

## Fi2

## OLS \#12 BEARING RETROFIT

ACCEPTANCE TEST REPORT
VOLUME I OF III
SUMMARY AND SPECIFICATION REQUIREMENTS
(CDRL 006A1)

Contract F04701-90-C-0028

Prepared For<br>UNITED STATES AIR FORCE Headquarters, Space Division Los Angeles, California

Prepared By

> WESTINGHOUSE ELECTRIC CORPORATION
> Defense and Electronics Center
> Baltimore, Maryland

$$
5: 3
$$

## TABLE OF CONTENTS

PAGE
1.0 Introduction ..... 1-1
1.1 Summary of System - Specific Parameters ..... 1-2
1.2 Specification Pass-Fail Summary ..... 1-5
1.3 Summary of OLS \#12 Testing ..... 1-8
1.4 Configuration \& Serialized Assemblies ..... 1-9
1.5 Thermal Vacuum Profiles ..... 1-20
1.6 Test History Calendar ..... 1-24
2.0 Development Specification Requirements ..... 2-1
2.1 Spectra ..... 2-1
2.2 Geometric Resolution ..... 2-5
2.2.1 Fine Geometric Resolution - Infrared ..... 2-5
2.2.2 Fine Geometric Resolution - Daytime Visual ..... 2-43
2.2.3 Smooth Geometric Resolution - Infrared ..... 2-62
2.2.4 Smooth Geometric Resolution - Daytime Visuai ..... 2-75
2.2.5 Smooth Geometric Resolution - Nighttime Visual ..... 2-88
2.2.6 Data Sampling ..... 2-101
2.3 Geometric Accuracy ..... 2-102
2.4 Radiometric Accuracy ..... 2-108
2.4.1 T Channel Radiometric Accuracy ..... 2-108
2.4.2 Daytime Radiometric Accuracy ..... 2-165
2.4.3 Nighttime Radiometric Accuracy. ..... 2-170
2.4.4 Gain Control Accuracy ..... 2-172
2.4.5 Gain Control Adjustability ..... 2-173
2.4.6 A/D Conversion \& Algorithms ..... 2-175
2.5 Radiometric Resolution ..... 2-176

## TABLE OF CONTENTS (Cont'd.)

PAGE
2.6 Noise ..... 2-178
2.6.1 T Channel Noise ..... 2-178
2.6.2 L Channel Noise - Day ..... 2-181
2.6.3 L Channe] Noise - Night ..... 2-183
2.6.4 Dark Current ..... 2-185
2.6.5 Stability ..... 2-186
2.6.6 Along-Track Noise Integration ..... 2-187
2.6.7 G1are Suppression ..... 2-188
2.7 Survivability ..... 2-189
2.8 Scan Angle ..... 2-190
2.9 Data Collection Rate ..... 2-191
2.10 Power ..... 2-192
2.11 Mass ..... 2-194
2.11.1 Total Mass ..... 2-194
2.11.2 Component Mass ..... 2-197
2.12 Cooler Transient Margin ..... 2-198
2.13 Design Features ..... 2-199
2.14 Redundant and Fallback Subsystems ..... 2-200
2.15 Environment ..... 2-202
2.16 Electromagnetic Compatibility ..... 2-203
3.0 INTERFACE SPECIFICATION REQUIREMENTS ..... 3-1
3.1 SSS Alignment Axes ..... 3-2
APPENDIX A - BVS 2579 - "Bearing Retrofit and Retest Plan for OLS 12 thru 16" ..... A-1
APPENDIX B - BVS 2600 - "RDS Rework and Retest Procedure for OLS 12, 13, 14, 15 and 16" ..... B-1

### 1.0 INTRODUCTION

The OLS \#12 Acceptance Test Report contains the technical data pertinent to the OLS \#12 AVE system. This document is intended to present the Acceptance Test data in terms of the requirements of the Prime Item Development Specification (DMSS-OLS-300) and Interface Specification (IS-YD-810A) for testing associated with bearing retrofit (BVS 2579). A copy of the signed-off BVS 2579 is included in this report as Appendix A. During this same period, BVS 2600 "RDS Rework and Retest Procedure" was performed. Testing for BVS 2600 was done at the functional level only. Therefore, no test results are included as part of this ATR. However, a copy of the signed-off BVS 2600 is included in this report as Appendix B. A special test of $T$ channe] stability was also performed during retrofit. Results of this testing are compiled in BVS 2698, "OLS 12 Stability Testing".

Test results and data have been reviewed and verified by Westinghouse Electric Corporation and USAF representatives. System performance data, test histories, data summaries and system analyses are included in this report. In addition, a complete set of system log books are on file at the contractor's facility and are available for review. The Test History is in log books K41477-- and K40502--.

It is intended that this report provide a complete summary of OLS 12 performance relative to all requirements. Therefore, data showing performance for requirements not verified as part of bearing retrofit are also provided. When data from previous tests are provided it will be so noted.

This Acceptance Test Report consists of 3 volumes as follows:
BVS 2691 OLS \#12 Summary and Specification Requirements
BVS 2692 OLS \#12 Acceptance Vibration Report
BVS 2693 OLS \#12 Alignment \& Synchronization Curves 1-1

```
1.1 Summary of System - Specific Parameters
OLS software Program = OLSPO2J.FS
Gain Constants and Sensor Switch Points
P(0) = 9.375 dB
P(1) = 51.75 dB
P(2) = 0 dB
P(3) = 29.75 dB
S(1) = 59.875 dB
S(2) = 22dB
S(3) = 33.75 dB
(These may change during Early Orbit Calibration.)
PMT HV EST \((\) A532 \()=3.634\) volts \(\pm .250 \mathrm{~V}\)
Cone Cooler S/N 024 with T detector S/N K-5
T Cold Patch EST (A549) curve - see Table next page.
T Cold Patch EST Voltage \(=2.211 \mathrm{~V} \pm .200 \mathrm{~V}\)
TGAIN Left \(=4\)
Right \(=4\) Both \(=4\)
TLEVEL vs M1 temperature range - see second page following for table
VDGA constant for PMTCAL \(=(0440)_{8}\)
Encoder Simulator Bias Constant \(=\) Prim -22 Redun -23
Encoder Simulator Separation Constant \(=\) Prime -7 Redund -6
```


## T COLD PATCH TEMP VS EST VOLTS <br> CONE COOLER S/N 024 <br> T DETECTOR S/N K-5

| I (deq k$)$ | EST (Volts) |
| :---: | :---: |
| 95 | 5.655 |
| 96 | 5.248 |
| 97 | 4.874 |
| 98 | 4.529 |
| 99 | 4.212 |
| 100 | 3.920 |
| 101 | 3.651 |
| 102 | 3.403 |
| 103 | 3.174 |
| 104 | 2.963 |
| 105 | 2.768 |
| 106 | 2.588 |
| 107 | 2.422 |
| 108 | 2.268 |
| 109 | 2.125 |
| 110 | 1.993 |
| 111 | 1.871 |
| 112 | 1.757 |
| 113 | 1.651 |
| 114 | 1.553 |
| 115 | 1.462 |
| 116 | 1.377 |
| 117 | 1.298 |
| 118 | 1.225 |
| 119 | 1.156 |
| 120 | 1.092 |
| 121 | 1.032 |
| 122 | 0.976 |
| 123 | 0.924 |
| 124 | 0.875 |
| 125 | 0.829 |

TLEVEL $\begin{gathered}\text { OLS \#12 } \\ \text { T DETECTOR } \mathrm{S} / \mathrm{N} \mathrm{K-5}\end{gathered}$

| TL | M1 TEMP $^{\circ} \mathrm{C}$ ) |  |
| :--- | ---: | ---: |
| 1111 | $-26.019^{\circ}$ | to |
| 1110 | $-21.069^{\circ}$ | $-21.069^{\circ}$ |
| 1101 | $-16.120^{\circ}$ | $-11.170^{\circ}$ |
| 1100 | $-11.170^{\circ}$ | $-6.221^{\circ}$ |
| 1011 | $-6.221^{\circ}$ | $-1.271^{\circ}$ |
| 1010 | $-1.271^{\circ}$ | $3.678^{\circ}$ |
| 1001 | $3.678^{\circ}$ | $8.628^{\circ}$ |
| 1000 | $8.628^{\circ}$ | $13.577^{\circ}$ |
| 0111 | $13.577^{\circ}$ | $18.527^{\circ}$ |
| 0110 | $18.527^{\circ}$ | $23.476^{\circ}$ |
| 0101 | $23.476^{\circ}$ | $28.426^{\circ}$ |
| 0100 | $28.426^{\circ}$ | $33.375^{\circ}$ |
| 0011 | $33.375^{\circ}$ | $38.325^{\circ}$ |
| 0010 | $38.325^{\circ}$ | $43.274^{\circ}$ |
| 0001 | $43.274^{\circ}$ | $48.224^{\circ}$ |
| 0000 | $48.224^{\circ}$ | $53.173^{\circ}$ |

TLEVEL command changes should be uplinked to the OLS as a function of M1 temperature to maximize $T$ Channel output accuracy.

### 1.2 Specification Pass-Fail Summary

The following sections of this Acceptance Test Report contain the test results as they pertain to the Development Specification requirements. Each Test Report paragraph heading is followed by the corresponding Segment Spec paragraph number in parentheses.

The table on the following page summarizes the OLS \#12 pass-fail status vs. Development Spec. paragraph number.

| DEVELOPMENT SPEC. PARAGRAPH NUMBER |  | PASS | FAIL |
| :---: | :---: | :---: | :---: |
| 3.2.1.1.1.1 | Infrared Spectrum | x |  |
| 3.2.1.1.1.2 | Vis-Day Spectrum | $x$ |  |
| 3.2.1.1.1.3 | Vis-Night Spectrum |  | $x$ |
| 3.2.1.1.2.1 | Fine Geometric Resolution - HRD | x |  |
| 3.2.1.1.2.1 | Fine Geometric Resolution - T | $x$ |  |
| 3.2.1.1.2.2 | Smooth Geometric Resolution - HRD | x |  |
| 3.2.1.1.2.2 | Smooth Geometric Resolution - T | x |  |
| 3.2.1.1.2.2 | Smooth Geometric Resolution - PMT | $x$ |  |
| 3.2.1.1.2.3 | Data Sampling | $x$ |  |
| 3.2.1.1.3.1 | Along Track Geometric Accuracy | X |  |
| 3.2.1.1.3.2/3 | Along Scan Geometric Accuracy | $x$ |  |
| 3.2.1.1.4.1.a | T Channel Radiometric Accuracy Repeatability | X |  |
| 3.2.1.1.4.1b | T Channel Radiometric Accuracy - Stability | x |  |
| 3.2.1.1.4.1c | T Channel Radiometric Accuracy - Fixed | $x$ |  |
| 3.2.1.1.4.2 | Daytime Radiometric Accuracy | x |  |
| 3.2.1.1.4.3 | Nighttime Radiometric Accuracy | $x$ |  |
| 3.2.1.1.4.5.1 | Terminator Location | x |  |
| 3.2.1.1.4.5.2 | Gain Change Rate | $x$ |  |
| 3.2.1.1.4.5.3 | Maximum Gain Settings | X |  |
| 3.2.1.1.4.5.4 | Commandable T-Channel Gain | x |  |
| 3.2.1.1.4.5.5 | Commandable T-Channel Level | x |  |
| 3.2.1.1.4.6.2/3 | A/D Conversions \& Algorithms | x |  |
| 3.2.1.1.5 | Radiometric Resolution | x |  |
| 3.2.1.1.6.1 | T Channel Noise | $x$ |  |
| 3.2.1.1.6.2 | L Channel Noise (Day) | x |  |
| 3.2.1.1.6.3 | L Channel Noise (Night) | $x$ |  |


| DEVELOPMENT SPEC. PARAGRAPH NUMBER |  | PASS | FAIL |
| :---: | :---: | :---: | :---: |
| 3.2.1.1.6.4 | Dark Current | $x$ |  |
| 3.2.1.1.6.5 | Stability | $x$ |  |
| 3.2.1.1.6.6 | Along-Track Noise Integration | x |  |
| 3.2.1.1.6.7 | Glare Suppression | $x$ |  |
| 3.2.1.1.7 | Survivability | $x$ |  |
| 3.2.1.1.8 | Scan Angle | x |  |
| 3.2.1.1.9 | Data Collection Rate | $x$ |  |
| 3.2.1.2 | Data Management | $x$ |  |
| 3.2.1.3.1 | $28 V$ Power | $x$ |  |
| 3.2.1.3.2 | 5V Power | $x$ |  |
| 3.2.2.1 | Total Mass | $x$ |  |
| 3.2.2.2 | Component Mass | $x$ |  |
| 3.2.2.3 | Cable Harness Mass | $x$ |  |
| 3.2.2.4 | Dimensional Limits | $x$ |  |
| 3.3 | Design Features | $x$ |  |
| 4.1/2 | Environment | x |  |
|  | Shipping \& Storage | $x$ |  |
| INTERFACE SPE | ARAGRAPH NUMBER |  |  |
| 3.1.3 | Al ignment | $x$ |  |

### 1.3 Summary of OLS \#12 Testing

03-25-91 Began RDS testing per BVS 2600
04-26-91 OSU(x) and SPS $(x)$ vibration
05-20-91 Began Bearing Retrofit testing per BVS 2579
05-23-91 SSS( $x, y, z$ ) vibration
05-29-91 Testing stopped to use OLS 12 units with BTM SSS for special BTM T channel testing per BVS 2654

06-09-91 PSU(x) vibration per BVS 2657 (repair of broken wire)
06-26-91 Resumed Bearing Retrofit testing
07-01-91 Began Thermal Vac testing
07-20-91 Break vacuum due to PMT failure
08-06-91 OLS 12 to Blue Room
08-07-91 Replaced PMT with PMT from OLS14
08-09-91 Replaced EST/LMD with unit from OLS14
08-13-91 SSS sine and random vibration
08-16-91 Thermal Vacuum ambient testing
08-19-91 Restarted Thermal Vacuum testing
09-19-91 Thermal Vacuum testing complete
10-27-91 Final Blue Room testing complete

### 1.4 Configuration and Serialized Assemblies <br> The configuration listing on the following pages includes the current configuration of the OLS \#12 as of 12-03-86.

| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| Key Drawing | 536R500G01 | F | 5007 |
| SSS Assembly | 640R800G08 | AM | 5007 |
| OSC Assy | 623R765G06 | AC | 5007 |
| HRD Assy | 623R754G04 | AB | 0006 |
| PWR Bd | 623R758G03 | R | 0006 |
| Pre Amp Bd | 623R506G03 | U | 0006 |
| T-Chan | 633R049G04 | $R$ | 0006 |
| T-Chan Bd | 633R178G03 | AD | 0006 |
| Modute | 623R727G01 | B | 5009 |
| Module | 623R727G01 | B | 5010 |
| VDGA/Lin Log | 644R150G03 | F | 5007 |
| Lin Log | 644R127G03 | P | 5007 |
| VDGA | 644R152G03 | P | 5007 |
| VDGA | 644R153G03 | $N$ | 5007 |
| Enc. OPT | 688R705H01 | C | 009 |
| PMT | 644R909G04 | P | 0007 |
| EMR Bd | 644R905G03 | D | 0007 |
| Switch Bd | 644R903G04 | M | 0007 |
| Doubler Bd | 644R907G02 | F | 0007 |
| Regulator Bd | 644R807G03 | H | 0006 |
| Pre Amp Bd | 644R935G03 | J | 5008 |
| HRD Post Amp | 644R220G04 | G | 5007 |
| Post Amp Bd | 644R228G04 | $A B$ | 5007 |
| EST/LMD | 644R219G03 | D | 0007 |
| EST/LMD Bd | 758R142G02 | E | 0007 |
| Heater Cont | 633R053G09 | J | 5015 |
| Elect Assy | 633R052G03 | V | 5015 |


| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| Heat Cont | 633R053G10 | J | 5016 |
| Elect Assy | 633R052G03 | $v$ | 5016 |
| Heat Cont | 633R053G11 | J | 5017 |
| Elect Assy | 633R052G03 | $v$ | 5017 |
| Heater Cont | 633R053G12 | J | 5018 |
| Elect Assy | 633R052G03 | V | 5018 |
| Rel Mech I | 640R701G02 | F | 5007 |
| Rel Mech II | 640R753G02 | H | 5007 |
| Rel Mech III | 640R381G02 | H | 5007 |
| T-Cl amp | 623R821G01 | G | - |
| T-Cal | 623R920G01 | B | - |
| Aux Encd | 640R846G04 | G | 5007 |
| Bd Assy | 640R825G04 | F | 5006 |
| Bd Assy | 640R844G04 | J | 5006 |
| Wire Dia | 682R239G03 | K | - |
| Wire Tab | 318R708 | B | - |
| Wire Tab | 315R386 | C | - |
| Wire Tab | 318R709 | (-) | - |
| Motor Assy | 623R894G01 | B | 73L0993 |
| IMC/M3 | 623R858G02 | D | 5007 |
| Cover, Cooler | 640R320G01 | (-) | 5007 |
| Cone Cooler | 9RA5216H01 | J | 024 |
| ENPA | 682R215G03 | M | 5007 |
| Al Bd | 682R167G03 | H | 5008 |
| A2 Bd | 682RI10G03 | T | 5007 |
| A3 Bd | 682RI12G03 | P | 5007 |
| Aux Encd B/U | 682R300G03 | C | 5007 |

[^0]| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| Al Bd | 682R149G03 | E | 5007 |
| A2 Bd | 682R151G03 | E | 5007 |
| BBI | KG43 |  | 026 |
| BB2 | KG43 |  | 027 |
| BB3 | KG43 |  | 028 |
| Ther. Blk. Kit | 661R564G03 | J | 5007 |
| GSSA/DOC | 640R790G03 | H | 5007 |
| GSSB | 633R906G01 | A | 5007 |
| PR1 | 688R461H01 | E | 040 |
| PR2 | 688R461H01 | E | 042 |
| PR3 | 688R461H01 | E | 043 |
| PR4 | 688R461H01 | E | 041 |
| Cable Assy | 9RA5255H09 | T | 006 |
| Cable Assy | 9RA5255H02 | T | 006 |
| Cable Assy | 9RA5255H03 | T | 007 |
| Cable Assy | 9RA5255H04 | T | 006 |
| Cable Assy | 9RA5255H10 | T | 003A |
| Cable Assy | 9RA5255H07 | T | 006 |
| Cable Assy | 9RA5255H06 | T | 501 |
| Cable Assy | 9RA8118G01 | F | - |
| Coax Assy | 644R327G01 | B | - |
| Coax Assy | 644R327G02 | B | - |
| Coax Assy | 644R327G03 | B | - |
| Coax Assy | 644R328G01 | C | - |
| Coax Assy | 644R328G02 | C | - |
| Coax Assy | 644R328G03 | C | - |
| Coax Assy | 644R328G04 | C | - |

$$
1-12
$$

| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| Coax Assy | 644R328G05 | C | - |
| Coax Cable | 644R328G06 | C | - |
| Coax Assy | 644R329G01 | $c$ | - |
| Coax Assy | 644R329G02 | c | - |
| Coax Assy | 644R329G03 | c | - |
| Coax Assy | 644R329G04 | c | - |
| Coax Assy | 644R329G05 | c | - |
| Coax Assy | 644R329G06 | C | - |
| Coax Assy | 644R329G07 | C | - |
| Coax Assy | 644R329G08 | C | - |
| SPS | 651R390G01 | AC | 5007 |
| Matrix | $651 R 342 G 03$ | AV | - |
| R/B | 644R665G04 | AE | 5012 |
| Matrix | 644R081G03 | L | - |
| Al Bd | 640R618G03 | F | 5014 |
| A2 Bd | 640R518G02 | P | 5013 |
| A3 Bd | 640R520G03 | P | 5013 |
| R/B | 644R665G04 | AE | 5013 |
| Matrix | 644R081G03 | L | - |
| Al Bd | 640R618G03 | F | 5014 |
| A2 Bd | 640R518G03 | P | 5014 |
| A3 Bd | 640R520G03 | P | 5014 |
| CU 1 | 640R612G02 | $J$ | 5013 |
| CU 1 | 640R612G02 | $\checkmark$ | 5014 |
| CU2 | 640R614G02 | J | 5012 |
| CU 2 | 640R614G02 | $K$ | 5013 |
| AU 1 | 640R608G02 | D | 5013 |


| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| AU 1 | 640R608G02 | D | 5014 |
| AU 2 | 640R610G02 | D | 5012 |
| AU 2 | 640R610G02 | D | 5013 |
| MCIX | 640R560G03 | L | 5014 |
| MC1X | 640R560G03 | L | 5015 |
| MC2X | 640R562G03 | U | 5014 |
| MC2X | 640R562G03 | U | 5015 |
| ROM | 640R530G03 | T | 5012 |
| ROM | 640R530G03 | T | 5013 |
| Core | 644R910H03 | K | 013 |
| Core | 644R910H03 | K | 014 |
| SDS2 | 640R442G03 | T | 5012 |
| SDS2 | 640R442G03 | N | 5013 |
| SDS3 | 640R444G03 | $N$ | 5012 |
| SDS3 | 640R444603 | N | 5013 |
| SDS4 | 640R446G03 | T | 5012 |
| SDS 4 | 640R446G03 | T | 5013 |
| SDS5 | 640R498G03 | p | 5012 |
| SDS5 | 640R498G03 | P | 5013 |
| CLSD | 640R458G03 | AD | 5012 |
| CLSD | 640R458G03 | AD | 5013 |
| SDSIX | 640R660G04 | AP | 5012 |
| SDSIX | 640R660G04 | AP | 5013 |
| FC-1 | 640R450G03 | AA | 5012 |
| FC-1 | 640R450G03 | AA | 5013 |
| FC-2 | 640R454G03 | V | 5012 |
| FC-2 | 640R454G03 | v | 5013 |


| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| FC-3 | 640R456G03 | Y | 5012 |
| FC-3 | 640R456G03 | $Y$ | 5013 |
| SDF-1 | 640R474G03 | AH | 5012 |
| SDF-1 | 640R474G03 | AH | 5013 |
| SDF-2 | 640R476G03 | AH | 5012 |
| SDF-2 | 640R476G03 | AH | 5013 |
| SDF-3X | 640R540G03 | H | 5012 |
| SDF-3X | 640R540G03 | H | 5013 |
| SDF-4X | 640R542G03 | H | 5012 |
| SDF-4X | 640R542G03 | H | 5013 |
| SDF-5X | 640R544G03 | $N$ | 5012 |
| SDF-5X | 640R544G03 | $N$ | 5013 |
| SDS-6 | 640R538G03 | U | 5012 |
| SDS-6 | 640R538G03 | U | 5013 |
| SDS-7 | 640R546G03 | $p$ | 5012 |
| SDS-7 | 640R546G03 | P | 5013 |
| 4B | 640R412G03 | P | 5012 |
| 4 B | 640R412G03 | P | 5013 |
| 7A | 640R414G03 | AB | 5012 |
| 7A | 640R414G03 | AB | 5013 |
| 7 B | 640R416G04 | AR | 5012 |
| 7 B | 640R416G04 | AR | 5013 |
| 1A | 640R400G03 | AK | 5014 |
| 1 A | 640R400G03 | AK | 5015 |
| 1B | 640R402G03 | $A D$ | 5012 |
| 1B | 640R402G03 | AD | 5013 |
| FBC | 640R448G03 | $N$ | 5012 |


| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| FBC | 640R448G03 | $N$ | 5013 |
| RAM | 640R558G03 | L | 5025 |
| RAM | 640R558G03 | L | 5026 |
| RAM | 640R558G03 | L | 5027 |
| RAM | 640R558G03 | L | 5028 |
| 2A | 640R488G03 | $Y$ | 5012 |
| 2 A | 640R488G03 | $Y$ | 5013 |
| 2B | 640R410G03 | W | 5012 |
| 2B | 640R410G03 | W | 5013 |
| 3 A | 640R404G03 | $Y$ | 5012 |
| 3 A | 640R404G03 | $Y$ | 5013 |
| 10x | 640R572G03 | $J$ | 5012 |
| 10X | 640R572G03 | J | 5013 |
| CLCL | 640R406G03 | AD | 5012 |
| CLCL | 640R406G03 | AD | 5013 |
| WF-1X | 640R566G03 | P | 5012 |
| WF-1X | 640R566G03 | P | 5013 |
| WF-2 | 640R432G03 | $Y$ | 5012 |
| WF-2 | 640R432G03 | $Y$ | 5013 |
| WF-3 | 640R622G02 | E | 5012 |
| WF-3 | 640R622G02 | $E$ | 5013 |
| WF-4 | 640R436G04 | L | 5012 |
| WF-4 | 640R436G04 | $L$ | 5013 |
| WF-5 | 640R438G03 | W | 5012 |
| WF-5 | 640R438G03 | W | 5013 |
| 9 A | 640R420G03 | AE | 5013 |
| 9A | 640R420G03 | AE | 5014 |


| DESCRIPTION | ASSEMBLY NO. | REV | S/N |
| :---: | :---: | :---: | :---: |
| 9BX | 640R586G04 | F | 5013 |
| 9BX | 640R586G04 | F | 5014 |
| 9CX | 640R570G03 | $N$ | 5013 |
| 9CX | 640R570G03 | N | 5014 |
| WF-6 | 640R568G03 | H | 5013 |
| WF-6 | 640R568G03 | H | 5014 |
| OSU | 640R960G03 | $Y$ | 5007 |
| Matrix | 522R783G02 | $Y$ | 5007 |
| Al | 640R522G03 | T | 5007 |
| A2 | 640R524G03 | $N$ | 5007 |
| Bottom | 644R047G03 | T | 5007 |
| Top | 644R046G02 | P | 5007 |
| SPU | 758R040G01 | L | 5007 |
| Matrix | 640R927G02 | $v$ | - |
| SSP-8 | 640R552G03 | E | 5014 |
| SSP-8 | 640R552G03 | E | 5015 |
| RTD-1 | 640R508G03 | AH | 5012 |
| RTD-1 | 640R508G03 | AH | 5013 |
| RTD-2 | 640R510G03 | AP | 5012 |
| RTD-2 | 640R510G03 | AP | 5013 |
| RTD-3 | 640R512G03 | $K$ | 5012 |
| RTD-3 | 640R512G03 | K | 5013 |
| RTD-4 | 640R526G03 | $N$ | 5013 |
| RTD-4 | 640R526G03 | $N$ | 5014 |
| RTD-5 | 640R514G03 | R | 5012 |
| RTD-5 | 640R514G03 | R | 5013 |
| SSP-1X | 640R550G03 | J | 5012 |


| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| SSP-1X | 640R550G03 | J | 5013 |
| SSP-2 | 640R462G03 | $v$ | 5012 |
| SSP-2 | 640R462G03 | $v$ | 5013 |
| SSP-3 | 640R464G03 | U | 5012 |
| SSP-3 | 640R464G03 | U | 5013 |
| SSP-4 | 640R466G03 | M | 5012 |
| SSP-4 | 640R466G03 | M | 5013 |
| SSP-5 | 640R468G03 | P | 5012 |
| SSP-5 | 640R468G03 | P | 5013 |
| SSP-6 | 640R470G03 | R | 5012 |
| SSP-6 | 640R470G03 | R | 5013 |
| SSP-7 | 640R472G03 | $v$ | 5012 |
| SSP-7 | 640R472G03 | $V$ | 5013 |
| SSP-9 | 640R554G03 | J | 5013 |
| SSP-9 | 640R554G03 | J | 5014 |
| PSU | 758R050G02 | Y | 5007 |
| Matrix | 640R620G01 | F | 0004 |
| RFI Plate | 690R891G01 | A | 5007 |
| Reg Assy | $682 R 089603$ | L | 5004 |
| Misc Bd | 644R302G03 | R | 5007 |
| T-Chan CG | 688R483G03 | G | 5007 |
| T-Left | 688R485G03 | F | 5007 |
| T-Rgt | 688R487G03 | G | 5007 |
| T-Chan BU | 688R489G03 | F | 5007 |
| T-Ana Fil | 688R491G03 | H | 5012 |
| T-Ana Fil | 688R491G03 | H | 5013 |
| L-Ana Fil | 688R493G03 | G | 5012 |

1-18

| DESCRIPTION | ASSEMBLY NO. | REV. | S/N |
| :---: | :---: | :---: | :---: |
| L-Ana Fil | 688R493G03 | G | 5013 |
| PSU TRA BLK | 640R998G03 | H | 5013 |
| PSU TRA BLK | 640R998G03 | H | 5014 |
| DME | 688R481G03 | G | 5013 |
| DME | 688R481G03 | G | 5014 |
| IMC | 644R864G03 | E | 5007 |
| Relay-1 | 688R501G03 | E | 5007 |
| +5V | 644R078G03 | P | 5007 |
| Relay-2 | 688R502G03 | 0 | 5007 |
| +12VDA | 688R499G03 | D | 5013 |
| +12VDA | 688R499G03 | D | 5014 |
| Dual ENPA | 640R616G02 | J | 5007 |
| Relay-3 | 688R503G03 | c | 5007 |
| -12V | 644R069G03 | $N$ | 5007 |
| Relay-5 | 688R505G03 | C | 5007 |
| Relay-4 | 688R504G03 | C | 5007 |
| $+12 \mathrm{~V} \mathrm{Vm}$ | 688R500G03 | C | 5007 |
| MC | 688R495G03 | F | 5013 |
| MC | 688R495G03 | F | 5014 |
| CPH | 688R497G03 | D | 5007 |
| Wire Tab | 318R249 | $N$ | - |
| Enable | 682R381G03 | D | 5007 |

### 1.5 Thermal Vacuum Profiles

The OLS \#12 AVE underwent a series of Thermal Vacuum Tests. The profiles on the next pages represent the history of pumpdowns, SSS temperature and M1 temperatures experienced by the OLS \#12 AVE.


OLS \#12 THERMAL VACUUM PROFILE

0

$\underset{+}{\mathrm{N}}$
운
MI TEMP

### 1.6 Test History Calendar

The test history calendar is a capsule look at the day-to-day progress of the OLS \#12 AVE throughout its testing period at WEC.


UNIT ols\#12








1-33
$\square$


### 2.0 DEVELOPMENT SPECIFICATION REQUIREMENTS

### 2.1 Spectra (3.2.1.1.1.1, 3.1.1.1.2, 3.1.1.1.3)

The original OLS \#12 Channel spectral responses were calculated by computer programs (GAINSET for L Channel and TGAIN for $T$ channel) utilizing vendor spectral response data for the detectors, mirror and lens transmissivities/ reflectivities, and solar \& lunar spectral radiance.

Since the original publication of OLS \#12 spectral response, more current data has become available for HRD and PMT spectral response. The HRD total spectral response has been recalculated using average telescope data for OLS \#13 thru OLS \#16, measured data for the OLS 16 relay optics and a typical HRD detector response.

The PMT total spectral response was re-calculated using replacement PMT (S/N 16) from OLS 14, measured OLS \#16 ORA and OLS \#16 telescope data. This total system PMT response, which is out of spec between 500 and 530 nm , is the one used to report the degree of spec compliance. The worst case point is at 520 nm , where the PMT response exceeds the specified maximum by $8 \%$ of the maximum response. The out-of-spec region is relatively small, and the increased response in the 500 nm region will have no significant effect on the night time visible imagery.

The OLS \#12 T channel and L Day spectral responses are within specification.

ATTACHMENTS: OLS \#12 HRD Channel Spectral Response.
OLS \#12 PMT Channel Spectral Response.
OLS \#12 T Channel Spectral Response.



$$
2-3
$$

(3)

(3)

2-4

### 2.2 GEOMETRIC RESOLUTION

2.2.1 Fine Geometric Resolution - Infrared (3.2.1.1.2.1)

### 2.2.1.1 Baseline (Orbit Nominal)

The TF Surface Resolution Parameter (SRP) is within the development specification limits.

The VAX Computer programs calculate and plot the Fine Primary SRP, and the T Right \& Left Fallback modes. In addition, all Specification required modes are tabulated and presented. The designations on the graphs are defined as follows:

TFP T Fine Primary Electronics
TFB T Fine Backup (Redundant) Electronics
TSP T Smooth Primary Electronics
TSB T Smooth Backup Electronics

ATTACHMENTS: TF Curves SRP Orbit Nominal
TF SRP Tables Orbit Nominal

SYSTEM 12.,SRP TF NORMAL,SSS=5. . M1 = - 8 , ,DATE: 914


ORBIT
NOMINAL
$1 . \square$

SYSTEM 12.,SRP TF L. FBAK,SSS=5 . .,M1=-8 , DRTE: 914


ORBIT
NOMINAL $X=F B N D R M A L S O S$

SYSTEM 12, SRP TF R FBAK, SSS $=5 \ldots, M 1=-8$, , DATE: 914

T. COMPLETE, SRP (NM)

|  | FLT. ND. = | 2 ENV. = | SS5= | SDEGC M1 = | -BDEGC DATE: | 914 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| SEG | SUR. DIST. (NM) | TFP | TFB | TSP | TSB |  |
| LFT | -750. | 0.725 | 0.722 | 1. 732 | 1. 675 |  |
| MID | -750. | 1. 253 | 0.000 | 1. 828 | 1. 782 |  |
| RGT | -750. | 0. 995 | 0.989 | 1.766 | 1.713 |  |
| LFT | 0. | 0.000 | 0. 000 | 0. 000 | 0.000 |  |
| MID | 0. | 0. 000 | 0. 000 | 0. 000 | 0. 000 |  |
| RGT | 0. | 0. 000 | 0. 000 | 0. 000 | 0. 000 |  |
| LFT | -431. | 0.397 | 0. 391 | 1. 461 | 1. 410 |  |
| MID | -431. | 0. 631 | 0.000 | 1. 488 | 1. 439 |  |
| RGT | -431. | 0. 485 | 0.479 | 1. 461 | 1. 410 |  |
| LFT | -398. | 0. 379 | 0. 373 | 1. 408 | 1. 359 |  |
| MID | -398. | 0. 614 | 0. 611 | 1. 429 | 1. 382 |  |
| RGT | -398. | 0. 454 | 0.447 | 1. 406 | 1. 357 |  |
| LFT | 0. | 0.000 | 0.000 | 0. 000 | 0. 000 |  |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |  |
| RGT | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |  |
| LFT | 0. | 0. 248 | O. 245 | 0.968 | 0. 935 |  |
| MID | 0. | 0. 255 | 0. 252 | 0.96日 | 0.935 |  |
| RGT | 0. | 0. 236 | 0. 234 | 0.766 | 0.932 |  |
| LFT | 0. | 0. 000 | 0. 000 | 0.000 | 0. 000 |  |
| MID | 0. | 0.000 | 0.000 | 0. 000 | 0.000 |  |
| RGT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |  |
| LFT | 398. | 0. 489 | 0. 484 | 1. 411 | 1. 363 |  |
| MID | 398. | 0. 568 | 0. 565 | 1. 420 | 1. 372 |  |
| RGT | 398. | 0. 355 | 0. 351 | 1. 397 | 1. 348 |  |
| LFT | 431. | 0. 521 | 0. 515 | 1. 466 | 1. 415 |  |
| MID | 431. | 0. 573 | 0.000 | 1. 479 | 1. 430 |  |
| RGT | 431. | 0. 377 | 0. 373 | 1. 455 | 1. 405 |  |
| LFT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |  |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |  |
| RGT | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |  |
| LFT | 757. | 1. 061 | 1. 054 | 1. 785 | 1. 735 |  |
| MID | 757. | 1. 449 | 0.000 | 1.900 | 1. 860 |  |
| RGT | 757. | 0. 695 | 0.692 | 1.715 | 1. 658 |  |


| SEG | SUR. DIST. (NM) | TFP | TFB | TSP | TSB |
| :---: | :---: | :---: | :---: | :---: | :---: |
| LFT | -750. | 0. 845 | 0. 841 | 0.770 | 0.745 |
| MID | -750. | 0.000 | O. 000 | 0.813 | 0.793 |
| RGT | -750. | 0.753 | 0.749 | 0.785 | 0.762 |
| LFT | 0. | 0.000 | 0.000 | 0. 000 | 0.000 |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |
| RGT | 0. | 0. 000 | 0.000 | 0. 000 | 0.000 |
| LFT | -431. | 0. 900 | 0. 885 | 0. 912 | 0. 881 |
| MID | -431. | 0.000 | 0.000 | 0.929 | 0. 899 |
| RGT | -431. | 0. 854 | 0.843 | 0.912 | 0.881 |
| LFT | -398. | 0. 903 | 0. 888 | 0.918 | 0. 886 |
| MID | -398. | 0. 934 | 0.930 | 0. 932 | 0.901 |
| RGT | -398. | 0. 867 | 0. 855 | 0.916 | 0.885 |
| LFT | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |
| RGT | 0. | 0.000 | 0.000 | 0.000 | 0. 000 |
| LFT | 0. | 0.902 | 0.892 | 0.922 | 0. 890 |
| MID | 0. | 0.882 | 0. 871 | 0.922 | 0.890 |
| RGT | 0. | 0.857 | 0.852 | 0.920 | 0.888 |
| EFT | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |
| RGT | 0. | 0. 000 | 0.000 | 0. 000 | 0.000 |
| LFT | 398. | 0.936 | 0.925 | 0.920 | 0. 889 |
| MID | 398. | 0. 865 | 0.860 | 0.926 | 0. 894 |
| RGT | 398. | 0.846 | 0.838 | 0. 911 | 0. 879 |
| LFT | 431. | 0.916 | 0.906 | 0.915 | 0. 884 |
| MID | 431. | 0. 000 | 0.000 | 0. 924 | 0. 893 |
| RGT | 431. | 0. 854 | 0. 844 | 0.909 | Q. 877 |
| LFT | 0. | 0. 000 | 0.000 | 0. 000 | 0. 000 |
| MID | 0. | 0. 000 | 0.000 | 0. 000 | 0.000 |
| RGT | 0. | 0. 000 | 0.000 | 0. 000 | 0.000 |
| LFT | 757. | 0. 790 | 0.785 | 0. 790 | 0.768 |
| MID | 757. | 0. 000 | 0. 000 | 0. 840 | 0. 823 |
| RGT | 757. | 0. 797 | 0.793 | 0. 759 | 0. 734 |

## TF, LEFT, PRIMARY

| FLT. NO. $=12$ | . $=$ | 4 SSS= | $M 1=-8 D E G C$ | DATE: |
| :---: | :---: | :---: | :---: | :---: |
| SUR. DIST. (NM) | SRP | ACTUAL (NM) | SRP RATII |  |
| -750. |  | 0. 725 | 0. 845 |  |
| 0. |  | 0.000 | 0.000 |  |
| -431. |  | 0. 397 | 0. 900 |  |
| -398. |  | 0.379 | 0.903 |  |
| 0. |  | 0.000 | 0.000 |  |
| 0. |  | 0. 248 | 0.902 |  |
| 0. |  | 0. 000 | 0.000 |  |
| 398. |  | 0. 489 | 0.936 |  |
| 431. |  | 0. 521 | 0.916 |  |
| 0. |  | 0.000 | 0.000 |  |
| 757. |  | 1. 061 | 0.790 |  |

TF, LEFT, BACKUP

| F゙LT. NO. $=12$ | . $=$ | 4 SSS= 5 | M1 = -8DEGC | DATE: |
| :---: | :---: | :---: | :---: | :---: |
| SUR. DIST. (NM) | SRP | ACTUAL (NM) | SRP RATID |  |
| -750. |  | 0. 722 | 0. 841 |  |
| 0. |  | 0.000 | 0.000 |  |
| -431. |  | 0. 391 | 0. 885 |  |
| -399. |  | 0. 373 | 0. 888 |  |
| 0. |  | 0. 000 | 0. 000 |  |
| 0. |  | 0. 245 | 0. 892 |  |
| 0. |  | 0. 000 | 0. 000 |  |
| 398. |  | 0. 484 | 0. 925 |  |
| 431. |  | 0. 515 | 0. 906 |  |
| 0. |  | 0. 000 | 0.000 |  |
| 757. |  | 1. 054 | 0.785 |  |

## TF, RIGHT, PRIMARY

FLT. ND. $=12$ ENV. $=4$ 5SS $=5 D E G C$ MI = -BDEGC DATE: 914
SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATID

| -750. | 0.995 | 0.753 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.485 | 0.854 |
| -398. | 0.454 | 0.869 |
| 0. | 0.000 | 0.000 |
| 0. | 0.236 | 0.857 |
| 0. | 0.000 | 0.000 |
| 398. | 0.355 | 0.846 |
| 431. | 0.377 | 0.854 |
| 0. | 0.000 | 0.000 |
| 757. | 0.695 | 0.797 |

TF RIGHT, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= 5DEGC M1 = -8DEGC DATE: 914
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIO
$-750$.
0.999
0.749
0.
0.000
0.000
$-431$.
0.479
0.843 -39日.
0.447
0. 855
0.
0.000
0. 000
0.
0. 234
0.852
0.
0.000
0.000
398.
0. 351
0. 838
431.
0. 373
0. 844
0.
0.000
0.000
757.
0. 692
0.793

### 2.2.1.2 Acceptance - Vibration

OLS \#12 underwent acceptance level SSS vibration per DMSS-OLS300 with cone cooler S/N 024 on May 23, 1991. The pre-to-post vibration SRP performance is shown on the attached curves and tables. ATTACHMENTS: TF SRP Curves Previbration.

TF SRP Tables Previbration.
TF SRP Curves Postvibration.
TF SRP Tables Postvibration.

SYSTEM 12.,SRP. TF NORMAL,SSS=5 . .,M1 = -8.,DATE: 707.


SYSTEM 12,,SRP. TF. L. FBAK,SSS=5 . ,M1=-8.,DATE:707.



SYSTEM 12 , SRP TF R FBAK,SSS=5 $\ldots, M 1=-8$, DATE: 707


T, COMPLETE, SRP (NM)

|  | FLT. ND. = | ENV. = | 55S= | 5DEGC M1= | -BDEGC DA |
| :---: | :---: | :---: | :---: | :---: | :---: |
| SEG | SUR. DIST. (NM) | TFP | TFB | TSP | TSB |
| LFT | -750. | 0.716 | 0. 708 | 1. 730 | 1. 665 |
| MID | -750. | 1. 297 | 0. 000 | 1.846 | 1.793 |
| RGT | -750. | 0. 987 | 0. 983 | 1.764 | 1.703 |
| LFT | -600. | 0. 544 | 0. 534 | 1. 715 | 1. 648 |
| MID | -600. | 1. 008 | 0.000 | 1.786 | 1.726 |
| RGT | -600. | 0. 716 | 0.708 | 1.725 | 1.660 |
| LFT | -431. | 0. 391 | 0. 387 | 1. 459 | 1. 403 |
| MID | -431. | 0.621 | 0.000 | 1. 489 | 1. 433 |
| RGT | -431. | 0.476 | 0. 469 | 1. 463 | 1. 407 |
| LFT | -398. | 0. 378 | 0. 374 | 1. 406 | 1. 351 |
| MID | -398. | 0.608 | 0.602 | 1. 429 | 1. 375 |
| RGT | -398. | 0. 445 | 0.438 | 1.410 | 1. 356 |
| LFT | -200. | 0. 280 | 0. 278 | 1. 100 | 1. 057 |
| MID | -200. | 0. 346 | 0. 341 | 1. 105 | 1.063 |
| RET | -200. | 0. 296 | 0. 293 | 1. 102 | 1. 059 |
| LFT | 0. | 0. 253 | 0. 251 | 0. 971 | 0.933 |
| MID | 0. | 0. 253 | 0. 251 | 0. 971 | 0.934 |
| RGT | 0. | 0. 232 | 0. 232 | 0. 968 | 0.931 |
| LFT | 200. | 0. 322 | 0. 317 | 1. 104 | 1. 061 |
| MID | 200. | 0. 321 | 0.316 | 1. 106 | 1. 063 |
| RGT | 200. | 0. 263 | 0. 263 | 1.096 | 1. 054 |
| LFT | 398. | 0.477 | 0. 469 | 1. 408 | 1. 354 |
| MID | 398. | 0.556 | 0. 549 | 1. 419 | 1. 365 |
| RGT | 398. | 0.345 | 0. 344 | 1. 398 | 1. 344 |
| LFT | 431. | 0.503 | 0. 495 | 1. 465 | 1. 408 |
| MID | 431. | 0. 561 | 0. 000 | 1. 48 ? | 1. 426 |
| RGT | 431. | 0. 367 | 0. 365 | 1. 456 | 1. 399 |
| LFT | 601. | 0.732 | 0. 724 | 1.726 | 1. 661 |
| MID | 601. | 0.924 | 0. 000 | 1. 766 | 1. 704 |
| RGT | 601. | 0.500 | 0. 492 | 1. 704 | 1.639 |
| LFT | 757. | 1. 037 | 1. 035 | 1.780 | 1. 722 |
| MID | 757. | 1. 424 | 0. 000 | 1. 894 | 1. 845 |
| RGT | 757. | 0.686 | 0.678 | 1.717 | 1.653 |


| SEG | SUR. DIST. <br> (NM) | TFP | TFB | TSP | TSB |
| :--- | :---: | :---: | :---: | :---: | ---: |
|  |  |  |  |  |  |
| LFT | -750. | 0.934 | 0.825 | 0.769 | 0.740 |
| MID | -750. | 0.000 | 0.000 | 0.921 | 0.797 |
| RGT | -750. | 0.747 | 0.744 | 0.784 | 0.757 |
| LFT | -600. | 0.897 | 0.882 | 0.971 | 0.837 |
| MID | -600. | 0.000 | 0.000 | 0.907 | 0.876 |
| RGT | -600. | 0.805 | 0.797 | 0.876 | 0.843 |
| LFT | -431. | 0.886 | 0.877 | 0.911 | 0.976 |
| MID | -431. | 0.000 | 0.000 | 0.930 | 0.895 |
| RGT | -431. | 0.838 | 0.826 | 0.914 | 0.879 |
| LFT | -398. | 0.901 | 0.891 | 0.917 | 0.881 |
| MID | -398. | 0.926 | 0.917 | 0.932 | 0.897 |
| RGT | -398. | 0.851 | 0.838 | 0.920 | 0.884 |
| LFT | -200. | 0.894 | 0.889 | 0.917 | 0.882 |
| MID | -200. | 0.946 | 0.931 | 0.922 | 0.886 |
| RGT | -200. | 0.870 | 0.862 | 0.919 | 0.883 |
| LFT | 0. | 0.921 | 0.914 | 0.925 | 0.889 |
| MID | 0. | 0.873 | 0.866 | 0.925 | 0.889 |
| RGT | 0. | 0.845 | 0.844 | 0.922 | 0.886 |
| LFT | 200. | 0.946 | 0.932 | 0.921 | 0.885 |
| MID | 200. | 0.877 | 0.864 | 0.922 | 0.886 |
| RGT | 200. | 0.839 | 0.838 | 0.914 | 0.878 |
| LFT | 398. | 0.912 | 0.898 | 0.918 | 0.883 |
| MID | 398. | 0.847 | 0.837 | 0.925 | 0.890 |
| RGT | 398. | 0.823 | 0.820 | 0.912 | 0.876 |
| LFT | 431. | 0.884 | 0.871 | 0.915 | 0.879 |
| MID | 431. | 0.000 | 0.000 | 0.926 | 0.891 |
| RGT | 431. | 0.831 | 0.827 | 0.909 | 0.874 |
| LFT | 601. | 0.821 | 0.813 | 0.876 | 0.843 |
| MID | 601. | 0.000 | 0.000 | 0.896 | 0.865 |
| RGT | 601. | 0.823 | 0.811 | 0.865 | 0.831 |
| LFT | 757. | 0.774 | 0.771 | 0.788 | 0.762 |
| MID | 757. | 0.000 | 0.800 | 0.838 | 0.816 |
| RGT | 757. | 0.787 | 0.777 | 0.760 | 0.731 |

## TF, LEFT, PRIMARY

FLT. ND. $=12$ ENV. $=4$ 5S5= 5DEGC M1=-8DEGC DATE: 707 SUR.DIST. (NM) SRP ACTUAL (NM) SRP RATID

| -750. | 0.716 | 0.834 |
| ---: | ---: | ---: |
| -600. | 0.544 | 0.897 |
| -431. | 0.391 | 0.886 |
| -398. | 0.378 | 0.901 |
| -200. | 0.280 | 0.894 |
| 0. | 0.253 | 0.921 |
| 200. | 0.322 | 0.946 |
| 398. | 0.477 | 0.912 |
| 431. | 0.503 | 0.884 |
| 601. | 0.732 | 0.821 |
| 757. | 1.039 | 0.774 |

TF, LEFT, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= 5DEGC M1=-BDEGC DATE: 707
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -750. | 0.708 | 0.825 |
| ---: | ---: | ---: |
| -600. | 0.534 | 0.882 |
| -431. | 0.387 | 0.877 |
| -398. | 0.374 | 0.891 |
| -200. | 0.278 | 0.889 |
| 0. | 0.251 | 0.914 |
| 200. | 0.317 | 0.932 |
| 398. | 0.469 | 0.898 |
| 431. | 0.495 | 0.871 |
| 601. | 0.724 | 0.813 |
| 757. | 1.035 | 0.771 |

TF, RIGHT, PRIMARY
FLT. NO. $=12$ ENV. $=4$ SSS= SDEGC M1 = -BDEGC DATE: 707 SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATIO

| -750. | 0.987 | 0.747 |
| ---: | ---: | ---: |
| -600. | 0.716 | 0.805 |
| -431. | 0.476 | 0.838 |
| -398. | 0.445 | 0.851 |
| -200. | 0.296 | 0.870 |
| 0. | 0.232 | 0.845 |
| 200. | 0.263 | 0.839 |
| 398. | 0.345 | 0.823 |
| 431. | 0.367 | 0.831 |
| 601. | 0.500 | 0.823 |
| 757. | 0.686 | 0.787 |

TF RIGHT, BACKUP
FLT. ND. $=12$ ENU. $=4$ 5SS= 5DEGC M1 = -BDEGC DATE: 707
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIO

| -750. | 0.983 | 0.744 |
| ---: | ---: | ---: |
| -600. | 0.708 | 0.797 |
| -431. | 0.469 | 0.826 |
| -398. | 0.438 | 0.838 |
| -200. | 0.293 | 0.862 |
| 0. | 0.232 | 0.844 |
| 200. | 0.263 | 0.838 |
| 398. | 0.344 | 0.820 |
| 431. | 0.365 | 0.827 |
| 601. | 0.492 | 0.811 |
| 757. | 0.678 | 0.777 |

SYSTEM 12 ,SRP TF R FBAK,SSS=5 , M1 = -8, DRTE: 914


SYSTEM 12,SRP TF NORMAL,SSS=5 , M1=-8 ,DATE:914


POST
VIB
$\stackrel{\square}{-}$



$N$
0
0
0


SYSTEM 12,SRP TF L FBAK,SSS=5 , M1 = - 8 , DATE: 914


POST
VIB

## T, COMPLETE, SRP (NM)



T, COMPLETE, SRP RATIO

| SEG | SUR. DIST. <br> (NM) | TFP | TFB | TSP | TSB |
| :--- | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |
| LFT | -750. | 0.845 | 0.841 | 0.770 | 0.745 |
| MID | -750. | 0.000 | 0.000 | 0.813 | 0.793 |
| RGT | -750. | 0.753 | 0.749 | 0.785 | 0.762 |
| LFT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| MID | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| RGT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| LFT | -431. | 0.900 | 0.885 | 0.912 | 0.881 |
| MID | -431. | 0.000 | 0.000 | 0.929 | 0.899 |
| RGT | -431. | 0.854 | 0.843 | 0.912 | 0.881 |
| LFT | -398. | 0.903 | $0.88 日$ | 0.918 | 0.886 |
| MID | -398. | 0.934 | 0.930 | 0.932 | 0.901 |
| RGT | -398. | 0.869 | 0.855 | 0.916 | 0.885 |
| LFT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| MID | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| RGT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| LFT | 0. | 0.902 | 0.892 | 0.922 | 0.890 |
| MID | 0. | 0.882 | 0.971 | 0.922 | 0.890 |
| RGT | 0. | 0.857 | 0.852 | 0.920 | 0.888 |
| LFT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| MID | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| RGT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| LFT | 398. | 0.936 | 0.925 | 0.920 | 0.889 |
| MID | 398. | 0.865 | 0.860 | 0.926 | 0.894 |
| RGT | 398. | 0.846 | 0.838 | 0.911 | 0.879 |
| LFT | 431. | 0.916 | 0.906 | 0.915 | 0.884 |
| MID | 431. | 0.000 | 0.000 | 0.924 | 0.893 |
| RGT | 431. | 0.854 | 0.844 | 0.909 | 0.877 |
| LFT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| MID | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| RGT | 0. | 0.000 | 0.000 | 0.000 | 0.000 |
| LFT | 757. | 0.790 | 0.785 | 0.790 | 0.768 |
| MID | 757. | 0.000 | 0.000 | 0.840 | 0.823 |
| RGT | 757. | 0.797 | 0.793 | 0.759 | 0.734 |

## TF, LEFT, PRIMARY

FLT. NO. $=12$ ENV. $=4$ SSS= SDEGC M1 = $\rightarrow$ BDEGC DATE: 914
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATIO

| -750. | 0.725 | 0.845 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.397 | 0.900 |
| -398. | 0.379 | 0.903 |
| 0. | 0.000 | 0.000 |
| 0. | 0.248 | 0.902 |
| 0. | 0.000 | 0.000 |
| 398. | 0.489 | 0.736 |
| 431. | 0.521 | 0.916 |
| 0. | 0.000 | 0.000 |
| 757. | 1.061 | 0.790 |

TF, LEFT, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= SDEGC M1 = -EDEGC DATE: 914
SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATIロ

| -750. | 0.722 | 0.841 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.391 | 0.885 |
| -398. | 0.373 | 0.898 |
| 0. | 0.000 | 0.000 |
| 0. | 0.245 | 0.892 |
| 0. | 0.000 | 0.000 |
| 398. | 0.484 | 0.925 |
| 431. | 0.515 | 0.904 |
| 0. | 0.000 | 0.000 |
| 757. | 1.054 | 0.795 |

$2-26$

TH, RI GUT, PRIMARY
FLT. NO. $=12$ END. $=4$ SSS= SDEGC M1=-EDEGC DATE: 914 SUR. DIST. (NM) SRP ACTUAL (NM) SP RATIO

| -750. | 0.995 | 0.753 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.485 | 0.854 |
| -398. | 0.454 | 0.869 |
| 0. | 0.000 | 0.000 |
| 0. | 0.336 | 0.857 |
| 0. | 0.000 | 0.000 |
| 398. | 0.355 | 0.846 |
| 431. | 0.377 | 0.854 |
| 0. | 0.000 | 0.000 |
| 757. | 0.695 | 0.797 |

TH RIGHT, BACKUP

FLT. ND. $=12$ ENS. $=4$ SSS $=5 D E G C$ MI = -BDEGC DATE: 914
SUR. DIST. (NM) SRP ACTUAL (NM) SP RATIO

| -750. | 0.989 | 0.749 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.479 | 0.843 |
| -398. | 0.447 | 0.855 |
| 0. | 0.000 | 0.000 |
| 0. | 0.234 | 0.852 |
| 0. | 0.000 | 0.000 |
| 398. | 0.351 | 0.838 |
| 431. | 0.373 | 0.844 |
| 757. | 0.000 | 0.000 |

### 2.2 Geometric Resolution (Cont'd)

2.2.1 Fine Geometric Resolution, Infrared (Cont'd) (3.2.1.1.2.1)

### 2.2.1.3 Acceptance - Thermal Vacuum

The attached TF SRP curves and tables demonstrate in-spec
performance at the thermal vacuum test limits. The Orbit Nominal SRP curves are contained in paragraph 2.2.1.1 and are not inciuded here.

ATTACHMENTS: TF SRP Curves Hot Limits
TF SRP Tables Hot Limits
TF SRP Curves Cold Limits
TF SRP Tables Cold Limits

SYSTEM 12,SRP TF NORMFL,SSS=7 , M1=12, DATE: 905


SYSTEM 12.,SRP TF L FBAK,SSS=7 ,M1=12,,DATE: 905

| - |  | HOT |
| :---: | :---: | :---: |
|  | -SPEC LIMIT | LIMIT |
|  | 0 =NORMAL SOS |  |

SYSTEM 12,SRP TF R FEAK, $\mathrm{SSS}=7 \quad, \mathrm{M1}=12$, DRTE: 905


|  | FLT. ND. $=$ | ENV. = | Sss= |  | GC DATE: |
| :---: | :---: | :---: | :---: | :---: | :---: |
| SEG | SUR. DIST. <br> (NM) | TFP | TFB | TSP | TSB |
| LFT | -750. | 0. 771 | 0.767 | 1. 751 | 1.697 |
| MID | -750. | 1. 252 | 0. 000 | 1. 833 | 1. 789 |
| RGT | -750. | 1. 015 | 1. 016 | 1. 777 | 1. 727 |
| - $\bar{T}$ | 0. | 0. 000 | 0. 000 | 0. 000 | 0. 000 |
| MID | 0. | 0. 000 | 0.000 | 0. 000 | 0.000 |
| RGT | 0. | 0. 000 | 0. 000 | 0. 000 | 0.000 |
| LFT | -431. | 0. 427 | 0. 424 | 1. 466 | 1. 417 |
| MID | -431. | 0. 642 | 0. 000 | 1. 493 | 1. 446 |
| RGT | -431. | 0. 491 | 0. 489 | 1. 469 | 1. 420 |
| LFT | -398. | 0. 399 | 0. 396 | 1. 413 | 1. 366 |
| MID | -398. | 0. 624 | 0. 622 | 1. 433 | 1. 388 |
| RGT | -398. | 0. 464 | 0. 462 | 1. 413 | 1. 365 |
| LFT | 0. | 0. 000 | 0. 000 | 0. 000 | 0. 000 |
| MID | 0. | 0. 000 | 0.000 | 0. 000 | 0. 000 |
| RGT | 0. | 0.000 | 0.000 | 0. 000 | 0. 000 |
| LFT | 0. | 0. 265 | 0. 263 | 0. 971 | 0.938 |
| MID | 0. | 0. 260 | 0. 258 | 0. 971 | 0.938 |
| RGT | 0. | 0. 238 | 0.236 | 0.968 | 0.935 |
| LFT | 0. | 0. 000 | 0. 000 | 0.000 | 0. 000 |
| MID | 0. | 0. 000 | 0. 000 | 0.000 | 0. 000 |
| RGT | 0. | 0. 000 | 0. 000 | 0.000 | 0. 000 |
| LFT | 398. | 0. 491 | 0. 489 | 1. 409 | 1. 362 |
| MID | 398. | 0. 567 | 0. 563 | 1. 421 | 1. 374 |
| RGT | 398. | 0. 367 | 0. 363 | 1. 399 | 1. 352 |
| LFT | 431. | 0. 521 | 0. 518 | 1. 468 | 1. 419 |
| MID | 431. | 0. 579 | 0.000 | 1. 483 | 1. 435 |
| RGT | 431. | 0. 394 | 0.390 | 1.458 | 1. 409 |
| LFT | 0. | 0. 000 | 0.000 | 0. 000 | 0. 000 |
| MID | 0. | 0.000 | 0. 000 | 0. 000 | O. 000 |
| RGT | 0. | 0. 000 | 0. 000 | 0. 000 | 0.000 1.744 |
| LFT | 757. | 1. 067 | 1. 069 | 1.792 | 1.744 1.876 |
| MID | 757. | 1. 475 | 0. 000 | 1. 916 | 1. 876 |
| RGT | 757. | 0. 729 | 0. 724 | 1. 728 | 1.673 |

## T, COMPLETE, SRP RATID

| SEG | SUR. DIST. (NM) | TFP | TFB | TSP | T5B |
| :---: | :---: | :---: | :---: | :---: | :---: |
| LFT | -750. | 0. 897 | 0. 893 | 0. 779 | 0. 755 |
| MID | -750. | 0. 000 | 0. 000 | 0. 815 | 0. 795 |
| RGT | -750. | 0.768 | 0. 769 | 0.790 | 0. 768 |
| LFT | 0. | 0.000 | 0.000 | 0.000 | 0. 000 |
| MID | 0. | 0. 000 | 0.000 | 0. 000 | 0. 000 |
| RGT | 0. | 0.000 | 0. 000 | 0.000 | 0. 000 |
| LFT | -431. | 0.766 | 0.960 | 0.916 | 0. 885 |
| MID | -431. | 0.000 | 0. 000 | 0.932 | 0. 903 |
| RGT | -431. | 0. 865 | 0. 861 | 0.918 | 0. 887 |
| LFT | -398. | 0. 952 | 0.944 | 0.922 | 0. 890 |
| MID | -39日. | 0. 750 | 0.946 | 0.934 | 0. 905 |
| RGT | -398. | 0. 887 | 0.883 | 0.921 | 0. 890 |
| LFT | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |
| RGT | 0. | 0. 000 | 0. 000 | 0.000 | 0. 000 |
| LFT | 0. | 0.965 | 0.956 | 0.924 | 0. 893 |
| MID | 0. | 0. 899 | 0. 890 | 0.925 | 0. 894 |
| RGT | 0. | 0. 865 | 0.858 | 0.922 | 0. 891 |
| LFT | 0. | 0. 000 | 0. 000 | 0.000 | 0.000 |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |
| RGT | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |
| LFT | 398. | 0.940 | 0.935 | 0.919 | 0. 888 |
| MID | 398. | 0. 863 | 0. 958 | 0.927 | 0.896 |
| RGT | 398. | 0. 176 | 0.866 | 0.912 | 0. 882 |
| LFT | 431. | 0. 917 | 0.911 | 0.917 | 0. 886 |
| MID | 431. | 0. 000 | 0.000 | 0.926 | 0. 896 |
| RGT | 431. | 0. 892 | 0. 883 | 0.910 | 0. 880 |
| LFT | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |
| RGT | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |
| LFT | 757. | 0. 795 | 0.796 | 0.793 | 0. 772 |
| MID | 757. | 0. 000 | 0.000 | 0. 848 | 0. 830 |
| RGT | 757. | 0. 835 | 0.830 | 0.764 | 0. 740 |

## "F, LEFT, PRIMARY

FLT. NQ. $=12$ ENV. $=4$ SSS= 7DEGC M1 = 12DEGC DATE: 905 SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID
$-750$.
0.
-431.
$-398$.
0. 771
0.000
0. 427
0. 000
0. 399

- 906

0. 000
1. 
2. 265
0.000
0.491
3. 
4. 521
5. 000
6. 952
0.000
0.965
7. 000
0.940
8. 
9. 067
0.917
0.000
10. 
11. 795

TF, LEFT, BACKUP
FLT. NO. $=12$ ENV. $=4$ 5S5= 7DEGC MI= 12DEGC DATE: 905
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -750. | 0.767 | 0.893 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.424 | 0.960 |
| -398. | 0.396 | 0.944 |
| 0. | 0.000 | 0.000 |
| 0. | 0.263 | 0.956 |
| 0. | 0.000 | 0.000 |
| 398. | 0.489 | 0.935 |
| 431. | 0.518 | 0.911 |
| 0. | 0.000 | 0.000 |
| 757. | 1.069 | 0.796 |

## TF, RIGHT, PRIMARY

FLT. NO. $=12$ ENV. $=4$ SSS= 7DEGC M1= IEDEGC DATE: 905
SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATIU

| -750. | 1.015 | 0.768 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.491 | 0.865 |
| -398. | 0.464 | 0.887 |
| 0. | 0.000 | 0.000 |
| 0. | 0.238 | 0.865 |
| 0. | 0.000 | 0.000 |
| 398. | 0.367 | 0.876 |
| 431. | 0.394 | 0.892 |
| 757. | 0.000 | 0.000 |
|  | 0.728 | 0.835 |

TF RIGHT, BACKUP

| FLT. NO. $=12$ | ENV. = | $4.555=7$ | 7DEGC | M1 $=$ | = 12DEGC | DATE: | 905 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| SUR. DIST. (NM) | SRP | ACTUAL (NM) | ) S | SRP R | RATID |  |  |
| -750. |  | 1.016 |  | 0.7 | 769 |  |  |
| 0. |  | 0.000 |  | 0. | 000 |  |  |
| -431. |  | 0.489 |  | 0. | 861 |  |  |
| -398. |  | 0.462 |  | O. | 883 |  |  |
| 0. |  | 0. 000 |  | 0.0 | 000 |  |  |
| 0. |  | 0. 236 |  | 0. 8 | 858 |  |  |
| 0. |  | 0. 000 |  | 0.0 | 000 |  |  |
| 398. |  | 0. 363 |  | 0.8 | 366 |  |  |
| 431. |  | 0. 390 |  | 0. E | 883 |  |  |
| 0. |  | 0.000 |  | 0.0 | 000 |  |  |
| 757. |  | 0. 724 |  | O. 8 | 330 |  |  |

SYSTEM 12.,SRP TF NORMPL,SSS=3. . M1 = -8, DRTE: 908



SYSTEM 12 ,,SRP. TF L FBAK,SSS=3. ,M1=-8, DATE: 908


1 : $\quad \vdots \quad \vdots$

SYSTEM 12,SRP TF R FBAK,SSS=3 , M1 = -8, DATE: 9 D8


T, COMPLETE, SRP (NM)


T, COMPLETE, SRP RATID

| SEG | SUR. DIST. (NM) | TFP | TFB | TSP | T5B |
| :---: | :---: | :---: | :---: | :---: | :---: |
| LFT | -750. | 0.798 | 0.790 | 0.762 | 0. 731 |
| MID | -750. | 0. 000 | 0.000 | 0.809 | 0. 784 |
| RGT | -750. | 0. 734 | 0. 728 | 0.779 | 0. 751 |
| LFT | 0. | 0. 000 | 0. 000 | 0.000 | 0.000 |
| MID | 0. | 0. 000 | 0.000 | 0. 000 | 0.000 |
| Ret | 0. | 0.000 | 0. 000 | 0.000 | 0. 000 |
| LFT | -431. | 0.857 | 0. 848 | 0. 907 | 0. 871 |
| MID | -431. | 0.000 | 0. 000 | 0.925 | 0. 889 |
| RGT | -431. | 0.806 | 0. 801 | 0.909 | 0. 872 |
| LFT | -398. | 0.862 | 0. 854 | 0.912 | 0. 875 |
| MID | -398. | 0. 918 | 0. 909 | 0.928 | 0. 892 |
| RGT | -398. | 0. 838 | 0. 833 | 0.917 | 0. 880 |
| LFT | 0. | 0. 000 | 0. 000 | 0. 000 | 0. 000 |
| MID | 0. | 0. 000 | 0.000 | 0.000 | 0. 000 |
| RGT | 0. | 0.000 | 0. 000 | 0. 000 | 0. 000 |
| LFT | 0. | 0. 943 | 0. 934 | 0.926 | 0. 889 |
| MID | 0. | 0. 861 | 0. 852 | 0.922 | 0. 885 |
| RGT | 0. | 0. 841 | 0. 835 | 0. 919 | 0. 888 |
| LFT | 0. | 0. 000 | 0.000 | 0. 000 | 0.000 |
| MID | 0. | 0. 000 | 0.000 | 0. 000 | 0. 000 |
| RGT | 0. | 0. 000 | 0. 000 | 0. 000 | 0. 000 |
| LFT | 398. | 0.910 | 0. 903 | 0.917 | 0. 880 |
| MID | 398. | 0. 847 | 0. 839 | 0.923 | 0. 886 |
| RGT | 398. | 0. 841 | 0. 833 | 0.909 | 0.873 |
| LFT | 431. | 0.901 | 0. 894 | 0.912 | 0.876 |
| MID | 431. | 0.000 | 0. 000 | 0. 922 | 0. 886 |
| RGT | 431. | 0. 822 | 0. 813 | 0.906 | 0. 869 |
| LFT | 0. | 0. 000 | 0. 000 | 0. 000 | 0. 000 |
| MID | 0. | 0. 000 | 0. 000 | 0.000 | 0.000 |
| RGT | 0. | 0. 000 | 0.000 | 0.000 | 0.000 |
| LFT | 757. | 0. 774 | 0.768 | 0.785 | 0. 759 |
| MID | 757. | 0. 000 | 0. 000 | 0.837 | 0.816 |
| RGT | 757. | 0. 778 | 0. 769 | 0. 755 | 0.725 |

## TF, LEFT, PRIMARY

FLT. ND. $=12$ ENV. $=4$ SSS= 3DEGC M1 $=-$ BDEGC DATE: 908
SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATIO

| -750. | 0.685 | 0.798 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.378 | 0.857 |
| -398. | 0.362 | 0.862 |
| 0. | 0.000 | 0.000 |
| 0. | 0.259 | 0.943 |
| 0. | 0.000 | 0.000 |
| 398. | 0.476 | 0.910 |
| 431. | 0.512 | 0.901 |
| 0. | 0.000 | 0.000 |
| 757. | 1.040 | 0.774 |

TF, LEFT, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= 3DEGC M1 = -GDEGC DATE: 908
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIU
$-750$.
0.
$-431$.
$-398$.
0.
0.
0.
398.
431.
0.
757.
0.678
0.790
0. 000
0. 375
0. 358
0.000
0. 257
0.000
0. 472
0. 508
0. 000

1. 032
2. 000
3. 848
4. 854
5. 000
0.934
6. 000
7. 903
0.894
0.000
0.768

TF,RIGHT, PRIMARY
FLT. ND. $=12$ ENV. $=4$ SSS= 3DEGC MI $=-$ BDEGC DATE: 908 SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -750. | 0.769 | 0.734 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.458 | 0.806 |
| -398. | 0.438 | 0.838 |
| 0. | 0.000 | 0.000 |
| 0. | 0.231 | 0.841 |
| 0. | 0.000 | 0.000 |
| 398. | 0.353 | 0.841 |
| 431. | 0.363 | 0.822 |
| 0. | 0.000 | 0.000 |
| 757. | 0.678 | 0.778 |

TF RIGHT, BACKUP
FLT. NO. $=12$ ENV. $=4$ 5SS= 3DEGC M1 = -BDEGC DATE: $90 B$
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIO

| -750. | 0.962 | 0.728 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.455 | 0.801 |
| -398. | 0.436 | 0.833 |
| 0. | 0.000 | 0.000 |
| 0. | 0.230 | 0.835 |
| 0. | 0.000 | 0.000 |
| 398. | 0.349 | 0.833 |
| 431. | 0.359 | 0.813 |
| 0. | 0.000 | 0.000 |
| 757. | 0.671 | 0.769 |

### 2.2 Geometric Resolution (Cont'd)

2.2.2 Fine Geometric Resolution - Daytime Visua] (3.2.1.1.2.1)
2.2.2.1 Baseline (Orbit Nominal)

The LF SRP is within the specification limits in both Primary and Redundant configurations.

ATTACHMENTS: LF SRP Curves Orbit Nominal
LF SRP Tables Orbit Nominal

SYSTEM 12.,SRP LF NORMAL,SSS=5 . .,M1=-8, , DATE:912


SYSTEM 12 ,,SRP LF FBRCK ,,SSS=5 . ,M1=--8, DATE:912


## LF, DAY, NORMAL, PRIMARY

FLT. NO. $=12$ ENV. $=4$ SSS= 5DEGC M1 = -BDEGC DATE: 912
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -800. | 0.474 | 0.982 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.304 | 0.933 |
| -398. | 0.396 | 0.980 |
| 0. | 0.000 | 0.000 |
| 0. | 0.330 | 0.957 |
| 0. | 0.000 | 0.000 |
| 438. | 0.398 | 0.987 |
| 0. | 0.310 | 0.947 |
| 800. | 0.000 | 0.000 |
|  | 0.462 | 0.958 |

L_F, DAY, NDRMAL, BACKUP

FLT. ND. $=12$ ENV. $=4$ SSS= 5DEGC M1 = -BDEGC DATE: 912
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -800. | 0.474 | 0.981 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.301 | 0.924 |
| -398. | 0.395 | 0.978 |
| 0. | 0.000 | 0.000 |
| 0. | 0.228 | 0.949 |
| 0. | 0.000 | 0.000 |
| $39 日$. | 0.397 | 0.985 |
| 431. | 0.306 | 0.939 |
| 0. | 0.000 | 0.000 |
| 800. | 0.462 | 0.956 |

## LF, DAY, FALLBACK, PRIMARY

FLT. ND. $=12$ ENV. $=4$ SSS= 5DEGC M1 = -GDEGC DATE: 912
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -787. | 0.615 | 0.926 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.332 | 0.950 |
| -398. | 0.316 | 0.951 |
| 0. | 0.000 | 0.000 |
| 0. | 0.205 | 0.938 |
| 0. | 0.000 | 0.000 |
| 398. | 0.322 | 0.969 |
| 431. | 0.340 | 0.972 |
| 788 | 0.000 | 0.000 |
| 0. | 0.635 | 0.954 |

LF, DAY, FALLBACK, BACKUP
FLT. NO. $=12$ ENV. $=4555=5 D E G C$ MI = -BDECC DATE: 912
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -787. | 0.614 | 0.925 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.329 | 0.941 |
| -398. | 0.313 | 0.942 |
| 0. | 0.000 | 0.000 |
| 0. | 0.203 | 0.928 |
| 0. | 0.000 | 0.000 |
| 398. | 0.319 | 0.959 |
| 431. | 0.337 | 0.963 |
| 0. | 0.000 | 0.000 |
| 788. | 0.634 | 0.953 |

### 2.2 Geometric Resolution (Cont'd)

### 2.2.2 Fine Geometric Resolution - Daytime Visual (Cont'd) <br> (3.2.1.1.2.1)

### 2.2.2.2 Acceptance - Vibration

OLS \#12 underwent Acceptance-level SSS vibration on May 23, 1991. The Pre-to-Post Vibration SRP performance is within Specification requirements and is shown on the attached curves and tables. No changes in SRP performance occurred as a result of vibration.

ATTACHMENTS: LF SRP Curves Pre-Vibration<br>LF SRP Tables Pre-Vibration<br>LF SRP Curves Post-Vibration<br>LF SRP Tables Post-Vibration

SYSTEM 12 , SRP LF NORMFL, SSS= 23 , M1 = 24 , DRTE:527


## LF, DAY, NORMAL, PRIMARY



LF: DAY, NORMAL, BACKUP
FLT. ND. $=12$ ENV. $=2$ SSS= 23DEGC M1=24DEGC DATE: 527
SUR.DIST. (NM) SRP ACTUAL (NM) SRP RATID

| -800. | 0.499 |  |
| ---: | ---: | ---: |
| 0. | 0.000 | 1.034 |
| -431. | 0.303 | 0.000 |
| -398. | 0.401 | 0.928 |
| 0. | 0.000 | 0.993 |
| 0. | 0.330 | 0.000 |
| 0. | 0.000 | 0.960 |
| 398 | 0.407 | 0.000 |
| 431. | 0.309 | 1.010 |
| 0. | 0.000 | 0.947 |
| 800. | 0.496 | 0.000 |
|  |  | 1.027 |

SYSTEM . 12 ,,SRP LF NORMAL, SSS= 23 , M1 = 24 , DRTE: 814


LF, DAY, NDRMAL, PR IMARY
FLT. NO. $=12$ ENV. $=2$ SSS= 23DEGC M1= 24DEGC DATE: 814 SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATID

| -800. | 0.493 | 1.022 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.305 | 0.937 |
| -398. | 0.401 | 0.973 |
| 0. | 0.000 | 0.000 |
| 0. | 0.234 | 0.974 |
| 0. | 0.000 | 0.000 |
| 398. | 0.406 | 1.007 |
| 431. | 0.310 | 0.950 |
| 0. | 0.000 | 0.000 |
| 800 | 0.504 | 1.044 |

LF, DAY, NDRMAL, BACKUP
FLT. ND. $=12$ ENV. $=2$ SSS= 23DEGC MI = 24DEGC DATE: 814
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATIC
$-800$
0.494

1. 023
2. 

0.000
0. 000
-431.
0. 303
0. 928
0.401
0.995
0. 000
0. 000
0.970
0.000

1. 008
2. 940
3. 000
4. 
5. 233
6. 000
0.406
7. 307
0.000
8. 045

### 2.2 Geometric Resolution (Cont'd)

### 2.2.2 Eine Geometric Resolution - Daytime Visual (Cont'd)

 (3.2.1.1.2.1)
### 2.2.2.3 Acceptance - Thermal Vacuum <br> OLS \#12 LF SRP is within the specification limits in both

 Primary and Redundant configurations. The Orbit Nominal curves are in paragraph 2.2.2.1 and are not included here.
## ATTACHMENTS: LF SRP Curve Hot Limit

LF SRP Tables Hot Limit
LF SRP Curves Cold Limit
LF SRP Tables Cold Limit

SYSTEM 12.,SRP LF NORMAL,SSS=7..., M1 = 12 , DATE: 903



SYSTEM 12 ,SRP LF FBACK , SSS=7 $7, M 1=12$, DRTE: 903


LF. DAY, FALLBACK, PRIMARY
FLT. NO. $=12$ ENV. $=4$ 5SS= 7DEGC M1 = 1PDEGC DATE: 903
SUR.DIST. (NM) SRP ACTUAL (NM) SRP RATID

| -787. | 0.625 | 0.941 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.334 | 0.954 |
| -398. | 0.317 | 0.955 |
| 0. | 0.000 | 0.000 |
| 0. | 0.307 | 0.947 |
| 0. | 0.000 | 0.000 |
| 478. | 0.321 | 0.965 |
| 0. | 0.341 | 0.975 |
| 788. | 0.000 | 0.000 |
| 0. | 0.645 | 0.969 |

LF. DAY, FALLBACK, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS $=7$ DEGC M1 $=12 D E G C$ DATE: 903
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -787. | 0.625 | 0.942 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.331 | 0.947 |
| -398. | 0.315 | 0.947 |
| 0. | 0.000 | 0.000 |
| 0. | 0.306 | 0.939 |
| 0. | 0.000 | 0.000 |
| 498. | 0.319 | 0.957 |
| 431. | 0.339 | 0.968 |
| 788. | 0.000 | 0.000 |
|  | 0.645 | 0.969 |

## LF, DAY, NDRMAL, PR IMARY

FLT. NG. $=12$ ENV. $=4$ SSS= 7DEGC MI = 1EDEGC DATE: 903
SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATID

| -800. | 0.480 | 0.995 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.305 | 0.935 |
| -398. | 0.397 | 0.989 |
| 0. | 0.000 | 0.000 |
| 0. | 0.230 | 0.960 |
| 0. | 0.000 | 0.000 |
| 498. | 0.401 | 0.993 |
| 0. | 0.300 | 0.920 |
| 800. | 0.000 | 0.000 |
|  | 0.476 | 0.986 |

LF, DAY, NORMAL, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= 7DEGC MI= 12DEGC DATE: 9O3
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIO

| -800. | 0.480 | 0.994 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 0.302 | 0.926 |
| -398. | 0.399 | 0.989 |
| 0. | 0.000 | 0.000 |
| 0. | 0.329 | 0.953 |
| 0. | 0.000 | 0.000 |
| 398. | 0.400 | 0.993 |
| 431. | 0.397 | 0.911 |
| 0. | 0.000 | 0.000 |
| 800. | 0.476 | 0.985 |

SYSTEM 12,SRP LF NORMAL,SSS=3 . .MI =-8, DATE: $9 \varnothing 8$

| $\begin{aligned} & -=\text { SPEC LIMIT } \\ & 0=+Z E O S \\ & x=-Z E S \end{aligned}$ |  |
| :---: | :---: |
|  |  |

COLD
LIMIT


SYSTEM 12 ,SRP LF FBACK , SSS=3 . ,M1 = -8, DATE: 908


LF, DAY, NDRMAL, PRIMARY
FLT. ND. $=12$ ENV. $=4$ SSS= 3DECC M1= -8DEGC DATE: 908
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIO

| -800. | 0.480 | 0.994 |
| ---: | ---: | ---: |
| $* * * * *$ | 0.000 | $* * * * * *$ |
| -431. | 0.305 | 0.935 |
| -398. | 0.394 | 0.977 |
| $* * * * *$ | 0.000 | $* * * * * *$ |
| 0. | 0.232 | 0.967 |
| $* * * * *$ | 0.000 | $* * * * * *$ |
| 398. | 0.400 | 0.990 |
| 431. | 0.304 | 0.931 |
| $* * * * *$ | 0.000 | $* * * * *$ |
| 800. | 0.462 | 0.957 |

LF, DAY, NDRMAL, BACKUP
FLT. ND. $=12$ ENU. $=4$ SSS= 3DEGC $M 1=-$ BDEGC DATE: 908
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIO

| -800. | 0.480 | 0.994 |
| :---: | ---: | ---: |
| $* * * * *$ | $* * * * *$ | $* * * * * *$ |
| -431. | 0.302 | 0.926 |
| -398. | 0.394 | 0.975 |
| $* * * * *$ | $* * * * *$ | $* * * * *$ |
| 0. | 0.230 | 0.960 |
| $* * * * *$ | $* * * * *$ | $* * * * *$ |
| 398. | 0.399 | 0.990 |
| 431. | 0.301 | 0.922 |
| $* * * * *$ | $* * * * * *$ | $* * * * *$ |
| 800. | 0.462 | 0.956 |

LF., DAY, FALLBACK, PRIMARY
FLT. ND. $=12$ ENV. $=4$ SSS= 3DECC M1 $=$-EDECC DATE: 908
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -787. | 0.619 | 0.932 |
| :---: | ---: | ---: |
| $* * * * *$ | 0.000 | $* * * * *$ |
| -431. | 0.342 | 0.979 |
| -398. | 0.318 | 0.957 |
| $* * * * *$ | 0.000 | $* * * * * *$ |
| 0. | 0.306 | 0.940 |
| $* * * * *$ | 0.000 | $* * * * *$ |
| 398. | 0.322 | 0.970 |
| 431. | 0.343 | 0.979 |
| $* * * * *$ | 0.000 | $* * * * * *$ |
| 788. | 0.641 | 0.963 |

LF, DAY, FALLBACK, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= 3DEGC M1 = -BDEGC DATE: 908
SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATID
$-787$.
*****
-431.
$-398$.

0 .
****휸․․
398.
431.
*****
7 89.
0. 620

0. 340
0.316

0. 204

0. 320
0.340
******
0.642
0.933
*
0.972
0.949
******苗
0.931

0.96 ?
0. 972
******
0. 964

### 2.2 Geometric Resolution (Cont'd)

2.2.3 Smoothed Geometric Resolution - Infrared (3.2.1.1.2.2)
2.2.3.1 Baseline (Orbit Nominal)

The TS SRP is within spec for all measured scan angles.

ATTACHMENTS: TS SRP Curve Orbit Nominal
TS SRP Tables Orbit Nominal

SYSTEM . 12., SRP tarey/TS NORM


SSS=5 . ,M1 = - 8 , DATE: 914 ORBIT
NOMINAL

TS，MID，PRIMARY
FLT．ND．$=12$ ENV．$=4$ SSS＝SDEGC MI $=-$ BDEGC DATE： 914
SUR．DIST．（NM）GRP ACTUAL（NM）SRP RATID

| -750. | 1.928 | 0.813 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | $1.48 日$ | 0.929 |
| -398. | 1.429 | 0.932 |
| 0. | 0.000 | 0.000 |
| 0. | 0.969 | 0.922 |
| 0. | 0.000 | 0.000 |
| 398. | 1.420 | 0.926 |
| 431. | 1.479 | 0.924 |
| 0. | 0.000 | 0.000 |
| 757. | 1.900 | 0.840 |

TS，MID，BACKUP
FLT．ND：$=12$ ENV $=4$ SSS＝SDEGC M1＝－日DEGC DATE： 914
SUR．DIST．（NM）SRP ACTUAL（NM）SRP RATID
$-750$.
0.
$-431$.
－398．
0.
0.
0.
398.
431.
0.
757.

1．7日2
0． 793
0.000

1． 439
1． 382
0． 000
0.935

0． 000
1． 372
1． 430
0.000

1． 860

0． 000
0． 899
0.901

0． 000
0.890
0.000
0.894

0． 893
0.000

0． 823

### 2.2 Geometric Resolution (Cont'd)

2.2.3 Smoothed Geometric Resolution - Infrared (Cont'd)
(3.2.1.1.2.2)

### 2.2.3.2 Acceptance - Vibration

OLS \#12 underwent acceptance level SSS vibration per DMSS-OLS300 with cone cooler $S / N 024$ on $5 / 23 / 91$. The pre-to-post vibration SRP performance is shown on the attached curves and tables.

| ATTACHMENTS: | TS | SRP curve pre-vibration |
| :--- | :--- | :--- |
|  | TS | SRP tables pre-vibration |
|  | TS | SRP curve post-vibration |
|  | TS | SRP tables post-vibration |

SYSTEM 12.,SRP LEM/TS. NORM


SSS=5 , M1 = - 8 ,,DATE:707

## PRE <br> VIB

$0=+Z E O S$
$x=-Z: E O S$


## TS, MID, PRIMARY

FLT. ND. $=12$ ENV. $=4$ SSS= 5DEGC M1= -GDEGC DATE: 707
SUR. DIST. (NM) SRP ACTUAL (NM) SRP RATIU

| -750. | 1.846 | 0.821 |
| ---: | ---: | ---: |
| -600. | 1.786 | 0.907 |
| -431. | 1.489 | 0.930 |
| -398. | 1.429 | 0.932 |
| -200. | 1.105 | 0.922 |
| 0. | 0.971 | 0.925 |
| 200. | 1.106 | 0.922 |
| 398. | 1.419 | 0.925 |
| 431. | 1.482 | 0.926 |
| 601. | 1.766 | 0.896 |
| 757. | 1.894 | 0.838 |

TS, MID, BACKUP

FLLT. ND. $=12$ ENV. $=4$ SSS $=$ SDEGC M1 = -GDEGC DATE: 707
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATIU

| -750. | 1.793 | 0.797 |
| ---: | ---: | ---: |
| -600. | 1.726 | 0.876 |
| -431. | 1.433 | 0.895 |
| -398. | 1.375 | 0.897 |
| -200. | 1.063 | 0.886 |
| 0. | 0.934 | 0.889 |
| 200. | 1.063 | 0.886 |
| 398. | 1.365 | 0.890 |
| 431. | 1.426 | 0.891 |
| 601. | 1.704 | 0.865 |
| 757. | 1.845 | 0.816 |

SYSTEM 12.,SRP /TS NORM


## TS. MID. PRIMARY

FLT. NO. $=12$ ENV. $=4$ SSS= 5DEGC M1 $=-8 D E G C$ DATE: 914
SUR.DIST. (NM) SRP ACTUAL (NM) SRP RATIU
$-750$.

1. 829
2. 813
3. 

-431.
-39日.
0.000
0. 000

1. 488
0.929
2. 429
0.000
0.932
3. 

0.968
0.000
0.000
0.922
0.
0.000

1. 420
2. 479
0.000
3. 900
0.926
0.924
0.000
4. 840

TS, MID, BACKUP

FLT. ND: $=12$ ENV. $=4$ SSS= SDEGC M1 = -EDEGC DATE: 914
SUR.DIST. (NM) SRP ACTUAL (NM) SRP RATID
-750.
0.
-431.
$-398$.
0.
0.
0.
398.
431.
0.
757.

1. 782
0.000
2. 439
3. 382
0.000
0.935
0.000
4. 372
5. 430
6. 000
7. 860
8. 793
9. 000
10. 899
11. 901
0.000
0.890
12. 000
13. 894
0.893
14. 000
15. 823

### 2.2 Geometric Resolution (Cont'd)

2.2.3 Smoothed Geometric Resolution - Infrared (3.2.1.1.2.2)

### 2.2.3.3 Acceptance - Thermal Vacuum

The TS SRP is within spec for the extremes of hot and cold Thermal Vacuum limit testing.

ATTACHMENTS: TS SRP Curve Hot Limits<br>TS SRP Tables Hot Limits<br>TS SRP Curve Cold Limits<br>TS SRP Tables Cold Limits

SYSTEM 12.,SRP NORM
 SSS=7 , M1 = 12 , DATE: 985
 0.4
$\otimes$

```
FLT. ND. = 12 ENV. = 4 5SS= 7DEGC M1= 12DEGC DATE: POS
```

SUR, DIST. (NM) SRP ACTUAL (NM) GRP RATID

| -750. | 1.833 | 0.815 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 1.493 | 0.932 |
| -398. | 1.433 | 0.934 |
| 0. | 0.000 | 0.000 |
| 0. | 0.971 | 0.925 |
| 0. | 0.000 | 0.000 |
| 398. | 1.421 | 0.927 |
| 431. | 1.483 | 0.926 |
| 0. | 0.000 | 0.000 |
| 757. | 1.916 | 0.848 |

TS, MID, BACKUP
FLT. NO. $=12$ ENV. $=4$ SSS= 7DEGC M1 = 12DEGC DATE: 905
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATIU
$-750$.
0.
-431 . -398.
0.
0.
0.
398.
431.
0.
757.

1. 789
0.000
2. 446
3. 388
0.000
0.938
0.000
4. 374
5. 435
0.000
6. 876
0.795
0.000
0.903
0.905
7. 000
8. 894
0.000
0.896
9. 896
10. 000
11. 830

SYSTEM 12.,SRP NORH/TS NOR



TS, MID, PRIMARY
FLT. ND. $=12$ ENV. $=4$ SSS= 3DEGC M1=-EDEGC DATE: 9OE
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATIU

| -750. | 1.818 | 0.809 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 1.480 | 0.925 |
| -398. | 1.424 | $0.92 日$ |
| 0. | 0.000 | 0.000 |
| 0. | 0.968 | 0.922 |
| 0. | 0.000 | 0.000 |
| 398. | 1.415 | 0.923 |
| 431. | 1.477 | 0.922 |
| 0. | 0.000 | 0.000 |
| 757. | 1.893 | 0.837 |

TS. MID, BACKUP

FLT. ND. $=12$ ENV. $=4$ SSS= 3DEGC M1 $=$-EDEGC DATE: FOB
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATID
$-750$.
0.
$-431$.
$-398$
0.
0.
0.
398.
431.
0.
757.

1. 764
2. 000
3. 423
4. 369
0.000
0.929
5. 000
6. 359
7. 419
0.000
8. 845
9. 784
0.000
10. 889
11. 892
12. 000
13. 885
14. 000
15. 886
16. 886
17. 000
18. 816

### 2.2 Geometric Resolution (Cont'd)

2.2.4 Smoothed Geometric Resolution - Davtime Visual (3.2.1.1.2.2)
2.2.4.] Baseline (Orbit Nominal)

The LS Day SRP is within spec limits at Orbit Nominal conditions.

ATTACHMENTS: LS Day SRP Curve - Orbit Nominal<br>LS Day SRP Tables - Orbit Nominal

SYSTEM 12.,SRP LS DAY/
SSS $=5 \ldots, M 1=-8$, DATE $: 912$


## LS, DAY, NGRMAL, PRIMARY

| FLT. NO. $=12$ |  | 4 SSS= 5 | M1 $=-8 \mathrm{PEGC}$ | DATE: |
| :---: | :---: | :---: | :---: | :---: |
| SUR. DIST. (NM) | SRP | ACTUAL (NM) | SRP RATIO |  |
| -800. |  | 1. 537 | 0. 654 |  |
| 0. |  | 0. 000 | 0. 000 |  |
| -431. |  | 1. 394 | 0.871 |  |
| -398. |  | 1. 350 | 0. 880 |  |
| 0. |  | 0. 000 | 0. 000 |  |
| 0. |  | 0.927 | 0. 883 |  |
| 0. |  | 0.000 | 0. 000 |  |
| 396. |  | 1. 347 | 0.878 |  |
| 431. |  | 1. 403 | 0. 876 |  |
| 0. |  | 0. 000 | 0.000 |  |
| 800. |  | 1. 542 | 0.656 |  |

LS, DAY, NORMAL, BACKUP
FLT. NO. $=12$ ENU. $=4$ SSS= 5DEGC M1=-BDEGC DATE: 912
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -800. | 1.547 | 0.658 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 1.403 | 0.876 |
| -398. | 1.359 | 0.886 |
| 0. | 0.000 | 0.000 |
| 0. | 0.933 | 0.888 |
| 0. | 0.000 | 0.000 |
| 398. | 1.355 | 0.884 |
| 431. | 1.412 | 0.882 |
| 0. | 0.000 | 0.000 |
| 800. | 1.552 | 0.660 |

2.2 Geometric Resolution (Cont'd)
2.2.4 Smoothed Geometric Resolution - Daytime Visual (Cont'd)
(3.2.1.1.2.2)

### 2.2.4.2 Acceptance - Vibration

The OLS \#12 SSS underwent acceptance level SSS vibrations per DMSS-0LS-300 on May 23, 1991. The LS Day SRP is within specification both before and after SSS vibration. No vibration-related changes in SRP were observed.

ATTACHMENTS: | LS Day SRP Curve | Pre-Vibration |  |
| :--- | :--- | :--- |
|  | LS Day SRP Tables | Pre-Vibration |
|  | LS Day SRP Curve | Post-Vibration |
|  | LS Day SRP Table | Post-Vibration |

SYSTEM 12, SRP. LS DAY/
$S S S=23 ., M 1=24$, DATE:527.

$\stackrel{\otimes}{\otimes}$

LS, DAY, NORMAL, PR IMARY


LS, DAY, NDRMAL, BACKUP
FLT. ND. $=12$ ENV. $=2 \cdot g 95=23 D E G C$ M1= 24DEGC DATE: 527
SUR.DIST. (NM) SRP ACTUAL.(NM) SRP RATID

| -800. | 1.573 | 0.670 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 1.425 | 0.890 |
| -398. | 1.378 | 0.898 |
| 0. | 0.000 | 0.000 |
| 0. | 0.948 | 0.902 |
| 0. | 0.000 | 0.000 |
| 398. | 1.378 | 0.898 |
| 431. | 1.429 | 0.892 |
| 0. | 0.000 | 0.000 |
| 800. | 1.586 | 0.675 |

$2-80$

SYSTEM 12.,SRP LS DAY/FM

$\mathrm{SSS}=23 ., \mathrm{MI}=24 .$, DATE: 814

0.8
$\stackrel{\nabla}{\Delta}$
$\stackrel{\otimes}{\bullet}$

| 0.0 | 2.0 | 4.0 <br> $S D / 10 \square-(N M)$ <br> $2-81$ | 6.0 | 8.0 |
| :---: | :---: | :---: | :---: | :---: |

LS, DAY, NGRMAL, PR IMARY
FLT. NO. $=12$ ENV. $=2$ SSSㅍ 23DEGC M1=24DECC DATE: 114
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -800. | 1.562 | 0.665 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 1.416 | 0.884 |
| -398. | 1.371 | 0.894 |
| 0. | 0.000 | 0.000 |
| 0. | 0.942 | 0.897 |
| 0. | 0.000 | 0.000 |
| 398. | 1.368 | 0.892 |
| 431. | 1.422 | 0.888 |
| 0. | 0.000 | 0.000 |
| 800. | 1.577 | 0.671 |

LS, DAY, NORMAL, BACKUP
FLT. ND. $=12$ ENU. $=2$ SSS= 23DEGC M1=24DEGC DATE: 814 SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -800. | 1.573 | 0.669 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 1.425 | 0.890 |
| -398. | 1.380 | 0.899 |
| 0. | 0.000 | 0.000 |
| 0. | 0.948 | 0.903 |
| 0. | 0.000 | 0.000 |
| 398. | 1.377 | 0.898 |
| 431. | 0.431 | 0.893 |
| 0. | 1.588 | 0.000 |
| 800. |  | 0.676 |

### 2.2 Geometric Resolution (Cont'd)

2.2.4 Smoothed Geometric Resolution - Daytime Visual (Cont'd) (3.2.1.1.2.2)

### 2.2.4.3 Acceptance - Thermal Vacuum

The LS Day SRP is within specification allowance over the entire range of temperatures.

ATTACHMENTS: LS Day SRP Curve Hot Limits<br>LS Day SRP Tables Hot Limits<br>LS Day SRP Curve Cold Limits<br>LS Day SRP Tables Cold Limits

SYSTEM 12 ,SRP LS DAY/T


SSS=7, , M1 = 12 , DATE: 903

## LS, DAY, NDRMAL, PR IMARY

FLT. ND. $=12$ ENV. $=4$ SSS= 7DEGC M1 = 1 PDEGC DATE: 903
SUR.DIST. (NM) SRP ACTUAL (NM) SRP RATIO

| -800. | 1.543 | 0.656 |
| ---: | ---: | ---: |
| 0. | 0.000 | 0.000 |
| -431. | 1.398 | 0.873 |
| -398. | 1.355 | 0.883 |
| 0. | 0.000 | 0.000 |
| 0. | 0.929 | 0.885 |
| 0. | 0.000 | 0.000 |
| 398. | 1.350 | 0.880 |
| 431. | 1.405 | 0.677 |
| 0. | 0.000 | 0.000 |
| 800 | 1.550 | 0.660 |

LS, DAY, NDRMAL, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= 7DEGC M1= 12DEGC DATE: 903
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

- 800. 

0. 

-431 .
$-398$.
0.
0.
0.
398.
431.
0.
800.

1. 553
2. 000
3. 40 g
4. 364
5. 000
0.935
0.000
6. 359
7. 414
8. 000
9. 561
10. 661
11. 000
12. 879
13. 889
0.000
14. 890
15. 000
16. 886
17. 883
18. 000
19. 664

SYSTEM .12.,SRP. LS DAY/I


## LS, DAY, NDRMAL, PRIMARY

FLT. ND. $=12$ ENV. $=4$ SSS= 3DEGC M1 $4=-$ GDEGC DATE: 908
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATID

| -800. | 1.533 | 0.652 |
| :---: | ---: | ---: |
| $* * * * *$ | $* * * * *$ | $* * * * * *$ |
| -431. | 1.393 | 0.870 |
| -398. | 1.348 | 0.878 |
| $* * * * *$ | $* * * * *$ | $* * * * * *$ |
| 0. | 0.925 | 0.881 |
| $* * * * *$ | $* * * * *$ | $* * * * *$ |
| 398. | 1.344 | 0.876 |
| 431. | 1.398 | 0.873 |
| $* * * * *$ | $* * * * *$ | $* * * * *$ |
| 800. | 1.537 | 0.654 |

LS, DAY, NDRMAL, BACKUP
FLT. ND.쿠 12 ENV. $=-4$ - $55 S=3 D E G C$. M1= -BDEGC DATE: - 908
SUR. DEEE (NM) SRP ACTUAL(NM) SRP RATIO

| -800; - | 1. 543 | 0.657 |
| :---: | :---: | :---: |
| ****** | 10. 897 | ****** |
| -431. | 1. 402 | 0. 876 |
| -398. | 1. 357 | 0. 884 |
| ***** | 10. 897 | ****** |
| 0 | 0. 931 | 0. 886 |
| \#****** | 10. 897 |  |
| 378. | 1. 353 | 0.882 |
| 431. | 1.408 | 0. 877 |
| ***** | 10. 897 | ********** |
| 800. | 1. 548 | 0.659 |

### 2.2 Geometric Resolution (Cont'd)

2.2.5 Smoothed Geometric Resolution - Nighttime Visual (3.2.1.1.2.2)

LS Night SRP routinely is deliberately adjusted to be close to spec limit during system integration in order to optimize PMT signal-to-noise ratio, at the expense of SRP margin.

### 2.2.5.1 Baseline (Orbit Nominal)

The LS Night SRP is within spec for all measured scan angles

ATTACHMENTS: LS Night SRP Curve - Orbit Nominal<br>LS Night SRP Table - Orbit Nominal

SYSTEM 12 ,SRP LS NITE . ,SSS=5 . .,M1 = - 8 ,,DRTE: 913


L5, NITE, NORMAL, PR IMARY

| FLT. ND. $=12$ | = | 4 S5S= 5 | M1 = - ${ }^{\text {EDEGC }}$ |
| :---: | :---: | :---: | :---: |
| SUR. DIST. (NM) | SRP | ACTUAL (NM) | SRP RATID |
| -799. |  | 2. 893 | 0.966 |
| 0. |  | 0. 000 | 0. 000 |
| -430. |  | 1. 624 | 0.972 |
| -397. |  | 2. 187 | 0.918 |
| 0. |  | 0. 000 | 0. 000 |
| 0. |  | 1. 193 | 0.884 |
| 0. |  | 0. 000 | 0. 000 |
| 397. |  | 2. 088 | 0.876 |
| 430. |  | 1. 622 | 0.971 |
| 0. |  | 0. 000 | 0.000 |
| 801. |  | 2. 881 | 0. 958 |

LS, NITE, NDRMAL, BACKUP
FLT. ND. $=12$ ENV. $=4$ SSS= 5DEGC M1= -BDEGC DATE: 913
SUR. DIST. (NM) SRP ACTUAL(NM) SRP RATIO
-799.
2. 919
0. 975
0.
-430.
-397.
0.
. 000

1. 639
2. 209
3. 000
4. 205
5. 000
6. 108
7. 637
8. 000
9. 000
10. 981
11. 927
12. 000
13. 892
14. 000
15. 885
16. 980
17. 907
18. 000
0.967

### 2.2 Geometric Resolution (Cont'd)

### 2.2.5 Smoothed Geometric Resolution - Nighttime <br> Visual (Cont'd) (3.1.2.2)

### 2.2.5.2 Acceptance - Vibration

The OLS \#12 SSS underwent Acceptance level SSS vibration per DMSS-OLS-300 on May 23, 1991. The pre-to-post vibration SRP performance is shown on the attached curves and tables.

ATTACHMENTS: LS Night SRP curve pre-vibration | LS Night SRP tables pre-vibration |  |
| ---: | :--- |
|  | LS Night SRP curve post-vibration |
|  | LS Night SRP tables post-vibration |



## LS: NITE, NORMAL, PRIMARY

FLT. ND. $=12$ ENV. $=2$ SSS= 23DEGC M1=23DEGC DATE: 810 SUR.DIST. (NM) SRP ACTUAL (NM) SRP RATIU

| -799. | 2.806 | 0.937 |
| ---: | ---: | ---: |
| -601. | 2.022 | 0.965 |
| -430. | 1.595 | 0.954 |
| -397. | 2.148 | 0.902 |
| -200. | 1.417 | 0.874 |
| 0. | 1.174 | 0.870 |
| 200. | 1.386 | 0.855 |
| 397. | 2.053 | 0.862 |
| 430. | 1.596 | 0.955 |
| 600 | 2.018 | 0.964 |
| 801. | 2.850 | 0.948 |

LS, NITE, NORMAL, BACKUP
FLT. ND. $=12$ ENV. $=2$ SSS= 23DEGC M1 = 23DEGC DATE: B10
SUR. BIST. (NM) SRP ACTUAL(NM) SRP RATID

| -799. | 2.829 | 0.944 |
| ---: | ---: | ---: |
| -601. | 2.039 | 0.973 |
| -430. | 1.607 | 0.962 |
| -397. | 2.165 | 0.909 |
| -200. | 1.430 | 0.892 |
| 0. | 1.184 | 0.877 |
| 200. | 1.398 | 0.863 |
| 397. | 2.070 | 0.869 |
| 430. | 1.608 | 0.962 |
| 600. | 2.036 | 0.972 |
| 801. | 2.871 | 0.955 |

SYSTEM 12.,SRP. LS NITE . , SSS=23.,M1=23 , DRTE: 815

$$
\begin{gathered}
\mathrm{N} \\
\mathrm{~N} \\
\\
\hdashline \begin{array}{c}
\text { SPEC LIMIT } \\
0=+Z \operatorname{ESS} \\
x=-Z E O S
\end{array}
\end{gathered}
$$

Post VIB
"

## L.S. NITE, NORMAL, PRIMARY

| FLT. NO. $=12$ | . $=$ | 2 SSS= 23 | M1 = 23DEGC | DATE: |
| :---: | :---: | :---: | :---: | :---: |
| SUR. DIST. (NM) | SRP | ACTUAL (NM) | SRP RATIO |  |
| -799. |  | 2. 802 | 0.936 |  |
| 0. |  | 0.000 | 0.000 |  |
| -430. |  | 1. 595 | 0. 955 |  |
| -397. |  | 2. 146 | 0. 901 |  |
| 0. |  | 0. 000 | 0.000 |  |
| 0. |  | 8. 175 | 0.870 |  |
| 0. |  | 0. 000 | 0.000 |  |
| 397. |  | 2. 053 | 0. 862 |  |
| 430. |  | 1. 593 | 0.953 |  |
| 0. |  | 0. 000 | 0. 000 |  |
| 801. |  | 2. 839 | 0. 944 |  |

LS, NITE, NORMAL, BACKUP
FLT. ND. $=12$ ENV. $=2555=23 D E G C$ M1= 23DEGC DATE: 815
SUR.DIST. (NM) SRP ACTUAL(NM) SRP RATIO
-799.
0.
$-430$.
-397.
0.
0.
0.
397.
430.
0.
801.
2. 825
0. 000

1. 608
2. 163
3. 000
4. 195
5. 000
6. 070
7. 605
8. 000
9. 862
10. 944
11. 000
12. 962
13. 908
14. 000
15. 878
16. 000
17. 869
18. 961
19. 000
20. 951

### 2.2 Geometric Resolution (Cont'd)

2.2.5 Smoothed Geometric Resolution - Nighttime

Visual (Cont'd) (3.1.2.2)

### 2.2.5.3 Acceptance - Thermal Vacuum

The LS Night SRP is within specification limits over the Acceptance temperature range. No temperature-related changes in SRP over the Acceptance temperature range were observed.

ATTACHMENTS: LS Night SRP Curve Hot Limits
LS Night SRP Tables Hot Limits
LS Night SRP Curve Cold Limits
LS Night SRP Tables Cold Limits

SYSTEM 12, SRP LS NITE ..,SSS=7. . M1 = 12 , DRTE: 903


LS, NITE, NORMAL, PRIMARY


LS. NITE, NORMAL, BACKUP
FLT MD. $=12$ ENY $=4$ SSS= 7DEGC MI = 12DEGC DATE: 903
GUR. DIST. (NH) SRP ACTUAL (NM) SRP RATIO

| -797 | 2.904 | 0.970 |
| ---: | ---: | ---: |
| 0 | 0.000 | 0.000 |
| -470. | 1.036 | 0.979 |
| -397 | 2.208 | 0.927 |
| 0. | 0.000 | 0.000 |
| 0. | 1.204 | 0.891 |
| 0. | 0.000 | 0.000 |
| 397 | 2.108 | 0.895 |
| 0. | 1.647 | 0.985 |
| 001 | 0.000 | 0.000 |
| 0. | 2.884 | 0.959 |

SYSTEM. 12 ,SRP. LS NITE .., SSS=3 . .,M1=-8 , DRTE: 908


## LS，NI TE，NORMAL，PRIMARY



LS，NITE，NORMAL，BACKUP
FLT．ND．$=12$ ENV．$=4$ SSS＝3DEGC M1＝－日DEGC DATE：9OB
SUR．DIST．（NM）SRP ACTUAL（NM）SRP RATID
$-799$.
0.
$-430$.
$-397$.
0.
0.
0.
397.
430.
0.
801.

2． 958
0． 000
1． 656
2． 228
0． 000
1． 217
0． 000
2． 126
1． 647
0． 000
2． 942

0． 988
0.000
0.991
0.935
0.000
0.903

0． 000
0.893

0．986
0.000

0．97日

### 2.2 Geometric Resolution (Cont'd)

### 2.2.6 Data Sampling (3.2.1.1.2.3)

The sampling frequency ratios for all modes of the 5D-3 OLS satisfy the specification requirements. The calculations are contained in the 50-3 OLS System Summary Report. The results are summarized below. The worst-case sampling frequency ratio for each mode is given.

MODE
LF Day - Normal
LF Day - Fallback
LS \& TS Day - Normal
LS Night - Normal
TF - Normal
TF Fallback - Normal Side of scan 3.28
IF Fallback - Abnormal Side of Scan
2.50
3.44

SAMPLING FREO. RATIO (Spec: > 2.4)
2.58
2.61
2.66
3.28

### 2.3 Geometric Accuracy (3.2.1.1.3.1, 3.2.1.1.3.2, 3.2.1.1.3.3)

The alignment and synchronization of the SSS determine the Geometric Accuracy. Geometric accuracy is specified in 3 categories (Repeatability, Stability \& Fixed Errors). Within these categories, accuracy is further specified for Along Track, Along Scan (Stored Data), Along Scan (Direct Data), and Along Scan (with digital delphi generation, i.e., the Encoder Simulator locked mode).

There was an observed shift in alignment between the extremes of M1 temperature of $-8^{\circ} \mathrm{C}$ and $+12^{\circ} \mathrm{C}$ on the order of 0.16 milliradians or less for all channels. There was also an observed shift in synchronization in all modes in OLS \#12 of approximately 0.1 milliradians or less between M1 temperature extremes of $-8^{\circ} \mathrm{C}$ and $+12^{\circ}$.

The Repeatability error is calculated using the rms difference of the measured alignment/synchronization (as a function of the variation of Ml temperature on orbit) from the hot-to-cold average values of alignment/ synchronization. The total Repeatability contribution for synchronization is determined by rss'ing the wow-flutter error (as measured in test $6 \times 11 . S T$ ) and the jitter error (which was negligible on OLS \#12) with the repeatability shift between the Hot and Cold Limits.

The Primary Alignment Reference Axes (REFPLN) are calculated in a computer program using HRD \& T channel Alignment and Sync data from several Orbit Nominal tests. The data determine a best-fit alignment with respect to the Interface Mounting Axes. The REFPLN Alignment and REFPLN Synchronization curves are included here. The remainder of the Alignment and Synchronization data for the OLS \#12 SSS are in BVS 2693, Vol. III of this Acceptance Test Report.

The 5D-3 System has a Fallback mode utilizing an encoder control track and Encoder Simulator. The spec limits and measured results are shown in Table 2.3-2.

ATTACHMENTS: OLS \#12 REFPLN ALIGNMENT
OLS \#12 REFPLN SYNCHRONIZATION
OLS \#12 ALIGN/SYNC vs SPEC, Table 2.3-1
OLS \#12 Encoder Simulator Sync, Table 2.3-2

## OLS\#I2 REFPLN ALIGNMENT



OLS\#12 REFPLN SYNCHRONIZATION


Table 2.3-1

## OLS \#12 ALIGN/SYNC vs. SPEC

all numbers in milliradians


Table 2.3-2
ALONG-SCAN GEOMETRIC ACCURACY WITH ENCODER SIMULATOR

|  | Stored | Direct Fine | Direct Smooth |
| :---: | :---: | :---: | :---: |
| Repeatability-Spec, mrad Measured | 1.0 | 1.1 | 2.2 |
|  | 0.08 | 0.08 | 0.10 |
| Stability - Spec, mrad Measured | 0.50 | 0.50 | 0.50 |
|  | 0.20 | 0.20* | 0.20* |
| Fixed - Spec, mrad ${ }_{\text {Measured }}$ | 10.0 | 10.0 | 10.0 |
|  | 0.88 | 0.88* | 0.88* |
| Total - Spec, mrad Calculated | 11.1 | 11.2 | 12.3 |
|  | 1.10 | 1.10 | 1.10 |

*Inferred from stored number

### 2.4 RADIOMETRIC ACCURACY

### 2.4.1 T Channel Radiometric Accuracy (3.2.1.1.4.1 a,b,c)

The overall one sigma accuracy of the OLS \#12 T Channel DC response is $0.73^{\circ} \mathrm{K}$ compared to a $1.1^{\circ} \mathrm{K}$ spec and therefore OLS \#12 does meet this specification requirement.

Table 2.4.1-1 presents the overall summary of performance, which is discussed more fully in sections 2.4.1.1, 2.4.1.2 and 2.4.1.3.

ATTACHMENTS:
Table 2.4.1-1
Table 2.4.1-2
Table 2.4.1-3
Table 2.4.1-4
Table 2.4.1-5
Table 2.4.1-6
Figure 2.4.1-1
Figure 2.4.1-2
Figure 2.4.1-3
Figure 2.4.1-4
Figure 2.4.1-5
Figure 2.4.1-6
Figure 2.4.1-7
Figure 2.4.1-8
Figure 2.4.1-9
Figure 2.4.1-10

Overall Contributors
$210^{\circ}$ to $310^{\circ} \mathrm{K}$ Best Straight Line Calibrations
T DC Response Compilation of Test Runs
BSL Equation T Right, Run \#ll
BSL Equation T Mid, Run \#ll
BSL Equation T Left, Run \#11
T DC Response Plots, Run \#l - Primary
T DC Response Plots, Run \#1 - Redundant
T DC Response Plots, Run \#2 - Primary
T DC Response Plots, Run \#2 - Redundant
T DC Response Plots, Run \#9 - Primary
T DC Response Plots, Run \#9 - Redundant
I DC Response Plots, Run \#10 - Primary
T DC Response Plots, Run \#10 - Redundant
T DC Response Plots, Run \#ll - Primary
T DC Response Plots, Run \#Il - Redundant

TABLE 2.4.1-1
OLS \#12
OVERALL CONTRIBUTORS TO T-CHANNEL RADIOMETRIC ACCURACY

| SPECIFICATION PARA. 3.1.4.1 | RMS <br> DEVIATION ( ${ }^{\circ} \mathrm{K}$ ) | SPECIFICATION <br> MAX <br> ONE SIGMA ERROR ( ${ }^{\circ} \mathrm{K}$ ) |
| :--- | :---: | :---: |
| a) Repeatability (<l day) | 0.262 | 0.42 |
| b) Stability (>1 day) | 0.62 | 0.80 |
| c) Fixed Deviations | 0.29 | 0.60 |
| TOTAL (RSS) ACCURACY | 0.73 | 1.10 |

## Discussion of T DC Response Test and Overview

The measurement of T DC Response is accomplished in the thermal vacuum chamber because the $T$ detector must be cooled to its operating temperature of near $110^{\circ} \mathrm{K}$. Two controlled blackbody targets variable in temperature over the dynamic range of $190^{\circ} \mathrm{K}$ to $310^{\circ} \mathrm{K}$ provide the absolute infrared radiance reference. The temperature of each target is measured by five thermocouples which have been calibrated against a precision platinum resistance temperature standard. The five thermocouples provide target temperature gradient information to indicate target stability as well as the capability to average the five for the reference target temperature. T Channel DC Response consists of comparing the average target temperature to the channel output voltage at the input to the $A / D$ converters for several target temperatures.

Eleven vacuum runs were made on OLS \#12. The T DC Response data from vacuum runs ( 1 through il) is compiled in Table 2.4.1-2 and -3, which show the equipment temperature environments and characteristics of each run. At the completion of testing, the $T$ channel gain pots were readjusted to set $\mathrm{T}_{\text {rgt }}=$ $\mathrm{T}_{\text {LFT }}=4$. The column headed "Data Points" indicates how many target temperatures were in that run. The characteristics of the response itself are indicated in three columns each for T LFT, T MID and T RGT. The compared channel response to target temperature results in a difference for each data point. This difference is corrected for Ml Temperature so that all data for a given run reflect the same $M 1$ temperature and the expected shaper circuit difference is subtracted. In this form the difference data for a given run should ideally be a linear function to target temperature. A linear leastsquares fit to the corrected data is used to determine the equation of the best straight line (BSL). In Table 2.4.1-2 the slope error, the $210^{\circ} \mathrm{K}$ ordinate and the RMS data fit values for these different BSL's are listed in the columns headed Slope, Ordinate at $210^{\circ} \mathrm{K}$, and RMS Dev.

$$
2-110
$$

In order to distinguish between gain and bias type effects caused by environment, the $310^{\circ} \mathrm{K}$ value that results when the $210^{\circ} \mathrm{K}$ BSL value is forced to zero difference is also calculated. Table 2.4.1-3 shows the pre-shaper \% gain Difference from Nominal, the Bias Diff. from Nominal (at $190^{\circ} \mathrm{K}$ ) where the pre-shaper Gain is forced to nominal, and the Ml temperature coefficient (K factor); in three columns each for TRGT, TMID and TLEFT.

Tables 2.4.1-4, -5 and -6 show the STS computer processed and gain compensated T DC Response Data of the final "Orbit Nominal" Run (Run \#ll) for TRGT, TMID, and TLEFT respectively. The fourth line down in the body of the Best Straight Line Equation, "RMS Deviation", is the RMS error (for $210^{\circ} \mathrm{K}$ to $310^{\circ} \mathrm{K}$ ) of the data points fitted to the best straight line. "FP" is T Fine Primary; "FB" is T Fine, Backup (Redundant); "SP" is T Smoothed, Primary; "SB" is T Smoothed, Backup (redundant). The SP and SB are not used for $T$ Right Only or for T Left Only in the Primary or Redundant normal modes; these are utilized only in the Fallback (slightly degraded) modes of operation. SP and SB are applicable to T Mid in normal Primary or Redundant modes.

Runs 1 and 2 together indicate the changes which accompany operation over the foreoptics cold-to-warm temperature range as indicated by M1 temperature.

Runs 3 and 4 together indicate the magnitude of the variation over the extremes of SSS temperature, $\left(+11^{\circ}\right.$ to $\left.-3^{\circ} \mathrm{C}\right)$; when compared to the $+3.2^{\circ} \mathrm{C}$ and $+4.6^{\circ} \mathrm{C}$ SSS run pairs with the corresponding M1 temperatures, (Runs 2 and 1 , respectively). However, changes between these runs are not only due to SSS temperature differences. The PSU, which contains the shaper networks was varied in temperature along with the SSS, from a 10 w of $0.5^{\circ} \mathrm{C}$ to a high of $+38.2^{\circ} \mathrm{C}$.

Figures 2.4.1-1 through 2.4.1-10 inclusive show, for Runs No. l through No. 11, (respectively), the test data points for Targets 1 and 2 and the BSL
plots for Right, Mid, and Left. (No BSL data plot was obtained for Runs 3 through 8 since they contain only 2 points).

The OLS \#12 average Ml coefficient (coupling factor) measured for the final run (\#11) was $0.207^{\circ} \mathrm{K}$ at $210^{\circ} \mathrm{K}$ scene per $1^{\circ} \mathrm{C}$ temperature change of Ml . The lower the Ml coefficient vaiue, the better the performance. The T Clamp leakage was -0.027\% T LEFT and -. 104\% RIGHT.

The Orbit Nominal BSL differences (from Table 2.4.1-4,5 and 6) between Fine Primary and Fine Backup are small, the largest being $0.38^{\circ} \mathrm{K}$ for T RGT, at the $310^{\circ} \mathrm{K}$ end. In the Smooth Primary and Backup modes, T RGT differs by $0.40^{\circ} \mathrm{K}$ (at $310^{\circ} \mathrm{K}$ ).

The difference between T LEFT and T RIGHT segments calibration (from Tables 2.4.1.3-4 and 5 ) is $0.56^{\circ} \mathrm{K}$ worst-case, vs. a spec 1 imit of $1.0^{\circ} \mathrm{K}$.
$210^{\circ}$ TO $310^{\circ} \mathrm{K}$ BEST STRAIGHT LINE CALCULATIONS

| date | RUN\# | $\begin{aligned} & R / L \\ & T G \end{aligned}$ | TL | $\begin{gathered} \text { \# OF } \\ \text { DDTA } \\ \text { POINTS } \end{gathered}$ | temperature ${ }^{\text {c }}$ |  |  | T RIGHT |  |  | T MID |  |  | iteft |  |  | COHMENTS |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  | Sss | M1 | PSU | SLOPE | ORD. <br> $2210^{\circ}$ | $\begin{aligned} & \text { RHS } \\ & \text { DEV } \end{aligned}$ | SLOPE | ORD. <br> - $210^{\circ}$ | $\begin{aligned} & \text { RHS } \\ & \text { DEV } \end{aligned}$ | SLOPE | ORD. <br> a $210^{\circ}$ | $\begin{aligned} & \text { RHS } \\ & \text { DEV } \end{aligned}$ |  |
| $\begin{aligned} & \text { TDCRH3A } \\ & \text { 07/08/91 } \\ & \hline \end{aligned}$ | 1 | 5/6 | 13 | 14 | 4.6 | -8.1 | 23.8 | . 0030 | . 16 | . 09 | -. 0056 | . 87 | . 05 | -. 0011 | . 44 | . 08 | COLD OPTIC LIMIT |
| $\begin{aligned} & \text { TDCRM3A } \\ & 07 / 13 / 91 \\ & \hline \end{aligned}$ | 2 | 5/6 | 9 | 8 | 3.2 | 12.4 | 23.5 | $-.0063$ | . 43 | . 05 | -. 0160 | 1.25 | . 07 | -. 0134 | . 88 | . 05 | HOT OPTIG Limit |
| $\begin{aligned} & \mathbf{T 1 2 1 1 2 3 1 8} \\ & 07 / 14 / 91 \end{aligned}$ | 3 | $5 / 6$ | 9 | 2 | 10.9 | 15.3 | 38.2 | $\cdot .0096$ | . 12 | . 00 | -. 0194 | . 94 | . 00 | $\cdot .0176$ | . 57 | . 00 | HOT SOAK \#1 |
| $\begin{aligned} & 712112318 \\ & 07 / 18 / 91 \end{aligned}$ | 4 | 5/6 | 13 | 2 | -2.4 | -10.8 | 0.5 | $\bigcirc .0008$ | . 63 | . 00 | -. 0072 | 1.40 | . 00 | $\cdot .0042$ | 1.01 | . 00 | COLD SOAK \#1 |
| 07/23/91 |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  | vaclum break PHT FAILURE |
| $\begin{array}{r} \text { T12172318 } \\ 08 / 25 / 91 \\ \hline \end{array}$ | 5 | $5 / 6$ | 9 | 2 | 10.8 | 16.0 | 37.9 | -. 0199 | . 50 | . 00 | -. 0269 | -1.34 | . 00 | -. 0208 | . 75 | . 00 | $\begin{aligned} & \text { HOT SOAK \#1 } \\ & \text { REPEAT } \end{aligned}$ |
| T121T231日 08/26/91 | 6 | 5/6 | 13 | 2 | -3.9 | -10.4 | 0.6 | -. 0066 | . 90 | . 00 | -. 0110 | 1.53 | . 00 | -. 0032 | 1.02 | . 00 | COLD SOAK \#1 REPEAT |
| T12172318 08/28/91 | 7 | 5/6 | 9 | 2 | 11.0 | 15.1 | 38.0 | -. 0223 | . 71 | . 00 | $-.0275$ | 1.37 | . 00 | $\cdot .0214$ | . 75 | . 00 | HOT SOAK \#2 |
| $\begin{aligned} & 102172318 \\ & 08 / 30 / 91 \end{aligned}$ | 8 | 5/6 | 13 | 2 | -2.7 | -10.6 | 0.5 | -.0088 | 1.03 | . 00 | $\cdots .0120$ | 1.57 | . 00 | $-.0043$ | 1.04 | . 00 | COLD SOAK \#2 |
| TDCRM3C 09/02/91 | 9 | 5/6 | 9 | 7 | 5.6 | 12.2 | 33.4 | $\cdot .0211$ | . 79 | . 04 | -. 0273 | 1.51 | . 08 | $\bigcirc .0197$ | . 94 | . 02 | Hot limit |
| TOCRH3B <br> 09/08/91 | 10 | 5/6 | 13 | 7 | 2.4 | -7.4 | 4.8 | -. 0138 | . 98 | . 02 | -. 0179 | 1.54 | . 07 | $\cdot .0111$ | 1.11 | . 04 | cold limit |
| TDCRM3C <br> 09/13/91 | 11 | 5/6 | 13 | 18 | 4.8 | -7.8 | 23.6 | $\cdot .0185$ | . 97 | . 05 | $\cdot .0222$ | 1.50 | . 07 | $\cdot .0148$ | 1.03 | . 05 | homikal |


|  | RGT/LFT |  |  |  | TEMPERATURE * ${ }^{\text {c }}$ |  |  | I RIGHT |  |  | T MID |  |  | T LEFT |  |  | COMAEHTS |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| DATE | $\underset{\substack{\text { RUH } \\ \#}}{ }$ | TG | TL | \# OF DATA POIHTS | SSS | H1 | PSU | \% Gail DIFF. FROH NOH. | BIAS DIFF. FROH NOM. | $\begin{gathered} \mathrm{K} \\ \text { FACTOR } \end{gathered}$ | $\begin{aligned} & \text { \% GAIH } \\ & \text { DIFF. } \\ & \text { FRON } \\ & \text { HOM. } \\ & \hline \end{aligned}$ | Blas DIFF. FROM NOH. | $\begin{gathered} \text { K } \\ \text { FACTOR } \\ \hline \end{gathered}$ | $\begin{aligned} & \text { \% GAIN } \\ & \text { DIFF. } \\ & \text { FROH. } \\ & \text { HOM. } \\ & \hline \end{aligned}$ | bIas DIFF. FROM HOM. | $\begin{gathered} K \\ \text { FACTOR } \\ \hline \end{gathered}$ |  |
| TDCRH3A 07/08/91 | 1 | 5/6 | 13 | 14 | 4.6 | -8. 1 | 23.8 | . 62 | . 56 | . 198 | . 04 | 1.28 | . 209 | . 27 | . 78 | . 209 | COLD OPTIC <br> LIMIT |
| TDCRMBB 07/13/91 | 2 | 5/6 | 9 | 8 | 3.2 | 12.4 | 23.5 | -. 52 | . 35 | . 198 | -1.17 | 1.22 | . 209 | -1.14 | . 68 | . 209 | $\begin{aligned} & \text { HOT OPTIC } \\ & \text { LIMIT } \end{aligned}$ |
| $\begin{aligned} & \text { T12122318 } \\ & 07 / 14 / 91 \end{aligned}$ | 3 | 5/6 | 9 | 2 | 10.9 | 15.3 | 38.2 | -1.34 | -. 66 | . 198 | -2.00 | . 14 | . 209 | -2.10 | -0.49 | . 209 | HOT SOAK \#1 |
| $\begin{aligned} & T 121 ז 231 \mathrm{~B} \\ & 07 / 18 / 91 \end{aligned}$ | 4 | 5/6 | 13 | 2 | - 2.4 | -10.8 | 0.5 | . 76 | 1.24 | . 198 | . 33 | 2.18 | . 209 | . 39 | 1.64 | . 209 | COLD SOAK \#1 |
| 07/23/91 |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  | vaculm break PMT FAILURE |
| $\begin{aligned} & T 1212318 \\ & 08 / 25 / 91 \end{aligned}$ | 5 | 5/6 | 9 | 2 | 10.8 | 16.0 | 37.9 | -2.51 | $\cdot .83$ | . 204 | -2.72 | . 28 | . 206 | -2.40 | -. 41 | . 210 | $\begin{aligned} & \text { HOT SOAK \#1 } \\ & \text { REPEAT } \\ & \hline \end{aligned}$ |
| $\begin{aligned} & \mathrm{T} 121 \mathrm{~T} 231 \mathrm{~B} \\ & 08 / 26 / 91 \end{aligned}$ | 6 | 5/6 | 13 | 2 | -3.9 | -10.4 | 0.6 | $\cdot .08$ | 1.28 | . 204 | -. 13 | 2.18 | . 206 | . 55 | 1.71 | . 210 | $\begin{aligned} & \text { COLD SOAK \#1 } \\ & \text { REPEAT } \end{aligned}$ |
| $\begin{aligned} & T 12112318 \\ & 08 / 28 / 91 \end{aligned}$ | 7 | $5 / 6$ | 9 | 2 | 11.0 | 15.1 | 38.0 | -2.67 | -. 61 | . 204 | -2.79 | . 29 | . 206 | -2.48 | -. 44 | . 210 | HOT SOAK \#2 |
| $\begin{aligned} & 121+2318 \\ & 08 / 30 / 91 \end{aligned}$ | 8 | 5/6 | 13 | 2 | -2.7 | -10.6 | 0.5 | -. 30 | 1.37 | . 204 | $-.24$ | 2.18 | . 206 | . 40 | 1.69 | . 210 | COLD SOAK W |
| $\begin{aligned} & \text { TDCRM3B } \\ & 09 / 02 / 91 \end{aligned}$ | 9 | 5/6 | 9 | 7 | 5.6 | 12.2 | 33.4 | -2.41 | $\bullet .15$ | . 204 | -2.62 | . 81 | . 206 | -2.04 | . 27 | . 210 | HOT LIMIT |
| $\begin{aligned} & \text { T0CRM3B } \\ & 09 / 08 / 91 \end{aligned}$ | 10 | 5/6 | 13 | 7 | 2.4 | -7.4 | 4.8 | -1.11 | . 88 | . 204 | -1.15 | 1.69 | . 206 | -. 56 | 1.34 | . 210 | COLD LIMIT |
| $\begin{aligned} & \text { TOCRM3C } \\ & 09 / 13 / 91 \end{aligned}$ | 19 | 5/6 | 13 | 18 | 4.8 | -7.8 | 23.6 | -1.83 | . 48 | . 204 | -1.85 | 1.24 | . 206 | -1.20 | . 87 | . 210 | NOMINAL |

## TABLE 2.4.1-4

```
OLS NUMBER 12
    T RGT DATA OF 09/12/91
    SSS AT 4.8C
    Ml AT -7.8C
    PSU TEMP = 23.6C
    Ml Coefficient = . }204\textrm{K}/\textrm{C
    T GAIN = 5
    T LEVEL = 13
    V2 <T Clamp> = 2.05404
    K9 <TL Step Size> = . }923
```


## BEST STRAIGHT LINE EQUATION

|  | FP | $(\Delta)$ | FB | SP | $(\Delta)$ | SB |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| BSL SLOPE | -0.0185 | - | -0.0215 | -0.0183 | - | -0.0215 |
| BSL AT 190K<K> | 1.34 | $(.01)$ | 1.33 | 1.31 | $(.02)$ | 1.29 |
| BSL AT 210K<K> | 0.97 | $(.07)$ | 0.90 | 0.94 | $(.08)$ | 0.86 |
| BSL AT 310K<K> | -0.88 | $(.38)$ | -1.26 | -0.89 | $(.40)$ | -1.29 |
| RMS DEVIATION<K> | 0.05 | - | 0.06 | 0.06 | - | 0.07 |
| BSL AT 310K; <br> 190 AT OV<K> | -1.30 | - | -1.67 | -1.30 | - | -1.70 |
| \% CHANGE FROM |  |  |  |  |  |  |
| $\quad$ NOM GAIN |  |  |  |  |  |  |


| OLS NUMBER 12 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| T MID DATA OF 09/12/91 |  |  |  |  |  |  |
| SSS AT 4.8C |  |  |  |  |  |  |
| M1 AT -7.8C |  |  |  |  |  |  |
| PSU TEMP $=23.6 \mathrm{C}$ |  |  |  |  |  |  |
| Ml Coefficient $=.206 \mathrm{~K} / \mathrm{C}$ |  |  |  |  |  |  |
| - T GAIN $=0$ |  |  |  |  |  |  |
| T LEVEL $=13$ |  |  |  |  |  |  |
| V2 <T Clamp> $=2.06055$ |  |  |  |  |  |  |
| K9 <TL Step Size> $=.9237$ |  |  |  |  |  |  |
| BEST STRAIGHT LINE EQUATION |  |  |  |  |  |  |
|  | FP | ( $\Delta$ ) | FB | SP | (4) | SB |
| BSL SLOPE | -0.0222 | - | -0.0244 | -0.0224 | - | -0.0242 |
| BSL AT 190K<K> | 1.94 | (.09) | 1.85 | 1.93 | (.08) | 1.85 |
| BSL AT 210K<K> | 1.50 | (.14) | 1.36 | 1.48 | (.12) | 1.36 |
| BSL AT 310K<K> | -0.72 | (.36) | -1.08 | -0.76 | (.30) | -1.06 |
| RMS DEVIATION<K> | 0.07 | - | 0.09 | 0.08 | - | 0.09 |
| BSL AT 310K; |  |  |  |  |  |  |
| 190 AT OV<K> | -1.31 | - | -1.65 | -1.35 | - | -1.63 |
| \% CHANGE FROM |  |  |  |  |  |  |
| NOM GAIN | -1.85 | - | -2.32 | -1.91 | - | -2.29 |
| BIAS DIFF FROM |  |  |  |  |  |  |
| NORMAL 190K<K> | 1.24 | - | 0.79 | 1.19 | - | 0.80 |

## TABLE 2.4.1-6

OLS NUMBER 12
T LFT DATA OF 09/12/91
SSS AT 4.9C
Ml AT -7.9C
PSU TEMP $=23.7 \mathrm{C}$
M1 Coefficient $=.210 \mathrm{~K} / \mathrm{C}$
T GAIN $=6$
T LEVEL = 13
V2 <T Clamp> $=2.06706$
K9 <TL Step Size> = . 9237

## BEST STRAIGHT LINE EQUATION

|  | FP | ( $\Delta$ | FB | SP | ( $\Delta$ ) | SB |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| BSL SLOPE | -0.0148 | - | -0.0154 | -0.0147 | - | -0.0154 |
| BSL AT 190K<K> | 1.32 | (.16) | 1.16 | 1.31 | (.18) | 1.13 |
| BSL AT 210K<K> | 1.03 | (.18) | 0.85 | 1.02 | (.19) | 0.85 |
| BSL AT 310K<K> | -0.45 | (.24) | -0.69 | -0.45 | (.26) | -0.71 |
| RMS DEVIATION<K> | 0.05 | - | 0.05 | 0.05 | - | 0.04 |
| BSL AT 310K; 190 AT OV<K> | -0.85 | - | -1.05 | -0.85 | - | -1.06 |
| \% CHANGE FROM NOM GAIN | -1.20 | - | -1.48 | -1.19 | - | -1.50 |
| BIAS DIFF FROM NORMAL 190K<K> | 0.87 | - | 0.47 | 0.87 | - | 0.42 |

FIGURE 2.4.1-1
COLD OPTIC



FIGURE 2.4.1-2


COLD OPTIC
LIMIT
NNNI

FIGURE 2.4.1-3

## HOT OPTIC LIMIT




FIGURE 2.4.1-5
HOT LIMIT


FIGURE 2.4.1-6


Figure 2.4.1-7
COLD LIMIT


FIGURE 2.4.1-8
COLD LIMIT



RUN \# 10

FIGURE 2.4.1-9
NOMINAL


Figure 2.4.1-10
NOMINAL


RUN \# 11

### 2.4 RADIOMETRIC ACCURACY <br> 2.4.1 T Channel Radiometric Accuracy (Cont'd) <br> 2.4.1.1 Repeatability (3.2.1.1.4.1a) <br> The 1 sigma Repeatability of $T$ Channel DC Response is <br> $0.262^{\circ} \mathrm{K}$ compared to a $0.42^{\circ} \mathrm{K}$ one sigma specification maximum and therefore OLS \#12 does meet this specification requirements.

## ATTACHMENTS

Table 2.4.1.1-1 Repeatability Contributors
Table 2.4.1.1-2 Gain and Bias Variations with Temperature Change
Table 2.4.1.1-3 Target Crosstalk, T Clamp Leakage Data

| TABLE 2.4.1.1-1OLS \#12REPEATABILITY CONTRIBUTORS SUMMARY |  |
| :---: | :---: |
| ERROR SOURCE | $\begin{aligned} & \text { ONE SIGMA } \\ & \text { ERROR }\left(K^{\circ}\right) \\ & \hline \end{aligned}$ |
| 1. Diurnal Ml Temperature Change $\left(4^{\circ} \mathrm{C}\right)$ <br> A. Quantization of I Level Command <br> B. Inability to Compensate Actual Effect Exactly | $\begin{aligned} & 0.19 \\ & 0.077 \end{aligned}$ |
| 2. Temperature Change $\mathrm{PSU} \pm 4.5^{\circ} \mathrm{C}$, $\mathrm{SSS} \pm 1^{\circ} \mathrm{C}$ <br> A. Effect due to Gain Change <br> B. Effect due to Bias Change | $\begin{aligned} & 0.066^{*} \\ & 0.058^{*} \end{aligned}$ |
| 3. T Clamp Shaper Compensation | 0.09 |
| 4. T Clamp Leakage | 0.104 |
| TOTAL RSS REPEATABILITY ERROR ( ${ }^{\circ} \mathrm{K}$ ) SPECIFICATION LIMIT, ${ }^{\circ} \mathrm{K}$, ONE SIGMA <br> *FROM TEST DATA (REDUCED) | $\begin{aligned} & 0.262 \\ & 0.42 \end{aligned} \operatorname{MAX} .$ |

## Discussion of Repeatability Calculations

1. Dinurnal M1 Temperature Change
A. The effects of M1 temperature (more properly the foreoptics temperature) are a Repeatability error source. The foreoptics thermal time constant is short enough to permit significant diurnal temperature variations. The ability to compensate for foreoptics temperature using the $T$ Level command greatly reduces this error but does not eliminate it. Although calculations enabling ground compensation smaller than the quantization of the $T$ Level command are possible, it is herein assumed that they will not generally be made. Therefore an error is ascribed due to the $T$ Level quantization as follows:
$0.294^{\circ} \mathrm{K}$ RMS T Level Cmd. Quantization Error at $210^{\circ} \mathrm{K}\left(1.02^{\circ} \times 1 \sqrt{12}\right)$ $x 0.642$ RMS Temperature Linearity Effects over $210-310^{\circ} \mathrm{K}$ dynamic range $=0.19^{\circ} \mathrm{K}$ RMS error
B. The fact that foreoptics temperature effect cannot be accurately predicted by the single monitor of M1 temperature means that in times of sharp transition the ability to compensate is impaired. It has been assumed that this error may be represented by a $1^{\circ} \mathrm{C}$ lag in M1 temperature during the $1 / 3$ of the orbit that sharp transistions occur. Therefore the inability to compensate the actual effect is ascribed the following error:

|  | $1^{\circ} \mathrm{C}$ | in M1 Temperature |
| :---: | :---: | :---: |
| x | $1 \sqrt{3}$ | RMS Over total orbit |
| x | 0.207 | T Left T Mid T Right average sensitivity coefficient of video at 210 K to Ml temperature change for OLS \#12 (K |
|  |  | factor) |
| $x$ | 0.642 | Temperature Linearity Effects over dynamic range |
|  | 0.077 | RMS error |

## Discussion of Repeatability Calculations

Total T channel gain change with temperature may be broken down into two components as follows:

```
Total Gain \Delta = [{PSU \DeltaT) < P PG] + [SSS \DeltaT) < S S ]
    where: }\quad\mp@subsup{P}{G}{}=PSU\mp@subsup{U}{}{G}\mathrm{ coefficient of gain,% per }\mp@subsup{}{}{\circ}\textrm{C}\mathrm{ .
    SG}=SSS\mathrm{ coefficient of gain, % per }\mp@subsup{}{}{\circ}\textrm{C}\mathrm{ .
```

Similarly for bias changes with temperature:

```
Total Bias \Delta = [(PSU \DeltaT) < P P B ] + [(SSS \DeltaT) < < S S ]
```



```
    SB
```

Data from two pairs of runs designated $A$ and $B$, were used to solve these equations simultaneously:

where: $\quad G=$ Total Gain change over temperature
$B=$ Total Bias change over temperature
$T_{p}=$ PSU Temperature change
$\mathrm{T}_{\mathrm{s}}=$ SSS Temperature change

Solved simultaneously for the temperature sensitivity factors, these equations can be reduced to the following:

$$
\begin{aligned}
& S_{G}=\frac{\left(T_{P A}\right)\left(G_{B}\right)-\left(T_{P B}\right)\left(G_{A)}\right.}{(T P A)\left(T_{S B}\right)-\left(T_{P B}\right)\left(T_{S A}\right)} \\
& P_{G}=\frac{G_{A}-\left(T_{S A}\right)\left(S_{G}\right)}{T_{P A}} \\
& S_{B}=\frac{\left(T_{P A}\right)\left(B_{B}\right)-\left(T_{P B}\right)\left(B_{A}\right)}{\left(T_{P A}\right)\left(T_{S B}\right)-\left(T_{P G}\right)\left(T_{S A}\right)} \\
& P_{B}=B_{A}-\left(T_{S A}\right)\left(S_{B}\right) \\
& T_{P A}
\end{aligned}
$$

2. SSS and PSU Temperature Change: Effect On Gain Change

The effects of SSS and PSU temperature change on gains were determined from parametric analysis of the four runs in Table 2.4.1.1-2. Solving simultaneous equations yielded sensitivity coefficients of gain change for both SSS and PSU temperature change. Temperature data from 5D-2 systems currentiy on-orbit indicates that the worst case SSS temperature variations are $1^{\circ} \mathrm{C}$ while worst case PSU temperature variations are $4.5^{\circ} \mathrm{C}$. Using the worst-case factors yields:
$S_{G}=.039 \%$ Gain change per degree SSS change
$\times 1^{\circ} \mathrm{C}$ temperature change
x $.31^{\circ} \mathrm{K}$ RMS over 210 K to 310 K range
$x 1 \sqrt{3}$ for uniform temperature distribution $=.007 \mathrm{deg}$
$P_{G}=-.082 \%$ Gain change per degree PSU change
$\times 4.5^{\circ}$ temperature change
$\mathrm{x} .31^{\circ} \mathrm{K}$ RMS over 210 K to 310 K range
$\times 1 \sqrt{3}$ for uniform temperature distribution $=-.066 \mathrm{deg}$

RSS'ing these two contributors yields 0.066 degree total.
3. SSS and PSU Temperature Change, Effect On Bias Change From Table 2.4.1.1-2:
$S_{B}=-.140$ deg Bias change per degree SSS change
$\times 1^{\circ}$ temperature change
$\times 0.642$ RMS Temperature Linearization Effects, 210 K to 310K
$\times 1 \sqrt{3}$ for uniform temperature distribution
$=-.052 \mathrm{deg}$
$P_{B}=-.015 \mathrm{deg}$ Bias change per degree PSU change
$\times 4.5^{\circ}$ temperature change
x 0.640 RMS Temperature Linearization Effects, 210K to 310K
$x 1 \sqrt{3}$ for uniform temperature distribution
$=-.025 \mathrm{deg}$
RSS'ing these two contributors yields 0.058 degree total.

## 4. T Clamp Shaper Compensation

The SSS temperature changes throughout each orbit are expected to cause a one sigma error of $0.146^{\circ} \mathrm{K}$ at 210 K due to the compensation for $T$ clamp temperature variation from 228 K to 253K. This times the 0.642 RMS Temperature Linearization Effect over the dynamic range equals $0.09^{\circ} \mathrm{K}$ RMS error.

## 5. T Clamp Leakage

An along scan variation (ASV) effect may be caused by some of the scene radiance being viewed at the time of $T$ clamp during the overscan period getting into the reference T Clamp value. This can happen if the T detector sensitivity extends slightly beyond the boundaries of $\mathrm{M4'}^{\prime}$ during the clamp time. The test performed for leakage is to view one target at $210^{\circ} \mathrm{K}$ during active scan while the other target (which is at the T Clamp angle) is varied over the $210^{\circ}$ to $310^{\circ} \mathrm{K}$ dynamic range. This data is presented in Table 2.4.1.1-3. The effect on the response to the active scan target is attributed to the T clamp optical leakage.

Using the OLS \#12 T data from Table 2.4.1.1-3, the $T$ clamp leakage contribution can be calculated as follows:

$$
\begin{aligned}
& T \text { clamp leakage ratio }(L R)=100 \% \times \Delta T \times \frac{\frac{\partial \mathrm{P}}{\Delta \mathrm{~T}} 214}{\Delta \mathrm{P}} \\
&=\Delta \mathrm{T} \times .50552 \%
\end{aligned}
$$

This calculation is performed in the MODE 4 data reduction of T121T221S. The ratio calculated is:

$$
\begin{array}{ll}
-0.032 \% & T \text { LEFT } \\
-0.122 \% & \text { T RIGHT }
\end{array}
$$

The peak error from $T$ clamp leakage (due to the $310^{\circ}$ background) can be calculated as follows:
peak error $=L R \times \Delta N \times\left(\frac{\partial P^{-1}}{\Delta T}\right)-1$

$$
=\Delta T \times \frac{\Delta N}{\Delta P} \times \frac{\frac{\partial P}{\Delta T_{214}}}{\frac{\bar{P}_{210}}{\Delta T}}
$$

$$
\begin{aligned}
& =\Delta T \times 0.8156 \times 1.0788 \\
& =\Delta T \times 0.880
\end{aligned}
$$

where:
$\Delta P=$ Difference in radiance between $210^{\circ}$ and $310^{\circ} \mathrm{K}$
$=16.742 \mathrm{E}-4 \mathrm{w} \mathrm{cm}^{-2} \mathrm{sr}^{-1} @ 310^{\circ} \mathrm{K}$
$-2.3468 \mathrm{E}-4 \mathrm{~W} \mathrm{~cm}^{-2} \mathrm{sr}^{-1}$ @ $210^{\circ} \mathrm{K}$
$=14.395 \mathrm{E}-4 \mathrm{w} \mathrm{cm}^{-2} \mathrm{sr}^{-1}$
and: $\quad \Delta N=$ Difference in radiance between $240^{\circ}$ and $310^{\circ} \mathrm{K}$
$=16.742 \mathrm{E}-4 \mathrm{~W} \mathrm{~cm}^{-2} \mathrm{sr}^{-1} @ 310^{\circ} \mathrm{K}$
$-5.001 \mathrm{E}-4 \mathrm{~W} \mathrm{~cm}^{-2} \mathrm{sr}^{-1} @ 240^{\circ} \mathrm{K}$
$11.741 \mathrm{E}-4 \mathrm{w} \mathrm{cm}^{-2} \mathrm{sr}^{-1}$
$\frac{\partial P}{\Delta T_{210}}=$ slope of radiance curve at $210^{\circ} \mathrm{K}=6.7452 \mathrm{E}-6$
$\frac{\partial \mathrm{P}}{\Delta \mathrm{T}}=$ slope of radiance curve at $214^{\circ} \mathrm{K}=7.277 \mathrm{E}-6$
$\Delta \mathrm{T}=$ measured change in response to $210^{\circ}$ target as the background is varied from $210^{\circ}$ to $310^{\circ} \mathrm{K}$.

RMS ERROR = PEAK ERROR
$\times 0.7605$ for RMS distribution of leakage radiance over dynamic range.
x 0.642 RMS Temperature Linearization Effect

FROM MODE 4 Data reduction:
Calculated RMS leakage error $=-0.027^{\circ} \mathrm{K}$ T LEFT
$=-0.104^{\circ} \mathrm{K}$ T RIGHT
The worst-case contribution to repeatability error by T-clamp leakage is therefore $-0.104^{\circ} \mathrm{K}$ RMS.

$$
2-134
$$

TABLE 2.4.1.1-2

|  |  | $\begin{aligned} & \text { SSS } \\ & \text { TEMP } \end{aligned}$ | $\begin{aligned} & \text { PSU } \\ & \text { TEMP } \end{aligned}$ | T RGT |  | T MID |  | T LFT |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | \% GAIN <br> DELTA <br> (\%) | $\begin{gathered} \text { BIAS CHG. } \\ 0190^{\circ} \mathrm{K} \\ \left({ }^{\circ} \mathrm{K}\right) \\ \hline \end{gathered}$ | \% GAIN DELTA (\%) | $\begin{gathered} \text { BIAS CHG. } \\ 190^{\circ} \mathrm{K} \\ \left({ }^{\circ} \mathrm{K}\right) \end{gathered}$ | \% GAIN <br> DELTA <br> (\%) | $\begin{gathered} \text { BIAS CHG. } \\ 0190^{\circ} \mathrm{K} \\ \left({ }^{\circ} \mathrm{K}\right) \end{gathered}$ |
| $\begin{gathered} M 1=-8^{\circ} \mathrm{C} \\ (\text { Rün } \mathrm{A}) \end{gathered}$ | RUN 11 <br> (NOMINAL TEST) | 4.8 | 23.6 | -1.83 | 0.48 | -1.85 | 1.24 | -1.20 | . 87 |
|  | RUN 8 (COLD SOAK | -2.7 | 0.5 | -. 30 | 1.37 | -. 24 | 2.18 | . 40 | 1.69 |
|  | $\text { RUN } 11-$ | $\begin{aligned} & 7.5 \\ & T_{S A} \end{aligned}$ | $\stackrel{23.1}{T_{P A}}$ | $-1.53$ | $-.89$ | $-1.61$ | $-.94$ | $-1.60$ | $-.82$ |
| $\begin{gathered} M 1=+12^{\circ} \mathrm{C} \\ \text { (Run B) } \end{gathered}$ | $\begin{gathered} \text { RUN } 7 \\ (H O T \text { SOAK } \\ \# 2) \end{gathered}$ | 11.0 | 38.0 | -2.67 | -. 61 | -2.79 | . 29 | -2.48 | -. 44 |
|  | RUN 9 $\begin{aligned} & \text { HOT } \\ & \text { LIMIT) } \\ & \hline \end{aligned}$ | 5.6 | 33.4 | -2.41 | -. 15 | -2.62 | . 81 | -2.04 | . 27 |
|  | RUN 7 RUN 9 | $\begin{gathered} 5.4 \\ \mathrm{~T}_{\text {SB }} \end{gathered}$ | $\begin{gathered} 4.6 \\ \mathrm{~T}_{\mathrm{PB}} \end{gathered}$ | $-.26$ | $\stackrel{-.46}{B_{B}}$ | $-_{\mathrm{G}_{\mathrm{B}}} 17$ | $-_{B_{B}}^{52}$ | $-._{B}^{44}$ | $-.71$ |
| Calculated Sensitivity Factors |  | $\begin{array}{ll} \text { SSS: } & \mathrm{S}_{\mathrm{G}} \\ \text { PSU: } & \mathrm{S}_{\mathrm{B}} \\ & \mathrm{P}_{\mathrm{B}}^{\mathrm{G}} \end{array}\left(\begin{array}{l} \% /{ }^{\circ} \mathrm{C} \\ \% /{ }^{\mathrm{C}} \\ \% /{ }^{\mathrm{C}} \\ { }^{\mathrm{C}} \\ { }^{\mathrm{C}} \mathrm{C} \end{array}\right\}$ |  | $\begin{aligned} & -.011 \\ & -.070 \end{aligned}$ | $\begin{aligned} & -.072 \\ & -.015 \star \end{aligned}$ | $\begin{gathered} .039 * \\ -.082^{*} \end{gathered}$ | $\begin{aligned} & -.085 \\ & -.013 \end{aligned}$ | $\begin{aligned} & -.031 \\ & -.059 \end{aligned}$ | $\begin{gathered} -.140^{*} \\ -.010 \end{gathered}$ |

TABLE 2.4.1.1-3
OLS \#12
target crosstalk, T CLAMP LEAKage Data*
$\begin{aligned} \text { SSS } & =+5^{\circ} \\ \text { M1 } & =-8^{\circ}\end{aligned}$

|  | $\begin{gathered} \mathrm{T} \\ \text { RIGHT } \\ \hline \end{gathered}$ | T MID | T CPL | T CPR | T LEFT |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{aligned} & \mathrm{T1} 210^{\circ} \\ & \text { [T2@310*] } \\ & (\mathrm{T} 121 \mathrm{~T} 231 \mathrm{G}) \end{aligned}$ | 0.32 | 1.00 | 0.08 | 0.75 | 0.57 | 07-06-91 |
| Difference, $\Delta T$ | 0.25 | 0.12 | 0.03 | 0.07 | 0.08 |  |
| $\begin{aligned} & T 1210^{\circ} \\ & {\left[T 2 @ 210^{\circ}\right]} \\ & (T 121 T 221 S) \\ & \hline \end{aligned}$ | 0.07 | 0.88 | 1.05 | 0.62 | 0.45 | 07-07-91 |
| $\begin{aligned} & T 2210^{\circ} \\ & {\left[T 10^{\circ} 310^{\circ}\right]} \\ & (T 131 T 221 A) \end{aligned}$ | 0.30 | 1.12 |  |  | 0.55 | 07-07-91 |
| Difference, $\Delta T$ | 0.14 | 0.21 |  |  | 0.13 |  |
| $\begin{aligned} & T 2210^{\circ} \\ & {\left[T 1 \text { @ } 210^{\circ}\right]} \\ & (T 1212215) \end{aligned}$ | 0.16 | 0.91 |  |  | 0.42 | 07-07-91 |
| Worst Case Data <br> From Tl21T221S. <br> Mode 4 Data <br> Reduction: <br> T clamp leakage Peak leakage er RMS leakage err | tio is at 210 at 210 | is <br> is | $\begin{aligned} & -0.032 \% \\ & -0.055 \mathrm{~K} \\ & -0.027 \mathrm{~K} \end{aligned}$ | $\begin{aligned} & -0.122 \% \\ & -0.213 K \\ & -0.104 K \end{aligned}$ |  |  |

*Data is FP Deviation in ${ }^{\circ} \mathrm{K}$

### 2.4 Radiometric Accuracy

### 2.4.1 T Channel Radiometric Accuracy (Cont'd)

### 2.4.1.2 Stability (3.2.1.1.4.1b)

The T Channel Radiometric Accuracy (Stability) analysis in the OLS 5D-3 System Summary Report Paragraph 3.5.1.2, predicts $0.62^{\circ} \mathrm{K} 1$ sigma error in stability compared to the $0.8^{\circ} \mathrm{K}$ maximum specification requirement. This calculation is applicable to all 5D-3 systems with $190-310^{\circ} \mathrm{K}$ total range.

## ATTACHMENTS

Table 2.4.1.2-1 Stability Contributors Summary
Table 2.4.1.2-2 Change in BSL $210^{\circ}$, $310^{\circ} \mathrm{K}$ Points Between Runs
Table 2.4.1.2-3 Change in $210^{\circ}, 310^{\circ}$ Output Deviation From Nominal ( ${ }^{\circ} \mathrm{K}$ ) between Power Supply 1 and Power Supply 2

## STABILITY CONTRIBUTORS SUMMARY

## ONE SIGMA

ERROR ( ${ }^{\circ} \mathrm{K}$ )

1. Shaped Bias
a) Open Loop Mirror Emissivity
0.1
b) T Clamp Shaper Compensation - Temperature
0.23

- Age
0.17
RSS Total
$0.30^{\circ} \mathrm{K}$
X RMS Temperature Linearization Effect
0.642
= RMS Shaped Bias Errors
$0.19^{\circ} \mathrm{K}$

2. Bias
a) Preshaper Gain

- Inner Stage Temperature
0.28
- Bias Current
0.24
- Amplifiers
0.22
b) Post Shaper DC Drift $\underline{\mathbf{0 . 1 2}}$
RSS Total = RMS Bias Error
$0.45^{\circ} \mathrm{K}$

3. Gain

Postshaper Gain Changes - Amplifier
over the $210-310 \mathrm{~K}$ range, ${ }^{\circ} \mathrm{K}$ RMS Error $\underline{\mathbf{0 . 3 8}}$
$\begin{array}{ll}\text { IOTAL RSS Stability Error (Total Dynamic Range) } & 0.62\end{array}$

Stability Error Specification ( ${ }^{\circ} \mathrm{K}, 1$ Sigma) $\quad 0.80$ Maximum

## Discussion of Stability Errors

The experimentally derived RMS change of the BSL(s) between runs was calculated to be $0.04^{\circ} \mathrm{K}, 0.04^{\circ} \mathrm{K}$ and $0.06^{\circ} \mathrm{K}$ for TRGT, TMID and TLEFT respectively. The two runs used were Run \#6 and Run \#8. The results verify the analytical estimate of the stability over time intervals greater than one day. This data is tabulated in Table 2.4.1.2-2.

As an additional check of stability, the Fine-Primary outputs of the three segments as a deviation from nominal ( ${ }^{\circ} \mathrm{K}$ ) at $210^{\circ}$ and $310^{\circ}$ were compared using power supply 1 data of TDCRM3C.ST and power supply 2 data of $6 \times 2 \times 3$ A.ST. (Both from Run \#11). This data is tabulated in Table 2.4.1.2-3. The deltas were calculated and RMS'd over the temperature range. The results are comparable to the "Change Between Runs" data.

TABLE 2.4.1.2-2
OLS \#12

CHANGE IN BSL 210, 310K POINTS BETWEEN RUNS

$$
S S S=+3^{\circ} \mathrm{C}, \mathrm{Ml}=-8^{\circ} \mathrm{C}
$$



TABLE 2.4.1.2-3
OLS \#12
T CHANNEL DC RESPONSE
DIFFERENCE BETWEEN POWER SUPPLIES 1 and 2
From Orbit Nominal (Run \#11), SSS $=+5^{\circ} \mathrm{C}, \mathrm{Ml}=-8^{\circ} \mathrm{C}$

|  | RIGHT |  | MID |  | LEFT |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\begin{aligned} & \text { TGT-1 } \\ & 210^{\circ} \mathrm{K} \\ & \hline \end{aligned}$ | $\begin{aligned} & \text { TGT-2 } \\ & 310^{\circ} \mathrm{K} \\ & \hline \end{aligned}$ | $\begin{aligned} & \hline \text { TGT-1 } \\ & 210^{\circ} \\ & \hline \end{aligned}$ | $\begin{array}{r} \text { TGT-2 } \\ 310^{\circ} \\ \hline \end{array}$ | $\begin{array}{r} \text { TGT-1 } \\ 210^{\circ} \\ \hline \end{array}$ | $\begin{array}{r} \text { TGT-2 } \\ 310^{\circ} \\ \hline \end{array}$ |
| FP DEV [K] <br> Power Supply 1 <br> TDCRM3C.ST <br> 09/13/91 | 0.97 | -0.87 | 1.51 | -0.62 | 0.94 | -0.49 |
| FP DEV [K] Power Supply 2 6X2X3A.ST | 0.98 | -0.93 | 1.56 | -0.65 | 1.00 | -0.55 |
| Change * K | 0.01 | 0.06 | 0.05 | 0.03 | 0.06 | 0.06 |
| RMS * K |  |  |  |  |  |  |

### 2.4 RADIOMETRIC ACCURACY

### 2.4.1 T Channel Radiometric Accuracy (Cont'd)

2.4.1.3 Fixed Deviations (3.2.1.1.4.1c)

The Fixed deviations for OLS \#12 are $0.29^{\circ} \mathrm{K}, 1$ sigma, compared to the $0.6^{\circ} \mathrm{K}$ specification maximum. The calibrateable portion of the fixed deviations is $0.17^{\circ} \mathrm{K}$ RMS compared to the $0.4^{\circ} \mathrm{K}$ RMS specification maximum. The Fixed deviation calibration for separate detector segments is $0.82^{\circ} \mathrm{K}$ (worst case) compared to the $1^{\circ} \mathrm{K}$ spec. maximum. The maximum along scan variation was $0.14^{\circ} \mathrm{K}$ RMS for TF (Right) and $0.12^{\circ} \mathrm{K}$ RMS for TS compared to the $0.2^{\circ} \mathrm{K}$ RMS specification maximum.

## ATTACHMENTS

Table 2.4.1.3-1 Fixed Deviations Contributors
Table 2.4.1.3-2 T Shaper Error Tabulation
Table 2.4.1.3-3 Target Deviation from Mean of Both Targets
Table 2.4.1.3-4 BSL Calibration Equations
Table 2.4.1.3-5 Fixed Deviation Calibration Differences for Separate Segments
Table 2.4.1.3-6 Along Scan Variation ( $265^{\circ}$ to $310^{\circ} \mathrm{K}$ ) within a Separate Segments
Table 2.4.1.3-7 Cone (Inner Stage) Patch Temp EST
Table 2.4.1.3-8 Cone Cooler Outer Stage Temp EST
Figure 2.4.1.3-1 5D3 Nominal Shaper Curve
Figure 2.4.1.3-2 Along Scan Variation, $T$ Right, $M 1=12^{\circ} \mathrm{C}$
Figure 2.4.1.3-3 Along Scan Variation, $T$ Mid, $M 1=12^{\circ} \mathrm{C}$
Figure 2.4.1.3-4 Along Scan Variation, $T$ Left, $M 1=12^{\circ} \mathrm{C}$
Figure 2.4.1.3-5 Along Scan Variation, $T$ Right, $M 1=-8^{\circ} \mathrm{C}$
Figure 2.4.1.3-6 Along Scan Variation, $T$ Mid, $M 1=-8^{\circ} \mathrm{C}$
Figure 2.4.1.3-7 Along Scan Variation, T Left, Ml $=-8^{\circ} \mathrm{C}$
Figure 2.4.1.3-8 Along Scan Variation, $T$ Auto $M 1=12^{\circ} \mathrm{C}$
Figure 2.4.1.3-9 Along Scan Variation, $T$ Auto $M 1=-8^{\circ} \mathrm{C}$

## FIXED DEVIATION CONTRIBUTORS

## DEVIATION SOURCE

1. Foreoptics Mirror Emissivity
2. T Clamp Shaper Compensation
3. Transfer Function
A. Non-Linearity $\quad 0.17 * 0.4^{\circ} \mathrm{K}$ Spec Max
B. Shaper Components Variation
0.10
C. Detector Spectrum Variation (included in 3A)
4. Test Targets
A. Temperature 0.10
B. Emissivity
0.10
C. Repeatability
0.04*

TOTAL (RSS) FIXED DEVIATION 0.29
FIXED DEV. SPECIFICATION LIMIT, ${ }^{\circ} \mathrm{K}$ ONE SIGMA
0.60 Maximum

* FROM TEST DATA ANALYSIS

5. Fixed Deviation BSL Calibrations Match for $\frac{\text { DATA }}{0.82} \frac{\text { SPEC MAX }}{1 .{ }^{\circ} \mathrm{K}}$ Separate Segments (Worst Case)
6. Along Scan Varations within a segment
$\left(265^{\circ}\right.$ to $\left.310^{\circ} \mathrm{K}\right)$ Worst Case

## Discussion of Fixed Deviation Tests and Calculations

1. Foreoptics Mirror Emissivity

The foreoptics mirror emissivity is a source of Fixed deviations as well as of Repeatability and Stability deviations. The correction in operation is made on M1 temperature only; whereas the entire foreoptics causes the offset phenomena. The ground calibration is made in the thermal vacuum chamber, where M1 is cooled radiatively via a cold tunnel, which fills the Ml view beyond the optical field of view of the system. Thus the foreoptics temperature distribution in the chamber differs somewhat from operation in space, especially in that M2 is colder in the chamber. It is not possible to determine accurately this effect based upon present knowledge. It is therefore assumed that it is equivalent to a $1^{\circ} \mathrm{C}$ difference in Ml temperature, or about 0.207 at $210^{\circ}$ using the actual OLS \#12 Ml coefficient ( $K$ factor). The RMS Temperature Linearization Effect, 0.642 , transforms this to a $0.13^{\circ} \mathrm{K}$ RMS contribution to fixed deviation.

## 2. T Clamp Shaper Compensation

The T Clamp shaper compensation contribution arises as follows. If the T Clamp emissivity were initially only 0.98 rather than the 0.995 used to calculate the compensation for T Clamp temperature, the error at $242^{\circ} \mathrm{K}$ would be $0.70^{\circ} \mathrm{K}$. Although this error would be compensated for when the $T$ channel adjustments were made, a change in T Clamp temperature to $256^{\circ} \mathrm{K}$ or $230^{\circ} \mathrm{K}$ would result in an error of $0.78^{\circ} \mathrm{K}$, producing an uncompensated error of $0.08^{\circ} \mathrm{K}$. Over the dynamic range this is equivalent to $0.09^{\circ} \mathrm{K}$ RMS.

## 3. Transfer Function

A. The departure of the $T$ channel radiometric transfer function from a linear relationship is not an error as such because it is known and compensation can be made for it. However, this type of deviation is included within the constraints of the Fixed deviation portion of the $T$ channel radiometric accuracy spec. The nominal T Channel non-linear transfer function (shaper) error is tabulated in Table 2.4.1.3-2 and plotted in Figure 2.4.1.3-1. The nominal shaper error is $0.15^{\circ} \mathrm{K}$ RMS. This calculation is made with the 50-2 shaper, which is also used on OLS \#12. The worst-case reduced test data (from Tables 2.4.1-4,5 \& 6) RMS Deviations of the points from the BSL for OLS \#12, are $0.07^{\circ} \mathrm{K}$ for T Right (Smooth Backup), $0.09^{\circ} \mathrm{K}$ for T MID (Fine and Smooth Backup) and $0.05^{\circ} \mathrm{K}$ for $T$ Left (Fine Primary \& Smooth Backup). The analytic value, $\left(0.15^{\circ} \mathrm{K}\right.$ RMS ) and the worst-case test value of $0.09^{\circ} \mathrm{K}$ are RSS'ed to become $0.17^{\circ} \mathrm{K}$ RMS for this Fixed deviation source versus the $0.4^{\circ} \mathrm{K}$ maximum spec allowance for the calibrateable portion of the Fixed deviations. B. Departure of shaper components from design nominal values may cause additional peak errors of $0.25^{\circ} \mathrm{K}$ and are assigned a one sigma error of $0.1^{\circ} \mathrm{K}$. These deviations would not be included in the measured deviation, because the data points are not close enough together to adequately detect them.
C. The $T$ detector spectrum difference from unit to unit is included in the calibration data and is therefore included in 3A, the non-linearity of transfer function.

## 4. Test Targets

The deviation of the reference test target indicated blackbody temperature from absolute is the result of contributions from three Fixed deviation sources.
A. The measurement of the averaged target temperature using the 5 thermocouples immersed in the target baseplate is subject to the accuracy of the PQL/Block V Thermocouple Aquisition and Control System calibrated per procedure MCS0116801B. A review of the calibration procedure and the equipment used has led to ascribing $0.1^{\circ} \mathrm{K}$ RMS to these sources.
B. The actual effective blackbody temperature of the target also deviates from that measured because of differing emissivity from that assumed and because the "true" radient temperature differs from the measured temperature using the thermocouples. The effective emissivity is a combination of true emissivity (which is better than 0.996 according to Eppley) and reflectance of up to 0.004 . An average emissivity of 0.998 is corrected for along with the thermocouple calibration. This source of deviation has been assigned 0.1 ${ }^{\circ} \mathrm{K}$ RMS .
C. Actual measurements with OLS \#12 system of the two Eppley IR Reference test targets over the $210^{\circ}$ to $310^{\circ} \mathrm{K}$ range reveal small differences between the targets. These differences, as tabulated in Table 2.4.1.3-3, represent the target differences from the mean of both targets for T LFT and T RGT averaged from file TDCRM3C.ST (Run \#11). The eleven differences when RMS'ed yield $0.04^{\circ} \mathrm{K}$ RMS deviation for this source.

## Fixed Deviation Calibrations for Separate Segments

The calibrations are represented by the best straight line (BSL) equations for the separate segments. The BSL deviation expressions (from ideal), in $y=m x+b$ form for the segments are tabulated in Table 2.4.1.3-4. In T Smooth the Right and Left detector segments are averaged and used across the entire scan line.

In order to determine the differences in calibrations for separate segments, (for comparison to the specification) the $T$ Fine BSL deviation equations of Table 2.4.1.3-4 were evaluated at $210^{\circ} \mathrm{K}$ and $310^{\circ} \mathrm{K}$. The 3 possible segment differences were then calculated. Also, the worst-case deviations between segments were taken from the Best Straight Line Plots (Figures 2.4.1-1 thru 2.4.1-12). These results are tabulated in Table 2.4.1.3-5. The calibration differences for separate segments are within the 1*K maximum spec throughout the dynamic range of 210 to 310 K for OLS \#12.

## Along Scan Variation ( $265^{\circ} \mathrm{K}$ to $310^{\circ} \mathrm{K}$ ) Within a Segment

The Along Scan Variation (ASV) in T DC Response is considered to be a Fixed deviation, since as a function of scan angle it does not vary as a function of time per se. Figures 2.4.1.3-2, 3, 4, 5, 6, 7, 8 and 9 show the deviations along-scan vs. surface distance (nmi/100) for T RGT, T MID, TLFT, and T AUTO. The vertical scale factor is $1.0^{\circ} \mathrm{K}$ per cm . Four figures are for $M 1=-8^{\circ} \mathrm{c}$ and four are for $M 1=+12^{\circ} \mathrm{C}$. The dip in the plots of $+5,-8$ data for the $290^{\circ} \mathrm{K}$ target at -550 nmi surface distance is due to the inadvertent deletion of a data file. The slight sawtooth effect on the $290^{\circ} \mathrm{K}$ plots is a result of the missing data's effect on the processing of the data. All data was within spec. The computer printed number to the right of each curve is the computed RMS deviation in millidegrees $K$ for the associated ASV plot. The

RMS ASV values are only printed for the target temperatures above $265^{\circ} \mathrm{K}$, i.e., the $270^{\circ}, 290^{\circ}$, and $310^{\circ} \mathrm{K}$ plots.

The T DC response BSL calibrations are taken using test scan angles of $-50^{\circ}(-600 \mathrm{nmi})$ for $T \mathrm{LFT},+50^{\circ}(+600 \mathrm{nmi})$ for $T \mathrm{RGT}$, and $+0^{\circ}(0 \mathrm{nmi})$ for $T$ MID.

The OLS \#12 has some ASV, but is within spec. The worst case (max) ASV RMS value within a segment for OLS \#12 was $0.14^{\circ} \mathrm{K}$ and is entered in Table 2.4.1.3-6 to compare with the specification limit.

TABLE 2.4.1.3-2

## I SHAPER ERROR LIST

The 190 to $310^{\circ} \mathrm{K}$ T Shaper used for OLS \#12 and up has 6 straight line segments of decreasing slope and 5 (inflections) diode break points. The significant features in the shaper curve are tabulated below:

| $\left({ }^{\frac{T}{K}}\right)$ | $\frac{\text { ERROR }}{\left({ }^{\circ} \mathrm{K}\right)}$ | REMARKS |
| :---: | :---: | :---: |
| 190 | 0 | End point adjusted to be an Ideal Curve |
| 195.5 | -0.35 | lst slope is parallel to Radiance (Smooth) Curve |
| 201.5 | 0 |  |
| 205 | +0.16 | lst diode cut-in |
| 209 | 0 |  |
| 214 | -0.215 | 2nd slope is parallel to Radiance (Smooth) Curve |
| 219.5 | 0 |  |
| 223.5 | +0.215 | 2nd diode cut-in |
| 228 | 0 |  |
| 235.5 | -0.28 | 3rd slope is parallel to Radiance Curve |
| 242 | 0 |  |
| 246.5 | +0.19 | 3 rd diode cut-in |
| 252 | 0 |  |
| 258 | -0.205 | 4th slope is parallel to Radiance Curve |
| 264.5 | 0 |  |
| 269.5 | +0.23 | 4th diode cut-in |
| 275.5 | 0 |  |
| 282 | -0.16 | 5th slope is parallel to Radiance Curve |
| 285.5 | 0 |  |
| 294 | +0.205 | 5th diode cut-in |
| 301 | 0 |  |
| 306 | -0.06 | 6th slope is parallel to Radiance Curve |
| 310 | 0 | End point adjusted to be an Ideal Curve |

The largest plus and minus errors in the $210 \mathrm{~K}-310^{\circ} \mathrm{K}$ range are $+0.23^{\circ}$ and $-0.28^{\circ} \mathrm{K}$ respectively.

The standard deviation $=0.15^{\circ} \mathrm{K}$ RMS over the $210^{\circ}$ to $310^{\circ} \mathrm{K}$ dynamic range.

TABLE 2.4.1.3-3
OLS \#12

## TARGET DEVIATION FROM MEAN OF BOTH TARGETS

| TARGET TEMP ( $\left.{ }^{\circ} \mathrm{K}\right)$ | DEVIATION ( $\left.{ }^{\circ} \mathrm{K}\right)$ |
| :--- | ---: |
| 210 | -0.01 |
| 220 | -0.07 |
| 230 | -0.02 |
| 240 | -0.01 |
| 250 | 0.03 |
| 260 | 0.01 |
| 270 | 0.02 |
| 280 | -0.01 |
| 290 | 0.04 |
| 300 | 0.02 |
| 310 | 0.09 |
|  |  |
| $R M S=\sqrt{\frac{\sum\left(\text { Dev. }{ }^{\circ} K\right)^{2}}{11}}=0.039^{\circ} \mathrm{K}$ |  |

## TABLE 2.4.1.3-4 <br> OLS \#12

BSL CALIBRATION EQUATIONS
(From Tables 2.4.1-4,5,6)

| $T$ FINE (Primary) |  |  | $\begin{array}{r} \text { EVA } \\ @ \quad 210^{\circ} \\ \hline \end{array}$ | uated $0310^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: |
| T-Right: | Error $=-0.0185(\mathrm{~T}-190)+1.34$ | $\left({ }^{\circ} \mathrm{K}\right)$ | +. 970 | -. 880 |
| T-Mid: | Error $=-0.0222(\mathrm{~T}-190)+1.94$ | $\left({ }^{\circ} \mathrm{K}\right)$ | +1.496 | -. 724 |
| T-Left: | Error $=-0.0148(\mathrm{~T}-190)+1.32$ | $\left({ }^{\circ} \mathrm{K}\right)$ | +1.024 | -. 456 |
| T FINE (Redundant) |  |  |  |  |
| T-Right: | Error $=-0.0215(\mathrm{~T}-190)+1.33$ | ( $\left.{ }^{\circ} \mathrm{K}\right)$ | +. 900 | -1.250 |
| T-Mid: | Error $=-0.0244(\mathrm{~T}-190)+1.85$ | (*K) | 1.362 | -1.078 |
| T-Left: | Error $=-0.0154(\mathrm{~T}-190)+1.16$ | ( ${ }^{\circ}$ ) | +. 852 | -. 688 |
| T SMOOTH (Primary | $\begin{aligned} & \text { Error }=-0.0224(T-190)+1.93 \\ & \text { SP MID) } \end{aligned}$ | ( $\left.{ }^{\circ} \mathrm{K}\right)$ |  |  |
| T SMOOTH (Redundan | $\begin{aligned} & \text { Error }=-0.0242(T-190)+1.85 \\ & - \text { SB MID) } \end{aligned}$ | ( ${ }^{\prime}$ ) |  |  |

## FIXED DEVIATION CALIBRATION DIFFERENCES FOR SEPARATE SEGMENTS

Calculated from Run \#12 BSL's in Table 2.4.1.3-4:

| DIFFERENCE <br> AT $210^{\circ} \mathrm{K}\left({ }^{\circ} \mathrm{K}\right)$ | DIFFERENCE <br> AT $310 \mathrm{~K}\left({ }^{\circ} \mathrm{K}\right)$ |
| :---: | :---: | | SPECIFICATION |
| :---: |
| $(\mathrm{MAX})$ |

PRIMARY

| T Mid to T Right | 0.53 | 0.16 | $1 \circ \mathrm{~K}$ |
| :--- | :--- | :--- | :--- |
| T Mid to T Left | 0.47 | 0.27 | $1^{\circ} \mathrm{K}$ |
| T Right to T Left | 0.05 | 0.42 | $1^{\circ} \mathrm{K}$ |

REDUNDANT

| T Mid to T Right | 0.46 | 0.17 | $1^{\circ} \mathrm{K}$ |
| :--- | :--- | :--- | :--- |
| T Mid to T Left | 0.51 | 0.39 | $1^{\circ} \mathrm{K}$ |
| T Right to T Left | 0.05 | $0.56^{*}$ | $1^{\circ} \mathrm{K}$ |

Worst Case Differences from Best Straight Line Plots (Figures 2.4.1-1 thru 2.4.1-12):

*WORST-CASE DATA

2-152

## TABLE 2.4.1.3-6 <br> OLS \#12

ALONG SCAN VARIATION ( $265^{\circ} \mathrm{K}$ to $310^{\circ} \mathrm{K}$ ) WITHIN A SEGMENT (From ASV Graphs)

| T-FINE | ERROR ( ${ }^{\circ} \mathrm{K}$ RMS) | LIMIT ( ${ }^{\circ} \mathrm{K}$ RMS) |
| :---: | :---: | :---: |
| T-Left Segment | 0.121 | 0.2 |
| T-Mid (Sum) Segment | 0.117 | 0.2 |
| T-Right Segment | 0.135 | 0.2 |
| T-SMOOTH |  |  |
| T-Sum | 0.117 | 0.2 |

## TABLE 2.4.1.3-7 <br> CONE COOLER S/N 024

| OLS-12 |  |
| :---: | :---: |
| CONE (INNER STAGE) PATCH TEMP. EST |  |
| TEMPERATURE $\cdot \mathrm{K}$ | PATCH EST, VOLTS |
| 95 | 5.655 |
| 96 | 5.248 |
| 97 | 4.874 |
| 98 | 4.529 |
| 99 | 4.212 |
| 100 | 3.920 |
| 101 | 3.651 |
| 102 | 3.403 |
| 103 | 3.174 |
| 104 | 2.963 |
| 105 | 2.768 |
| 106 | 2.588 |
| 107 | 2.422 |
| 108 | 2.268 |
| 109 | 2.125 |
| 110 | 1.993 |
| 111 | 1.871 |
| 112 | 1.757 |
| 113 | 1.651 |
| 114 | 1.553 |
| 115 | 1.437 |
| 116 | 1.377 |
| 117 | 1.298 |
| 118 | 1.225 |
| 119 | 1.156 |
| 120 | 1.092 |
| 121 | 1.022 |
| 122 | .976 |
| 123 | .924 |
| 124 |  |
| 125 | .875 |
|  |  |

TABLE 2.4.1.3-8
CONE COOLER OUTER STAGE TEMP EST
OLS \#12
T CONE TEMP EST (EST \#33)

| I (DEG K) | EST VOLTS | T (DEG K) | EST VOLTS |
| :---: | :---: | :---: | :---: |
| 158 | 4.8221 | 194 | 4.1282 |
| 159 | 4.8181 | 196 | 4.0328 |
| 160 | 4.8136 | 198 | 3.93 |
| 161 | 4.8088 | 200 | 3.8195 |
| 162 | 4.8035 | 202 | 3.7016 |
| 163 | 4.7978 | 204 | 3.5769 |
| 164 | 4.7915 | 206 | 3.4468 |
| 165 | 4.7848 | 208 | 3.3115 |
| 166 | 4.7775 | 210 | 3.1719 |
| 167 | 4.7695 | 212 | 3.0292 |
| 168 | 4.7609 | 214 | 2.8844 |
| 169 | 4.7515 | 216 | 2.7386 |
| 170 | 4.7414 | 218 | 2.5924 |
| 171 | 4.7306 | 220 | 2.4475 |
| 172 | 4.7188 | 222 | 2.305 |
| 173 | 4.7063 | 224 | 2.1659 |
| 174 | 4.6926 | 226 | 2.0302 |
| 175 | 4.678 | 228 | 1.8995 |
| 176 | 4.6622 | 230 | 1.7735 |
| 177 | 4.6454 | 235 | 1.4832 |
| 178 | 4.6273 | 240 | 1.2308 |
| 179 | 4.608 | 245 | 1.0159 |
| 180 | 4.5874 | 250 | 0.8359 |
| 181 | 4.5654 | 255 | 0.6873 |
| 182 | 4.5418 | 260 | 0.5650 |
| 183 | 4.517 | 265 | 0.4653 |
| 184 | 4.4904 | 270 | 0.3842 |
| 185 | 4.4622 | 275 | 0.3182 |
| 186 | 4.4323 | 280 | 0.2646 |
| 187 | 4.4008 | 285 | 0.2207 |
| 188 | 4.3673 | 290 | 0.1852 |
| 189 | 4.3322 | 295 | 0.1560 |
| 190 | 4.2951 | 300 | 0.1320 |
| 192 | 4.2156 | 305 | 0.1123 |

FIGURE 2.4.1.3-1






FIG. 2.4.1.3-5
ASV


SYSTEM 12, QATE: 707 TIME 831, SSS=5,M1=-8,TG=6, $\quad$, $=13$



SYSTEM 12 , DATE: 712 TIME 349 SSS $=5, M 1=12, T G=6, T L=8$


FIG. 2.4.1.3-8
ASV



FIG. 2.4.1.3-9
ASV

Ø. $\quad$.
SO 100
$4 . \square$
$8 . \square$

### 2.4 Radiometric Accuracy (Cont'd)

### 2.4.2 Davtime Radiometric Accuracy (3.2.1.1.4.2)

OLS \#12 achieved the 7\% absolute radiance requirement by setting the HRD channel gain as shown on the L channel DC Response plot, using the calibrated light source (VULS).

The analysis of the calibration accuracy and the L-Day channel stability show within-specification performance. The gain ratios (PMH/PML, PML/HRD, and PMH/HRD) were measured during bearing retrofit retest using the VULS during Acceptance Test in test $6 \times 2 \times 1.5 T$, and vary less than $0.3 \%$ from the average of the ratios. The gain ratios measured in test $6 \times 2 \times 2 . S T$ using a less accurate test method show greater variation.

The plot of L DC Response contains the calculated sensor switch points, $S(x)$, and relative gains $P(X)$, which are stored in the OLS Constants Memory page zero, locations 071 through 077. $\mathrm{P}(2)$ and $\mathrm{S}(2)$ represent the bypass of the PMT $1 / 9$ mode, which is not usually implemented on orbit.

OLS \#12 exhibited l.10dB drop in transmission from room temperature to $+5^{\circ} \mathrm{C}$. The correct light level corresponding to $2.12 \times 10^{-2} \mathrm{~W} / \mathrm{cm}^{2}-\mathrm{sr}$ is changed by $3 \%(0.26 \mathrm{~dB})$ relative to factory adjustment based on updated HRD spectral data used in the GAINSET program. Also, OLS \#12 exhibited a lower optics transmission than typical OLS units by 2 dB in the HRD channel. Thus $\mathrm{P}(0)$ must be reset to 6.0 (nom) $+1.10+0.26+2=9.36 \mathrm{~dB}$. Rounding off to the nearest $1 / 8 \mathrm{~dB}$ gives 9.375 dB as the new setting for $\mathrm{P}(0)$.

The Sl value used for OLS 12 DC response adjustment is 2.96 v . Also, the Gl value (HRD to PMT gain offset factor due to differing spectra) used was 4.37 db and the lunar/solar gain ratio used was 1.033 .

P1 is derived using the PMT LO/HRD average gain value of 49.98 dB with a compensation for the HRD Loss and PMT Gain Ratios with temperature from the bearing retrofit retest data as plotted in figure 2.4.2-2 and converted to dB of 1.10 dB and .61 dB , respectively. The Pl value is $49.98+0.61+1.10=51.69$ (rounded to nearest $1 / 8$ th $d B=51.75$ ).

| ATTACHMENT: | OLS \#12 | L Channel DC Response Plot |
| :--- | :--- | :--- |
|  | Table 2.4.2-1 | OLS \#12 DC Response Stability |
|  | Table 2.4.2-2 | OLS \#12 PMT/HRD DC Response vs. SSS Temp. |

Table 2.4.2-1
OLS \#12 L DC Response Stability

## Stability vs. Time ( $6 \times 2 \times 1$. ST data using VULS)

| DATE | $\frac{\text { PMT HI }}{\text { PMT LO }}$ | $\frac{\text { PMT LO }}{\text { HRD }} d B$ | $\frac{\text { PMT HI }}{\text { HRD }}$ |
| :---: | :---: | :---: | :---: |
| 08/10/91 | 29.74 | 49.98 | 79.71 |
| 08/14/91 | 29.73 | 49.95 | 79.68 |
| 10/26/91 | 29.73 | 50.15 | 79.88 |
| 10/27/91 | 29.71 | 49.82 | 79.53 |
| Average (Direct Multiple) | $\begin{aligned} & 29.73 \mathrm{~dB} \\ & (30.65) \end{aligned}$ | $\begin{gathered} 49.98 \\ (315.50) \end{gathered}$ | $\begin{aligned} & 79.70 \mathrm{~dB} \\ & (9660.51) \end{aligned}$ |

Stability vs. Temperature ( $6 \times 2 \times 2$. ST data using half sphere source)

| DATE | ENVIRONMENT | $\frac{\text { PMT HI }}{\text { PMT LO }}$ | $\frac{\text { PMT LO }}{\text { HRD }} d B$ | $\frac{\text { PMT HI }}{\text { HRD }}$ |
| :---: | :---: | :---: | :---: | :---: |
| 08/18/91 | TV Amb | 29.74 dB | 39.06 dB | 68.80 dB |
| DATE | ENVIRONMENT | $\frac{\text { PMT HI }}{\text { PMT LO }} \text { dB }$ | $\frac{\text { PMT LO }}{\text { HRD }}$ | $\frac{\text { PMT HI }}{\text { HRD }}$ |
| 08/22/91 | +5/-8 | 29.65 | 40.85 | 70.79 |
| 08/25/91 | +12/+15 | 29.69 | 40.23 | 69.93 |
| 08/26/91 | -2/-11 | 29.63 | 41.42 | 71.06 |
| 08/28/91 | +12/+15 | 29.76 | 40.44 | 70.19 |
| 08/30/91 | -2/-11 | 29.78 | 41.47 | 71.25 |
| 09/02/91 | +7/+12 | 29.68 | 40.66 | 70.35 |
| 09/07/91 | +3/-8 | 29.68 | 41.06 | 70.74 |
| 09/12/91 | +5/-8 | 29.69 | 40.89 | 70.58 |
| $\overline{\text { Average }}$ |  | 29.70 dB | 40.88 dB | 70.57 dB |

TABLE 2.4.2-2. PMT/HRD DC RESPONSE vs. SSS TEMPERATURE


### 2.4 Radiometric Accurack, (Cont'd)

### 2.4.3 Nighttime Radiometric Accuracy (3.2.1.1.4.3)

The PMT accuracy is required to degrade by no more than $60 \%$ from its initial accuracy at time of Acceptance testing to end of 3 year orbital life.

The 5D-3 OLS System Summary Report indicates PMT channel stability to be within $25.7 \%$ over the mission life so that the PMT meets this specification requirement. The above figures do not include corrections utilizing on-board LED calibration. The PMT CAL LED is extremely stable, and has an essentially constant output over the mission life.

The DC response curve of the OLS \#12 PMT is shown in the L Channel DC Response curve in paragraph 2.4.2. Unlike the HRD, the PMT optics transmission appears typical.

ATTACHMENT: Table 2.4.3-1 PMT CAL Baseline data (See para 2.4.2 attachment for L. Chan. DC Response curve)

TABLE 2.4.3-1
PMT CAL BASELINE DATA

The PMT Cal voltage EST is a monitor of PMT sensitivity and stability characteristics. In order to relate on-orbit measurement of PMT CAL to measurement during Acceptance Test, the following data are provided. PMT CAL Voltage (EST \#40) and PMT BU (Back-up) CAL V (EST \#41) are sampled 500 times in test $6 \times 6 \times 2 . S T$. The PMT Cal voltage EST output is tabulated below for all runs after PMT replacement.

TABLE 2.4.3-1
PMT CAL BASELINE DATA

| DATE | SSS TEMP | OUTPUT VOLTAGE (mV) |  |
| :---: | :---: | :---: | :---: |
|  |  | $\begin{aligned} & \text { PMT CAL V } \\ & \text { (EST \#40) } \end{aligned}$ | $\begin{aligned} & \text { PMT BU CAL V } \\ & \text { (EST \#41) } \end{aligned}$ |
| 08-09-91 | +25 | 2478 | 2475 |
| 08-14-91 | +25 | 2499 | 2487 |
| 09-03-91 | +12 | 2378 | 2372 |
| 09-08-91 | +3 | 2361 | 2356 |
| 09-14-91 | +5 | 2381 | 2376 |
|  | AVERAGE | 2419 | 2413 |
| Max change | m AVERAGE | 3.31\% | 3.07\% |

### 2.4 RADIOMETRIC ACCURACY. (Cont'd)

### 2.4.4 Gain Control Accuracy (3.1.4.4)

Along-Scan Gain Control (ASGC) accuracy is within the specification limit of 4 dB of the smooth monotonic curve drawn through the nominal values of Gain Value versus Scense Solar Elevation (GVVSSE), tabulated in para 3.2.1.1.1.4 of DMSS-0LS-300.

System Tests $5 \times 6 \times 3$.ST and $5 \times 5 \times 6$. ST exercise the ASGC function through various combinations of extremes of slope and bias adjustment; and measure channel output for a wide range of values. These are automatically compared against stored test limits which ensure spec compliance.

An analysis using the ASGC software algorithm showing 3.25 dB maximum error is summarized in the System Summary Report paragraph 3.5.4.

ATTACHMENTS: None.

### 2.4 RADIOMETRIC ACCURACY, (Cont'd)

### 2.4.5 Gain Control Adjustability (3.2.1.1.4.4 et al)

The OLS \#12 Gain Control Adjustability is the same as for 5D-1 systems. The OLS 5D-3 System Summary Report paragraph 3.5.5 contains the analysis required to demonstrate conformance with the specifications, with additional information below. TERMINATOR LOCATION (3.1.4.5.1)

The GNC Command (an uplink command) has a sub mode (GNC 100 X ) which allows the GVVSSE bias (terminator location) to be adjusted by $X$ degrees, where $X$ is a 6-bit 2's complement word with an LSB of $0.5^{\circ}$. This results in the required range of $\pm 15.5$ degrees.

GAIN CHANGE RATE (3.1.4.5.2)
The GNC command submode (GNC 101 X ) allowed the GVVSSE slope to be varied by a factor of $1+X$ over $\pm 48 \%$ in $1.6 \%$ increments where $X$ is a 6-bit $2^{\prime}$ s complement word with an LSB of $2^{-6}$. This results in the required range of $\pm 48 \%$. However, the BRDF change in the $L$ channel gain calculation required the deletion of this capability to meet timing limitations in the OLSP. The 886 spec must be revised to reflect this change.

MAXIMUM GAIN SETTING (3.2.1.1.4.5.3)
The maximum ASGC gain is commandable. An operational value is determined in Early Orbit Calibrations. The value is stored in the Operational Constants Memory location page $\theta$ Address 104 (BCMAX). The ASGC function \& performance are exercised in tests $5 \times 6 \times 3$.ST \& $5 \times 6 \times 6 . S T$

COMMANDABLE T-CHANNEL GAIN (3.2.1.1.4.5.4)
The T-Channel Commandable Gain is exercised in test $6 \times 8 \times 2$.ST. The channe 1 output is measured for the entire range of commandable gains. The T Channel Gain is required to have the capability of being varied by command to be between $146 \%$ and $149 \%$ of the established minimum gain value ( 0 dB ).

The actual percentage of TGAIN change was not measured as part of bearing retrofit. However, previously this was measured for OLS \#12 as $49.8 \%$ for T Right and $49.2 \%$ for T Left.

Each step of TGAIN is required to be between $1.7 \%$ and $3.7 \%$ above the preceeding lower gain value. Measured gain steps on OLS \#16 ranged from $1.86 \%$ to $3.47 \%$, within specification.

COMMANDABLE T CHANNEL LEVEL (3.2.1.1.4.5.5)
The T Channel Commandable Level is also exercised in test $6 \times 8 \times 2.5 T$. The specification requires that TLEVEL be variable over at least a $14^{\circ}$ range in steps of $1.1^{\circ} \mathrm{K}$ or less. Measured results during the original OLS 12 testing were $15.12^{\circ}$ range and step sizes between $0.969^{\circ} \mathrm{K}$ to $1.027^{\circ} \mathrm{K}$ worst-case; all within specification.

### 2.4 RADIOMETRIC ACCURACY, Cont'd

### 2.4.6 A/D Conversions \& Algorithms (3.2.1.1.4.6.2 \&

3.2.1.1.4.6.3)

DMSS-OLS-300 specifies that the Stored Smooth Algorithm accuracy with an ideal A/D shall be verified by analysis. OLS 5D-3 System Summary Report contains the analysis which shows that the LS \% Full Scale Deviation does meet the specification. The results are summarized below:


The Actual A/D Conversion Radiometric Accuracy was measured in system test $6 \times 10 . S T$. The worst-case results from the OLS \#12 Thermal Vacuum runs (Cold Limit \& Orbit Nominal) are summarized below:

| A/D | $\begin{gathered} \text { BSL SLOPE } \\ (\% \text { DEV FROM IDEAL) } \end{gathered}$ | BSL OFFSET <br> (\% OF FULL SCALE) | RMS DEV FROM BSL (\% OF FULL SCALE) |
| :---: | :---: | :---: | :---: |
| SDF-L PRIM | -0.33 | 0.18 | 0.04 |
| RED | -0.19 | 0.10 | 0.03 |
| SDF-T PRIM | 0.17 | -0.12 | 0.15 |
| RED | 0.38 | -0.16 | 0.17 |
| RTD-F PRIM | -0.33 | 0.18 | 0.04 |
| RED | -0.18 | 0.08 | 0.05 |
| SPEC | $\pm 1.0$ | $\pm 1.0$ | 0.5 |
| RTD-S PRIM | -0.23 | 0.08 | 0.02 |
| RED | -0.26 | 0.06 | 0.02 |
| SDS-L PRIM | -0.23 | 0.06 | 0.04 |
| RED | 0.20 | 0.06 | 0.06 |
| SDS-T PRIM | 0.10 | -0.30 | 0.06 |
| RED | 0.50 | -0.20 | 0.08 |
| SPEC | $\pm 0.5$ | $\pm 0.5$ | 0.25 |

### 2.5 RADIOMETRIC RESOLUTION (3.2.1.1.5 et al.)

DMSS-OLS-300 apportions the Radiometric Resolution verification between Test and Analysis.

The Fine and Direct Smoothed Radiometric Resolution (para.
3.2.1.1.5.1), as well as the Stored Smoothed A/D Converter Error (para. 3.2.1.1.5.2), were measured in System Test $6 \times 10$. ST during the OLS \#12 vacuum runs and are tabulated below:

PEAK DEV FROM BSL

| A/D |  |  |  |  | (\% OF FULL SCALE) | SPEC |
| :--- | :--- | :---: | :---: | :---: | :---: | :---: |
| SDF-L | PRIM | 0.07 | $\pm 0.8 \%$ |  |  |  |
|  | RED | -0.06 | $\pm 0.8 \%$ |  |  |  |
| SDF-T | PRIM | 0.33 |  |  |  |  |
|  | RED | -0.30 | $\pm 0.8 \%$ |  |  |  |
| RTD-F | PRIM | 0.08 |  |  |  |  |
|  | RED | 0.12 |  |  |  |  |
| RTD-S | PRIM | 0.04 |  |  |  |  |
|  | RED | -0.04 | $\pm 0.5 \%$ |  |  |  |
| SDS-L | PRIM | -0.07 |  |  |  |  |
|  | RED | 0.07 |  |  |  |  |
| SDS-T | PRIM | -0.10 | $\pm 0.5 \%$ |  |  |  |
|  | RED | -0.16 |  |  |  |  |

The Stored Smoothed Algorithms Resolution with Ideal A/D (para. 3.2.1.1.5.2.1) are verified by Analysis in OLS 5D-3 System Summary Report, and are summarized in Table 2.5.1.

TABLE 2.5-1
Stored Smoothed Algorithms Resolution With Ideal A/D

| ITEM | SPEC | ACTUAL |
| :--- | :--- | :--- |
| Accuracy |  |  |
| LS Algorithm | $< \pm 2.2 \%$ | $-1.70 \%$ |
|  |  | $+2.09 \%$ |
| TS Algorithm | $< \pm 0.4 \%$ | $\pm 0.39 \%$ |

Resolution
LS Algorithm
$<1.6 \% \quad 1.57 \%$
TS Algorithm

| Population 1 Density | $25 \%$ | $25 \%$ |
| :---: | :---: | :---: |
| Quantization | $<0.8 \%$ | $0.78 \%$ |
| Population 2 Density | $75 \%$ | $75 \%$ |
| Quantization | $<0.4 \%$ | $0.39 \%$ |
| Population Distribtution | Uniform | Uniform |
| Quantization Capability | $0.4 \%$ | $0.4 \%$ |

### 2.6 NOISE

### 2.6.1 T Channel Noise (3.2.1.1.6.1)

The $T$ Channel noise equivalent temperature difference (NETD) is specified between 210 K and 310 K , although the $T$ channe responsivity extends down to 190K.

The NETD is measured during T Channel DC response tests in the Thermal Vacuum Chamber.

The Channel views a stable blackbody target at 210 K (worst-case noise). The channel output is sampled \& the noise (std. deviation) of 2000 samples is converted to NETD using the following formula:

NETD $=$ [Avg Noise in Volts * 24 */Volt] * 1.074 (the shaper slope correction)

The OLS \#12 NETD is in-spec. The noise in the T Right segment is $11.8 \%$ larger than in the $T$ Left segment.

|  | TF | TS | TS Fallback |
| :---: | :---: | :---: | :---: |
| SPEC | $2.2{ }^{\circ} \mathrm{K}$ | $0.90^{\circ} \mathrm{K}$ | $1.3^{\circ} \mathrm{K}$ |
| Worst-Case Measured NETD | $0.695{ }^{\circ} \mathrm{K}$ | $0.279{ }^{\circ} \mathrm{K}$ | $0.393^{\circ} \mathrm{K}$ |
| Worst-Case Average NETD | $0.662^{\circ} \mathrm{K}$ | $0.273{ }^{\circ} \mathrm{K}$ | $0.370^{\circ} \mathrm{K}$ |

ATTACHMENT: Table 2.6.1-1 OLS \#12 Primary Side NETD
Table 2.6.1-2 OLS \#12 Redundant Side NETD

Table 2.6.1-1


* Worst Case Measured
** Shaper Slope Correction Factor $=1.074$

Table 2.6.1-2

*Worst Case Measured
**Shaper Slope Correction Factor $=1.074$

### 2.6 NOISE (Cont'd)

### 2.6.2 L-Channel Noise (Day) (3.2.1.1.6.2)

The L Channel Noise is measured using the calibrated Variable Uniform Light Source (VULS). Dark noise is measured in test $6 \times 3 \times 1 . S T$ and shot noise is measured in $6 \times 3 \times 5$.ST. The OLS \#12 HRD is in-spec for the entire range of illumination. Worst-case HRD SNR exceeds the specification. In summary:

| LIGHT LEVEL | SNR |  |  |
| :---: | :---: | :---: | :---: |
|  | SPEC | PRIOR OLS 12 TESTING WORST CASE MEASURED | RETEST WORSE CASE (FROM GRAPH) |
| $5.5 \times 10^{-5}$ | 5 | 7.45 | 7.8 |
| $5.5 \times 10^{-4}$ | 34.8 | 68.1 | 88.1 |
| $1.1 \times 10^{-3}$ | 62.3 | 116 | 140 |
| $2.2 \times 10^{-3}$ | 112 | 189 | 201 |
| $5.5 \times 10^{-3}$ | 200 | 251 | 235 |
| ATTACHMENT: | OLS \# | 12 HRD Channel SNR Gr | aph |



### 2.6 NOISE (Cont'd)

### 2.6.3 L Channel Noise (Night) 3.2.1.1.6.3)

The PMT dark noise is measured in all environments in Tests $6 \times 3 \times 1$.ST, $6 \times 3 \times 2$.ST and $6 \times 3 \times 4$.ST. The SNR is calculated from the measured noise (std. deviation of multiple voltage samples) vs.
light level and compared against spec values.
The minimum SNR from Bearing Retrofit retest is 7.8 at $8.0 \times$ $10^{-9}$ watts $/ \mathrm{cm}^{2}-s r$. The minimum SNR from Bearing Retrofit Retest is 16.0 . The worst case combined PMT shot noise and dark noise SNR from bearing retrofit testing is 7.01 calculated as SNR $=1 / \sqrt{1 /(\text { SNR dark })^{2}+1 /(\text { SNR shot })^{2}}$.

ATTACHMENT: OLS \#12 PMT channel SNR graph.

### 2.6 NOISE (Cont'd)

### 2.6.4 Dark Current (3.2.1.1.6.4)

The Dark Current (the PMT noise with no signal input) is determined from the graph of PMT SNR in paragraph 2.6.3. The Dark Noise SNR is calculated from data gathered during PMT Smoothed Noise measurements. These measurements are made in Test $6 \times 3 \times 1$. ST during Thermal Vacuum testing. For the OLS \#12 bearing retrofit retest, the average Dark Noise SNR of 5 measurements at $8 \times 10^{-9}$ watts $/ \mathrm{cm}^{2}-\mathrm{SR}$ is 16.2 , or $37.0 \%$ of the noise corresponding to an SNR of 6 . The MINIMUM Dark Noise SNR measured at $8 \times 10^{-9}$ watts $/ \mathrm{cm}^{2}$-SR was 16.0 , or $37.5 \%$ of the noise corresponding to a SNR of 6 . This is well within the spec requirement for the dark current to be $50 \%$ or less of the noise corresponding to an SNR of 6.00 .

### 2.6 NOISE, (Cont'd)

2.6.5 Stability (3.2.1.1.6.5) (L - Channel (night)

The OLS 5D-3 System Summary Report contains the analysis for this spec requirement.

The loss in sensitivity after 3 years on orbit of the PMT channel will be < $23 \%$. This would require 2.27 dB change in VDGA gain to compensate and over 17 dB of VDGA gain is available.

ATTACHMENT: None.

### 2.6 NOISE (Cont'd)

2.6.6 Along-Track Noise Integration (3.2..1.1.6.6)

OLS 50-3 System Summary Report contains the analysis which concludes that the OLS 5D-3 algorithm is consistently above 0.6 times the SNR resulting from perfect integration with 8 bit $A / D$ for $T$ Channel; and above $1 / \sqrt{2}$ times the SNR resulting from perfect integration with 6-bit A/D for L channel. Therefore, the Along-Track Noise Integration is in-spec.

ATTACHMENT: None.
2.6 NOISE, (Cont'd)
2.6.7 Glare Suppression (3.2.1.1.6.7)

OLS 5D-3 System Summary Report contains the analysis which verifies that the OLS does provide effective protection against solar glare for sun angles between $75^{\circ}$ and $95^{\circ}$.

ATTACHMENTS: None.

### 2.7 SURVIVABILITY (3.2.7)

The OLS 5D-3 System Summary Report contains calculations of survivability. See BVS 2353 (Verification of Survivability Requirements) for further details.

ATTACHMENTS: None.

### 2.8 SCAN ANGLE (3.2.1.1.8)

Tests $6 \times 7 \times 1$. ST and $6 \times 7 \times 3$.ST (End of Scan Vignette for HRD \& T channels respectively) measure the delphi number at which $1 \%$ vignetting of scene begins to occur. The measured delphis enable calculation of the altitude needed to obtain contiguous coverage at the equator. The contiguous coverage requirement is based on the average of $+Z$ and $-Z$ scan angles. For OLS \#12, the following results were obtained:

| CHANNEL | DELPHI | SCAN ANGLE | CONTIGUOUS COVERAGE ABOVE: |  |
| :--- | :---: | :---: | :---: | :---: |
| +Z HRD | +991.2 | $+55.97^{\circ}$ | $427.09 \mathrm{n} . \mathrm{mi}$. |  |
| -Z HRD | -990.2 | $-55.91^{\circ}$ | $428.25 \mathrm{n} . \mathrm{mi}^{\circ}$ |  |
| +Z T | +981.0 | $+55.39^{\circ}$ | $438.34 \mathrm{n} . \mathrm{mi}$. |  |
| -Z T | -986.0 | $-55.68^{\circ}$ | $432.70 \mathrm{n} . \mathrm{mi}$. |  |

Thus, both the HRD channel and the $T$ channel meet the requirements for contiguous coverage above 440 naut. mi, since both channels will provide contiguous coverage for all altitudes above 433.38 naut. mi.

ATTACHMENTS: None.

### 2.9 DATA COLLECTION RATE (3.2.1.1.9)

OLS \#12 does scan the field of view at the prescribed $11.88+/-$ . 12 Hz rate. This parameter is measured in Scanner Functional tests $5 \times 12 \times 1$. ST (Primary Side) and $5 \times 12 \times 2$.ST (Redundant Side). The test results are summarized below for all TV tests:

| Date | Frequency, Hz <br> Redundant |  |
| :--- | :---: | :---: |
| 07-06-91 <br> Optic Limit | 11.90 | 11.91 |
| 09-02-91 <br> Hot Limit | 11.89 | 11.89 |
| 09-06-91 <br> Cold Limit | 11.91 | 11.90 |
| 09-11-91 <br> Orbit Nom. | 11.90 | 11.90 |

ATTACMENTS: None.

### 2.10 POWER (3.3.1 and 3.3.2)

Both +28 V and +5 V power is measured and monitored continuously throughout all of the test sequence.

The power required in the 8 Development-Spec-defined modes is tabulated below. The Development Spec Power Profile is measured in test $5 \times 2 \times 11$. ST for modes 1 through 8 and $5 \times 2 \times 2$.ST for mode 0 .

10V power consumption was not tested on OLS \#12. The current system test equipment is not capable of monitoring 10 V power. Analysis of the components using S/C supplied 10 V power indicates that 5D-3 10V power consumption is in spec.

DMSS-OLS-300 limits 28 V power consumed for SSS thermal control to 23 watts maximum. SSS heater power consumption was not measured on OLS \#12. Analysis of the heater resistances and tolerances indicates that 5D-3 SSS heater power consumption is in spec.

OLS \#12 28 V power consumption is in spec for all modes in the primary and redundant configurations. Fallback (dual power) configuration power consumption is also in spec. See the attached table for further details.

ATTACHMENTS: OLS \#12 Power Profile

## OLS \#12 POWER PROFILE

| SINGLE POWER |  |  |  | DUAL POWER |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| MODE/LIMIT | $\begin{gathered} \text { TV } \\ +5 /-8 \\ 07-04-91 \\ \hline \end{gathered}$ | $\begin{gathered} \text { TV } \\ \text { HOT LIMIT } \\ 09-01-91 \\ \hline \end{gathered}$ | $\begin{gathered} \text { TV } \\ \text { COLD LIMIT } \\ 09-06-91 \\ \hline \end{gathered}$ | $\begin{aligned} & 28 \mathrm{~V} \\ & \text { LIMIT } \end{aligned}$ | WORST CASE (CALCULATED) |
| 0 88W | 53 | 53 | 53 | 131W | 88 |
| 1 105W | 82 | 82 | 81 | 148W | 117 |
| 2116 W | 88 | 89 | 88 | 159W | 124 |
| 3 125W | 95 | 96 | 93 | 168W | 131 |
| 4 157W | 131 | 131 | 128 | 200W | 166 |
| 5167 W | 137 | 139 | 135 | 210W | 174 |
| 6 198W | 166 | 167 | 162 | 241W | 202 |
| 7 207W | 178 | 180 | 172 | 250W | 215 |
| 8 218W | 185 | 187 | 178 | 261W | 222 |
| $\begin{gathered} 5 \mathrm{~V} \\ \text { MODE/LIMIT } \\ \hline \end{gathered}$ |  |  |  |  |  |
| 0 4.3W | 3 | 3 | 3 |  |  |
| 1 4.3W | 3 | 4 | 3 |  |  |
| 24.3 W | 3 | 4 | 3 |  |  |
| 3 4.3W | 3 | 4 | 3 |  |  |
| 4 4.3W | 3 | 4 | 3 |  |  |
| 5 4.3W | 3 | 4 | 3 |  |  |
| 64.3 W | 4 | 4 | 3 |  |  |
| 7 4.3W | 4 | 4 | 3 |  |  |
| 8 4.3W | 3 | 4 | 3 |  |  |

### 2.11 MASS

2.11.1 Total Mass (3.4.1)

The weights of all OLS \#12 components were not measured as part of bearing retrofit. The data taken on 12-03-86 during the original OLS 12 selloff are provided for reference. The tape recorder and encrypter serial numbers are those belonging to the system at OLS \#12 sell-off and may change.

All Westinghouse furnished parts meet their center of gravity specification limits and their maximum specified weight allocation. All Typical encrypters exceed the spec limit in center-of-gravity. The encrypters are GFE to WEC and their weight and C.G. are not controlled by WEC.

The total weight of the OLS \#12 AVE is 291.01 pounds, (less BBX's, but including GSSA/DOC \& Test Cable), vs. a spec limit of 298 pounds.

ATTACHMENT: OLS \#12 Weight and Center-of-Gravity Tables
TABLE 1
WESTINGHOUSE FURNISHED PARTS SUPPLIED WITH OLS 12 SYSTEM

|  |  |  | $\bar{x}$ |  |  | $\gamma$ |  |  | I |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| unit | SER．no． | SPEC | MPR | ACT | SPEC | MPR | ACt | SPEC | HPR | Act | $\begin{aligned} & \text { MAX* } \\ & \text { SPEC } \end{aligned}$ | $\begin{gathered} \text { MPR** } \\ \mathrm{W} / \mathrm{OCONT}^{2} \\ \hline \end{gathered}$ | $\begin{aligned} & \text { HPR** } \\ & { }^{\prime} \text { COMS } \end{aligned}$ | ACt |
| sss | 5007 | 1．8土．5 | 1．8さ．5 | 1.86 | 6．2土．5 | 6．2土．5 | 6.10 | $0.7 \pm .5$ | $0.7 \pm .6$ | 0.59 | 59.0 | 53.29 | 54.35 | 54.64 |
| SPS | 5007 | $3.0 \pm .5$ | 3． $0 \pm .5$ | 2.94 | 13．8¹．0 | $13.8 \pm 1.0$ | 13.76 | $8.6 \pm .8$ | 8． $6 \pm .8$ | 8.56 | 70.0 | 68.15 | 69.50 | 69.00 |
| SPU | 5007 | 3．0土．5 | 3．0．$\pm .5$ | 3.00 | 6．6さ． 5 | 6．6さ． 5 | 6.55 | 6． $0 \pm .5$ | $6.0 \pm .5$ | 5.84 | 18.0 | 17.00 | 17.34 | 17.32 |
| PSU | 5007 | 2．3土． 5 | 2．3．5 | 2.78 | $7.0 \pm .6$ | 7．0土． 6 | 6.72 | $7.2 \pm .5$ | 7．2£． 5 | 7.15 | 27.0 | 25.60 | 26.10 | 26.31 |
| OSU | 5007 | $1.2 \pm .25$ | $1.2 \pm .25$ | 1.27 | 4．0さ． 5 | $4.0 \pm .5$ | 4.35 | 3．0£． 5 | 3． $0 \pm .5$ | 2.72 | 4.0 | 3.47 | 3.53 | 3.52 |
| GSSA／DDC | 5007 | 4．2さ．5 | 4．2 $\pm .5$ | 4.11 | $+0.1 \pm .3$ | $+0.1 \pm .3$ | 0.15 | 2．4土 ． 5 | 2．4土． 5 | 2.37 | 9.0 | 7.83 | 7.99 | 8.10 |
| PR1 | 040 | 3．45さ． 25 | 3．45さ． 25 | 3.29 | $6.36 \pm .25$ | 6． $36 \pm .25$ | 6.13 | 4． $23 \pm .25$ | 4．23£． 25 | 4.14 | 22.75 | 21.14 | 21.57 | 21.46 |
| PR2 | 041 | 3．45t． 25 | 3．45土． 25 | 3.38 | 6．36さ． 25 | 6．36 $\pm .25$ | 6.38 | $4.23 \pm .25$ | $4.23 \pm .25$ | 4.28 | 22.75 | 21.14 | 21.56 | 21.44 |
| PR3 | 042 | 3．45さ． 25 | 3．45土． 25 | 3.32 | $6.36 \pm .25$ | 6．36 .25 | 6.19 | 4． $23 \pm .25$ | 4．23£． 25 | 4.29 | 22.75 | 21.14 | 21.56 | 21.40 |
| PR4 | 043 | $3.45 \pm .25$ | 3．45£． 25 | 3.40 | $6.36 \pm .25$ | 6．36 $\pm 25$ | 6.28 | 4．23土． 25 | $4.23 \pm .25$ | 4.30 | 22.75 | 21.14 | 21.56 | 21.33 |
| cables | （1） | － | － | － | － | － | － | － | － | － | 22.0 | 20．88 | 21.30 | 20.49 |
| test |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| cable | （2） | － | － | － | － | － | － | － | － | － | 6.0 | 6.0 | 6.0 | 6.0 |
| ＊DMSS－DLS－300，SCN 011． 20 Nov． 1987 <br> ＊＊503 Mass Properties Report， 18 Hov． 1988 |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  | 298 | 286.78 | 292.36 | 291.01 |

TABLE 2
government furnished parts supplied with ols 12 SYSTEM


### 2.11 MASS (Cont'd)

2.11.2 Component Mass (3.4.2, 3.4.3)

The mass of the individual components of the OLS \#12 AVE are tabulated below.

| Component | Spec | Measured |
| :---: | :---: | :---: |
| SSS | 59.0 | 54.64 |
| SPS | 70.0 | 69.00 |
| SPU | 18.0 | 17.32 |
| PSU | 27.0 | 26.31 |
| OSU | 4.0 | 3.52 |
| GSSA/DOC | 9.00 | 8.10 |
| PRI | 22.75 | 21.46 |
| PR2 | 22.75 | 21.44 |
| PR3 | 22.75 | 21.40 |
| PR4 | 22.75 | 21.33 |
| BB1 | 3.67 | 3.62 |
| BB2 | 3.67 | 3.66 |
| BB3 | 3.66 | 3.72 |
| Cables | 32.00 | 20.49 |

The cable figure does not include Special Sensor cables which are not supplied by WEC.

### 2.12 COOLER TRANSIENT MARGIN (3.2.2.5)

The $T$ channel cone cooler transient cooldown cooling capacity margin is tested by the file $7 \times 7 . S T$. An external power supply provides $1 / 2$ Watt of heating to the outer stage of the cooler during a normal cooldown. On OLS \#12 bearing retrofit, cone cooler S/N 024 successfully reached its operating set-point with $1 / 2$ watt of external power applied, demonstrating the required margin.

## ATTACHMENTS: None

### 2.13 DESIGN FEATURES

The following design features of the 5D-3 OLS are addressed in the analyses. The analyses are contained in the OLS 5D-3 System Summary Report. The requirements of the Design Features are met or exceeded in each category.

SUBJECT
DESIGN INTERFACES
RELIABILITY
WEAROUT/CONSUMPTION
STORAGE
CONTAMINATION CONTROL
CORROSION OF METAL PARTS
MAINTAINABILITY
INTERCHANGEABILITY

SPEC. PARA.
(3.1.2)
(3.2.3)
(3.2.3.3)
(3.2.3.4)
(3.2.3.5)
(3.2.3.6)
(3.2.4)
(3.3.5)

ATTACHMENTS: None.

### 2.14 REDUNDANT AND FALLBACK SUBSYSTEMS (3.2.3.1 \& 3.2.3.2)

Paragraph 3.2.3.1 of the development spec requires the OLS to incorporate the following redundant subsystems:
(1) Along-Scan Gain Control
(2) Main Bus Power Supplies
(3) Data Processors
(4) Memories
(5) I/0 Interfaces
a.
Bus Controls
b.

I/O Controls
c.

S/C Interfaces
d.

WOW/Flutter Signals/Clock Drivers
e.

Drive Motor Controls
f.

Sensor Controls
g.

Gain Controls
h.

Encoder Processors
(6) Data Channels
a.
Smooth Video Filters
b.
Fine Video Filters
c.
d.
SDS Channel
e.
SDF Channel
f.
RTD Channel
g.
Special Sensor Processors
h.
Output Data Multiplexers
(7) Output Switching Unit Oscillator and Clock Circuits
(8) Digital Tape Recorders - (Three of Four Required)
(9) Output Data Channels - (Three of Four Required)

Paragraph 3.2.3.2 requires the OLS to incorporate the following fallback subsystems:
(1) IMC Shut-Off Mode.
(2) HRD - Detector Single Segment Select, with associated electronics.
(3) T-Detector Single Segment Select, with associated electronics.
(4) Digital Generation of Delphi Scanner Clock.
(5) PMT Shut-Off Mode

Redundant and fallback subsystems are verified by test during the normal test flow either by repetition of the relevant test on the alternate subsystem (in the case of a redundant subsystem) or by tests designed to verify the specific subsystem (in the case of a fallback subsystem). In each case, any out-of-specs or anomalies are reported as part of the relevant Test Report paragraph.

The following enviromental requirements are addressed in analyses and are contained in the OLS 5D-3 System Summary Report. The Thermal Vacuum (except CHA), Random Vibration, and shock requirements are verified by the sucessful completion of the approved Test Procedure.

The 5D-3 Environmental requirements of the Development Specification are met or exceeded by the OLS \#12 AVE.

| SUBJECT | SPEC. PARA |
| :--- | :--- |
| GROUND ENVIRONMENT | 20.2 .1 |
| LAUNCH ENVIRONMENT | 20.2 .2 |
| THERMAL VACUUM (CHA) | 20.2 .2 .1 |
| ACCELERATION | 20.2 .2 .5 |
| CHARGED PARTICLE ENVIRONMENT | 20.2 .2 .6 |
| LAUNCH PRESSURE PROFILE | 20.2 .2 .8 |
| ACCOUSTIC FIELD | 20.2 .2 .9 |
| TRANSPORATION \& HANDLING ENVIRONMENT 20.2 .3 |  |

ATTACHMENTS: None.

### 2.16 ELECTROMAGNETIC COMPATABILITY (3.3.2)

OLS \#12 EMC testing was conducted per Westinghouse documents BVS 2049 (Block 5D-3 Electromagnetic Compatability Test Plan) and T928546 (Block 5D-3 Electromagnetic Interference Test Procedure) during the original OLS \#12 testing. This testing was conducted in two phases. Phase one consisted of testing with the BTM SSS in the unpowered launch configuration on $3 / 4 / 85$ thru $3 / 9 / 85$. Phase two consisted of testing using the OLS \#12 system conducted 8/22/85 thru 8/29/85. The results of this testing are reported separately in Volume $V$ of the original Qualification Test Report. A summary of the EMC test results is included in table 2.16.1 - EMC Test Results. OLS \#12 meets all DMSS-OLS-300 EMC requirements.

ATTACHMENTS: Table 2.16.1 - EMC Test Results

TABLE 2.16.1

## ELECTROMAGNETIC COMPATABILITY

TEST RESULTS

| Test | T928546 <br> Test Procedure <br> Section | Data Complete | Result |
| :--- | :---: | :---: | :--- |$|$| Expose unpowered |
| :--- |
| BTM SSS with all <br> covers and room <br> Temperature T <br> Detector <br> installed to 200 <br> V/m for 5 minutes |

### 3.0 INTERFACE SPECIFICATION REDUIREMENTS

Electrical Interface parameters are measured in the OLS Detailed Electrical Test (T927989) and the OLS Special Sensor Detailed Electrical Test (T927992). These tests demonstrated conformance with all applicable Interface Specification requirements. The only Interface related system measurements that vary significantly from system to system are the SSS Alignment axes which are included here.

### 3.1 SSS ALIGNMENT AXES

The OLS \#12 SSS Reference Plane axes are within the specification allowances. The results of system test and calculation are given below. The designations are those in Interface Spec IS-YD-810, para. 3.2.7ff.

## SECONDARY REFERENCE AXES

 IO PRIMARY AXES
## SECONDARY REFERENCES AXES

$X_{R-p}=0.764 \mathrm{mrad}=158 \mathrm{arc} \mathrm{sec}$ TO MOUNTING (INTERFACE) AXES
$Y_{R-P}=0.467 \mathrm{mrad}=96 \mathrm{arcsec}$ $Z_{R-p}=0.626 \mathrm{mrad}=129 \mathrm{arc} \mathrm{sec}$ $X_{R-M}=0.780 \mathrm{mrad}=161 \mathrm{arc} \sec$ $Y_{R-M}=0.496 \mathrm{mrad}=102 \mathrm{arc} \mathrm{sec}$ $Z_{R-M}=0.658 \mathrm{mrad}=136 \mathrm{arcsec}$

These are within the specification limits of 600 arc seconds. The Mounting (Interface) Axes to Primary axes are also calculated, using the computer program REFPLN and are given below:

$$
\begin{aligned}
& X_{\text {H-P }}=0.027 \mathrm{mrad}=67 \mathrm{arc} \mathrm{sec} \\
& Y_{\mathrm{H}-\mathrm{P}}=0.272 \mathrm{mrad}=56 \mathrm{arc} \mathrm{sec} \\
& Z_{\mathrm{M}-\mathrm{P}}=0.270 \mathrm{mrad}=56 \mathrm{arc} \mathrm{sec}
\end{aligned}
$$

These are within the specification limits of 120 arc seconds.
olsfRICuTIOM

J．Spengler
M．Barrett
5．Nichols
M．Little
R．Bark
R．Rum
P．Kiefer
V．Williams
6．Pollock
B．Spencer
R．Lieske
ل．Scilipoti


BUS 2579
－DATE 05 November 1991 ORIGINATOR 2 APPROVED，Q\＆RA RUB， APPROVED，ENGRG

REVISION


## APPENDIX A

# BEARING RETROFIT <br> AND RETEST PLAN FOR ILS 12 THRU 16 

Support and Services Contract No．F04701－90－C－0028

Prepared for
UNITED STATES AIR FORCE Headquarters，Space Systems Division

Los Angeles，California

Prepared by
WESTINGHOUSE ELECTRIC CORPORATION Defense and Electronics Center

Baltimore，Maryland

## REVISION SHEET \& NOTES PAGE

NOTICE: Unless otherwise instructed, the marked-up pages showing actual changes incorporated in a new Rev. will be maintained in the BVS Master File for future reference and the remainder of the document will be discarded when the new Rev. is filed.

| $\begin{aligned} & \overline{\text { REVISION }} \\ & \text { LETIER } \\ & \hline \end{aligned}$ | $\begin{gathered} \text { REVISION } \\ \text { DATE } \\ \hline \end{gathered}$ | AFFECTED PAGES | $\begin{aligned} & \text { REVISION } \\ & \text { MADE BY } \end{aligned}$ |
| :---: | :---: | :---: | :---: |
| A | 4/6/90 | 1,2,*3-6, 7-11,*12, 13-16,*17-20, 21, 22 | J. SMUTKO |
| B | 8/7/90 | 1, 2, 16, 16a | D. OMETZ |
| C | 8/26/90 | -1, 2, 16b, 17 | J. SMUTK0 |
| D | 9/21/90 | $1,2,5,6,18,21$ | G. POLLOCK |
| E | 2/20/91 | 1, 2, 17-23 | R. BARK |
| F | 2/28/91 | 1,2,17 | R. BARK |
| G | 5/20/91 | 22, 23 | M. BARRETT |
| H | 5/22/91 |  | M. Barrett |
| $J$ | 6/28/91 | 1-3, 22-23 | SCILIPOTI |
| $k$ | 8/29/91 | $1-2,18,22-23$ | SCILIPOTI |
| L | 11/05/91 | $1,2,23,24$ | G. POLLOCK |

(* INDICATES PAGE \# IS ONLY CHANGE TO THAT PAGE)

Notes:
WP51 \JSm01.di

### 1.0 INTRODUCTION

This document describes the detailed rework and test verification plan for replacing the oscillating assembly bearings in the SSS with new bearings having improved lubrication (Ny 188B).

Included are step-by-step instructions, with check-off lines for all inspection, mechanical, optical and electrical test operations.

### 2.0 REFERENCES

This document references the following Westinghouse procedures:
$9 R A 3681$ SSS Assembly Procedure
$9 R A 4026$ SSS Handling Procedure
9TA9354 Mechanical Operations for SSS Oscillating Assembly Bearing Retrofit

T927002 SSS/DME Compatibility Test Specification
T927686 OLS System Acceptance Test Procedure
OLS Program Directives:
PD 024
PD 026
PD 027
PD 030
PD 044
PD 045
PD 055
SQL 0735 Vibration Procedure

### 3.0 REWORK AND TEST PLAN FOR ILS:

$\qquad$
$\Delta$ 3.1 Charge labor for this effort to the Block 5 Support and Services contract. Present G.0. number, valid thru 9/30/91, is 53741. Task assignments are as follows:

BAAA Non-recurring Engineering
BABA OLS-12
BALA OLS-14
BALA OLS-15
BALA OLS-16
Material and Travel G.0. is 53742.
3.2 The modification is accomplished by working revision notice G931B. This consists of replacing the two bearing pairs in oscillating Assembly 623R765.

Special instructions have been written to supplement the ANs and describe the mechanical operations necessary to retrofit scanner bearings after an SSS has been fully assembled, see 9TA9354.
3.3 For those systems in the field, return the system to WEC, Baltimore for rework and retest. Follow all applicable handing procedures including Program Directives 024, 026 and 027. As an option, the PSU and SSS only may be returned if another system is available to support the retest effort.
3.4 INCOMING INSPECTION AND SYSTEM TESTS.

OPERATION
Unpack
Record serial nos. of reed. units:


Attach copy of incoming
DD1149 to this BVS for control purposes

WEC Receiving Inspection
AFPRO Inspection

## Baseline Electrical Tests - Deleted

### 3.5 OPTICAL ALIGNMENT BASELINE MEASUREMENTS

Prior to bearing retrofit certain optical tests must be performed in order to accumulate baseline data with which to compare readings taken after the work is completed. This is necessary so that alignment integrity can be verified after the SSS has been partially disassembled and reassembled. These tests will be the same as some of the tests performed in 9RA3681 "Assembly and alignment procedure SSS assembly". However, there may be slight differences in technique because the assembly status of the SSS will not be exactly the same as in the normal building sequence. The steps herein, then, will be excerpts from that procedure and all step numbers referred to will be taken from 9RA3681.

Data should be noted in the applicable flight log book and used for post retrofit alignment comparisons.

In order to perform the necessary tests, the PMT and HRD detector must be removed. The spring assemblies must also be tied in order to permit positioning of the telescope.

## STEPS FROM 9RA3681 TO BE PERFORMED

## STEPS

VERIFICATION DATE
"Adjustment of optical alignment, test and integration facility" prepares the facility for required tests.
"SSS assembly mounting procedure" Mounts the SSS to the test facility reference interface for testing.
"Alignment of the oscillating assembly rotating axis with the Moore table axis." - positions the SSS for optical measurements.
"Mounting interface alignment measurements" - Determines SSS reference axis position in relation to the OATIF mirrors/SSS mounting interface.


A test will be performed to determine the position of the HRD detector prior to removal as follows:
Clamp a Gaertner bench microscope to the $T / T$ table aligning the microscope reticle with the T/T reticle. Observing the HRD detector through the microscope, center the reticle on at least 2 corners of each segment of the detector. Note the $\mathrm{T} / \mathrm{T} \mathrm{Y}$ and Z axis positions for each point observed.

"Oscillating assembly transmission test" - Determines \% transmission of telescope prior to retrofit.

Inspect mirror M1 to determine if a scatter test should be performed. If, on inspection, Ml appears very dirty perform a scatter test per step 19 of 9RA3681.

$30 \cdot+40$

## STEPS FROM 9RA3681 TO BE PERFORMED

STEPS
20.4 "HRD detector alignment check" - checks alignment of the HRD in relation to the ORA field splitter. Illumination via the PMT light as described in step 20.4 .5 should not be necessary. If the light from the T/T point source is insufficient to view the HRD the PMT must be removed prior to performing step 20.4.

Before proceeding remove the HRD and PMT if still installed. Mount the PMT and HRD reticles.
14.2 to 14.3 "M1 centering test" - verifies that the optical beam is centered on M1 prior to retrofit.
16.1.1 to
16.1.4, 16.1.10, \& 16.1.19 to 16.1.26
16.2.1 to 16.2.4, 16.2.11, 16.2.20 to 16.2.27, \& 16.2.29
15.2.2 to 15.2 .7 \& 15.2.16 to 15.2.18
15.2.19.6 to 15.2.19.8
"T-Cal alignment..." Determines proretrofit T-Cal end of scan position. Make no adjustments.
"T-Clamp alignment..." - Determines pereretrofit T-Clamp end of scan position. Make no adjustments. "Encoder optics alignment" - Deter facet from 15.1.15 referred to in steps 15.2 .6 and 15.2 .7 will be assumed to be facet 8 .

Defines minimum allowable voltages and angular displacement on the faceted

VERIFICATION DATE
 ring:


## STEPS

VERIFICATION
DATE
15.3.1 to "Encoder nadir adjustment" - Determines 15.3.6 Note: Make NO adjustmint in
15.3.6
15.3.12
15.4.1 to 15.4.26

Omit steps
15.4.18, 15.4.19, 15.4.20 \& 15.4.23
15.5

Note: Make
NO adjust-
mints in
15.5.12,
15.5.13 or
15.5.16
15.6 .1 to 15.6.3
15.9

Note: Make NO adjustmint in 15.9.13 or 15.9.16
"SSS Preparation for primary aux. encoder alignment." - Prepares SSS for testing of primary aux. encoder.
"Primary aux. encoder alignment" Performs tests to determine electrooptical position of the primary aux. encoder.

"Back-up auxiliary encoder alignment."

"Encoder nadir alignment error" -
Provides a formula for determining nadir alignment error.
"Encoder linearity and signal amplitude measurements." - Determine pre-retrofit position for facets of the encoder $W / R$ to the target translator. In steps 15.4.17, 15.4.22 and 15.4 .25 only a sampling of the numbered pulses shall be taken. Sample pulses 15,60 , 97 and 142.

- Determines pre-retrofit electrooptical position of the back-up aux. encoder.




## STEPS FROM 9RA3681 TO BE PERFORMED

15.1.1 to
15.1.15 Omit step 15.1.14
"Faceted ring angular measurements" - Determines optical positions of the facets of the polygon ring. In steps 15.1.4, 15.1.8, 15.1 .10 and 15.1.13 where the step refers to specific facets, it shall be required to perform the procedure only on a sample of the facets. A facet shall be chosen at the beginning, the middle and as near to the end as can be seen. Perform the procedure on facets 1,8 and 14 if these are accessible. Step 15.1.14 will be omitted and in step 15.1.15 the facet closest to the mean facet $Y$ axis reading will be assumed to be facet 8.

Cover the HRD and PMT reticles and ORA parts with lens tissue. Data should be noted in the applicable flight log book and used for post retrofit alignment comparisons.

$100 c t a$

### 3.6 MECHANICAL OPERATIONS

Perform the operations outlined in 9TA9354. This procedure describes all the mechanical operations necessary to retrofit scanner bearings after an SSS has been fully assembled. This step-by-step procedure includes check-off lines for each operation and inspection point.

Following completion of the procedure, attach the working copy of $9 T A 9354$ to this BVS.

Verification of completion
Inspection


1-.78-91
$2 / 22 / 91$

### 3.7 OPTICAL RE-ALIGNMENT

Optical Tests after Bearing Retrofit
After the bearing retrofit certain optical tests from 9RA3681 must be performed both for comparison to baseline tests as well as to ensure the unit is ready for integration tests. The bulk of these post bearing retrofit tests are the same as the optical baseline tests discussed in section 3.5. Record data in the applicable system SSS log book.

In order to perform the necessary tests, the PMT and HRD detector must be removed. The spring assemblies must also be tied in order to permit positioning of the telescope.

## STEPS FROM 9RA3681 TO BE PERFORMED

VERIFICATION
DATE

8.0 | "Adjustment of optical alignment, |
| :--- |
| test and integration facility"- |
| prepares the facility for required |
| tests. |


9.0 "SSS assembly mounting procedure" Mounts the SSS to the test facility reference interface for testing.
11 "Alignment of the oscillating assembly rotating axis with the Moore table axis." - positions the SSS for optical measurements.
"Mounting interface alignment measurements" - Determines SSS reference axis position in relation to the OATIF mirrors/SSS mounting interface.

20 NaCl 96

20 Nous 90
13.0 to 13.3, "M3, M5 and M5 Mask Alignment" -
18
"Oscillating assembly transmission test" - Determines \% transmission of telescope.

20 Now. 90
Inspect mirror Ml to determine if a scatter test should be performed. If, on inspection, Ml appears very dirty perform a scatter test per step 19 of 9RA3681.

201604.90

BUS 2579

STEPS
"HRD detector alignment check" - checks
alignment of the HRD in relation to the
ORA field splitter. Illumination via
the PMT light as described in step
20.4 .5 should not be necessary. If the
light from the T/T point source is in-
sufficient to view the HRD the PMT must
be removed prior to performing step
20.4.
Before proceeding remove the HRD and PMT if still installed. Mount the PMT and HRD reticles.
14.2 to 14.3 "Ml centering test" - verifies that the optical beam is centered on M1.
"T-Cal alignment..." Determines T-Cal end of scan position. Make no adjustments.
"T-Clamp alignment..." - Determines T-Clamp end of scan position. Make no adjustments.
16.1.1 to
16.1.4, 16.1.10, \&
16.1.19 to 16.1.26
16.2.1 to 16.2.4, 16.2.11,

## VERIFICATION

DATE 16.2.20 to 16.2.27, \& 16.2.29

## LeA



20110190


## STEPS FROM 9RA3681 TO BE PERFORMED



## STEPS FROM 9RA3681 TO BE PERFORMED

STEPS
VERIFICATION
DATE


### 3.7.1 PMT ASSEMBLY SPECTRAL RESPONSE STABILITY CHECK - OLS-16 Only

With the PMT assembly removed during the optical re-alignment, a check of the PMT spectral response will be performed to check spectral stability for any evidence of a shift since the last PMT spectral response made on 07/30/88.

The test will be performed in accordance with T-361A88, test paragraph 9.12 - Spectral Response and Effective Sensitivity.

## STEPS TO BE PERFORMED

## STEPS

## Inspection of PMT (Damage Verification)

## Install in Transport Case

Spectral Response Test from T-361A88, Para. 9.12
Inspect PMT prior to SSS Installation for Damage
Reinstall PMT on SSS
Inspection (W \& DPRO)
3.7.2 THERMAL BLANKET UPGRADE PER ECP-25-OLS-16 only

In place of the thermal blanket hardware originally installed on OLS-16, install the following oscillating assembly insulation covers and



## In pla 0LS-16, ins insulation:

540R561G01
540R561G02
540R562H01
54OR562HO2
540R563HiO1
Inspection
-16a-
BVS 2579

## 3.7 .2 (Cont'd.)

Install ECP-25 upgraded bracket on the 1A8 HRD/PMT Postamplifier Assembly as follows:

## STEPS

Remove the cover assembly, 644R288, from the HRD/PMT Postamplifier, 644R220 located on the SSS Be careful not to disturb the potentiometer adjustments. It may be necessary to cut the RTV used to stake the potentiometers if it has adhered to the cover.

On the cover assembly, 644R288G01, replace mount 522R838GO1 and 432R269G01 with items 16 and 17 respectively on the Thermal Blanket Retrofit drawing, 765R630. (540R564G01 and 540R584GO1) Re-mark the cover assembly to 644R288G02.

Inspect modified cover
Inspect 644R220, OK to reinstall cover, WEC \& DPRO Install modified cover on 1A8 postamplifer

## Inspection

(Note - retest of 644R220 postamplifier not required, will be tested at system level).

VERIFICATION
DATE


### 3.8 SSS TEST PER T927002

Disconnect SSS main cable connector lA9P2


2/6/91
Perform the test procedures of T927002 including 50 hour bearing confidence test, scanner centering, scanner frequency, and limit switch adjustments if required. It is not necessary to repeat paragraph 4.16, $T$ detector bias current measurement.

Reconnect 1A9P2
Data Review
Inspection

*3.8.1 Perform encoder optics ambient funtignal test per T927002, paragraph 4.12.4 JS TECH (Done 2/21/91)
*3.8.2 Apply additional adhesive to encoder optics assembly per RN GL54D. DATE $2-27-91$ G.J. 5 . MANN
 INSP


NOTE: Note after 24 hours the $5 S S$ may be removed from the handling fixture and installed on the base plate/test block.
*3.8.3 Seven day cure at room temperature.
DATE COMPLETE $3-4-9$
*3.8.4 Reinstall cover and torque screws to 4 to $6 \mathrm{in} / \mathrm{lbs}$.
G.JS. MANUF

| INSP |
| :--- |
| $3-8^{2}-91$ |
| DATE |

*3.8.5 Repeat step 3.8.1
TECH JJ, DATE

*NOTE: FOr OLS 16 perform this action after completion of paragraph
3.13 of this BVS.
3.9 AMBIENT SYSTEM TESTS

QUICKTESTN.ST
QUICKTESTR.ST
6X2X1.ST
AHCIIPT.ST - R
APCIIPT.ST - $P$
AHSFBIIPT.ST - R
6X3X1.ST - P
6×3X2.ST - R
$6 \times 3 \times 5 . S T-R$
MHClIPT.ST - R
6X5X1.ST - P
6X7X1.5T
6X7X2.ST
6X9.ST
7X8.ST
Data Review

VERIFICATION DATE


## STEPS

3.10 THERMAL VACUUM ADJUST

Deleted
3.11 VIBRATION - SSS Only

Inspection per PD045
checkpoint 3a checkpoint 3b

Notify AFPRO Warved per $5 . k$ info.


Vibrate SSS, 3 Axis, acceptance level per 1927686 para. 3.5

WEC Inspection per PD 055
AFPRO
PD 045 Checkpoint \#4


### 3.12 POST-VIBRATION, AMBIENT

Perform the test procedures of T927686 paragraph 3.6 except delete paragraph 3.6.5 and in paragraph 3.6.3, only the following test files need to be run:

NOMINAL CONFIGURATION TESTS



REDUNDANT CONFIGURATION TESTS


A Execute $6 \times 2 \times 3$.ST by entering "DSK $6 \times 2 \times 3.5 T$ ". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

In paragraphs 3.7.5.1.1, 3.7.10.3, 3.7.11.3, and 3.7.12.2, delete the following test files -
$5 \times 2 \times 16 \quad B B$ Signature
$5 \times 4 \times 1,2,3,4 \quad$ Core Tests
5X8X1,2 DMDM
$5 \times 10 \times 1,2,3,4 \quad$ Output Data Switching
5X14×1,2,3,4 SSP Formatter Tests
5×16×1,2,3,4
A/D Tests

Add a one day nominal temperature $T$ channel stability test between the two soak cycles by performing the following at approximately 2 hour intervals:

Execute $6 \times 2 \times 3 . S T$ by entering "DSK $6 \times 2 \times 3$.ST". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

## Delete paragraph 3.7.12.6.7

Add a day at the beginning and a day at the end of the nominal temperature piateau for additional $T$ channel stability by performing the following at approximately 2 hour intervals:

Add a day at the beginning and a day at the end of the nominal temperature plateau for additional $T$ channel stability by performing the following at approximately 2 hour intervals:

Execute $6 \times 2 \times 3.5 T$ by entering "DSK $6 \times 2 \times 3.5 T$ ". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

SPS Coax Connector repair on JIO due to defective female contact per NR 20250959.

- Remove top cover. of SPS
- Remove 2 P.C. boards - A241 \& A242
- Remove 540 R913G01 cable
- Remove coax connector J10 and replace

Remove coax connect
with new connector

- Circuit check
- WEC Insp.
- OPRO Insp.


Verify a minimum of 500 hours of scanner operation with new bearings has been performed in vacuum. Any deficiency should be made up at this time.

VERIFICATION
DATE

Verify completion of Thermal Vacuum


- OK to reinstall 913 cable in SPS
- Reinstal1 boards A241 \& A242
- OK to cover Insp.
- WEC Insp.
- DPRO Insp.
- Install cover
- WEC Insp.
- Reinstall buffer connector

3.16 Due to male contact pin damage on cables 644R329G02 and G03, replace OLS-12 cables with OLS-14 cables.


### 3.17 Final Ambient

For 0LS-16 perform additional adhesive operation pr RN GL54D prior to final ambient. See paragraph 3.8.1 thru 3.8.5.

For OLS 12 only perform the following post coax connector repair tests:

NEWONI.ST
QKTESTN.ST
5X10X1SS.ST
NEWON2.ST
QKTESTR.ST
5X10X2SS.ST
Perform T927686 paragraph 3.8, Final Scan Plane Definition.

Perform T927685 paragraph 3.9, Inspection, Data Review, and Preparation for Shipment except Delete paragraph 3.9.2, Weight and Center of Gravity.

Pin Retention
Inspection
Data Review
AFPRO
Pack
Ship


```
OLS PROGRAM DIRECTIVE
```

-CONTINUATION
SHEET-
DIRECTIVE NO. 045

DATE
12/12/88

CHECKPOINT 3.a (Before Vibration) Blue Room


#### Abstract

A. T channe adjustment tool was removed from PSU per Program Directive \#046. PSU was inspected before and after cover was installed. B. SSS and PSU pots are staked per PS 82560SA.


C. SSS mirrors are staked.
D. Thrust bearing (9RA3893) and limit switch assembly 758R962 have been removed for vibration. Buffer connectors are to remain installed.
INSP

| I NOTE 1: |
| :--- |
| WHEN APPLICABLE, |
| RECORD ACTIONS |
| PERFORMED IN SYSTEM |
| LOG BOOK. |

NOTE 2:
"CAUTION"
BUMP, SHOCK,
ABRUPT MOVEMENT
OF TAPE RECORDERS
can cause severe
DAMAGE.
$\square$

CHECKPOINT 3.b
(Before Vibration, in PQL)

## RESP

A. Vibration area is clean and PQL procedures 735 and 737 fallowed.

B. Vibration test equipment is within calibration date.
C. System monitoring equipment is within calibration date.
D. Clean room hats, gowns, masks and gloves are available and in use.


QE


```
NOTE 1:
    WHEN APPLICABLE,
    RECORD ACTIONS
    PERFORMED IN SYSTEM
    LOG 800K.
```

NOTE 2:
"CAUTION"
BUMP. SHOCK,
ABRUPT MOVEMENT
of tape recorders
can cause severe
DAMAGE.

```
OLS PROGRAM DIRECTIVE
```

-CONTINUATION
$\qquad$
12/12/88

CHECKPOINT \#4
(After Vibration)

|  |  | RESP |
| :---: | :---: | :---: |
| A. Westinghouse and Air Force post vibration inspection per BVS PD 055 has been performed on all assembles, including GSSA, GSSB and Blankets. (Verify that SSS and PSU pots are staked; or stake per PS 82560SA) | ( | INSP |
| B. The thrust bearing (9RA3893) and limit switch assembly 758R962 have been installed for thermal vacuum acceptance test. | $\frac{1}{1}$ | MFG INSP |
| C. Area is clean and contains no miscellaneous parts or extraneous hardware. | $15$ | QE |
| D. Anti-static mats and wrist straps are in place and ready for use. | \|r+1] | QE |
| E. Test equipment checked per Program Directive \#O22 less paragraphs IID and IIE. 033 | ? | QE |
| F. Perform Test Equipment Operational check per PD 022 , paragraphs IID and IIE. | TiRE | TD |
| G. The SPS, SPU, PSU, OSU, TCP, Recorders and BB's are interconnected with system cable and ground bus per 9R07845. Handling Procedures 9RA4220, 9RA4225 and 9RA4026 were followed. | PEP | TD |

## NOTE $1:$

WHEN APPLICABLE, RECORD ACTIONS PERFORMED IN SYSTEM LOG 800K.

```
NOTE 2:
    "CAUTION"
    BUMP, SHOCK,
    ABRUPT MOVEMENT
    OF TAPE RECORDERS
    CAN CAUSE SEVERE
    DAMAGE.
```



Distribution:
J. Spangle
M. Barrett
S. Nichols
B. Ditch
R. Bark
M. Epperly
G. Pollock
R. Baum
B. Spencer
A. Whyms


## APPENDIX B

RDS REWORK AND RETEST PROCEDURE

ILS 12
Contract F04701-90-C-0028

Prepared For
UNITED STATES AIR FORCE
Headquarters, Space Division
Los Angeles, California

Prepared By
WESTINGHOUSE ELECTRIC CORPORATION
Electronics Systems Group
Baltimore, Maryland

$$
B-1
$$

## REVISION SHEET



| WPF EP.lah | PAGE <br> 2 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

$B-2$

### 1.0 Introduction

This document describes the detailed rework and retest plan for SPS and OSU units returned from the fieid for incorporation of Real-time Data Smooth. A copy of this document will serve as a checklist for accomplishing the rework and retest procedures.

Incorporation of RDS into an OLS requires the modification of the
following assembijes:
${ }^{9 C}$ and SDF-5 boards in the SPS SPS Matrix Plate
OSU-1 and OSU-2 boards in the OSU
OSU Matrix Plate
OSU Top Cover Assembly
Record Serialization of Units to be reworked here:


Record Assembly Serial Numbers here:

|  | Prime 9C Board <br> Redundant 9C Board <br> Prime SDF-5 Board <br> Redundant SDF-5 Board | (6518342)(775RO76 or 775R077)(775RO76 or $775 R 077$ )(775R078 or 775079 )(775R078 or 775R079) | SNSNSNSNSN | $\frac{0001}{5013}$ |
| :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |
|  |  |  |  | 5 |
| 6408 640 4846 |  |  |  | $\frac{5012}{5012}$ |
|  |  |  |  | 5013 |
|  | OSU Mother Plate Assy | (522R783) |  |  |
|  | OSU Top Cover Assy | (644R046) | SN | $\frac{520.000}{50.9}$ |
| ${ }_{\text {r2\% }}$ | OSU-1 Board | (775R080) | SN | 5007 |
| 2h603 | -2 | (775R081) | SN | 5007 |


| WPF EP.lah | PAGE <br> 3 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS -2600 | REV |
| :--- | :---: | :---: | :---: | :---: |
|  |  |  |  |  |

### 2.0 Rework and Assembly Retest Plan

2.1 Pre-Rework

Incoming Inspection of Returned Units SPS (651R390) WEC

SPS (651R390) DPRO
OSU (640R960) WEC
OSU (640R960) DPRO

Mfg/Date $\frac{\text { Verification }}{\text { Insp }} /$ Date
$1 / 4 / 41$

2.2 Rework and Inspection

The RN numbers listed below are for reference only - All assemblies should be configured to their latest revision.

System Rework (536R500) GG42D
SPS Chassis Rework (651R390) OLS-12,
OLS-13

SPS Mother Plate Assy Rework (651R342)
OLS-12, 14, $15 \& 16$ GG35D OLS-13 GG700


SPS Matrix Plate Wiring Rework (wiretabs 322R959 or 322R960) OLS-12

GG17D
OLS-13 GG67D
OLS-14 to 16
GG16D
9 C board assy rework (775R076 or 775R077)
OLS-12 GG10D, GG15D \& GG20D
 OLS-13 OLS-14 to 16

GG69D
GG11D, GG15D \& GG21D
SDF-5 board assy rework (775R078 \& 775R079)

OLS-12
OLS-13 OLS-14 to 16

GG08D, GG14D \& GG18D
G668D
GG09D, GG14D \& GG19D


| WPF EP.lah | PAGE <br> 4 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> - |
| :--- | :---: | :---: | :---: | :---: |

$R-4$

OSU chassis rework (640R960) S/N 5007, $5009-5011$ GG33D
S/N 5008

Verification


OSU Top Cover Rework (644R046) S/N 5007, 5009-5011 S/N 5008 GG32D GG65D

OSU Mother Plate Rework (522R783) S/N 5007, 5009-5011 GG34D S/N 5008 GG65D


OSU Matrix Plate Wiring Rework (wiretab 322R958)

S/N 5007-5011
GG22D
OSU-1 board assy rework (775R080) S/N 5007, 5009-5011

GG12D, GGI50
GG54A GG5 GG63D

OSU-2 board assy rework (775R081) S/N 5007, 5009-5011 GG13D, GG15D \& GG24D S/N 5008


S/N 5008


| WPF EP. lah | PAGE <br> 5 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> 8 |
| :--- | :---: | :---: | :---: | :---: |
| $8-5$ |  |  |  |  |

### 2.3 Assembly Level Retest




| WPF EP.lah | PAGE <br> 6 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV |
| :--- | :---: | :---: | :---: | :---: |


Room Temperature Retest per paragraph 4.3 of T814A76

Are Coat Data Review
WEC Inspection - OK to Coat


DPRO Inspection - OK to Coat $\qquad$
Conformal Coat
Eight Non-powered Temperature Cycles
Hi/Low Temperature Test per paragraph 4.7 of T814A76

Data Review Complete $\qquad$
WEC Inspection - Assembly Complete
DPRO Inspection - Assembly Complete


* open ten on eff for coning oik $2 / 2 / \mathrm{a}$, OK $2 / \mathrm{T} / 91$.

| WPF EP.lah | PAGE <br> 7 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BUS -2600 | REV <br> - |
| :--- | :---: | :---: | :---: | :---: |



| WPF EP.1ah | PAGE <br> 8 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV |
| :--- | :---: | :---: | :---: | :---: |

$$
R-Q
$$

2.3.4 Redundant Side SDF-5 Retest (775R078/775R079)

Room Temperature Retest per paragraph 4.3 of T814A78
$1 / 26 / 91$
$1 / 31 / 91$


Pre Coat Data Review
WEC Inspection - OK to Coat
DPRO Inspection - OK to Coat
Conformal Coat
Eight Non-powered Temperature Cycles
Hi/Low Temperature Test per paragraph 4.7 of T814A78

1/3191 ME JAMES
$3 / 4 / 91$


Data Review Complete
WEC Inspection - Assembly Complete
DPRO Inspection - Assembly Complete


| WPF EP.lah | PAGE <br> 9 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> - |
| :--- | :---: | :---: | :---: | :---: |

B-9


| WPF EP.lah | PAGE <br> 10 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> - |
| :--- | :---: | :---: | :---: | :---: |

$B-10$


| WPF EP.lah | PAGE <br> 11 | FSCA NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> - |
| :--- | :---: | :---: | :---: | :---: |

B-1/
2.3.7

OSU Assembiy Retest (640R960)

|  | $S N$S007 <br> Date <br> $2 / 15 / 91$ |
| :---: | :---: |

-2-15-1991
shiclat 4.7 of T814A56

Data Review Complete
WEC Inspection - Assembly Complete
DPRO Inspection - Assembly Complete

* open item \#\% 2/1s/a1g\&K
ind smpar an test.

| WPF EP. Iah | PAGE <br> 12 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> - |
| :--- | :---: | :---: | :---: | :---: |

$B-12$
3.0 Subsystem Level Retest Procedure
3.1 Ambient Subsystem Verification Date Verification

Rework Complete - No unexplained Open Items on ICT

Checkpoint A of PD 045 (attach copy)


Run the following Test files (Room Temperature):

NEWONI'SS.ST

- QKTESTN.ST
$5 \times 20 \times 1 \leq 5.5 T$-RDSTETSS.ST
NEWON2SE.ST
QKTESTR.ST

5X18X1SS. ST
5X18X2SS. ST
5X18X3SS. ST
5X18×4SS. ST



### 3.1.1 Ambient Encrypter Verification

Note: To allow for scheduling and security constraints the ambient encrypter verification may be performed out of sequence, however, paragraph 3.1:1 must be complete prior to starting paragraph 3.3.9, Thermal Cycle \#8.

Instal] the KG-46 data encrypter and KG-28 decrypter. Check out the KG-28 set-up using the ST-19 verification procedure

Run the following test files:
NEWONISS.ST
$9 \times 20 \times 3 \leq 5.57$ ROSTSILSI


| WPF EP.lah | PAGE <br> 13 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

$B-13$

```
OLS PROGRAM DIRECTIVE
```

-CONTINUATION
SHEET-

DIRECTIVE NO $\qquad$ DATE $\qquad$

CHECKPOINT \#A
(Before Subsystem Test, in Bleck-5 Lean Room, per T927000)
Pol Thermal Clamber
A. Area is clean and contains no miscellaneous parts or extraneous hardware.
B. The antistatic mat and wrist straps are in place and ready for use.
C. Unit and System cable connector pins are checked and none are bent or pushed in. (NOTE: This can be verified at presystem unit inspection and the buffer connectors are inspected and installed at this point per P0034)
D. Verify correct color code on buffer connectors to certify inspected, tested and approved connectors per PD 034 (Appendix B). SPS, SP, PS U, os $4, \downarrow$ CA ACES
E. Test equipment configuration checked per Program Directive \#cz less paragraphs I If - and- IE P
F. Perform Test Equipment Operational Check per poozz paragraphs IID and IIE.
G. Review open ICT items on the SPS, SPU, PSU and OSU, and evaluate closure prior to moving to subsystem
 LNG TET. $13 / 2 / 2 / 5$

R


QI
 INSP


INSP


DE



OE
H. TCP, SPS, SPU, PSU and OSU are connected to. system cable connectors and each is grounded to the ground bus per 9RD7845. 065 SYFFEM ders2

1. Item annotated on ICT that units are ready for subsystem test.


TD $P Q L$ 15 OE

## NOTE 1:

WHEN APPLICABLE, RECORD ACTIONS PERFORMED IN SYSTEM LOG BOOK.

## NOTE 2: <br> "CAUTION"

BUMP, . SHOCK, ABRUPT MOVEMENT OF TAPE RECORDERS CAN CAUSE SEVERE DAMAGE.

Page 3 of 14

NEWON2SS.ST
$5 \times 20 \times 455.5 T$ ROSTST.ST
NEWONDI. ST
$5 \times 2$ ex $355.5 T \cdot$ ROSTSTSS.SF
NEWOND2.ST
$5 \times 20 \times 455.5 T$ ROSTSF.SF

4/25/91
H/25/91
$4 / 25 / 91$
4/25/91
$+/ 25 / 91$
+/25/91


### 3.2 SPS and OSU Vibration

Note: To allow for scheduling, OSU and SPS vibration may occur before ambient tests per paragraph 3.1. Vibration must be completed prior to starting paragraph 3.3. Later vibrations due to rework or RN incorporation shall be recorded on the units ICT.

Pre-Vib Data Review
WEC Inspection - OK to Vibrate
DPRO Inspection - OK to Vibrate
Vibrate SPS - x-axis, random only 503 acceptance level per PQL737, nonpowered


Vibrate OSU - x-axis, random only 503 acceptance level per PQL737,


WEC Inspection - Post Vib
DPRO Inspection - Post Vib
$\frac{4 / 26 / 51}{4 / 2 \cdot 9}$

### 3.3 Thermal Test

Checkpoint B of PD-045
(attach copy)
Install Thermocouples (PQL operation)




形


| WPF EP.lah | PAGE <br> 14 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

$B-15$

```
OLS PROGRAM DIRECTIVE
```

(Before Subsystem Thermal Test, in PQL, per T927000)
A. PQL thermal chamber is clean and area contains no miscellaneous parts or extraneous hardware.
B. Assembly and cable connectors (buffered and unbuffered) were checked before cabling up, and no damage was found. (Note: Where buffer connectors are in-place, check the viewable side; do not remove from assembly.) VERIFY COLOR CODE ON BUFFER CONNECTORS PER PD OJ, APPENDIX B.
C. SPS. SPU, PSU and OSU tranported per 9RA4220.

0. Assemblies SPS, SPU and TCP are connected to system cable connectors and each is grounded to the ground bus per 9R07845.
E. Test equipment is in calibration as required in Program Directive, \#O22 less paragraphs IID and IIE. 033


QI
F. Perform Test Equipment operational check per PD -022 paragraphs IID and IIE.

## NOTE 1:

WHEN APPLICABLE. RECORD ACTIONS PERFORMED IN SYSTEM LOG 800K.

```
NOTE 2:
    "CAUTION"
    BUMP, SHOCK,
    ABRLPT MOVEMENT
    OF TAPE RECORDERS
    CAN CAUSE SEvERE
    DAMAGE.
```


### 3.3.1 Ambient Verification

Run the following Test Files (Room Temperature):


| WPF EP. 7 ah | PAGE <br> 15 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

B-17

### 3.3.2 Thermal Cycle \#1

3.3.2.1 Hot Temperature
Allow chamber air in the vicinity
of the SPS to stabilize at $+50^{\circ} \mathrm{C}$ $+4^{\circ} /-0^{\circ}$ for 2 hours. During this time the OLS shall not have power on.

Run $5 \times 18 \times 15 S . S T$
NEWONI.ST
S×20入/5S.5T ROSSTSISS.SI
Enter CON 042
Enter OLS OFF

### 3.3.2.2 Cold Temperature

Allow chamber air in the vicinity of the SPS to stabilize at $-10^{\circ} \mathrm{C}$ $+0^{\circ} /-4^{0}$ for 2 hours. During this time the OLS shall not have power on.

$$
\text { Run } 5 \times 18 \times 1 S S .5 T
$$

NEWDNZ. ST
$5 \times 20 \times 255.57$ RESSTSTSS.ST
Enter CON 042
Enter OLS OFF

Date

5/4/91

5/4/91
Iteslar
5/4/01)

Date


Verification Data five


Verification


| WPF EP.lah | PAGE <br> 16 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

R-1R

### 3.3.3 Thermal Cycle \#2



| WPF EP. lah | PAGE <br> 17 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS -2600 | REV <br> $C$ |
| :---: | :---: | :---: | :---: | :---: |
| $B-19$ |  |  |  |  |

### 3.3.4 Thermal Cycle \#3

### 3.3.4.1 Hot Temperature Date Verification Date 但ure

Allow chamber air in the vicinity of the SPS to stabilize at $+50^{\circ} \mathrm{C}$ $+4^{\circ} \%-0^{\circ}$ for 2 hours. During this time the OLS shall not have power on.

Run $5 \times 18 \times 3$ SS. ST
NEWONI.ST
$5 \times 20 \times 155.5 T$ ROSSTSFSS.ST
Enter CON 042
Enter OLS OFF

### 3.3.4.2 Cold Temperature



Allow chamber air in the vicinity of the SPS to stabilize at $-10^{\circ} \mathrm{C}$ $+0^{\circ} /-4^{\circ}$ for 2 hours. During this time the OLS shall not have power on.

Run $5 \times 18 \times 3$ SS. ST
NEWONR.ST
$5 \times 20 \times 2=5.57$ RDSSTSTS5.ST
Enter CON 042
Enter OLS OFF


| WPF EP.lah | PAGE <br> 18 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

$B-20$

### 3.3.5 Thermal Cycle \#4



| WPF EP.lah | PAGE <br> 19 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

$$
B-2 \mid
$$

### 3.3.6 Thermal Cycle \#5



| WPF EP. 1ah | PAGE <br> 20 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> $C$ |
| :--- | :---: | :---: | :---: | :---: |

$B-2 Z$

### 3.3.7 Thermal Cycle \#6



| WPF EP.lah | PAGE <br> 21 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> $C$ |
| :--- | :---: | :---: | :---: | :---: |

### 3.3.8 Thermal Cycle \#7

### 3.3.8.1 Hot Temperature Date Verification bate nite

Allow chamber air in the vicinity of the SPS to stabilize at $+50^{\circ} \mathrm{C}$ $+4^{\circ} /-0^{\circ}$ for 2 hours. During this time the OLS shall not have power on.

Run $5 \times 18 \times 3 S S . S T$ NEUONI. ST
$5 \times 20 \times 15 S .5 T$ ROSSTSTSS.ST
Enter CON 042
Enter OLS OFF


### 3.3.8.2 Cold Temperature

Date
Verification
Allow chamber air in the vicinity of the SPS to stabilize at $-10^{\circ} \mathrm{C}$ $+0^{\circ} /-4^{\circ}$ for 2 hours. During this time the OLS shall not have power on.

Run $5 \times 18 \times 3$ SS. ST
$5 \times 20 \times 255.5 T$ ROSSTSTSS.ST
Enter CON 042
Enter OLS OFF


| WPF EP.lah | PAGE <br> 22 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BUS -2600 | REV <br> $C$ |
| :--- | :---: | :---: | :---: | :---: |

$R-24$

### 3.3.9 Thermal Cycle $\# 8$

### 3.3.9.1 Hot Temperature

Allow chamber air in the vicinity of the SPS to stabilize at $+50^{\circ} \mathrm{C}$ $+4^{\circ} \%-0^{\circ}$ for 2 hours. During this time the OLS shall not have power on.
sliolor


Run the following dual prime Test Files:


| WPF EP.lah | PAGE <br> 23 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BUS -2600 | REV <br> $C$ |
| :--- | :---: | :---: | :---: | :---: |



| WPF EP.1ah | PAGE <br> 24 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS-2600 | REV <br> C |
| :--- | :---: | :---: | :---: | :---: |

$$
R-כ G
$$

Install the KG-46 data encrypter and KG-28 decrypter. Check out the KG-28 set-up using the ST-19 verification procedure.


Run the following test files:
NEWONDI.ST
$5 \times 20 \times 355.5 T$ ROSSTSTSSE.ST
$5 \times 20 \times 455.5 T$ ROSNTSTESE. 57
Enter CON 042
Enter OLS OFF
Remove the KG-46 encrypter and K6-28 decrypter.

### 3.3.9.2 Cold Temperature

Allow chamber air in the vicinity of the SPS to stabilize at $-10^{\circ} \mathrm{C}$ $+0^{\circ} /-4^{0}$ for 2 hours. During this time the. OLS shall not have power on.

5/ul9,
Run the following dual prime Test Files:


| WPF EP. Jat | PAGE <br> 25 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BUS 2600 | REV <br> $C$ |
| :---: | :---: | :---: | :---: | :---: |


| －5x12x15S．ST |  | （1970） | Data new $9 \Leftrightarrow \sin s \pi_{1}$ |
| :---: | :---: | :---: | :---: |
| $\checkmark 5 \times 13 \times 1$ SS．ST | slular | （3m） | و\％simet |
| － $5 \times 13 \times 3$ SS．ST | sluelar | （1i6） | Sas stisit |
| － $5 \times 14 \times 155.5 \mathrm{ST}$ | Slerlar | （3IV） | 50． $513 / 9 / 9$ |
| $\checkmark 5 \times 16 \times 155.5 T$ | slular | （ $\frac{314}{\text { A }}$ ） | 妨 Shasi |
| －5X17XISS．ST | slular | （14id） | eas 5130， |
| －3， | NA Geo | 12 |  |
| $\rightarrow 5 \times 2 \times 155.5 T$ | $5(1119)$ | （1499） |  |
| －5x2x2SS．ST | $571191$ | 等乐） |  |

Run the following dual redundant Test Files：


| WPF EP．lah | PAGE <br> 26 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BVS－2600 | REV <br> $C$ |
| :--- | :---: | :---: | :---: | :---: |

$R-28$


Install the KG-46 data encrypter and KG-28 decrypter. Check out the KG-28 set-up using the ST-19 verification procedure.


Run the following test files:
NEWOND1.ST
$5 \times 20 \times 355.51$ AOSETSTFSE.ST
NEWOND2. ST

Enter CON 042

Turn OLS and TCP OFF, bring chamber from cold to $+50 \pm 10^{\circ} \mathrm{C}$ and allow

$5 / 13 / 91$


5/B2/21 $\qquad$
to soak for 2 hours minimum to prevent moisture from condensing on units
Return chamber to room temperature
and remove thermocouples and remove thermocouples.
Subsystem Data Review
Subsystem Test Complete
Final WEC Inspection
Final DPRO Inspection


| WPF EP.lah | PAGE <br> 27 | FSCM NO <br> 97942 | DOCUMENT NUMBER <br> BUS -2600 | REV <br> $C$ |
| :---: | :---: | :---: | :---: | :---: |
| $B-29$ |  |  |  |  |

BVS 2693

DATE 20 March 1992
ORIGINATOR $\frac{f+f e i l i n=t}{\text { J. Scilipoti }}$
REV $\qquad$
F/2
OLS \#12 BEARING RETROFIT
ACCEPTANCE TEST REPORT
VOLUME III OF III
ALIGNMENT AND SYNCHRONIZATION CURVES
(CDRL 006A1)

## Contract F04701-90-C-0028

Prepared For<br>UNITED STATES AIR FORCE Headquarters, Space Division Los Angeles, California

Prepared By<br>WESTINGHOUSE ELECTRIC CORPORATION<br>Defense and Electronics Center<br>Baltimore, Maryland

## $517$

## TABLE OF CONTENTS

PAGE

1. REFPLN Plots ..... 1
2. Align/Sync Plotted with Respect to REFPLN ..... 4
Alignment - HRD ..... 5

- T. ..... 6
- PMT ..... 7
Synchronization - HRD ..... 8
- T. ..... 9
- PMT ..... 10

3. Alignment/Synchronization For All Modes ..... 12
Alignment - HRD ..... 13

- T. ..... 18
- PMT ..... 23
Synchronization - HRD SDF \& SDS ..... 28
- HRD RTDF \& RTDS ..... 36
- T SDF \& SDS ..... 42
- T RTDF \& RTDS ..... 50
- PMT SDS \& RTDS ..... 52

4. Synchronization Using Backup Encoder \& Encoder Simulator ..... 62
AS Along Scan (Synchronization)
AT Along Track (Alignment)
SD Surface Distance
SDF Stored Data Fine
SDS Stored Data Smooth
RTD F Real Time Data - Fine
RTD S Real Time Data - Smoothed
H HRD Channel
T T(Therma1) Channel
P PMT Channe]
5. REFPLN PLOTS

These are the computer-generated least-squares fits to OLS \#12 HRD and T Channel Stored Data Fine (SDF) Alignment and Synchronization data taken from the final $+5{ }^{\circ} \mathrm{C}$ SSS/ $-8 \cdot{ }^{\circ} \mathrm{C}$ Ml Thermal Vacuum run (Orbit Nominal).

For OLS \#12, data from Thermal Vacuum Runs with M1 $=+12^{\circ} \mathrm{C}$ was also used to take into account any Alignment and Synchronization sensitivity to Ml temperature.

REFPLN is a computer program which generates the Alignment and Synchronization which represents the line-of-sight (LOS) or "look-angles" of the SSS with respect to the mounting (Interface) axes.

The curves are plotted as Error in milliradians from the OLS Interface Axes (essentially the spacecraft PMP axes), vs. ground surface distance along scan from subtrack (Nadir).
(An error of 0.1 milliradian at 450 naut. mi. altitude represents a ground position error of . 045 naut. mi. at nadir.)

## OLS\#/2 REFPLN ALIGNMENT



OLS\#12 REFPLN SYNCHRONIZATION

2. ALIGNMENT AND SYNCHRONIZATION FOR ALL MODES AT +5: SSS TEMP PLOTTED WITH RESPECT TO REFPLN

These curves are the difference between the Alignment and Synchronization curves at SSS $=+5 \cdot \mathrm{C}$ (Orbit Nominal) and the REFPLN Plots.

The curves represent the expected angular errors from the line-of-sight (REFPLN) axes for OLS data taken in the various modes of operation at orbit nominal conditions.

The curves were generated by averaging the data at $M 1=-8^{\circ}$ and $M 1=$ +12 . and then finding the difference between the average and the REFPLN.

OLS\#12 REFPLN ALIGNMENT - HRD SDF


-r -


OLS\#12 REFPLN SYNCHRONIZATION - HRD SDF


## OLS\#12 REFPLN SYNCHRONIZATION - T SDF




OLS\#12 REFPLN SYNCHRONIZATION - PMT RTDS


## 3. OLS \#12 ALIGNMENT \& SYNCHRONIZATION FOR ALL MODES

The following graphs are the measured OLS \#12 Alignment \& Synchronization with respect to the mounting (Interface) axes, for the following conditions.

Orbit Nominal (SSS $=+5^{\circ}, \mathrm{Ml}=-8^{\circ}$ )
Hot Limits $\quad\left(S S S=+7^{\circ}, M 1=+12^{\circ}\right)$
Cold Limits $\quad\left(S S S=+3^{\circ}, M 1=-8^{\circ}\right)$
Pre Vibration (Acceptance Level)
Post Vibration (Acceptance Level)











SYSTEM 12
IMC-NORM . PMT ALIGN . RTD-S ..... SSS=5. .,M1 = -8., DATE: 913
$\stackrel{\Upsilon}{-}]$ ORBIT


SYSTEM 12 IMC-NORM PMT ALIGN . RTD-S SSS=7..,M1= 12,,DATE: 903. N
-
HOT


IMC-NDRM . PMT ALIGN . RTD-S ..... SSS=3., M1 = -8, DATE: 907


SYSTEM 12 IMC-NORM PMT ALIGN RTD-S

SSS=23.,M1=24 ,DATE: 812


IMC-NORM . PMT ALIGN $\quad$ RTD-S $\quad . \quad . \quad S S S=22, M 1=22$, DATE: 818










 IMC-NORM . HRD . .SYNC . . RTD-F ..... SSS=7..,M1= 12, DATE: 903

 $-\square .4$
$\stackrel{\infty}{\infty}$
$\stackrel{N}{\square}$




SYSTEM 12

IMC-NORM

HRD SYNC
RTD-S
$S S S=7 \ldots, M 1=12$, DATE: $9 \boxed{ }$


## - 0.4

$\stackrel{\infty}{\infty}$
1
$-1.2$

$-4.0$


SD/100


SYSTEM 12 IMC-NORM T SYNC SDF $\quad$ SSS=5 . . M1 = -8., DATE: 914

\(\xrightarrow{\Upsilon}] \quad \begin{gathered}ORBIT<br><br>\end{gathered}\)



SYSTEM 12
IMC-NORM T . SYNC. SDF . . . SSS=3. ., M1 = =-8, DATE: 908







SYSTEM 12 IMC-NORM T . . SYNC .... RTD-F.....SSS=5 .,M1=-8, DATE: 708



SYSTEM . 12 ...AS . . PMT . . MODE=SS , . SSS =5 . ., M1 =-8 , DATE: 913.


SYSTEM 12 AS . PMT $\quad$ MODE $=S S, \quad S S S=7 \ldots, M 1=12$, DATE: 903


SYSTEM $12 \ldots$ AS $\ldots$ PMT $\ldots$ MODE $=$ SS . . SSS $=3 \ldots, M 1=-8$, , DATE: 907.



SYSTEM . 12 . AS .... PMT .... MODE $=S S$, $. S S S=22$, , M1 $=22$, DATE: 818


SYSTEM 12 AS . . PMT MODE $=$ DS , $\quad S S S=5 \quad, M 1=-8$, DATE: 913
$\left.\begin{array}{c}\dot{\sim} \\ \dot{\sim}\end{array}\right] \quad$ ORBIT


SYSTEM . 12 . AS .... PMT ... MODE=DS , $\mathrm{SSS}=7 \ldots, \mathrm{M1}=12$, DATE: 903


SYSTEM 12 AS . PMT MODE=DS , $S S S=3 \quad, M 1=-8$, ,DATE: 907


SYSTEM $12 \ldots$ AS AMT $\quad$ MODE $=D S, S S S=23, M 1=24$, DATE: 812


SYSTEM 12 AS PMT MODE=DS , $\quad S S S=22, M 1=22$, DATE: 818

4. SYNCHRONIZATION USING BACKUP ENCODER \& ENCODER SIMULATOR

The synchronization accuracy of the backup encoder track \& delphi generator are measured in the HRD SDF mode during acceptance test.

The curve labelled $A$ is taken with I/O X, using the Backup Encoder Control Track and Encoder Deiphi Generation. This curve can be compared to an HRD SDF sync curve using the Primary Encoder Control Track.

The curve labelled B is taken with $1 / 0 \mathrm{X}$, using the Primary Encoder Control Track, and encoder Simulator Delphi Generation. This curve is plotted as milliradians error from Interface Axis vs. Surface distance.

The curve labelled C is taken with I/O Y, using the Backup Encoder Control Track and Encoder Simulator Delphi Generation. This curve is plotted the same as the B curve.

The Bias and Separation constants used for bearing retrofit for OLS \#12 Primary Encoder are Bias $=-22$ and Separation $=-7$. The Backup Encoder constants used were Bias $=-23$ and Separation $=-6$. These constants are operationally adjustable to account for the effects of non-sinusoidal motion of the scanner.

SYSTEM . 12, ,AS/AT,H/T/P, MODE= SDF ,SSS=5 , M1 = -8, DATE: 915
凹.

SYSTEM . $12 ., A S / A T, H / T / P, M O D E=\ldots ., S S S=5 \ldots, M 1=-8$, DATE: 915


SYSTEM. 12, ,AS/AT,H/T/P, MODE $=\ldots \quad, \quad, \quad$ SSS $=5 \ldots, M 1=-8$, ,DATE: 915
$\stackrel{\infty}{\underset{\sim}{f}}\left[\begin{array}{l}\text { ORBIT } \\ \hdashline \text { NOMINAL }\end{array}\right.$

SYSTEM . 12, AS $/ A T, H / T / P, M O D E=S D F, S S S=7 \ldots, M 1=12$, DATE: 905



SYSTEM . 12.,AS/AT,H/T/P, MODE = SDF ,SSS $=7 \ldots, M 1=12$, DATE: 905


SYSTEM 12, AS/AT, H/T/P, MOCE $=$ SDF $, S S S=3 \ldots, M 1=-8$, DATE: 907


SYSTEM . 12, AS/AT,H/T./P,MODE= SDF , SSS $=3 \ldots, M 1=-8$, DATE: 907



SYSTEM . 12, ,AS/AT, $(1) T / P, M O D E=S D F, S S S=5 \ldots, M 1=-8$, DATE: 708


SYSTEM 12, ,AS/AT, ©HTT/P, MODE $=$ SDF , $, S S S=5 \ldots, M 1=-8$, DATE: 708


SYSTEM 12, AS/AT, (H)T/P, MODE $=$ SDF $, S S S=5 \ldots, M 1=-8$, DATE: 708


SYSTEM 12, ,AS/AT,H/T/P, MODE $=5 D F, S S S=5 \quad, M 1=-8$, ,DATE: 915


SYSTEM 12, ,AS/AT,H/T/P,MODE $=\ldots, \quad, \mathrm{SSS}=5, \mathrm{M} 1=-8$, DATE: 915


POST VIB

SYSTEM 12, AS $/ A T, H / T / P, M O D E=\quad, \quad$ SSS $=5 \ldots, M 1=-8$, , DATE: 915



[^0]:    1-11

