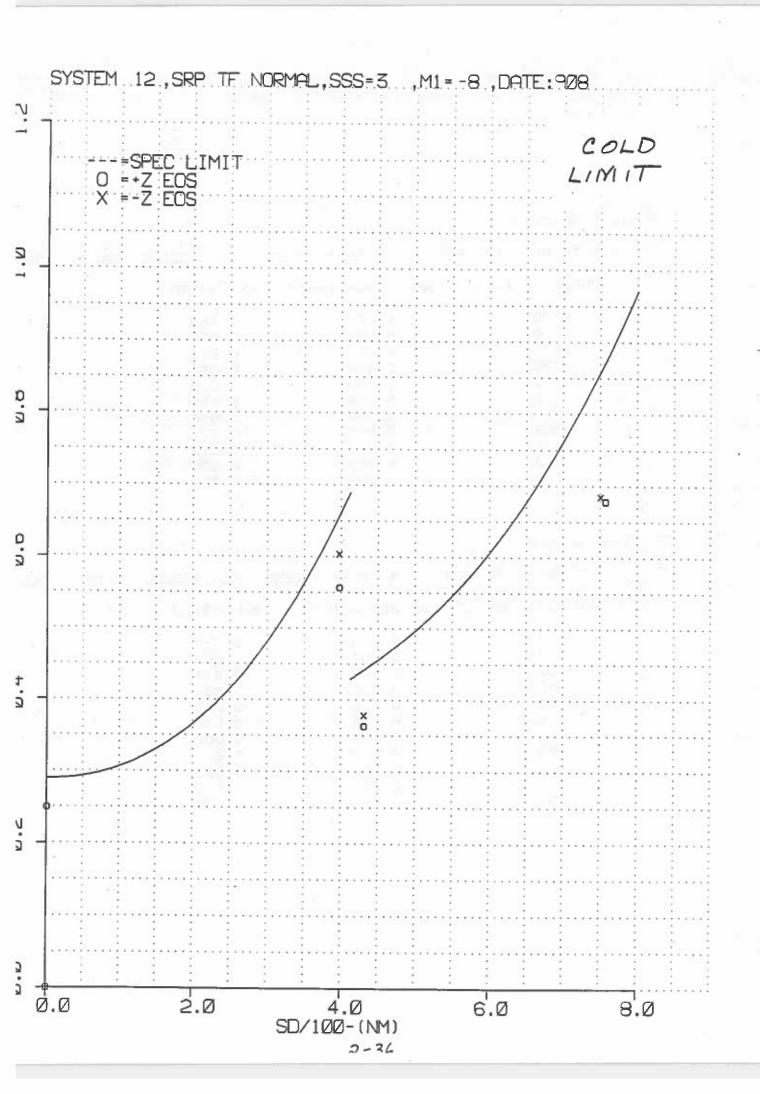
FL.	1. NO. = 12	ENV. =	4 555=	/DEGC M1= 12DE	GC DATE:	905
SUF	R. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
	-750.		0. 767	0. 893		
	Ο.		0.000	0. 000		
	-431.		0. 424	0. 960		- 0
	-398.		0. 396	0. 944		
	0.		0. 000	0. 000	54	
	Ο.		0. 263	0. 956		
	0.		0.000	0.000		
	398.		0. 489	0. 935		
	431.		0. 518	0. 911		
	Ο.		0.000	0. 000		
	<b>757</b> .		1.069	0. 796		

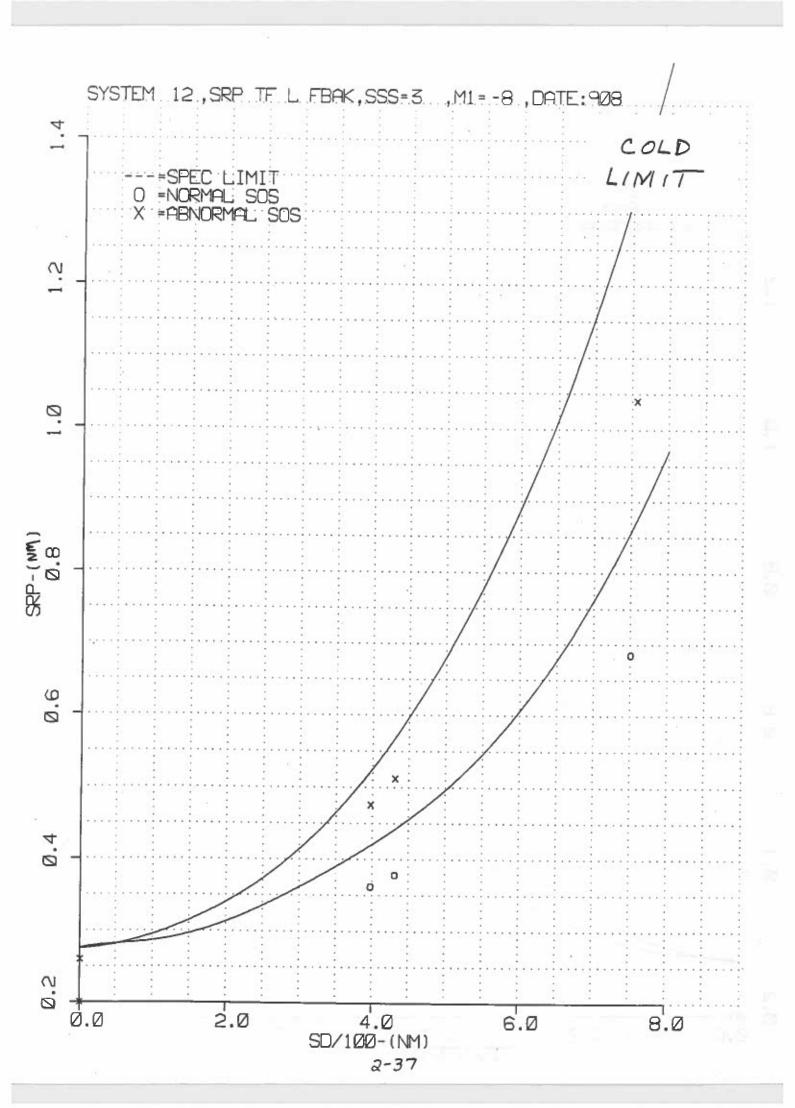
# TF, RIGHT, PRIMARY

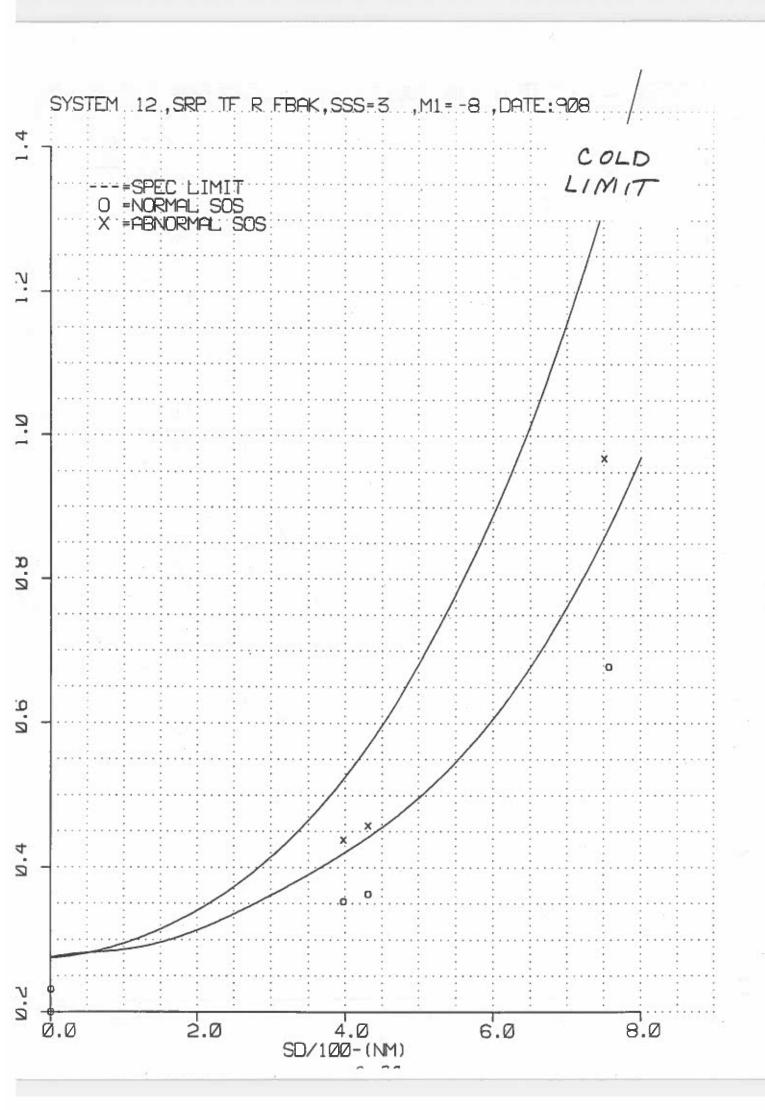
ENV. =	4 555=	7DEGC M1	= 12DEGC	DATE:	905
SRP	ACTUAL (N	M) SRP	RATIO		
	1.015	0.	768		
	0.000	0.	000		
	0. 491	0.	365		
	0. 464				
	0.000			*	
	0. 238	0.	365		
	0. 000	0.	000		
	0. 367	0. (	376		
	0. 394	0. (	392		
	0. 000	0. (	000		
	0. 728				
		SRP ACTUAL(N 1.015 0.000 0.491 0.464 0.000 0.238 0.000 0.367 0.394 0.000	SRP ACTUAL (NM) SRP (1. 015 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0. 0.	SRP ACTUAL (NM) SRP RATIO  1. 015	SRP ACTUAL (NM) SRP RATIO  1. 015

# TF RIGHT, BACKUP

FLT. NO. = 12	ENV. =	4 SSS=	7DEGC	M1=	12DEGC	DATE:	905
SUR. DIST. (NM)	SRP	ACTUAL (N	M) S	RP RA	TIO		
<b>-75</b> 0.		1. 016		0. 76	9		
Ο.		0.000		0.00			
<b>-431</b> .		0. 489		0. 86			
-398.		0. 462		0.88			
0.		0.000		0. 00			
0.		0. 236		0. 85			
0.		0. 000		0. 00			
<b>39</b> 8.		0. 363		0.86			
431.		0. 390		0. 88			
Ο.		0. 000		0. 00			
757.		0.724		0.00	_		







# T, COMPLETE, SRP (NM)

34	FLT. NO. =	12 ENV. =	4 SSS=	3DEGC M1=	-BDEGC DATE:	908
SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB	
LFT	-750.	0 (05		=	Auto-	
MID	-750.	0. 685	0. 678	1.713	1. 645	
RGT	-750.	1. 248	0. 000	1.818	1. 764	
LFT		0. 969	0. 962	1. 752	1. 688	
MID	0.	0. 000	0. 000	0. 000	0. 000	
	0.	0. 000	0.000	0. 000	0. 000	
RGT	0.	0. 000	0. 000	0. 000	0.000	
LFT MID	-431. -431.	0. 378	0. 375	1. 453	1. 395	
RGT	-431. -431.	0. 614	0. 000	1. 480	1. 423	
LFT	-431. -398.	0. 458	0. 455	1. 455	1. 397	
MID	-376. -378.	0. 342	0. 358	1. 398	1. 342	
RGT	-378. -398.	0. 604	0. 597	1. 424	1. 369	
LFT		0. 438	0. 436	1. 407	1. 350	
MID	0.	0. 000	0. 000	0. 000	0. 000	
RGT	0.	0. 000	0. 000	0. 000	0. 000	
LFT	0.	0.000	0. 000	0. 000	0. 000	
MID	0.	0. 259	0. 257	0. 973	0. 934	
RGT	0.	0. 249	0. 247	0. 968	0. 929	
LFT	0.	0. 231	0. 230	0. 965	0. 926	
MID	0.	0. 000	0. 000	0. 000	0. 000	
RGT	0.	0. 000	0. 000	0. 000	0. 000	
LFT	0. 3 <del>9</del> 8.	0. 000	0.000	0. 000	0. 000	
MID	378.	0. 476	0. 472	1. 406	1. 350	
RGT	378. 3 <b>9</b> 8.	0. 556	0. 551	1. 415	1. 359	
LFT	431.	0. 353	0. 349	1. 394	1. 339	
MID	431.	0. 512	0. 508	1. 461	1. 403	
RGT	431.	0. 567	0. 000	1. 477	1. 419	
LFT		0. 363	0. 359	1. 451	1. 393	
MID	0.	0. 000	0. 000	0. 000	0. 000	
RGT	0.	0. 000	0. 000	0. 000	0. 000	
LFT	0. 757	0. 000	0. 000	0. 000	0. 000	
MID	757.	1. 040	1.032	1. 775	1. 715	
RGT	757.	1. 443	0. 000	1. 893	1.845	
reg I	<b>757</b> .	0. 678	0. 671	1. 706	1. 639	

T, COMPLETE, SRP RATIO

SEG	SUR. DIST.	TFP	TFB	TSP	TSB
LFT	<b>-750</b> .	0. 798	0. 790	0.762	0. 731
MID	<b>−750</b> .	0.000	0.000	0. 809	0.784
RGT	<b>−750</b> .	0. 734	0. 728	0. 779	0. 751
LFT	0.	0. 000	0. 000	0.000	0. 000
MID	0.	0. 000	0.000	0. 000	0. 000
RGT	0.	0.000	0. 000	0. 000	0. 000
LFT	-431.	0. 857	0. 848	0. 907	0. 871
MID	-431.	0. 000	0.000	0. 925	0. 889
RGT	-431.	0. 806	0. 801	0. 909	0. 872
LFT	-398.	0. 862	0. 854	0. 912	0. 875
MID	-398.	0. 918	0. 909	0. 928	0. 892
RGT	-398.	0. 838	0. 833	0. 917	0. 880
LFT	0.	0. 000	0. 000	0. 000	0.000
MID	0.	0. 000	0.000	0.000	0. 000
RGT	0.	0.000	0.000	0000	0. 000
LFT	0.	0. 943	0. 934	0. 926	0. 889
MID	0.	0.861	0. 852	0. 922	0. 885
RGT	0.	0.841	0. 835	0. 919	0. 882
LFT	0.	0.000	0.000	0. 000	0.000
MID	0.	0.000	0.000	0.000	0. 000
RGT	0.	0.000	0. 000	0.000	0. 000
LFT	398.	0. 910	0. 903	0. 917	0. 880
MID	398.	0. 847	0. 839	0. 923	0.886
RGT	378.	0. 841	0. 833	0. 909	0. 873
LFT	431.	0. 901	0. 894	0. 912	0. 876
MID	431.	0.000	0. 000	0. 922	0. 884
RGT	431.	0. 822	0.813	0. 906	0.869
LFT	0.	0.000	0.000	0. 000	0. 000
MID	0.	0. 000	0.000	0. 000	0. 000
RGT	0.	0.000	0. 000	0. 000	0. 000
LFT	757.	0.774	0. 768	0. 785	0. 759
MID	757.	0.000	0. 000	0. 837	0.816
RGT	757.	0. 778	0. 769	0. 755	0. 725

# TF, LEFT, PRIMARY

FLT. NO. = 12	ENV. =	4 SSS=	3DEGC	M1= -BDEGC	DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (N	M) S	RP RATIO	£	
<b>−75</b> 0.	(	D. 685		0. 798		
0.	(	0. 000		0. 000		
<b>-431</b> .	(	). 378		0. 857		
-398.	(	0. 362		0. 862		
0.	(	0. 000		0. 000		
0.	(	). 259		0. 943		
Ο.	(	0. 000		0. 000		
398.	(	). 476		0. 910		
431.	(	). 512		0. 901		
0.	(	0.000		0. 000		
<b>757</b> .	1	. 040		0. 774		

# TF, LEFT, BACKUP

FLT. NO. = 12	ENV. =	4 SSS=	3DEGC	M1= -BDEGC	DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (N	M) 5	SRP RATIO		
<b>−750</b> .		0. 678		0. 790		
0.		0.000		0. 000	72	
<b>-431</b> .		0. 375		0. 848		
-378.		0. 358		0. 854		
0.		0.000		0. 000		
Ο.		0. 257		0. 934		
0.		0. 000		0. 000		
398.		0. 472		0. 903		0.0
431.		0. 508		0. 894		
0.		0. 000		0. 000		
<b>757</b> .		1. 032		0. 768		

### TF, RIGHT, PRIMARY

FLT. NO. = 12	ENV. =	4 SSS= 3DE	EGC M1= -BDEGC	DATE: 908	
SUR. DIST. (NM)	SRP	ACTUAL(NM)	SRP RATIO		
<b>−75</b> 0.		0. 969	0. 734		
0.		0. 000	0. 000		
-431.		0. 458	0. 804		
-378.		0. 438	0. 838		
0.		0. 000	0. 000		
0.		0. 231	0. 841		
0.	4	0. 000	0. 000		
398.		0. 353	O. 841		
431.		0. 343	0. 822		
0.		0. 000	0. 000		
<b>757.</b>		0. 67B	O. 77B		

### TF RIGHT, BACKUP

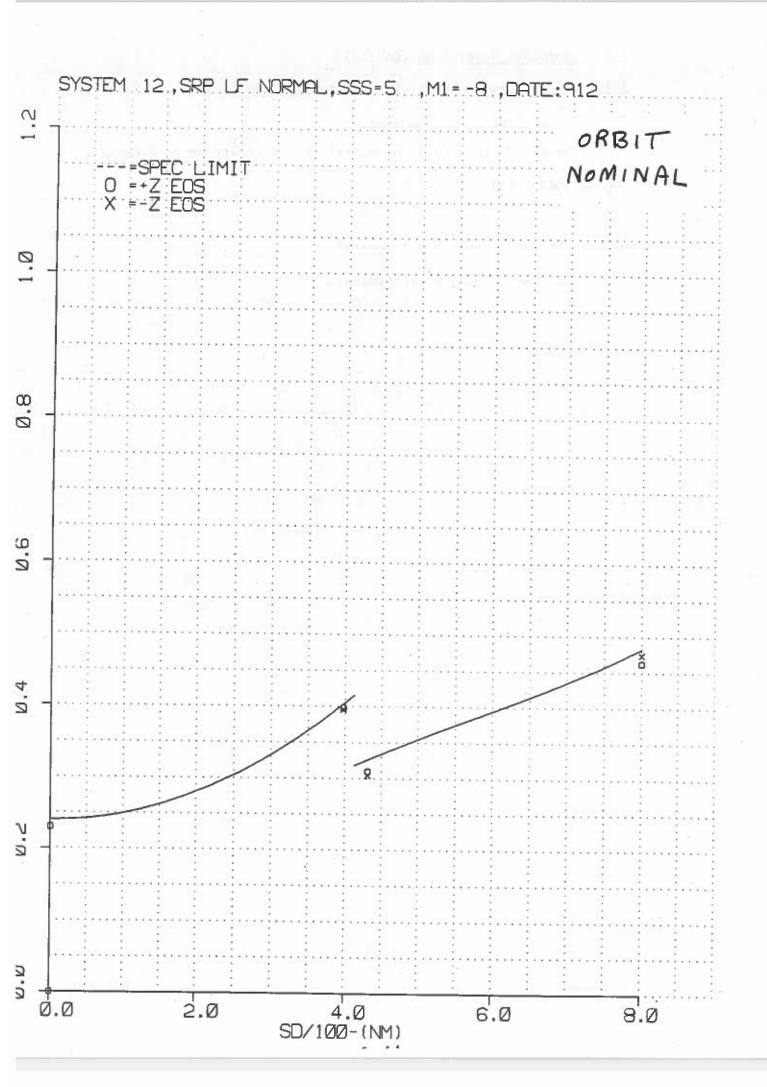
FLT. NO. = 12	ENV. =	4 SSS=	3DEGC	M1= -BDEGC	DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (N	IM) S	RP RATIO		
-750.		0. 962		0. 728		
0.		0. 000		0.000		
-431.		0. 455		0. 801		
-398.		0. 436		0. 833		
0.		0. 000		0.000		
0.		0. 230		0. 835		
0.		0. 000		0. 000		
398.		0. 349		0. 833		
431.		0. 359		0. 813		
0.		0. 000		0. 000		
<b>757</b> .		0. 671		0. 769		

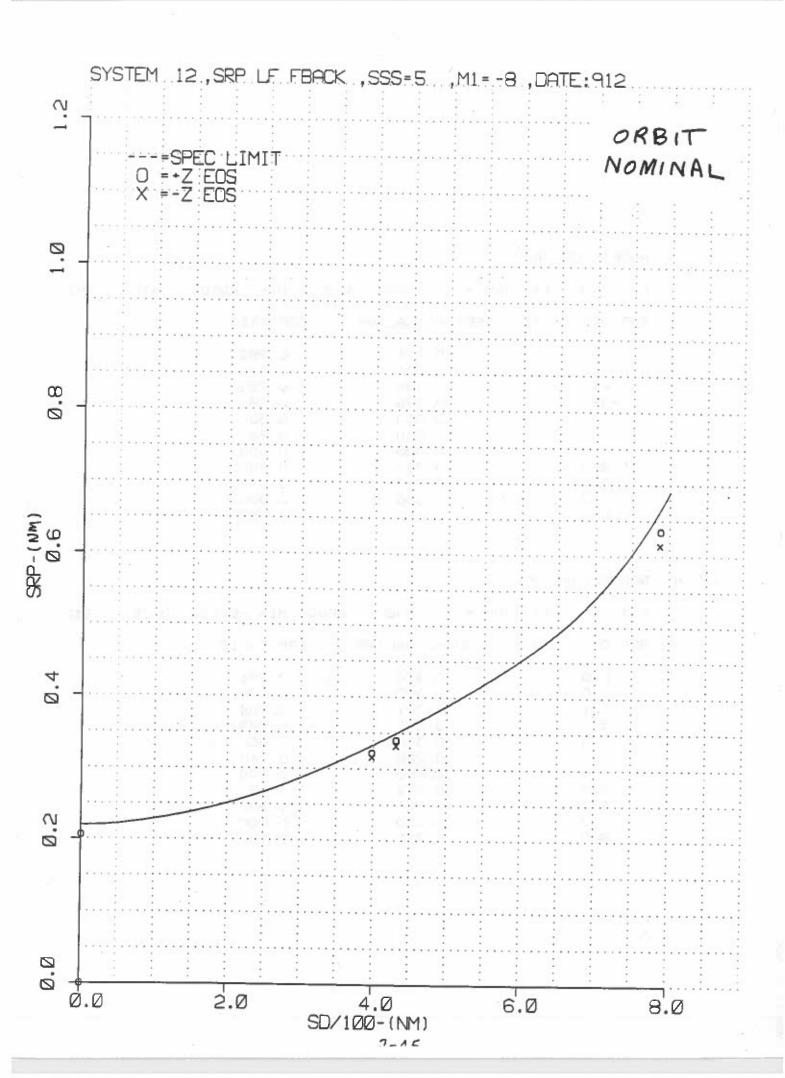
- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.2 Fine Geometric Resolution Daytime Visual (3.2.1.1.2.1)
- 2.2.2.1 <u>Baseline (Orbit Nominal)</u>

The LF SRP is within the specification limits in both Primary and Redundant configurations.

ATTACHMENTS: LF SRP Curves Orbit Nominal

LF SRP Tables Orbit Nominal





FLT. NO. = 12	ENV. =	4 555=	5DEGC M	1= -8DEGC	DATE:	912
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP	RATIO	8	
-800.		0. 474	0.	. 982		
Q.		0.000	0.	. 000		
-431.		0. 304	0.	. 933		
-398.		0. 396	0.	. 980	3:	
0.		0.000	0.	. 000		0 2
Ο.		0. 230	0.	. 957		
0.		0.000	0.	. 000		
398.		0. 398	0.	. 987		
431.		0. 310	0.	949		
0.		0. 000	0.	. 000		
800.		0. 462	0.	. 958		

FET. NO 12	EIVV. —	4 333-	SDEGC	MI= -ODEG	C DAII	=:	415
SUR. DIST. (NM)	SRP	ACTUAL (N	M) S	RP RATIO			
<b>-800</b> .		0. 474	4	0. 981			
0.		0.000		0. 000			
-431.		0. 301		0. 924			
-398.		0. 395		0. 978			
0.		0. 000		0. 000			
0.		0. 228		0. 949			
0.		0.000		0. 000		*30	
398.		0. 397		0. 985			
431.		0. 306		0. 939			
0.		0. 000		0. 000			
B00.		0. 462		0. 956			

# LF, DAY, FALLBACK, PRIMARY

FLT. NO. = 12	ENV. =	4 555=	5DEGC	M1= -8DE(	GC DATE:	912
SUR. DIST. (NM)	SRP	ACTUAL (N	M) 9	RP RATIO		
<b>-787</b> .		0. 615		0. 926		
0.		0. 000		0. 000		
<b>-431</b> .		0. 332		0. 950		30
-398.		0. 316		0. 951		
0.		0. 000	2.0	0.000		
0.		0. 205		0. 938		
0.		0. 000		0. 000		
398.		0. 322		0. 969		
431.		0. 340		0. 972		
0.		0. 000		0. 000		
700		0.635		0 954		

### LF, DAY, FALLBACK, BACKUP

FL1. NU. = 12	ENV. = 4	555=	SDEGC	MI= -ADEGC	DATE:	912
SUR. DIST. (NM)	SRP A	CTUAL (N	M) S	RP RATIO		
<b>-787</b> .	0	. 614		0. 925		
0.	0	. 000		0. 000		
-431.	0	. 329		0. 941		
-398.	0	. 313		0. 942		
0.	. 0	. 000		0. 000		
0.	0	. 203		0. 928		
0.	0	. 000		0. 000		
398.	0	. 319		0. 959		
431.	0	. 337		0. 963		
0.	0	. 000		0. 000		
788.	0	. 634		0. 953		

- 2.2 Geometric Resolution (Cont'd)
- 2.2.2 <u>Fine Geometric Resolution Daytime Visual</u> (Cont'd)
  (3.2.1.1.2.1)

### 2.2.2.2 Acceptance - Vibration

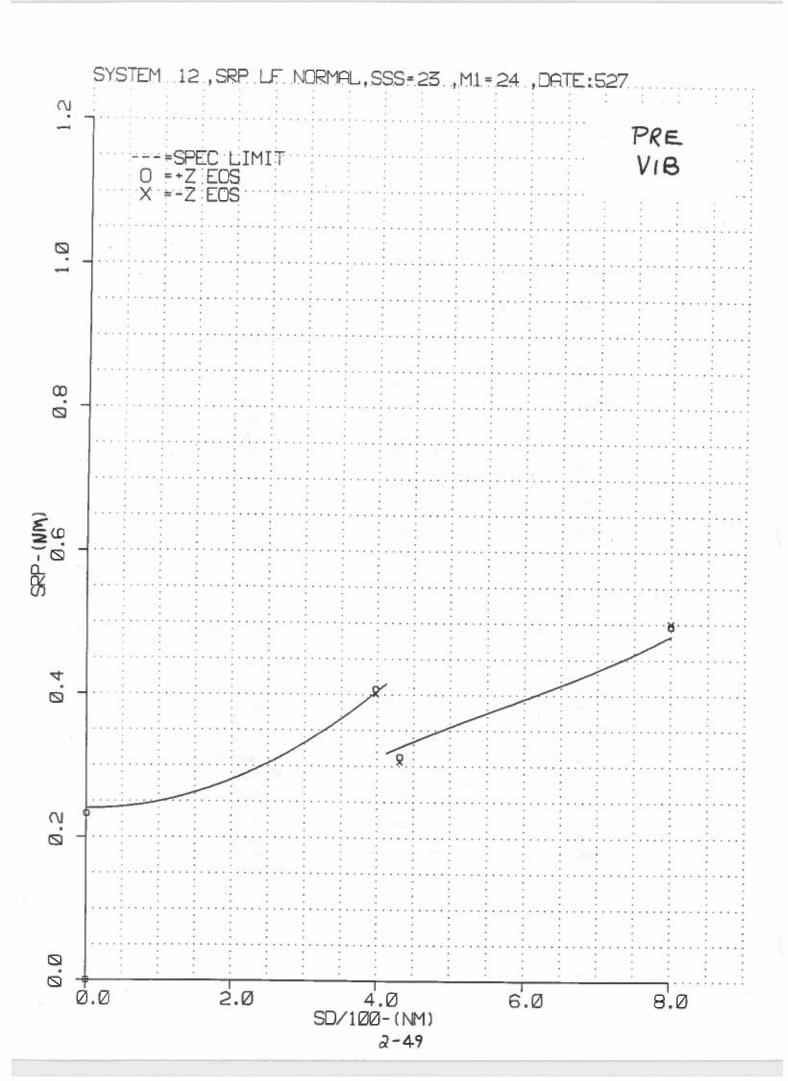
OLS #12 underwent Acceptance-level SSS vibration on May 23, 1991. The Pre-to-Post Vibration SRP performance is within Specification requirements and is shown on the attached curves and tables. No changes in SRP performance occurred as a result of vibration.

ATTACHMENTS: LF SRP Curves Pre-Vibration

LF SRP Tables Pre-Vibration

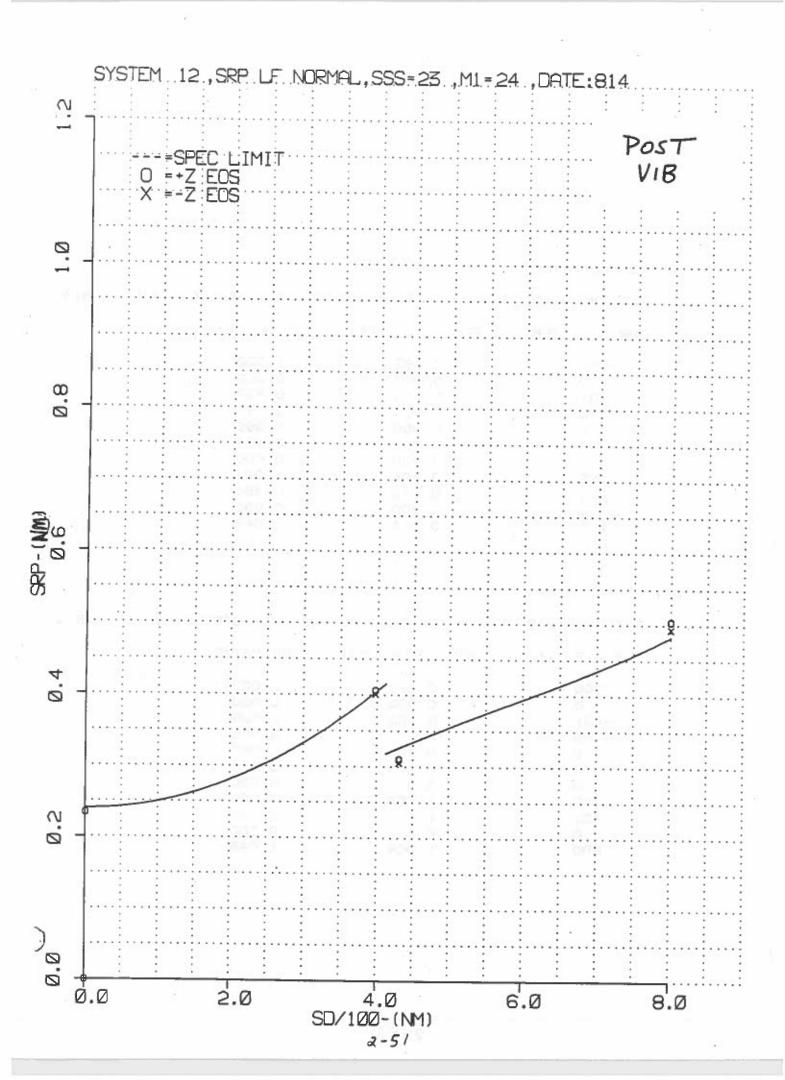
LF SRP Curves Post-Vibration

LF SRP Tables Post-Vibration



FLT. NO. = 12	ENV. =	2 SSS= 23DEGC	M1= 24DEGC	DATE: 527	P
SUR. DIST. (NM)	SRP	ACTUAL (NM)	SRP RATIO		
-B00.		0. 500	1. 035		
0.		0.000	0. 000		
<b>-431</b> .		0. 306	0. 939		
-398.		0. 401	0. 994		
0.		0.000	0. 000		
0.		0. 232	0. 968		
0.		0. 000	0. 000		
398.		0.408	1. 011		
431.		0. 313	O. 958		
0.		0. 000	0. 000		
800.	97	0. 496	1. 028		

FLT. NO. = 12	ENV. =	2 SSS= 23DEG	C M1= 24DEGC	DATE: 527	
SUR. DIST. (NM)	SRP	ACTUAL(NM)	SRP RATIO		
<b>-800</b> .		0. 499	1. 034		
0.		0. 000	0. 000		
-431.		0. 303	0. 928	* *	
-398.		0. 401	0. 993		
0.		0.000	0. 000		
0.		0. 230	0. 960		
Ο.		0.000	0. 000		
398.		0. 407	1.010		
431.		0. 309	0. 947		
0.		0. 000	0. 000		
800		0 484	1 027		



FLT. NO. = 12	ENV. =	2 SSS= 23DE	GC M1= 24DEGC	DATE:	814
SUR. DIST. (NM)	SRP	ACTUAL(NM)	SRP RATIO	6	
<b>-800</b> .		0. 493	1.022		
0.	е	0. 000	0. 000		
-431.		0. 305	0. 937		
-398.		0. 401	0. 993		
0.		0. 000	0.000		
0.		0. 234	0. 974		
0.		0. 000	0.000		
398.		0. 406	1.007		
431.		0. 310	0. 950		
0.		0. 000	0.000		
800.		0. 504	1.044		

FLT. NO. = 12 ENV. =	2 SSS= 23DEGC M1= 24DEGC	DATE:	814
SUR. DIST. (NM) SRF	ACTUAL(NM) SRP RATIO		
-800.	0. 494		
0.	0.000 0.000		
-431.	0. 303 0. 928		
-398.	0. 401 0. 995		
0.	0. 000 0. 000		
0.	0. 233 0. 970		
0.	0, 000 0, 000		
398.	0. 406 1. 008		
431.	0. 307 0. 940		J.
0.	0. 000 0. 000		-157
800	0 504 1, 045		

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.2 <u>Fine Geometric Resolution Daytime Visual</u> (Cont'd) (3.2.1.1.2.1)

# 2.2.2.3 Acceptance - Thermal Vacuum

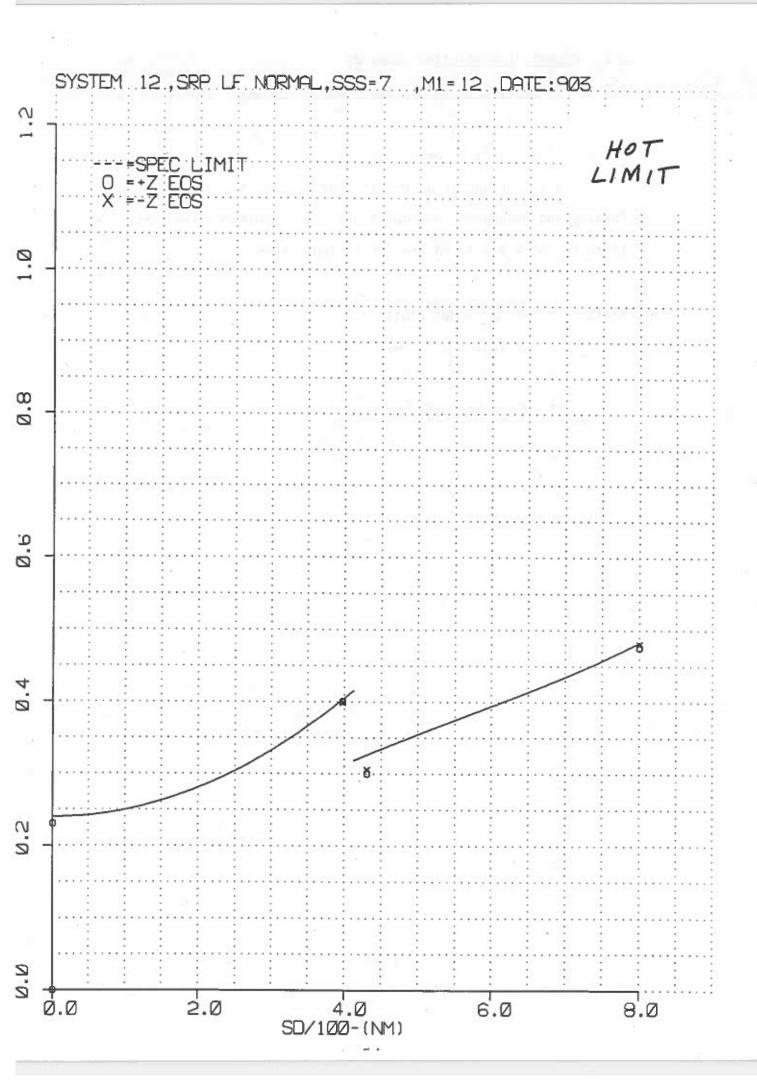
OLS #12 LF SRP is within the specification limits in both Primary and Redundant configurations. The Orbit Nominal curves are in paragraph 2.2.2.1 and are not included here.

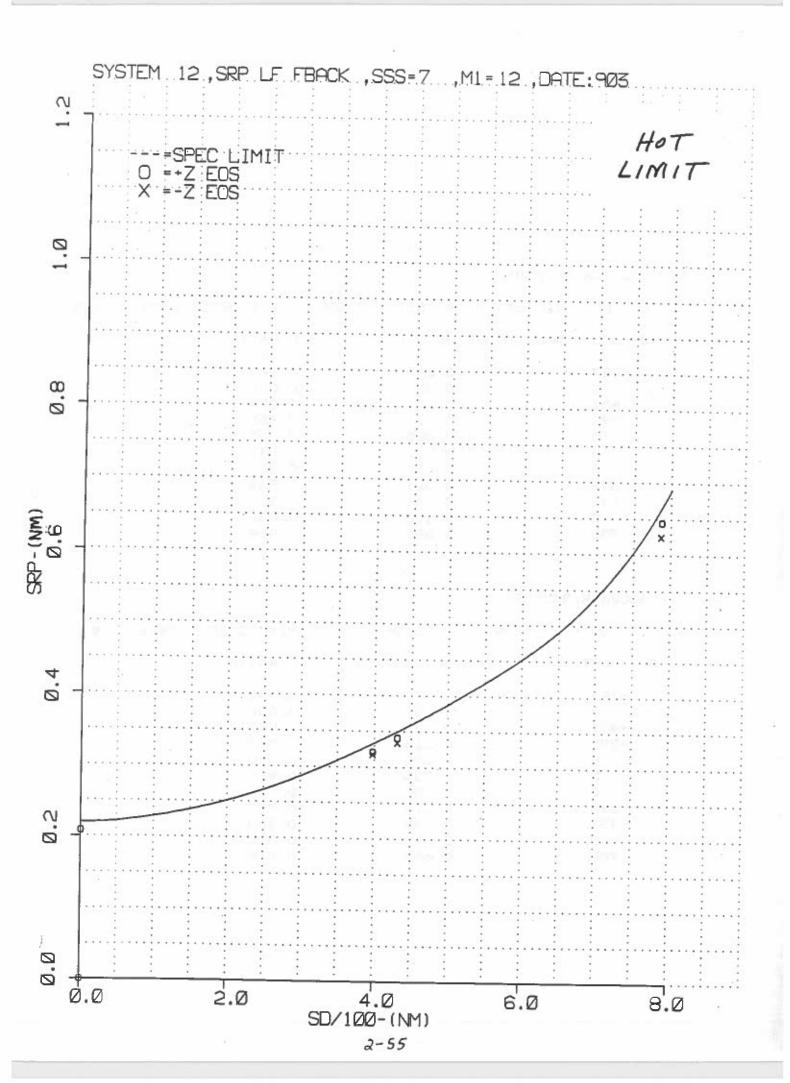
ATTACHMENTS: LF SRP Curve Hot Limit

LF SRP Tables Hot Limit

LF SRP Curves Cold Limit

LF SRP Tables Cold Limit





#### LF.DAY, FALLBACK, PRIMARY

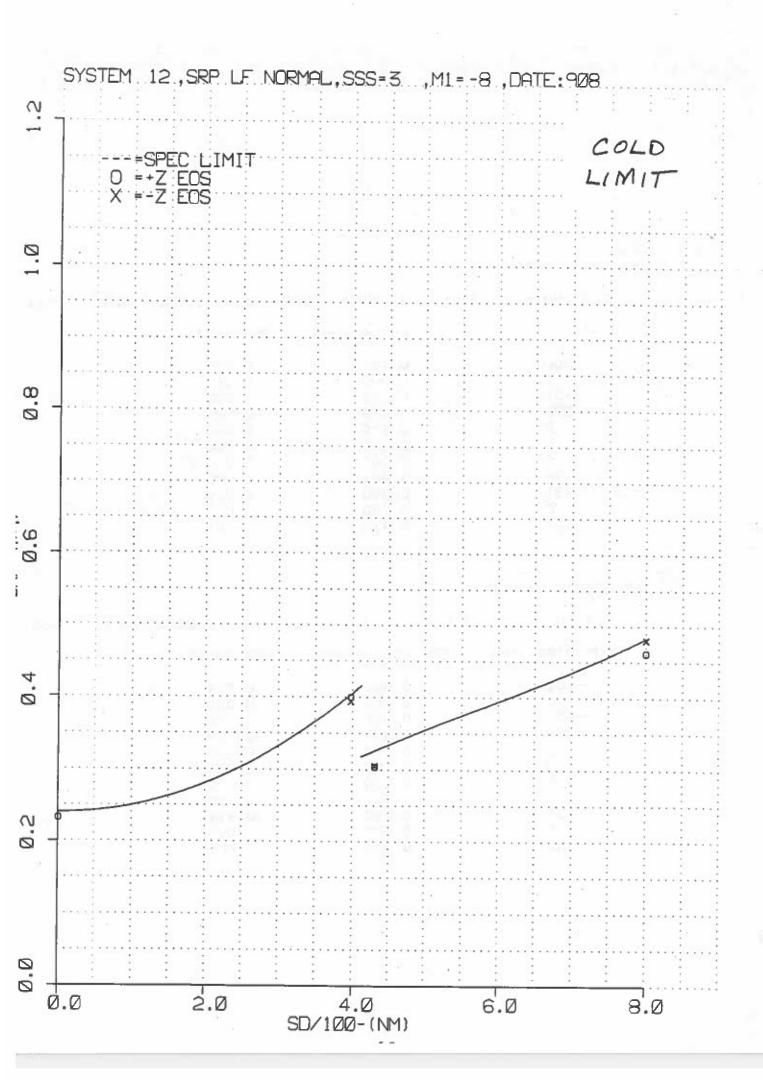
FLT. NO. = 12	ENV. =	4 555=	7DEGC M	11= 12DE	GC DATE:	903
SUR. DIST. (NM)	SRP	ACTUAL (N)	1) SRP	RATIO		
-787.		0. 625	0	. 941		
0.		0. 000	0	. 000		
-431.		0. 334	0	. 954		
-398.		0. 317	0	. 955		
0.		0.000	0	. 000		
0.		0. 207	0	. 947		
Ο.		0. 000	0	. 000		
378.		0. 321	0	. 965		
431.		0. 341	0	). 975		
Ο.		0.000	0	. 000		
788.		0. 645	0	. 969		

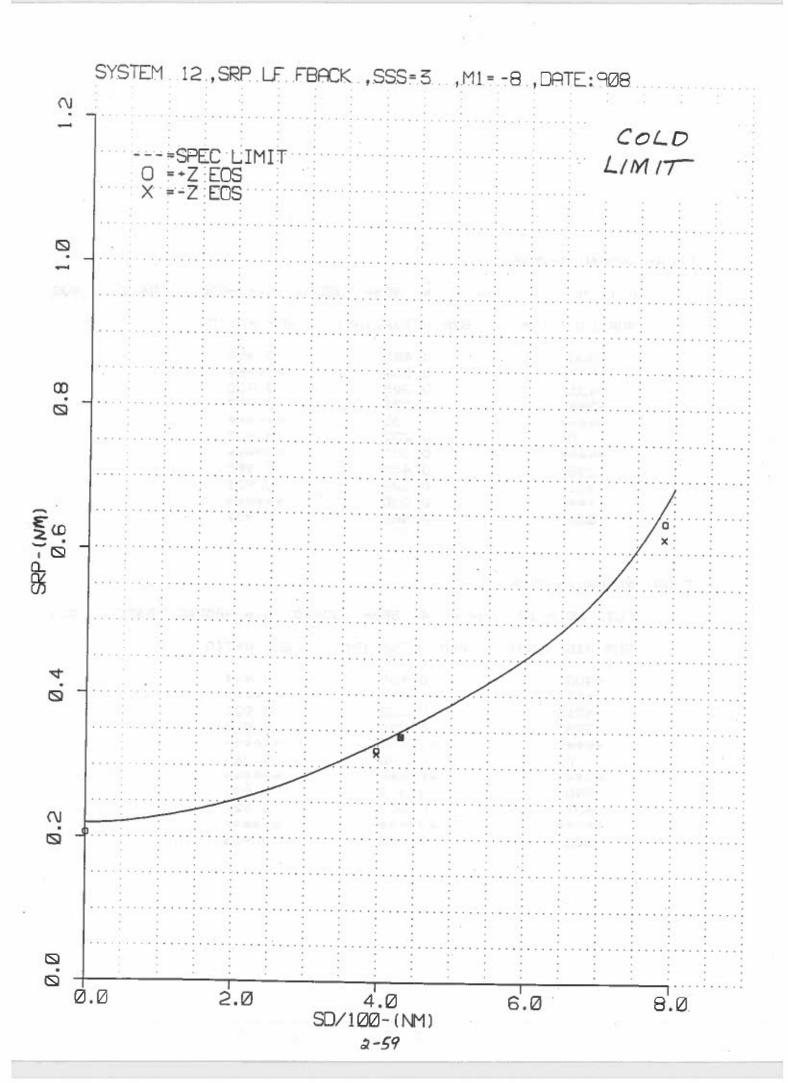
# LF. DAY, FALLBACK, BACKUP

FLT. NO. = 12	ENV. =	4 555=	7DEGC M1= 12DEGC	DATE:	903
SUR. DIST. (NM)	SRP	ACTUAL (N	1) SRP RATIO		
-787.		0. 625	0. 942		
0.		0. 000	0. 000		
-431.		0. 331	0. 947		
-398.		0. 315	0. 947		
0.		0. 000	0. 000		111
0.		0. 204	0. 939		
0.		0. 000	0. 000		
398.		0. 318	0. 957		
431.	•	0. 339	o. 968		
Ο.		0. 000	0. 000		
788		0.645	0. 969		

FLT. NO. = 12	ENV. =	4 555=	7DEGC M1= 12DEGC	DATE:	903
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
-800.		0. 480	0. 995		
0.		0.000	0. 000		
-431.		0. 305	0. <del>9</del> 35		
-378.		0. 399	0. 789		
0.		0.000	0. 000		
0.		0. 230	0. 960		
0.		0. 000	0. 000		
398.		0. 401	0. 993		
431.		0. 300	0. 920		
0.		0.000	0. 000		
800.		0. 476	0. 986		

FLT. NO. = 12	ENV. =	4 SSS=	7DEGC M1= 12DE	EGC DATE: 903
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO	
-800.		0. 480	0. 994	
0.		0. 000	0. 000	
<b>-431</b> .		0. 302	0. 926	E
-378.		0. 399	0. 989	
0.		0.000	0. 000	
0.		0. 229	0. 953	
Ο.		0. 000	0, 000	
398.		0. 400	0. 993	
431.		0. 297	0. 911	
Ο.		0. 000	0, 000	
800.		0. 476	0. 985	





FLT. NO. = 12	ENV. =	4 999=	3DEGC M1= -8DEGC	DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
<b>-800</b> .		0. 480	0. 994	-25	
****		0.000	****		
-431.		0. 305	0. 935		
-398.		0. 394	0. 977		
****		0.000	*****		
0.		0. 232	0. 967		
****		0. 000	****		
378.		0. 400	0. 990		
431.		0. 304	0. 931		
****		0. 000	****		
800.		0. 462	0. 957		

FLT. NO. = 12	ENV. = 4 SSS= 3	DEGC M1= -BDEGC	DATE:	908
SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO		
<b>-800</b> .	0. 480	0. 994		
****	*****	*****		
-431.	0. 302	0. 926		
-398.	0. 394	0. 975		
****	*****	****		
0.	0. 230	0. 960		
****	*****	****		
398.	0. 379	0. 990		
431.	0. 301	0. 922		
****	****	*****		
800.	0. 462	0. 956		

### LE, DAY, FALLBACK, PRIMARY

FLT. NO. = 12	ENV. =	4 555=	3DEGC M1= -BDEGC	DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
<b>-787</b> .		0. 619	0. 932		
****		0. 000	****		
-431.		0. 342	0. 979		
-398.		0.318	0. 957		
****		0. 000	****		
0.		0. 206	0. 940		
****		0. 000	****		
398.		0. 322	0. 970		
431.		0. 343	0. 979		
****		0. 000	***		
<b>78</b> 8.		0. 641	0. 963		

### LF, DAY, FALLBACK, BACKUP

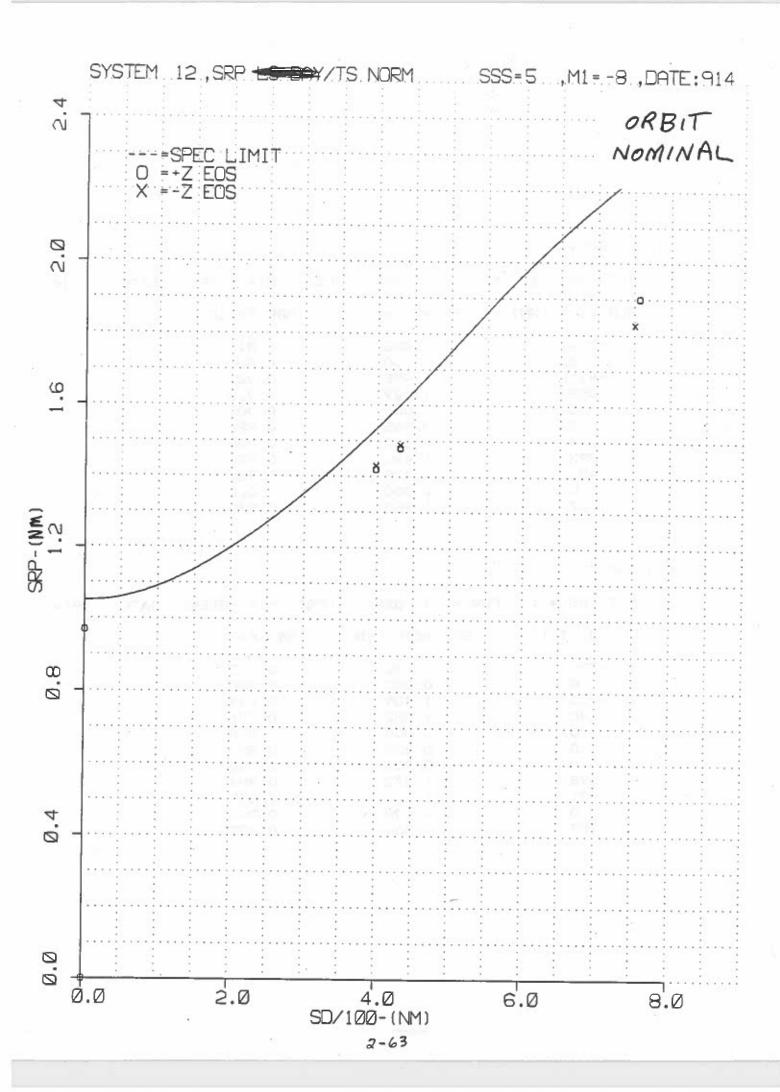
ENV. = 4 SSS=	3DEGC M1= -8DEGC	DATE:	908
SRP ACTUAL (N	1) SRP RATIO		
0. 420	0. 933		
****	****		
0. 340	0. <i>972</i>		
0. 316	0. 949		
****	****		
0. 204	0. 931		
****	****		
0. 320	0. 962		
0. 340	0. 972		
*****	****		
0. 642	0. 964		
	SRP ACTUAL (NN  0. 620  ******  0. 340  0. 316  ******  0. 204  ******  0. 320  0. 340  ******	SRP ACTUAL (NM) SRP RATIO  0. 620 0. 933  ****** 0. 340 0. 972 0. 316 0. 949  ****** 0. 204 0. 931  ****** 0. 320 0. 962 0. 340 0. 972  ******	SRP ACTUAL(NM) SRP RATIO  0. 620 0. 933  ****** 0. 340 0. 972 0. 316 0. 949  ****** 0. 204 0. 931  ****** 0. 320 0. 962 0. 340 0. 972  ******  ******

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.3 Smoothed Geometric Resolution Infrared (3.2.1.1.2.2)
- 2.2.3.1 Baseline (Orbit Nominal)

The TS SRP is within spec for all measured scan angles.

ATTACHMENTS: TS SRP Curve Orbit Nominal

TS SRP Tables Orbit Nominal



# TS, MID, PRIMARY

FL1. NU. = 12	ENV. =	4 555=	5DEGC M1= -BDEGC	DATE:	914
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO	N	
<b>−750</b> .		1. 828	0. 813		
0.		0. 000	0.000		
-431.		1.488	0. 929		
-398.		1. 429	0. 932		
0.		0.000	0. 000		
Ο.		0. 968	0. 922		
Ο.		0. 000	0. 000		
378.		1. 420	0. 926		
431.		1. 479	0. 924		
<b>○</b> 0.		0. 000	0. 000		
<b>757</b> .		1. 900	O. 840		

### TS, MID, BACKUP

FLT. NO. = 12	ENV. ≃	4 555=	5DEGC M1= -BDEGC	DATE: 914
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO	*
<b>−750</b> . □		1. 782	0. 793	
0.		0.000	0. 000	
<b>-431</b> .		1. 439	0. 899	
-378.		1.382	0. 901	
Ο.		0.000	0. 000	
0.		0. 935	0. 890	
٥.		0.000	0. 000	
378.		1. 372	0. 894	
431.		1. 430	0. 893	
0.		0. 000	0. 000	
<b>757</b> .		1.860	0. 823	

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.3 <u>Smoothed Geometric Resolution Infrared</u> (Cont'd)
  (3.2.1.1.2.2)

### 2.2.3.2 Acceptance - Vibration

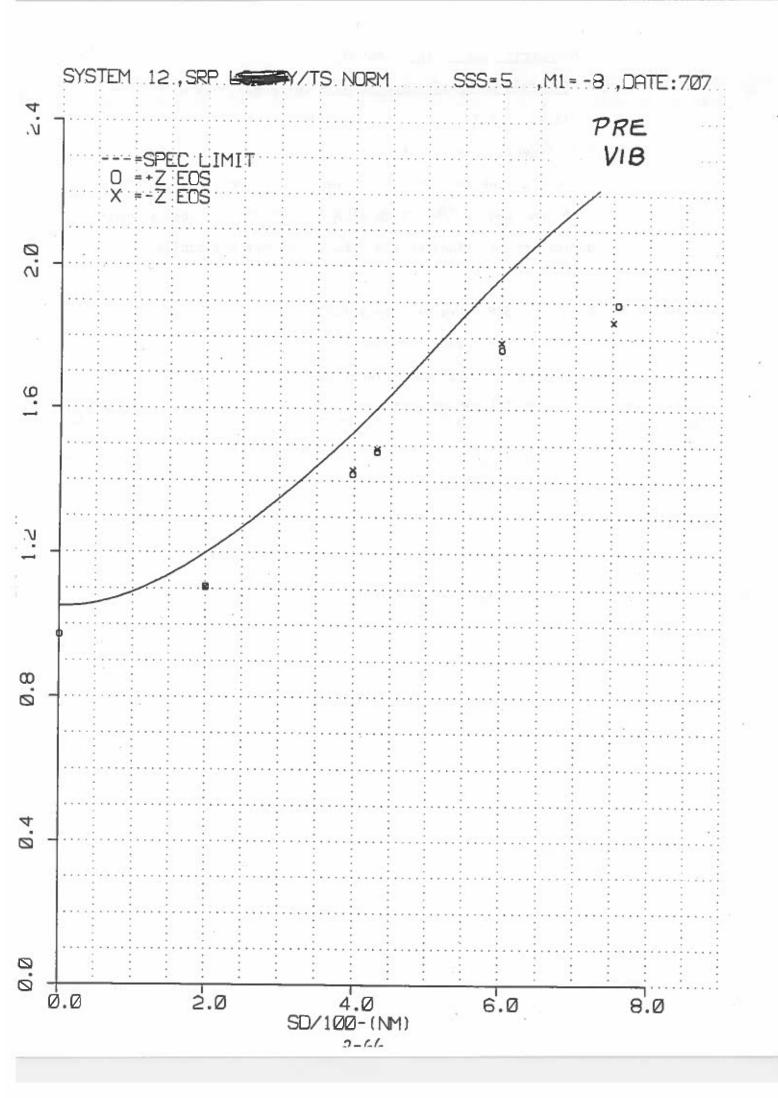
OLS #12 underwent acceptance level SSS vibration per DMSS-OLS-300 with cone cooler S/N 024 on 5/23/91. The pre-to-post vibration SRP performance is shown on the attached curves and tables.

ATTACHMENTS: TS SRP curve pre-vibration

TS SRP tables pre-vibration

TS SRP curve post-vibration

TS SRP tables post-vibration



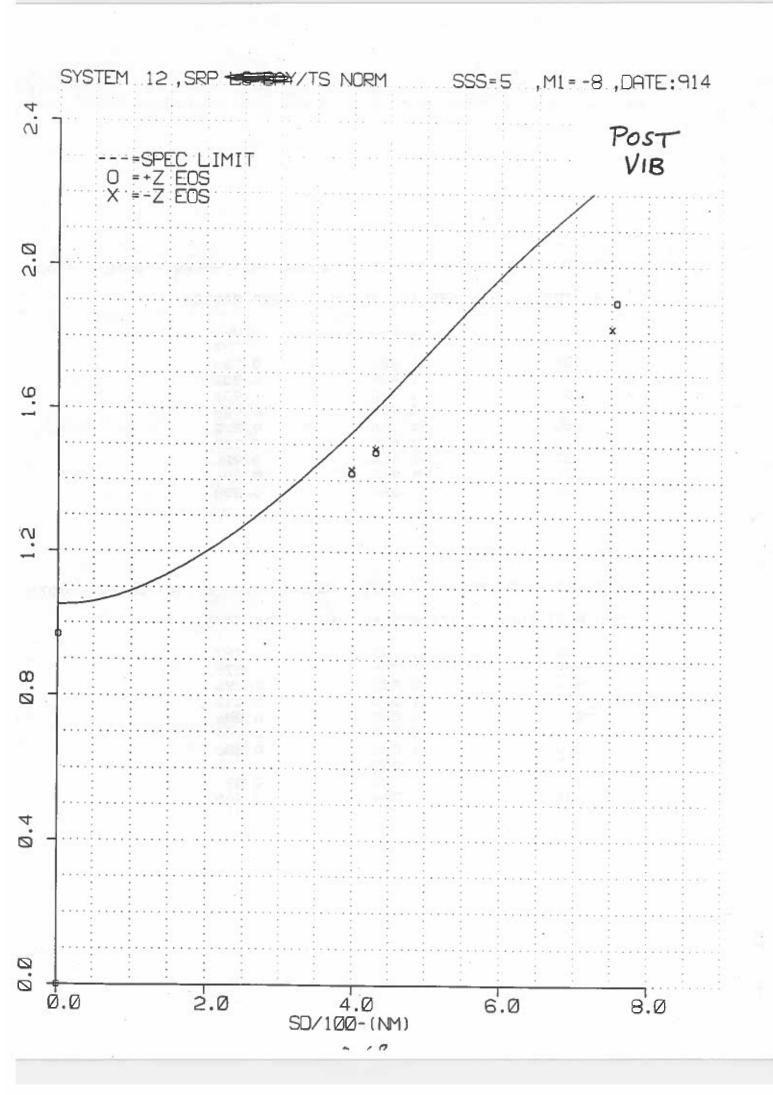
### TS, MID, PRIMARY

FLT. NO. = 12	ENV. =	4 SSS=	5DEGC M1= -8DEGC	DATE:	707
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO	*	
<b>−75</b> 0.		1. 846	0. 821		
-600.		1. 786	0. 907		
<b>-431</b> .		1. 489	0. 930		
-398.		1. 429	0. 932		
<b>-200</b> .		1. 105	0. 922		
0.		0. 971	0. 925		
200.		1. 106	0. 922		
398.		1. 419	0. 925		
431.		1. 482	0. 926		•
<b>401</b> .		1.766	0. 896		
757.		1.894	0. 838		

# TS, MID, BACKUP

SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATIO  -750.	FLT. NO. = 12	ENV. =	4 555=	5DEGC M1= -8DEGC	DATE:
-600.       1.726       0.876         -431.       1.433       0.895         -398.       1.375       0.897         -200.       1.063       0.886         0.0934       0.889         200.       1.063       0.886         398.       1.365       0.890         431.       1.426       0.891         401.       1.704       0.865	SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO	
	-600. -431. -398. -200. 0. 200. 398. 431. 601.		1. 726 1. 433 1. 375 1. 063 0. 934 1. 063 1. 365 1. 426 1. 704	0. 874 0. 895 0. 897 0. 886 0. 889 0. 884 0. 890 0. 891 0. 865	

707



# TS, MID, PRIMARY

FLT, NO. = 12	ENV. =	4 SSS≃	5DEGC M1= -8DEGC	DATE:	914
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
-750. 0. -431. -398. 0. 0. 0. 398. 431.		1.828 0.000 1.488 1.429 0.000 0.968 0.000 1.420 1.479	0.813 0.000 0.929 0.932 0.000 0.922 0.000 0.924		
0. 7 <b>5</b> 7.	**	0. 000 1. <del>7</del> 00	0. 000 0. 840		

### TS, MID, BACKUP

4 SSS= 5DEGC	M1= -BDEGC DATE:	914
ACTUAL(NM) S	RP RATIO	
1.782	0. 793	
0. 000		
1. 439		0.6
1.382	0. 901	
0. 000	0. 000	
0. 935	0. 890	
0. 000	0. 000	
1. 372	0. 894	
1. 430	0. 893	
0. 000	0. 000	
1.860	0. 823	
	ACTUAL(NM) S  1. 782 0. 000 1. 439 1. 382 0. 000 0. 935 0. 000 1. 372 1. 430 0. 000	ACTUAL(NM) SRP RATIO  1. 782

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.3 Smoothed Geometric Resolution Infrared (3.2.1.1.2.2)
- 2.2.3.3 Acceptance Thermal Vacuum

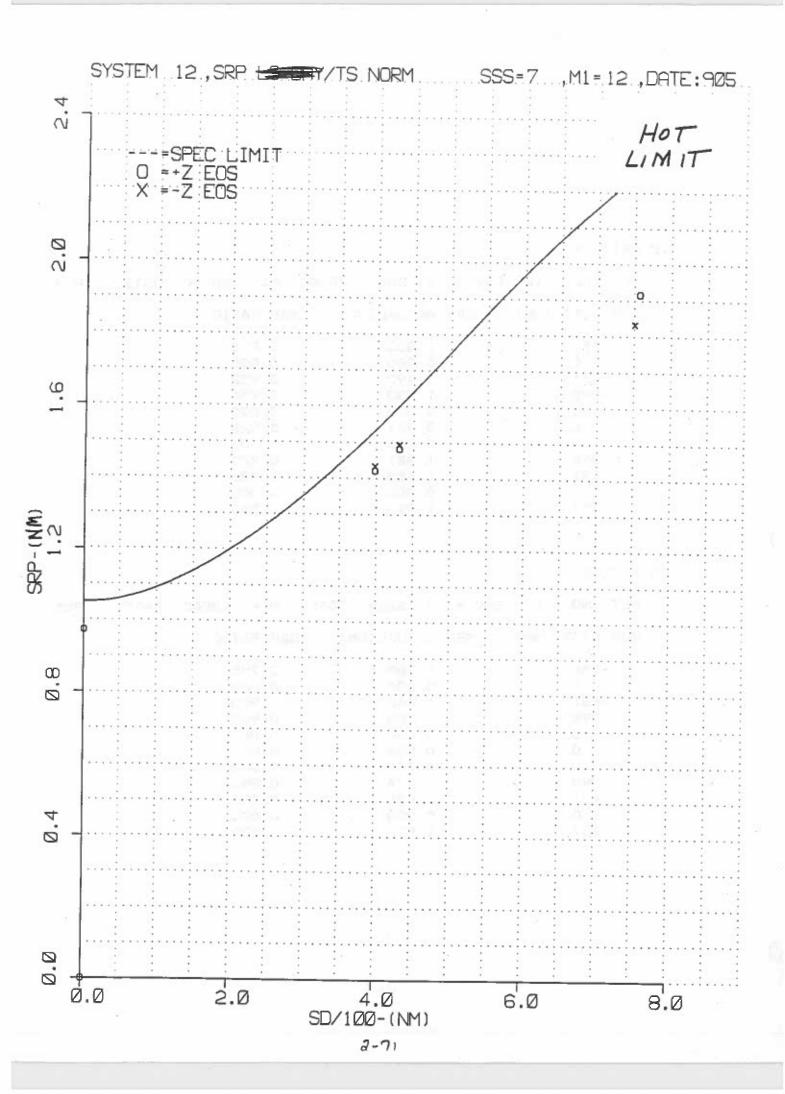
The TS SRP is within spec for the extremes of hot and cold Thermal Vacuum limit testing.

-ATTACHMENTS: TS SRP Curve Hot Limits

TS SRP Tables Hot Limits

TS SRP Curve Cold Limits

TS SRP Tables Cold Limits

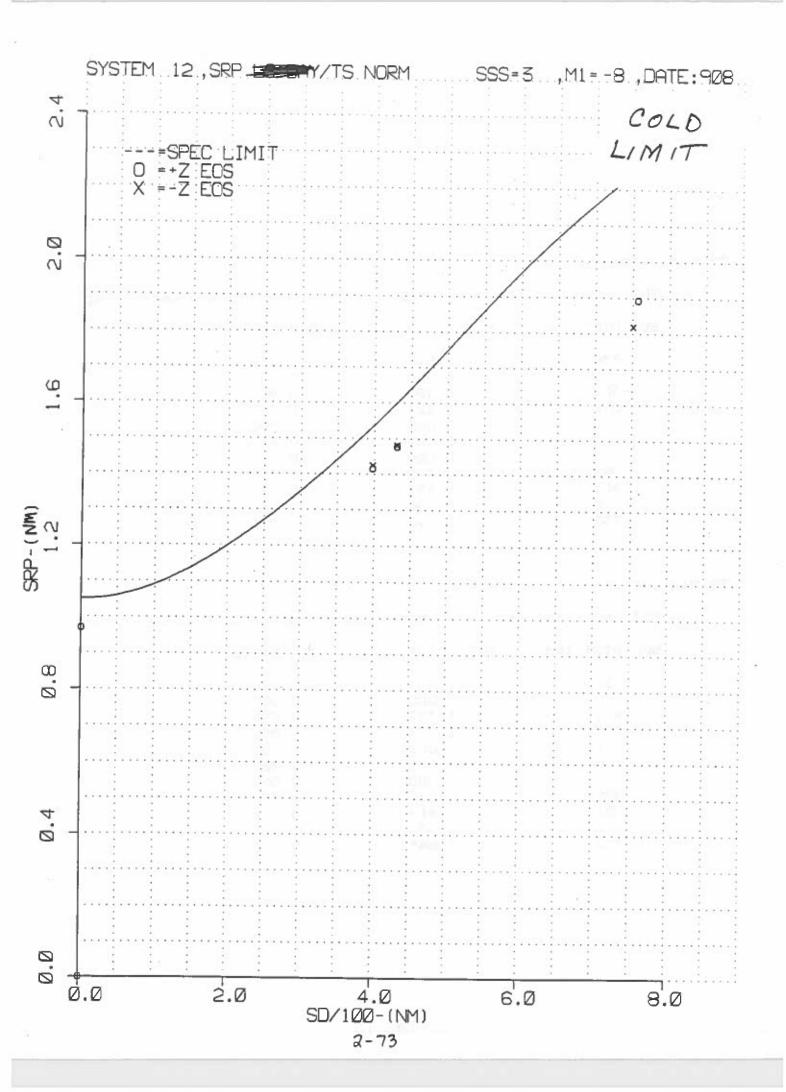


# TE MID, PRIMARY

FLT. NO. = 12 ENV. =	4 555= 7DEGC	M1= 12DEGC DATE:	905
SUR. DIST. (NM) SF	P ACTUAL (NM)	SRP RATIO	
<b>-75</b> 0.	1.833	0. 815	
0.	0.000	0. 000	
-431.	1. 493	0. 932	
-398.	1. 433	0. 934	
O. :	0. 000	0. 000	
0.	0. 971	0. 925	
0.	0. 000	0. 000	
398.	1. 421	0. 927	
431.	1. 483	0. 926	
0.	0. 000	0. 000	
5 <b>757</b> .	1. 916	0. 848	

## TS, MID, BACKUP

FLI. NO. = 12	ENV. = 4	SSS=	7DEGC 1	M1= 12DEG0	DATE:	905
SUR. DIST. (NM)	SRP AC	TUAL (NM	) SRF	RATIO		
<b>-750</b> .	1.	789	(	D. 795		
Ο.	0.	000	C	0. 000		
-431.	1.4	446	C	0. 903		
-378.	1.3	388	C	0. 905		
0.	0.	000	C	0. 000		
0.	0.4	938	C	D. 894		
0.	0.	000	C	0. 000		
398.	1.3	374	C	D. 896		
431.	1.4	435	C	D. 896		
0.	0. (	000	C	0. 000		
<b>757.</b>	1.1	<b>37</b> 6	C	D. 830	27 - 52	



#### TS, MID, PRIMARY

FLT. NO. = 12	ENV. =	4 555=	3DEGC M1= -BDEGC	DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (NI	M) SRP RATIO		
<b>−75</b> 0.		1. 818	0. 809		
Ο.		0.000	0. 000		
<b>-431</b> .		1. 480	0. 925		
<b>-398</b> .		1. 424	0. 928		
0.		0.000	0. 000		
0.		0. 968	0. 922		
0.		0.000	0. 000		
398.		1. 415	0. 923	(2)	
431.		1. 477	0. 922		
0.		0.000	0. 000		
<b>75</b> 7.		1. 893	0. 837		

### TS, MID, BACKUP

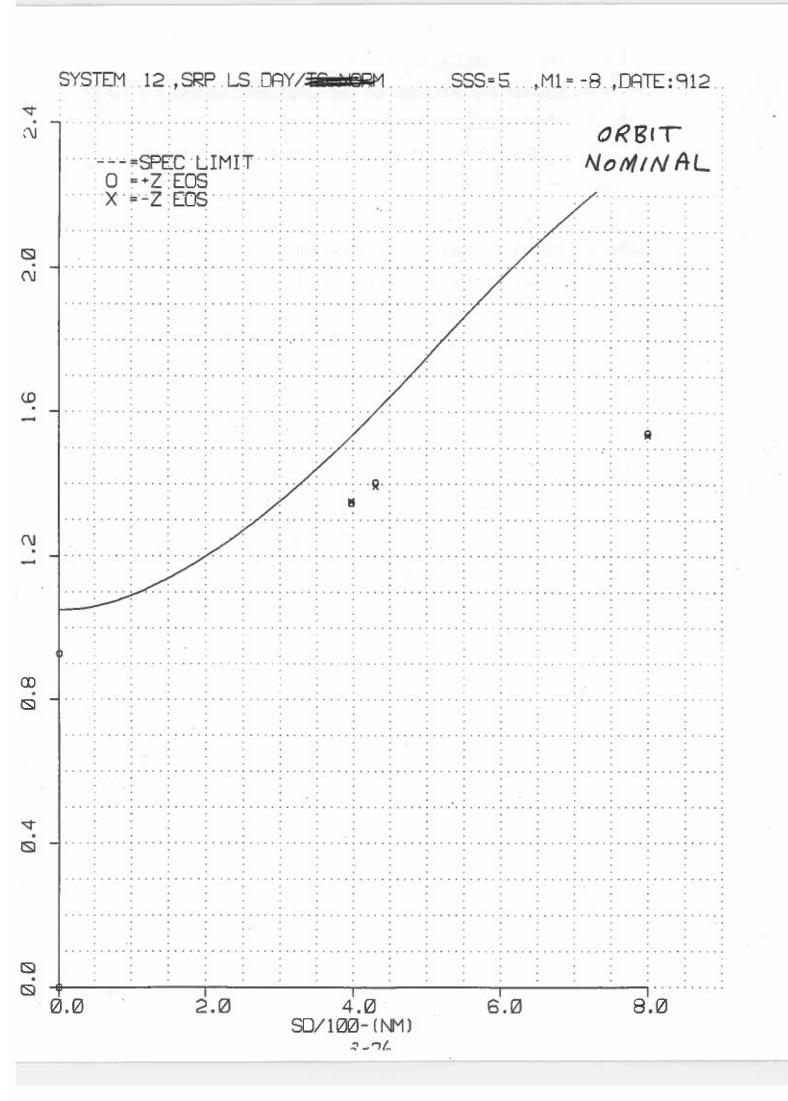
FLT. NO. = 12	ENV. =	4 555=	3DEGC M1= -8DEGC	DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
-750. 0. -431. -398. 0. 0. 0. 398. 431. 0. 757.		1. 764 0. 000 1. 423 1. 369 0. 000 0. 929 0. 000 1. 359 1. 419 0. 000 1. 845	0. 784 0. 000 0. 889 0. 892 0. 000 0. 885 0. 000 0. 886 0. 886 0. 000 0. 816		•
		<del>-</del>			

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.4 <u>Smoothed Geometric Resolution Daytime Visual</u> (3.2.1.1.2.2)
- 2.2.4.1 Baseline (Orbit Nominal)

The LS Day SRP is within spec limits at Orbit Nominal conditions.

ATTACHMENTS: LS Day SRP Curve - Orbit Nominal

LS Day SRP Tables - Orbit Nominal



FLT. NO. = 12	ENV. =	4 555=	5DEGC M1= -8DEGC	DATE:	912
SUR. DIST. (NM)	SRP	ACTUAL (NM	SRP RATIO		
<b>-800</b> .		1. 537	0. 654		
0.		0. 000	0. 000		
<b>-431</b> .		1. 394	0. 871		
-398.		1. 350	0.880		
0.		0. 000	0. 000		
0.		0. 927	0. 883	- 19	
0.		0. 000	0. 000		
39B.		1. 347	0. 878		
431.		1. 403	0. 876		
0.		0. 000	0. 000		
800.		1. 542	0. 454		

# LS, DAY, NORMAL, BACKUP

FLT. NO. = 12	ENV. =	4 555=	5DEGC M1= -BDEGC	DATE:	912
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
<b>-800</b> .		1. 547	0. 658		
0.		0. 000	0. 000		
-431.		1.403	0. 876		
<i>−</i> 398.		1. 359	0. 886		
0.		0. 000	0. 000		
0.		0. 933	0. 888		
0.		0. 000	0. 000		
398.		1. 355	0. 884		
431.		1.412	0.882		
0.		0. 000	0. 000		
800.		1. 552	0. 660		

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.4 <u>Smoothed Geometric Resolution Daytime Visual</u> (Cont'd)
  (3.2.1.1.2.2)

## 2.2.4.2 Acceptance - Vibration

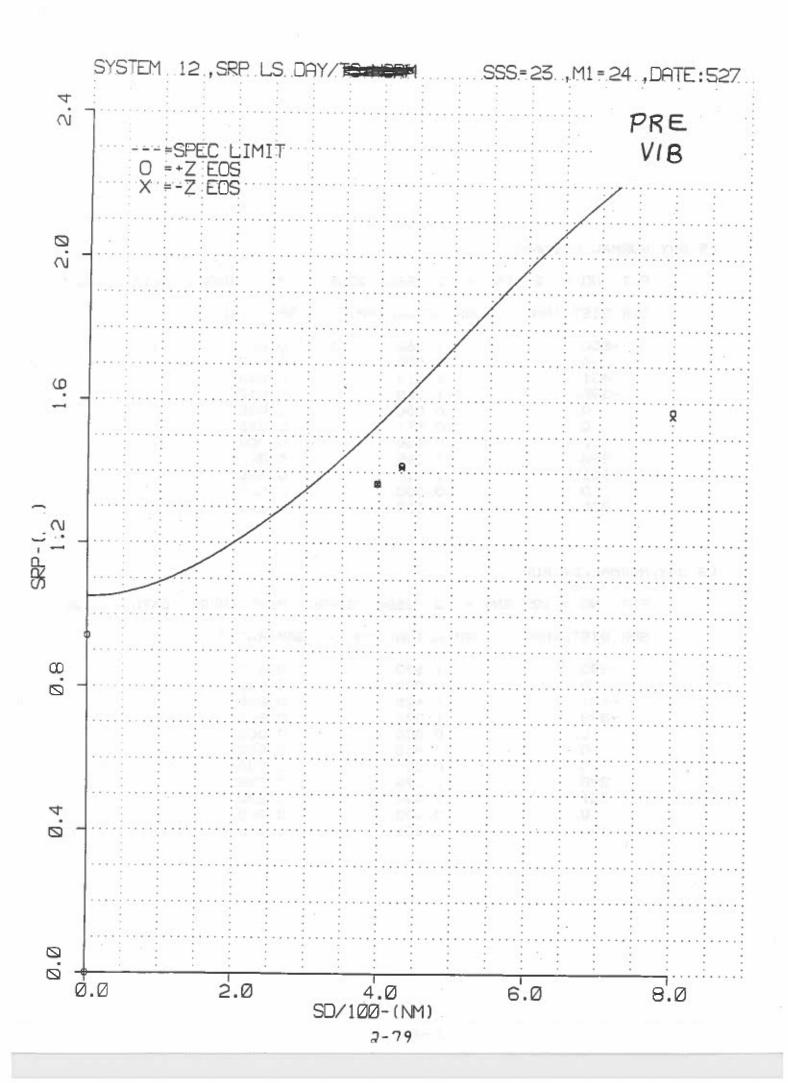
The OLS #12 SSS underwent acceptance level SSS vibrations per DMSS-OLS-300 on May 23, 1991. The LS Day SRP is within specification both before and after SSS vibration. No vibration-related changes in SRP were observed.

ATTACHMENTS: LS Day SRP Curve Pre-Vibration

LS Day SRP Tables Pre-Vibration

LS Day SRP Curve Post-Vibration

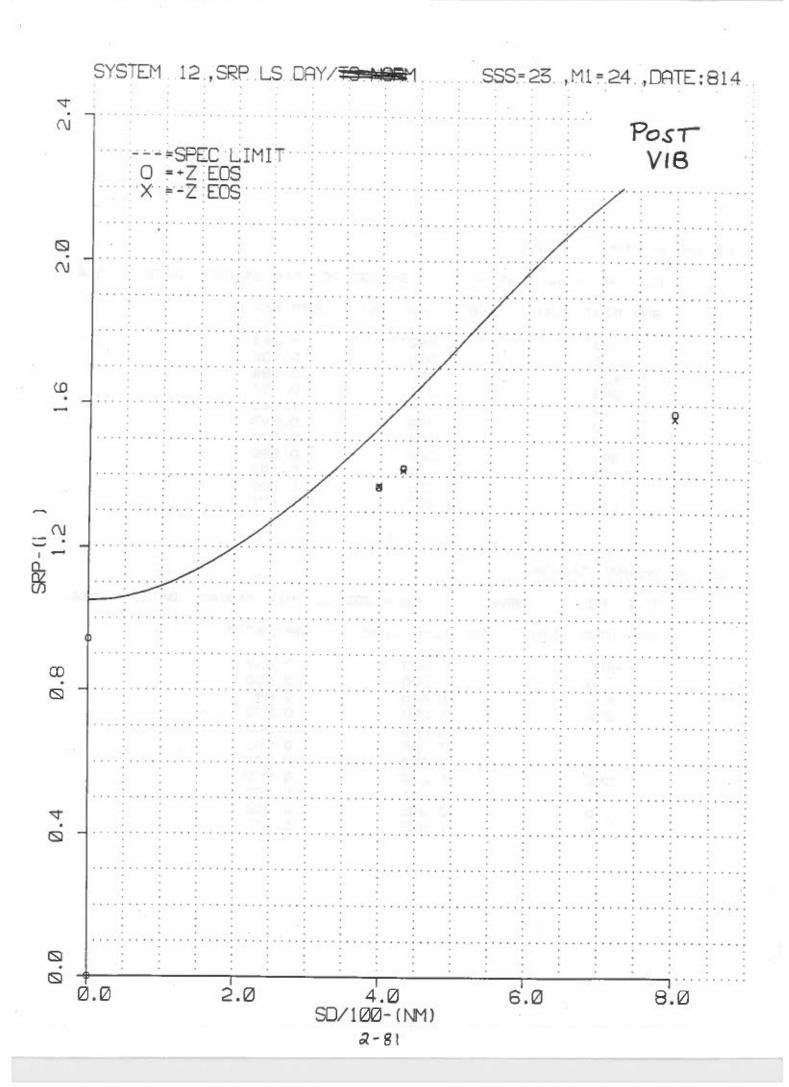
LS Day SRP Table Post-Vibration



FLT. NO. = 12	ENV. =	2 SSS= 23DEGC	M1= 24DEGC	DATE:	527
SUR. DIST. (NM)	SRP	ACTUAL (NM)	RP RATIO	•	
<b>-800</b> .		1. 562	0. 665		- C1
0. -431.		0. 000	0. 000 0. 883		
-378.		1. 368	0. 892		
0.	59	0. 000	0. 000		
0.		0. 941	0. 896		
0.		0. 000	0.000		63
398.		1.368	0. 892		
431.		1. 419	0. 884		
Ο.		0. 000	0.000		
800.		1. 574	0. 670	20	

## LS, DAY, NORMAL, BACKUP

FLI. NO 12 ENV	2 222# 230560	NI = SADEGC	DATE:	32/
SUR. DIST. (NM) SRP	ACTUAL (NM)	SRP RATIO	56	
<b>-800</b> .	1. 573	0. 670		
0.	0. 000	0.000		
-431.	1. 425	0. 890		
-398.	1. 378	Q. 898		
Ο.	0. 000	0. 000	***	
0.	0. 948	0. 902		
0.	0. 000	0.000		
378.	1.378	0. 898		
431.	1. 429	0. 892		
0.	0.000	0.000		
800.	1.586	0. 675		



FLT. NO. = 12	ENV. =	2 555= 231	EGC M1= 24DEGC	DATE:	814
SUR. DIST. (NM)	SRP	ACTUAL (NM)	SRP RATIO		
<b>-800</b> .		1. 562	0. 665		
0.		0. 000	0.000		
-431.		1. 416	0. 884		
-398.		1. 371	0. 894	0	
0.		0. 000	0. 000		
0.		0. 942	0. 897		
O.		0. 000	0. 000		
378.		1. 368	0. 872		
431.	13	1. 422	0. 888		
		0. 000	0. 000		
0.		1. 577	0. 671		
800.		4. 077			

## LS, DAY, NORMAL, BACKUP

FLT. NO. = $12$	ENV. =	5 222= 53DEGC	MI= SANEGO	DHIE. GIT
SUR. DIST. (NM)	SRP	ACTUAL (NM)	SRP RATIO	
-воо. о.		1. 573 0. 000 1. 425	0. 669 0. 000 0. 890	
-431. -398. 0.		1. 380 0. 000	0. 899 0. 000	e . ļ
0. 0. 378.		0. 948 0. 000 1. 377	0. 903 0. 000 0. 898	
431. 0. 800.		1. 431 0. 000 1. 588	0. 893 0. 000 0. 676	

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.4 <u>Smoothed Geometric Resolution Daytime Visual</u> (Cont'd)
  (3.2.1.1.2.2)

## 2.2.4.3 Acceptance - Thermal Vacuum

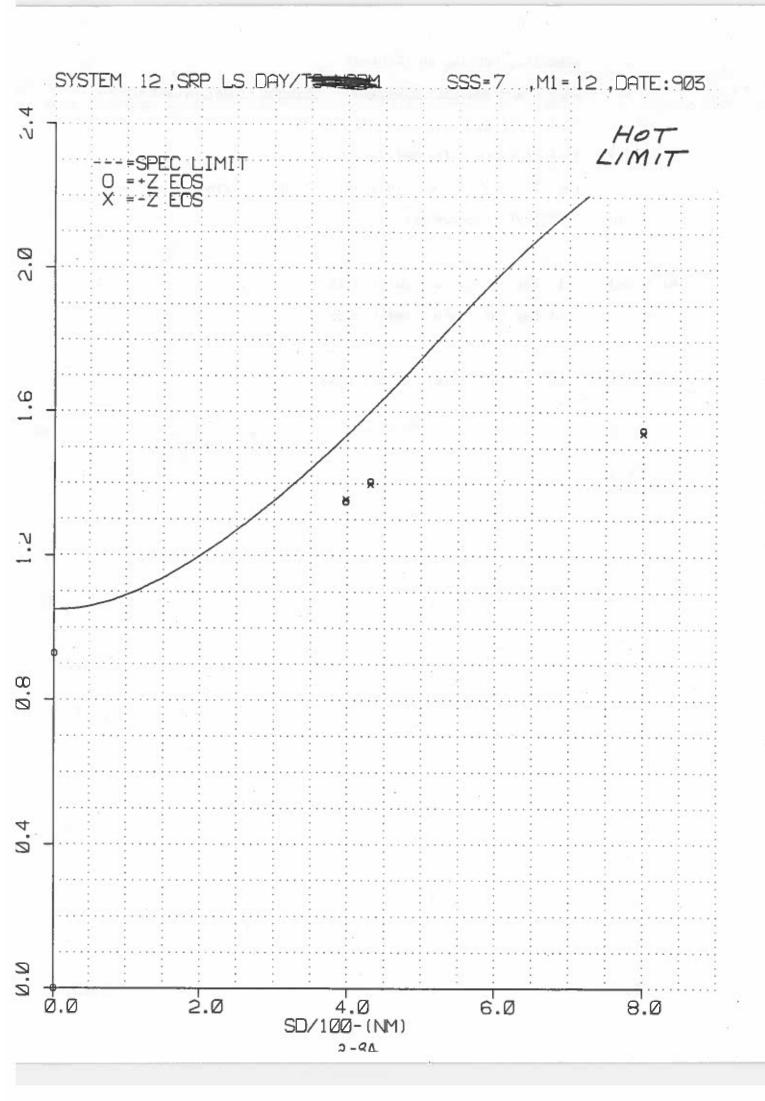
The LS Day SRP is within specification allowance over the entire range of temperatures.

ATTACHMENTS: LS Day SRP Curve Hot Limits

LS Day SRP Tables Hot Limits

LS Day SRP Curve Cold Limits

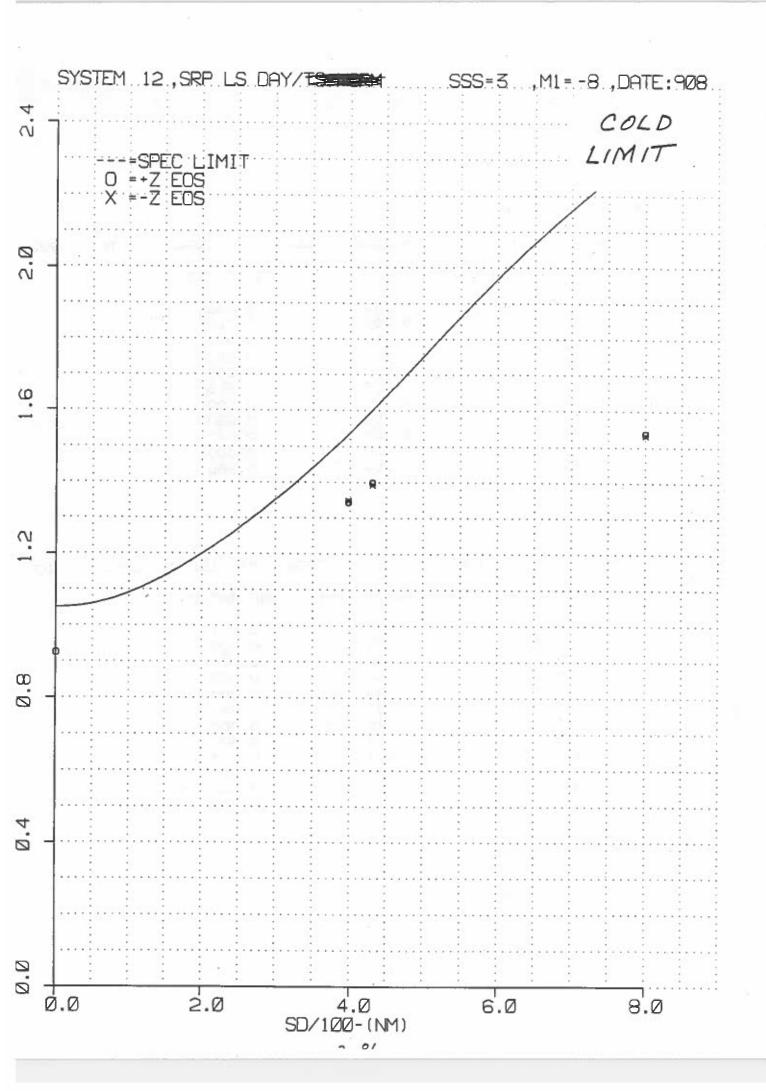
LS Day SRP Tables Cold Limits



ENV. =	4 555=	7DEGC M1=	12DEGC	DATE: 903
SRP	ACTUAL (N	M) SRP RA	TIO	
	1. 543	0. 65	6	
	0. 000	0.00	0	
	1. 378	0. 87	3	
	1. 355	0.88	3	
	0.000			
	0. 929			
	0.000			
	1. 350	0.88	0	
	1. 405			
	0. 000		4.15	
	1. 550			
		SRP ACTUAL (NI  1. 543 0. 000 1. 378 1. 355 0. 000 0. 929 0. 000 1. 350 1. 405 0. 000	SRP ACTUAL (NM) SRP RA  1. 543	SRP ACTUAL (NM) SRP RATIO  1. 543

## LS, DAY, NORMAL, BACKUP

FLT. NO. = 12	ENV, =	4 SSS=	7DEGC M1= 12DEGC	DATE: 903
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO	
-800. 0. -431.		1. 553 0. 000 1. 408	0. 661 0. 000 0. 879	
-398. 0. 0. 0.		1. 364 0. 000 0. 935 0. 000	0. 889 0. 000 0. 890	
398. 431. 0. 800.		1. 359 1. 414 0. 000 1. 561	0. 000 0. 884 0. 883 0. 000 0. 444	×



FLT. NO. = 12	ENV. = 4 SSS= 3	DEGC M1= -8DEGC	DATE:	908
SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO		
<b>-800</b> .	1. 533	0. 452		
****	****	****		
-431.	1. 393	0. 870		
-398.	1. 348	0. 878		
***	****	*****		
0.	0. 925	0. 881		
****	****	****		
378.	1. 344	0. 876		
431.	1.398	0. 873		
****	*****	****		
800.	1. 537	0. 654		

### LS, DAY, NORMAL, BACKUP

FLT. ND. = 12 ENV. = \_4 SSS= 3DEGC M1= -BDEGC DATE: \_908

SUR. DISTE (NM)	SRP ACTUAL(NM)	SRP RATIO
-B002~.	1. 543	0. 657
<b>米米米格</b> 二。	10. 897	****
<b>-431</b> . **	1. 402	0. 876
-378.	1. 357	0. 884
****	10. 897	****
0	0. 931	0. 886
****	10: 897	****
398.	1. 353	0. 882
431.	1. 408	0. 879
****	10. 897	*****
° 800.	1. 548 ·	0. 459

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.5 <u>Smoothed Geometric Resolution Nighttime Visual</u> (3.2.1.1.2.2)

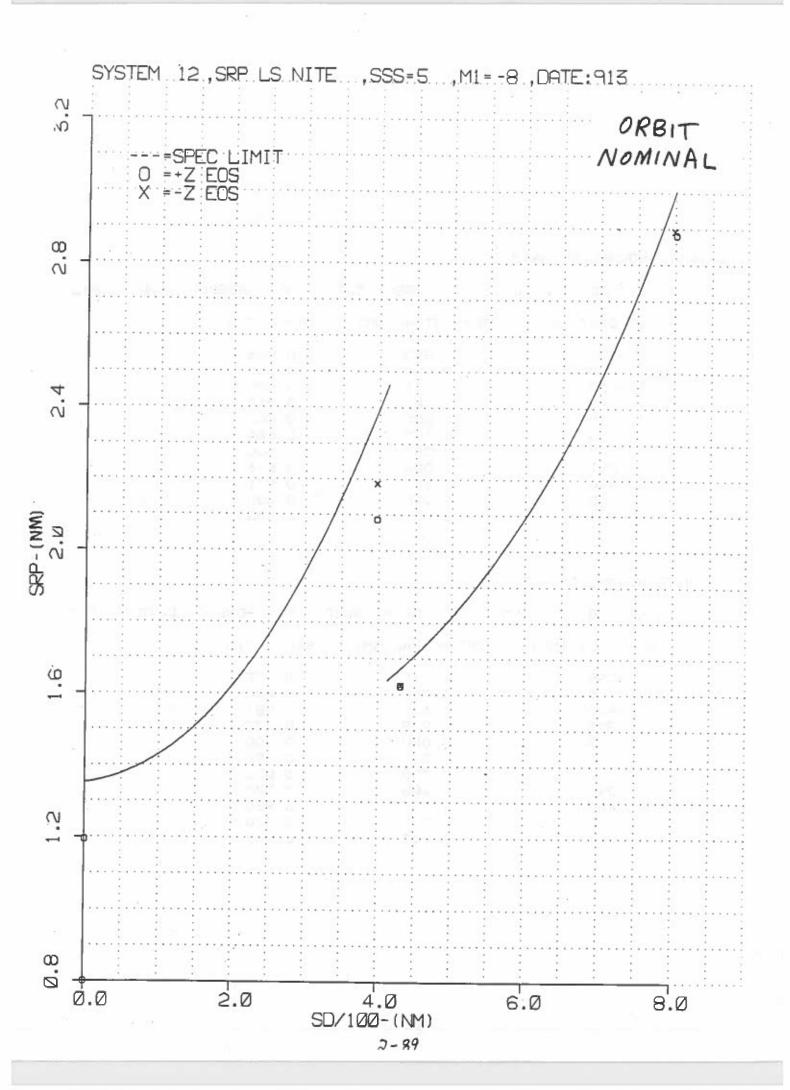
LS Night SRP routinely is deliberately adjusted to be close to spec limit during system integration in order to optimize PMT signal-to-noise ratio, at the expense of SRP margin.

### 2.2.5.1 Baseline (Orbit Nominal)

The LS Night SRP is within spec for all measured scan angles

ATTACHMENTS: LS Night SRP Curve - Orbit Nominal

LS Night SRP Table - Orbit Nominal



FLT. NO. = 12	ENV. =	4 555=	5DEGC M1= -BDEGC	DATE:	913
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
<b>-799</b> .		2. 873	0. 966		
0.		0. 000	0. 000		1557
-430.		1.624	0. 972		
-397.		2. 187	0. 918		
0.		0.000	0. 000		
0.		1. 193	0.884		
0.		0. 000	0. 000		
<b>397</b> .		2. 086	0. 876		
430.		1.622	0. 971		
0.		0.000	0. 000		
801		2 881	0.958		

FLT. NO. = 12	ENV. =	4 555=	5DEGC M1= -8DEGC	DATE:	913
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SRP RATIO		
<b>-799</b> .		2. 919	0. 975		
0.		0. 000	0. 000		
· <b>-430</b> .		1. 639	0. 981		
<del>-</del> 397.		2. 209	0. 927		
0.		0. 000	0. 000		
0.		1. 205	0. 892		
0.		0. 000	0. 000		
397.		2. 108	0. 885		
430.		1. 637	0. 980		
<b>O.</b>	2.	0.000	0. 000		
801.		2. 907	<b>0. 967</b>		

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.5 <u>Smoothed Geometric Resolution Nighttime</u>

<u>Visual</u> (Cont'd) (3.1.2.2)

## 2.2.5.2 Acceptance - Vibration

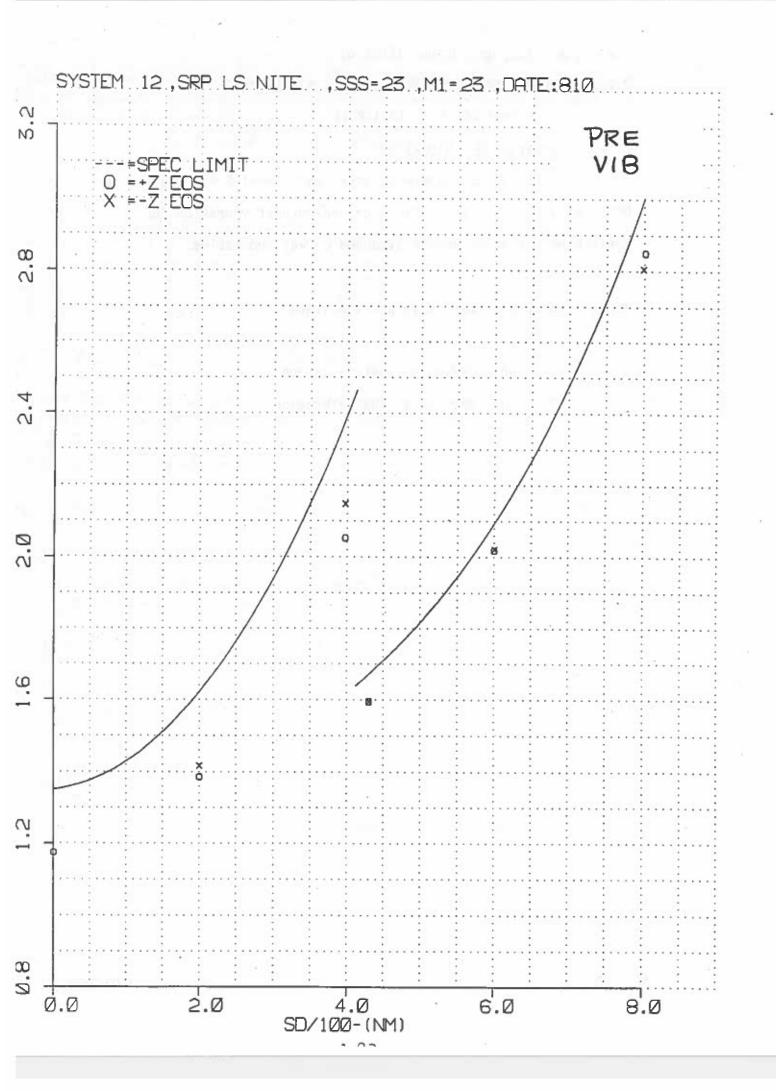
The OLS #12 SSS underwent Acceptance level SSS vibration per DMSS-OLS-300 on May 23, 1991. The pre-to-post vibration SRP performance is shown on the attached curves and tables.

ATTACHMENTS: LS Night SRP curve pre-vibration

LS Night SRP tables pre-vibration

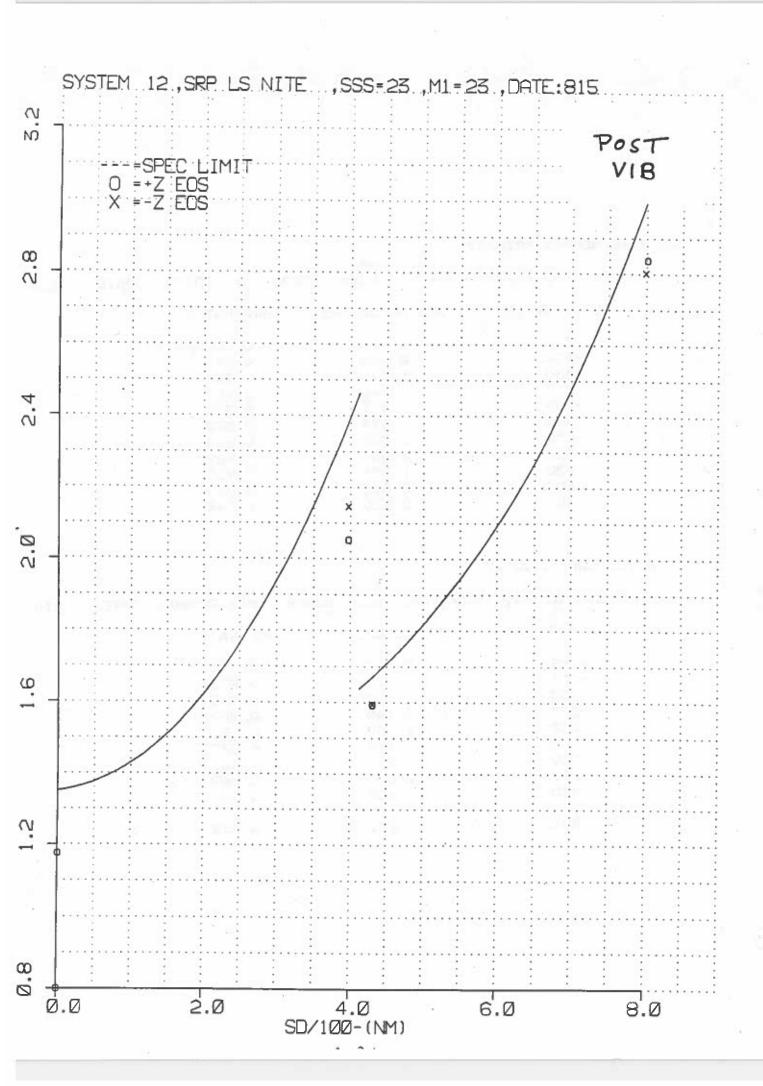
LS Night SRP curve post-vibration

LS Night SRP tables post-vibration



FLT. NO. = 12	ENV. =	2 SSS= 23DEGC	M1= 23DEGC	DATE: 810
SUR. DIST. (NM)	SRP	ACTUAL(NM) S	RP RATIO	
-799. -601. -430. -397. -200. 0. 200. 397. 430. 600. 801.		2. 806 2. 022 1. 595 2. 148 1. 417 1. 174 1. 386 2. 053 1. 596 2. 018 2. 850	0. 937 0. 965 0. 954 0. 902 0. 874 0. 870 0. 855 0. 862 0. 955 0. 964 0. 948	

FLT. NO. = 12	ENV. =	2 SSS=	23DEGC 1	11= 23DEGC	DATE:	810
SUR. DIST. (NM)	SRP	ACTUAL (N	IM) SRF	P RATIO		
<b>-799</b> .		2. 828	i:: (	), 944		
-601.		2. 039	C	). 973		
<b>-430</b> .		1. 607		). 962		
<del>-</del> 397.		2. 165		), 909		
<b>-200</b> .		1.430	C	). 882		
О.		1. 184		). 877		
200.		1.398		). 863		
397.		2. 070	O	). 869		
430.		1. 608		). 962		
<b>600</b> .		2. 036	_	. 972		
801		2 071		055		



F	LT. NO. = 12	ENV. =	2 SSS= 23DEG0	M1= 23DEGC DATE	: 815
S	UR. DIST. (NM	) SRP	ACTUAL (NM)	SRP RATIO	
	<b>-799</b> .		2. 802	0. 936	
	0.		0.000	0. 000	
	<b>-430</b> .		1. 595	0. 955	
	<b>-397</b> .	e.	2. 146	0. 901	
	0.	520	0.000	0. 000	
	Ο.		1. 175	0. 870	
	О.	10	0. 000	0. 000	
	397.		2. 053	0. 862	
	430.		1. 593	0. 953	
	O		0.000	0. 000	
	801.		2. 839	0. 944	

FLT. NO. = 12	ENV. =	2 SSS= 23DEGC	M1= 23DEGC	DATE:	815
SUR. DIST. (NM)	SRP	ACTUAL(NM) S	RP RATIO		
-799. 0. -430. -397. 0. 0. 397. 430. 0.	¥0	2. 825 0. 000 1. 608 2. 163 0. 000 1. 185 0. 000 2. 070 1. 605 0. 000 2. 862	0. 944 0. 000 0. 962 0. 908 0. 000 0. 878 0. 000 0. 869 0. 961 0. 000 0. 951		

- 2.2 <u>Geometric Resolution</u> (Cont'd)
- 2.2.5 <u>Smoothed Geometric Resolution Nighttime</u>

  <u>Visual</u> (Cont'd) (3.1.2.2)

2.2.5.3 Acceptance - Thermal Vacuum

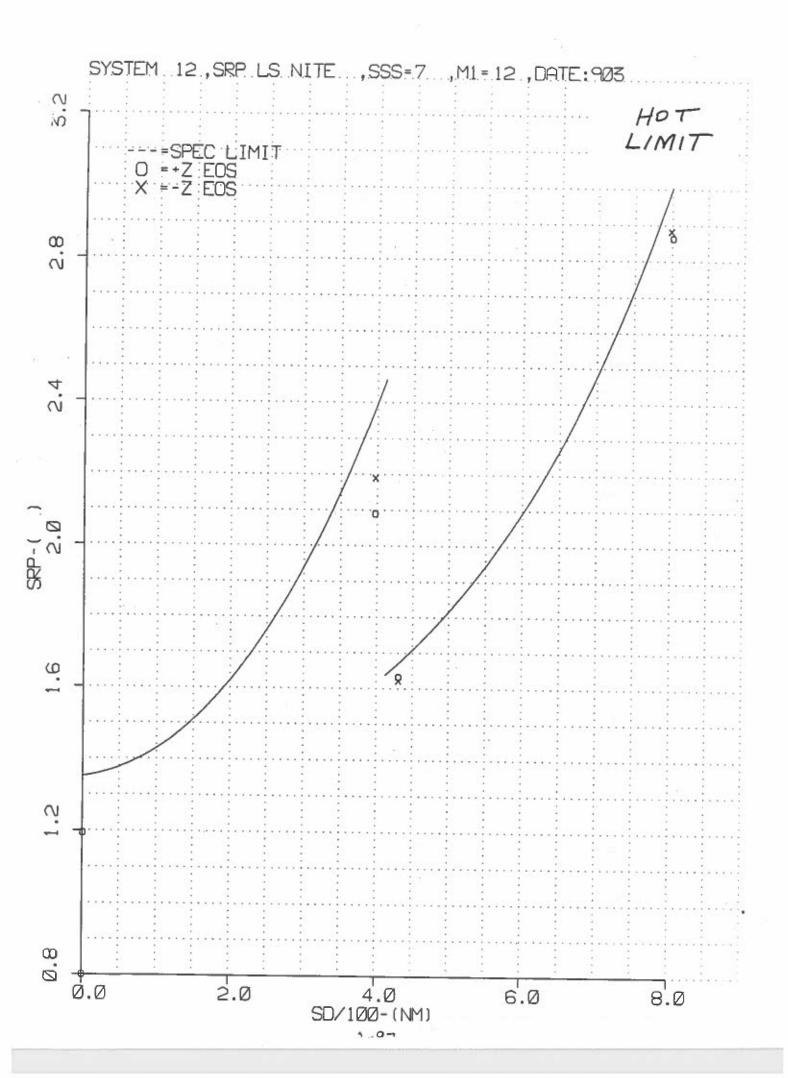
The LS Night SRP is within specification limits over the Acceptance temperature range. No temperature-related changes in SRP over the Acceptance temperature range were observed.

ATTACHMENTS: LS Night SRP Curve Hot Limits

LS Night SRP Tables Hot Limits

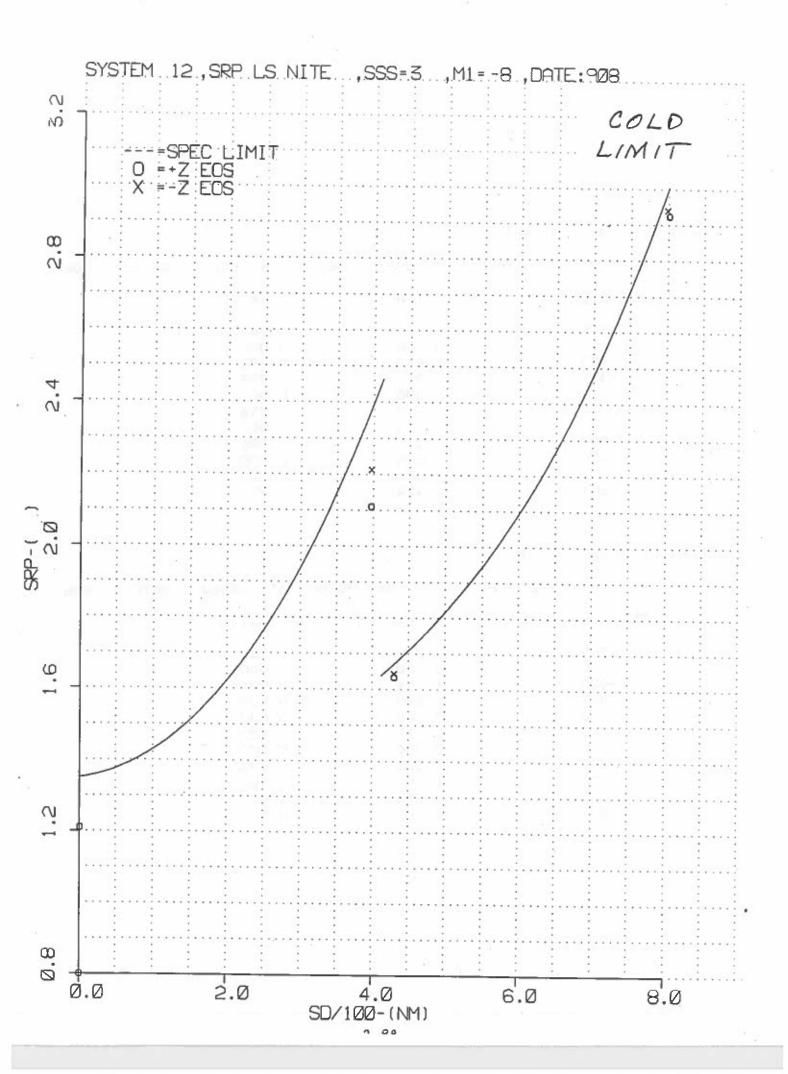
LS Night SRP Curve Cold Limits

LS Night SRP Tables Cold Limits



FLT. NO. = 12	ENV. =	4 5SS≃ 7	DEGC M1= 12DEGC	DATE:	903
SUR. DIST. (NM)	SRP	ACTUAL(NM)	SRP RATIO		
-799. 0.		2. 883 0. 000	0. 963 0. 000		
-430. -397		1:622	0.971 0.919		
O. O.		0.000 1.194	0. 000 0. 884		
O.		0.000	0.000		
397. 430.		2.090 1.632	0. 877 0. 977		
0. 801.		0,000 2,863	0.000 0.952		

FL1, NB, = 12	ENV. =	4 SSS=	7DEGC M1= 12DEGC	DATE:	903
SUR DIST (NM)	SRP	ACTUAL (N	M) SRP RATIO		
-799. 0. -430. -397. 0. 0. 397. 430. 0. B01.	76 98	2. 904 0. 000 1. 636 2. 208 0. 000 1. 204 0. 000 2. 108 1. 647 0. 000 2. 884	0. 970 0. 000 0. 979 0. 927 0. 000 0. 891 0. 000 0. 885 0. 985 0. 000 0. 959		



FLT.	ND. = 12	ENV. =	4	SSS=	3DEGC	M1	= -8DEGC	DATE:	908
SUR.	DIST. (NM)	SRP	AC <sup>-</sup>	TUAL (N	M)	SRP	RATIO		
_	·799.		2. 9	741		0.	<b>782</b>		
	0.		0. (	000		0.	000	12	
_	430.		1. 6	543		0.	<del>9</del> 83		
	397.		.2. 2	212		0.	929		
	0.		Q. (	000		0.	000		
	0.		1. 2	210		0.	896		
	0.		0.0	000		0.	000		
	397.		2. :	110		0.	884		
	430.		1. 6	534		0.	978		
	0.		0. 0	000		0.	000		
	801.		2. 9	724			972		
	<b>601.</b>		2. 4	724		<b>O</b> . '	972		

PE1. NO. = 12	ENV. =	4 555=	BDEGC	WI= -BDEG	C DATE:	908
SUR. DIST. (NM)	SRP	ACTUAL (N	M) SF	RP RATIO		
- <b>799</b> .		2. 958		0. 988		
0.		0. 000		0.000		
<b>-430</b> .		1. 656		0. 991		
<del>-</del> 397.		2. 228		0. 935		
0.		0. 000		0.000		
٥.	¥6	1. 219		0. 903		
0.		0. 000		0.000		
397.		2. 126		0. 893		
430.		1.647		0. 986		
0.		0. 000		0. 000		
801.		2. 942		0. 978		

## 2.2 <u>Geometric Resolution</u> (Cont'd)

#### 2.2.6 <u>Data Sampling</u> (3.2.1.1.2.3)

The sampling frequency ratios for all modes of the 5D-3 OLS satisfy the specification requirements. The calculations are contained in the 5D-3 OLS System Summary Report. The results are summarized below. The worst-case sampling frequency ratio for each mode is given.

MODE	<pre>SAMPLING FREQ. RATIO (Spec: &gt; 2.4)</pre>
LF Day - Normal	2.58
LF Day - Fallback	2.61
LS & TS Day - Normal	2.50
LS Night - Normal	2.66
TF - Normal	3.44
TF Fallback - Normal Side of scar	3.28
TF Fallback - Abnormal Side of So	can 3.28

# 2.3 Geometric Accuracy (3.2.1.1.3.1, 3.2.1.1.3.2, 3.2.1.1.3.3)

The alignment and synchronization of the SSS determine the Geometric Accuracy. Geometric accuracy is specified in 3 categories (Repeatability, Stability & Fixed Errors). Within these categories, accuracy is further specified for Along Track, Along Scan (Stored Data), Along Scan (Direct Data), and Along Scan (with digital delphi generation, i.e., the Encoder Simulator locked mode).

There was an observed shift in alignment between the extremes of M1 temperature of -8°C and +12°C on the order of 0.16 milliradians or less for all channels. There was also an observed shift in synchronization in all modes in OLS #12 of approximately 0.1 milliradians or less between M1 temperature extremes of -8°C and +12°.

The Repeatability error is calculated using the rms difference of the measured alignment/synchronization (as a function of the variation of M1 temperature on orbit) from the hot-to-cold average values of alignment/ synchronization. The total Repeatability contribution for synchronization is determined by rss'ing the wow-flutter error (as measured in test 6x11.ST) and the jitter error (which was negligible on OLS #12) with the repeatability shift between the Hot and Cold Limits.

The Primary Alignment Reference Axes (REFPLN) are calculated in a computer program using HRD & T channel Alignment and Sync data from several Orbit Nominal tests. The data determine a best-fit alignment with respect to the Interface Mounting Axes. The REFPLN Alignment and REFPLN Synchronization curves are included here. The remainder of the Alignment and Synchronization data for the OLS #12 SSS are in BVS 2693, Vol. III of this Acceptance Test Report.

The 5D-3 System has a Fallback mode utilizing an encoder control track and Encoder Simulator. The spec limits and measured results are shown in Table 2.3-2.

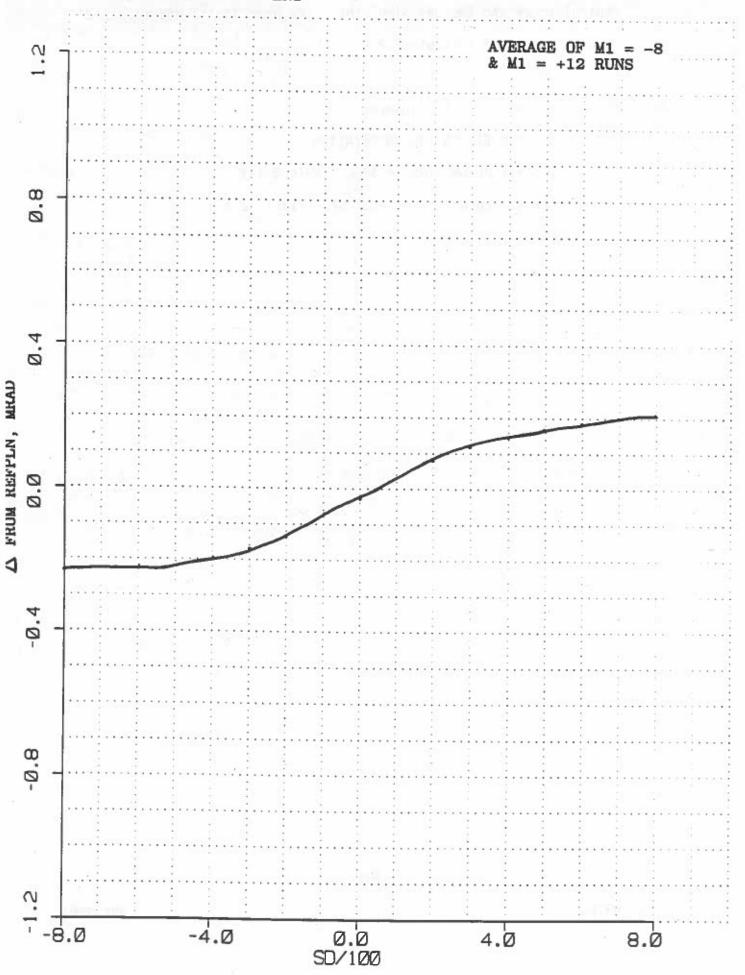
ATTACHMENTS: OLS #12 REFPLN ALIGNMENT

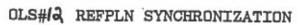
OLS #12 REFPLN SYNCHRONIZATION

OLS #12 ALIGN/SYNC vs SPEC, Table 2.3-1

OLS #12 Encoder Simulator Sync, Table 2.3-2







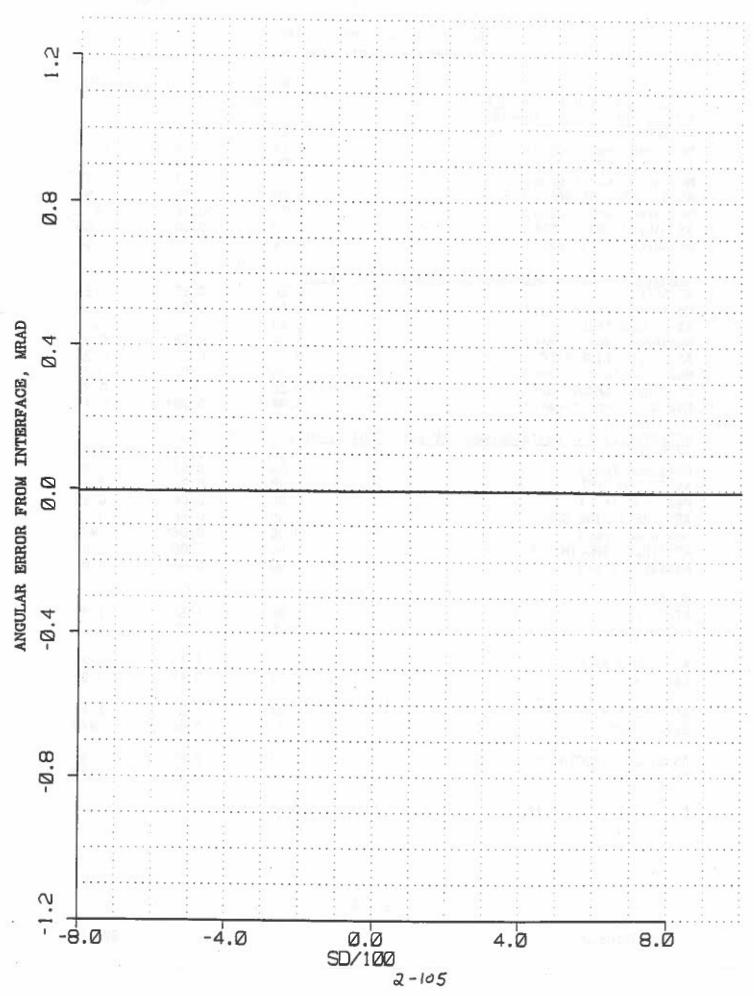


Table 2.3-1

OLS #12 ALIGN/SYNC vs. SPEC all numbers in milliradians

	HRD	<u>T</u>	<u>PMT</u>
FIXED - Delta between "REFPLN"  & Optic Hot - Cold Average			
AT SPEC	0.45	0.70	0.60
Measured (worst-case)	0.19 0.80	0.25 0.80	0.14 1.90
AS STORED SPEC Measured (worst-case)	0.26	0.33	.76
AS DIRECT FINE SPEC	0.80	0.80	1.90
Measured (worst-case)	0.24	0.36	N/A 1.90
AS DIRECT SMOOTH SPEC Measured (worst-case)	0.80 0.24	0.80 0.36	0.97
Heasured (worst-case)			
STABILITY - Delta Between Pre & Post - Vibratio	<u>on</u>	0.55	0.55
AT SPEC Measured (worst-case)	0.50 0.10	0.03	0.09
AS STORED SPEC	0.20	0.25	0.25
Measured (worst-case)	0.04	0.08	0.14
AS DIRECT FINE SPEC	0.20 0.04*	0.25 0.08*	0.25 N/A
Measured (worst-case) AS DIRECT SMOOTH SPEC	0.20	0.25	0.25
Measured (worst-case)	0.04*	0.08*	0.14
REPEATABILITY - Delta between TV Hot & Cold Lin	nite		
AT SPEC	0.20	0.22	0.20
Measured (rms)	0.03	0.02	0.02
AS STORED SPEC	0.30 0.02	0.30	0.30
Measured (rms) AS DIRECT FINE SPEC	0.50	0.50	0.50
Measured (rms)	0.06	0.06*	N/A
AS DIRECT SMOOTH SPEC	2.00 0.08	2.00 0.06*	2.00 0.08
Measured (rms)	0.08	0.00~	0.00
TOTAL -	19	<i>a</i>	
AT SPEC	1.00 0.29	1.30 0.29	1.20 0.23
Calculated	0.29	0.23	0.23
AS STORED SPEC	1.16	1.19	2.29
Calculated	0.31	0.43	0.93
AS DIRECT FINE SPEC	1.34	1.36	2.46
Calculated	0.31	0.46	N/A
	0.01	2.02	2 02
AS DIRECT SMOOTH SPEC Calculated	2.81 0.33	2.82 0.46	3.92 1.13
Calculated	V. U		

Table 2.3-2
ALONG-SCAN GEOMETRIC ACCURACY WITH ENCODER SIMULATOR

	Stored	Direct Fine	Direct Smooth
Repeatability-Spec, mrad	1.0	1.1	2.2
Measured		0.08	0.10
Stability - Spec, mrad	0.50	0.50	0.50
Measured	0.20	0.20*	0.20*
Fixed - Spec, mrad	10.0	10.0	10.0
Measured	0.88	0.88*	0.88*
Total - Spec, mrad	11.1	11.2	12.3
Calculated	1.10	1.10	1.10

<sup>\*</sup>Inferred from stored number

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#### 2.4 RADIOMETRIC ACCURACY

## 2.4.1 T Channel Radiometric Accuracy (3.2.1.1.4.1 a,b,c)

The <u>overall</u> one sigma accuracy of the OLS #12 T Channel DC response is 0.73°K compared to a 1.1°K spec and therefore OLS #12 does meet this specification requirement.

Table 2.4.1-1 presents the overall summary of performance, which is discussed more fully in sections 2.4.1.1, 2.4.1.2 and 2.4.1.3.

ATTACHMENTS:	
Table 2.4.1-1	Overall Contributors
Table 2.4.1-2	210° to 310°K Best Straight Line
	Calibrations
Table 2.4.1-3	T DC Response Compilation of Test Runs
Table 2.4.1-4	BSL Equation T Right, Run #11
Table 2.4.1-5	BSL Equation T Mid, Run #11
Table 2.4.1-6	BSL Equation T Left, Run #11
Figure 2.4.1-1	T DC Response Plots, Run #1 - Primary
Figure 2.4.1-2	T DC Response Plots, Run #1 - Redundant
Figure 2.4.1-3	T DC Response Plots, Run #2 - Primary
Figure 2.4.1-4	T DC Response Plots, Run #2 - Redundant
Figure 2.4.1-5	T DC Response Plots, Run #9 - Primary
Figure 2.4.1-6	T DC Response Plots, Run #9 - Redundant
Figure 2.4.1-7	T DC Response Plots, Run #10 - Primary
Figure 2.4.1-8	T DC Response Plots, Run #10 - Redundant
Figure 2.4.1-9	T DC Response Plots, Run #11 - Primary
Figure 2.4.1-10	T DC Response Plots, Run #11 - Redundant

TABLE 2.4.1-1

OLS #12

OVERALL CONTRIBUTORS TO T-CHANNEL RADIOMETRIC ACCURACY

SPECIFICATION PARA. 3.1.4.1	RMS DEVIATION (°K)	SPECIFICATION MAX ONE SIGMA ERROR (°K)
a) Repeatability (<1 day)	0.262	0.42
b) Stability (>1 day)	0.62	0.80
c) Fixed Deviations	0.29	0.60
TOTAL (RSS) ACCURACY	0.73	1.10

#### Discussion of T DC Response Test and Overview

The measurement of T DC Response is accomplished in the thermal vacuum chamber because the T detector must be cooled to its operating temperature of near 110°K. Two controlled blackbody targets variable in temperature over the dynamic range of 190°K to 310°K provide the absolute infrared radiance reference. The temperature of each target is measured by five thermocouples which have been calibrated against a precision platinum resistance temperature standard. The five thermocouples provide target temperature gradient information to indicate target stability as well as the capability to average the five for the reference target temperature. T Channel DC Response consists of comparing the average target temperature to the channel output voltage at the input to the A/D converters for several target temperatures.

Eleven vacuum runs were made on OLS #12. The T DC Response data from vacuum runs (1 through 11) is compiled in Table 2.4.1-2 and -3, which show the equipment temperature environments and characteristics of each run. completion of testing, the T channel gain pots were readjusted to set  $T_{RGT}$  =  $T_{LET}$ = 4. The column headed "Data Points" indicates how many target temperatures were in that run. The characteristics of the response itself are indicated in three columns each for T LFT, T MID and T RGT. The compared channel response to target temperature results in a difference for each data point. This difference is corrected for M1 Temperature so that all data for a given run reflect the same M1 temperature and the expected shaper circuit difference is subtracted. In this form the difference data for a given run should ideally be a linear function to target temperature. A linear leastsquares fit to the corrected data is used to determine the equation of the best straight line (BSL). In Table 2.4.1-2 the slope error, the 210°K ordinate and the RMS data fit values for these different BSL's are listed in the columns headed Slope, Ordinate at 210°K, and RMS Dev.

In order to distinguish between gain and bias type effects caused by environment, the 310°K value that results when the 210°K BSL value is forced to zero difference is also calculated. Table 2.4.1-3 shows the pre-shaper % gain Difference from Nominal, the Bias Diff. from Nominal (at 190°K) where the pre-shaper Gain is forced to nominal, and the M1 temperature coefficient (K factor); in three columns each for TRGT, TMID and TLEFT.

Tables 2.4.1-4, -5 and -6 show the STS computer processed and gain compensated T DC Response Data of the final "Orbit Nominal" Run (Run #11) for TRGT, TMID, and TLEFT respectively. The fourth line down in the body of the Best Straight Line Equation, "RMS Deviation", is the RMS error (for 210°K to 310°K) of the data points fitted to the best straight line. "FP" is T Fine Primary; "FB" is T Fine, Backup (Redundant); "SP" is T Smoothed, Primary; "SB" is T Smoothed, Backup (redundant). The SP and SB are not used for T Right Only or for T Left Only in the Primary or Redundant normal modes; these are utilized only in the Fallback (slightly degraded) modes of operation. SP and SB are applicable to T Mid in normal Primary or Redundant modes.

Runs 1 and 2 together indicate the changes which accompany operation over the foreoptics cold-to-warm temperature range as indicated by M1 temperature.

Runs 3 and 4 together indicate the magnitude of the variation over the extremes of SSS temperature, (+11° to -3°C); when compared to the +3.2°C and +4.6°C SSS run pairs with the corresponding M1 temperatures, (Runs 2 and 1, respectively). However, changes between these runs are not only due to SSS temperature differences. The PSU, which contains the shaper networks was varied in temperature along with the SSS, from a low of 0.5°C to a high of +38.2°C.

Figures 2.4.1-1 through 2.4.1-10 inclusive show, for Runs No. 1 through No. 11, (respectively), the test data points for Targets 1 and 2 and the BSL

plots for Right, Mid, and Left. (No BSL data plot was obtained for Runs 3 through 8 since they contain only 2 points).

The OLS #12 average M1 coefficient (coupling factor) measured for the final run (#11) was 0.207°K at 210°K scene per 1°C temperature change of M1. The lower the M1 coefficient value, the better the performance. The T Clamp leakage was -0.027% T LEFT and -.104% RIGHT.

The Orbit Nominal BSL differences (from Table 2.4.1-4,5 and 6) between Fine Primary and Fine Backup are small, the largest being 0.38°K for T RGT, at the 310°K end. In the Smooth Primary and Backup modes, T RGT differs by 0.40°K (at 310°K).

The difference between T LEFT and T RIGHT segments calibration (from Tables 2.4.1.3-4 and 5) is 0.56°K worst-case, vs. a spec limit of 1.0°K.

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TABLE 2.4.1-2 0LS #12 210° TO 310°K BEST STRAIGHT LINE CALCULATIONS

	COMMENTS	COLD OPTIC	HOT OPTIC LIMIT	SOAK #1	D SOAK #1	VACUUM BREAK PHT FAILURE	HOT SOAK #1 REPEAT	COLD SOAK #1 REPEAT	HOT SOAK #2	D SOAK #2	HOT LIMIT	COLD LIMIT	NOMINAL
-	<u> </u>			TOH O	COLD	VAC	7.8			COLD			
	RMS	80.	50.	0.	00.		00.	00.	9-	9.	.02	70.	-05
T LEFT	ORD. a 210*	77.	8.	.57	1.01		57.	1.02	27.	1.04	76.	1.11	1.03
,	SLOPE	0011	0134	0176	0042		0208	0032	0214	0043	0197	0111	0148
	RMS	.05	70.	00.	00.		00.	.00	00.	00.	88	20.	20.
T MID	oko. a 210*	.87	1.25	76.	1.40		1.34	1.53	1.37	1.57	1.51	1.54	1.50
	SLOPE	0056	0160	0194	0072		0269	0110	0275	0120	0273	0179	0222
	RMS	60*	.05	.00	00.		8	.00	00.	00.	.04	.02	.05
T RIGHT	ORD. a210*	.16	.43	.12	.63		.50	06.	12.	1.03	62.	96.	76.
	SLOPE	.0030	0063	0096	0008		0199	0066	0223	0088	0211	0138	-,0185
٥٠	PSU	23.8	23.5	38.2	0.5		37.9	9.0	38.0	0.5	33.4	4.8	23.6
TEMPERATURE	H.	-8.1	12.4	15.3	-10.8	H =	16.0	-10.4	15.1	-10.6	12.2	-7.4	-7.8
1	SSS	4.6	3.2	10.9	-2.4		10.8	-3.9	11.0	-2.7	5.6	2.4	4.8
# OF	POINTS	14	60	2	2		2	2	2	2	7	2	18
	=	5	٥	6	13	-	٥	13	٥	₽.	٥	13	13
70	16 16	2/6	2/6	9/9	5/6	1	9/9	9/9	9/9	2/6	2/6	9/9	9/9
	RUN#	-	N	м	7	*.	ıs ,	9	2	80	٥	10	=
	DATE	TDCRM3A 07/08/91	TDCRM3A 07/13/91	T1217231B 07/14/91	T12172318 07/18/91	07/23/91	T12172318 08/25/91	T121T231B 08/26/91	T121T231B 08/28/91	11211231B 08/30/91	TDCRM3C 09/02/91	TDCRM3B 09/08/91	TDCRM3C

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TABLE 2.4.1-3 OLS #12 T DC RESPONSE COMPILATION OF TEST RUNS

		RGT,	RGT/LFT		TEMP	TEMPERATURE .	٠. د		T RIGHT			T MID			T LEFT		
DATE	₩.	7.0	1	# OF DATA POINTS	SSS	E	PSU	% GAIN DIFF. FROM NOM.	BIAS DIFF. FROM NOM.	K FACTOR	% GAIN DIFF. FROM NOM.	BIAS DIFF. FROM NOM.	K FACTOR	% GAIN DIFF. FROM NOM.	BIAS DIFF. FROM NOM.	K FACTOR	COMMENTS
TDCRM3A 07/08/91	-	2/6	13	14	4.6	-8.1	23.8	- 62	.56	.198	70	1.28	.209	72.	.78	500	COLD OPTIC LIMIT
TDCRM3B 07/13/91	2	9/9	6	80	3.2	12.4	23.5	52	.35	198	-1.17	1.22	.209	-1.14	89.	-209	HOT OPTIC LIMIT
T121T231B 07/14/91	м	9/9	0	2	10.9	15.3	38.2	-1.34	99	.198	-2.00	.14	-209	-2.10	65.0-	.209	HOT SOAK #1
11211231B 07/18/91	4	5/6	13	2	. 2.4	-10.8	0.5	.76	1.24	198	.33	2.18	.209	.39	1.64	.209	COLD SOAK #1
07/23/91			=						H			L	- , - 				VACUUM BREAK PMT FAILURE
T12172318 08/25/91	2	9/9	٥	2	10.8	16.0	37.9	-2.51	83	.204	-2.72	.28	.206	-2.40	41	.210	HOT SOAK #1 REPEAT
T12172318 08/26/91	6	9/9	13	2	-3.9	-10.4	9.0	80	1.28	-204	13	2.18	.206	.55	1.71	.210	COLD SOAK #1 REPEAT
11211231B 08/28/91	7	9/9	6	2	11.0	15.1	38.0	-2.67	- 61	-204	-2.79	.29	.206	-2.48	77	.210	HOT SOAK #2
T12172318 08/30/91	8	5/6	13	2	-2.7	-10.6	0.5	- 30	1.37	.204	24	2.18	.206	05-	1.69	.210	COLD SOAK #2
TDCRM38 09/02/91	6	9/9	0	7	5.6	12.2	33.4	-2.41	.15	-204	-2.62	.81	.206	-2.04	.27	.210	HOT LIMIT
TDCRM3B 09/08/91	10	2/6	£	7	2.4	4.7-	4.8	-1.11	88.	.204	-1.15	1.69	.206	56	1.34	.210	COLD LIMIT
TDCRM3C 09/13/91	F	9/9	13	18	4.8	-7.8	23.6	-1.83	. 48	.204	-1.85	1.24	-206	-1.20	.87	.210	NOMINAL

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#### TABLE 2.4.1-4

OLS NUMBER 12

T RGT DATA OF 09/12/91

SSS AT 4.8C

M1 AT -7.8C

PSU TEMP = 23.6C

M1 Coefficient = .204 K/C

T GAIN = 5

T LEVEL = 13

V2 < T Clamp > = 2.05404

**K9 <TL Step Size> = .9237** 

#### BEST STRAIGHT LINE EQUATION

	FP	(A)	FB	SP	(A)	SB
BSL SLOPE	-0.0185	- D	-0.0215	-0.0183	-	-0.0215
BSL AT 190K <k></k>	1.34	(.01)	1.33	1.31	(.02)	1.29
BSL AT 210K <k></k>	0.97	(.07)	0.90	0.94	(80.)	0.86
BSL AT 310K <k></k>	-0.88	(.38)	-1.26	-0.89	(.40)	-1.29
RMS DEVIATION <k></k>	0.05	D) 1 II	0.06	0.06	-	0.07
BSL AT 310K;						
190 AT OV <k></k>	-1.30	15	-1.67	-1.30	-	-1.70
% CHANGE FROM						
NOM GAIN	-1.83	-2	-2.36	-1.83	-	-2.40
BIAS DIFF FROM						
NORMAL 190K <k></k>	0.48	1.00	0.09	0.43	- 1200	0.01

#### TABLE 2.4.1-5

OLS NUMBER 12

T MID DATA OF 09/12/91

SSS AT 4.8C

M1 AT -7.8C

PSU TEMP = 23.6C

M1 Coefficient = .206 K/C

T GAIN = 0

T LEVEL = 13

V2 < T Clamp > = 2.06055

**K9 <TL Step Size> = .9237** 

#### BEST STRAIGHT LINE EQUATION

	FP	(A)	FB	SP	(A)	SB
BSL SLOPE	-0.0222	13/4/40	-0.0244	-0.0224	-	-0.0242
BSL AT 190K <k></k>	1.94	(.09)	1.85	1.93	(80.)	1.85
BSL AT 210K <k></k>	1.50	(.14)	1.36	1.48	(.12)	1.36
BSL AT 310K <k></k>	-0.72	(.36)	-1.08	-0.76	(.30)	-1.06
RMS DEVIATION <k></k>	0.07	-	0.09	0.08	-	0.09
BSL AT 310K;						
190 AT OV <k></k>	-1.31	-	-1.65	-1.35	-	-1.63
% CHANGE FROM						
NOM GAIN	-1.85	-	-2.32	-1.91	-	-2.29
BIAS DIFF FROM						
NORMAL 190K <k></k>	1.24	-	0.79	1.19	- /	0.80

#### TABLE 2.4.1-6

OLS NUMBER 12

T LFT DATA OF 09/12/91

SSS AT 4.9C

M1 AT -7.9C

PSU TEMP = 23.7C

M1 Coefficient = .210 K/C

T GAIN = 6

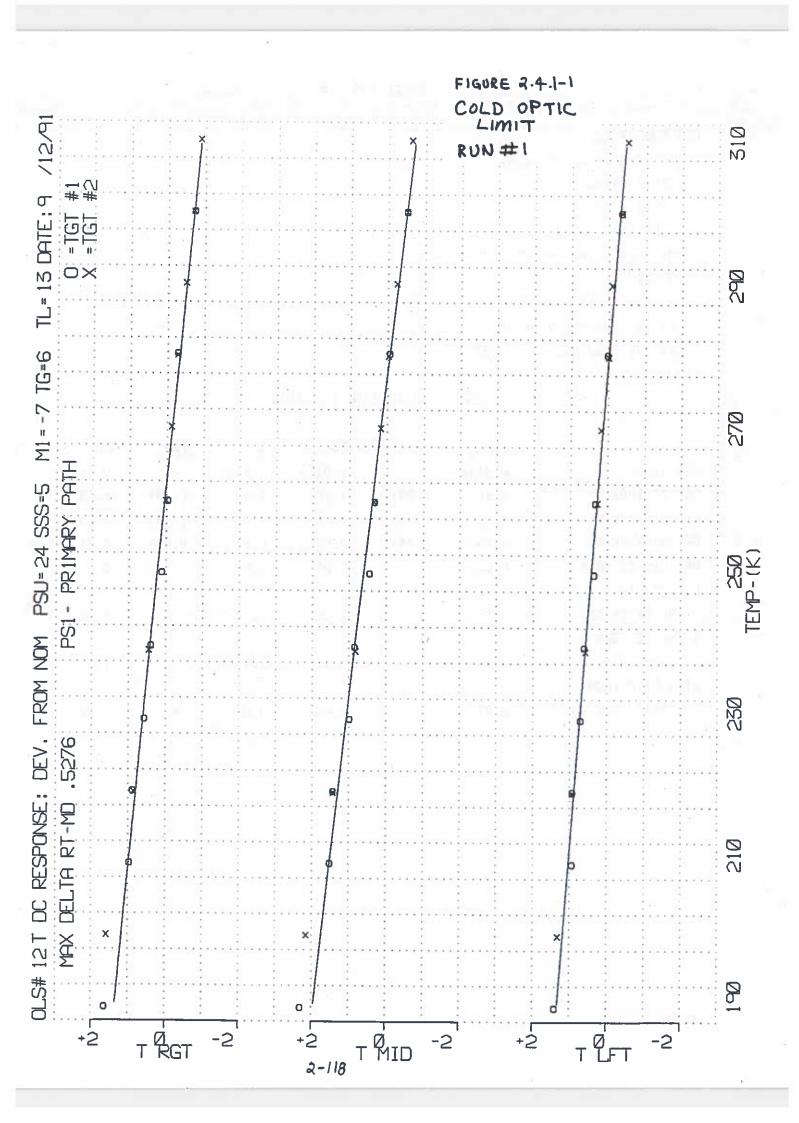
T LEVEL = 13

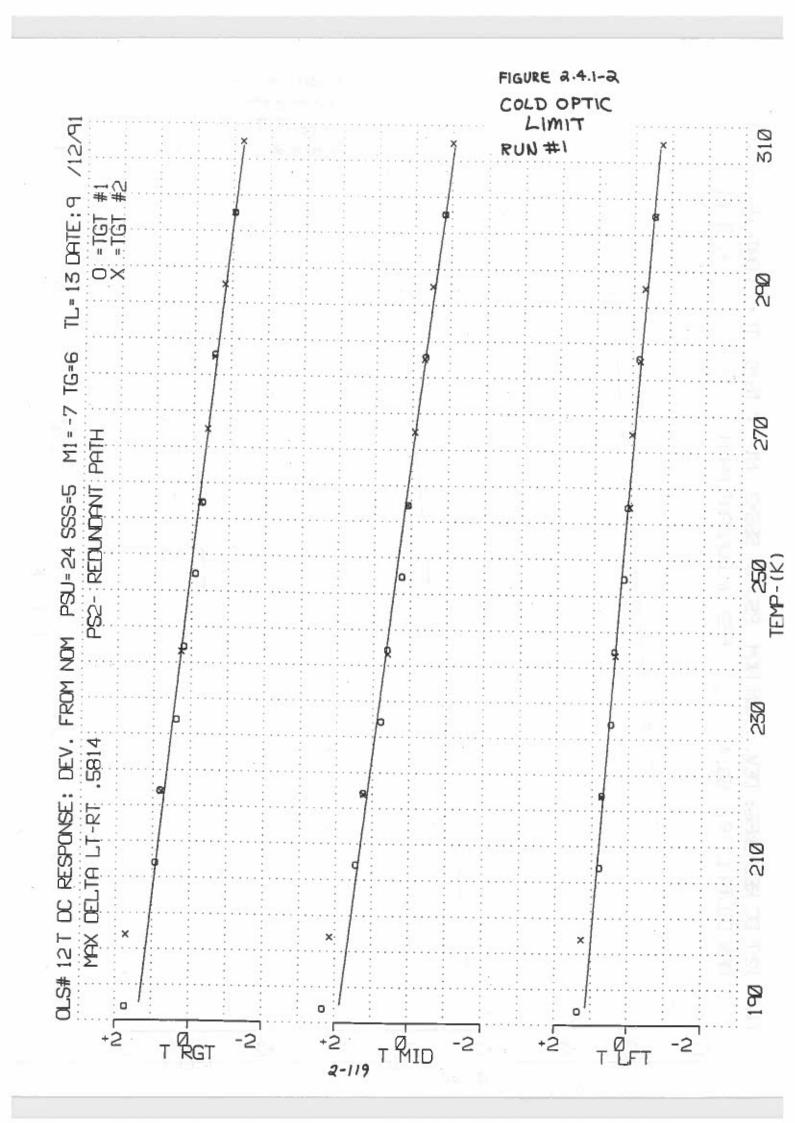
V2 < T Clamp > = 2.06706

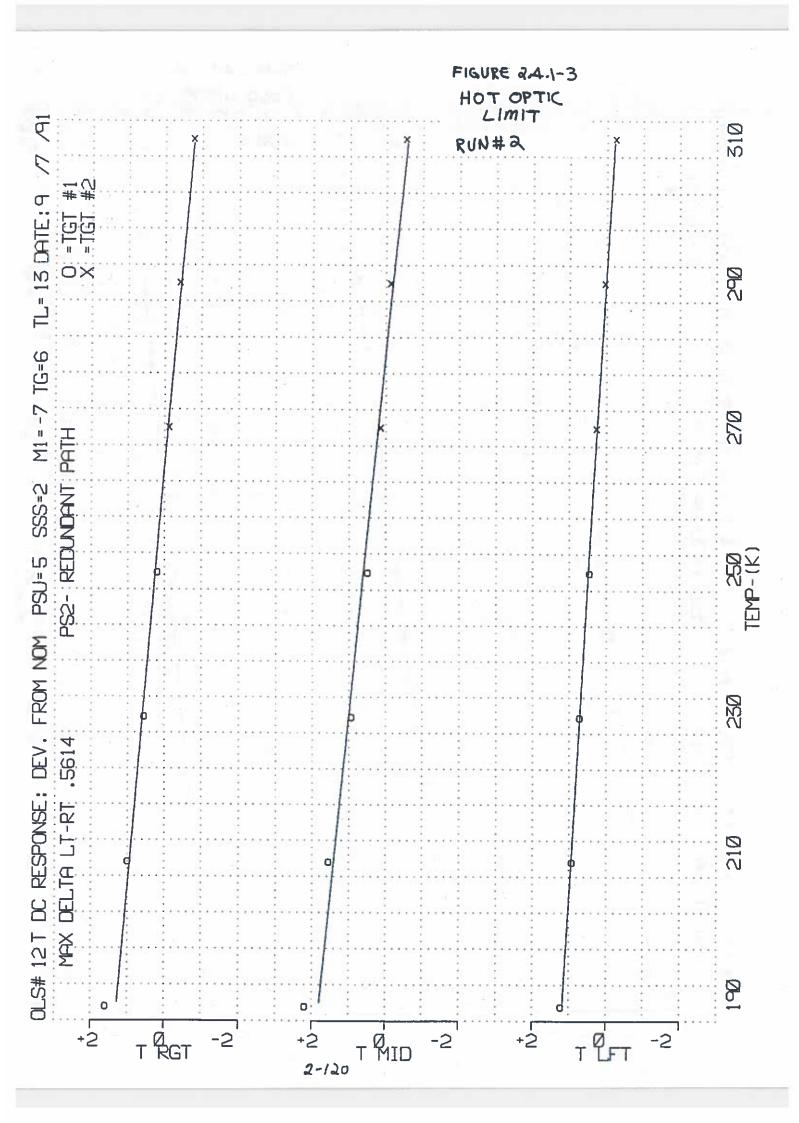
**K9 <TL Step Size> = .9237** 

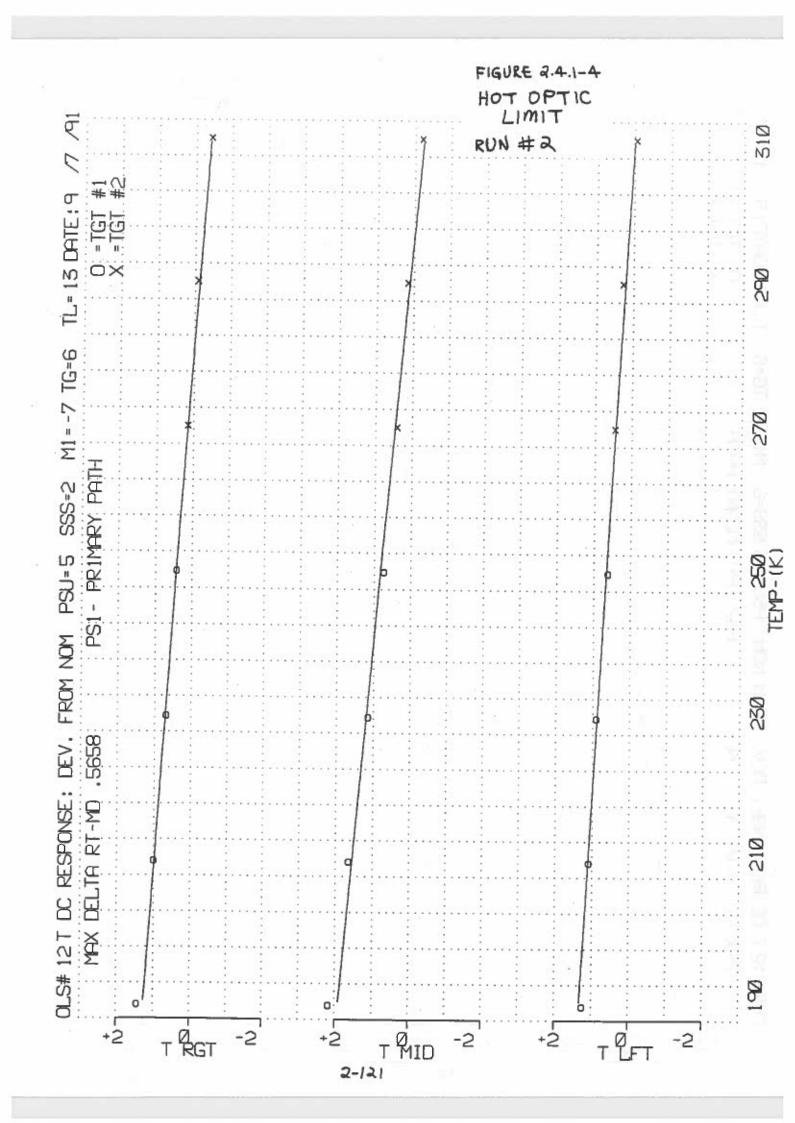
#### BEST STRAIGHT LINE EQUATION

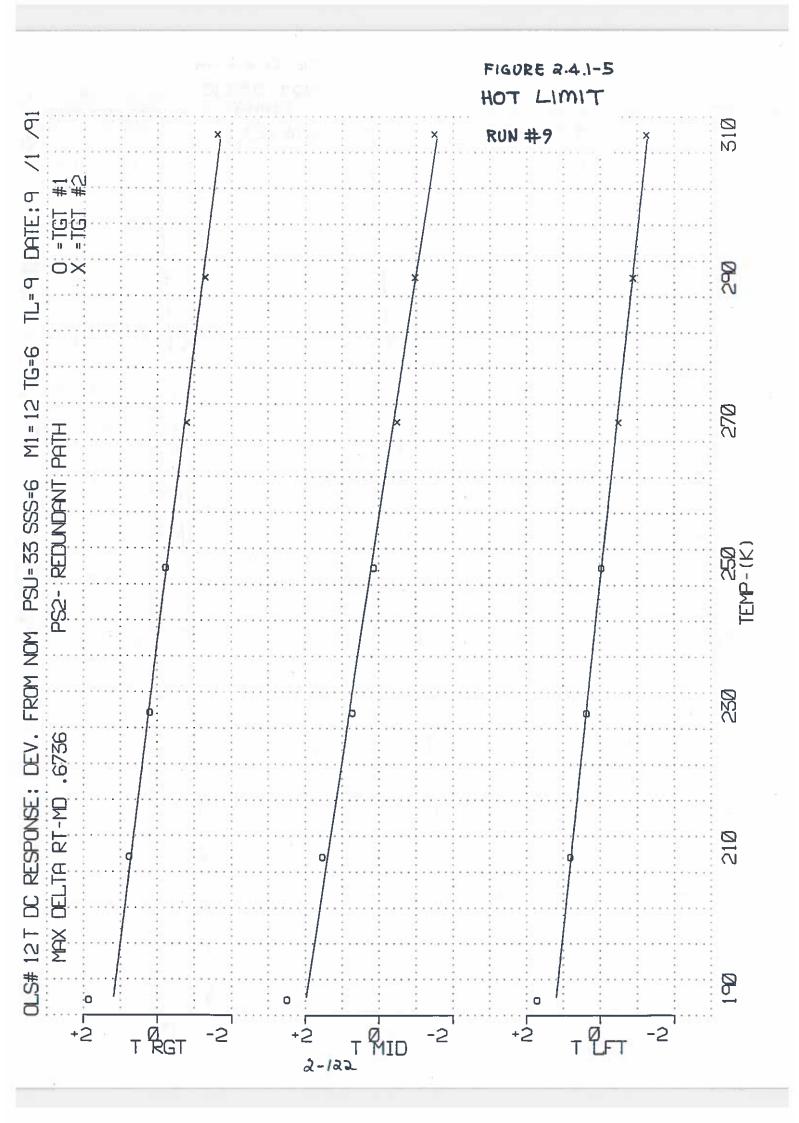
- 19	FP	(A)	FB	SP	(A)	SB
BSL SLOPE	-0.0148	-	-0.0154	-0.0147	-	-0.0154
BSL AT 190K <k></k>	1.32	(.16)	1.16	1.31	(.18)	1.13
BSL AT 210K <k></k>	1.03	(.18)	0.85	1.02	(.19)	0.85
BSL AT 310K <k></k>	-0.45	(.24)	-0.69	-0.45	(.26)	-0.71
RMS DEVIATION <k></k>	0.05	-	0.05	0.05	- 11	0.04
BSL AT 310K;	50 KI					
190 AT OV <k></k>	-0.85	-	-1.05	-0.85	170	-1.06
% CHANGE FROM						1 L
NOM GAIN	-1.20	-	-1.48	-1.19	~	-1.50
BIAS DIFF FROM						
NORMAL 190K <k></k>	0.87	-	0.47	0.87	2	0.42











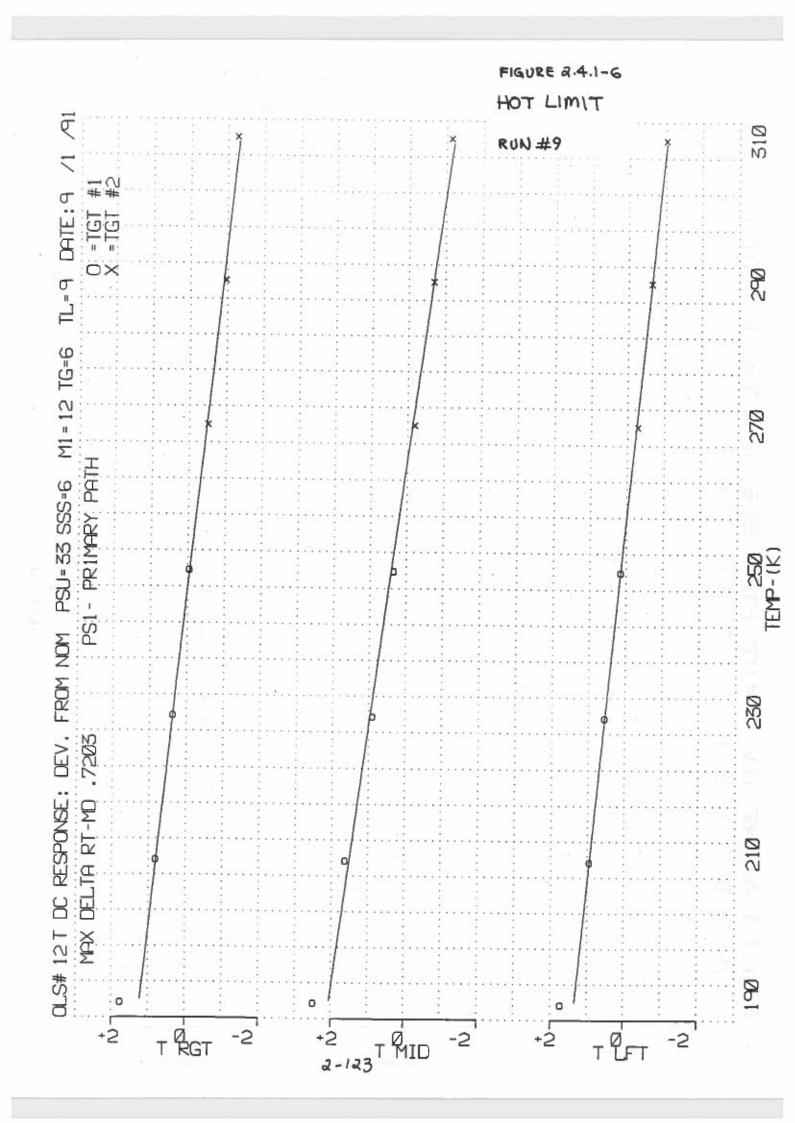
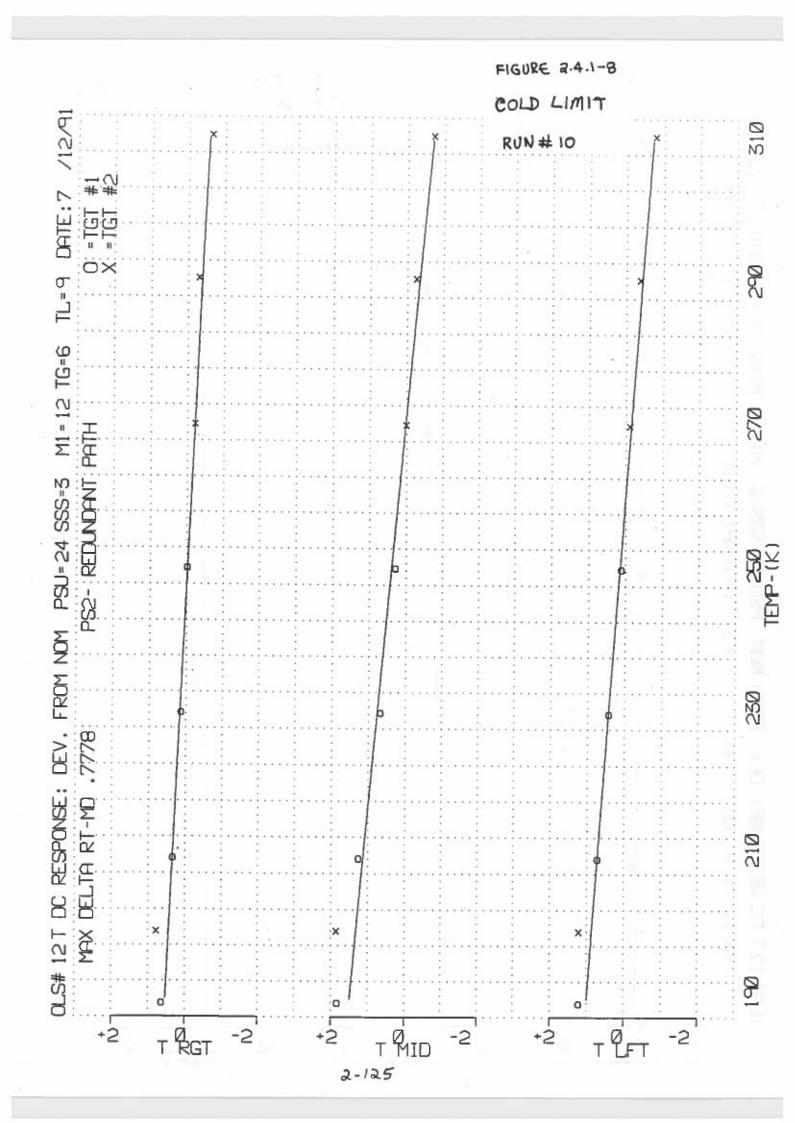


FIGURE a.4.1-7 COLD LIMIT **310** RUN #10 TL=9 DATE:7 0 =TGT X =TGT 290 OLS# 12T DC RESPONSE; DEV, FROM NOM PSU-24 SSS-3 M1=12 TG-6
MAX DELTA RT-MD .8244 PS1- PRIMARY PATH 270 230 210 190 +2 +2 +2 T RGT TET T MID



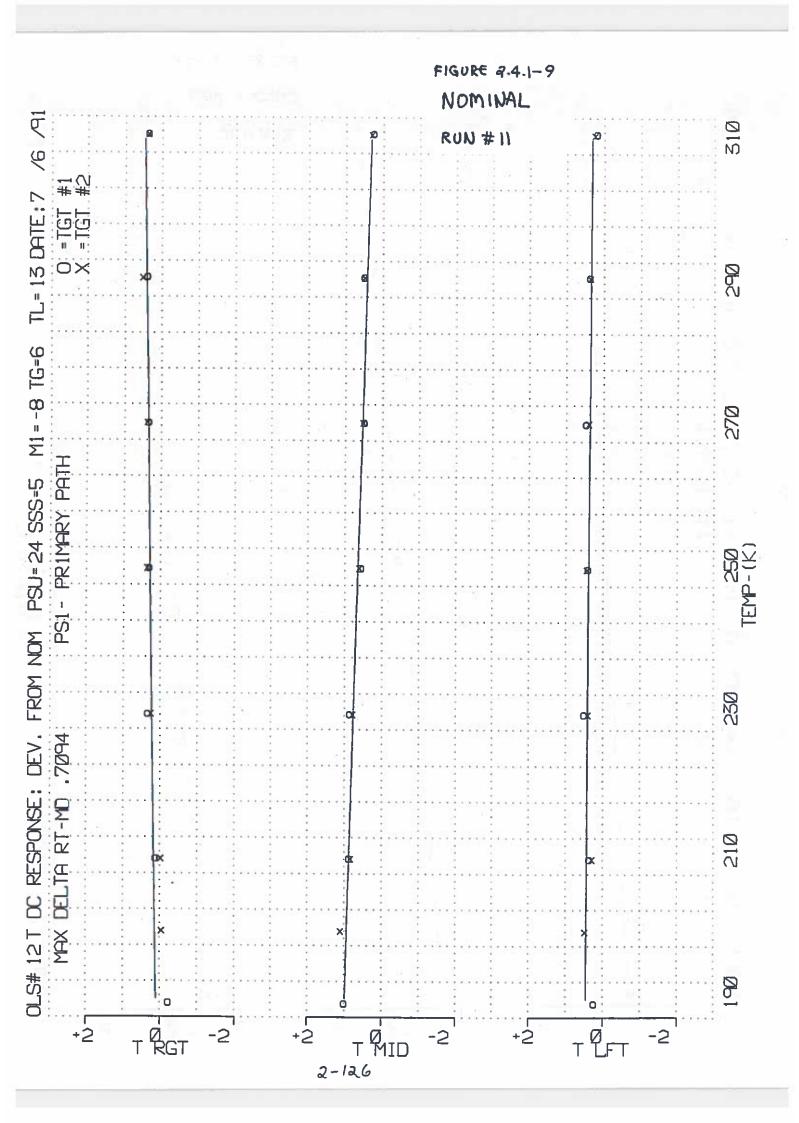


FIGURE 24.1-10 NOMINAL 4 510 RUN # 11 9 TL= 13 DATE: 7 0 = TGT X = TGT 290 M1=-8 TG=6 270 PS2- REDUNDANT PATH OLS# 12 T DC RESPONSE: DEV. FROM NOM PSU=24 SSS=5 JEMP-(K) 230 MAX DELTA RT-MD .6619 190 +5 +5 T RGT T MID 2-127 ÖFT.

#### 2.4 RADIOMETRIC ACCURACY

- 2.4.1 T Channel Radiometric Accuracy (Cont'd)
- 2.4.1.1 Repeatability (3.2.1.1.4.1a)

The 1 sigma Repeatability of T Channel DC Response is 0.262°K compared to a 0.42°K one sigma specification maximum and therefore OLS #12 does meet this specification requirements.

#### **ATTACHMENTS**

Table 2.4.1.1-1 Repeatability Contributors

Table 2.4.1.1-2 Gain and Bias Variations with Temperature Change

Table 2.4.1.1-3 Target Crosstalk, T Clamp Leakage Data

TABLE 2.4.1.1-1 OLS #12 REPEATABILITY CONTRIBUTORS SUMMARY	
ERROR_SOURCE	ONE SIGMA ERROR (K°)
<ol> <li>Diurnal M1 Temperature Change (4°C)</li> <li>A. Quantization of T Level Command</li> <li>B. Inability to Compensate Actual Effect Exactly</li> </ol>	0.19 0.077
2. Temperature Change PSU ± 4.5°C, SSS ± 1°C  A. Effect due to Gain Change B. Effect due to Bias Change	0.066* 0.058*
3. T Clamp Shaper Compensation	0.09
4. T Clamp Leakage	0.104
TOTAL RSS REPEATABILITY ERROR (°K) SPECIFICATION LIMIT, °K, ONE SIGMA  *FROM TEST DATA (REDUCED)	0.262 0.42 MAX.

### Discussion of Repeatability Calculations

- 1. Dinurnal M1 Temperature Change
  - A. The effects of M1 temperature (more properly the foreoptics temperature) are a Repeatability error source. The foreoptics thermal time constant is short enough to permit significant diurnal temperature variations. The ability to compensate for foreoptics temperature using the T Level command greatly reduces this error but does not eliminate it. Although calculations enabling ground compensation smaller than the quantization of the T Level command are possible, it is herein assumed that they will not generally be made. Therefore an error is ascribed due to the T Level quantization as follows:
    - 0.294°K RMS T Level Cmd. Quantization Error at 210°K (1.02° x 1 √ 12) x 0.642 RMS Temperature Linearity Effects over 210-310°K dynamic range = 0.19°K RMS error
  - B. The fact that foreoptics temperature effect cannot be accurately predicted by the single monitor of M1 temperature means that in times of sharp transition the ability to compensate is impaired. It has been assumed that this error may be represented by a 1°C lag in M1 temperature during the 1/3 of the orbit that sharp transistions occur. Therefore the inability to compensate the actual effect is ascribed the following error:

1°C\_\_\_\_ Lag in Ml Temperature

x 1 √ 3 RMS Over total orbit
x 0.207 T Left T Mid T Right average sensitivity coefficient of video at 210K to M1 temperature change for OLS #12 (K

factor)
0.642 Temperature Linearity Effects over dynamic range.

= 0.077°K RMS error

Х

#### Discussion of Repeatability Calculations

Total T channel gain change with temperature may be broken down into two components as follows:

Total Gain 
$$\triangle$$
 = [(PSU  $\triangle$ T) x P<sub>G</sub>] + [SSS  $\triangle$ T) x S<sub>G</sub>] where: P<sub>G</sub> = PSU coefficient of gain, % per °C. S<sub>G</sub> = SSS coefficient of gain, % per °C.

Similarly for bias changes with temperature:

Total Bias 
$$\Delta = [(PSU \Delta T) \times P_B] + [(SSS \Delta T) \times S_B]$$
  
where:  $P_B = PSU$  coefficient of bias, °K per °C.  $S_B = SSS$  coefficient of bias, °K per °C.

Data from two pairs of runs designated A and B, were used to solve these equations simultaneously:

Gain 
$$\triangle$$
, Run  $A = G_A = (T_{PA})(P_G) + (T_{SA})(S_G)$   
Gain  $\triangle$ , Run  $B = G_B = (T_{PB})(P_G) + (T_{SB})(S_G)$   
Bias  $\triangle$ , Run  $A = B_A = (T_{PA})(P_B) + (T_{SA})(G_B)$   
Bias  $\triangle$ , Run  $B = B_B = (T_{PB})(P_B) + (T_{SB})(S_B)$   
where:
$$G = Total \ Gain \ change \ over \ temperature$$

$$B = Total \ Bias \ change \ over \ temperature$$

$$T_P = PSU \ Temperature \ change$$

$$T_S = SSS \ Temperature \ change$$

Solved simultaneously for the temperature sensitivity factors, these equations can be reduced to the following:

$$S_{G} = (T_{PA})(G_{B}) - (T_{PB})(G_{A})$$

$$T_{TPA}(T_{SB}) - (T_{PB})(T_{SA})$$

$$P_{G} = G_{A} - (T_{SA})(S_{G})$$

$$T_{PA}$$

$$S_{B} = (T_{PA})(B_{B}) - (T_{PB})(B_{A})$$

$$T_{PA}(T_{SB}) - (T_{PB})(T_{SA})$$

$$P_{B} = B_{A} - (T_{SA})(S_{B})$$

$$T_{PA}$$

$$2-131$$

2. SSS and PSU Temperature Change: Effect On Gain Change

The effects of SSS and PSU temperature change on gains were determined from parametric analysis of the four runs in Table 2.4.1.1-2. Solving simultaneous equations yielded sensitivity coefficients of gain change for both SSS and PSU temperature change. Temperature data from 5D-2 systems currently on-orbit indicates that the worst case SSS temperature variations are 1°C while worst case PSU temperature variations are 4.5°C. Using the worst-case factors yields:

S<sub>G</sub> = .039% Gain change per degree SSS change x 1°C temperature change x .31°K RMS over 210K to 310K range

 $\times$  1  $\sqrt{3}$  for uniform temperature distribution = .007 deg

P<sub>G</sub> = -.082% Gain change per degree PSU change x 4.5° temperature change x .31°K RMS over 210K to 310K range

 $\times$  1  $\sqrt{3}$  for uniform temperature distribution = -.066 deg

RSS'ing these two contributors yields 0.066 degree total.

3. SSS and PSU Temperature Change, Effect On Bias Change From Table 2.4.1.1-2:

 $S_{\rm B}$  = -.140 deg Bias change per degree SSS change

x 1° temperature change

x 0.642 RMS Temperature Linearization Effects, 210K to 310K

x 1  $\sqrt{3}$  for uniform temperature distribution = -.052 deg

 $P_{\rm B}$  = -.015 deg Bias change per degree PSU change

x 4.5° temperature change

x 0.640 RMS Temperature Linearization Effects, 210K to 310K

 $x 1 \sqrt{3}$  for uniform temperature distribution = -.025 deg

RSS'ing these two contributors yields 0.058 degree total.

#### 4. T Clamp Shaper Compensation

The SSS temperature changes throughout each orbit are expected to cause a one sigma error of 0.146°K at 210K due to the compensation for T clamp temperature variation from 228K to 253K. This times the 0.642 RMS Temperature Linearization Effect over the dynamic range equals 0.09°K RMS error.

#### 5. T Clamp Leakage

An along scan variation (ASV) effect may be caused by some of the scene radiance being viewed at the time of T clamp during the overscan period getting into the reference T Clamp value. This can happen if the T detector sensitivity extends slightly beyond the boundaries of M4' during the clamp time. The test performed for leakage is to view one target at 210°K during active scan while the other target (which is at the T Clamp angle) is varied over the 210° to 310°K dynamic range. This data is presented in Table 2.4.1.1-3. The effect on the response to the active scan target is attributed to the T clamp optical leakage.

Using the OLS #12 T data from Table 2.4.1.1-3, the T clamp leakage contribution can be calculated as follows:

T clamp leakage ratio (LR) = 
$$100\% \times \Delta T \times \frac{\frac{\partial P}{\Delta T}}{\Delta P}$$
  
=  $\Delta T \times .50552\%$ 

This calculation is performed in the MODE 4 data reduction of T121T221S. The ratio calculated is:

The peak error from T clamp leakage (due to the 310° background) can be calculated as follows:

peak error = LR x 
$$\triangle N$$
 x  $(\frac{\partial P^{-1}}{\triangle T_{210}})$  -1

$$= \Delta T \times \frac{\Delta N}{\Delta P} \times \frac{\frac{\partial P}{\Delta T}}{\frac{\partial P}{\partial P}}$$

$$= \Delta T \times 0.8156 \times 1.0788$$

$$= \Delta T \times 0.880$$

where:

ΔP = Difference in radiance between 210° and 310°K

$$= 14.395E-4 \text{ w cm}^{-2} \text{ sr}^{-1}$$

and:  $\Delta N$  = Difference in radiance between 240° and 310°K

$$\frac{\partial P}{\Delta T}$$
 = slope of radiance curve at 210°K = 6.7452 E-6

$$\frac{\partial P}{\Delta T}$$
 = slope of radiance curve at 214°K = 7.277 E-6

ΔT = measured change in response to 210° target as the background is varied from 210° to 310°K.

RMS ERROR = PEAK ERROR

x 0.7605 for RMS distribution of leakage radiance over dynamic

range.

x 0.642 RMS Temperature Linearization Effect

FROM MODE 4 Data reduction:

The worst-case contribution to repeatability error by T-clamp leakage is therefore  $-0.104\,^{\circ}\text{K}$  RMS.

GAIN AND BIAS VARIATIONS WITH TEMPERATURE CHANGE (M1 TEMP CORRECTED TO +12.C OR -8.C)

				Ь	T RGT	I	T MID	T LFT	LFT
		SSS TEMP	PSU TEMP	% GAIN DELTA (%)	BIAS CHG. @ 190*K (*K)	% GAIN DELTA (%)	BIAS CHG. @ 190°K (*K)	% GAIN DELTA (%)	BIAS CHG. @ 190°K (°K)
	RUN 11 (NOMINAL TEST)	4.8	23.6	-1.83	0.48	-1.85	1.24	-1.20	.87
M1 = -8°C (Run A)	RUN 8 (COLD SOAK)	-2.7	0.5	30	1.37	24	2.18	.40	1.69
	RUN 11- RUN 8	7.5 T <sub>sA</sub>	23.1 T <sub>PA</sub>	-1.53 G <sub>A</sub>	89 8	-1.61 G <sub>A</sub>	94 B <sub>A</sub>	-1.60 G <sub>A</sub>	82 B <sub>A</sub>
	RUN 7 (HOT SOAK #2)	11.0	38.0	-2.67	61	-2.79	.29	-2.48	44
M1=+12°C (Run B)	RUN 9 (HOT LIMIT)	5.6	33.4	-2.41	15	-2.62	.81	-2.04	.27
<b>.</b>	RUN 7- RUN 9	5.4 T <sub>SB</sub>	4.6 T <sub>PB</sub>	26 G <sub>B</sub>	46 B <sub>B</sub>	17 18	52 B <sub>B</sub>	44 G <sub>B</sub>	71 B <sub>B</sub>
Calcu Sensi Facto	Calculated Sensitivity Factors	SSS:	Ss. %%°C)	011	072	.039*	085	031	140*
*LINDCT CAC	CACE WAITIES								

\*WORST CASE VALUES

2-135

TABLE 2.4.1.1-3
OLS #12
TARGET CROSSTALK, T CLAMP LEAKAGE DATA\*

SSS = +5° M1 = -8°

	T RIGHT	T MID	T CPL	T CPR	T LEFT	#
T1 210° [T2 @ 310°] (T121T231G)	0.32	1.00	0.08	0.75	0.57	07-06-91
Difference, AT	0.25	0.12	0.03	0.07	0.08	- 15.
T1 210° [T2 @ 210°] (T121T221S)	0.07	0.88	1.05	0.62	0.45	07-07-91
T2 210° [T1 @ 310°] (T131T221A)	0.30	1.12			0.55	07-07-91
Difference, △T	0.14	0.21	1-618		0.13	
T2 210° [T1 @ 210°] (T121T221S)	0.16	0.91			0.42	07-07-91
Worst Case Data From T121T221S.ST Mode 4 Data Reduction:	*					74
T clamp leakage r Peak leakage erro RMS leakage error	r at 210	°K is K is	-0.032% -0.055K -0.027K	-0.122% -0.213K -0.104K		

<sup>\*</sup>Data is FP Deviation in \*K

#### 2.4 Radiometric Accuracy

- 2.4.1 <u>T Channel Radiometric Accuracy</u> (Cont'd)
- 2.4.1.2 Stability (3.2.1.1.4.1b)

The T Channel Radiometric Accuracy (Stability) analysis in the OLS 5D-3 System Summary Report Paragraph 3.5.1.2, predicts 0.62°K 1 sigma error in stability compared to the 0.8°K maximum specification requirement. This calculation is applicable to all 5D-3 systems with 190-310°K total range.

#### **ATTACHMENTS**

Table 2.4.1.2-1 Stability Contributors Summary

Table 2.4.1.2-2 Change in BSL 210°, 310°K Points Between Runs

Table 2.4.1.2-3 Change in 210°, 310° Output Deviation From

Nominal (°K) between Power Supply 1 and Power

Supply 2

## TABLE 2.4.1.2-1

## STABILITY CONTRIBUTORS SUMMARY

			ONE SIGMA ERROR (°K)
1.	Shaped Bias		
	a) Open Loop Mirror Emissivity	0.1	
	b) T Clamp Shaper Compensation - Temperature	0.23	
	- Age	0.17	
	RSS Total	0.30°K	
	X RMS Temperature Linearization Effect = RMS Shaped Bias Errors	0.642	0.19°K
2.	Bias des automorphism de la companie	3112 10	
	a) Preshaper Gain - Inner Stage Temperature	0.28	
	- Bias Current	0.24	
	- Amplifiers	0.22	
	b) Post Shaper DC Drift	0.12	
	RSS Total = RMS Bias Error		0.45°K
3.	<u>Gain</u>		
	Postshaper Gain Changes - Amplifier		
	over the 210-310K range, °K RMS Error		0.38
	TOTAL RSS Stability Error (Total Dynamic Range)		0.62
	Stability Error Specification (°K, 1 Sigma)		0.80 Maximum

#### Discussion of Stability Errors

The experimentally derived RMS change of the BSL(s) between runs was calculated to be 0.04°K, 0.04°k and 0.06°K for TRGT, TMID and TLEFT respectively. The two runs used were Run #6 and Run #8. The results verify the analytical estimate of the stability over time intervals greater than one day. This data is tabulated in Table 2.4.1.2-2.

As an additional check of stability, the Fine-Primary outputs of the three segments as a deviation from nominal (°K) at 210° and 310° were compared using power supply 1 data of TDCRM3C.ST and power supply 2 data of 6X2X3A.ST. (Both from Run #11). This data is tabulated in Table 2.4.1.2-3. The deltas were calculated and RMS'd over the temperature range. The results are comparable to the "Change Between Runs" data.

TABLE 2.4.1.2-2

OLS #12

## CHANGE IN BSL 210, 310K POINTS BETWEEN RUNS

SSS = +3°C, M1 = -8°C

	TG I	TL	T RGT		T MID		T LFT	
	R/L		210	310	210	310	210	310
T121T231B 08-26-91	5/6	13	.90	.25	1.53	0.42	1.02	0.70
T121T231B 08-30-91	5/6	13	1.03	.15	1.57	0.36	1.04	0.61
Change Between Runs			0.13	0.10	0.04	0.06	0.02	0.09
RMS Change	MS Change		0.12		0.05		0.07	

TABLE 2.4.1.2-3

## OLS #12

# T CHANNEL DC RESPONSE

## DIFFERENCE BETWEEN POWER SUPPLIES 1 and 2

From Orbit Nominal (Run #11), SSS = +5°C, M1 = -8°C

	RIC	GHT	M	ID	LEFT		
d lm 'rann	TGT-1 210°K	TGT-2 310°K	TGT-1 210°	TGT-2 310°	TGT-1 210°	TGT-2 310°	
FP DEV [K] Power Supply 1 TDCRM3C.ST 09/13/91	0.97	-0.87	1.51	-0.62	0.94	-0.49	
FP DEV [K] Power Supply 2 6X2X3A.ST	0.98	-0.93	1.56	-0.65	1.00	-0.55	
Change *K	0.01	0.06	0.05	0.03	0.06	0.06	
RMS °K	0.0	04	0	.04	0.06		

#### 2.4 RADIOMETRIC ACCURACY

## 2.4.1 T Channel Radiometric Accuracy (Cont'd)

## 2.4.1.3 <u>Fixed Deviations</u> (3.2.1.1.4.lc)

The Fixed deviations for OLS #12 are 0.29°K, 1 sigma, compared to the 0.6°K specification maximum. The calibrateable portion of the fixed deviations is 0.17°K RMS compared to the 0.4°K RMS specification maximum. The Fixed deviation calibration for separate detector segments is 0.82°K (worst case) compared to the 1°K spec. maximum. The maximum along scan variation was 0.14°K RMS for TF (Right) and 0.12°K RMS for TS compared to the 0.2°K RMS specification maximum.

#### **ATTACHMENTS**

Table 2.4.1.3-1	Fixed Deviations Contributors
Table 2.4.1.3-2	T Shaper Error Tabulation
Table 2.4.1.3-3	Target Deviation from Mean of Both Targets
Table 2.4.1.3-4	BSL Calibration Equations
Table 2.4.1.3-5	Fixed Deviation Calibration Differences for
	Separate Segments
Table 2.4.1.3-6	Along Scan Variation (265° to 310°K) within a
	Separate Segments
Table 2.4.1.3-7	Cone (Inner Stage) Patch Temp EST
Table 2.4.1.3-8	Cone Cooler Outer Stage Temp EST
Figure 2.4.1.3-1	5D3 Nominal Shaper Curve
Figure 2.4.1.3-2	Along Scan Variation, T Right, M1 = 12°C
Figure 2.4.1.3-3	Along Scan Variation, T Mid, M1 = 12°C
Figure 2.4.1.3-4	Along Scan Variation, T Left, Ml = 12°C
Figure 2.4.1.3-5	Along Scan Variation, T Right, Ml = -8°C
Figure 2.4.1.3-6	Along Scan Variation, T Mid, M1 = -8°C
Figure 2.4.1.3-7	Along Scan Variation, T Left, M1 = -8°C

Figure 2.4.1.3-8 Along Scan Variation, T Auto M1 = 12°C Figure 2.4.1.3-9 Along Scan Variation, T Auto M1 = -8°C

### OLS #12

# FIXED DEVIATION CONTRIBUTORS

DEV	IATION SOURCE		ONE SIGMA ERROR (°K)	
1.	Foreoptics Mirror Emissivity		0.13*	
2.	T Clamp Shaper Compensation		0.09	
3.	Transfer Function			
	A. Non-Linearity		0.17* 0.4°K Spec M	ax
	B. Shaper Components Variation		0.10	
	C. Detector Spectrum Variation (included in 3A	.)		
4.	Test Targets		1	
	A. Temperature		0.10	
	B. Emissivity		0.10	
	C. Repeatability		0.04*	
TOT	AL (RSS) FIXED DEVIATION		0.29	
FIX	ED DEV. SPECIFICATION LIMIT, °K ONE SIGMA		0.60 Maximum	
* F	ROM TEST DATA ANALYSIS			
		DATA	SPEC MAX	
5.	Fixed Deviation BSL Calibrations Match for Separate Segments (Worst Case)	0.82	1.°K	
6.	Along Scan Varations within a segment (265° to 310°K) Worst Case	0.14K R	1S 0.2°K RMS	

# Discussion of Fixed Deviation Tests and Calculations

# 1. Foreoptics Mirror Emissivity

The foreoptics mirror emissivity is a source of Fixed deviations as well as of Repeatability and Stability deviations. The correction in operation is made on MI temperature only; whereas the entire foreoptics causes the offset phenomena. The ground calibration is made in the thermal vacuum chamber, where MI is cooled radiatively via a cold tunnel, which fills the MI view beyond the optical field of view of the system. Thus the foreoptics temperature distribution in the chamber differs somewhat from operation in space, especially in that M2 is colder in the chamber. It is not possible to determine accurately this effect based upon present knowledge. It is therefore assumed that it is equivalent to a 1°C difference in MI temperature, or about 0.207at 210° using the actual OLS #12 MI coefficient (K factor). The RMS Temperature Linearization Effect, 0.642, transforms this to a 0.13°K RMS contribution to fixed deviation.

# 2. T Clamp Shaper Compensation

The T Clamp shaper compensation contribution arises as follows. If the T Clamp emissivity were initially only 0.98 rather than the 0.995 used to calculate the compensation for T Clamp temperature, the error at 242°K would be 0.70°K. Although this error would be compensated for when the T channel adjustments were made, a change in T Clamp temperature to 256°K or 230°K would result in an error of 0.78°K, producing an uncompensated error of 0.08°K. Over the dynamic range this is equivalent to 0.09°K RMS.

- Transfer Function
- A. The departure of the T channel radiometric transfer function from a linear relationship is not an error as such because it is known and compensation can be made for it. However, this type of deviation is included within the constraints of the Fixed deviation portion of the T channel radiometric accuracy spec. The nominal T Channel non-linear transfer function (shaper) error is tabulated in Table 2.4.1.3-2 and plotted in Figure 2.4.1.3-1. The nominal shaper error is 0.15°K RMS. This calculation is made with the 5D-2 shaper, which is also used on OLS #12. The worst-case reduced test data (from Tables 2.4.1-4,5 & 6) RMS Deviations of the points from the BSL for OLS #12, are 0.07°K for T Right (Smooth Backup), 0.09°K for T MID (Fine and Smooth Backup) and 0.05°K for T Left (Fine Primary & Smooth Backup). The analytic value, (0.15°K RMS) and the worst-case test value of 0.09°K are RSS'ed to become 0.17°K RMS for this Fixed deviation source versus the 0.4°K maximum spec allowance for the calibrateable portion of the Fixed deviations.
- B. Departure of shaper components from design nominal values may cause additional peak errors of 0.25°K and are assigned a one sigma error of 0.1°K. These deviations would not be included in the measured deviation, because the data points are not close enough together to adequately detect them.
- C. The T detector spectrum difference from unit to unit is included in the calibration data and is therefore included in 3A, the non-linearity of transfer function.

### 4. Test Targets

The deviation of the reference test target indicated blackbody temperature from absolute is the result of contributions from three Fixed deviation sources.

- A. The measurement of the averaged target temperature using the 5 thermocouples immersed in the target baseplate is subject to the accuracy of the PQL/Block V Thermocouple Aquisition and Control System calibrated per procedure MCS0116801B. A review of the calibration procedure and the equipment used has led to ascribing 0.1°K RMS to these sources.
- B. The actual effective blackbody temperature of the target also deviates from that measured because of differing emissivity from that assumed and because the "true" radient temperature differs from the measured temperature using the thermocouples. The effective emissivity is a combination of true emissivity (which is better than 0.996 according to Eppley) and reflectance of up to 0.004. An average emissivity of 0.998 is corrected for along with the thermocouple calibration. This source of deviation has been assigned 0.1°K RMS.
- C. Actual measurements with OLS #12 system of the two Eppley IR Reference test targets over the 210° to 310°K range reveal small differences between the targets. These differences, as tabulated in Table 2.4.1.3-3, represent the target differences from the mean of both targets for T LFT and T RGT averaged from file TDCRM3C.ST (Run #11). The eleven differences when RMS'ed yield 0.04°K RMS deviation for this source.

### Fixed Deviation Calibrations for Separate Segments

The calibrations are represented by the best straight line (BSL) equations for the separate segments. The BSL deviation expressions (from ideal), in y = mx + b form for the segments are tabulated in Table 2.4.1.3-4. In T Smooth the Right and Left detector segments are averaged and used across the entire scan line.

In order to determine the differences in calibrations for separate segments, (for comparison to the specification) the T Fine BSL deviation equations of Table 2.4.1.3-4 were evaluated at 210°K and 310°K. The 3 possible segment differences were then calculated. Also, the worst-case deviations between segments were taken from the Best Straight Line Plots (Figures 2.4.1-1 thru 2.4.1-12). These results are tabulated in Table 2.4.1.3-5. The calibration differences for separate segments are within the 1°K maximum spec throughout the dynamic range of 210 to 310K for OLS #12.

### Along Scan Variation (265°K to 310°K) Within a Segment

The Along Scan Variation (ASV) in T DC Response is considered to be a Fixed deviation, since as a function of scan angle it does not vary as a function of time per se. Figures 2.4.1.3-2, 3, 4, 5, 6, 7, 8 and 9 show the deviations along-scan vs. surface distance (nmi/100) for T RGT, T MID, TLFT, and T AUTO. The vertical scale factor is  $1.0^{\circ}$ K per cm. Four figures are for M1 = -8°c and four are for M1 = +12°C. The dip in the plots of +5, -8 data for the 290°K target at -550nmi surface distance is due to the inadvertent deletion of a data file. The slight sawtooth effect on the 290°K plots is a result of the missing data's effect on the processing of the data. All data was within spec. The computer printed number to the right of each curve is the computed RMS deviation in millidegrees K for the associated ASV plot. The

RMS ASV values are only printed for the target temperatures above 265°K, i.e., the 270°, 290°, and 310°K plots.

The T DC response BSL calibrations are taken using test scan angles of -50° (-600nmi) for T LFT, +50° (+600nmi) for T RGT, and +0° (0 nmi) for T MID.

The OLS #12 has some ASV, but is within spec. The worst case (max) ASV RMS value within a segment for OLS #12 was 0.14°K and is entered in Table 2.4.1.3-6 to compare with the specification limit.

# T SHAPER ERROR LIST

The 190 to 310°K T Shaper used for OLS #12 and up has 6 straight line segments of decreasing slope and 5 (inflections) diode break points. The significant features in the shaper curve are tabulated below:

(°K)	411	ERROR (°K)	<u>REMARKS</u>			
190 195.5		0 -0.35		adjusted to		Curve (Smooth) Curve
201.5 205		0 +0.16	lst diode	140		3
209 214 219.5		-0.215 0	2nd slope	e is parallel	to Radiance	(Smooth) Curve
223.5 228		+0.215	2nd diode	cut-in		
235.5 242		-0.28 0	3rd slope	e is parallel	to Radiance	Curve
246.5 252		+0.19 0	3rd diode			
258 264.5		-0.205 0		e is parallel	to Radiance	Curve
269.5 275.5		+0.23	4th diode		to Dodinso	Summer :
282 285.5 294		-0.16 0 +0.205	5th diode	e is parallel	to Radiance	curve
301 306		0 -0.06		is parallel	to Radiance	Curve
310		0		adjusted to		

The largest plus and minus errors in the 210K-310°K range are +0.23° and -0.28°K respectively.

The standard deviation = 0.15°K RMS over the 210° to 310°K dynamic range.

TABLE 2.4.1.3-3

OLS #12

# TARGET DEVIATION FROM MEAN OF BOTH TARGETS

TARGET TEMP (°K)	DEVIATION (°K)
210	-0.01
220	-0.07
230	-0.02
240	-0.01
250	0.03
260	0.01
270	0.02
280	-0.01
290	0.04
300	0.02
310	0.09

$$RMS = \sqrt{\frac{\sum (Dev. °K)^2}{11}} = 0.039°K$$

# OLS #12

### **BSL\_CALIBRATION EQUATIONS**

(From Tables 2.4.1-4,5,6)

							EVA	LUATED
T FINE (Pri	mary) =						<u>@ 210°</u>	<u>@ 310°</u>
T-Right:	Error =	-0.0185	(T-190)	+ 1.34	(	°K)	+.970	880
T-Mid:	Error =	-0.0222	(T-190)	+ 1.94	(	°K)	+1.496	724
T-Left:	Error =	-0.0148	(T-190)	+ 1.32	, : (	°K)	+1.024	456
		45						
T FINE (Red	lundant)							
T-Right:	Error =	-0.0215	(T-190)	+ 1.33	(	°K)	+.900	-1.250
T-Mid:	Error =	-0.0244	(T-190)	+ 1.85	(	°K)	1.362	-1.078
T-Left:	Error =	-0.0154	(T-190)	+ 1.16	(	°K)	+.852	688
T SMOOTH (Primary -		-0.0224	(T-190)	+ 1.93	(	°K)		
					•	- N		
T SMOOTH (Redundant			(T-190)	+ 1.85	(	°K)		

TABLE 2.4.1.3-5

OLS #12

# FIXED DEVIATION CALIBRATION DIFFERENCES FOR SEPARATE SEGMENTS

Calculated from Run #12 BSL's in Table 2.4.1.3-4:

	DIFFERENCE AT 210°K (°K)	DIFFERENCE AT 310K (*K)	SPECIFICATION (MAX)
PRIMARY		11 27	
T Mid to T Right	0.53	0.16	1•K
T Mid to T Left	0.47	0.27	1°K
T Right to T Left	0.05	0.42	1°K
REDUNDANT		ر خور طرعا	
T Mid to T Right	0.46	0.17	1°K
T Mid to T Left	0.51	0.39	1°K
T Right to T Left	0.05	0.56*	1°K

Worst Case Differences from Best Straight Line Plots (Figures 2.4.1-1 thru 2.4.1-12):

RUN #		PRIMA	ARY PATH		REDUND	ANT PATH	SPI	EC. MAX.
1		0.71°	RGT-MID		0.66°	RGT-MID		1°K
2	*	0.82°	RGT-MID	*	0.78°	RGT-MID		1°K
9		0.72°	RGT-MID		0.67°	RGT-MID		1°K
10		0.57°	RGT-MID		0.56°	RGT-MID		1°K
11		0.53°	RGT-MID		0.58°	RGT-MID		1°K

\*WORST-CASE DATA

# OLS #12

# ALONG SCAN VARIATION (265°K to 310°K) WITHIN A SEGMENT

(From ASV Graphs)

	ONE SIGMA	SPEC
T-FINE	ERROR (°K RMS)	LIMIT (°K RMS)
T-Left Segment	0.121	0.2
T-Mid (Sum) Segment	0.117	0.2
T-Right Segment	0.135	0.2
T-SMOOTH		
T-Sum	0.117	0.2

# CONE COOLER S/N 024

OLS-12

# CONE (INNER STAGE) PATCH TEMP. EST

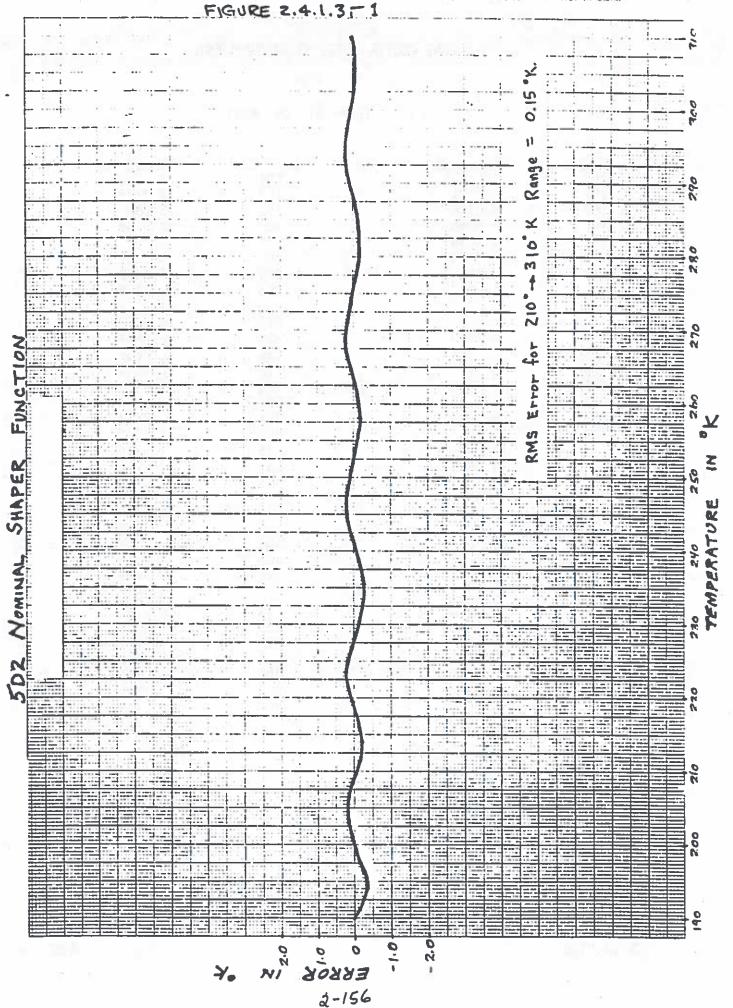
TEMPERATURE *K		PATCH EST,	VOLTS
95		5.655	
96	hart Frederick	5.248	
97		4.874	
98		4.529	
99		4.212	
100		3.920	
101		3.651	
102		3.403	
103		3.174	
104		2.963	
105		2.768	
		2.588	
106		2.422	
107			
108		2.268	
109		2.125	
110		1.993	
111		1.871	
112		1.757	
113		1.651	
114		1.553	
115		1.462	
116		1.377	
117		1.298	
118		1.225	
119		1.156	
120		1.092	
	9	1.022	
121			
122		.976	0.04
123		.924	
124		.875	
125		.829	

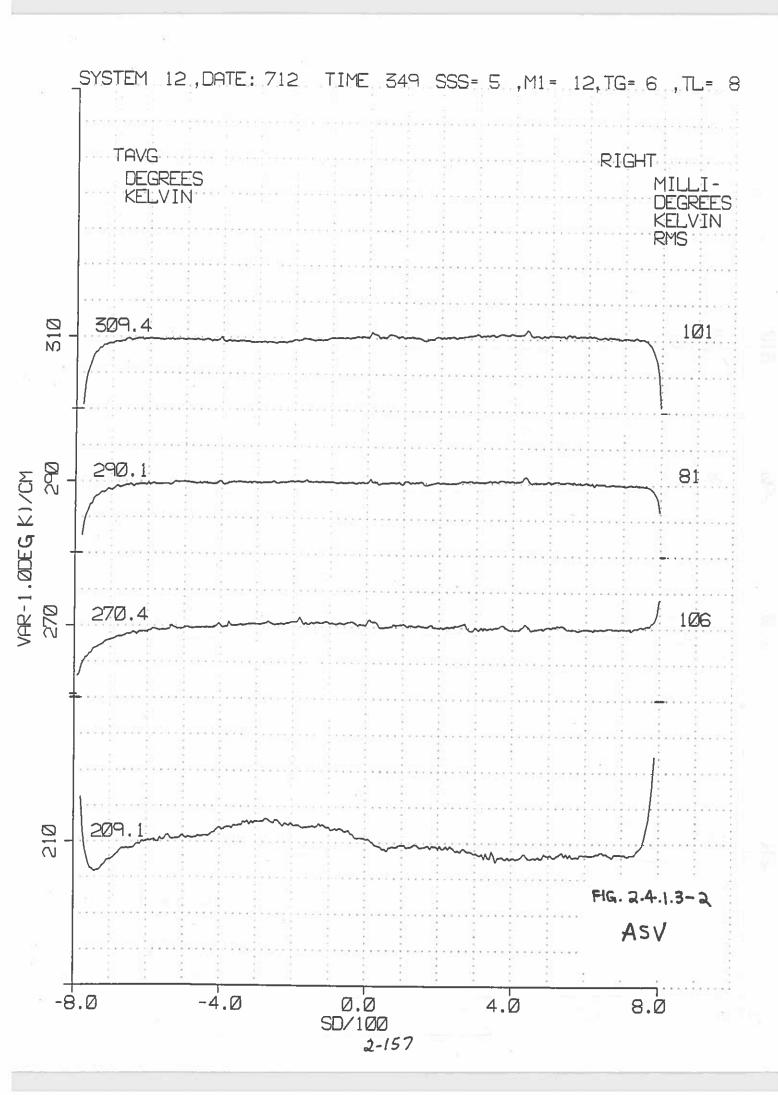
# CONE COOLER OUTER STAGE TEMP EST

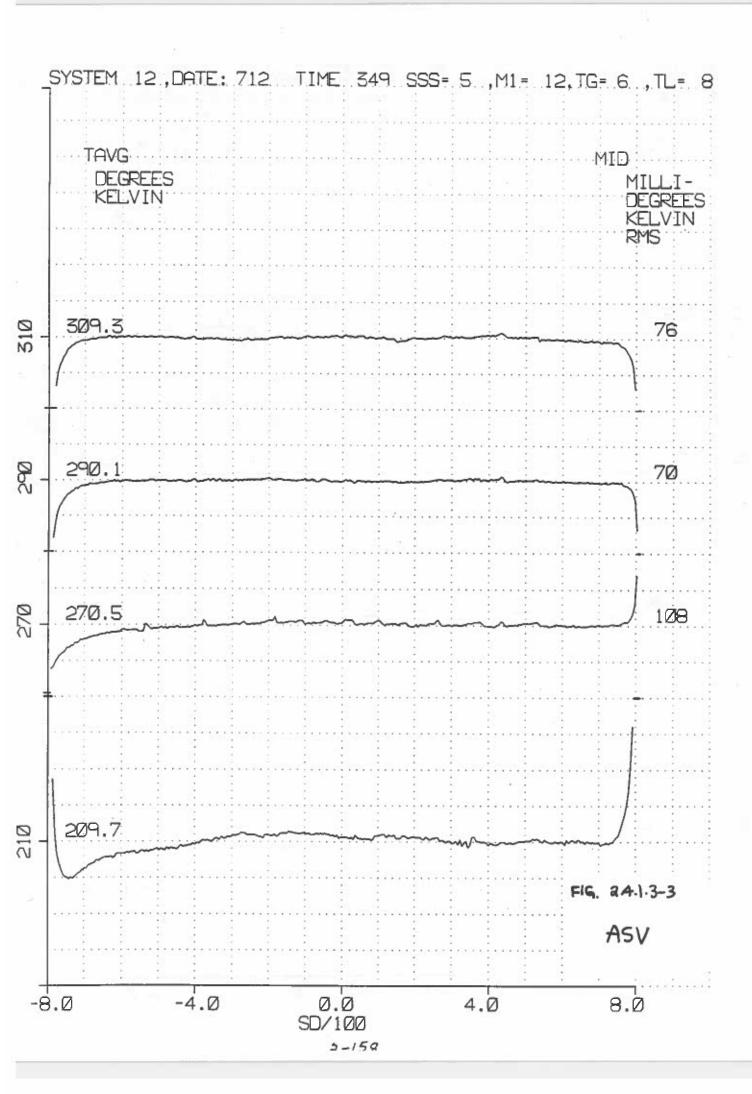
# OLS #12

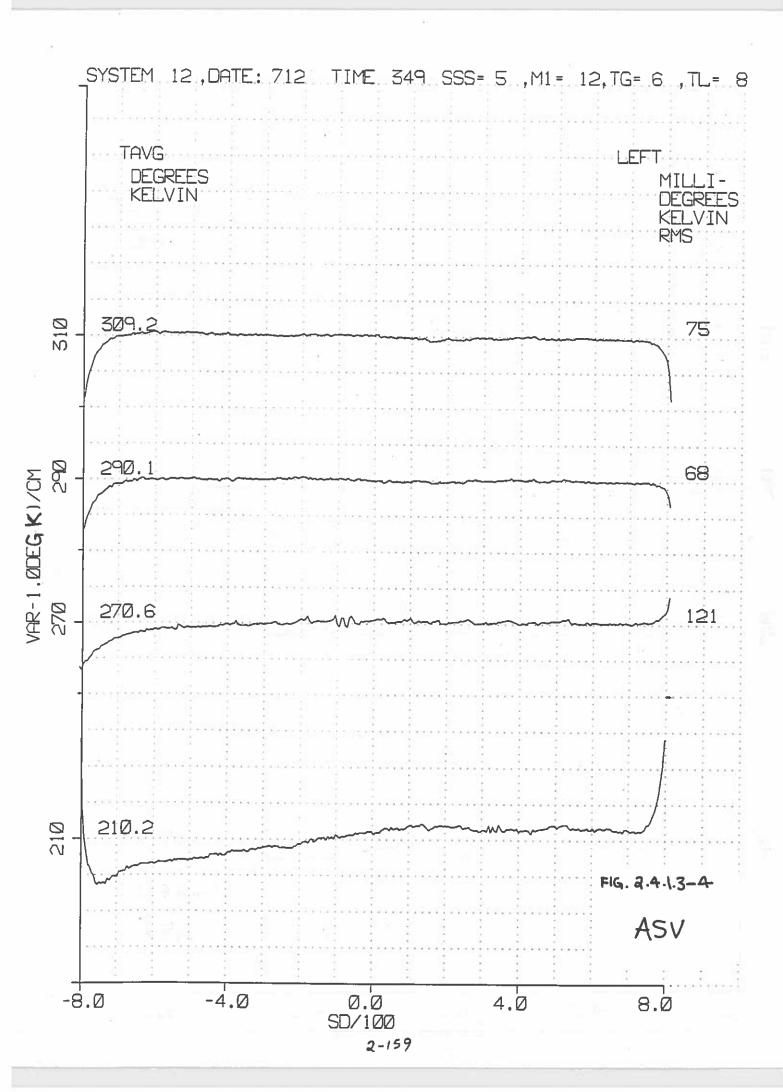
# T CONE TEMP EST (EST #33)

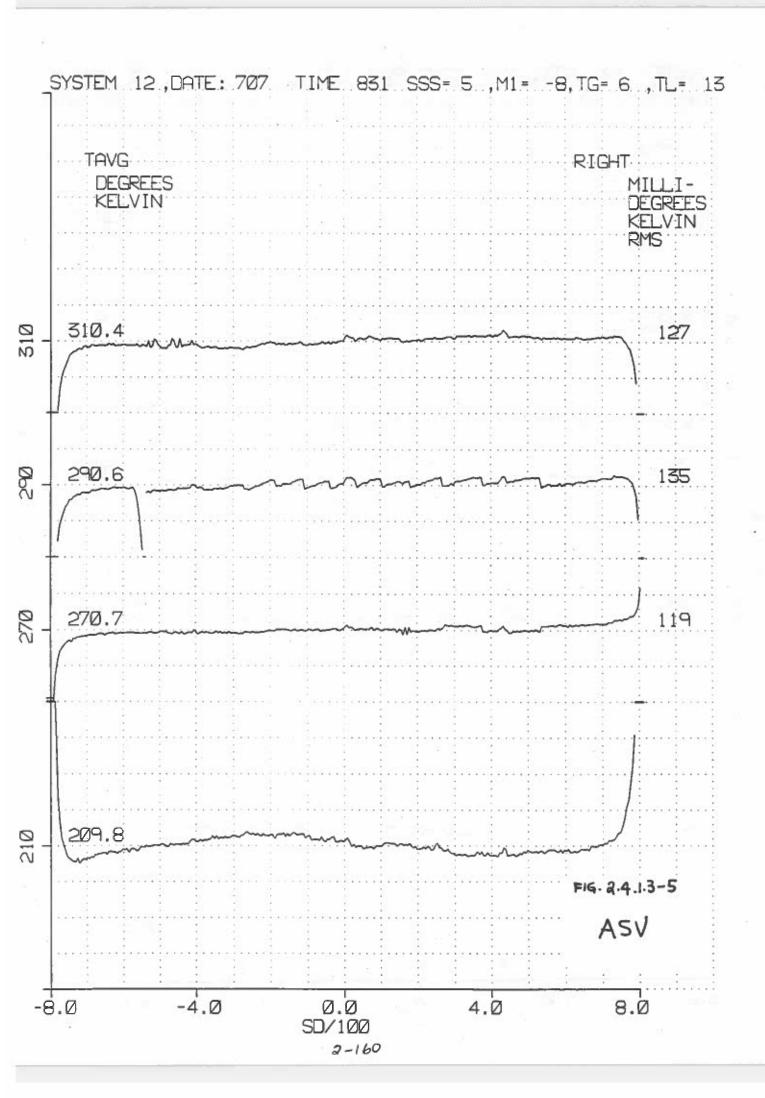
T (DEG K)  158 159 160 161 162 163 164 165 166 167 168 169 170 171 172 173 174 175 176 177 178 179 180 181 182 183 184 185 186 187 188 189 190 192	4.821 4.8181 4.8136 4.8088 4.8035 4.7978 4.7915 4.7848 4.7775 4.7695 4.7609 4.7515 4.7414 4.7306 4.7188 4.7063 4.6926 4.678 4.6926 4.6454 4.622 4.6454 4.6273 4.608 4.5654 4.5654 4.5654 4.5654 4.517 4.4904 4.4622 4.4323 4.4008 4.3673 4.3322 4.2951 4.2156		T (DEG K) 194 196 198 200 202 204 206 208 210 212 214 216 218 220 222 224 226 228 230 235 240 245 250 255 260 265 270 275 280 285 290 295 300 305	EST VOLTS 4.1282 4.0328 3.93 3.8195 3.7016 3.5769 3.4468 3.3115 3.1719 3.0292 2.8844 2.7386 2.5924 2.4475 2.305 2.1659 2.0302 1.8995 1.7735 1.4832 1.2308 1.0159 0.8359 0.6873 0.5650 0.4653 0.3842 0.3182 0.2646 0.2207 0.1852 0.1560 0.1320 0.1123
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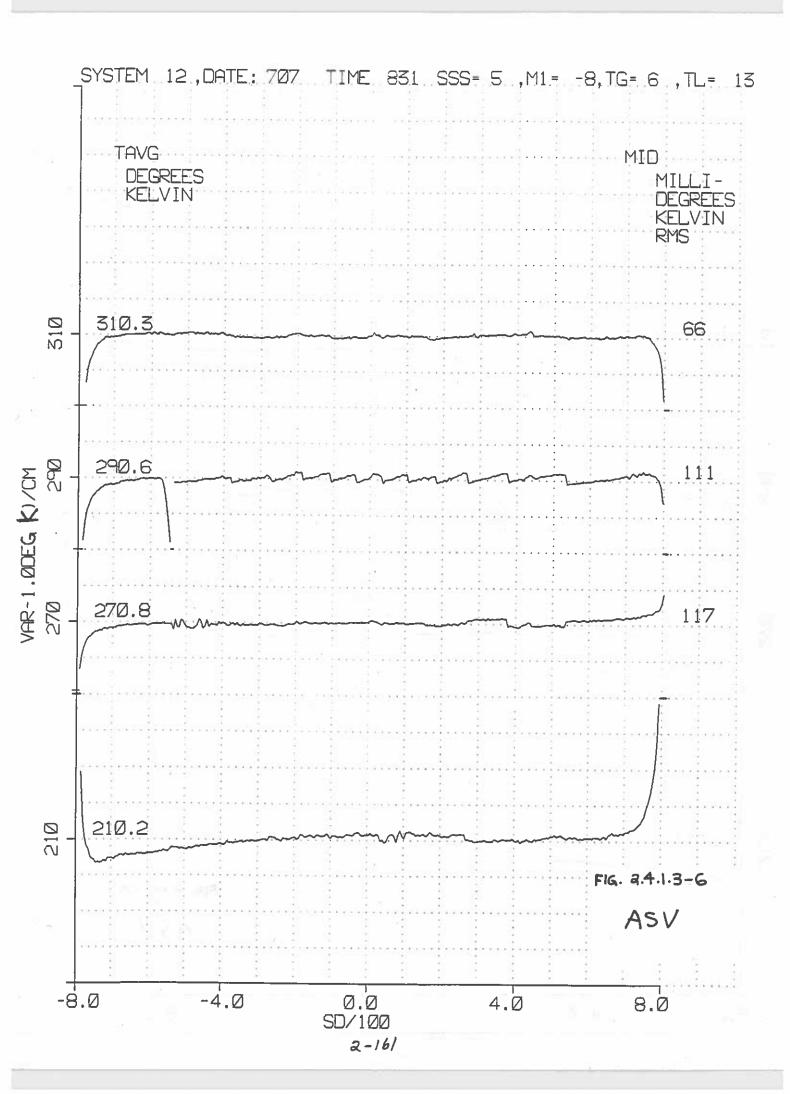


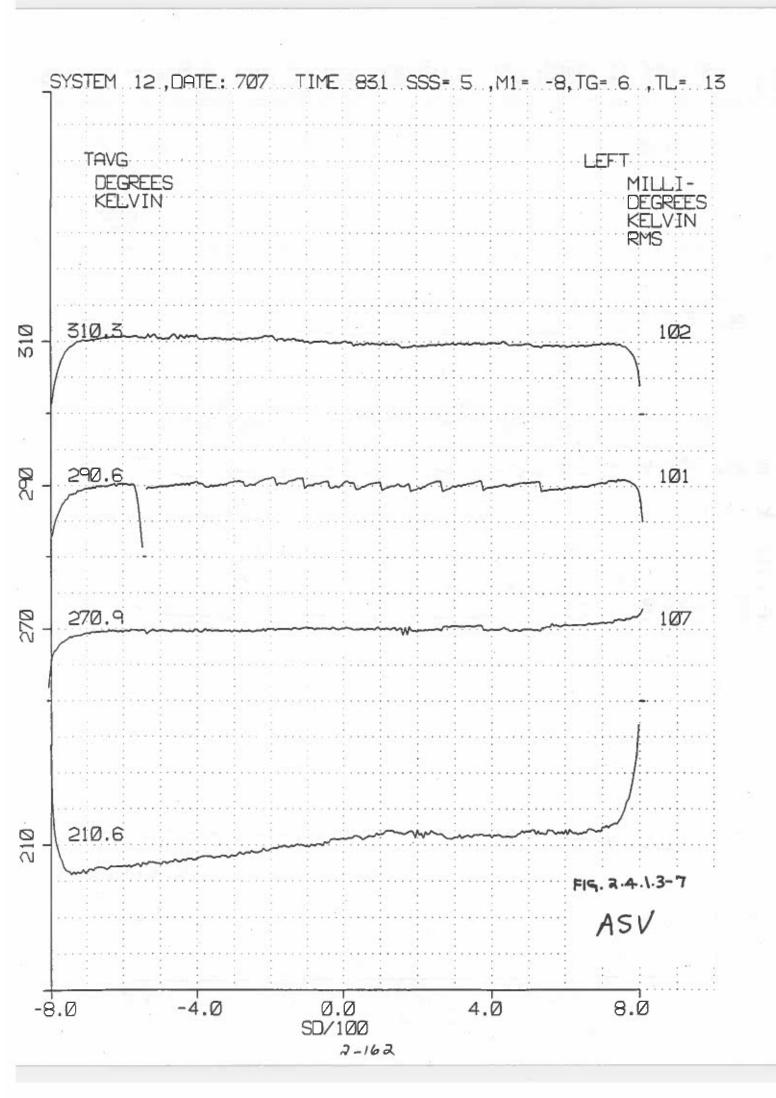


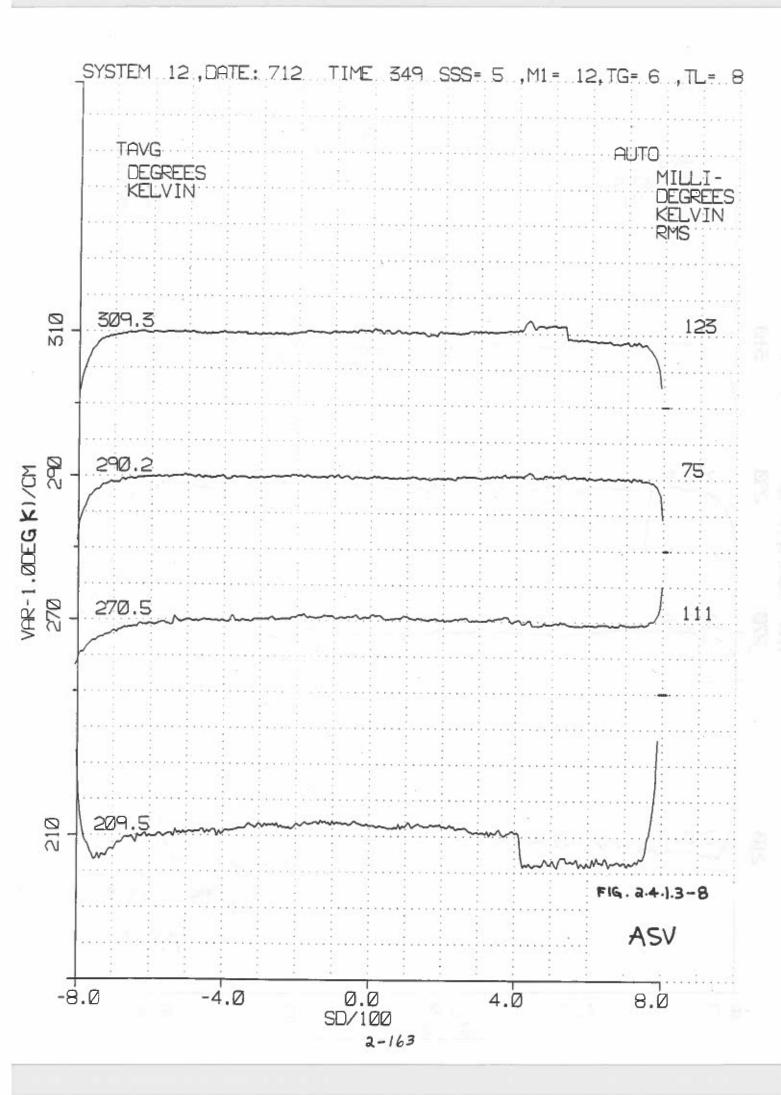


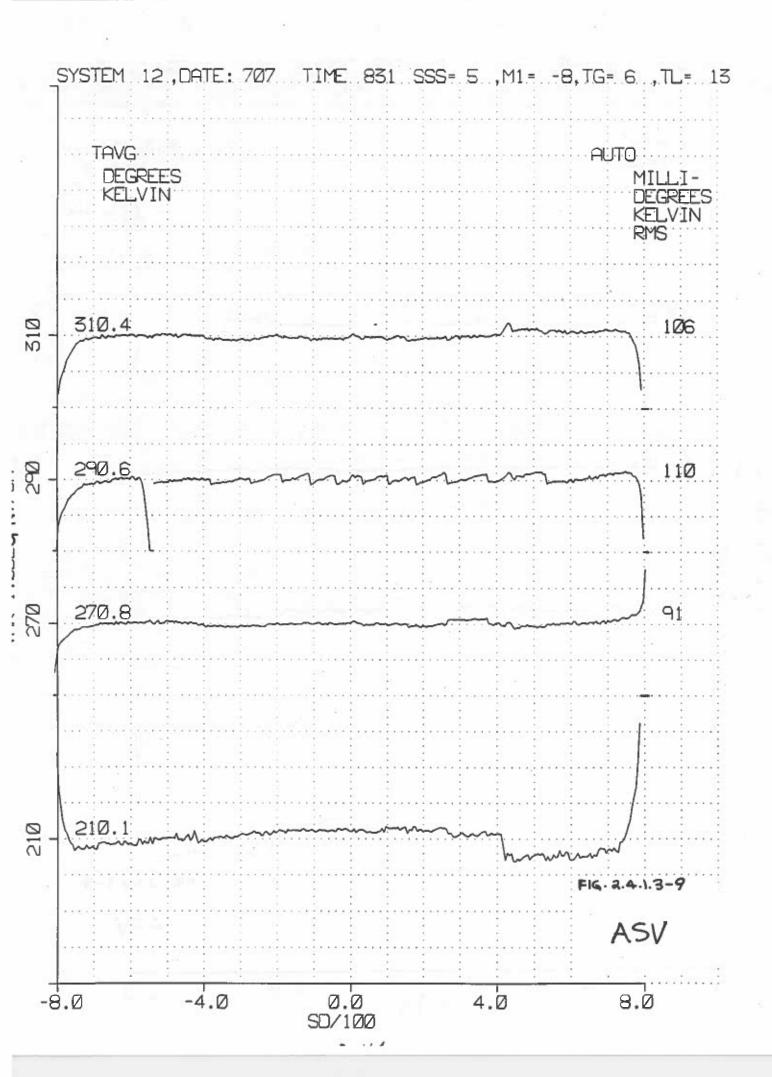












### 2.4 Radiometric Accuracy (Cont'd)

#### 2.4.2 Daytime Radiometric Accuracy (3.2.1.1.4.2)

OLS #12 achieved the 7% absolute radiance requirement by setting the HRD channel gain as shown on the L channel DC Response plot, using the calibrated light source (VULS).

The analysis of the calibration accuracy and the L-Day channel stability show within-specification performance. The gain ratios (PMH/PML, PML/HRD, and PMH/HRD) were measured during bearing retrofit retest using the VULS during Acceptance Test in test 6x2x1.ST, and vary less than 0.3% from the average of the ratios. The gain ratios measured in test 6x2x2.ST using a less accurate test method show greater variation.

The plot of L DC Response contains the calculated sensor switch points, S(x), and relative gains P(X), which are stored in the OLS Constants Memory page zero, locations 071 through 077. P(2) and S(2) represent the bypass of the PMT 1/9 mode, which is not usually implemented on orbit.

OLS #12 exhibited 1.10dB drop in transmission from room temperature to  $+5^{\circ}$ C. The correct light level corresponding to  $2.12 \times 10^{-2}$  w/cm<sup>2</sup> -sr is changed by 3% (0.26dB) relative to factory adjustment based on updated HRD spectral data used in the GAINSET program. Also, OLS #12 exhibited a lower optics transmission than typical OLS units by 2dB in the HRD channel. Thus P(0) must be reset to 6.0 (nom) + 1.10 + 0.26 +2 = 9.36dB. Rounding off to the nearest 1/8 dB gives 9.375dB as the new setting for P(0).

The S1 value used for OLS 12 DC response adjustment is 2.96 v. Also, the G1 value (HRD to PMT gain offset factor due to differing spectra) used was 4.37 db and the lunar/solar gain ratio used was 1.033.

P1 is derived using the PMT LO/HRD average gain value of 49.98 dB with a compensation for the HRD Loss and PMT Gain Ratios with temperature from the bearing retrofit retest data as plotted in figure 2.4.2-2 and converted to dB of 1.10 dB and .61 dB, respectively. The P1 value is 49.98 + 0.61 + 1.10 = 51.69 (rounded to nearest 1/8th dB = 51.75).

ATTACHMENT: OLS #12 L Channel DC Response Plot

Table 2.4.2-1 OLS #12 DC Response Stability

Table 2.4.2-2 OLS #12 PMT/HRD DC Response vs. SSS Temp.

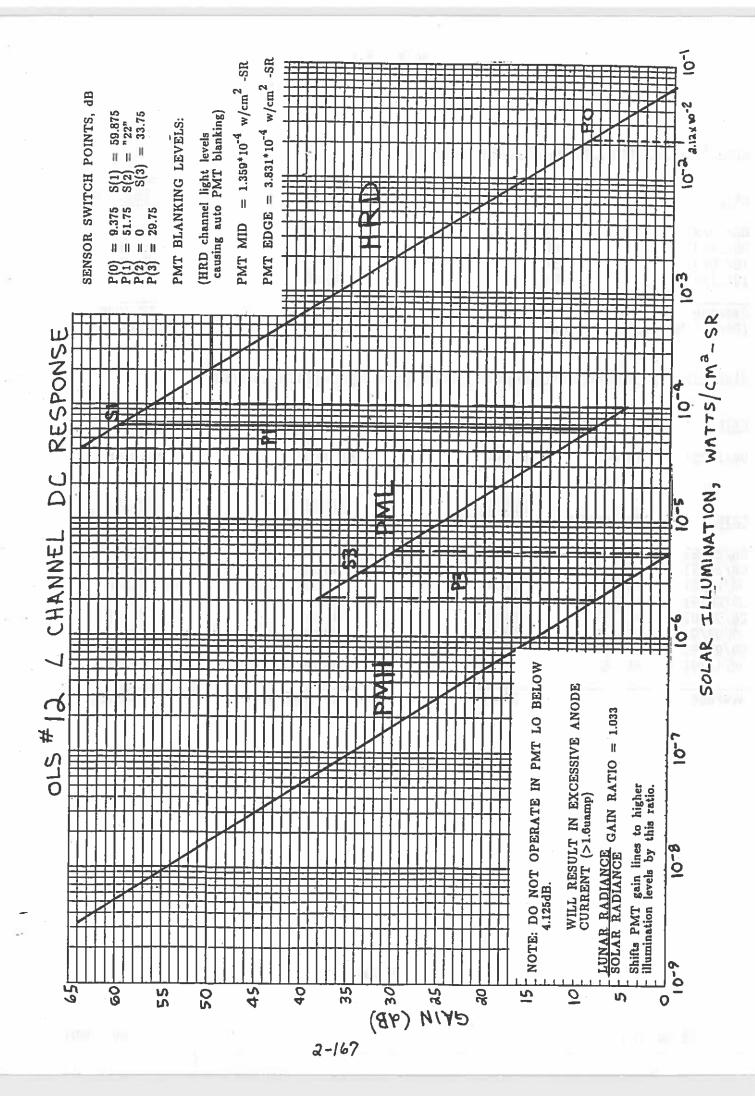


Table 2.4.2-1
OLS #12 L DC Response Stability

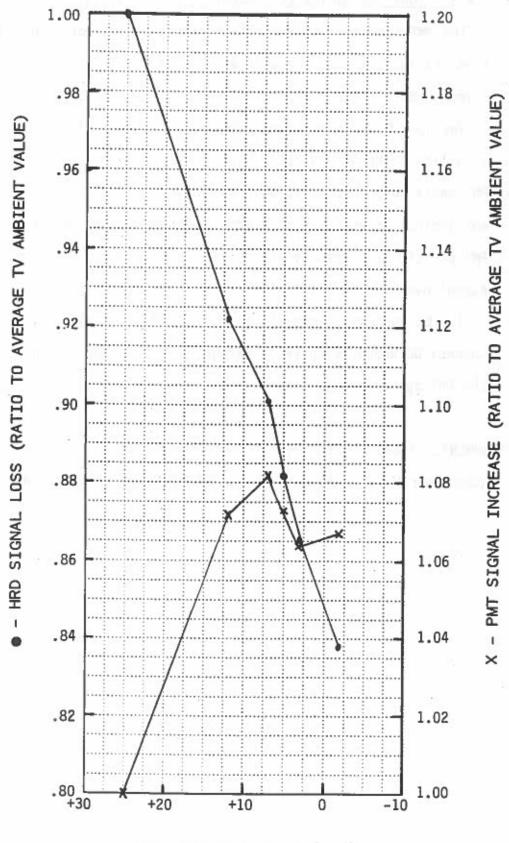
# Stability vs. Time (6x2x1.ST data using VULS)

DATE	PMT HI dB	PMT LO dB	PMT HI HRD dB
08/10/91	29.74	49.98	79.71
08/14/91	29.73	49.95	79.68
10/26/91	29.73	50.15	79.88
10/27/91	29.71	49.82	79.53
Average	29.73dB	49.98dB	79.70dB
(Direct Multiple)	(30.65)	(315.50)	(9660.51)

# Stability vs. Temperature (6x2x2.ST data using half sphere source)

<u>DATE</u>	ENVIRONMENT	PMT LO dB	PMT_LO HRD d8	PMT HI dB
08/18/91	TV Amb	29.74dB	39.06dB	68.80dB
DATE	ENVIRONMENT	PMT HI dB	PMT LO HRD dB	PMT HI HRD dB
08/22/91 08/25/91 08/26/91 08/28/91 08/30/91 09/02/91 09/07/91 09/12/91	+5/-8 +12/+15 -2/-11 +12/+15 -2/-11 +7/+12 +3/-8 +5/-8	29.65 29.69 29.63 29.76 29.78 29.68 29.68 29.69	40.85 40.23 41.42 40.44 41.47 40.66 41.06 40.89	70.79 69.93 71.06 70.19 71.25 70.35 70.74 70.58
Average		29.70dB	40.88dB	70.57dB

TABLE 2.4.2-2. PMT/HRD DC RESPONSE vs. SSS TEMPERATURE



SSS TEMPERATURE (DEG C)

# 2.4 Radiometric Accuracy, (Cont'd)

# 2.4.3 Nighttime Radiometric Accuracy (3.2.1.1.4.3)

The PMT accuracy is required to degrade by no more than 60% from its initial accuracy at time of Acceptance testing to end of 3 year orbital life.

The 5D-3 OLS System Summary Report indicates PMT channel stability to be within 25.7% over the mission life so that the PMT meets this specification requirement. The above figures do not include corrections utilizing on-board LED calibration. The PMT CAL LED is extremely stable, and has an essentially constant output over the mission life.

The DC response curve of the OLS #12 PMT is shown in the L Channel DC Response curve in paragraph 2.4.2. Unlike the HRD, the PMT optics transmission appears typical.

ATTACHMENT: Table 2.4.3-1 PMT CAL Baseline data

(See para 2.4.2 attachment for L. Chan. DC Response curve)

# TABLE 2.4.3-1 PMT CAL BASELINE DATA

The PMT Cal voltage EST is a monitor of PMT sensitivity and stability characteristics. In order to relate on-orbit measurement of PMT CAL to measurement during Acceptance Test, the following data are provided. PMT CAL Voltage (EST #40) and PMT BU (Back-up) CAL V (EST #41) are sampled 500 times in test 6x6x2.ST. The PMT Cal voltage EST output is tabulated below for all runs after PMT replacement.

TABLE 2.4.3-1
PMT CAL BASELINE DATA

<u>DATE</u>	SSS TEMP	OUTPUT VOLTAGE (mV)		
		PMT CAL V (EST #40)	PMT BU CAL V (EST #41)	
08-09-91	+25	2478	2475	
08-14-91	+25	2499	2487	
09-03-91	+12	2378	2372	
09-08-91	+3	2361	2356	
09-14-91	+5	2381	2376	
			33	
	AVERAGE	2419	2413	
Max change	from AVERAGE	3.31%	3.07%	

# 2.4 RADIOMETRIC ACCURACY, (Cont'd)

# 2.4.4 Gain Control Accuracy (3.1.4.4)

Along-Scan Gain Control (ASGC) accuracy is within the specification limit of 4dB of the smooth monotonic curve drawn through the nominal values of Gain Value versus Scense Solar Elevation (GVVSSE), tabulated in para 3.2.1.1.1.4 of DMSS-OLS-300.

System Tests 5x6x3.ST and 5x6x6.ST exercise the ASGC function through various combinations of extremes of slope and bias adjustment; and measure channel output for a wide range of values. These are automatically compared against stored test limits which ensure spec compliance.

An analysis using the ASGC software algorithm showing 3.25dB maximum error is summarized in the System Summary Report paragraph 3.5.4.

ATTACHMENTS: None.

#### 2.4 RADIOMETRIC ACCURACY, (Cont'd)

### 2.4.5 Gain Control Adjustability (3.2.1.1.4.4 et al)

The OLS #12 Gain Control Adjustability is the same as for 5D-1 systems. The OLS 5D-3 System Summary Report paragraph 3.5.5 contains the analysis required to demonstrate conformance with the specifications, with additional information below.

### TERMINATOR LOCATION (3.1.4.5.1)

The GNC Command (an uplink command) has a sub mode (GNC  $10\ 0\ X$ ) which allows the GVVSSE bias (terminator location) to be adjusted by X degrees, where X is a 6-bit 2's complement word with an LSB of  $0.5^{\circ}$ . This results in the required range of  $\pm 15.5$  degrees.

### GAIN CHANGE RATE (3.1.4.5.2)

The GNC command submode (GNC 10 1 X) allowed the GVVSSE slope to be varied by a factor of 1+X over  $\pm$  48% in 1.6% increments where X is a 6-bit 2's complement word with an LSB of  $2^{-6}$ . This results in the required range of  $\pm$  48%. However, the BRDF change in the L channel gain calculation required the deletion of this capability to meet timing limitations in the OLSP. The 886 spec must be revised to reflect this change.

#### MAXIMUM GAIN SETTING (3.2.1.1.4.5.3)

The maximum ASGC gain is commandable. An operational value is determined in Early Orbit Calibrations. The value is stored in the Operational Constants Memory location page  $\theta$  Address 104 (BCMAX). The ASGC function & performance are exercised in tests 5x6x3.ST & 5x6x6.ST

### COMMANDABLE T-CHANNEL GAIN (3.2.1.1.4.5.4)

The T-Channel Commandable Gain is exercised in test 6x8x2.ST. The channel output is measured for the entire range of commandable gains. The T Channel Gain is required to have the capability of being varied by command to be between 146% and 149% of the established minimum gain value (0 dB).

The actual percentage of TGAIN change was not measured as part of bearing retrofit. However, previously this was measured for OLS #12 as 49.8% for T Right and 49.2% for T Left.

Each step of TGAIN is required to be between 1.7% and 3.7% above the preceding lower gain value. Measured gain steps on OLS #16 ranged from 1.86% to 3.47%, within specification.

# COMMANDABLE T CHANNEL LEVEL (3.2.1.1.4.5.5)

The T Channel Commandable Level is also exercised in test 6x8x2.ST. The specification requires that TLEVEL be variable over at least a 14° range in steps of 1.1°K or less. Measured results during the original OLS 12 testing were 15.12° range and step sizes between 0.969°K to 1.027°K worst-case; all within specification.

### 2.4 RADIOMETRIC ACCURACY, Cont'd

2.4.6 <u>A/D Conversions & Algorithms</u> (3.2.1.1.4.6.2 & 3.2.1.1.4.6.3)

DMSS-OLS-300 specifies that the Stored Smooth Algorithm accuracy with an ideal A/D shall be verified by analysis. OLS 5D-3 System Summary Report contains the analysis which shows that the LS % Full Scale Deviation does meet the specification. The results are summarized below:

<u>Mode</u>	% Full Scale Deviation	Analysis
LS	$\leq \pm 2.2\%$	-1.70%, $+2.09%$
TS	$\leq \pm 0.4\%$	± 0.39%

The Actual A/D Conversion Radiometric Accuracy was measured in system test 6x10.ST. The worst-case results from the OLS #12 Thermal Vacuum runs (Cold Limit & Orbit Nominal) are summarized below:

A/D	BSL SLOPE (% DEV FROM IDEAL)	BSL OFFSET (% OF FULL SCALE)	RMS DEV FROM BSL (% OF FULL SCALE)
SDF-L PRIM	-0.33	0.18	0.04
RED	-0.19	0.10	0.03
SDF-T PRIM	0.17	-0.12	0.15
RED	0.38	-0.16	0.17
RTD-F PRIM	-0.33	0.18	0.04
RED	-0.18	0.08	0.05
SPEC	±1.0	±1.0	0.5
RTD-S PRIM	-0.23	0.08	0.02
RED	-0.26	0.06	0.02
SDS-L PRIM	-0.23	0.06	0.04
RED	0.20	0.06	0.06
SDS-T PRIM	0.10	-0.30	0.06
RED	0,50	-0.20	0.08
SPEC	<u>+</u> 0.5	±0.5	0.25

# 2.5 <u>RADIOMETRIC RESOLUTION</u> (3.2.1.1.5 et al.)

DMSS-OLS-300 apportions the Radiometric Resolution verification between Test and Analysis.

The Fine and Direct Smoothed Radiometric Resolution (para. 3.2.1.1.5.1), as well as the Stored Smoothed A/D Converter Error (para. 3.2.1.1.5.2), were measured in System Test 6x10.ST during the OLS #12 vacuum runs and are tabulated below:

А	/D	PEAK DEV FROM BSL (% OF FULL SCALE)	SPEC
SDF-L	PRIM RED	0.07	±0.8%
SDF-T	PRIM RED	0.33 -0.30	<u>+</u> 0.8%
RTD-F	PRIM RED	0.08 0.12	<u>+</u> 0.8%
RTD-S	PRIM RED	0.04	<u>+</u> 0.25%
SDS-L	PRIM RED	-0.07 0.10	<u>+</u> 0.5%
SDS-T	PRIM RED	-0.10 -0.16	<u>+</u> 0.5%

The Stored Smoothed Algorithms Resolution with Ideal A/D (para. 3.2.1.1.5.2.1) are verified by Analysis in OLS 5D-3 System Summary Report, and are summarized in Table 2.5.1.

# TABLE 2.5-1 Stored Smoothed Algorithms Resolution With Ideal A/D

<u>ITEM</u>	SPEC	<u>ACTUAL</u>
Accuracy		
LS Algorithm	< <u>+</u> 2.2%	-1.70% +2.09%
TS Algorithm	< <u>+</u> 0.4%	<u>+</u> 0.39%
Resolution		2 1
LS Algorithm	<1.6%	1.57%
TS Algorithm		
Population 1 Density Quantization	25% <0.8%	25% 0.78%
Population 2 Density Quantization	75% <0.4%	75% 0.39%
Population Distributtion	Uniform	Uniform
Quantization Capability	0.4%	0.4%

### 2.6 NOISE

# 2.6.1 <u>T Channel Noise</u> (3.2.1.1.6.1)

The T Channel noise equivalent temperature difference (NETD) is specified between 210K and 310K, although the T channel responsivity extends down to 190K.

The NETD is measured during T Channel DC response tests in the Thermal Vacuum Chamber.

The Channel views a stable blackbody target at 210K (worst-case noise). The channel output is sampled & the noise (std. deviation) of 2000 samples is converted to NETD using the following formula:

NETD = [Avg Noise in Volts \* 24 °/Volt] \* 1.074 (the shaper slope correction)

The OLS #12 NETD is in-spec. The noise in the T Right segment is 11.8% larger than in the T Left segment.

		TS	<u>TS Fallback</u>
SPEC	2.2°K	0.90°K	1.3°K
Worst-Case Measured NETD	0.695°K	0.279°K	0.3 <del>9</del> 3°K
Worst-Case Average NETD	0.662°K	0.273°K	0.370°K

ATTACHMENT: Table 2.6.1-1 OLS #12 Primary Side NETD

Table 2.6.1-2 OLS #12 Redundant Side NETD

Table 2.6.1-1

				OLS #	12 PRIMAR	A SIDE M	:10	_		7//
							Nois	e mV		
DATE	SSS	M1	G R/L	TL	FINE RGT	SMOOTH RGT	FINE	SMOOTH MID	FINE LFT	SMOOTH LFT
07/06/91	5	-8	5/6	13	26.97*	15.24	17.21	10.14	23.28	12.50
07/14/91	12	15	5/6	9	25.46	13.99	16.93	10.26	21.99	11.63
07/18/91	-2	-11	5/6	13	26.64	15.25*	17.32	10.18	24.28*	12.60
08/22/91	5	-8	5/6	13	26.32	14.00	17.04	10.76	22.90	12.25
08/24/91	5	12	5/6	9	25.38	14.67	16.60	10.50	22.11	11.83
08/25/91	12	15	5/6	9	25.17	13.65	16.60	10.25	22.02	11.45
08/26/91	-2	-11	5/6	13	25.64	14.41	16.60	10.20	22.78	12.52
08/28/91	12	15	5/6	9	24.22	14.01	15.95	10.07	21.94	11.58
08/30/91	-2	-11	5/6	13	26.17	14.63	16.78	10.35	23.27	12.51
09/01/91	7	12	5/6	9	25.60	13.63	16.61	10.61	22.74	11.90
09/07/91	3	-8	5/6	_ 13	25.15	14.49	16.86	10.03	22.03	12.10
			AVERA NETD	GE	25.70 0.617	14.36 0.345	16.77 0.402	10.30 0.247	22.67 0.544	12.08 0.290
NETD Correct	cion fo	r Shap	er Slo	pe**	0.662	0.370	0.432	0.265	0.584	0.311

<sup>\*</sup> Worst Case Measured
\*\* Shaper Slope Correction Factor = 1.074

Table 2.6.1-2

			(	DLS #1	2 REDUNDA	NT SIDE N	IETD						
100						Noise mV							
DATE	SSS	M1	G R/L	TL	FINE RGT	SMOOTH RGT	FINE	SMOOTH MID	FINE LFT	SMOOTH LFT			
07/05/91	5	-8	6/6	12	26.58	14.73	18.53*	10.47	23.56	11.97			
07/11/91	5	12	5/6_	8	25.77	14.31	17.82	10.84*	23.59	12.72*			
09/12/91	5	-8	5/6	13	24.75	13.86	16.03	10.50	22.56	12.24			
AVERAGE NETD  NETD Correction for Shaper Slope**				25.70 0.617 0.662	14.30 0.343 0.369	17.46 0.419 0.450	10.60 0.254 0.273	23.24 0.558 0.599	12.31 0.295 0.317				

<sup>\*</sup>Worst Case Measured
\*\*Shaper Slope Correction Factor = 1.074

### 2.6 NOISE (Cont'd)

### 2.6.2 <u>L-Channel Noise (Day)</u> (3.2.1.1.6.2)

The L Channel Noise is measured using the calibrated Variable Uniform Light Source (VULS). Dark noise is measured in test 6x3x1.ST and shot noise is measured in 6x3x5.ST.

The OLS #12 HRD is in-spec for the entire range of illumination. Worst-case HRD SNR exceeds the specification. In summary:

		SNR	
LIGHT LEVEL	SPEC	PRIOR OLS 12 TESTING WORST CASE MEASURED	RETEST WORSE CASE (FROM GRAPH)
5.5 x 10 <sup>-5</sup>	5	7.45	7.8
$5.5 \times 10^{-4}$	34.8	68.1	88.1
$1.1 \times 10^{-3}$	62.3	116	140
$2.2 \times 10^{-3}$	112	189	201
5.5 x 10 <sup>-3</sup>	200	251	235

OLS #12 HRD CHANNEL SNR

SOLAR ILLUMINATION WATTS/CM - SR

### 2.6 NOISE (Cont'd)

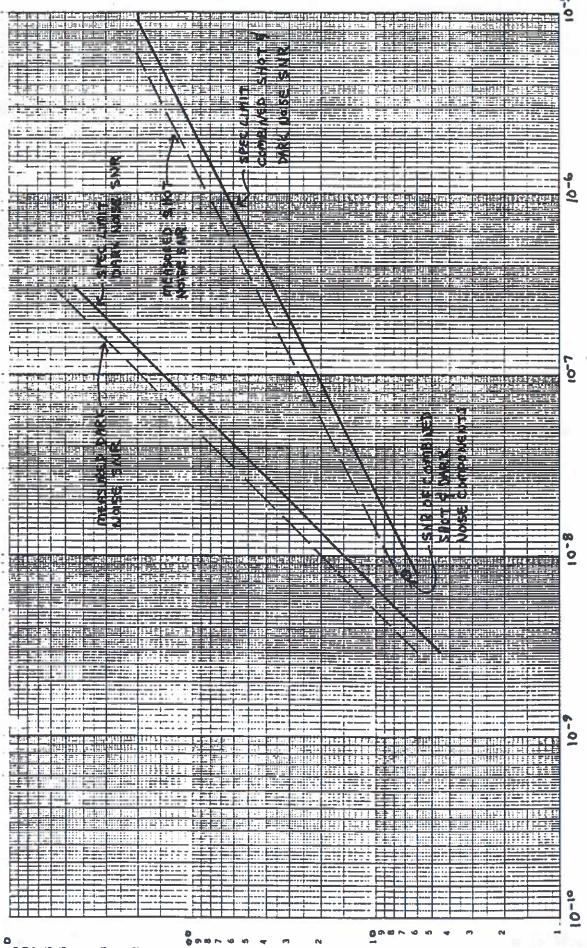
2.6.3 <u>L Channel Noise (Night)</u> 3.2.1.1.6.3)

The PMT dark noise is measured in all environments in Tests 6x3x1.ST, 6x3x2.ST and 6x3x4.ST. The SNR is calculated from the measured noise (std. deviation of multiple voltage samples) vs. light level and compared against spec values.

The minimum SNR from Bearing Retrofit retest is 7.8 at 8.0 x  $10^{-9}$  watts/cm<sup>2</sup>-sr. The minimum SNR from Bearing Retrofit Retest is 16.0. The worst case combined PMT shot noise and dark noise SNR from bearing retrofit testing is 7.01 calculated as

 $SNR=1/\sqrt{1/(SNR \, dark)^2} + 1/(SNR \, shot)^2$ .

ATTACHMENT: OLS #12 PMT channel SNR graph.



SOLAR ILLUMINATION

WATTS/CM2-SR

**SNR** 0 -010A

### 2.6 NOISE (Cont'd)

### 2.6.4 <u>Dark Current</u> (3.2.1.1.6.4)

The Dark Current (the PMT noise with no signal input) is determined from the graph of PMT SNR in paragraph 2.6.3. The Dark Noise SNR is calculated from data gathered during PMT Smoothed Noise measurements. These measurements are made in Test 6x3x1.ST during Thermal Vacuum testing. For the OLS #12 bearing retrofit retest, the average Dark Noise SNR of 5 measurements at  $8 \times 10^{-9}$  watts/cm²-SR is 16.2, or 37.0% of the noise corresponding to an SNR of 6. The MINIMUM Dark Noise SNR measured at  $8 \times 10^{-9}$  watts/cm²-SR was 16.0, or 37.5% of the noise corresponding to a SNR of 6. This is well within the spec requirement for the dark current to be 50% or less of the noise corresponding to an SNR of 6.00.

### 2.6 NOISE, (Cont'd)

2.6.5 <u>Stability</u> (3.2.1.1.6.5) (L - Channel (night)

The OLS 5D-3 System Summary Report contains the analysis for this spec requirement.

The loss in sensitivity after 3 years on orbit of the PMT channel will be < 23%. This would require 2.27dB change in VDGA gain to compensate and over 17dB of VDGA gain is available.

### 2.6 NOISE (Cont'd)

2.6.6 Along-Track Noise Integration (3.2..1.1.6.6)

OLS 5D-3 System Summary Report contains the analysis which concludes that the OLS 5D-3 algorithm is consistently above 0.6 times the SNR resulting from perfect integration with 8 bit A/D for T Channel; and above  $1/\sqrt{2}$  times the SNR resulting from perfect integration with 6-bit A/D for L channel. Therefore, the Along-Track Noise Integration is in-spec.

### 2.6 NOISE, (Cont'd)

2.6.7 Glare Suppression (3.2.1.1.6.7)

OLS 5D-3 System Summary Report contains the analysis which verifies that the OLS does provide effective protection against solar glare for sun angles between 75° and 95°.

### 2.7 SURVIVABILITY (3.2.7)

The OLS 5D-3 System Summary Report contains calculations of survivability. See BVS 2353 (Verification of Survivability Requirements) for further details.

### 2.8 <u>SCAN ANGLE</u> (3.2.1.1.8)

Tests 6x7x1.ST and 6x7x3.ST (End of Scan Vignette for HRD & T channels respectively) measure the delphi number at which 1% vignetting of scene begins to occur. The measured delphis enable calculation of the altitude needed to obtain contiguous coverage at the equator. The contiguous coverage requirement is based on the average of +Z and -Z scan angles.

For OLS #12, the following results were obtained:

CHANNEL	DELPHI	SCAN ANGLE	CONTIGUOUS COV	ERAGE ABOVE:
+Z HRD	+991.2	+55.97°	427.09 n. mi.	427.67 avg.
-Z HRD	-990.2	-55.91°	428.25 n. mi.	
+Z T	+981.0	+55.39°	438.34 n. mi.	435.52 avg.
-z T	-986.0	-55.68*	432.70 n. mi.	

Thus, both the HRD channel and the T channel meet the requirements for contiguous coverage above 440 naut. mi, since both channels will provide contiguous coverage for all altitudes above 433.38 naut. mi.

### 2.9 DATA COLLECTION RATE (3.2.1.1.9)

OLS #12 does scan the field of view at the prescribed 11.88 +/.12 Hz rate. This parameter is measured in Scanner Functional tests
5x12x1.ST (Primary Side) and 5x12x2.ST (Redundant Side).

The test results are summarized below for all TV tests:

Date	Frequ <u>Primary</u>	ency, Hz <u>Redundant</u>
07-06-91 Optic Limit	11.90	11.91
09-02-91 Hot Limit	11.89	11.89
09-06-91 Cold Limit	11.91	11.90
09-11-91 Orbit Nom.	11.90	11.90

### 2.10 POWER (3.3.1 and 3.3.2)

Both +28V and +5V power is measured and monitored continuously throughout all of the test sequence.

The power required in the 8 Development-Spec-defined modes is tabulated below. The Development Spec Power Profile is measured in test 5x2x11.ST for modes 1 through 8 and 5x2x2.ST for mode 0.

10V power consumption was not tested on OLS #12. The current system test equipment is not capable of monitoring 10V power.

Analysis of the components using S/C supplied 10V power indicates that 5D-3 10V power consumption is in spec.

DMSS-OLS-300 limits 28V power consumed for SSS thermal control to 23 watts maximum. SSS heater power consumption was not measured on OLS #12. Analysis of the heater resistances and tolerances indicates that 5D-3 SSS heater power consumption is in spec.

OLS #12 28V power consumption is in spec for all modes in the primary and redundant configurations. Fallback (dual power) configuration power consumption is also in spec. See the attached table for further details.

ATTACHMENTS: OLS #12 Power Profile

OLS #12 POWER PROFILE

P.	SINGLE	DUAL POWER				
28V MODE/LIMIT	TV +5/-8 07-04-91	TV HOT LIMIT 09-01-91	TV COLD LIMIT 09-06-91	28V LIMIT	WORST CASE (CALCULATED)	
∉0 88₩	53	53	53	131W	88	
1 105W	82	82	81	148W	117	
2 116W	88	89	88	159W	124	
3 125W	95	96	93	168W	131	
4 157W	131	131	128	200W	166	
5 167W	137	139	135	210W	174	
6 198W	166	167	162	241W	202	
7 207W	178	180	172	250W	215	
8 218W	185	,187	178	261W	222	
5V MODE/LIMIT		De n		1-11 45	Maria	
0 4.3W	3	3	3	amilia	K 1 &	
1 4.3W	3	4	3			
2 4.3W	3	4	m 1 m3 1 m	-1 fit	Bu Bulla	
3 4.3W	3	4	3			
4 4.3W	3	4	3			
5 4.3W	3	4	3			
6 4.3W	4	4	3			
7 4.3W	4	4	3		ε	
8 4.3W	3	4	3 :		*1	

### 2.11 MASS

### 2.11.1 <u>Total Mass</u> (3.4.1)

The weights of all OLS #12 components were not measured as part of bearing retrofit. The data taken on 12-03-86 during the original OLS 12 selloff are provided for reference. The tape recorder and encrypter serial numbers are those belonging to the system at OLS #12 sell-off and may change.

All Westinghouse furnished parts meet their center of gravity specification limits and their maximum specified weight allocation. All Typical encrypters exceed the spec limit in center-of-gravity. The encrypters are GFE to WEC and their weight and C.G. are not controlled by WEC.

The total weight of the OLS #12 AVE is 291.01 pounds, (less BBX's, but including GSSA/DOC & Test Cable), vs. a spec limit of 298 pounds.

ATTACHMENT: OLS #12 Weight and Center-of-Gravity Tables

WESTINGHOUSE FURNISHED PARTS SUPPLIED WITH OLS 12 SYSTEM SUMMARY OF WEIGHT AND CENTER GRAVITY

è	ACT	54.64	69.00	17.32	26.31	3.52	8.10	21.46	21.44	21.40	21.33	20.49	:	6.0
HT	MPR**	54.35	69.50	17.34	26.10	3.53	7.99	21.57	21.56	21.56	21.56	21.30	1	6.0
WEIGHT	MPR**	53.29	68.15	17.00	25.60	3.47	7.83	21.14	21.14	21.14	21.14	20.88		6.0
	MAX* SPEC	59.0	70.0	18.0	27.0	4.0	9.0	22.75	22.75	22.75	22.75	22.0		6.0
	ACT	0.59	8.56	5.84	7.15	2.72	2.37	4.14	4.28	4.29	4.30	ı		-
_7	MPR	9.14.0	8.6+.8	6.0±.5	7.2±.5	3,0+.5	2.4+.5	4.23±.25	4.23±.25	4.23±.25	4.23±.25	ī	ĺ	-
	SPEC	0.7±.5	8.6+.8	6.0±.5	7.2±.5	3.0+.5	2.4+.5	4.23±.25	4.23±.25	4.23±.25	4.23±.25	£		•
	ACT	6.10	13.76	6.55	6.72	4.35	0.15	6.13	6.38	6.19	6.28	ı	Ì	-
>-	MPR	6.2±.5	13.8±1.0	6.6±.5	7.0±.6	4.0+.5	+0.1±.3	6.36±.25	6.36±.25	6.36±.25	6.36±.25	·		-
97	SPEC	6.2±.5	13.8+1.0	6.6±.5	7.0±.6	4.0+.5	+0.1±.3	6.36±.25	6.36±.25	6.36±.25	6.36±.25			1
	ACT	1.86	2.94	3,00	2.78	1.27	4.11	3.29	3.38	3,32	3.40	1		-
×	MPR	1.8±.5	3.0+.5	3.0+.5	2.35	1.2±.25	4.2+.5	3.45±.25	3.45±.25	3.45±.25	3.45±.25	F	7.	
	SPEC	1.8+5	3.0+.5	3.0+.5	2,3±.5	1.2±.25	4.2+.5	3.45±.25	3.45±.25	3.45±.25	3.45±.25	ı		= - : A
•	SER. NO.	2002	5007	2005	2002	2005	2002	040	041	042	043	<u> </u>	11	(2)
	UNIT	\$55	SPS	SPU	PSU	USO	GSSA/DOC	PR1	PR2	PR3	PR4	CABLES	TEST	CABLE

\*\*5DMSS-OLS-300, SCN 011, 20 Nov. 1987

291.01

292.36

286.78

298

TOTAL WEIGHT

(1) SERIAL NUMBERS ARE AS RECORDED ON DATA SHEET (2) A MASS ALLOCATION OF 6 LBS. HAS BEEN ASSIGNED FOR TEST CABLE FROM THE TOTAL OLS MASS ALLOCATION.

TEST CABLE IS PROVIDED AND CONTROLLED BY THE SPACECRAFT INTEGRATOR.

2-195

## TABLE 2

# GOVERNMENT FURNISHED PARTS SUPPLIED WITH OLS 12 SYSTEM SUMMARY OF WEIGHT AND CENTER GRAVITY

	ACT	3.62	3.66	3.72
두	W CONT	3.59	3.59	3.59
WEIGHT	MPR W/O CONT	3.34	3.33	3.33
	MAX SPEC	3.59	3.59	3.59
	ACT	2.18	2.18	2.20
	MPR	2.2±.1	2.2±.1	2.2+.1
	SPEC	2.2±.1	2.2±.1	2.79 2.2+.1
	ACT	2.75	2.75	2.79
+	MPR	2.7±.1	2.7±.1	2.7+.1
	SPEC	2.7±.1	2.7±.1	2.7+.1
	ACT	1.86	1.83	1.83
×	MPR -	1.8±.1	1.8+.1	1.8+.1
	SPEC	1.8±.1		1.8+.1
	UNIT SER. NO.	032	031	046
	UNIT	198	882	883

10.77 TOTAL WEIGHT

11.00

10.77

10.00

### 2.11 MASS (Cont'd)

### 2.11.2 <u>Component Mass</u> (3.4.2, 3.4.3)

The mass of the individual components of the OLS #12 AVE are tabulated below.

Component	<u>Spec</u>	Measured
SSS	59.0	54.64
SPS	70.0	69.00
SPU	18.0	17.32
PSU	27.0	26.31
OSU	4.0	3.52
GSSA/DOC	9.00	8.10
PR1	22.75	21.46
PR2	22.75	21.44
PR3	22.75	21.40
PR4	22.75	21.33
BB1	3.67	3.62
BB2	3.67	3.66
BB3	3.66	3.72
Cables	32.00	20.49

The cable figure does not include Special Sensor cables which are not supplied by WEC.

### 2.12 COOLER TRANSIENT MARGIN (3.2.2.5)

The T channel cone cooler transient cooldown cooling capacity margin is tested by the file 7x7.ST. An external power supply provides 1/2 Watt of heating to the outer stage of the cooler during a normal cooldown. On OLS #12 bearing retrofit, cone cooler S/N 024 successfully reached its operating set-point with 1/2 watt of external power applied, demonstrating the required margin.

### 2.13 DESIGN FEATURES

The following design features of the 5D-3 OLS are addressed in the analyses. The analyses are contained in the OLS 5D-3 System Summary Report. The requirements of the Design Features are met or exceeded in each category.

SUBJECT	SPEC. PARA.
DESIGN INTERFACES	(3.1.2)
RELIABILITY	(3.2.3)
WEAROUT/CONSUMPTION	(3.2.3.3)
STORAGE	(3.2.3.4)
CONTAMINATION CONTROL	(3.2.3.5)
CORROSION OF METAL PARTS	(3.2.3.6)
MAINTAINABILITY	(3.2.4)
INTERCHANGEABILITY	(3.3.5)

### 2.14 REDUNDANT AND FALLBACK SUBSYSTEMS (3.2.3.1 & 3.2.3.2)

Paragraph 3.2.3.1 of the development spec requires the OLS to incorporate the following redundant subsystems:

- (1) Along-Scan Gain Control
- (2) Main Bus Power Supplies
- (3) Data Processors
- (4) Memories
- (5) I/O Interfaces
  - a. Bus Controls
  - b. I/O Controls
  - c. S/C Interfaces
  - d. WOW/Flutter Signals/Clock Drivers
  - e. Drive Motor Controls
  - f. Sensor Controls
  - g. Gain Controls
  - h. Encoder Processors
- (6) Data Channels
  - a. Smooth Video Filters
  - b. Fine Video Filters
  - c. T-Channel Post Amplifiers
  - d. SDS Channel
  - e. SDF Channel
  - f. RTD Channel
  - g. Special Sensor Processors
  - h. Output Data Multiplexers
- (7) Output Switching Unit Oscillator and Clock Circuits
- (8) Digital Tape Recorders (Three of Four Required)

- (9) Output Data Channels (Three of Four Required)
  Paragraph 3.2.3.2 requires the OLS to incorporate the following
  fallback subsystems:
  - (1) IMC Shut-Off Mode.
  - (2) HRD Detector Single Segment Select, with associated electronics.
  - (3) T-Detector Single Segment Select, with associated electronics.
  - (4) Digital Generation of Delphi Scanner Clock.
  - (5) PMT Shut-Off Mode

Redundant and fallback subsystems are verified by test during the normal test flow either by repetition of the relevant test on the alternate subsystem (in the case of a redundant subsystem) or by tests designed to verify the specific subsystem (in the case of a fallback subsystem). In each case, any out-of-specs or anomalies are reported as part of the relevant Test Report paragraph.

### 2.15 ENVIRONMENT

The following environmental requirements are addressed in analyses and are contained in the OLS 5D-3 System Summary Report. The Thermal Vacuum (except CHA), Random Vibration, and shock requirements are verified by the successful completion of the approved Test Procedure.

The 5D-3 Environmental requirements of the Development Specification are met or exceeded by the OLS #12 AVE.

SUBJECT	SPEC. PARA
GROUND ENVIRONMENT	20.2.1
LAUNCH ENVIRONMENT	20.2.2
THERMAL VACUUM (CHA)	20.2.2.1
ACCELERATION	20.2.2.5
CHARGED PARTICLE ENVIRONMENT	20.2.2.6
LAUNCH PRESSURE PROFILE	20.2.2.8
ACCOUSTIC FIELD	20.2.2.9
TRANSPORATION & HANDLING ENVI	RONMENT 20.2.3

### 2.16 <u>ELECTROMAGNETIC COMPATABILITY</u> (3.3.2)

OLS #12 EMC testing was conducted per Westinghouse documents BVS 2049 (Block 5D-3 Electromagnetic Compatability Test Plan) and T928546 (Block 5D-3 Electromagnetic Interference Test Procedure) during the original OLS #12 testing. This testing was conducted in two phases. Phase one consisted of testing with the BTM SSS in the unpowered launch configuration on 3/4/85 thru 3/9/85. Phase two consisted of testing using the OLS #12 system conducted 8/22/85 thru 8/29/85. The results of this testing are reported separately in Volume V of the original Qualification Test Report. A summary of the EMC test results is included in table 2.16.1 - EMC Test Results. OLS #12 meets all DMSS-OLS-300 EMC requirements.

ATTACHMENTS: Table 2.16.1 - EMC Test Results

### TABLE 2.16.1

## ELECTROMAGNETIC COMPATABILITY TEST RESULTS

		<u></u>	
Test	T928546 Test Procedure Section	Data Complete	Result
Expose unpowered BTM SSS with all covers and room Temperature T Detector installed to 200 V/m for 5 minutes	5.3	03/09/85	SSS operated correctly following exposure
Measure OLS-12 Radiated Emissions system signature	6.4	08/26/85	Signature measurement completed. This test is for information only - no limits are specified.
Expose operating OLS #12 SSS to .25V/m for 5 min.	5.4	08/28/86	System operated correctly at ambient during and following exposure.
Expose unpowered OLS #12 SSS with all covers installed to 12.5V/m for 5 min.	5.5	08/29/86	System operated correctly in ambient and in post EMI TV testing following exposure. No failures traceable to EMI testing occured.

### 3.0 INTERFACE SPECIFICATION REQUIREMENTS

Electrical Interface parameters are measured in the OLS Detailed Electrical Test (T927989) and the OLS Special Sensor Detailed Electrical Test (T927992). These tests demonstrated conformance with all applicable Interface Specification requirements. The only Interface related system measurements that vary significantly from system to system are the SSS Alignment axes which are included here.

### 3.1 SSS ALIGNMENT AXES

The OLS #12 SSS Reference Plane axes are within the specification allowances. The results of system test and calculation are given below. The designations are those in Interface Spec IS-YD-810, para. 3.2.7ff.

### SECONDARY REFERENCE AXES TO PRIMARY AXES

### SECONDARY REFERENCES AXES TO MOUNTING (INTERFACE) AXES

 $X_{p-p} = 0.764 \text{ mrad} = 158 \text{ arc sec}$ 

 $X_{R-M} = 0.780 \text{ mrad} = 161 \text{ arc sec}$ 

 $Y_{R-P} = 0.467 \text{ mrad} = 96 \text{ arc sec}$ 

 $Y_{R-M} = 0.496 \text{ mrad} = 102 \text{ arc sec}$ 

 $Z_{R-P} = 0.626 \text{ mrad} = 129 \text{ arc sec}$ 

 $Z_{R-M} = 0.658 \text{ mrad} = 136 \text{ arc sec}$ 

These are within the specification limits of 600 arc seconds.

The Mounting (Interface) Axes to Primary axes are also calculated, using the computer program REFPLN and are given below:

 $X_{M-P} = 0.027 \text{ mrad} = 67 \text{ arc sec}$ 

 $Y_{M-P} = 0.272 \text{ mrad} = 56 \text{ arc sec}$ 

 $Z_{M-P} = 0.270 \text{ mrad} = 56 \text{ arc sec}$ 

These are within the specification limits of 120 arc seconds.

### DISTRIBUTION

- J. Spangler
- M. Barrett

- R. Bark
- R. Raum
- P. Kiefer V. Williams
- G. Pollock
- 8. Spencer
- R. Lieske
- J. Scilipoti

DATE	<u>05</u>	November	199	
TOTOTNAT	-00		10	

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APPROVED, Q&RA RWE

APPROVED, ENGRG

REVISION

BVS

### APPENDIX A

BEARING RETROFIT AND RETEST PLAN FOR OLS 12 THRU 16

Support and Services Contract No. F04701-90-C-0028

Prepared for

UNITED STATES AIR FORCE Headquarters, Space Systems Division Los Angeles, California

Prepared by

WESTINGHOUSE ELECTRIC CORPORATION Defense and Electronics Center Baltimore, Maryland

BVS No.: 2579

### REVISION SHEET & NOTES PAGE

NOTICE: Unless otherwise instructed, the marked-up pages showing actual changes incorporated in a new Rev. will be maintained in the BVS Master File for future reference and the remainder of the document will be discarded when the new Rev. is filed.

REVISION LETTER	REVISION DATE	AFFECTED PAGES	REVISION MADE BY
A B C D E F G H J K L	4/6/90 8/7/90 8/26/90 9/21/90 2/20/91 2/28/91 5/20/91 5/22/91 6/28/91 8/29/91	1,2, *3-6, 7-11, *12, 13-16, *17-20, 21, 22 1, 2, 16, 16a 1, 2, 16b, 17 1, 2, 5, 6, 18, 21 1, 2, 17-23 1, 2, 17 22, 23 22 1-3, 22-23 1-2, 18, 22-23 1, 2, 23, 24	J. SMUTKO D. OMETZ J. SMUTKO G. POLLOCK R. BARK R. BARK M. BARRETT M. BARRETT SCILIPOTI SCILIPOTI G. POLLOCK

(\* INDICATES PAGE # IS ONLY CHANGE TO THAT PAGE)

Notes:

WP51\JSm01.dl

### 1.0 INTRODUCTION

This document describes the detailed rework and test verification plan for replacing the oscillating assembly bearings in the SSS with new bearings having improved lubrication (Nye 188B).

Included are step-by-step instructions, with check-off lines for all inspection, mechanical, optical and electrical test operations.

### 2.0 REFERENCES

This document references the following Westinghouse procedures:

SSS Assembly Procedure SSS Handling Procedure 9RA3681 9RA4026

9TA9354 Mechanical Operations for SSS Oscillating Assembly

Bearing Retrofit

T927002 SSS/DME Compatibility Test Specification

OLS System Acceptance Test Procedure T927686

OLS Program Directives:

PD 024

PD 026

PD 027

PD 030 PD 044

PD 045

PD 055

PQL 0735 Vibration Procedure

### 3.0 REWORK AND TEST PLAN FOR OLS:

Charge labor for this effort to the Block 5 Support and Services Δ 3.1 contract. Present G.O. number, valid thru 9/30/91, is 53741. Task assignments are as follows:

BAAA Non-recurring Engineering

BABA OLS-12

BACA OLS-14

BADA OLS-15

BAEA OLS-16

Material and Travel G.O. is 53742.

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- 3.2 The modification is accomplished by working revision notice G931B.
  - This consists of replacing the two bearing pairs in oscillating Assembly 623R765.
  - Special instructions have been written to supplement the RNs and describe the mechanical operations necessary to retrofit scanner bearings after an SSS has been fully assembled, see 9TA9354.
- 3.3 For those systems in the field, return the system to WEC, Baltimore for rework and retest. Follow all applicable handling procedures including Program Directives 024, 026 and 027. As an option, the PSU and SSS only may be returned if another system is available to support the retest effort.

3.4			AND SYSTEM TESTS.		
	<u>OPERAT</u>	ION		VERIFICATION	DATE
	Unpack			<u>G.S.</u>	9/21/90
	Record	serial nos. of	f rcvd. units:		•
	SSS	640R800G08	S/N .5007		
	or	758R750G0 <u>м/A</u>	S/N		
	PSU		S/N		
	SPS	651R390G0	S/N		55
	SPU	758R040G0	S/N		
	OSU	640R960G0	S/N		
				59	
	22 - 2712				
		(2) (1) (1) (1) (1) (1) (1) (1) (1) (1) (1	<u> </u>		
			•		
	Attach	copy of incomi 49 to this BVS	ng		
	purp		ioi colletol	-G.S.	9/21/90

WEC Receiving Inspection

AFPRO Inspection

Baseline Electrical Tests - Deleted

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### 3.5 OPTICAL ALIGNMENT BASELINE MEASUREMENTS

Prior to bearing retrofit certain optical tests must be performed in order to accumulate baseline data with which to compare readings taken after the work is completed. This is necessary so that alignment integrity can be verified after the SSS has been partially disassembled and reassembled. These tests will be the same as some of the tests performed in 9RA3681 "Assembly and alignment procedure SSS assembly". However, there may be slight differences in technique because the assembly status of the SSS will not be exactly the same as in the normal building sequence. The steps herein, then, will be excerpts from that procedure and all step numbers referred to will be taken from 9RA3681.

Data should be noted in the applicable flight log book and used for post retrofit alignment comparisons.

In order to perform the necessary tests, the PMT and HRD detector must be removed. The spring assemblies must also be tied in order to permit positioning of the telescope.

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### STEPS FROM 9RA3681 TO BE PERFORMED

STEPS		VERIFICATION DATE
8.0	"Adjustment of optical alignment, test and integration facility" - prepares the facility for required tests.	With Near Jock 90
9.0	"SSS assembly mounting procedure" - Mounts the SSS to the test facility reference interface for testing.	20 1/2-160x/ 10+90
11	"Alignment of the oscillating assembly rotating axis with the Moore table axis." - positions the SSS for optical measurements.	With May 3 out 90
12	"Mounting interface alignment measure- ments" - Determines SSS reference axis position in relation to the OATIF mirrors/SSS mounting interface.	With Monf 50ct 90
	A test will be performed to determine the position of the HRD detector prior to removal as follows:  Clamp a Gaertner bench microscope to the T/T table aligning the microscope reticle with the T/T reticle. Observing the HRD detector through the microscope, center the reticle on at least 2 corners of each segment of the detector. Note the T/T Y and Z axis positions for each point observed.	22/h-Mon/ 30ct 90
18	"Oscillating assembly transmission test" - Determines % transmission of telescope prior to retrofit.	Withelian 30ct 90
	Inspect mirror M1 to determine if a scatter test should be performed. If, on inspection, M1 appears very dirty perform a scatter test per step 19 of 9RA3681.	With hon 30ct 90
		-

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	10	
STEPS		<u>VERIFICATION</u> <u>DATE</u>
20.4	"HRD detector alignment check" - checks alignment of the HRD in relation to the ORA field splitter. Illumination via the PMT light as described in step 20.4.5 should not be necessary. If the light from the T/T point source is insufficient to view the HRD the PMT must be removed prior to performing step 20.4.	WHACKSonf 4 Oct 90
	Before proceeding remove the HRD and PMT if still installed. Mount the PMT and HRD reticles.	With least 4 oct 90
14.2 to 14.3	"M1 centering test" - verifies that the optical beam is centered on M1 prior to retrofit.	WHIN- 700 50ct 90
16.1.1 to 16.1.4, 16.1.10, & 16.1.19 to 16.1.26	"T-Cal alignment" Determines pre- retrofit T-Cal end of scan position. Make no adjustments.	Whileof 80ct 90
16.2.1 to 16.2.4, 16.2.11, 16.2.20 to 16.2.27, & 16.2.29	"T-Clamp alignment" - Determines pre- retrofit T-Clamp end of scan position. Make no adjustments.	Whaleof 80ct 90
15.2.2 to 15.2.7 & 15.2.16 to 15.2.18	"Encoder optics alignment" - Determines max. and min. clock voltages. The facet from 15.1.15 referred to in steps 15.2.6 and 15.2.7 will be assumed to be facet 8.	WHAtlan 18 Oct 90
15.2.19.6 to 15.2.19.8	Defines minimum allowable voltages and angular displacement on the faceted ring.	With Short 80ctor

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<u>STEPS</u>		<u>VERIFICATION</u>	DATE
15.3.1 to 15.3.6 Note: Make NO adjust- ment in 15.3.6	"Encoder nadir adjustment" - Determines pre-retrofit position of the encoder at the Nadir position W/R to target translater position.	W.Ahelloon	8actap
15.3.12	"Encoder nadir alignment error" - Provides a formula for determining nadir alignment error.	With land	80ct90
15.4.1 to 15.4.26 Omit steps 15.4.18, 15.4.19, 15.4.20 & 15.4.23	"Encoder linearity and signal amplitude measurements." - Determine pre-retrofit position for facets of the encoder W/R to the target translator. In steps 15.4.17, 15.4.22 and 15.4.25 only a sampling of the numbered pulses shall be taken. Sample pulses 15, 60, 97 and 142.	WH Mileon	
15.5 Note: Make <u>NO</u> adjust- ments in 15.5.12, 15.5.13 or 15.5.16	"Back-up auxiliary encoder alignment." - Determines pre-retrofit electro- optical position of the back-up aux. encoder.	With Heart	
15.6.1 to 15.6.3	"SSS Preparation for primary aux. encoder alignment." - Prepares SSS for testing of primary aux. encoder.	A WHM- Keen	
15.9 Note: Make <u>NO</u> adjust- ment in 15.9.13 or 15.9.16	"Primary aux. encoder alignment" - Performs tests to determine electro- optical position of the primary aux. encoder.	WHI Leay	

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**STEPS** 

<u>VERIFICATION</u> DATE

15.1.1 to 15.1.15 Omit step 15.1.14

"Faceted ring angular measurements"

- Determines optical positions of
the facets of the polygon ring. In
steps 15.1.4, 15.1.8, 15.1.10 and
15.1.13 where the step refers to
specific facets, it shall be required
to perform the procedure only on a
sample of the facets. A facet shall
be chosen at the beginning, the
middle and as near to the end as can
be seen. Perform the procedure on
facets 1, 8 and 14 if these are
accessible. Step 15.1.14 will be
omitted and in step 15.1.15 the facet
closest to the mean facet Y axis
reading will be assumed to be facet 8.

Cover the HRD and PMT reticles and ORA parts with lens tissue. Data should be noted in the applicable flight log book and used for post retrofit alignment comparisons.

Mr Willam 100ct 90

Withelland 10 oct 90

#### 3.6 MECHANICAL OPERATIONS

Perform the operations outlined in 9TA9354. This procedure describes all the mechanical operations necessary to retrofit scanner bearings after an SSS has been fully assembled. This step-by-step procedure includes check-off lines for each operation and inspection point.

Following completion of the procedure, attach the working copy of 9TA9354 to this BVS.

Verification of completion
Inspection

<u>G.J.S.</u> 1-78-91

#### 3.7 OPTICAL RE-ALIGNMENT

Optical Tests after Bearing Retrofit

After the bearing retrofit certain optical tests from 9RA3681 must be performed both for comparison to baseline tests as well as to ensure the unit is ready for integration tests. The bulk of these post bearing retrofit tests are the same as the optical baseline tests discussed in section 3.5. Record data in the applicable system SSS log book.

In order to perform the necessary tests, the PMT and HRD detector must be removed. The spring assemblies must also be tied in order to permit positioning of the telescope.

STEPS		VERIFICATION	DATE
8.0	"Adjustment of optical alignment, test and integration facility" - prepares the facility for required		
a: 8	tests.	inthe front	10112120
9.0	"SSS assembly mounting procedure" - Mounts the SSS to the test facility reference interface for testing.	With Thori	12 1120
11	"Alignment of the oscillating assembly rotating axis with the Moore table axis." - positions the SSS for optical measurements.	With Then	2 <u>0 Nav.</u> 90
12	"Mounting interface alignment measurements" - Determines SSS reference axis position in relation to the OATIF mirrors/SSS mounting interface.	Will-short	20 Nov. 90
13.0 to 13.3,	"M3, M5 and M5 Mask Alignment" - Determines whether adjustments are needed in M3 and M5. (Hopefully M5 will not need alignment. If it does, consult an optical engineer because the ORA must be removed to align M5. Then it will be necessary to realign the ORA when it is reinstalled).	Brstle 1/_ 1	
18	"Oscillating assembly transmission test" - Determines % transmission of telescope.	Million .	20 Nov.90
	Inspect mirror MI to determine if a scatter test should be performed. If, on inspection, MI appears very dirty perform a scatter test per step 19 of 9RA3681.	Wolfe Theory	<u> 20 Novi</u> . 90

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STEPS	10	<u>VERIFICATION</u>	DATE
20.4	"HRD detector alignment check" - checks alignment of the HRD in relation to the ORA field splitter. Illumination via the PMT light as described in step		
	20.4.5 should not be necessary. If the light from the T/T point source is insufficient to view the HRD the PMT must be removed prior to performing step		
	20.4.	WHI Soon	20 Nov. 90
	Before proceeding remove the HRD and PMT if still installed. Mount the PMT and HRD reticles.	NA WHA	
14.2 to 14.3	"M1 centering test" - verifies that the optical beam is centered on M1.	WHI Chank	20 Nov.90
16.1.1 to 16.1.4, 16.1.10, & 16.1.19 to	"T-Cal alignment" Determines T-Cal end of scan position. Make no adjustments.	- Time	
16.1.26		with But	27 11-17-
16.2.1 to 16.2.4, 16.2.11, 16.2.20 to	"T-Clamp alignment" - Determines T-Clamp end of scan position. Make no adjustments.		
16.2.27, & 16.2.29		with the of	=- 1 Walter

STEPS		VERIFICATION	DATE
15.2.2 to 15.2.7 & 15.2.16 to 15.2.18	"Encoder optics alignment" - Determines max. and min. clock voltages. The facet from 15.1.15 referred to in steps 15.2.6 and 15.2.7 will be assumed to be facet 8.	WHI Hand	2211-19
15.2.19.6 to 15.2.19.8	Defines minimum allowable voltages and angular displacement on the faceted ring.	Mithiland	20 110120
15.3.1 to 15.3.6 Note: Make <u>NO</u> adjust- ment in 15.3.6	"Encoder nadir adjustment" - Determines position of the encoder at the Nadir position W/R to target translater position.		7 / 11 / 1 <sub>2</sub>
15.3.12	"Encoder nadir alignment error" - Provides a formula for determining nadir alignment error.	Diffichent	
15.4.1 to 15.4.26 Omit steps 15.4.18, 15.4.19, 15.4.20 & 15.4.23	"Encoder linearity and signal amplitude measurements." - Determine position for facets of the encoder W/R to the target translator. In steps 15.4.17, 15.4.22 and 15.4.25 only a sampling of the numbered pulses shall be taken. Sample pulses 15, 60, 97 and 142.	noth-Visery	
15.5 Note: Make <u>NO</u> adjust- ments in 15.5.12, 15.5.13 or 15.5.16	"Back-up auxiliary encoder alignment." - Determines electro-optical position of the back-up aux. encoder.	White Soas	11 1920 900
15.6.1 to 15.6.3	"SSS Preparation for primary aux. encoder alignment." - Prepares SSS for testing of primary aux. encoder.	Win Shari	11 130 900
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**STEPS** <u>VERIFICATION</u> DATE 15.9 "Primary aux. encoder alignment" -Note: Make Performs tests to determine electro-NO adjustoptical position of the primary aux. ment in encoder. 15.9.13 or 15.9.16 15.1.1 to "Faceted ring angular measurements" 15.1.15 - Determines optical positions of Omit step the facets of the polygon ring. In steps 15.1.4, 15.1.8, 15.1.10 and 15.1.14 15.1.13 where the step refers to specific facets, it shall be required to perform the procedure only on a sample of the facets. A facet shall be chosen at the beginning, the middle and as near to the end as can be seen. Perform the procedure on facets 1, 8 and 14 if these are accessible. Step 15.1.14 will be omitted and in step 15.1.15 the facet closest to the mean facet Y axis reading will be assumed to be facet 8. Reinstall HRD & PMT (OLS-16 see next page, 16a) A test will be performed to determine the position of the HRD detector after reinstallation as follows: Clamp a Gaertner bench microscope to the T/T table aligning the microscope reticle with the T/T reticle. Observing the HRD detector through the microscope, center the reticle on at least 2 corners of each segment of the detector. Note the T/T Y and Z axis positions for each point observed. Inspection

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## 3.7.1 PMT ASSEMBLY SPECTRAL RESPONSE STABILITY CHECK - OLS-16 Only

With the PMT assembly removed during the optical re-alignment, a check of the PMT spectral response will be performed to check spectral stability for any evidence of a shift since the last PMT spectral response made on 07/30/88.

The test will be performed in accordance with T-361A88, test paragraph 9.12 - Spectral Response and Effective Sensitivity.

#### STEPS TO BE PERFORMED

**STEPS** 

		VERTICATION / DATE
Inspection of PMT (Damage Install in Transport Case	Verification)	N/A_
Spectral Response Test from	m T-361A88, Para. 9.12	2 second that I have dentited
Inspect PMT prior to SSS I	nstallation for Damage	Secure Control of the
Reinstall PMT on SSS		
Inspection ( <u>W</u> & DPRO)	*	
3.7.2 THERMAL BLANKET UPGRA	ADE PER ECP-28 - OLS-16 only	
In place of th OLS-16, install the insulation:	e thermal blanket hardware following oscillating assemb	originally installed on ly insulation covers and
		VERIFICATION, DATE
	540R561G01 540R561G02	N/A
1 Vien.	540R562H01	
	540R562H02	11
	540R563H01	
Inspection		
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#### 3.7.2 (Cont'd.)

Install ECP-25 upgraded bracket on the 1A8 HRD/PMT Postamplifier Assembly as follows:

<u>STEPS</u>	VERIFICATION	DATE
Remove the cover assembly, 644R288, from the HRD/PMT Postamplifier, 644R220 located on the SSS Be careful not to disturb the potentiometer adjustments. It may be necessary to cut the RTV used to stake the potentiometers if it has adhered to the cover.	NA	
On the cover assembly, 644R288G01, replace mount 522R838G01 and 432R269G01 with items 16 and 17 respectively on the Thermal Blanket Retrofit drawing, 765R630. (540R564G01 and 540R584G01) Re-mark the cover assembly to 644R288G02.	official file for the same of	in Sin
Inspect modified cover	-1 =195 III====	
Inspect 644R220, OK to re-install cover, WEC & DPRO	EXPLOYED WITH THE	
Install modified cover on 1A8 postamplifer	F	
Inspection (Note - retest of 644R220 postamplifier not required, will be tested at system level).		

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STEPS		VERIFICATION	DATE
3.8	SSS TEST PER T927002		= = =
	Disconnect SSS main cable connector 1A9P2	2,7	2/6/91
	Perform the test procedures of T927002 including 50 hour bearing confidence test, scanner centering, scanner frequency, and limit switch adjustments if required. It is not necessary to repeat paragraph 4.16,		2/2/2
	T detector bias current measurement.	0,0,	2/2//
	Reconnect 1A9P2	Ju,	2/6/9
	Data Review		2/21/91
	Inspection	· ·	2/27/91
*3.8.1	Perform encoder optics ambient funtional test 4.12.4 TECH (Dobe 2/21/91)	per T927002, pa	ragraph
*3.8.2	Apply additional adhesive to encoder optics as	sembly per RN GL5	4D.
	DATE 2-27-91 G.J.S. MANNF INSP NOTE: Note after 24 hours the SSS may be remore fixture and installed on the base plate,	DPRO the hand	
*3.8.3	Seven day cure at room temperature.		
	DATE COMPLETE 3-6-91		
*3.8.4	Reinstall cover and torque screws to 4 to 6 in,  GIS MANUF  DATE 3-8-91	′lbs.	
*3.8.5	Repeat step 3.8.1	20	
	TECH		8.

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\*NOTE: For OLS 16 perform this action after completion of paragraph 3.13 of this BVS.

### **STEPS** AMBIENT SYSTEM TESTS 3.9 QUICKTESTN.ST QUICKTESTR.ST 6X2X1.ST AHC11PT.ST - R APC11PT.ST - P AHSFB11PT.ST - R 6X3X1.ST - P 6X3X2.ST - R 6X3X5.ST - R MHC11PT.ST - R 6X5X1.ST - P 6X7X1.ST 6X7X2.ST

6X9.ST 7X8.ST Data Review

VERIFICATION	DATE
DIA DIA RIV	5/2/9/08 5/2/9/08 5/2/9/08 5/2/9/08 5/2/9/08 5/2/9/08
RMC RMC RMC DIA	5/22/91 on 5/22/91 re-10 5/22/91 ex 5/22/91 on 5/22/91 on
DEP	5/23/91 OK 5/23/91

STEPS		VEDIETCATION	DATE II
7		VERIFICATION	DATE
3.10	THERMAL VACUUM ADJUST		
	Deleted		
3.11	<u>VIBRATION</u> - SSS Only		
	Inspection per PD045 checkpoint 3a checkpoint 3b		5/23/91
	Notify AFPRO Waived pur S. Kink	OF	5/23/91
	Vibrate SSS, 3 Axis, acceptance level per T927686 para. 3.5	RAS/	5/23/91
es in	WEC Inspection per PD 055	(50)	5/24/91
	MAIVED PER C. STUM	P (IZI)	15

PD 045 Checkpoint #4

#### 3.12 POST-VIBRATION, AMBIENT

Perform the test procedures of T927686 paragraph 3.6 except delete paragraph 3.6.5 and in paragraph 3.6.3, only the following test files need to be run:

#### NOMINAL CONFIGURATION TESTS

2			<u>VERIFICATION</u>	DATE
	mry 5X1X1.ST	Primary Configuration Power	DICH	5/24/91
	may 5X1X4.ST	Load Operational Prog Processor C	DIA	5/24/91
	5X1X6.ST	Initialize Primary Configuration	DH4	5/24/91
	may 5X2X1.ST	Quiescent Mode Power	SM	524/91 OK
	my 5X2X2.ST	Primary Configuration	Ath-	1/2 H/91 ON
	my 5X2X3.ST	Primary Configuration Dual I/O	DIH.	\$24/91 OK
	5X2X4.ST	Primary Config. Dual Formatter	DNA	3/24/91 OK
	SX2X5.ST	Primary Config. Dual Formatters & I/C	D	Shalgion
	MA9 5X3X1.ST	Primary Configuration EST Check	M-	5/2/9/120-12
	mg 5X5X1.ST	HRD Analog Test	214	5/27/91 OK
	5X5X2.ST	PMT Analog Test	Ost ,	5/27/91 OK
	2004 5X5X3.ST	T-Channel Analog Test	YIX	5/27/91 OK
	747 5X6X1.ST	PGC	RMP	3/27/91 OK
	may 5X6X2.ST	ATGC	RMC	5/27/91 OK
_	mg 5X6X3.ST	ASGC Ø(	W BAC	5/28/9/ OK
	mor 5X12X1.ST	Scanner Functional	DH -	5/34/9, ox
	V			-

#### REDUNDANT CONFIGURATION TESTS

	×	VERIFICATION	DATE
mg 5X1X2.ST	Redundant Configuration Power	DyH-	5/24/91
meg 5X1X5.ST	Load Operational Prog Processor D	211	5/24/91
5X1X7.ST	Initialize Redundant Configuration	Alst	5/24/91
71/19 5X3X2.ST	Redundant Configuration EST Check	DIA	JD8/9/126-1:
5X6X4.ST	PGC	RMC	5/28/91 OK
MG 5X6X5.ST	ATGC	RMC	5/28/91 OK
MOGY 5X6X6.ST	ASGC (QC)	RATE	5/28/91 01
7009 5X12X2.5	Scanner Functional	DH-	5/24/91 OK
(3.6.4)			
MA 6.1	IMC HRD A/S - Redundant - AHSF3PTI.ST	DIA_	J-7/9, ox
mg, 6.1.1	HRD A/S - Redundant - AHSF7PT.ST	DSH	5/27/91 OK
mg 6.1.2	PMT A/S - Primary - APC7PT.ST	RME	5/28/91 OM
mog 6.1.4	Backup Encoder HRD Sync AHSFB9PT.ST	RMC RMC	5/25/91-5/29/91
6X2X1.ST	L DC Response (must precede 6X3X1.ST)	D34-	5/27/91 0x
mg 6X2X4.ST	T Chan Elec DC Response	DEP	5/28/91 OK
may 6X3X1.ST	L Chan Dark Noise - Primary	DIA	5/27/91 OK
my 6.4.1	HRD MTF - Primary - MHA7PT.ST	DM-	5/2/9/ RC-10
mpg 6.4.2	PMT MTF - Redundant - MPA7PT.ST	JAH.	5/28/91 or
6X4X3A.ST	Ambient T MTF - Redundant	DIA	D\$/41 ox
mag 6X6X2.ST	PMT CAL	SM	5/27/91 or
mpg 6X7X2.ST	990 Test	DIA	JD 19100
TX5.ST	Actuator EST Test	DAA	18/91 ar.
Mag 7x8.5T	Scanner Rest	DIA	5/24 91 OK

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#### 3.13 Thermal Vacuum Acceptance Test

Perform the Thermal Vacuum Test procedures per T927686, paragraph 3.7. Do not install encrypters, use the BBT simulator box. Attach the system test log sheets to this BVS.

The following tests may be deleted from T927686 for this retest:

In paragraph 3.7.3.1, delete the following test files -

5X4X2 Core Test 5X4X3 Core Test 5X16X2 A/D Test 5X16X4 A/D Test

Do a T channel electrical adjust by adding test "TSET.ST" to paragraph 3.7.3.2.

Delete paragraphs 3.7.3.4 and 3.7.3.5.

Add one day of T stabilization testing at the completion of Optics Limit with ml=-8 by performing the following at approximately 2 hour intervals:

Execute 6X2X3.ST by entering "DSK 6X2X3.ST". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

In paragraphs 3.7.5.1.1, 3.7.10.3, 3.7.11.3, and 3.7.12.2, delete the following test files -

5X2X16
BB Signature
5X4X1,2,3,4
Core Tests
5X8X1,2
DMDM

5X10X1,2,3,4
Output Data Switching
5X14X1,2,3,4
SSP Formatter Tests
5X16X1,2,3,4
A/D Tests

ADD 4x9x1.57 to P3.7.5.1.2 logors and almost or some tote

Add a one day nominal temperature T channel stability test between the two soak cycles by performing the following at approximately 2 hour intervals:

Execute 6X2X3.ST by entering "DSK 6X2X3.ST". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

Delete paragraph 3.7.12.6)7

Add a day at the beginning and a day at the end of the nominal temperature plateau for additional T channel stability by performing the following at approximately 2 hour intervals:

(JSm01.dl)

Δ

4

-22-

Add a day at the beginning and a day at the end of the nominal temperature plateau for additional T channel stability by performing the following at approximately 2 hour intervals:

Execute 6X2X3.ST by entering "DSK 6X2X3.ST". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

<u>VERIFICATION</u> DATE Verify a minimum of 500 hours of scanner operation with new bearings has been performed in vacuum. Any deficiency should be made up at this time. Verify completion of Thermal Vacuum SPS Coax Connector repair on JIO due to defective female contact per NR 20250959. Remove top cover of SPS (MFG) - Remove 2 P.C. boards - A241 & A242 (MFG) - Remove 640 R913G01 cable (MFG) - Remove coax connector J10 and replace with new connector (MFG) - Circuit check (TEST) - WEC Insp. (INSP) NEW SELECT - DPRO Insp. 11/5/91 (DPRO PER M. LITTLE INSP) - OK to reinstall 913 cable in SPS - Reinstall boards A241 & A242 - OK to cover Insp. - WEC Insp. - DPRO Insp. NOW SELECT PER C. STUMP. - Install cover - WEC Insp.

3.16 Due to male contact pin damage on cables 644R329G02 and G03, replace OLS-12 cables with OLS-14 cables.

- Reinstall buffer connector

(JSm01.dl)

3.15

-23-

#### 3.17 Final Ambient

For OLS-16 perform additional adhesive operation pr RN GL54D prior to final ambient. See paragraph 3.8.1 thru 3.8.5.

For OLS 12 only perform the following post coax connector repair tests:

**NEWON1.ST** 

QKTESTN.ST

5X10X1SS.ST

**NEWON2.ST** 

OKTESTR.ST

5X10X2SS.ST

Perform T927686 paragraph 3.8, Final Scan Plane Definition.

Perform T927686 paragraph 3.9, Inspection, Data Review, and Preparation for Shipment except Delete paragraph 3.9.2, Weight and Center of Gravity.

Pin Retention Inspection Data Review

**AFPRO** 

Pack

Ship

11/8/97 11/8/97 11/8/97 11/8/97 11/8/97

### OLS PROGRAM DIRECTIVE

-CONTINUATION SHEET-

DIRECTIVE NO. \_\_\_045

DATE 12/12/88

# CHECKPOINT 3.a (Before Vibration) Blue Room

	· · · · · ·		
		,	RESP
Α.	T channel adjustment tool was removed from PSU per Program Directive #046. PSU was	WZA	MFG
	inspected before and after cover was installed.	1 <i>4/1</i> 21	INSP
В.	SSS and PSU pots are staked per PS 82560SA.	हिंग्डी	MFG
		<b></b>	INSP
C.	SSS mirrors are staked.	EJS.	MFG
	THE RESIDENCE OF THE PARTY OF T	12	INSP
D.	Thrust bearing (9RA3893) and limit switch assembly 758R962 have been removed for vibra-	45	MFG
20	tion. Buffer connectors are to remain installed.	( <u>33)</u>	INSP
ε.	Pre-vibration inspection of the SSS, SPS, SPU, PSU, OSU, GSSA, GSSB and blankets has been performed. (NOTE: If the covers have not been removed from the SPS, SPU and OSU since unit verification at Checkpoint 2.a, step C, perform only external pre-vibration inspection; do not	الله المالية ا	INSP
	remove covers.).		
F.	Presented to Air Force for pre-vibration inspection. Warred J. Kirk DRRs.	(Titaly	INSP

#### NOTE 1:

WHEN APPLICABLE, RECORD ACTIONS PERFORMED IN SYSTEM LOG BOOK.

NOTE 2:

"CAUTION"

BUMP, SHOCK,

ABRUPT MOVEMENT

OF TAPE RECORDERS

CAN CAUSE SEVERE

DAMAGE.

Page 8 of 14

inly. INSP

Protective connector covers are in place on all

unit external connectors and the SSS.

#### OLS PROGRAM DIRECTIVE

-CONTINUATION

SHEET-

DIRECTIVE NO. <u>045</u>

DATE \_ 12/12/88

## CHECKPOINT 3.b (Before Vibration, in PQL)

<u>RESP</u>

A. Vibration area is clean and PQL procedures 735 and 737 followed.

132K

QE(PQL)

B. Vibration test equipment is within calibration date.

M

QE(PQL)

C. System monitoring equipment is within calibration date.

A Section

QE

 Clean room hats, gowns, masks and gloves are available and in use.

0E

NOTE 1: WHEN AP

WHEN APPLICABLE,
RECORD ACTIONS
PERFORMED IN SYSTEM |
LOG BOOK.

NOTE 2:

"CAUTION"
BUMP, SHOCK,
ABRUPT MOVEMENT
OF TAPE RECORDERS
CAN CAUSE SEVERE
DAMAGE.

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### OLS PROGRAM DIRECTIVE

-CONTINUATION SHEET-

DIRECTIVE NO. \_\_045

DATE \_\_12/12/88

## CHECKPOINT #4 (After Vibration)

	3.		RESP
A.	Westinghouse and Air Force post vibration inspection per BVS PD 055 has been performed on all assembles, including GSSA, GSSB and Blankets. (Verify that SSS and PSU pots are staked; or stake per PS 82560SA)	(E)	INSP
8.	The thrust bearing (9RA3893) and limit switch assembly 758R962 have been installed for thermal vacuum acceptance test.	111	MFG INSP
C.	Area is clean and contains no miscellaneous parts or extraneous hardware.		QE
D.	Anti-static mats and wrist straps are in place and ready for use.	I TO	QE
E.	Test equipment checked per Program Directive #022 less paragraphs IID and IIE. 633		QE <sub>.</sub>
F.	Perform Test Equipment Operational check per PD 022, paragraphs IID and IIE.	TEP	TD
G.	The SPS, SPU, PSU, OSU, TCP, Recorders and BB's are interconnected with system cable and ground bus per 9R07845. Handling Procedures 9RA4220, 9RA4225 and 9RA4026 were followed.	THE	TD

	NOTE 1:
	WHEN APPLICABLE,
;	RECORD ACTIONS
	PERFORMED IN SYSTEM
	LOG BOOK.

NOTE 2:

"CAUTION"

BUMP, SHOCK,

ABRUPT MOVEMENT

OF TAPE RECORDERS

CAN CAUSE SEVERE

DAMAGE.

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d/01320

that it selections are also the selections DISTRIBUTION:

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8VS 2600 DATE 19 February 91

ORIGINATOR M. Epperly

QUALITY ASSURANCE RW

REVISION\_

RDS REWORK AND RETEST PROCEDURE

For OLS 12, 13, 14, 15 and 16

ols 12

Contract F04701-90-C-0028

Prepared For

UNITED STATES AIR FORCE Headquarters, Space Division Los Angeles, California

Prepared By

WESTINGHOUSE ELECTRIC CORPORATION Electronics Systems Group Baltimore, Maryland

#### REVISION SHEET

Revision Letter	Revision Date	Affected Pages	Revision Made by
. 0.	25 Jul 90	Released	
Α	14 Aug 90	Pages 2 and 13-27	- W
В	5 Oct 90	Pages 2-5	Rennenkampf
С	19 Feb 91	Pages 1, 2 and 13-27	Epperly
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WPF EP.lah	PAGE	FSCM NO	DOCUMENT NUMBER	REV
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1.0 <u>Introduction</u>

This document describes the detailed rework and retest plan for SPS and OSU units returned from the field for incorporation of Real-time Data Smooth. A copy of this document will serve as a checklist for accomplishing the rework and retest procedures.

Incorporation of RDS into an OLS requires the modification of the following assemblies:

9C and SDF-5 boards in the SPS SPS Matrix Plate OSU-1 and OSU-2 boards in the OSU OSU Matrix Plate OSU Top Cover Assembly

Record Serialization of Units to be reworked here:

 OLS 536R500G o/
 5007

 SPS 651R390G o/
 5007

 OSU 640R960G o/
 5007

Record Assembly Serial Numbers here:

640163701703	SPS Mother Plate Assy Prime 9C Board Redundant 9C Board Prime SDF-5 Board Redundant SDF-5 Board	(651R342) (775R076 or 775R077) (775R076 or 775R077) (775R078 or 775R079) (775R078 or 775R079)	SN SN SN SN SN	0001 5013 5014 5012 5013
640R522G0S 640R524G03	OSU Mother Plate Assy OSU Top Cover Assy OSU-1 Board OSU-2 Board	(522R783) (644R046) (775R080) (775R081)	SN SN SN SN	3075-0001 5007 5007 5007

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E Mine	23(1)	120010	313 2333	

#### 2.0 Rework and Assembly Retest Plan

2.1	Pre-Rework		Verific Mfq/Date	
	Incoming Inspection of Return SPS (651R390) WEC	ned Units	rii q/ bate	Insp/Date 1/4/91
	SPS (651R390) DPRO			,
	OSU (640R960) WEC	Zonati dan ji		1/4/91
68	OSU (640R960) DPRO			
2.2	Rework and Inspection The RN numbers listed below ar be configured to their latest	re for <u>reference</u> revision.	only - All asse	mblies should
	System Rework (536R500)	GG42D		
	SPS Chassis Rework (651R390) OLS-12, 14, 15 & 16 OLS-13	GG <u>36</u> D GG71D	2/6/9/	(23 th)
	SPS Mother Plate Assy Rework OLS-12, 14, 15 & 16 OLS-13	(651R342) GG35D GG70D	1/29/9/545	1/28/91
	SPS Matrix Plate Wiring Rewor (wiretabs 322R959 or 322R960) OLS-12 OLS-13 OLS-14 to 16	GG17D GG67D GG16D	1/28/9/	/24/21
	9C board assy rework (775R076	or 775R077) GG10D, GG15D & GG69D GG11D, GG15D &	_	<u> 3/15/9/</u>
	SDF-5 board assy rework (775Re OLS-12 OLS-13 OLS-14 to 16	078 & 775R079) GG08D, GG14D & GG68D GG09D, GG14D &		2/15/9/

. — — — — — — — — — — — — — — — — — — —		Pil .		
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<u>Verification</u> Insp/Date OSU chassis rework (640R960) S/N 5007, 5009-5011 GG33D S/N 5008 GG67D OSU Top Cover Rework (644R046) S/N 5007, 5009-5011 GG32D S/N 5008 GG65D OSU Mother Plate Rework (522R783) S/N 5007, 5009-5011 GG34D S/N 5008 GG66D OSU Matrix Plate Wiring Rework (wiretab 322R958) S/N 5007-5011 GG22D OSU-I board assy rework (775R080) S/N 5007, 5009-5011 GG12D, GG15D & GG23D GG54A, GG55A & GG56A S/N 5008 GG63D OSU-2 board assy rework (775R081) S/N 5007, 5009-5011 GG13D, GG15D & GG24D S/N 5008 GG64D

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			2	

## 2.3 Assembly Level Retest

2.3.1	Prime Side 9C Retest (775R076	<u>/775R077)</u> SN	5014
		Date	Verification
Rework Co	omplete - No open items on	1/14/91	(0.3)
Room Temp	perature Retest per paragraph 3 of T814A76	1/21/91	
Pre Coat	Data Review	47 7	16/2/21
WEC Inspe	ection - OK to Coat	1/26/41	
DPRO Insp	pection - OK to Coat	1/29/91	31
Conformal	Coat	2-2-91	
Eight Non	-powered Temperature Cycles	2-3-91	
Hi/Low Te 4.7	emperature Test per paragraph of T814A76	2-4-91	
Data Revi	ew Complete	2/5/91	<u> </u>
WEC Inspe	ction - Assembly Complete	2/7/91	
DPRO Insp	ection - Assembly Complete		2491
* OPEN	ITEM on CH FOR COATing &	ex 2/2/4, OK2/1/4,	1 ,

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	2.3.2 Redundant Side 9C Retest (775R076	/775R077) SN	5013
	Rework Complete - No open items on ICT	1-14-91	S& IS
	Room Temperature Retest per paragraph 4.3 of T814A76	1-21-91	111
	Pre Coat Data Review		1/22/91
	WEC Inspection - OK to Coat		1/26/41
	DPRO Inspection - OK to Coat		1/28/91
	Conformal Coat	- OD-	2/2/91
	Eight Non-powered Temperature Cycles	• (1)	2/3/91
	Hi/Low Temperature Test per paragraph 4.7 of T814A76	(3.86)	02-05-1991
	Data Review Complete		2/6/91
	WEC Inspection - Assembly Complete		2/7/91
	DPRO Inspection - Assembly Complete		2/8/91
- ,	open iten on eft FOR COATing JEK 7/2/21 OK	< 3/7/91 i	

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#

2.3.3 Prime Side SDF-5 Retest (77	5R078/775R079) SN	5012
The state of	<u>Date</u>	Verification
Rework Complete - No open items on ICT	1/24/91	
Room Temperature Retest per paragraph 4.3 of T814A78	1/31/91	
Pre Coat Data Review	1/31/91	Me JAMES
WEC Inspection - OK to Coat	2/4/91	(6,0)
DPRO Inspection - OK to Coat	2/4/91	1.4
Conformal Coat	2/4/91	
Eight Non-powered Temperature Cycles	2/9/91	
Hi/Low Temperature Test per paragraph 4.7 of T814A78	0/10/91	
Data Review Complete	2/11/81	
WEC Inspection - Assembly Complete	2-11-91	(ITME)
DPRO Inspection - Assembly Complete	2/11/91	

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2.3.4 Redundant Side SDF-5 Retest (	775R078/775R079) SN 50/3
Rework Complete - No open items on ICT	1/26/91
Room Temperature Retest per paragraph 4.3 of T814A78	1/31/91
Pre Coat Data Review	1/31/91 MEJAMES
WEC Inspection - OK to Coat	2/4/91
DPRO Inspection - OK to Coat	2/4/91
Conformal Coat	2/8/9/
Eight Non-powered Temperature Cycles	2/9/9/
Hi/Low Temperature Test per paragraph 4.7 of T814A78	3/10/91
Data Review Complete	2/11/91
WEC Inspection - Assembly Complete	2-11-91
DPRO Inspection - Assembly Complete	Mufal

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2.3.5 <u>OSU-1 Retest (775R080)</u>	SN 5007	
	Date	<u>Verification</u>
Rework Complete - No open items on ICT	2/2/91	-
Room Temperature Retest per paragraph 4.3 of T814A80	2/3/91	
Pre Coat Data Review	2/4/41	<u>(i)</u>
WEC Inspection - OK to Coat	2/4/91	
DPRO Inspection - OK to Coat	2/6/91	
Conformal Coat	2/6/91	(53.13)
Eight Non-powered Temperature Cycles	2-11-91	(三日)
Hi/Low Temperature Test per paragraph 4.7 of T814A80	2-11-91	
Data Review Complete	2/13/91	(33)
WEC Inspection - Assembly Complete	2/13/91	(38)
DPRO Inspection - Assembly Complete	2/13/91	

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2.3.6	OSU-2 Retest (775R081)	sn <u>5007</u>	_1
*		<u>Date</u>	Verification
Rework C	omplete - No open items on T	1/18/91	Sais
Room Tem	perature Retest per paragraph 3 of T814A81	1/29/91	
Pre Coat	Data Review	<u> </u>	1/29/91
WEC Inspe	ection - OK to Coat	11 over 181 Sept.	1/3//9/
DPRO Inst	pection - OK to Coat	as a memory of poor of	1/31/91
Conformal	Coat	1/31/91	Se is
Eight Nor	n-powered Temperature Cycles	02-06-1991	
Hi/Low Te 4.7	emperature Test per paragraph of T814A81	2/7/91	0489
Data Revi	ew Complete		2/7/91
WEC Inspe	ction - Assembly Complete	2-7-91	
DPRO Insp	ection - Assembly Complete		2/8/91

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2.3.7 OSU Assembly Retest (640R960)	SN	5007
	Date	Verification
**Rework Complete - No open items on ICT	2/15/91	(270)
Room Temperature Retest per paragraph 4.1 and 4.2 of T814A56	02-15-1991	<u>***</u>
Hi/Low Temperature Test per paragraph 4.7 of T814A56	2/16/91	
Data Review Complete	2/18/91	Aw
WEC Inspection - Assembly Complete	2-19-91	
DPRO Inspection - Assembly Complete & open iten # 80 2/15/918EK  No Empael on test.	2/19/91	
No Empael on test.		

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## 3.0 Subsystem Level Retest Procedure Date <u>Verification</u> 3.1 Ambient Subsystem Verification Rework Complete - No unexplained Open Items on ICT Checkpoint A of PD 045 (attach copy) Run the following Test Files (Room Temperature) NEWONI'S. ST OKTESTN.ST 5XZCX/SS.ST. - RDSTSTSS.ST NEWON235.ST QKTESTR.ST 5 X2 0 X 2 S S S / - ROSTSTSS-ST 5X18X1SS.ST 5X18X2SS.ST 5X18X3SS.ST 5X18X4SS.ST 3.1.1 Ambient Encrypter Verification Note: To allow for scheduling and security constraints the ambient encrypter verification may be performed out of sequence, however, paragraph 3.1:1 must be complete prior to starting paragraph 3.3.9, Thermal Cycle #8. Install the KG-46 data encrypter and KG-28 decrypter. Check out the KG-28 set-up using the ST-19 verification procedure Run the following test files: **NEWONISS.ST**

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5x20x3 55.57 ROSISI SI

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	OLS PROGRAM DIRECTIVE	12
-CONTINUATION SHEET-	DIRECTIVE NO. 045 DATE 12/12/88	

	(Before Subsystem Test, in Block 5 Clean Room, per Pal Thermal Chamber	T927000)		
Α.	Area is clean and contains no miscellaneous parts or extraneous hardware.		<u>RESP</u> QE	
в.	The anti-static mat and wrist straps are in place and ready for use.		ЭĐ	
C.	Unit and System cable connector pins are checked and none are bent or pushed in. (NOTE: This can		INSP	
	be verified at presystem unit inspection and the buffer connectors are inspected and installed at this point per PDO34)			
D.	Verify correct color code on buffer connectors to certify inspected, tested and approved connectors per PD 034 (Appendix B). SPS, SPW, PSW, OSW, + CARCES	<b>1</b>	INSP	
E.	Test equipment configuration checked per Program Directive #022 less paragraphs IID and III		QE	PQL
F.	Perform Test Equipment Operational Check per PD022 paragraphs IID and IIE.	VOER	TD	PQL
G.	Review open ICT items on the SPS, SPU, PSU and OSU, and evaluate closure prior to moving to subsystem		QE	
	WRING TEST. STONE,	MIL	TD MFG	ä
н.	TCP, SPS, SPU, PSU and OSU are connected to. system cable connectors and each is grounded to the ground bus per 9RD7845.	15EV	TD	POL
I.	Item annotated on ICT that units are ready for subsystem test.	1	QE	
j v	TE 1: NHEN APPLICABLE, RECORD ACTIONS PERFORMED IN SYSTEM OG BOOK.  NOTE 2: "CAUTION" BUMP, SHOCK, ABRUPT MOVEMENT OF TAPE RECORDERS	<u></u>		

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			43
	NEWON2SS.ST	4/25/91	
	5XZOXYSS.ST ROSTST.ST	1/25/91	10/4°
	NEWOND1.ST	4/25/91	Ala)
	5 KZCX 355.57 RDSTSTSS.ST	4/25/91	Cara Ala
	NEWOND2.ST	4/25/91	
	SX ZOXYSS.ST ROSTST.ST	4/25/91	
3.2	SPS and OSU Vibration		en E
	Note: To allow for scheduling, OSU and SPS vibration may occur before ambient tests per paragraph 3.1. Vibration must be completed prior to starting paragraph 3.3. Later vibrations due to rework or RN incorporation shall be recorded on the units ICT.		Maria Cara Cara Cara Cara Cara Cara Cara
	Pre-Vib Data Review	4/26/91	JEH
2	WEC Inspection - OK to Vibrate	31/26/91	Ser Ser
	DPRO Inspection - OK to Vibrate	4/26/91	- BET
	Vibrate SPS - x-axis, random only 5D3 acceptance level per PQL737, nonpo	POL 4/26/9/ owered	[F]
	Vibrate OSU - x-axis, random only 5D3 acceptance level per PQL737, nonpo	1 POL 4 26/91	
	WEC Inspection - Post Vib	4/26/91	
	DPRO Inspection - Post Vib	4/2:13	DPRO WAVE PER R.BA
3.3	Thermal Test	HIREE	3114
	Checkpoint B of PD-045 (attach copy)	4/26/51	.8
	Install Thermocouples (PQL operation)	4/27/91	PQL

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06512 4/25/91 31 3600

## OLS PROGRAM DIRECTIVE

-CONTINUATION SHEET-

DIRECTIVE NO. 045

DATE \_\_12/12/88

# (Before Sub-system Thermal Test, in PQL, per T927000)

A. PQL thermal chamber is clean and area contains no miscellaneous parts or extraneous hardware.

RESP OE

B. Assembly and cable connectors (buffered and unbuffered) were checked before cabling up, and no damage was found. (Note: Where buffer connectors are in-place, check the viewable side; do not remove from assembly.) VERIFY COLOR CODE ON BUFFER CONNECTORS PER PD 034, APPENDIX B.

INSP

C. SPS, SPU, PSU and OSU tranported per 9RA4220. Dent-

MFG

 Assemblies SPS, SPU and TCP are connected to system cable connectors and each is grounded to the ground bus per 9RD7845.

DESK-JEH 1

E. Test equipment is in calibration as required in Program Directive #022 less paragraphs IID and IIE.

(3)

QE

F. Perform Test Equipment operational check per PD\_022 paragraphs IID and IIE.

TRH

TD

NOTE 1:

WHEN APPLICABLE,
RECORD ACTIONS
PERFORMED IN SYSTEM
LOG BOOK.

NOTE 2:

"CAUTION"

BUMP, SHOCK,

ABRUPT MOVEMENT

OF TAPE RECORDERS

CAN CAUSE SEVERE

DAMAGE.

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# 3.3.1 Ambient Verification

Run the following Test Files (Room Temperature):

	Remo Date	Verification Date A
NEWON1.ST	5/2 4/26/91	\$ \$17. SH 47.
QKTESTN.ST	5/6 4/26/91	18 17 SES
ラメス@X/SS.57-ROSTSTSS.ST	1/2/2 4/26/91	200 5/14 201 4har
NEWON2.ST	5/2 4/26/91	A(A) 809 4 Nun
QKTESTR.ST	A2 1/26/91	AIA) 30 5/1/
5x20x255.57 RDSTSTSS.ST	J2 426/9	2869) 95/7M AIA
5X18X1SS.ST	5/2/91	90,5171
5X18X2SS.ST	5/2/91	20,5171
5X18X3SS.ST	5/3/91	207 SITH.
5X18X4SS.ST	5/3/91	90,500

			1.5	
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# 3.3.2 Thermal Cycle #1

	3.3.2.1 Hot Temperature	<u>Date</u>	Verification Pata Rev.
	Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have		~
	power on.	5/4/91	- FM
	Run 5x18x1SS.ST NEWONI.ST	5/4/9	RAN 126-5
	SXZUX 155.5T ROSSISISS SI	5/4/91	RMC 200 57MI
	Enter CON 0 42	5/4/91	497
	Enter OLS OFF	5/4/91	AJK.
	3.3.2.2 Cold Temperature	<u>Date</u>	<u>Verification</u>
	Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.	5/NG1	Rub?
•	Run 5x18x1SS.ST NEWON 2.ST STXZ Y Z 55.5 T RDSSTSTSS.ST	5/4/91	12C-5 900
	Enter CON 0 42	5/4/91	
	Enter OLS OFF	5/4/91	P. S.

WPF EP.lah	PAGE	FSCM NO	DOCUMENT NUMBER	REV
	16	97942	BVS-2600	C

# 3.3.3 Thermal Cycle #2

	3.3.3.1 Ho	Temperature	<u>Date</u>	Verification Coto
	or the SPS	per air in the vicinit to stabilize at +50°C 2 hours. During	у	73 57
	this time to power on.	the OLS shall not have	5/4/91	2085 AIA
	Run	5x18x2SS.ST NEWON! ST	5/5/91	904 517.
5KZ 0	X / 55.5 T		<u>5/5/91</u>	909 5/71
		Enter CON 0 42	<u>.5/5/91</u>	
		Enter OLS OFF	<u> 5/5/91</u>	1301
	3.3.3.2 Col	d Temperature	<u>Date</u>	<u>Verification</u>
	+0°/-4° for	er air in the vicinity to stabilize at -10°C 2 hours. During		Ε.
	this time to power on.	he OLS shall not have	5/5/91	(1301) (Ala)
5	Run	5x18x2SS.ST NEW ON 2. ST	5/5/91	(314g) 90, 517,
5×20	x 2 55.57	RDSSTSTSS.ST	5/5/91	(316) 909 5/7/
		Enter CON 0 42	5/5/91	AND
		Enter OLS OFF	5/5/91	(316)

WPF EP.lah	PAGE	FSCM NO	DOCUMENT NUMBER	- REV
	17	97942	BVS-2600	С

### 3.3.4 Thermal Cycle #3

Run

5 XZOX155.57

3.3.4.1 Hot Temperature

Date

Verification Pata Reve

Allow chamber air in the vicinity of the SPS to stabilize at  $+50^{\circ}$ C  $+4^{\circ}/-0^{\circ}$  for 2 hours. During this time the OLS shall not have power on.

5x18x3SS.ST

NEWOWI. ST ROSSTSTSS.ST

Enter CON 0 42

Enter OLS OFF

5/5/9/ 5/5/9/ 5/5/9/

199 50A1

(編集)

3.3.4.2 Cold Temperature

Date

<u>Verification</u>

Allow chamber air in the vicinity of the SPS to stabilize at  $-10^{\circ}\text{C}$   $+0^{\circ}/-4^{\circ}$  for 2 hours. During this time the OLS shall not have power on.

Run 5x18x3SS.ST NEWON2.ST

5x70x2=5.57 RDSSTSTSS.ST

Enter CON 0 42

Enter OLS OFF

5/5/9/ 5/5/9/ 5/5/9/ 5/5/9/

904 517191

(1) 300 5171

Eeg .

## 3.3.5 <u>Thermal Cycle #4</u>

	3.3.5.1 Hot	Temperature	<u>Date</u>	Verification Data A
	of the SPS +4°/-0° for	per air in the vicinit to stabilize at +50°C 2 hours. During the OLS shall not have		<u>11301</u> )
	Run	5x18x4SS.ST Newoni.ST	5/6/91	99 5/2
5 X Z	0×155.57		.5/4/91	209 5/7
		Enter CON 0 42	5/4/41	1301)
		Enter OLS OFF	-5/6/91	
	3.3.5.2 Col	d Temperature	<u>Date</u>	<u>Verification</u>
	+0°/-4° for	er air in the vicinity to stabilize at -10°C 2 hours. During he OLS shall not have		
	power on.		5/6/91	
	Run	5x18x4SS.ST NEWON 2.ST	5/6/91	90 517.
5 XZ	0XZ55.57	RDSSTSTSS.ST	5/6/91	111 923 517K
		Enter CON 0 42	5/10/91	Ala
		Enter OLS OFF	5/6/01	

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#### 3.3.6 Thermal Cycle #5

3.3.6.1 Hot Te	emperature
Allow chamber of the SPS to	air in the vicinity stabilize at +50°C

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x1SS.ST NEW ON! ST SX20X/SS.ST RDSSTSTSS.ST

Enter CON 0 42

Enter OLS OFF

1.10

Date

5/7/91

5/7/91

3/7/9/

3.3.6.2 Cold Temperature

Allow chamber air in the vicinity of the SPS to stabilize at  $-10^{\circ}$ C  $+0^{\circ}/-4^{\circ}$  for 2 hours. During this time the OLS shall not have power on.

Run 5x18x1SS.ST NEWON 2.ST 5x20x25557 RDSSTSTSS.ST

Enter CON 0 42

Enter OLS OFF

<u>Date</u>

<u>Verification</u>

5/7/9/ 5/8/9/ 5/8/9/ 5/8/9/ 2555 AIA 2009 5 1 1019 1

Verification Data Revi

30g 5MM

Seg 517191

120-5

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# 3.3.7 Thermal Cycle #6

3.3.7.1 Hot Temperature	<u>Date</u>	Verification Onto
Allow chamber air in the vicinit of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During	у	
this time the OLS shall not have power on.	5/8/91	
Run 5x18x2SS.ST VEWONI.ST	5/9/91	12 C:
5 x 20x1 \$5. 5 7 RUSSISISS.ST	-5/9/91	904 9/10/
Enter CON 0 42	.5/9/91	atan' ,
Enter OLS OFF	5/9/91	
	**	
3.3.7.2 Cold Temperature	<u>Date</u>	<u>Verification</u>
Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During	<del></del>	<u>Verification</u>
Allow chamber air in the vicinity of the SPS to stabilize at alogo	<del></del>	<u>Verification</u>
Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.  Run 5x18x2SS.ST	- 11 * 1 * 1 * 1 * 1 * 1 * 1 * 1 * 1 * 1	93. 5/A
Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.	5/9/91	
Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.  Run 5x18x2SS.ST  VEWONZ.ST	<u>5/9/91</u> <u>5/9/91</u>	93. 5/A

	R REV	DOCUMENT NUMBER BVS-2600		PAGE 21	WPF EP.lah
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#### 3.3.8 Thermal Cycle #7

3.3.8.1 Hot Temperature

Date

Verification Date Re-

Allow chamber air in the vicinity of the SPS to stabilize at  $+50^{\circ}$ C  $+4^{\circ}/-0^{\circ}$  for 2 hours. During this time the OLS shall not have power on.

5/9/91

1265

Run 5x18x3SS.ST NEWON (ST 5 X Z O X / SS.S ) ROSSTSTSS-ST

5/9/91

903 5/67

Enter CON 0 42

Enter OLS OFF

V7/9/

<u>Date</u>

<u>Verification</u>

3.3.8.2 Cold Temperature

Allow chamber air in the vicinity of the SPS to stabilize at  $-10^{\circ}$ C  $+0^{\circ}/-4^{\circ}$  for 2 hours. During this time the OLS shall not have power on.

5/10/91

316 907 5/1019

120-5

Run 5x18x3SS.ST

5 X 2 UX 2 SS. ST ROSSTSTSS. ST

5/10/9

Enter CON 0 42

5/10/91

Enter OLS OFF

Spolar

#### 3.3.9 Thermal Cycle #8

### 3.3.9.1 Hot Temperature

Allow chamber air in the vicinity of the SPS to stabilize at +50°C

+4°/-0° for 2 hours. During Pata Re this time the OLS shall not have power on. 5/10/91 Run the following dual prime Test Files: /\_\_ NEWOND1.ST 5/10/91 Jeg 5/13 5x3x1SS.ST 5/10/91 907 919 5x3x5SS.ST 90, 5/11 5x5x1SS.ST 907 5/13 288E 5x6x1SS.ST 907 9/13 5x8x1SS.ST 909 5117. 5X9X1SS.ST 120-8 90 511. 5X10X1SS.ST 95 5/13, 5X11X1SS.ST Seg. 5/17/1 5X12X1SS.ST Seg SIMM. - 5X13X1SS.ST 959 SIT/7 5x13x3SS.ST 903 5/13/71 5X14X1SS.ST 903 5/17. 5X16X1SS.ST 95 5/3/9 5X17X1SS.ST 989, 54/1491

12

903 5/13/91

20, 9/19/

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5X19X3SS.ST

✓ 5X2X1SS.ST

- 5X2X2SS.ST

Run the following dual prime Test Files: - NEWOND2.ST 83 5ln19 5x3x2SS.ST 5x3x6SS.ST 905 513191 5x5x2SS.ST 909, 5/DA1 5x6x2SS.ST 803 5/13/91 CB 5/13/91 5X9X2SS.ST 949 5/3/71 803 5/13/91 5X11X2SS.ST 969 5/17/71 5X12X2SS.ST SA FIBRI 5x13x2SS.ST 90, 5113171 5x13x4SS.ST 90% SINTI → 5X14X2SS.ST 5X16X2SS.ST 5X17X2SS.ST OLS 5X19X4SS.ST

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Install the KG-46 data encrypter and KG-28 decrypter. Check out Pata Kevi the KG-28 set-up using the ST-19 SEP verification procedure. 4/13/91 Run the following test files: NEWOND1.ST 5/13/90 5 x 20x 3 55.37 RDSSTSTSSE 5/13/91 **NEWOND2.ST** 5/13/91 RDSSTSTSSE.ST 5/13/91 929 5/11/9, 5 X20X455.57 Enter CON 0 42 Enter OLS OFF Remove the KG-46 encrypter and K6-28 decrypter. 3.3.9.2 Cold Temperature Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on. 5/11/91 Run the following dual prime Test Files: NEWOND1.ST 5/11/91 9eg 5/13/ 5x3x1SS.ST 5/11/91 5x3x5SS.ST 5/11/91 5/13/9 5x5x1SS.ST 5/11/91 903 9/13/7 5x6x1SS.ST 5/11/91 903 5/13/7/ 5x8x1SS.ST 5/11/91 99 9/13/91 5X9X1SS.ST 5/11/91 90 91BAI 5X10X1SS.ST ✓ 5X11X1SS.ST 99 813171

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			(20)	VATA /LEVIL
	SX12X1SS.ST	5/11/91	(110)	90 910191
	✓ 5x13x1SS.ST	5/11/91	Ale	90% SIAR
		5/11/91	(3165) (A)(A)	904 9/13191
	√5X14X1SS.ST	5/11/91	(31 <b>G</b> )	803 5/13/21
	✓ 5X16X1SS.ST	5/11/91	(AIA)	905 5113191
	✓ 5X17X1SS.ST	5/11/91	110	Jos Flister
	<del>-5X19X355.51-</del>	NA FOR OL.	5#12	
-	5X2X1SS.ST	5/11/91	AIA 12	c-6
	€X2X2SS.ST	5/11/91	499	नुन्द डीगगा
Run the	following dual red	undant Test Files:		
	NEWOND2.ST	5/11/91	2893 AIA	967 5/17/91
	√ 5x3x2SS.ST	5/11/91	AIA	SA SIBAL
	5x3x6SS.ST	5/11/91	38634	SA 5171
		5/11/91	4/4	98 5/0R1
	✓ 5x6x2SS.ST	5/4/91	Z 3 12	2007 EH5171
	5x8x2SS.ST	5/11/91	AAA	963 510171
	5X9X2SS.ST	5/11/91	(ALC	903 FIR191
	5X10X2SS.ST	5/11/91	1414	99 519171
	5X11X2SS.ST	5/11/91	286	ורוביאל בסב
	✓ 5X12X2SS.ST	Slulgi	- (AIA)	20 5113191
		<u>slulgi</u>	296	gos sinai
	✓5x13x4SS.ST	5/11/91	(E)	907 510 hi
	∠ 5X14X2SS.ST	5/11/91	SHE)	27 510R1
	≥5X16X2SS.ST	5/11/ai	(a) Ta	207 5/13/71
-9	5X17X2SS.ST	5/11/91	(File)	Jez 5/11.11
		= ( -(1)		

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5X16X2ss.sT	
5X17X2S9.ST	

ser primions

SYLOYAGG ST

NA FOR OLS #12

Install the KG-46 data encrypter and KG-28 decrypter. Check out the KG-28 set-up using the ST-19 verification procedure.

5/13/91

Run the following test files:

NEWOND1.ST

5/13/91

(110) Jos 51.4191

5X20X3SS. STROSSTSTSSE.ST

5/13/91

(314) 903 5/Hh

NEWOND2.ST

5/13/91

(B) (3) 5/14/1

5 XZ OXY 5 5. ST RDSSTSTSSE.ST

5/13/11

2000

Enter CON 0 42

5/13/11

TOOP

Turn OLS and TCP OFF, bring chamber from cold to  $+50 \pm 10^{\circ}$  C and allow to soak for 2 hours minimum to prevent moisture from condensing on units

5/17/71

1720

Return chamber to room temperature and remove thermocouples.

5/17/91

DEP

Subsystem Data Review

5/17/91

sup

Subsystem Test Complete

5/17/91

DEP

Final WEC Inspection

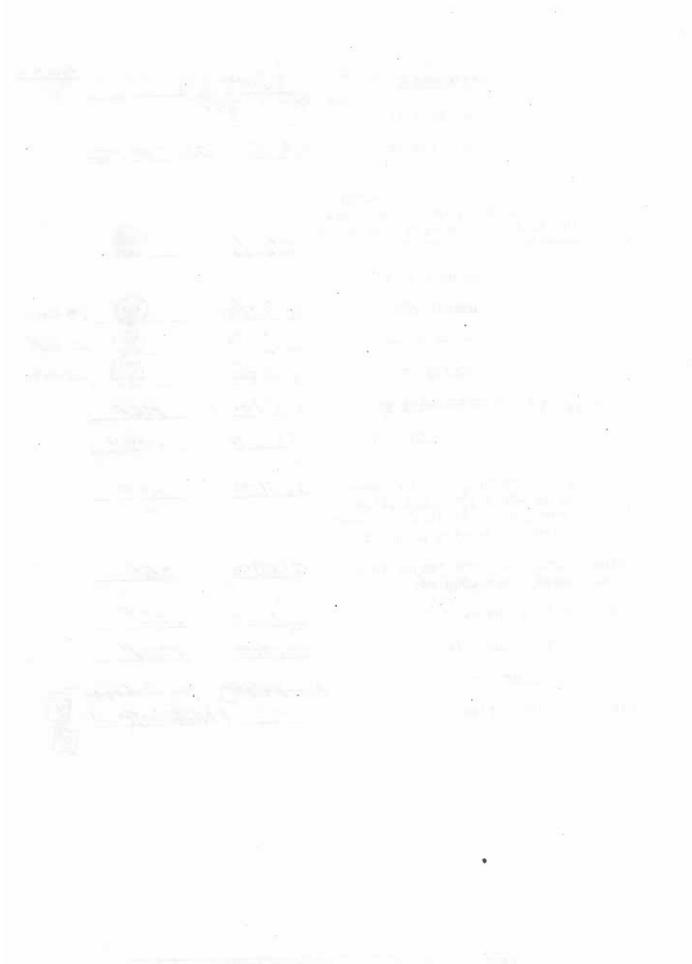
3/11/91

Supper 5

Final DPRO Inspection

- TO INSPECTION.

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			99	



BVS	2693	
DATE	20 March 1992	
ORIGINATOR _	Halisti J. Scilipoti	
	∪ J. Scfilipoti	
	REV -	

F12

OLS #12 BEARING RETROFIT

ACCEPTANCE TEST REPORT
VOLUME III OF III
ALIGNMENT AND SYNCHRONIZATION CURVES

(CDRL 006A1)

Contract F04701-90-C-0028

Prepared For

UNITED STATES AIR FORCE Headquarters, Space Division Los Angeles, California

Prepared By

WESTINGHOUSE ELECTRIC CORPORATION
Defense and Electronics Center
Baltimore, Maryland

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### **ABBREVIATIONS**

AS	Along Scan (Synchronization)			
AT	Along Track (Alignment)			
SD	Surface Distance			
SDF	Stored Data Fine			
SDS	Stored Data Smooth			
RTD F	Real Time Data - Fine			
RTD S	Real Time Data - Smoothed			
Н	HRD Channel			
Т	T(Thermal) Channel			

PMT Channel

#### 1. REFPLN PLOTS

These are the computer-generated least-squares fits to OLS #12 HRD and T Channel Stored Data Fine (SDF) Alignment and Synchronization data taken from the final +5°C SSS/ -8°C Ml Thermal Vacuum run (Orbit Nominal).

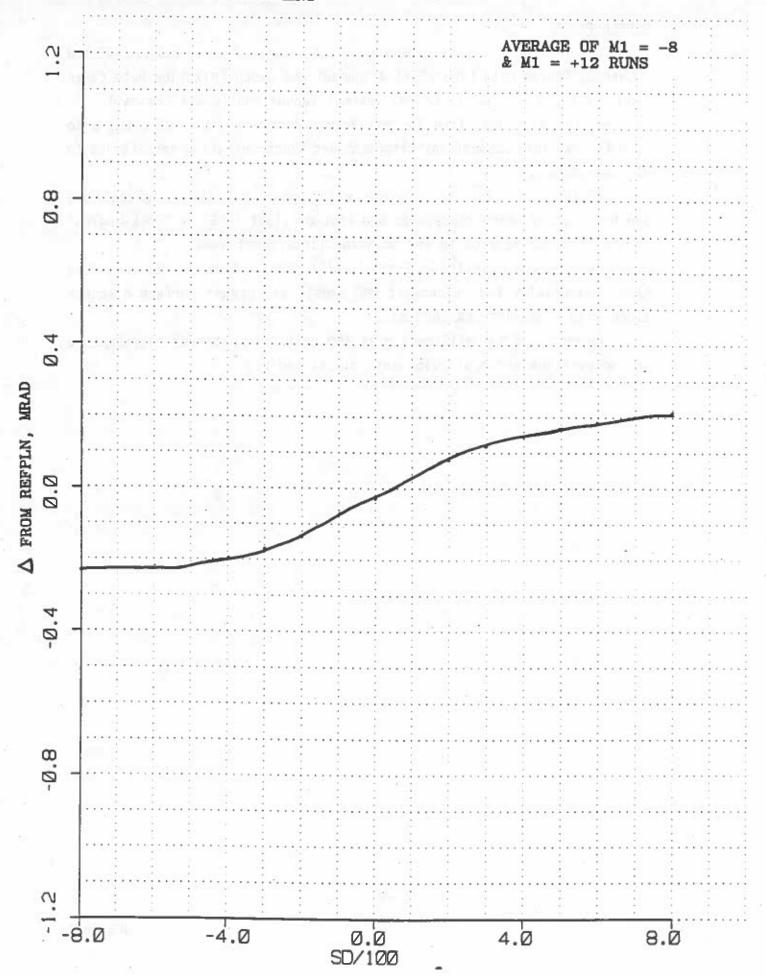
For OLS #12, data from Thermal Vacuum Runs with M1 =  $\pm$ 12°C was also used to take into account any Alignment and Synchronization sensitivity to M1 temperature.

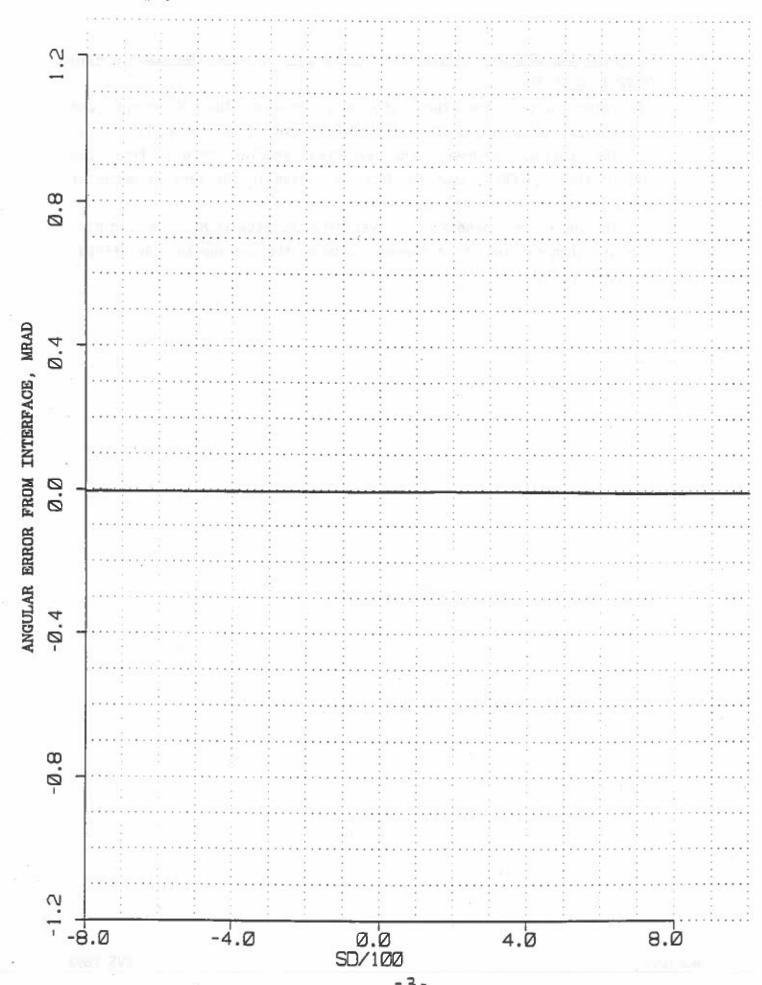
REFPLN is a computer program which generates the Alignment and Synchronization which represents the line-of-sight (LOS) or "look-angles" of the SSS with respect to the mounting (Interface) axes.

The curves are plotted as Error in milliradians from the OLS Interface Axes (essentially the spacecraft PMP axes), vs. ground surface distance along scan from subtrack (Nadir).

(An error of 0.1 milliradian at 450 naut. mi. altitude represents a ground position error of .045 naut. mi. at nadir.)





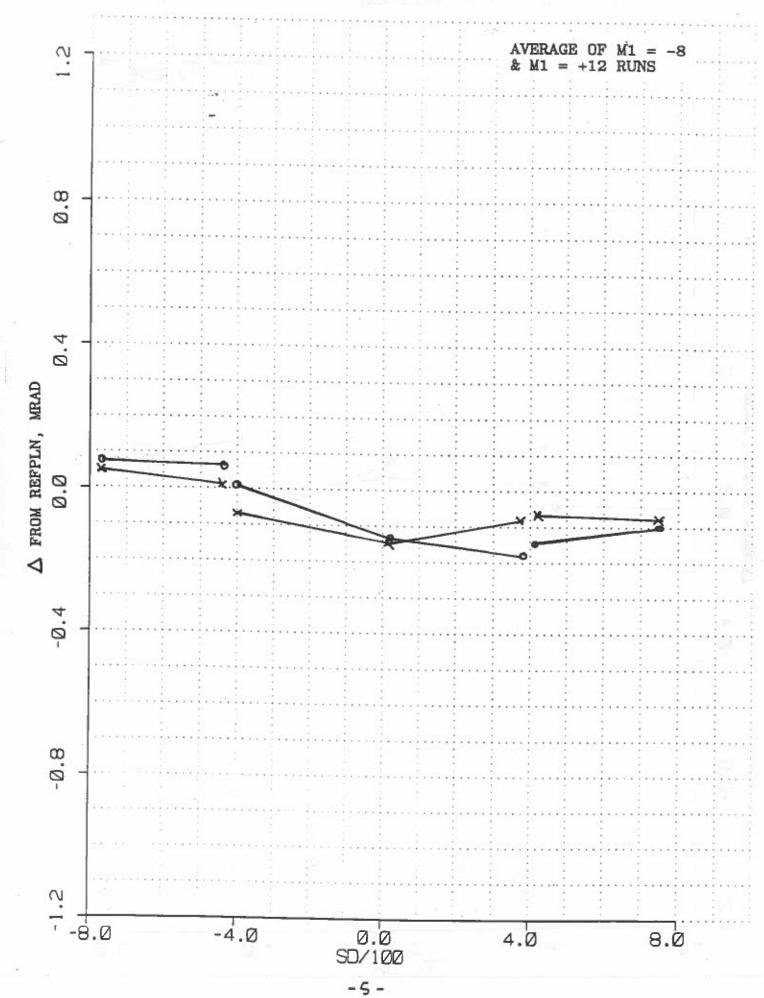


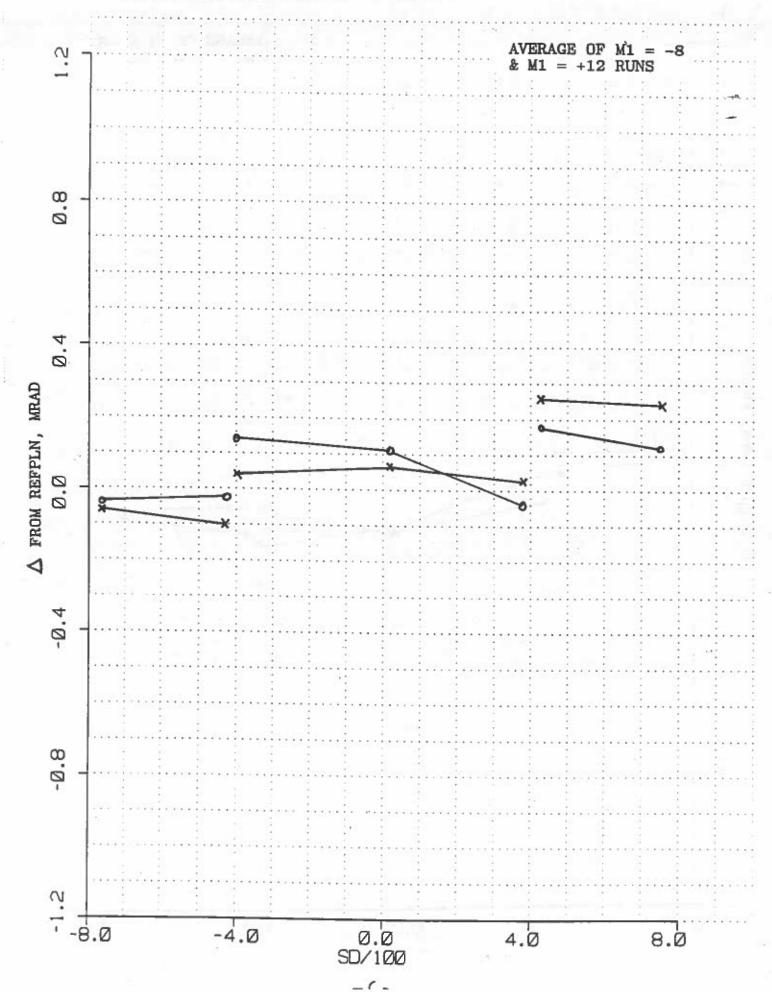
# 2. ALIGNMENT AND SYNCHRONIZATION FOR ALL MODES AT +5. SSS TEMP PLOTTED WITH RESPECT TO REFPLN

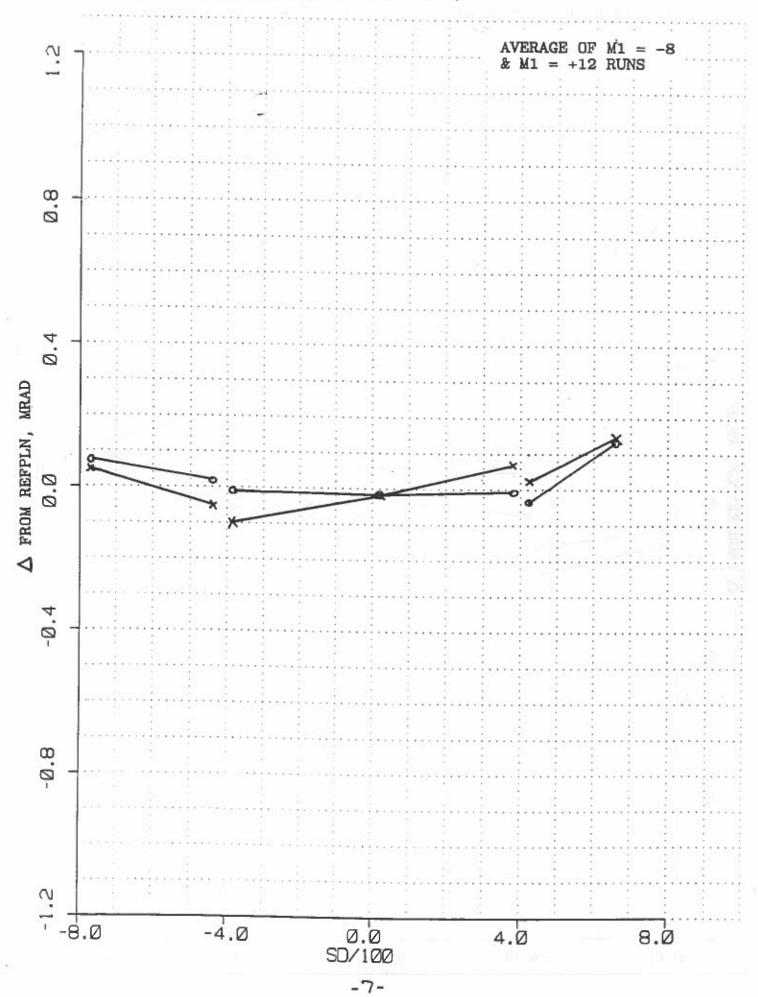
These curves are the difference between the Alignment and Synchronization curves at  $SSS = +5 \cdot C$  (Orbit Nominal) and the REFPLN Plots.

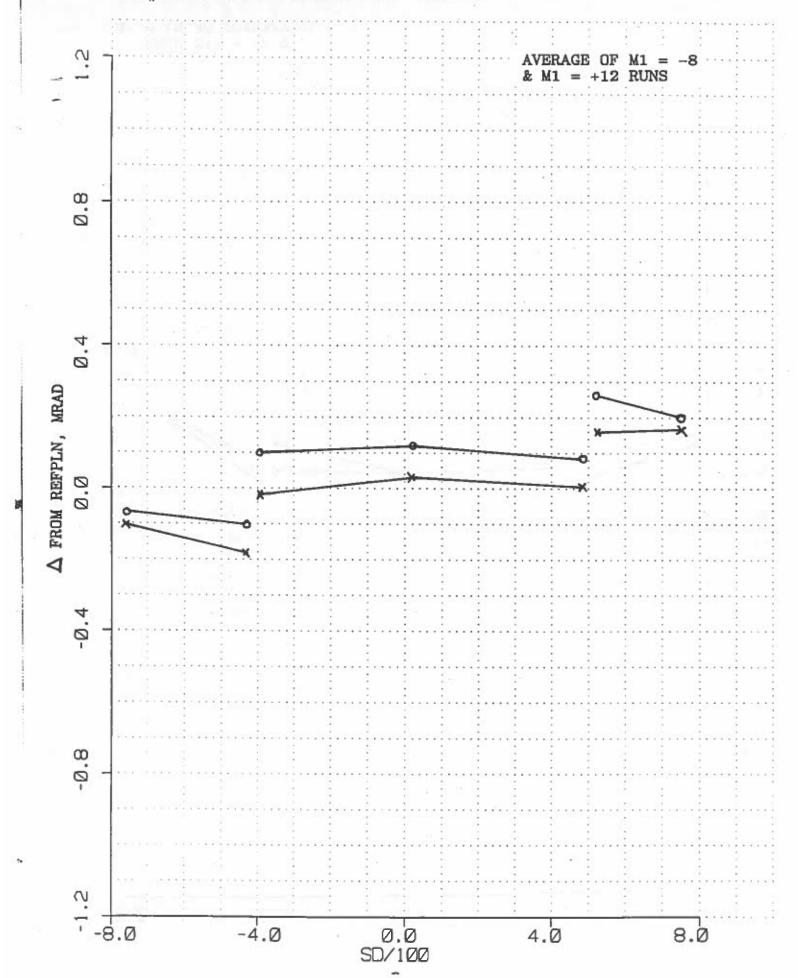
The curves represent the expected angular errors from the line-of-sight (REFPLN) axes for OLS data taken in the various modes of operation at orbit nominal conditions.

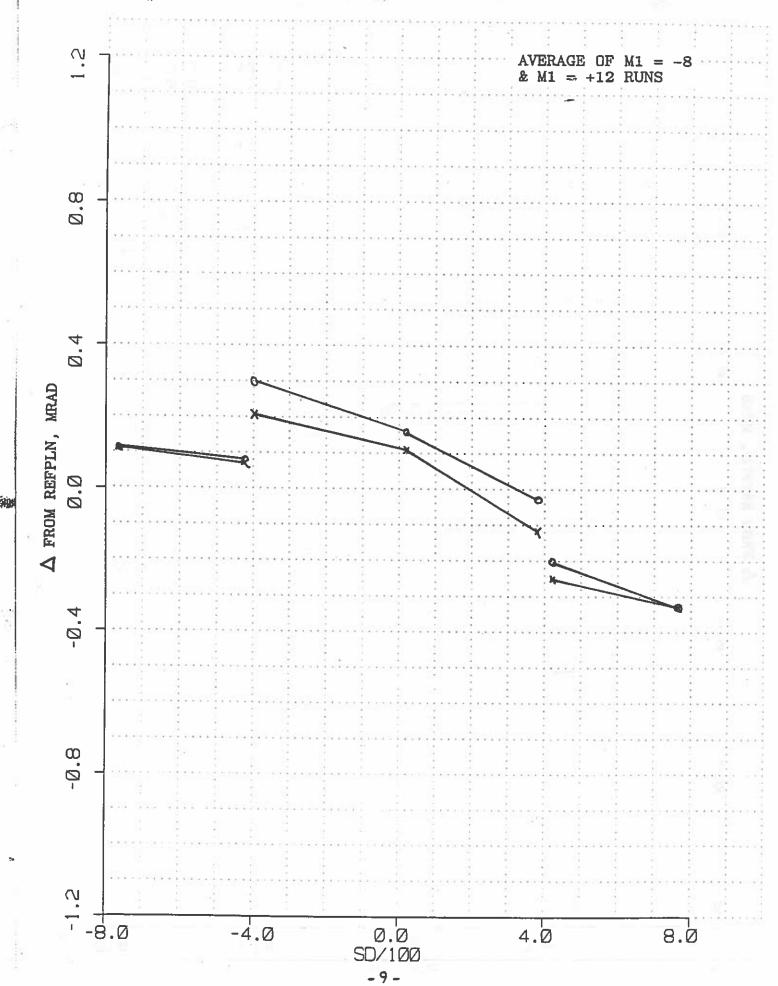
The curves were generated by averaging the data at  $M1 = -8^{\circ}$  and  $M1 = +12^{\circ}$  and then finding the difference between the average and the REFPLN.

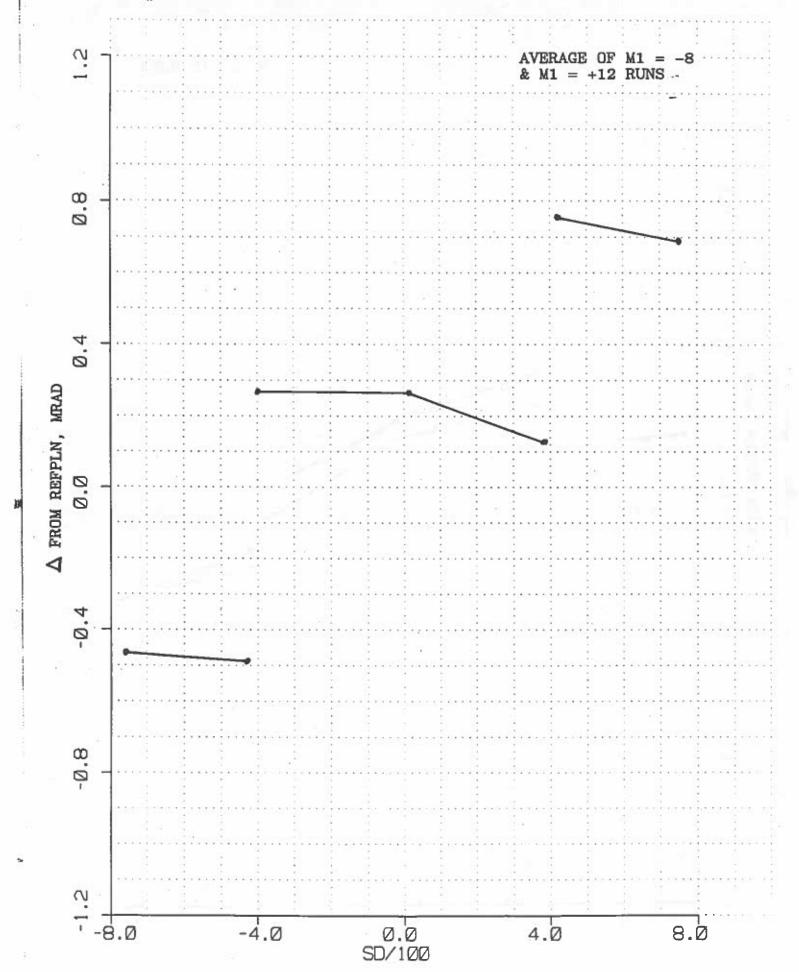


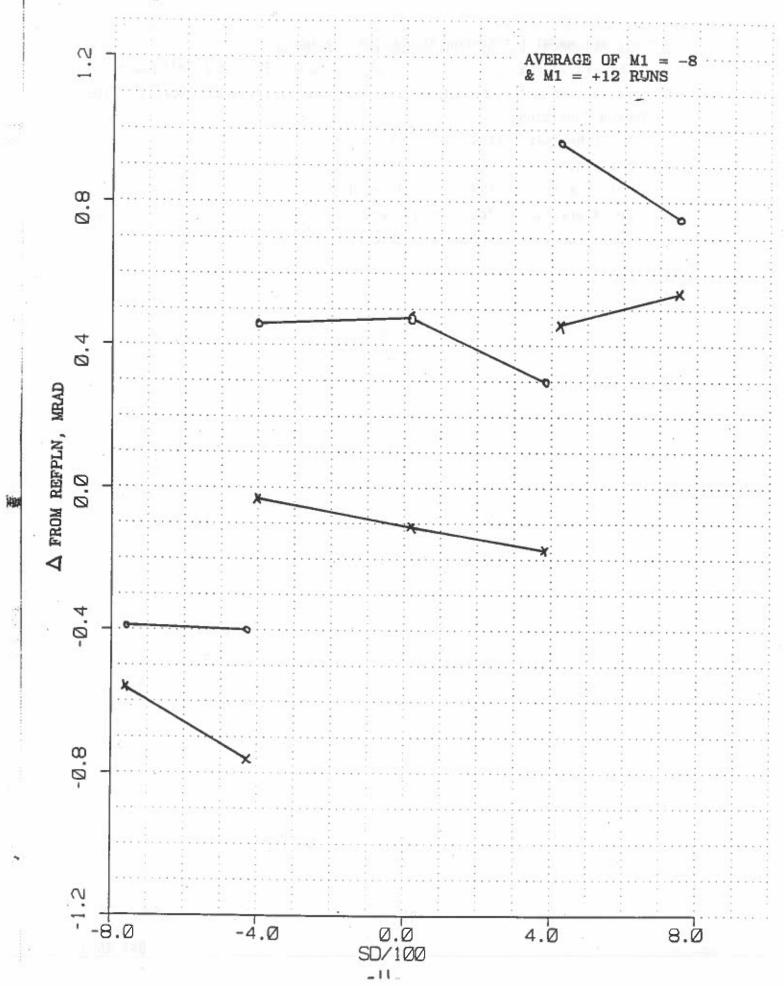








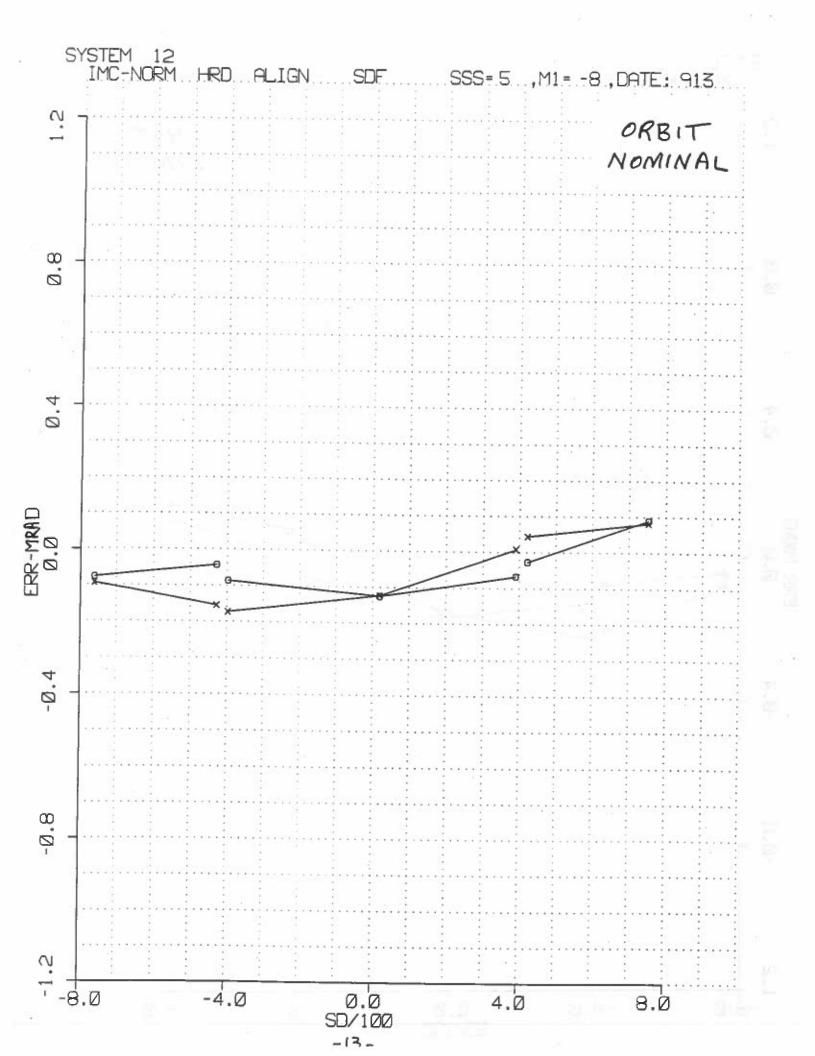


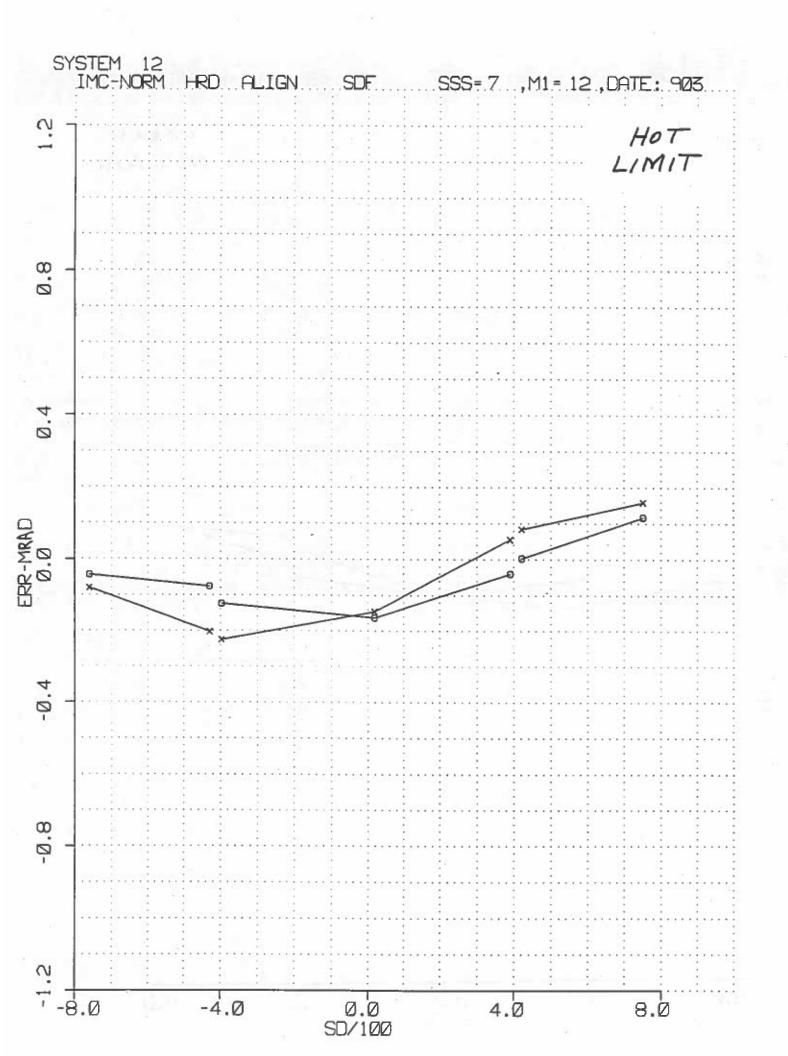


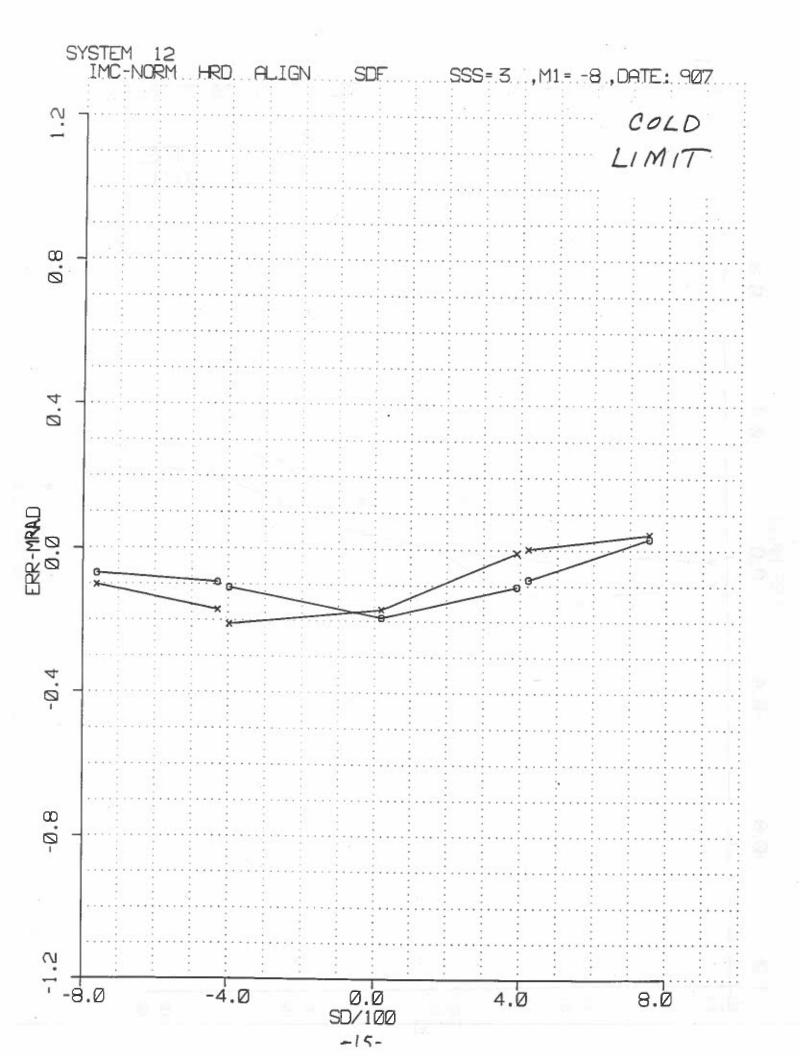
### 3. OLS #12 ALIGNMENT & SYNCHRONIZATION FOR ALL MODES

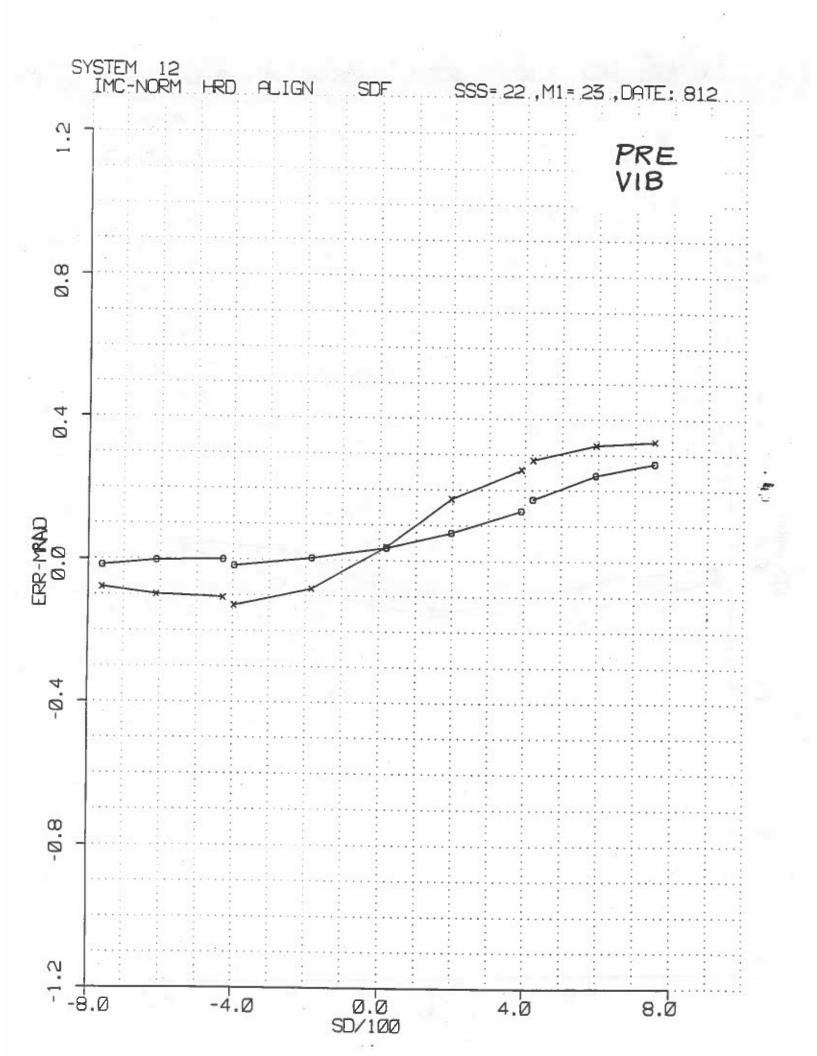
The following graphs are the measured OLS #12 Alignment & Synchronization with respect to the mounting (Interface) axes, for the following conditions.

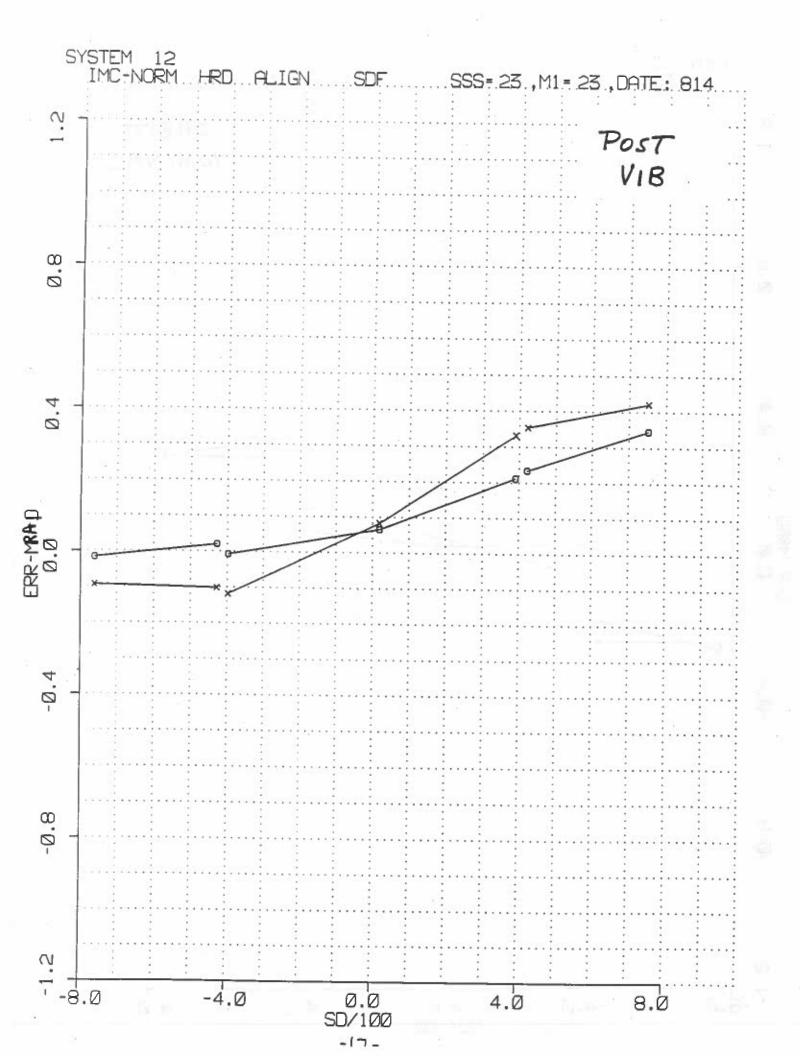
Orbit Nominal (SSS = +5°, M1 = -8°)
Hot Limits (SSS = +7°, M1 = +12°)
Cold Limits (SSS = +3°, M1 = -8°)
Pre Vibration (Acceptance Level)
Post Vibration (Acceptance Level)

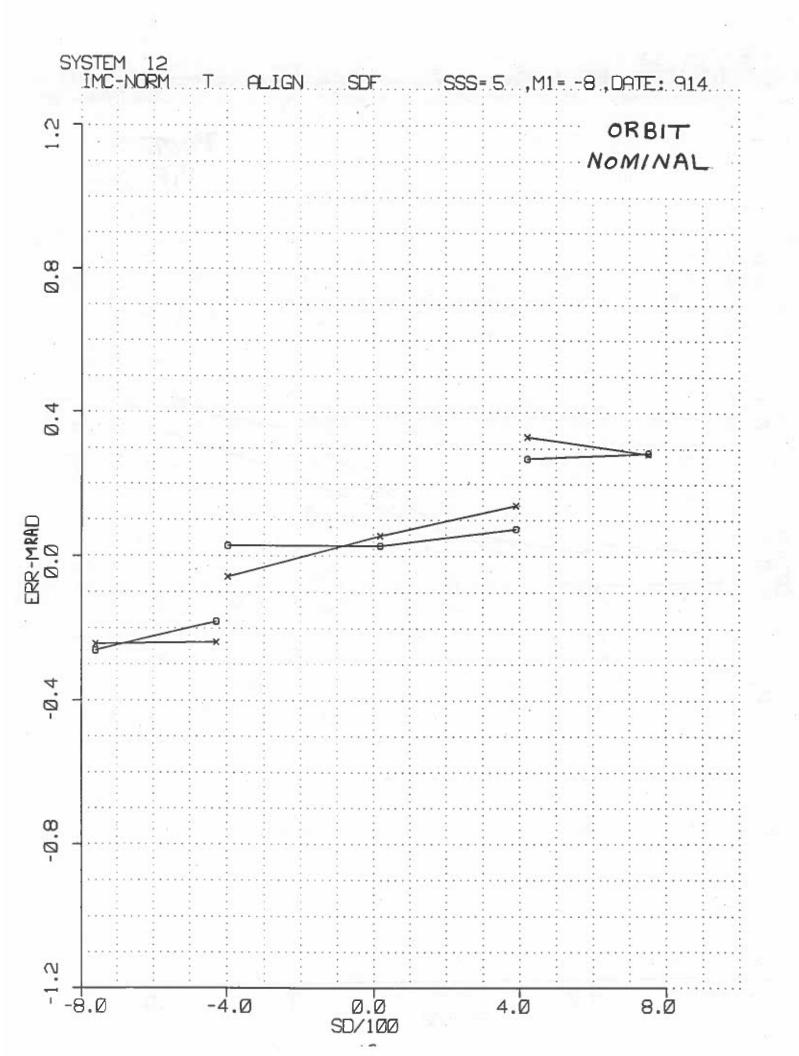


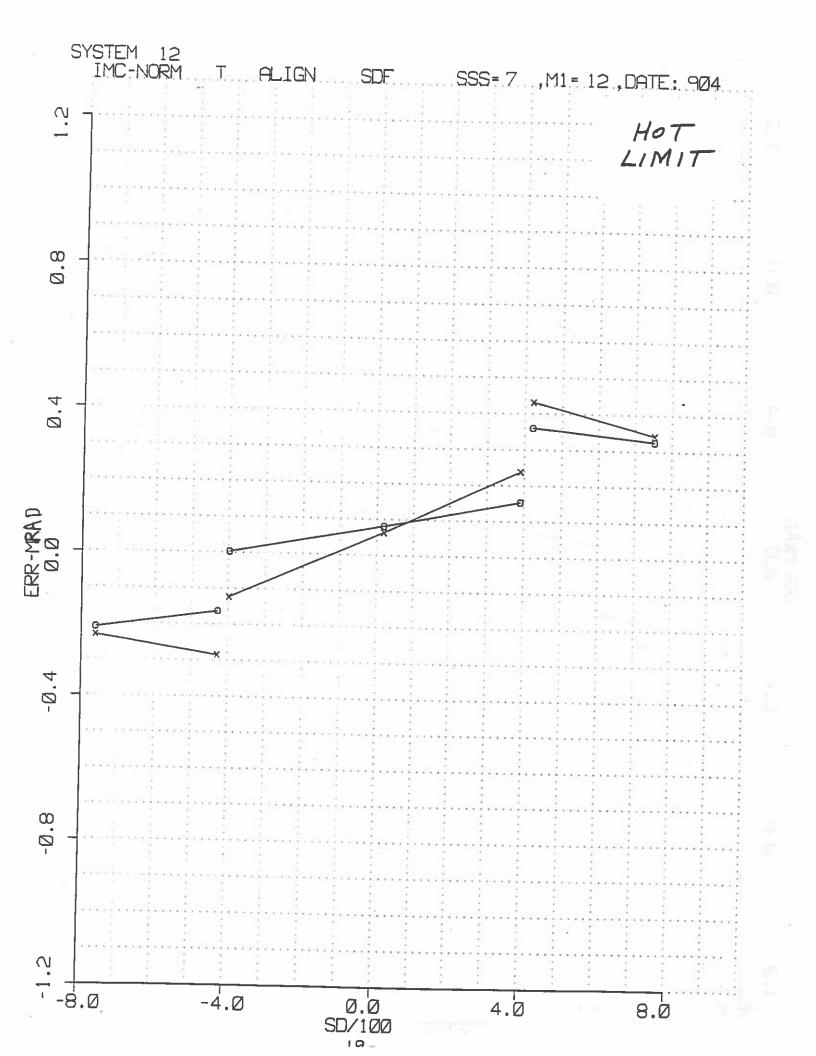


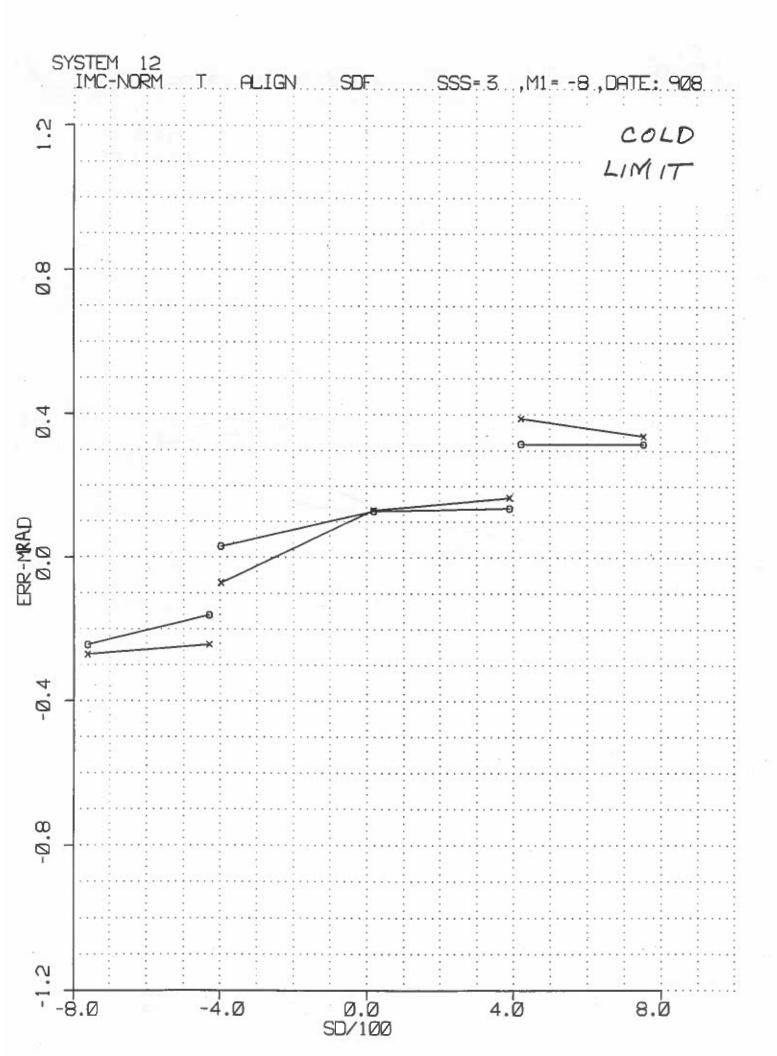


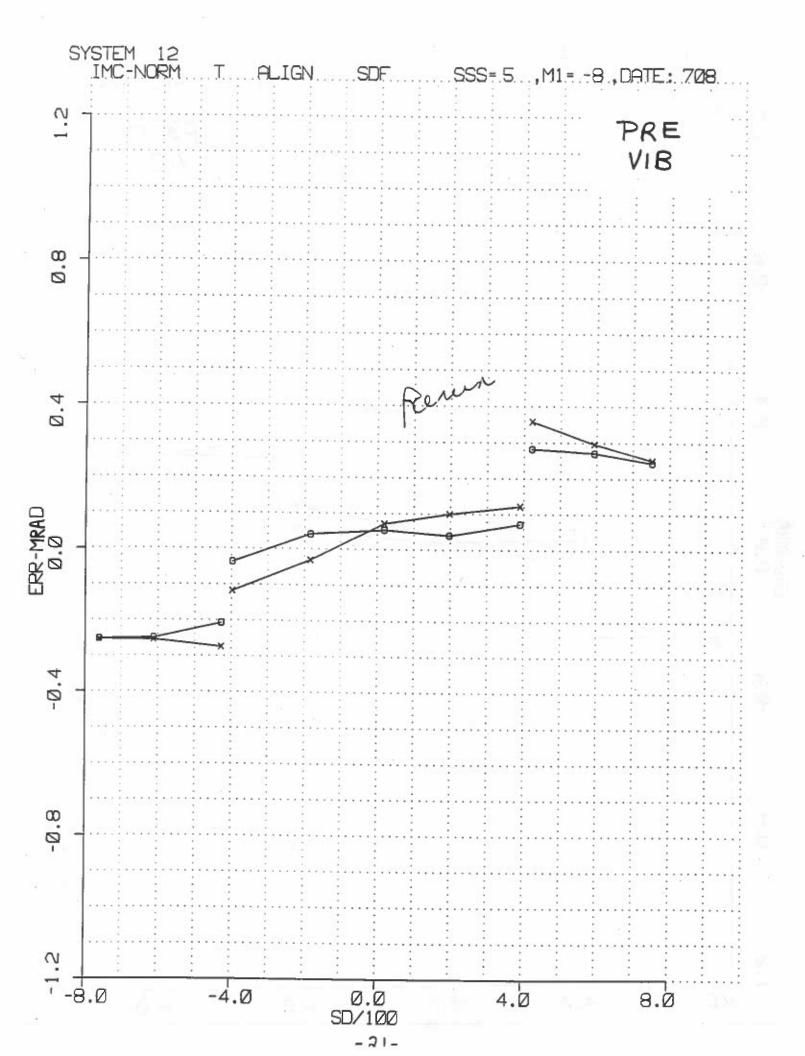


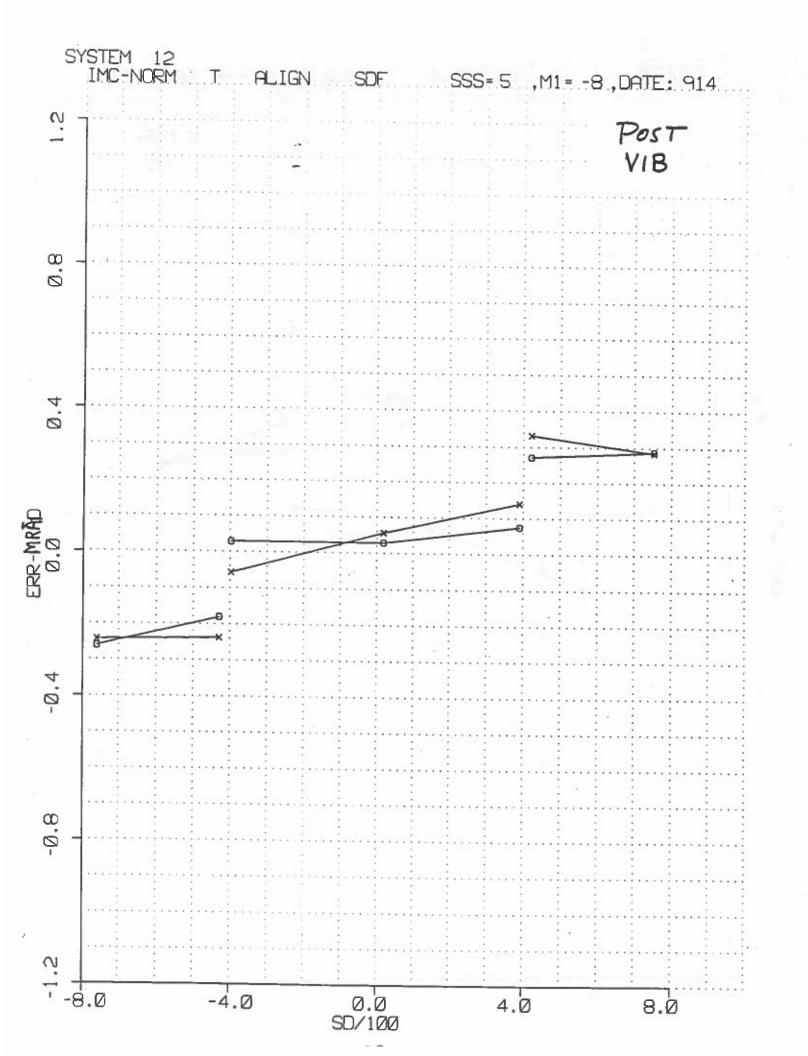


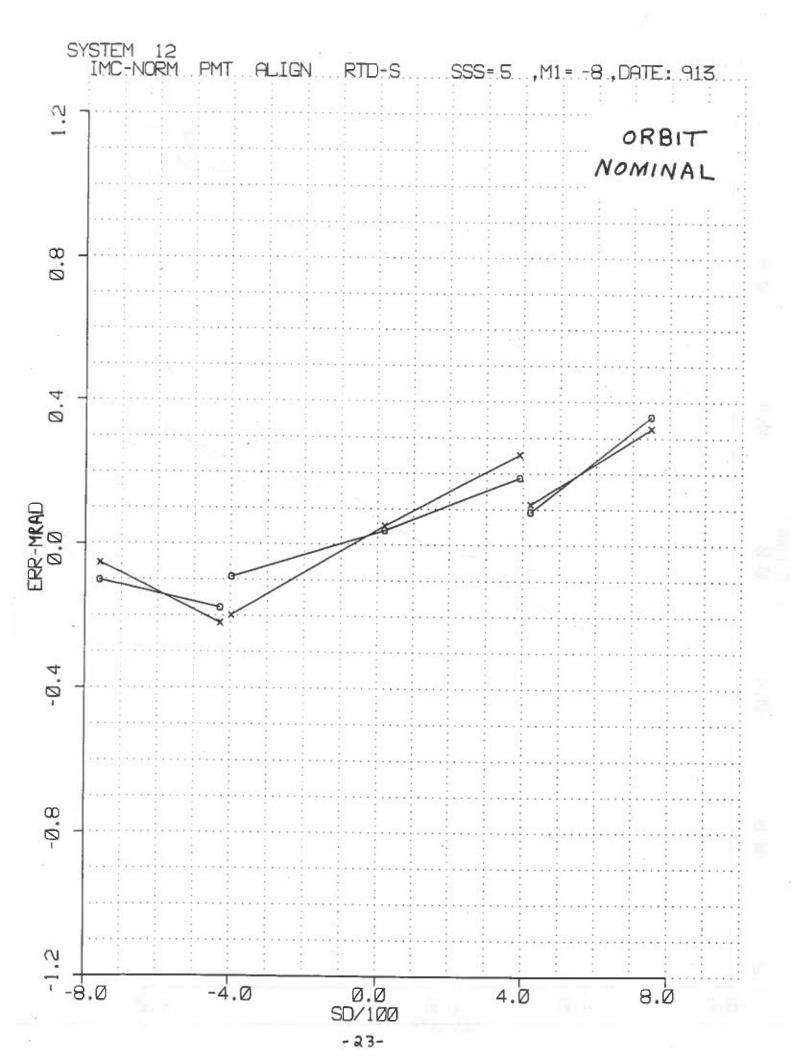


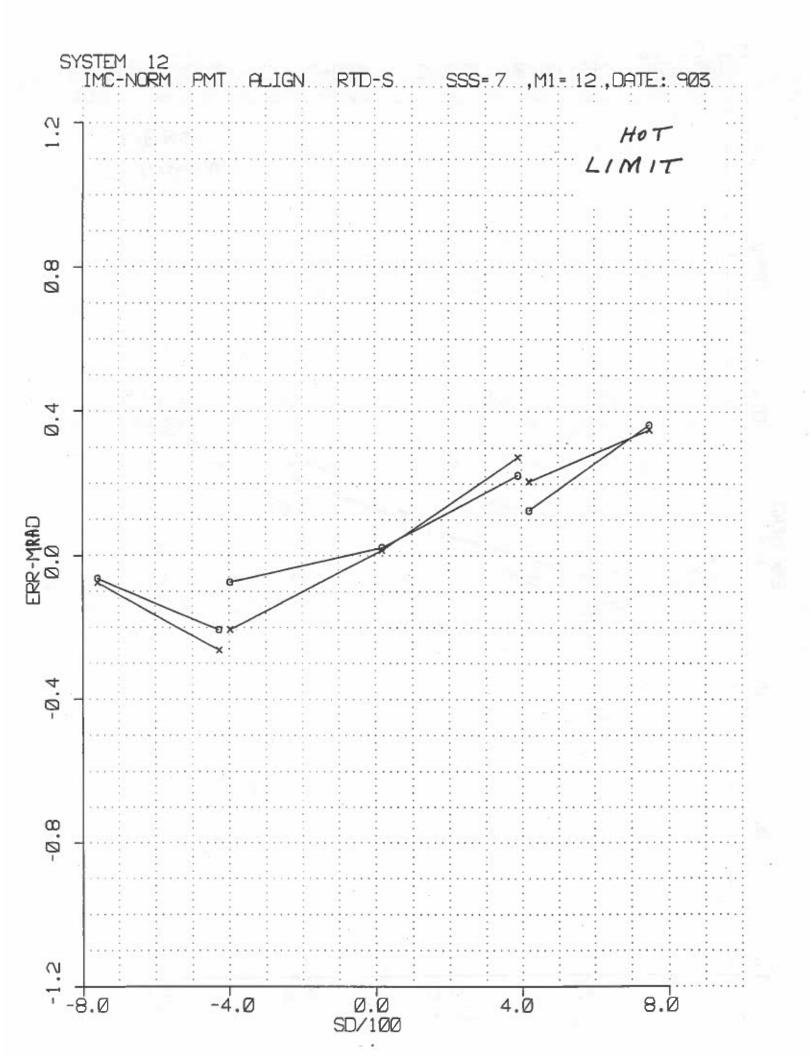


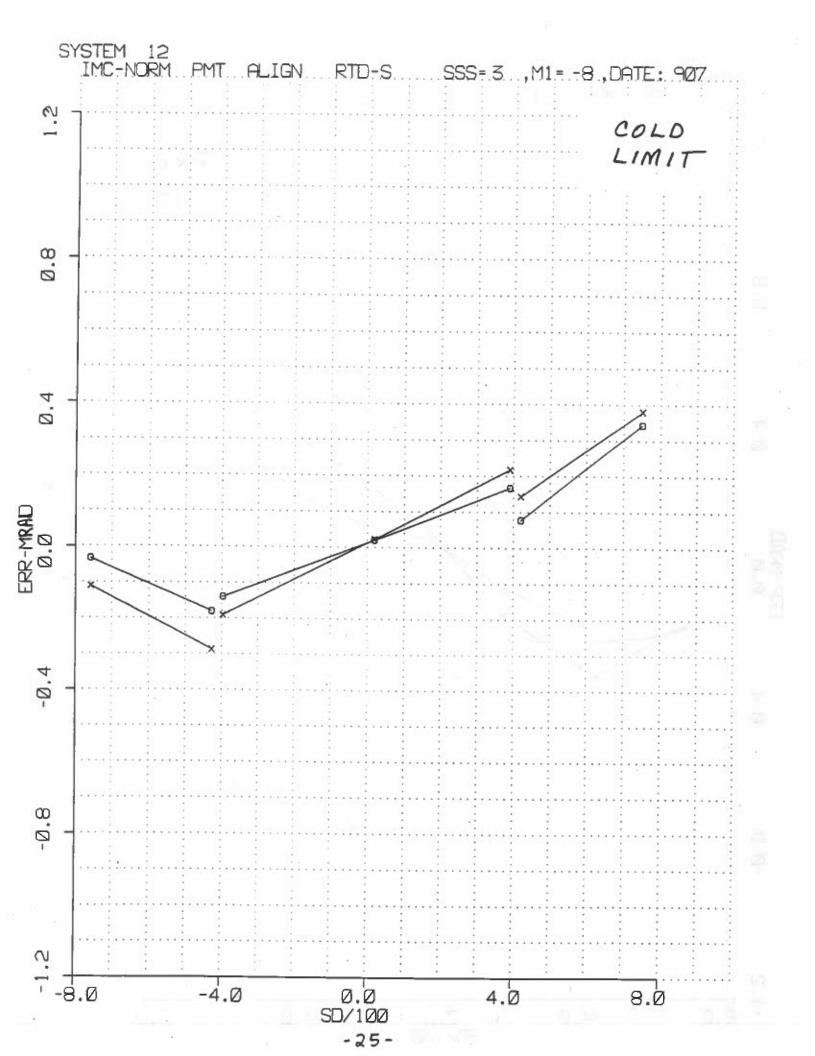


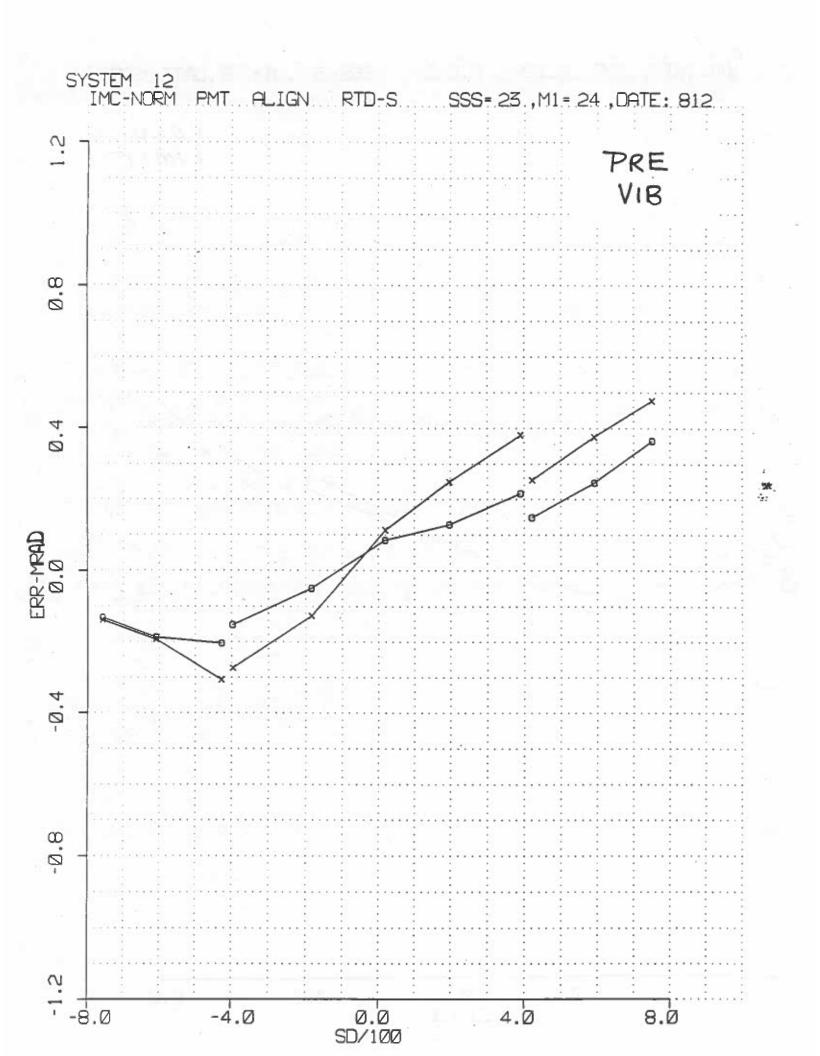


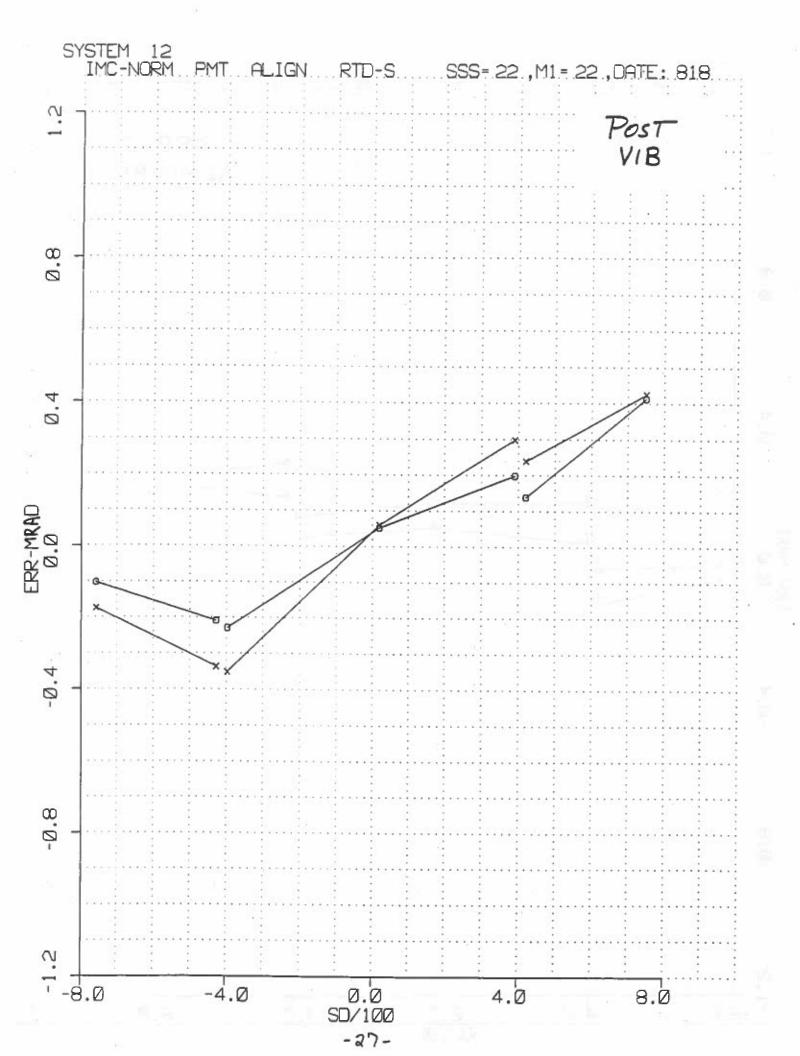


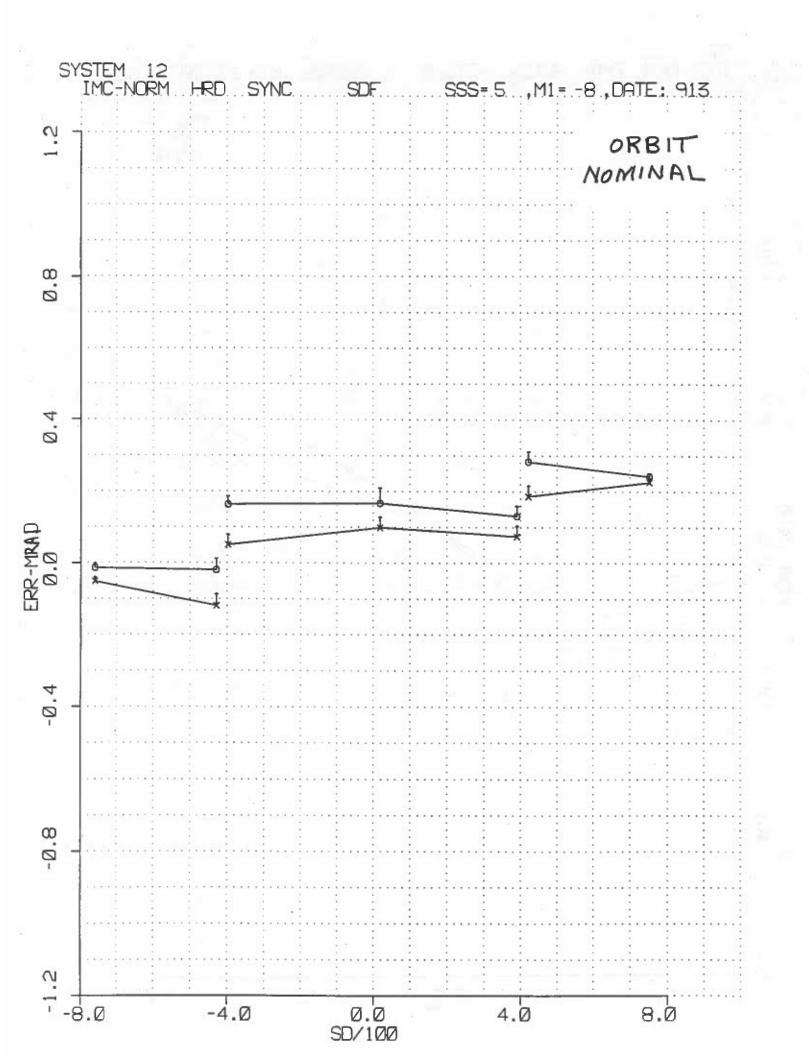


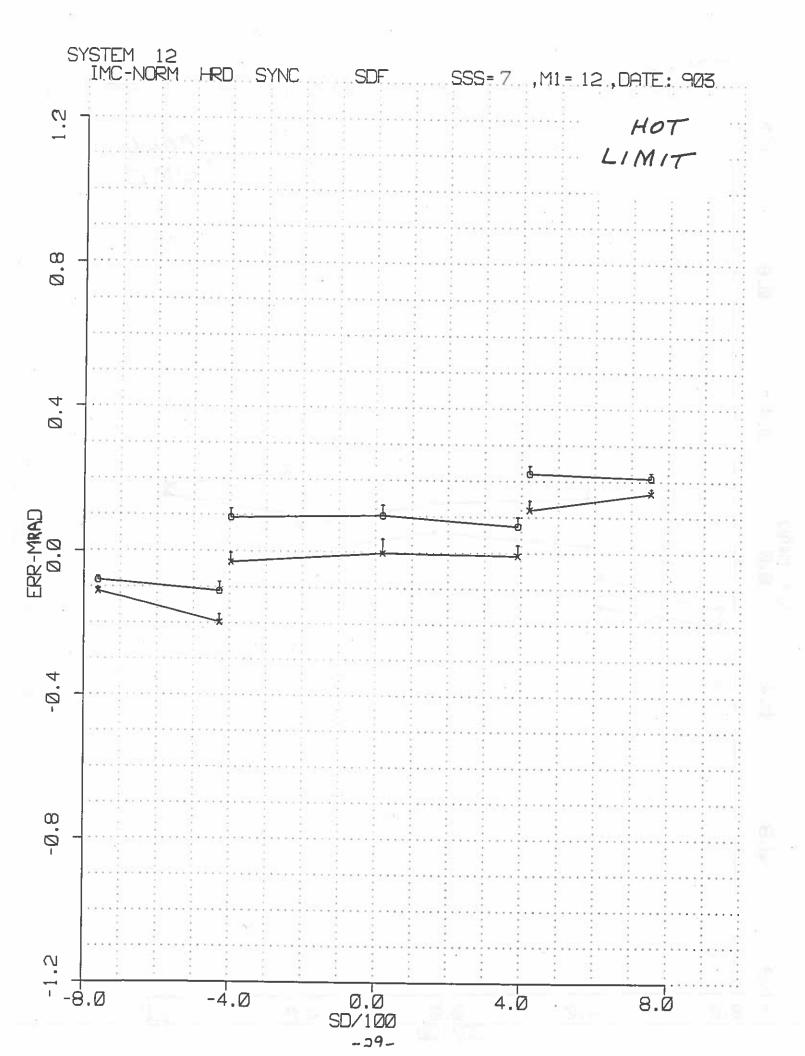


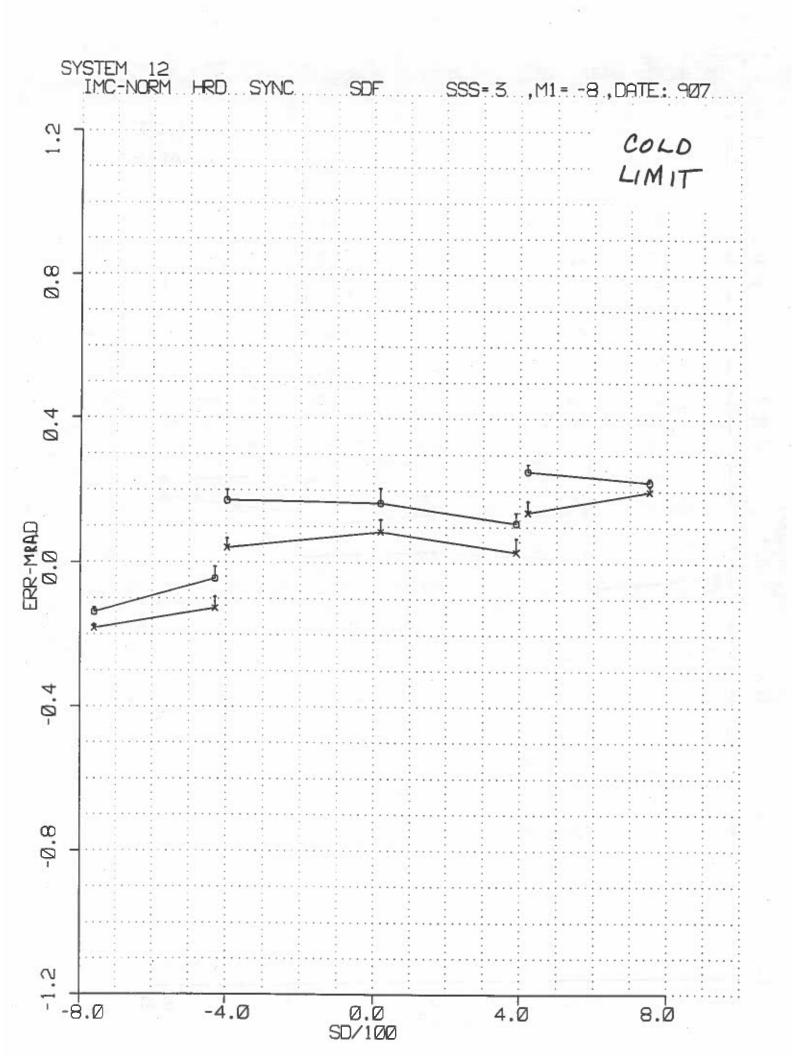


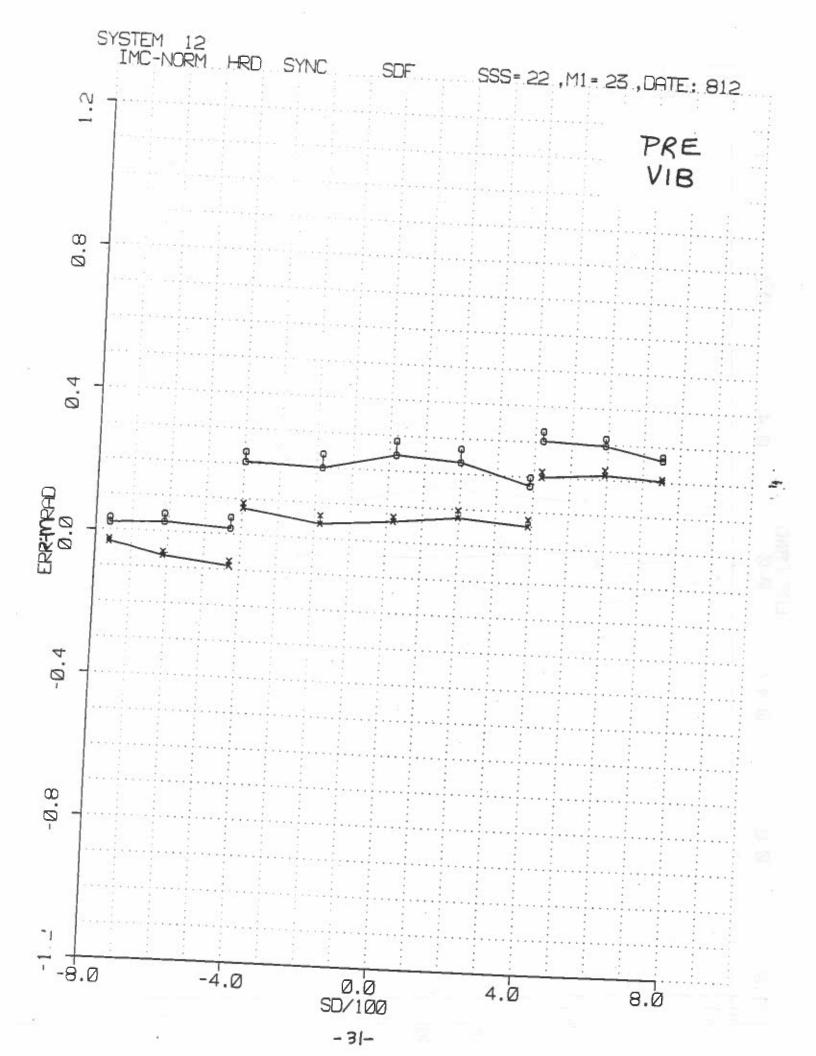


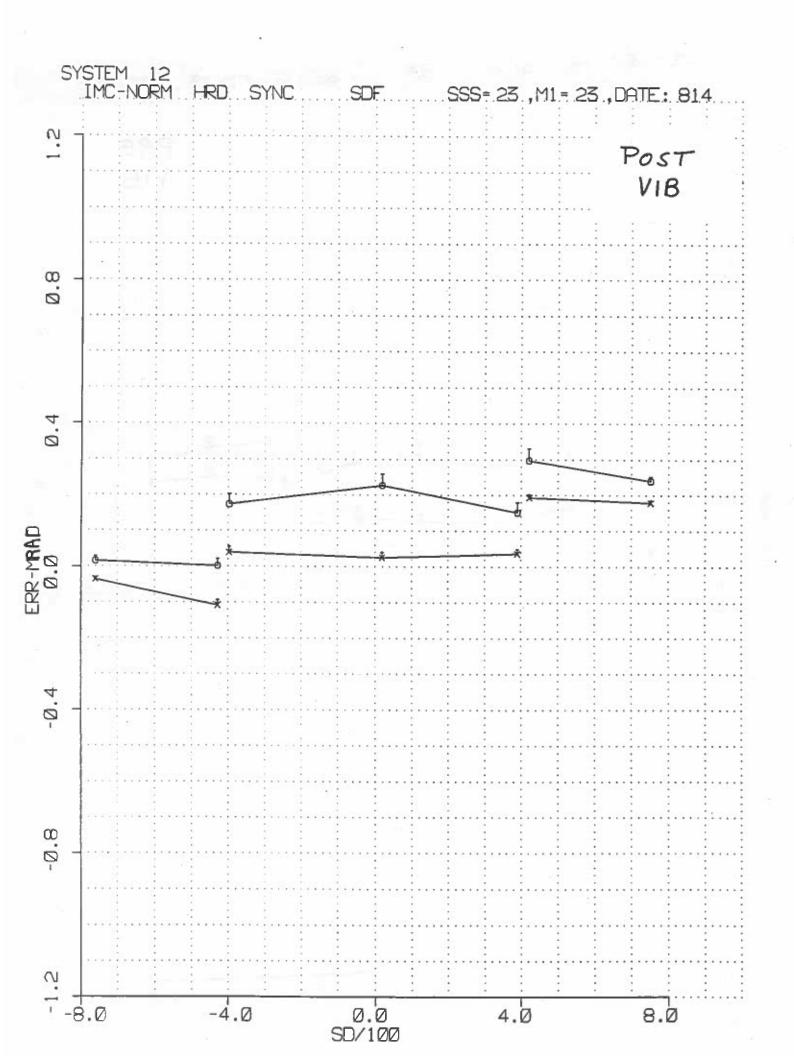


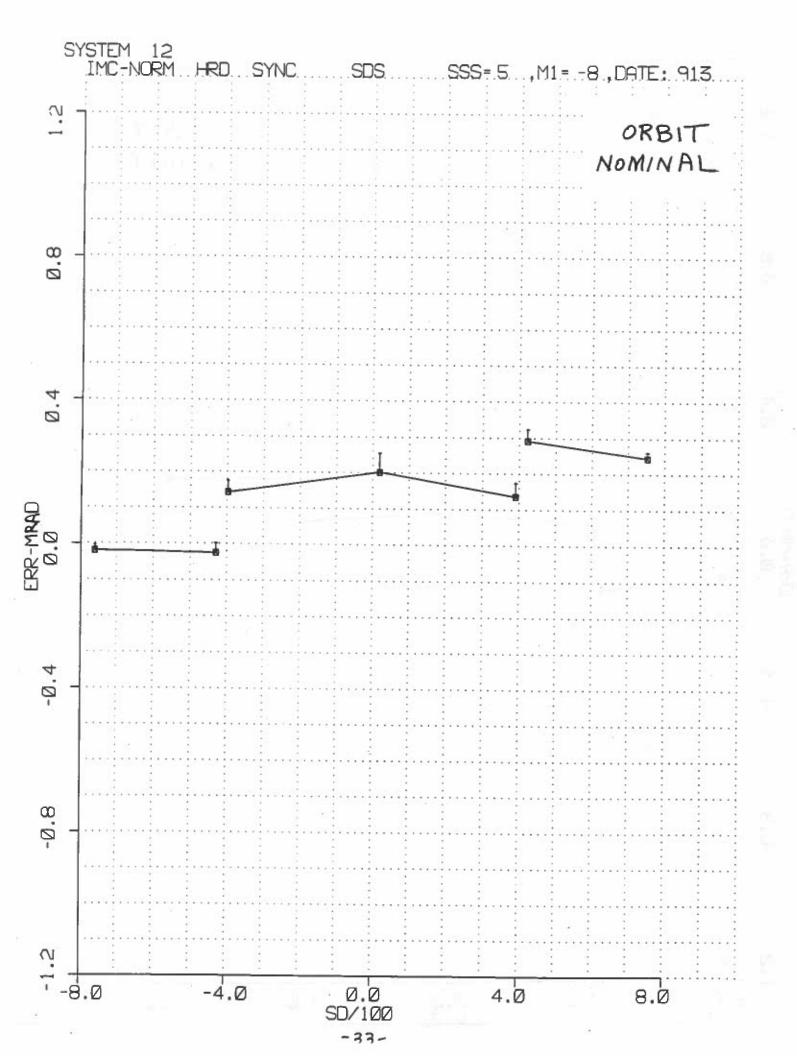


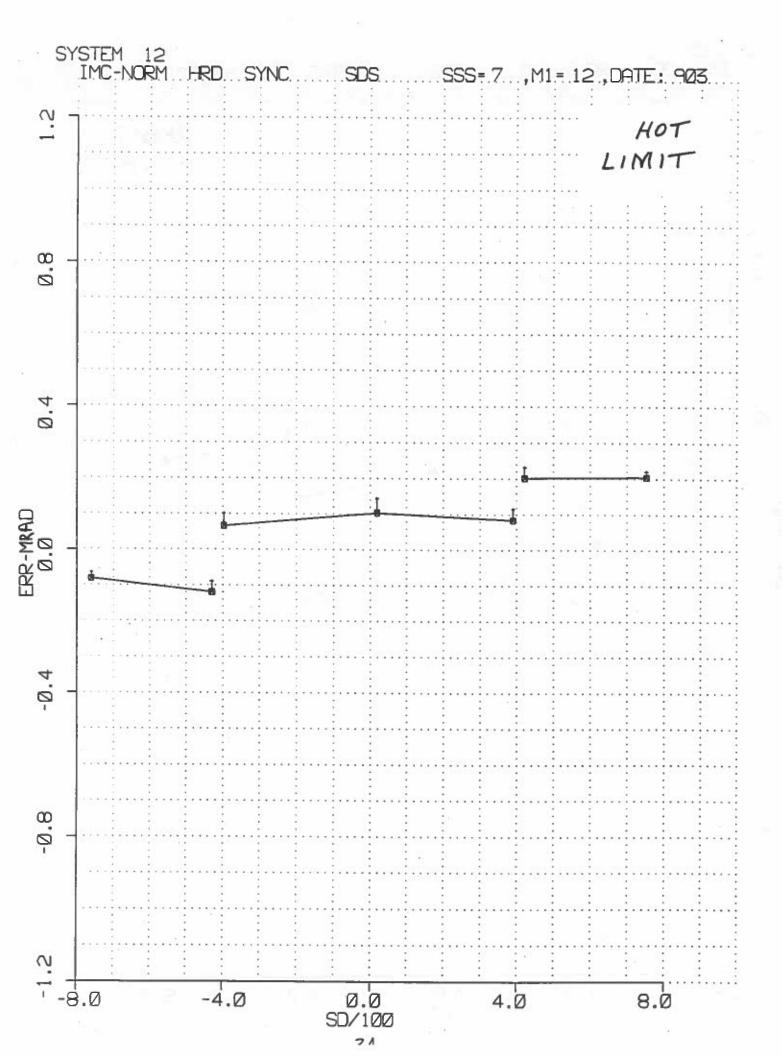


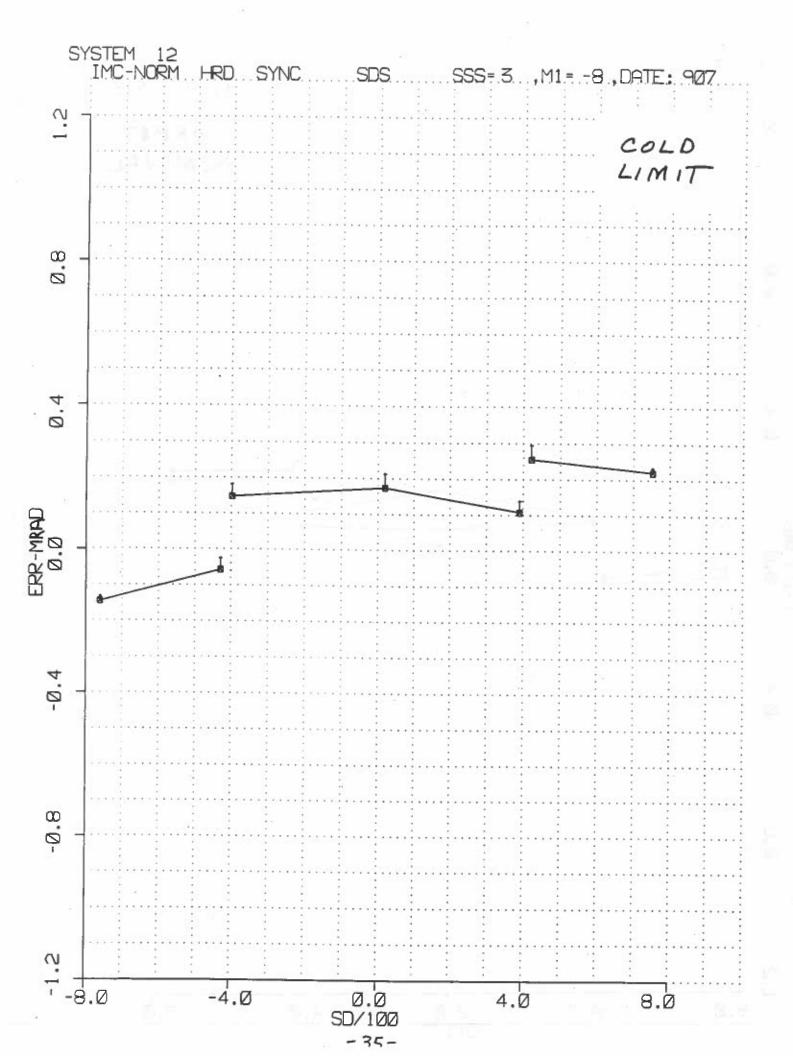


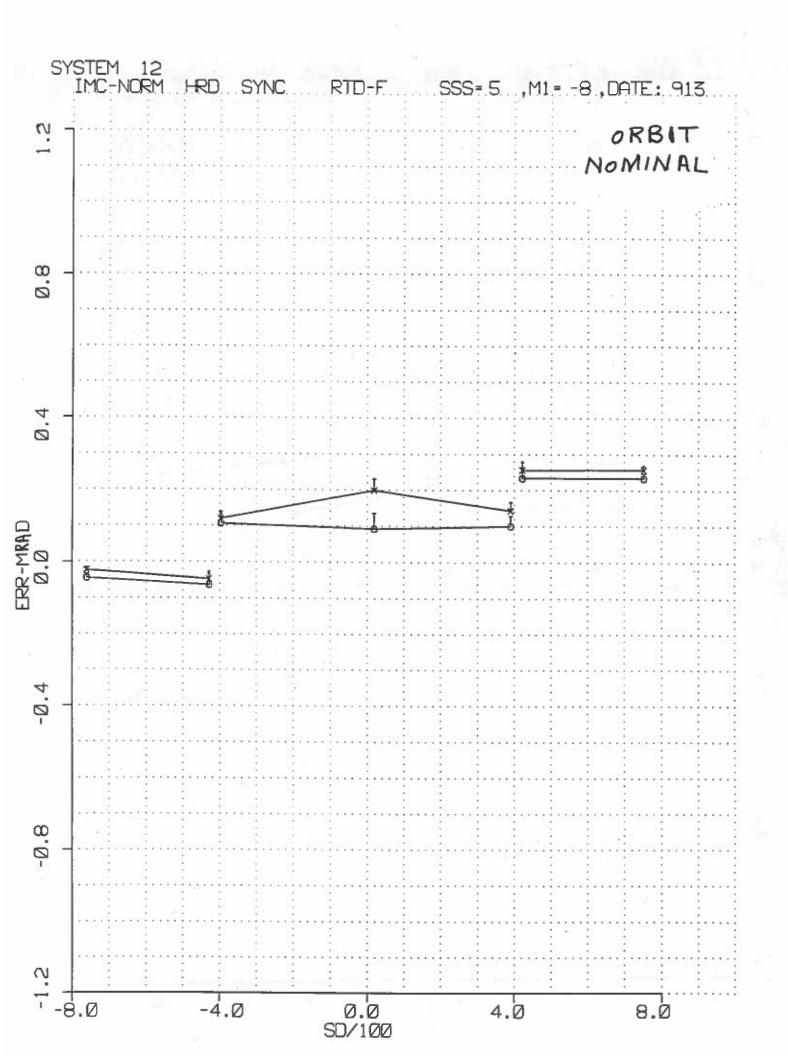


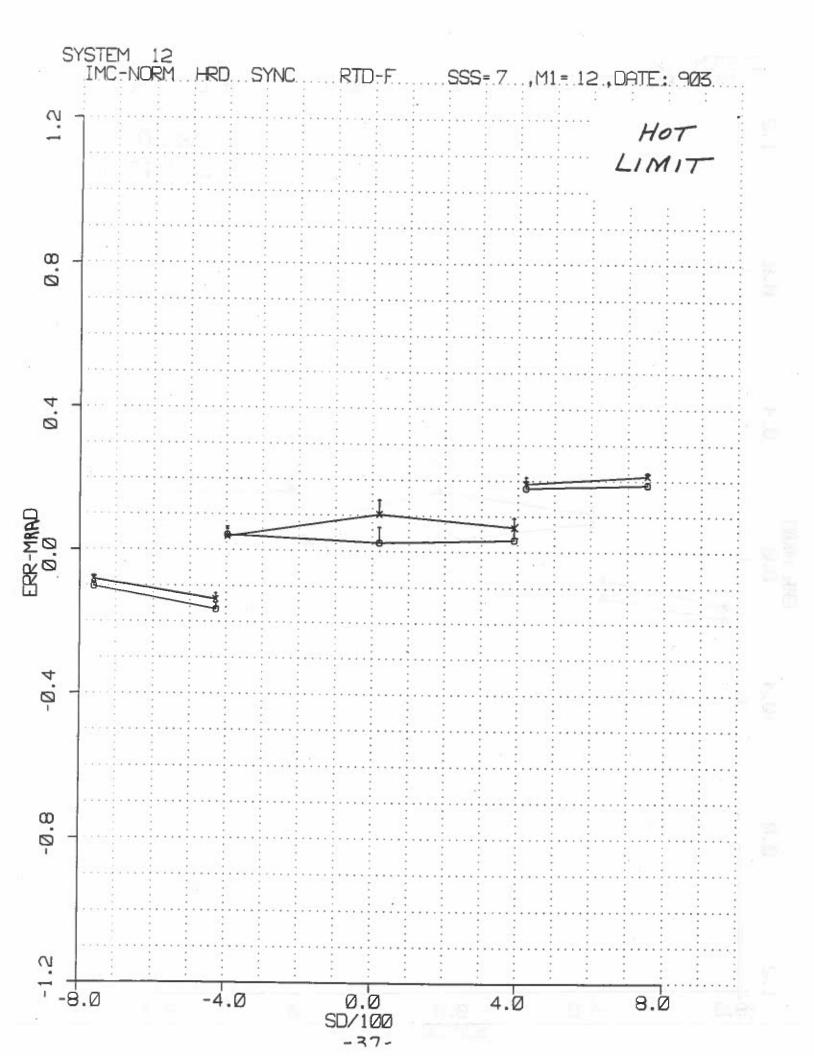


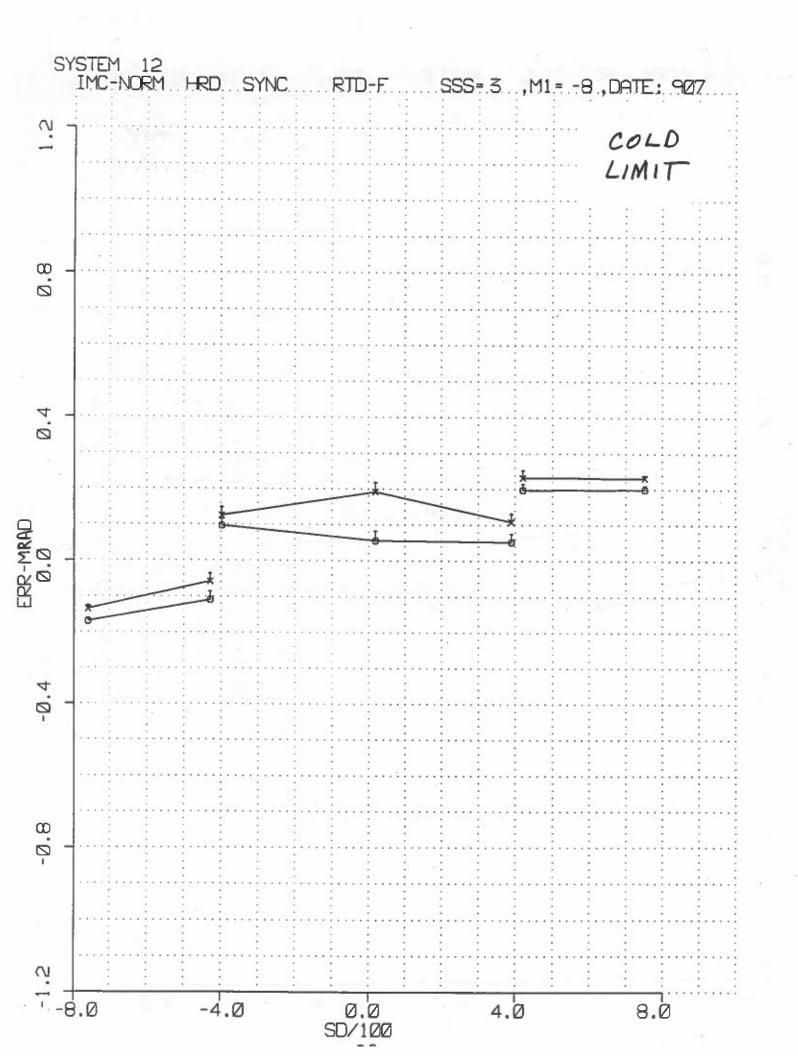


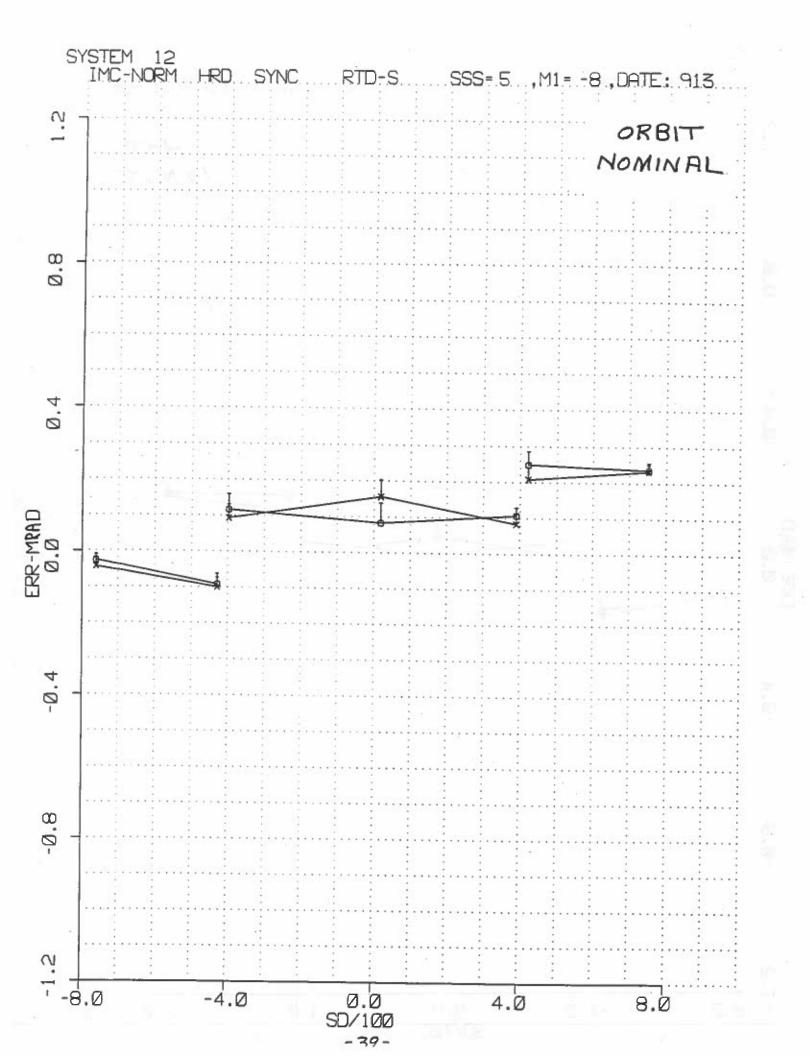


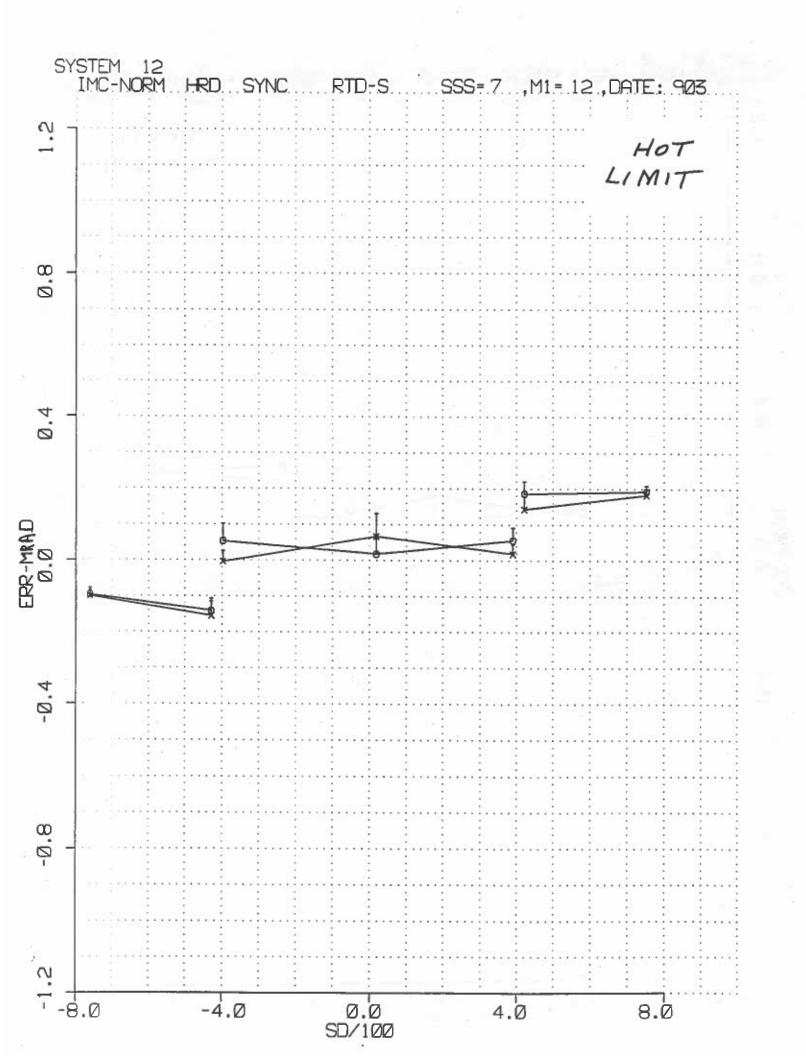


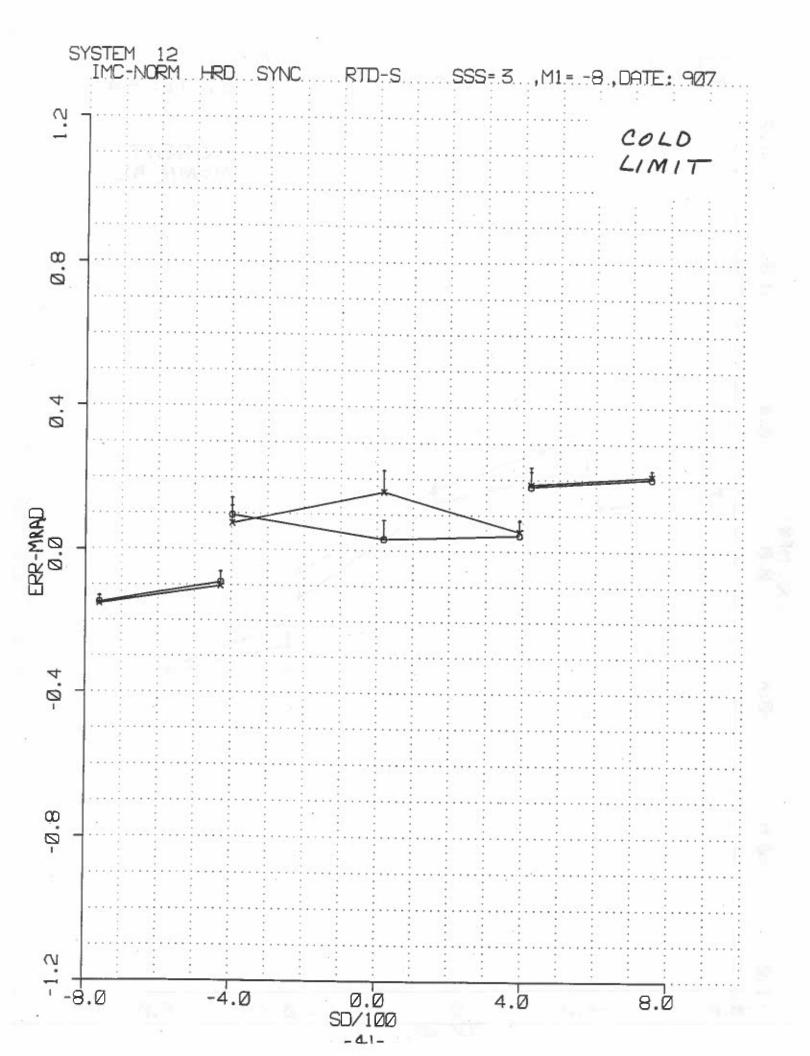


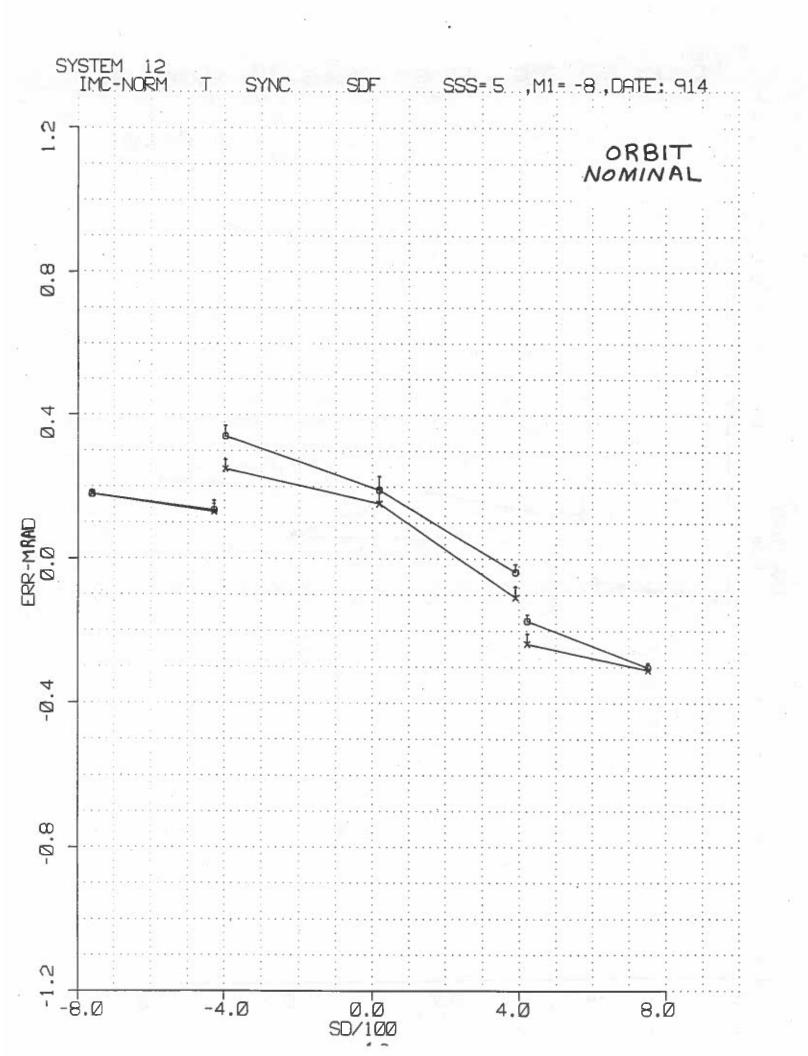


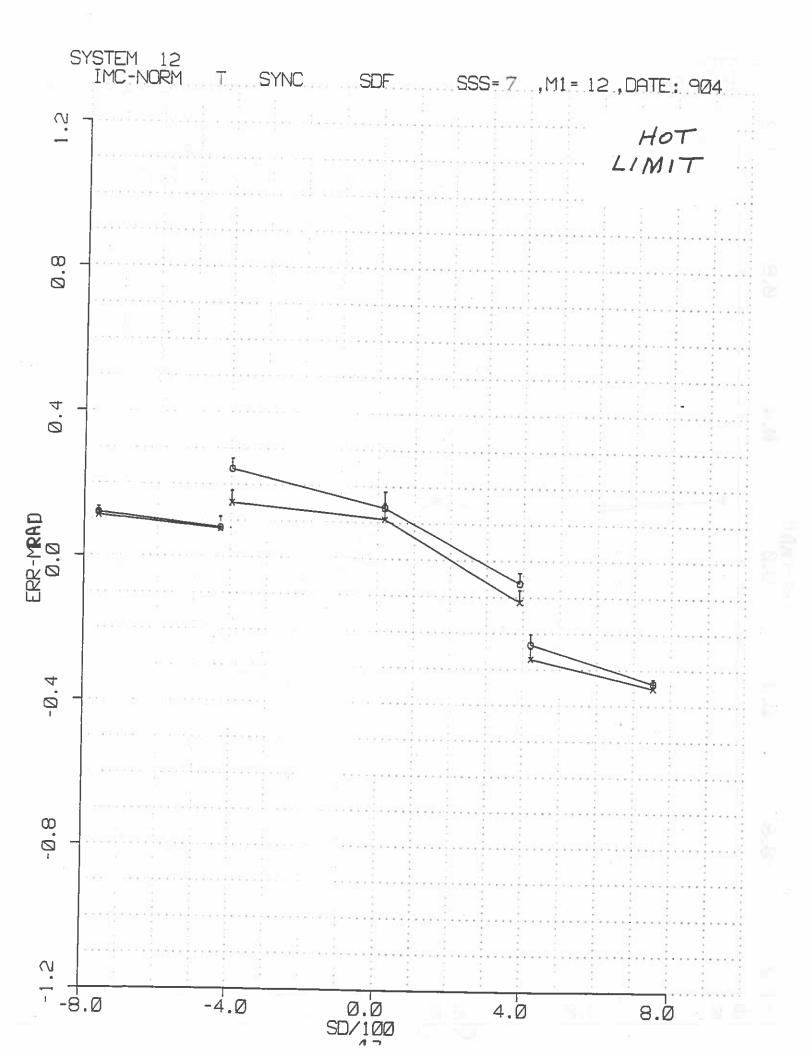


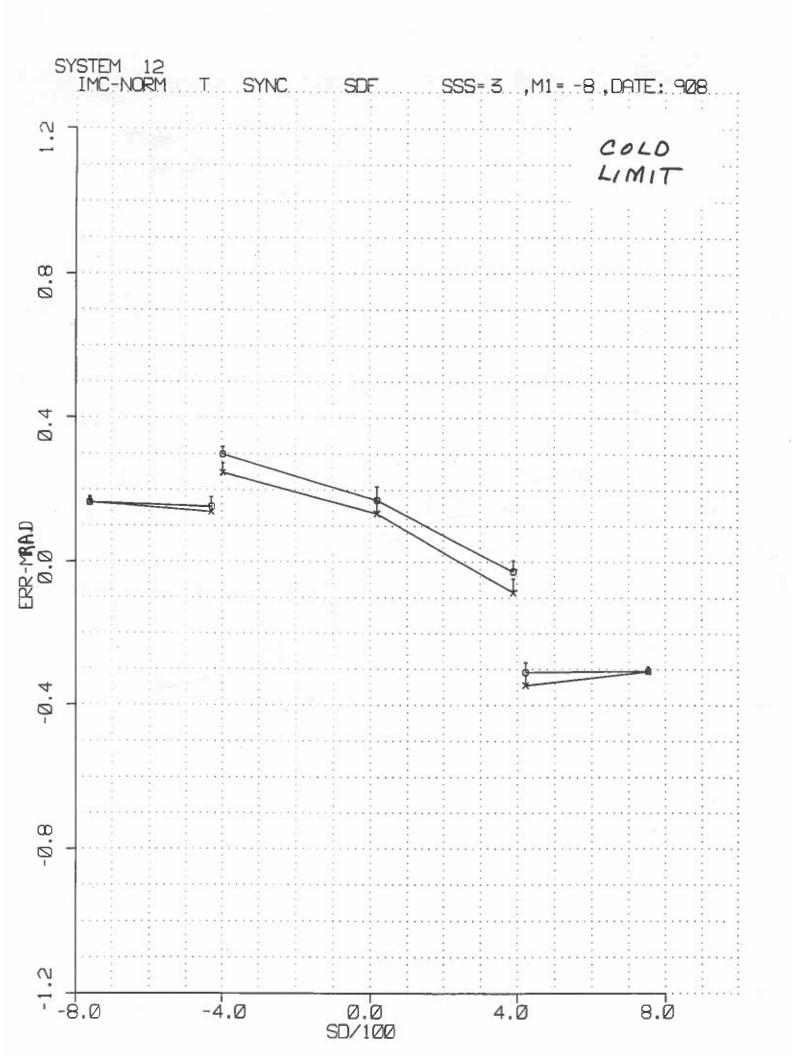


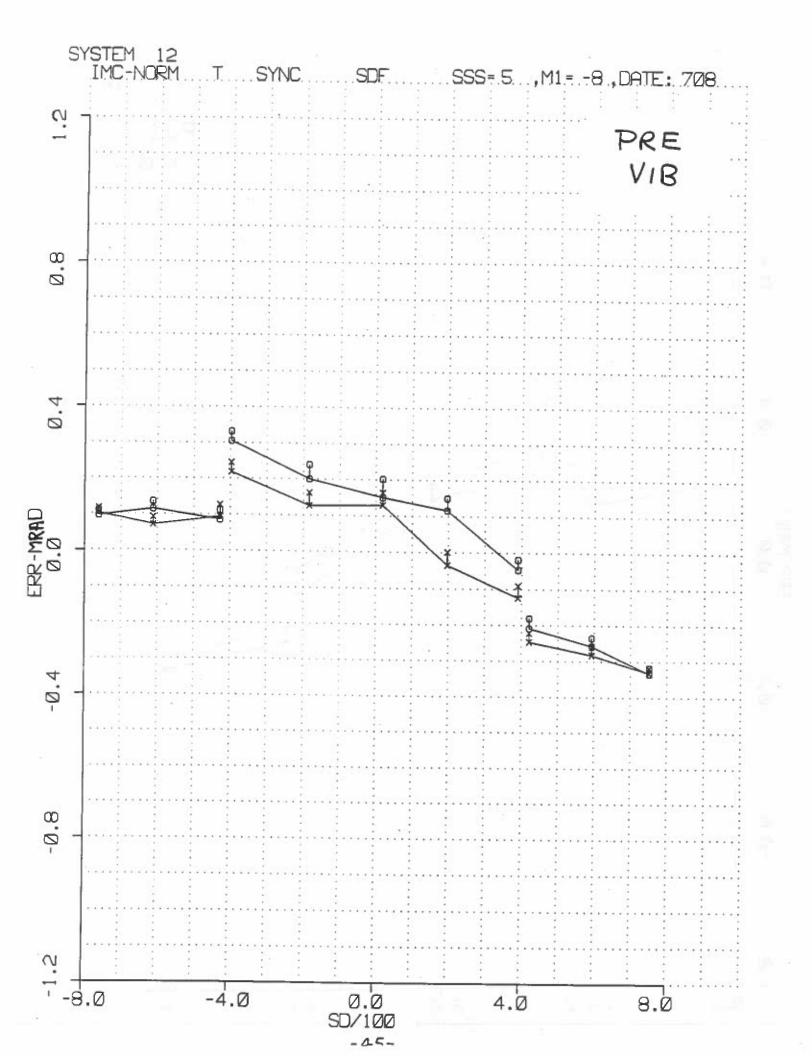


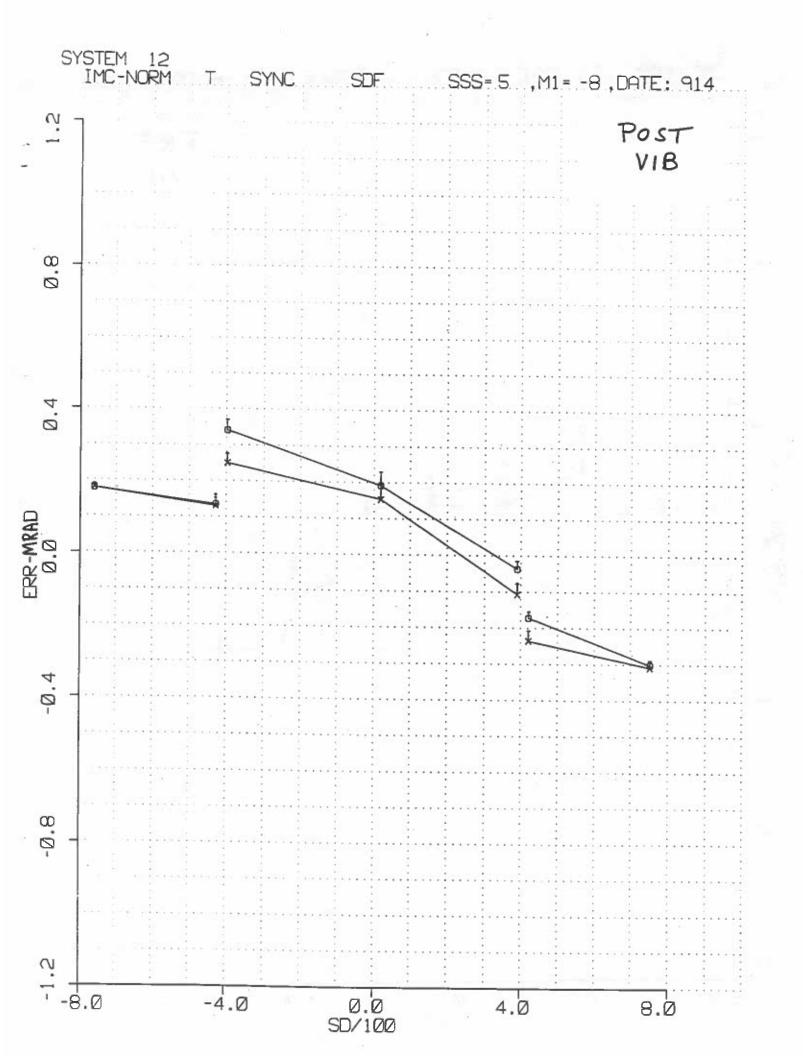


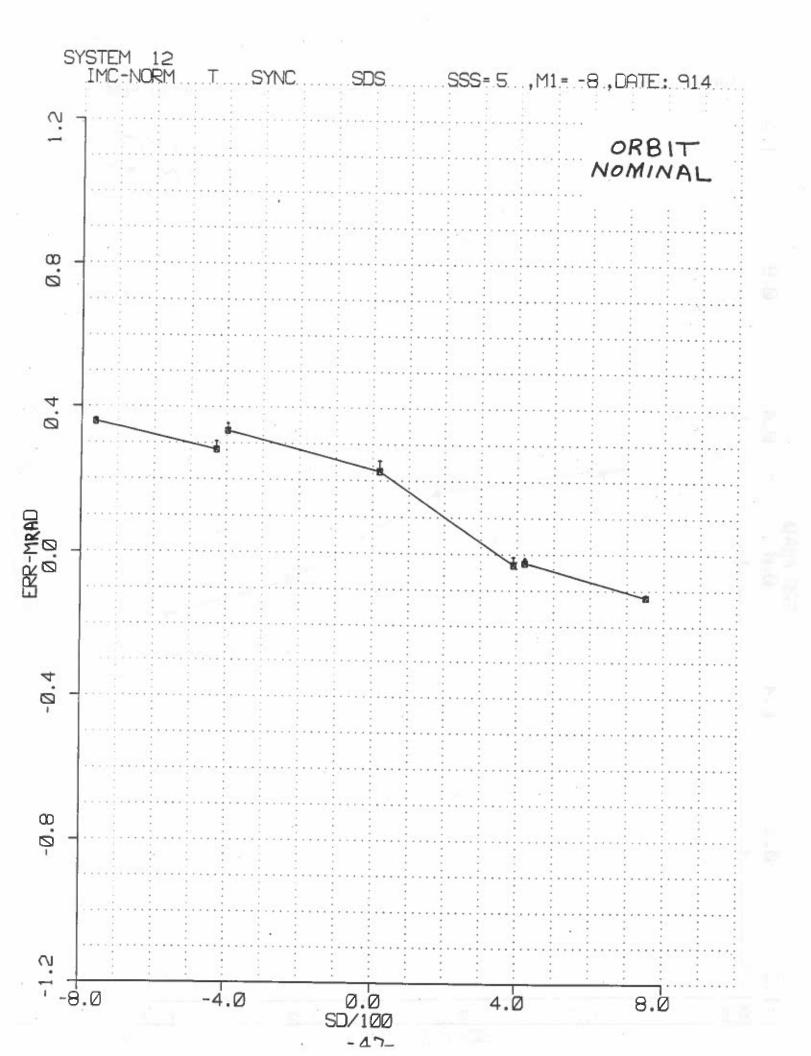


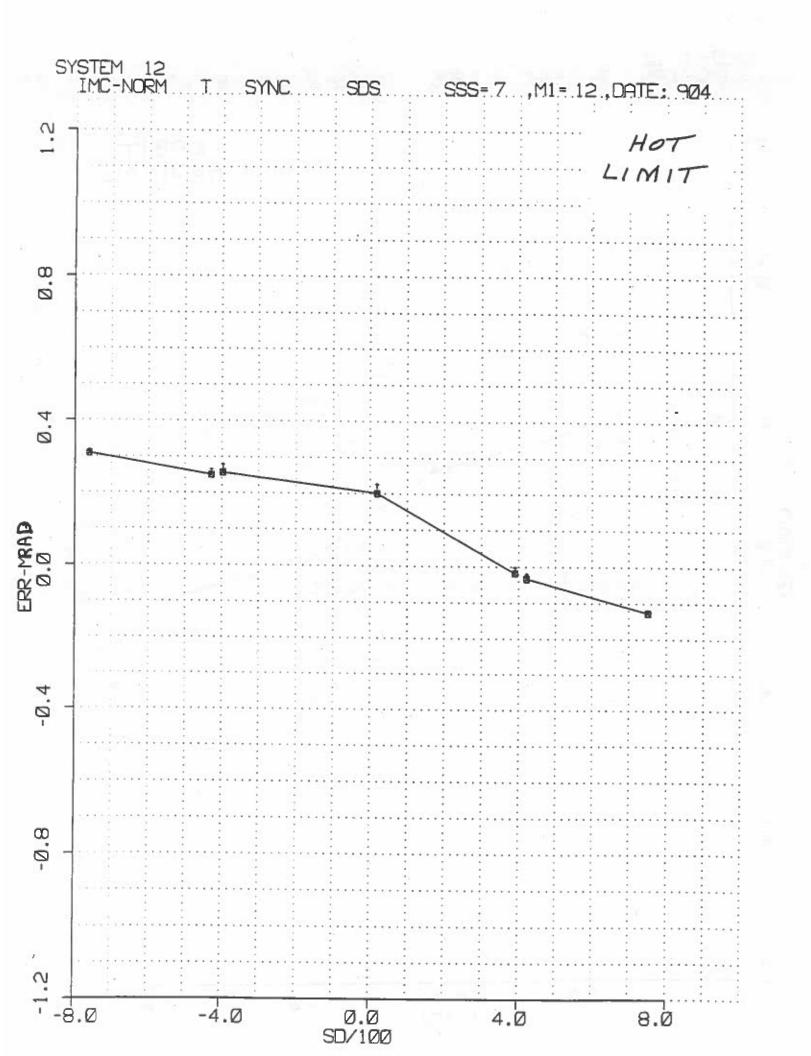


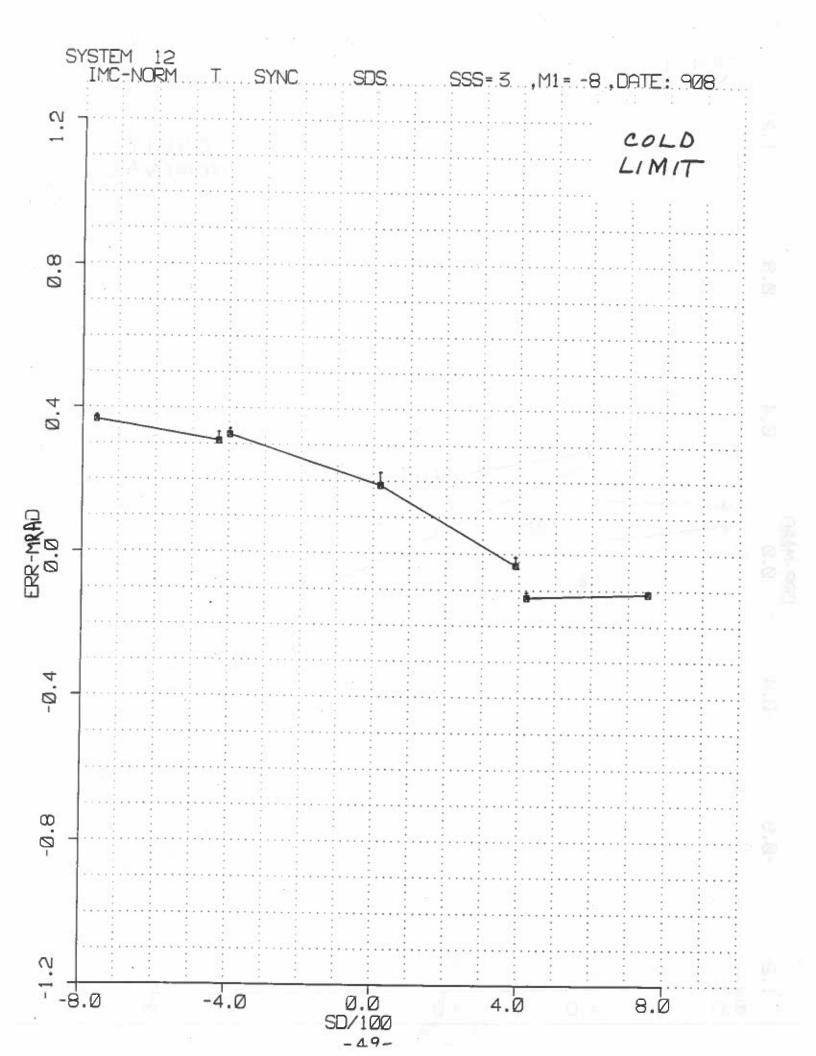


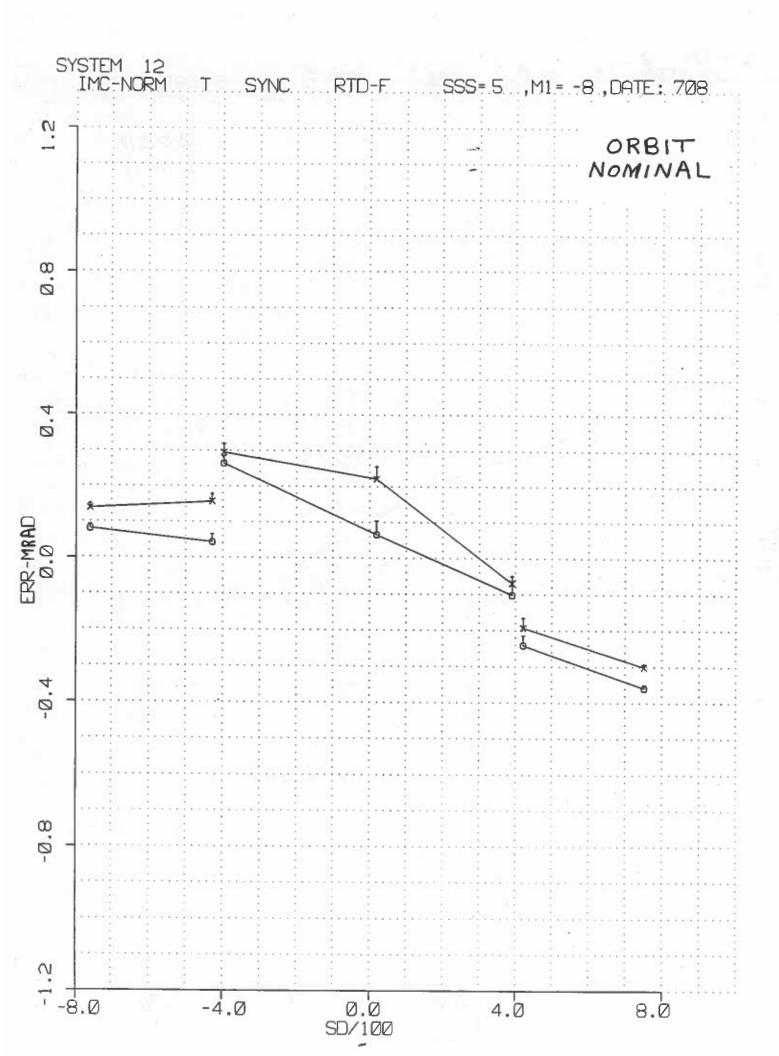


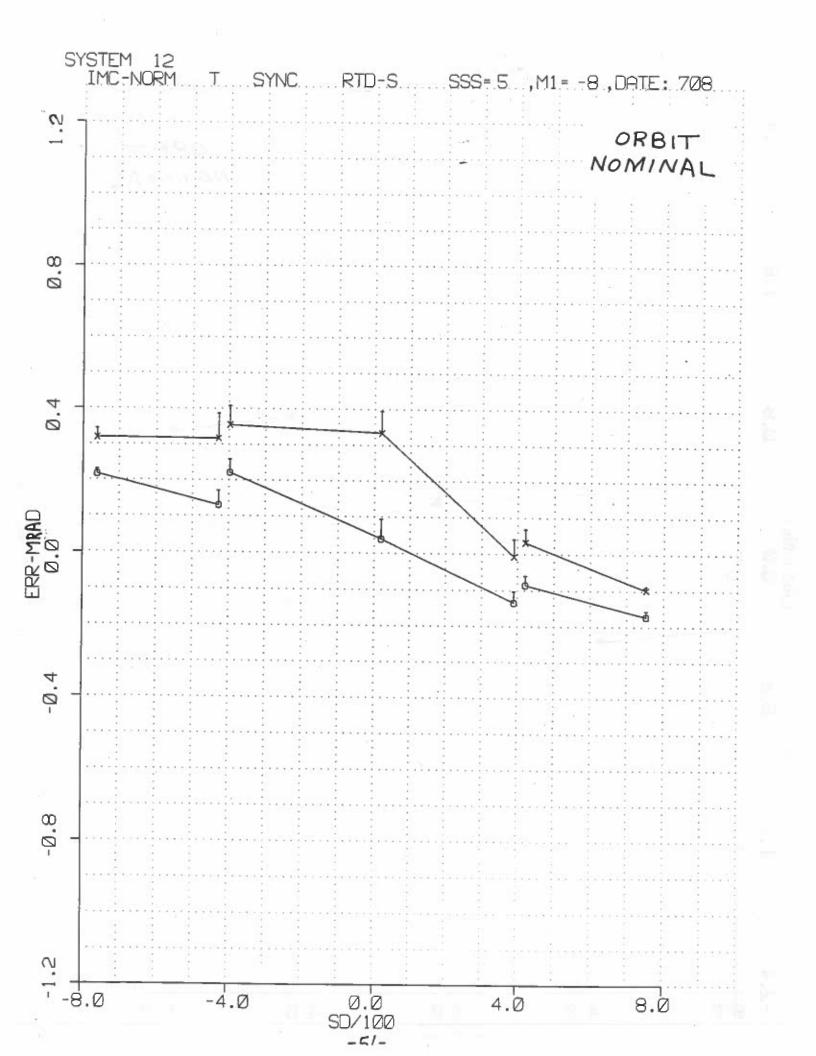


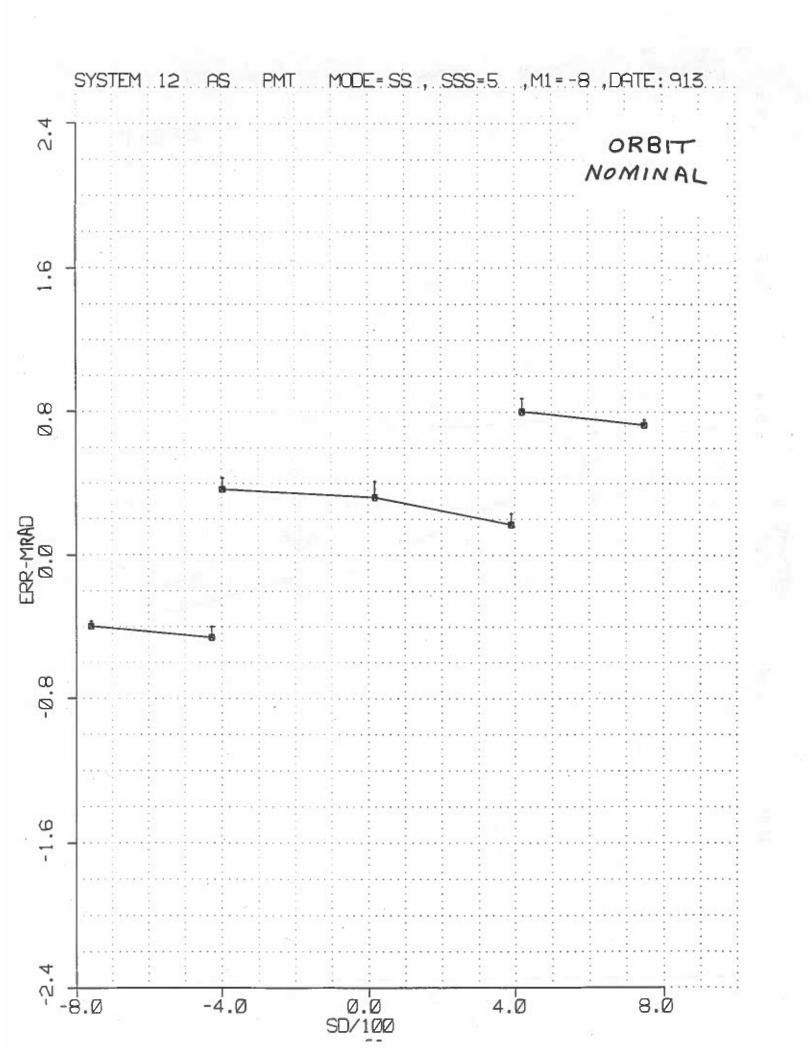


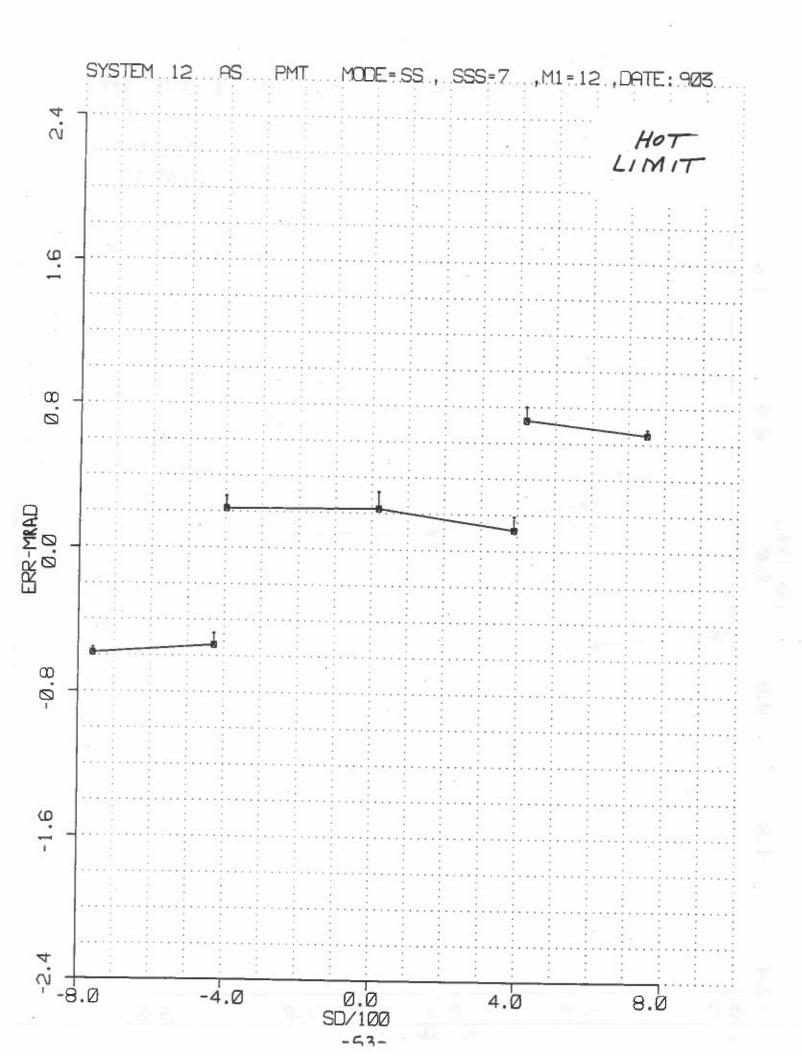


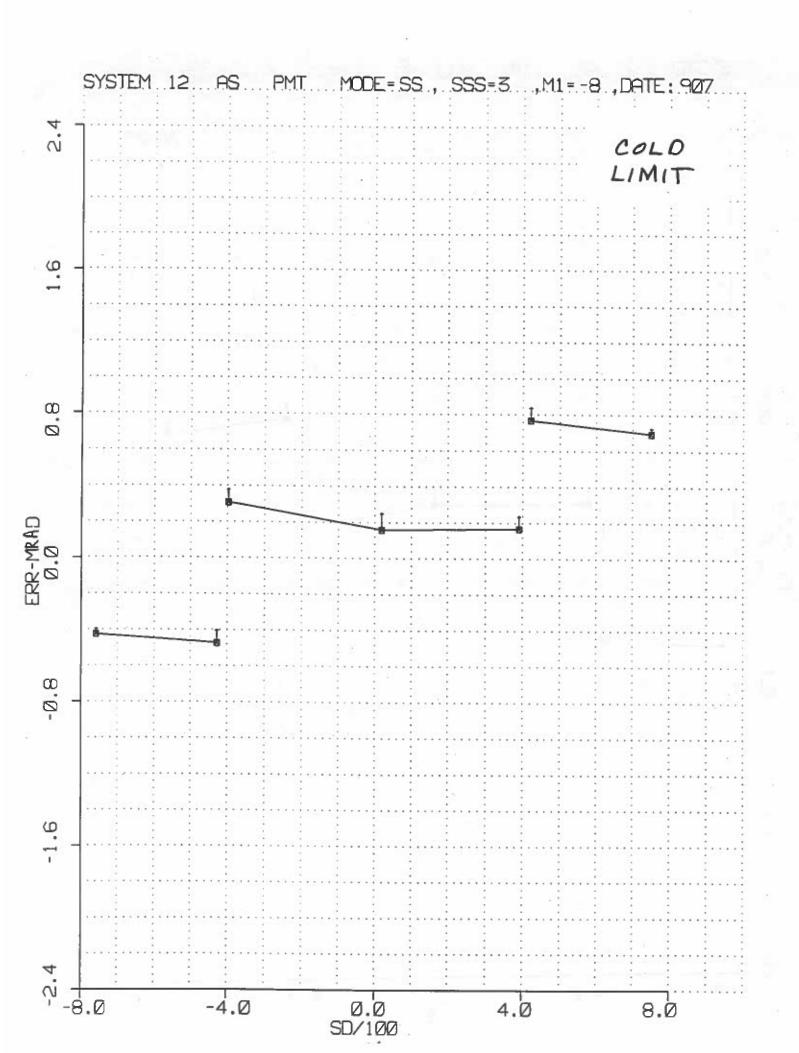


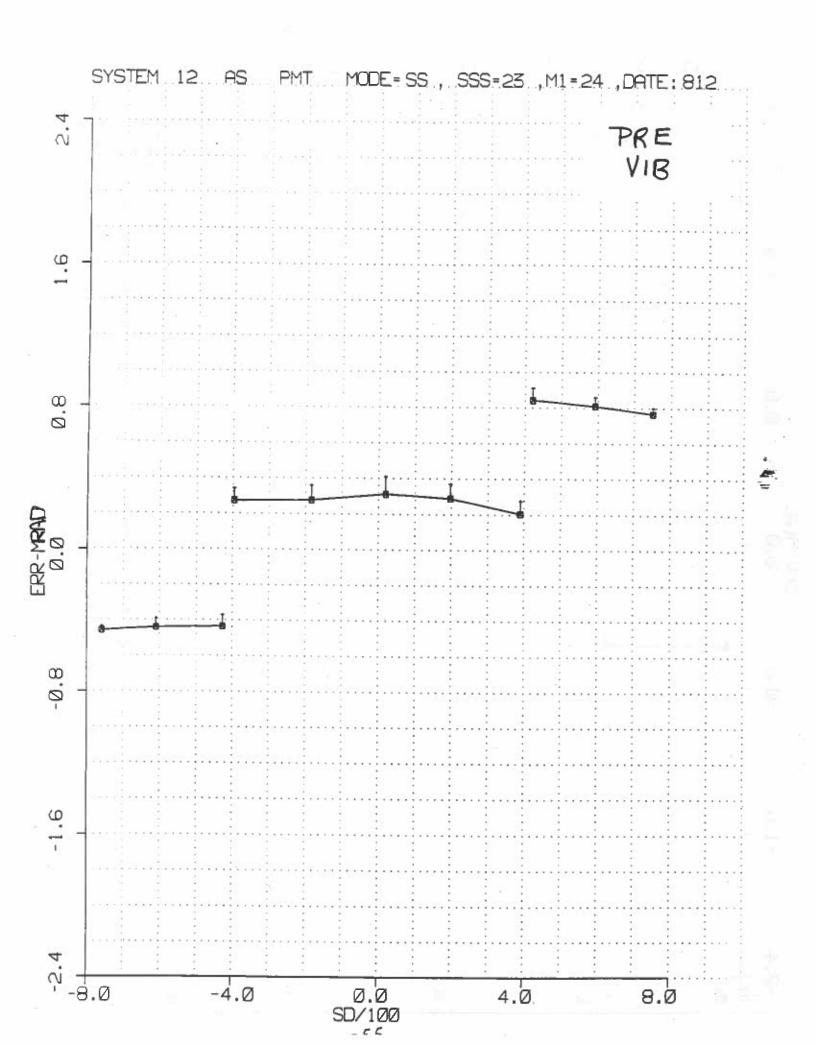


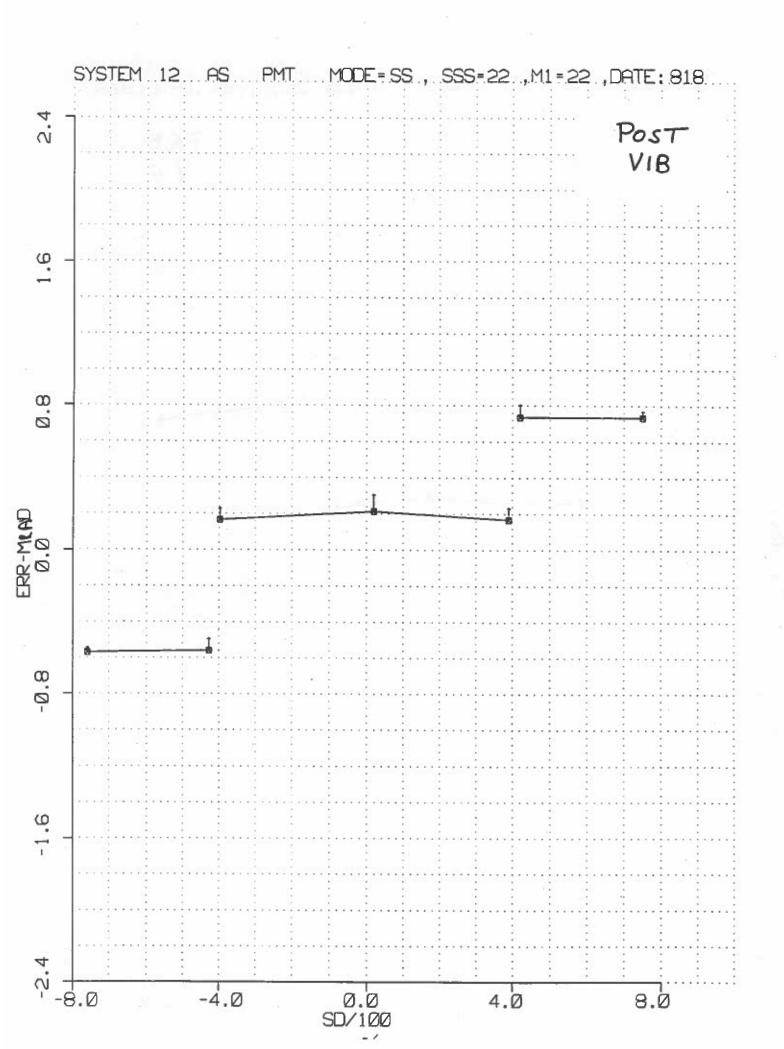


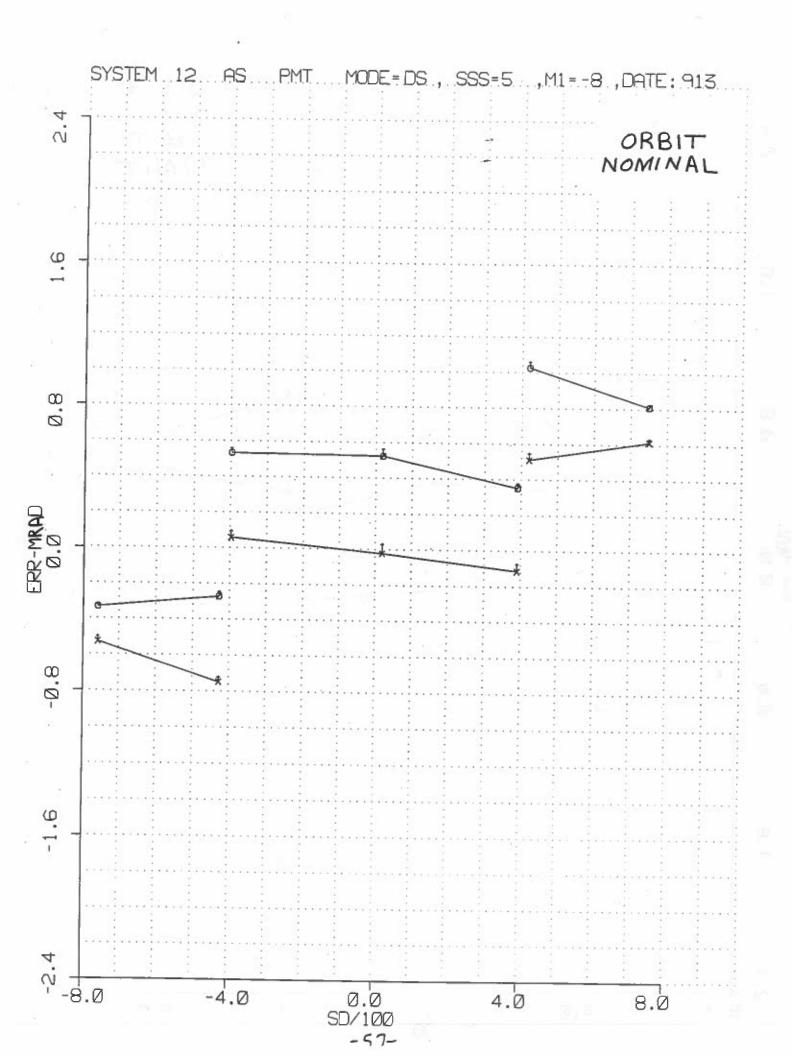


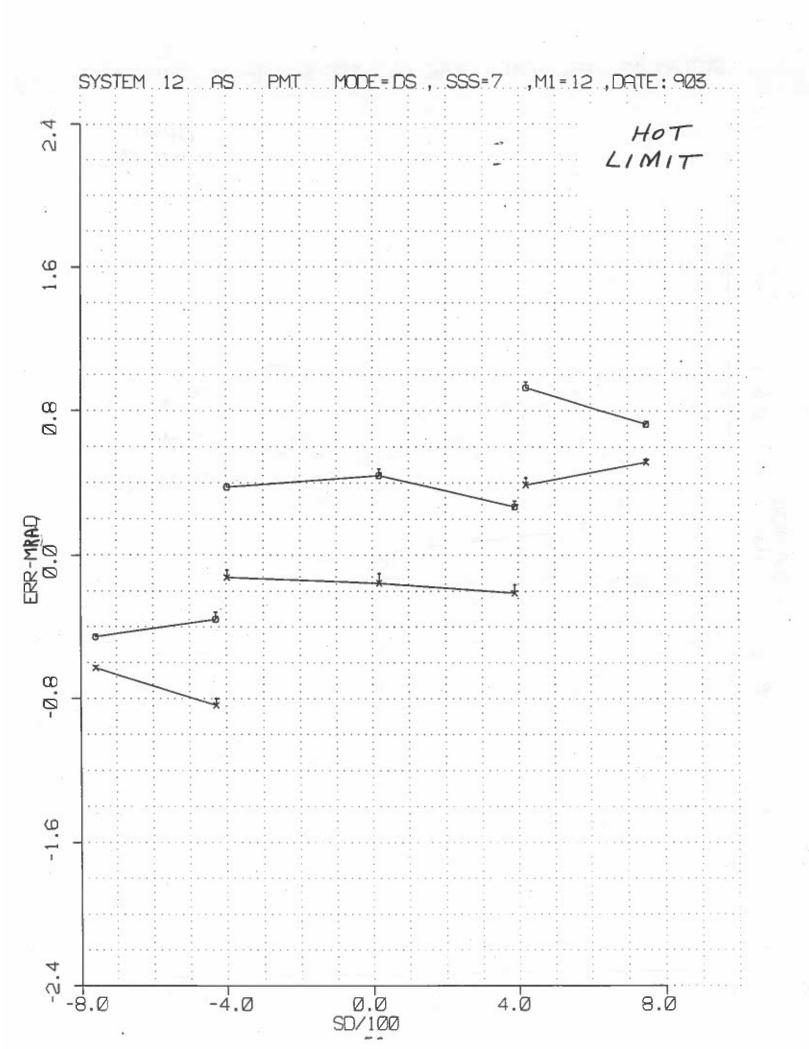


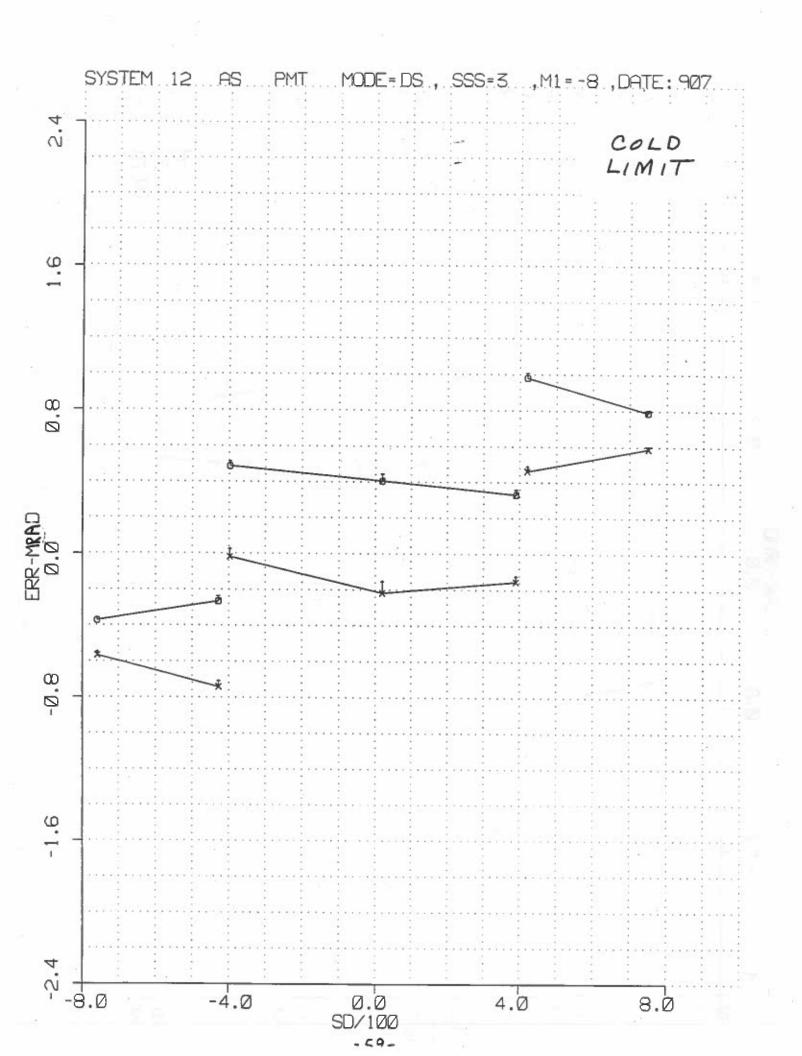


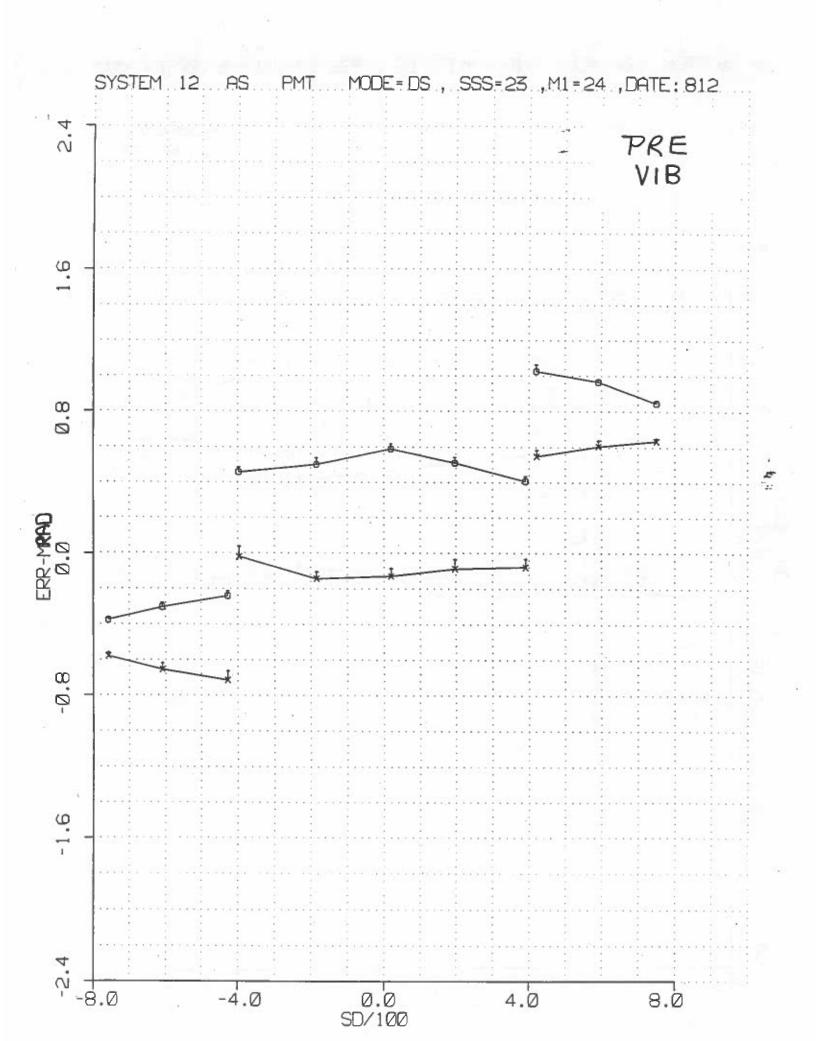


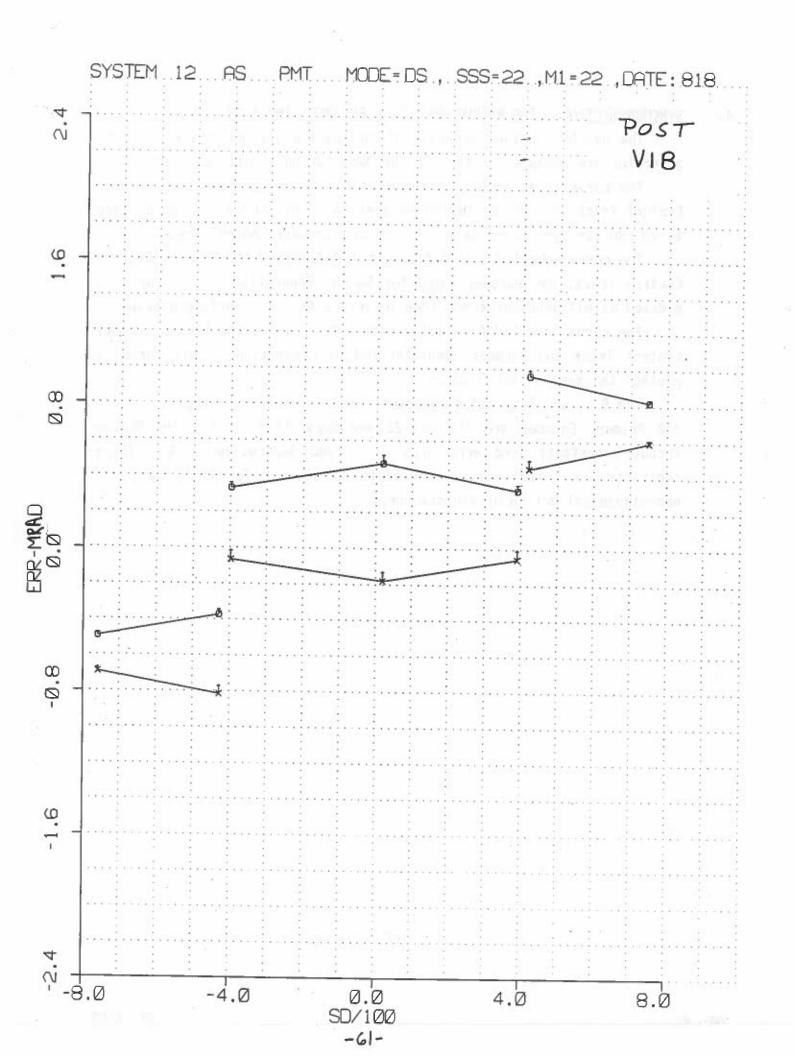












## 4. SYNCHRONIZATION USING BACKUP ENCODER & ENCODER SIMULATOR

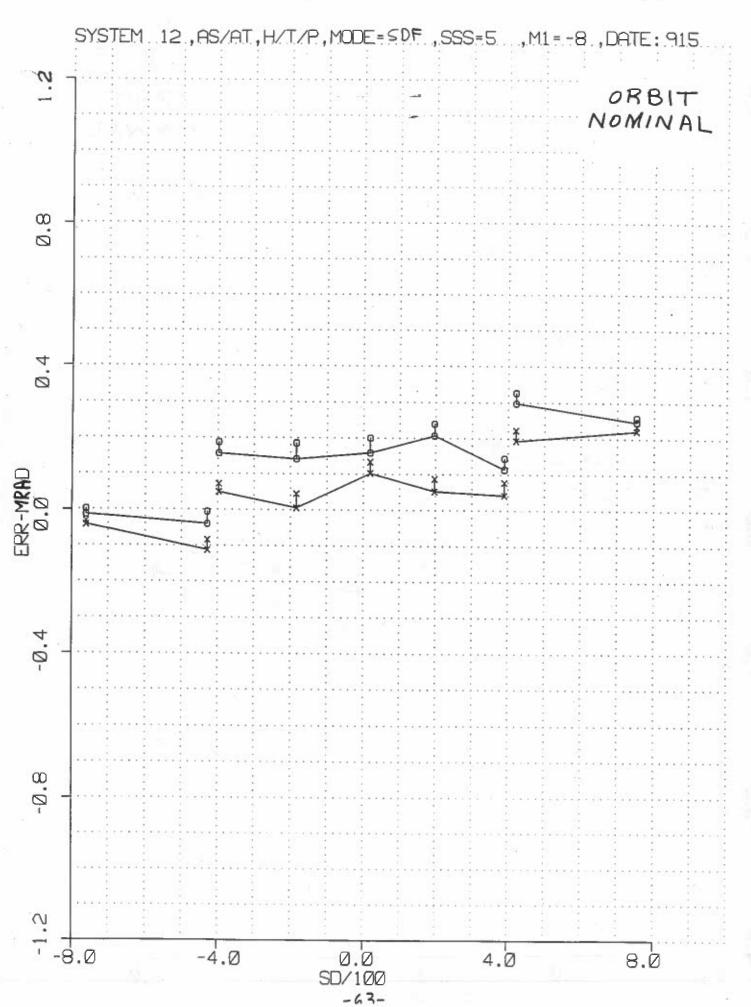
The synchronization accuracy of the backup encoder track & delphi generator are measured in the HRD SDF mode during acceptance test.

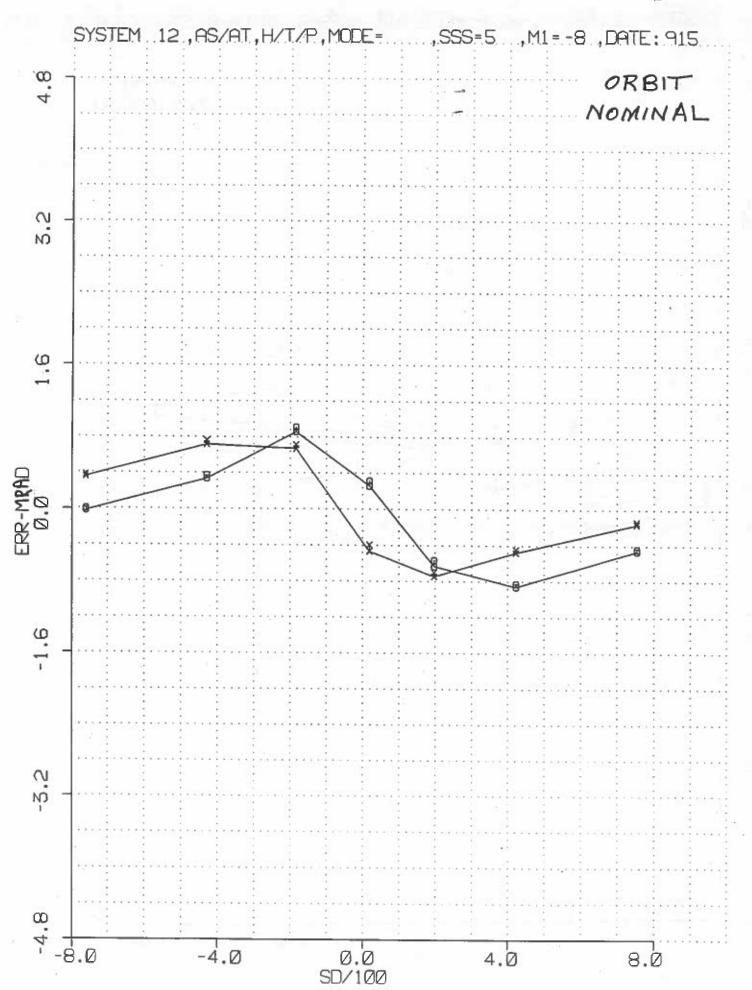
The curve labelled A is taken with I/O X, using the Backup Encoder Control Track and Encoder Delphi Generation. This curve can be compared to an HRD SDF sync curve using the Primary Encoder Control Track.

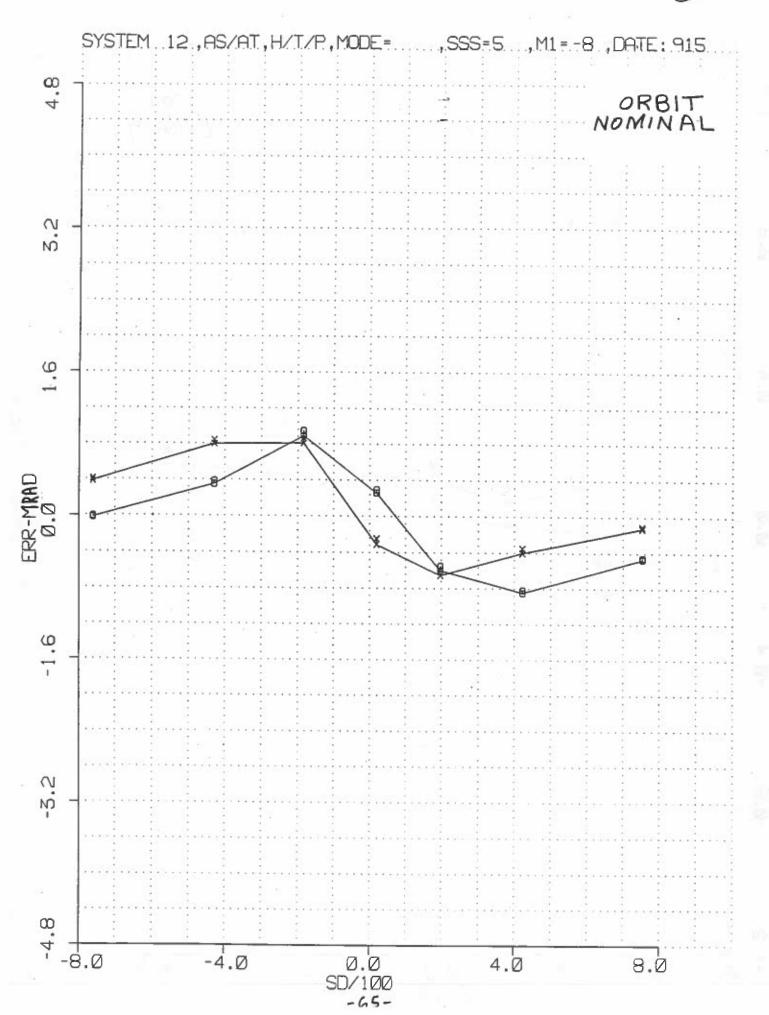
The curve labelled B is taken with I/O X, using the Primary Encoder Control Track, and encoder Simulator Delphi Generation. This curve is plotted as milliradians error from Interface Axis vs. Surface distance.

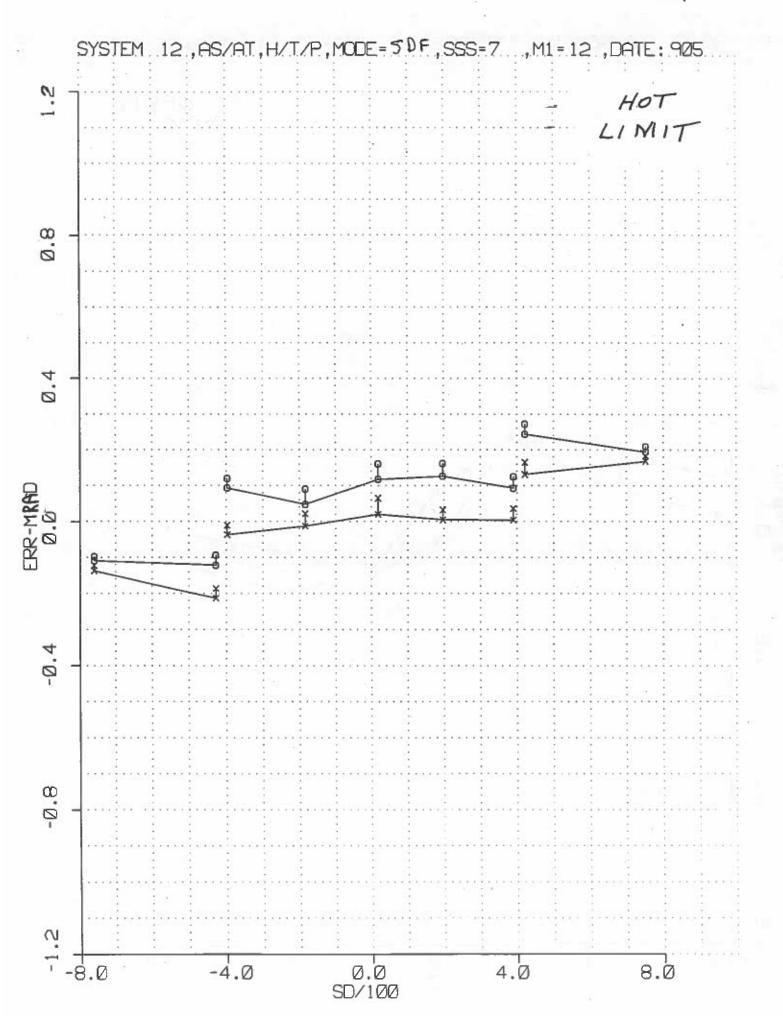
The curve labelled C is taken with I/O Y, using the Backup Encoder Control Track and Encoder Simulator Delphi Generation. This curve is plotted the same as the B curve.

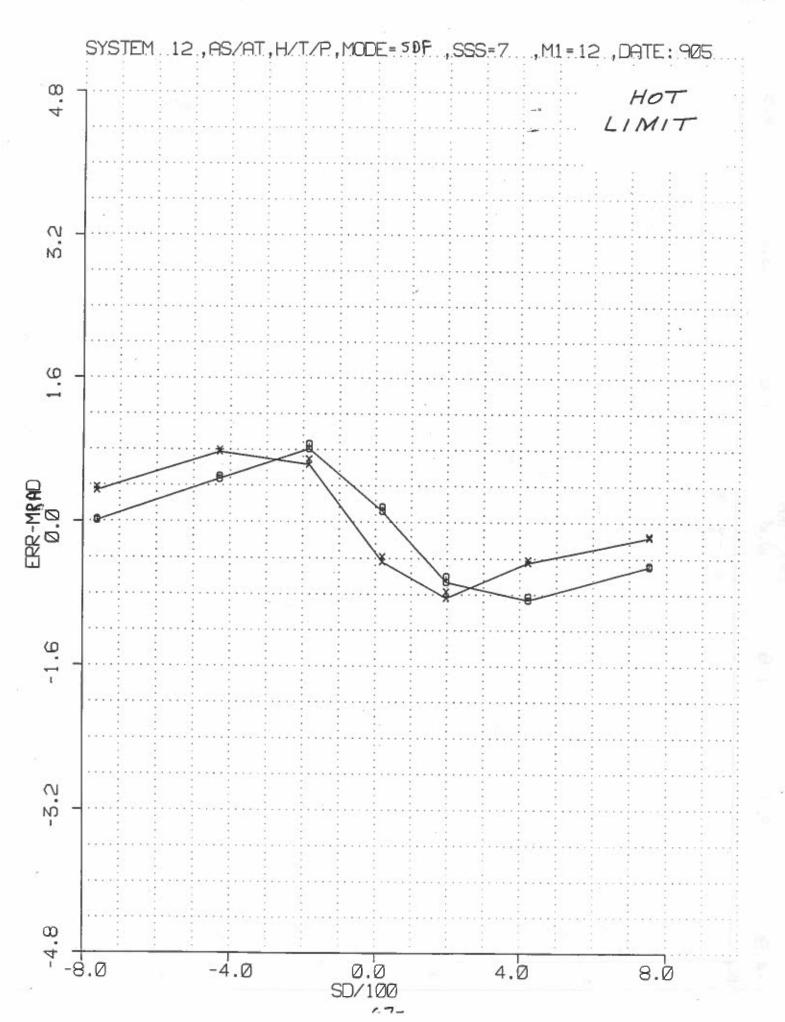
The Bias and Separation constants used for bearing retrofit for OLS #12 Primary Encoder are Bias = -22 and Separation = -7. The Backup Encoder constants used were Bias = -23 and Separation = -6. These constants are operationally adjustable to account for the effects of non-sinusoidal motion of the scanner.

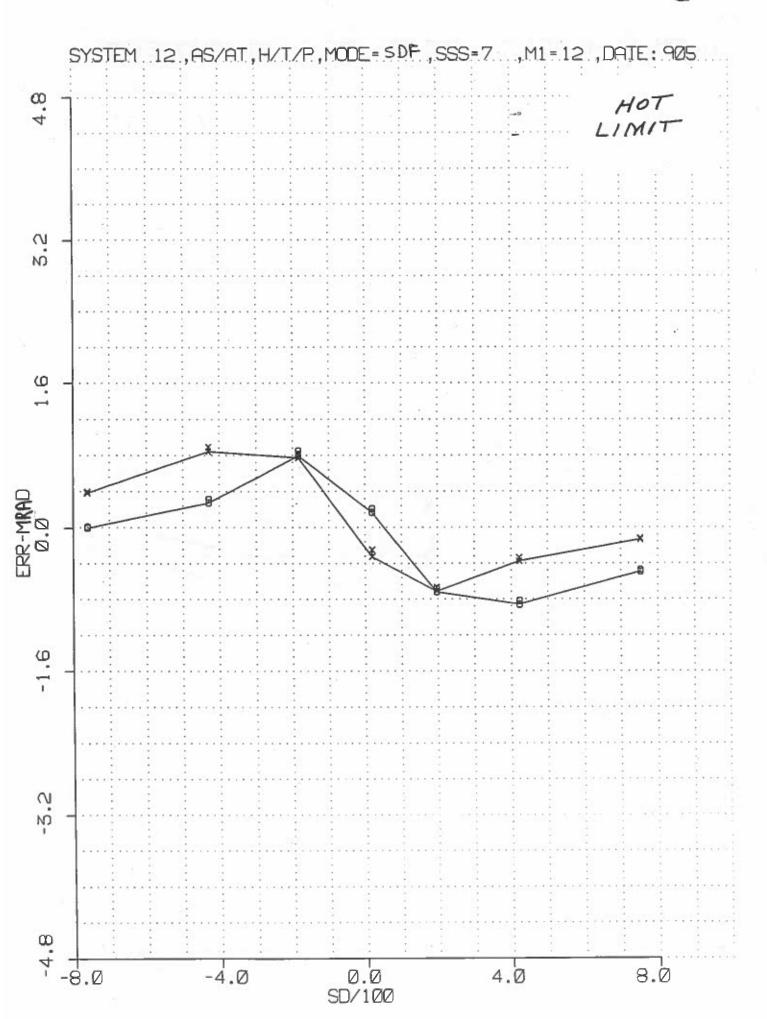


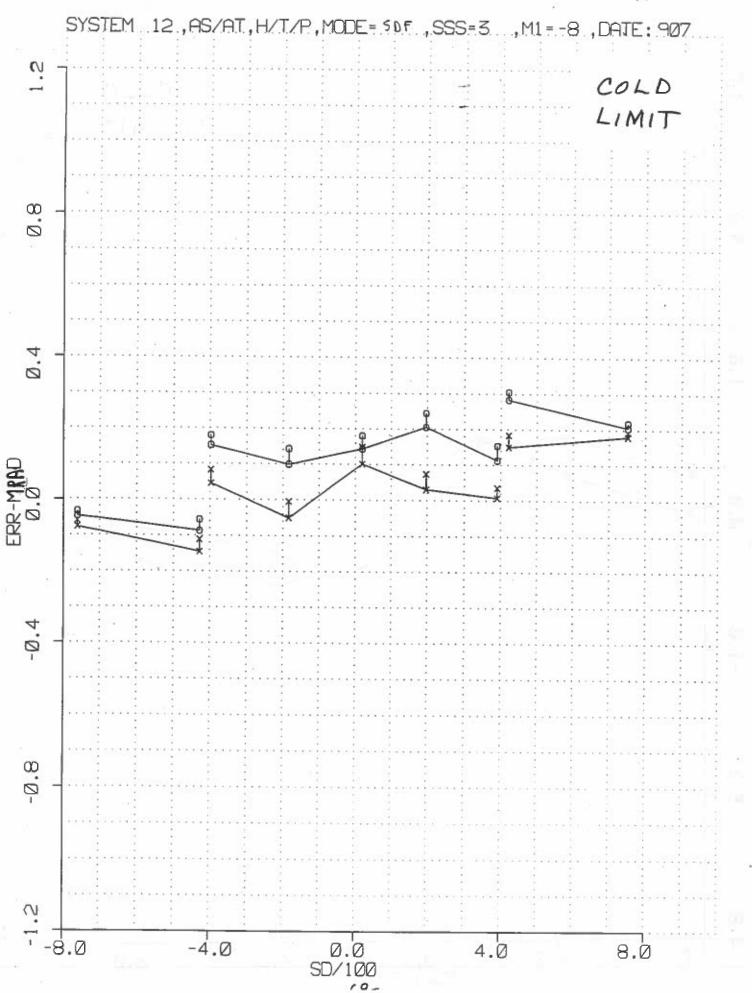


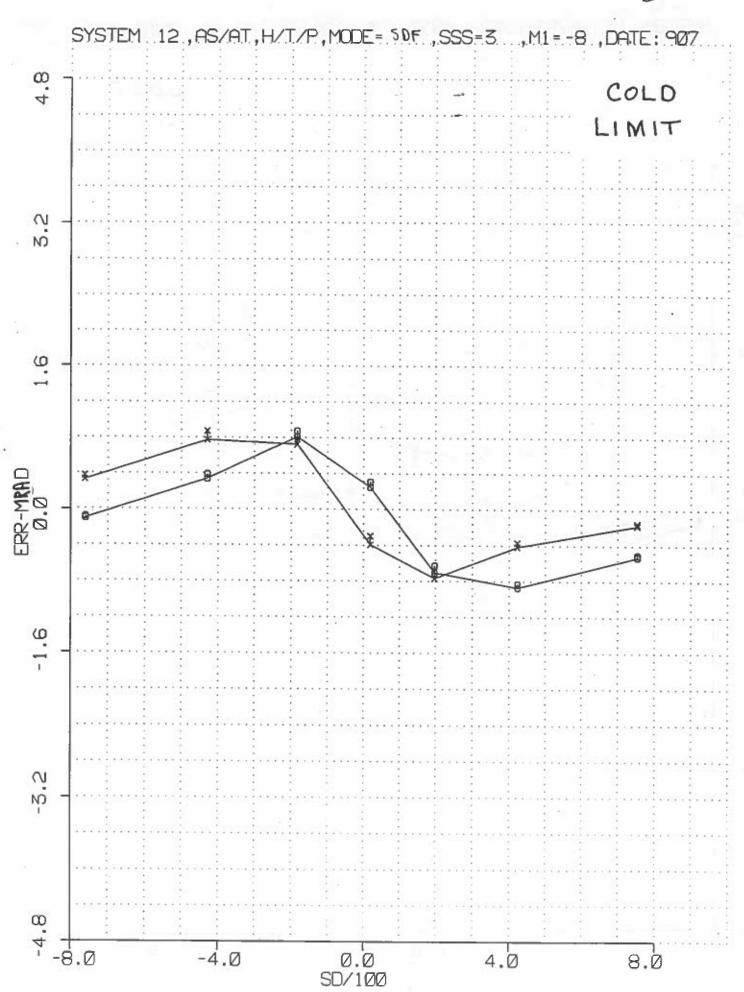


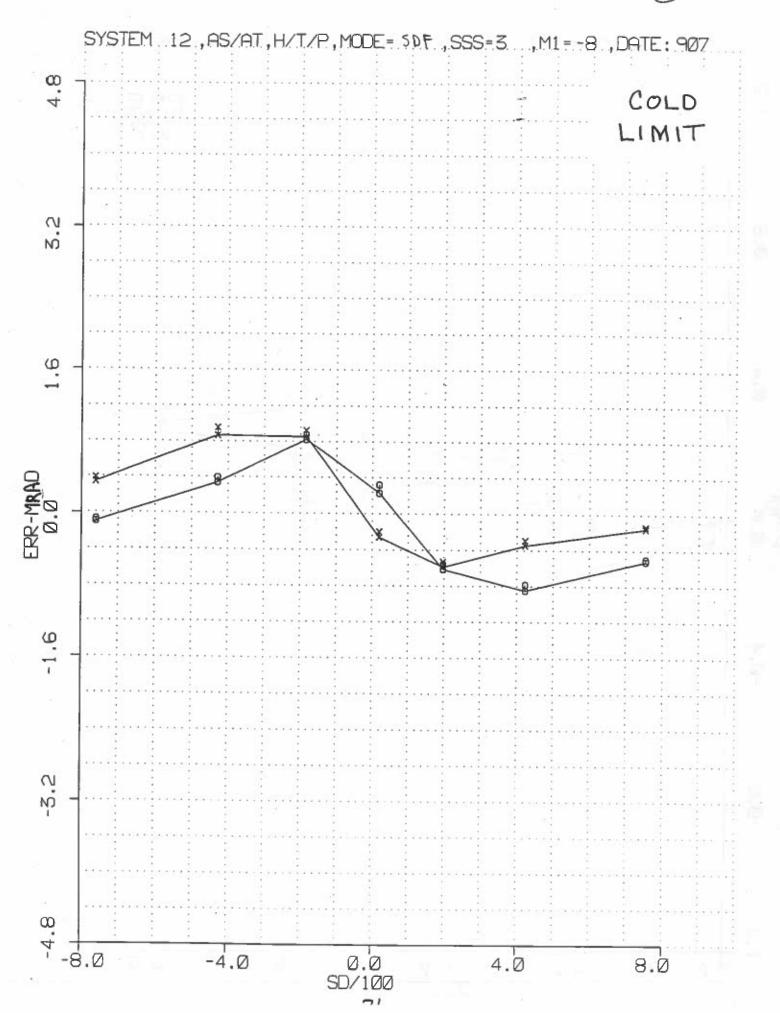


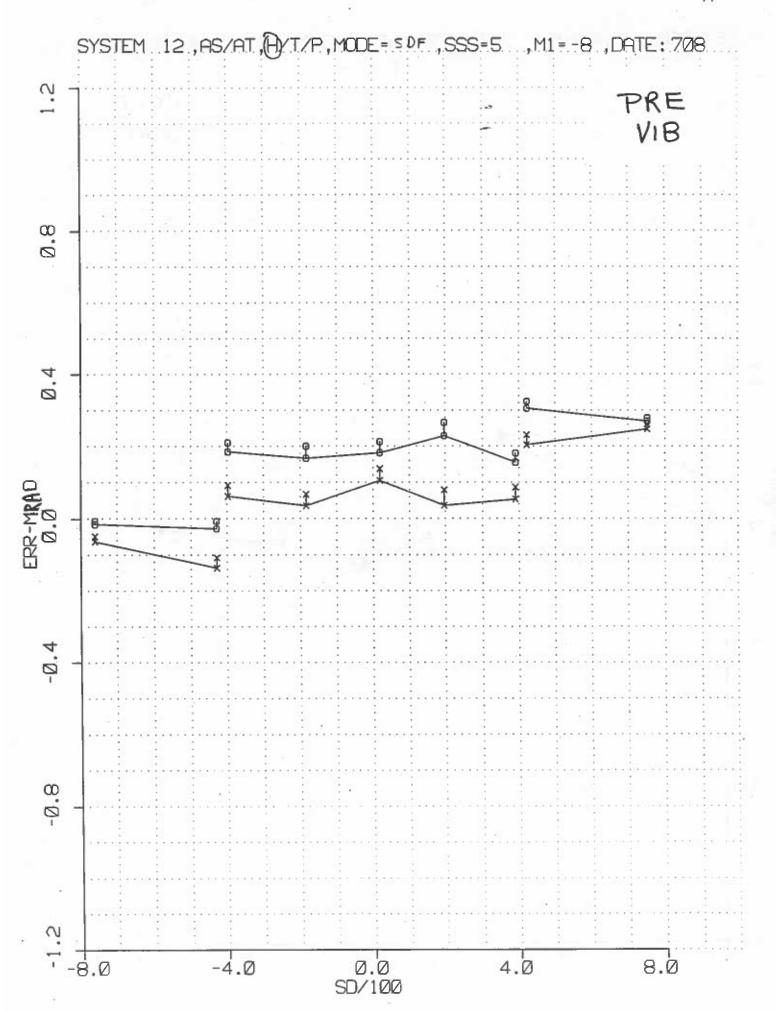


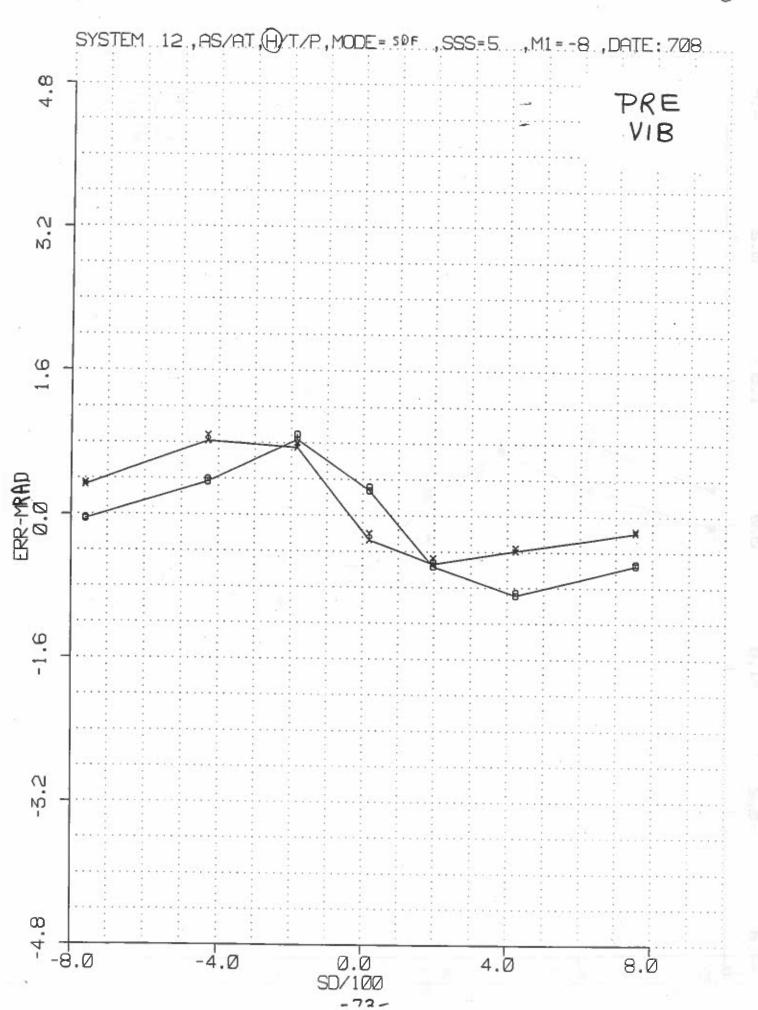


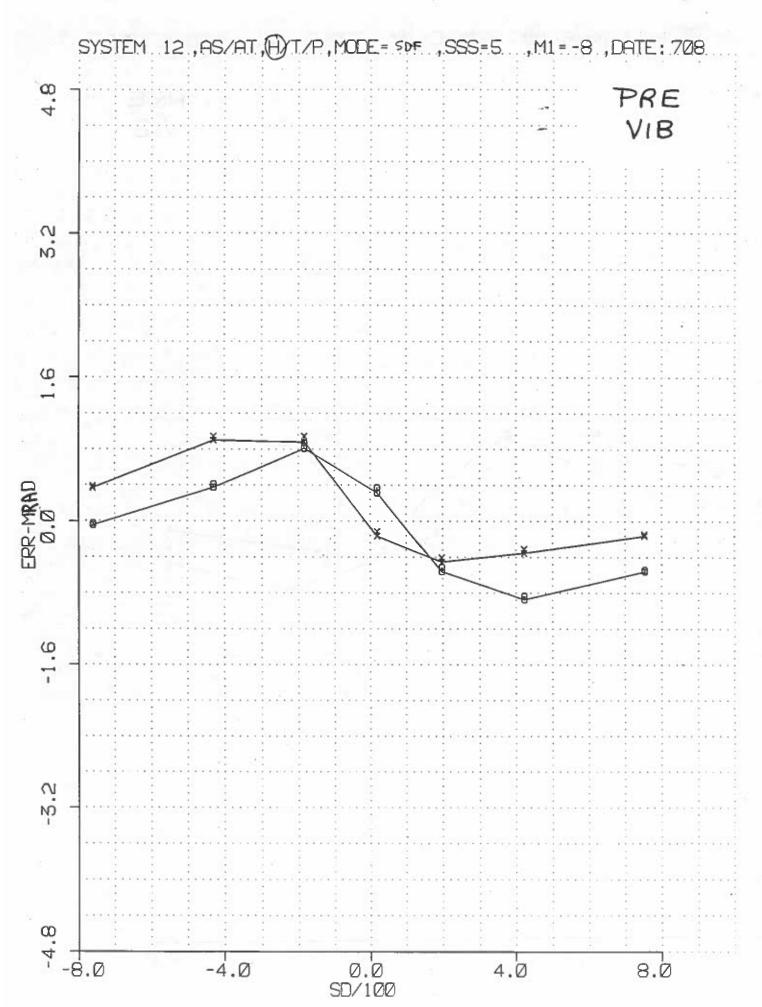


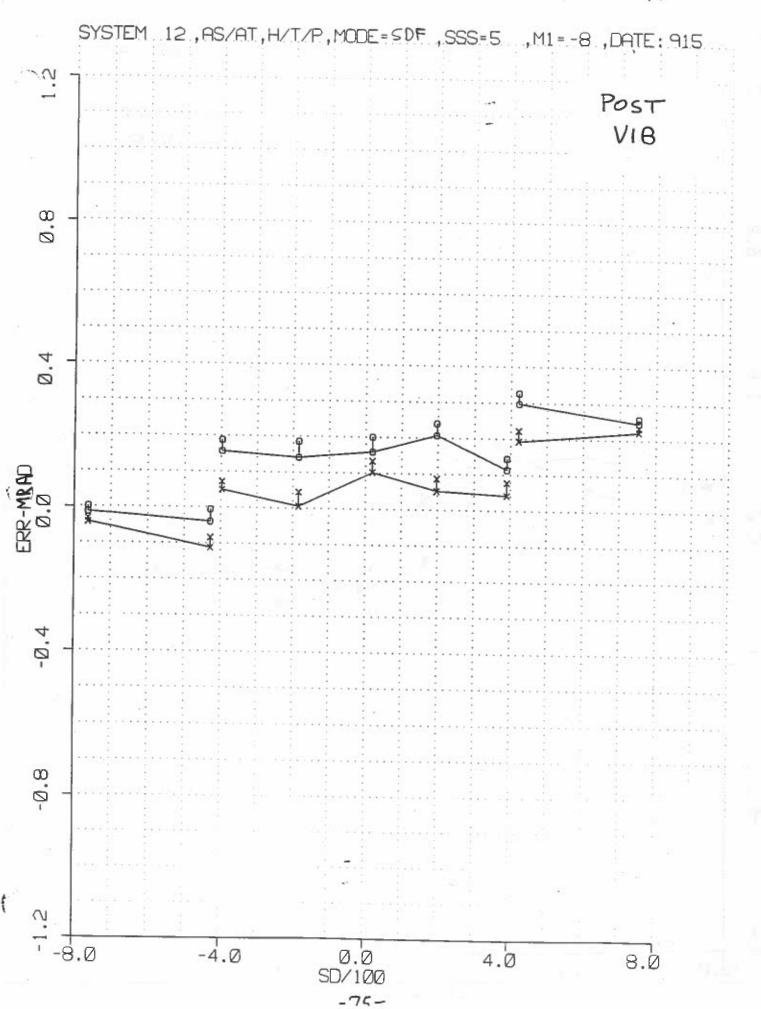


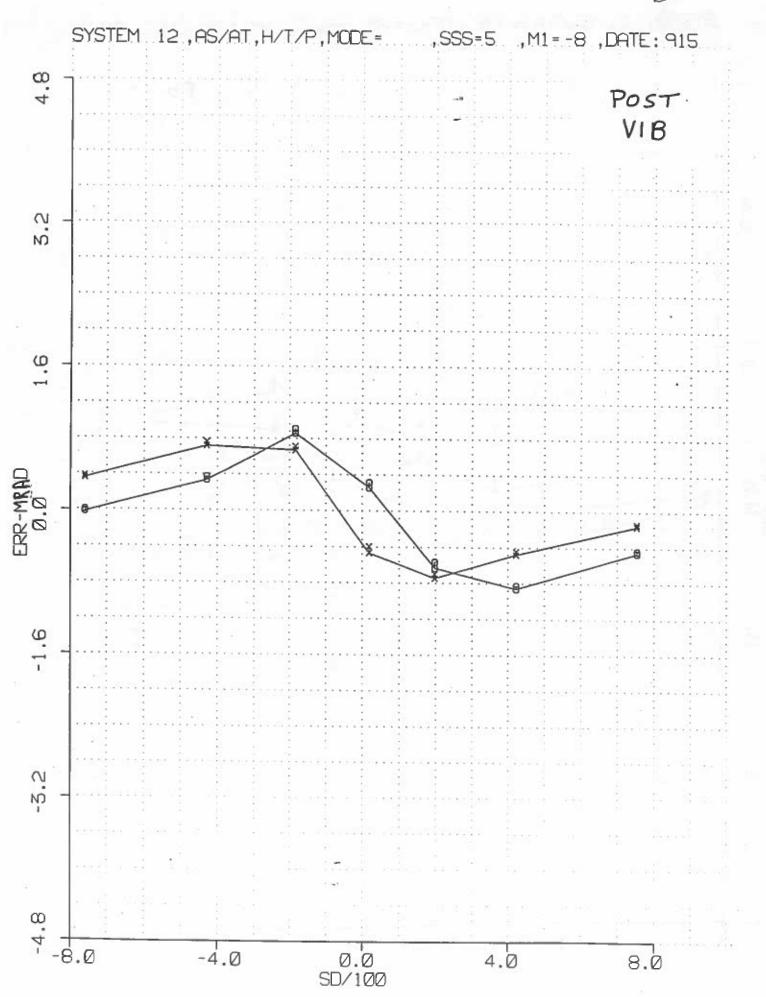


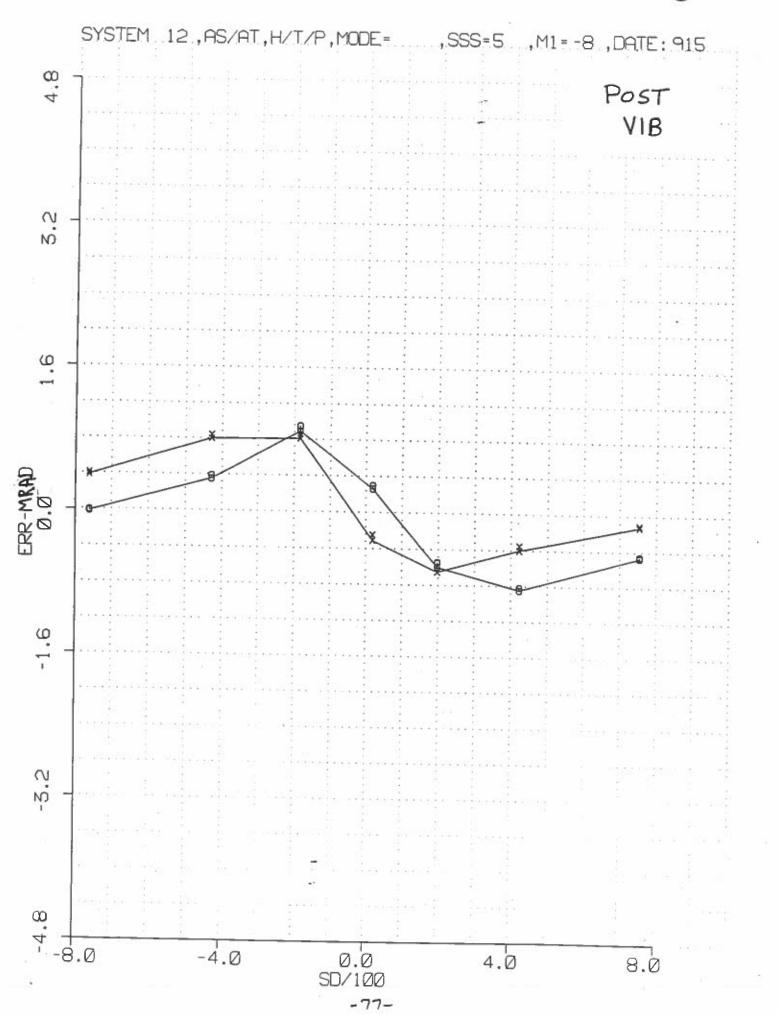












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