

BVS 2731

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REV -

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F13

OLS #14 BEARING RETROFIT

ACCEPTANCE TEST REPORT
VOLUME I OF III
SUMMARY AND SPECIFICATION REQUIREMENTS

(CDRL 006A1)

Contract F04701-90-C-0028

Prepared For

UNITED STATES AIR FORCE
Headquarters, Space Division
Los Angeles, California

Prepared By

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Defense and Electronics Center
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1.0 INTRODUCTION

The OLS #14 Acceptance Test Report contains the technical data pertinent to the OLS #14 AVE system. This document is intended to present the Acceptance Test data in terms of the requirements of the Prime Item Development Specification (DMSS-OLS-300) and Interface Specification (IS-YD-810A) for testing associated with bearing retrofit (BVS 2579). A copy of the signed-off BVS 2579 is included in this report as Appendix A. During this same period, BVS 2600 "RDS Rework and Retest Procedure" was performed. Testing for BVS 2600 was done at the functional level only. Therefore, no test results are included as part of this ATR. However, a copy of the signed-off BVS 2600 is included in this report as Appendix B.

Test results and data have been reviewed and verified by Westinghouse Electric Corporation and USAF representatives. System performance data, test histories, data summaries and system analyses are included in this report. In addition, a complete set of system log books are on file at the contractor's facility and are available for review. The Test History is in log books K32956--, and K40096-- and K42327--.

It is intended that this report provide a complete summary of all areas of OLS 14 requirements. Therefore, requirements not tested as part of bearing retrofit are also provided. When previously tested data is provided it will be so noted.

This Acceptance Test Report consists of 3 volumes as follows:

BVS 2731	OLS #14 Summary and Specification Requirements
BVS 2732	OLS #14 Acceptance Vibration Report
BVS 2733	OLS #14 Alignment & Synchronization Curves

1.1 Summary of System - Specific Parameters

OLS software Program = OLSP02J.FS

Gain Constants and Sensor Switch Points

P(0) = 9.875 dB
P(1) = 50.00 dB
P(2) = 0 dB
P(3) = 29.75 dB
S(1) = 59.875 dB
S(2) = 22 dB
S(3) = 33.75 dB

(These may change during Early Orbit Calibration.)

PMT HV EST (A532) = 3.635 volts \pm .250V

Cone Cooler S/N 026 with T detector S/N J-5

T Cold Patch EST (A549) curve - see Table next page.

T Cold Patch EST Voltage = 2.18V \pm .200V

TGAIN Left = 4
Right = 4
Both = 4

TLEVEL vs M1 temperature range - see second page following for table

VDGA constant for PMTCAL = (0440)₈

Encoder Simulator Bias Constant = Prim -18 Redun -17

Encoder Simulator Separation Constant = Prime +8 Redund +8

T COLD PATCH TEMP VS EST VOLTS
CONE COOLER S/N 026
T DETECTOR S/N J-5
OLS-14

<u>T (deg k)</u>	<u>EST (Volts)</u>
95	5.935
96	5.513
97	5.124
98	4.766
99	4.435
100	4.129
101	3.847
102	3.587
103	3.346
104	3.124
105	2.919
106	2.729
107	2.553
108	2.390
109	2.239
110	2.100
111	1.970
112	1.850
113	1.738
114	1.635
115	1.538
116	1.448
117	1.365
118	1.287
119	1.215
120	1.147
121	1.084
122	1.025
123	0.970
124	0.919
125	0.870

OLS #14
 TLEVEL VS M1 TEMPERATURE RANGE
 T DETECTOR S/N J-5

<u>TL</u>	<u>M1 TEMP(°C)</u>	
1111	-22.032°	to -16.513°
1110	-16.513°	-10.993°
1101	-10.993°	-5.474°
1100	-5.474°	0.045°
1011	0.045°	5.564°
1010	5.564°	11.034°
1001	11.034°	16.603°
1000	16.603°	22.122°
0111	22.122°	27.641°
0110	27.641°	33.161°
0101	33.161°	38.680°
0100	38.680°	44.199°
0011	44.199°	49.718°
0010	49.718°	55.238°
0001	55.238°	60.757°
0000	60.757°	66.276°

TLEVEL command changes should be uplinked to the OLS as a function of M1 temperature to maximize T Channel output accuracy.

1.2 Specification Pass-Fail Summary

The following sections of this Acceptance Test Report contain the test results as they pertain to the Development Specification requirements. Each Test Report paragraph heading is followed by the corresponding Segment Spec paragraph number in parentheses.

The table on the following page summarizes the OLS #14 pass-fail status vs. Development Spec. paragraph number.

DEVELOPMENT SPEC. PARAGRAPH NUMBER	PASS	FAIL
3.2.1.1.1.1 Infrared Spectrum	x	
3.2.1.1.1.2 Vis-Day Spectrum	x	
3.2.1.1.1.3 Vis-Night Spectrum	x	
3.2.1.1.2.1 Fine Geometric Resolution - HRD	x	
3.2.1.1.2.1 Fine Geometric Resolution - T	x	
3.2.1.1.2.2 Smooth Geometric Resolution - HRD	x	
3.2.1.1.2.2 Smooth Geometric Resolution - T	x	
3.2.1.1.2.2 Smooth Geometric Resolution - PMT	x	
3.2.1.1.2.3 Data Sampling	x	
3.2.1.1.3.1 Along Track Geometric Accuracy	x	
3.2.1.1.3.2/3 Along Scan Geometric Accuracy	x	
3.2.1.1.4.1.a T Channel Radiometric Accuracy Repeatability	x	
3.2.1.1.4.1b T Channel Radiometric Accuracy - Stability	x	
3.2.1.1.4.1c T Channel Radiometric Accuracy - Fixed	x	
3.2.1.1.4.2 Daytime Radiometric Accuracy	x	
3.2.1.1.4.3 Nighttime Radiometric Accuracy	x	
3.2.1.1.4.5.1 Terminator Location	x	
3.2.1.1.4.5.2 Gain Change Rate	x	
3.2.1.1.4.5.3 Maximum Gain Settings	x	
3.2.1.1.4.5.4 Commandable T-Channel Gain	x	

DEVELOPMENT SPEC. PARAGRAPH NUMBER	PASS	FAIL
3.2.1.1.4.5.5 Commandable T-Channel Level	X	
3.2.1.1.4.6.2/3 A/D Conversions & Algorithms	X	
3.2.1.1.5 Radiometric Resolution	X	
3.2.1.1.6.1 T Channel Noise	X	
3.2.1.1.6.2 L Channel Noise (Day)	X	
3.2.1.1.6.3 L Channel Noise (Night)	X	
3.2.1.1.6.4 Dark Current	X	
3.2.1.1.6.5 Stability	X	
3.2.1.1.6.6 Along-Track Noise Integration	X	
3.2.1.1.6.7 Glare Suppression	X	
3.2.1.1.7 Survivability	X	
3.2.1.1.8 Scan Angle	X	
3.2.1.1.9 Data Collection Rate	X	
3.2.1.2 Data Management	X	
3.2.1.3.1 28V Power	X	
3.2.1.3.2 5V Power	X	
3.2.2.1 Total Mass	X	
3.2.2.2 Component Mass	X	
3.2.2.3 Cable Harness Mass		X*
3.2.2.4 Dimensional Limits	X	
3.3 Design Features	X	
4.1/2 Environment	X	
5.1 Shipping & Storage	X	
INTERFACE SPEC PARAGRAPH NUMBER		
3.1.3 Alignment	X	

* This was not tested as part of Bearing Retrofit.

1.3 Summary of OLS #14 Testing

Aug-91 Returned to WEC for Bearing Retrofit & RDS
rework and testing

09-17-92 RDS (BVS 2600) - Began RDS Testing

09-21-92 SPS (x) Vibration

09-21-92 OSU (x) Vibration

09-30-92 Completed RDS Testing

11-04-92 Began System Testing per BVS 2579

12-11-92 SSS (x,y,z) Vibration

12-17-92 Post-Vibration Testing

01-27-93 OLS 14 to Thermal VAC Chamber

02-02-93 Thermal Vac Ambient Testing

02-05-93 Shim Replacement in SSS (Ref. AM 14A-12)

02-11-93 Thermal Vacuum Testing

03-04-93 Break VAC for Troubleshooting

03-11-93 Resume Thermal Vac Testing After STS Repairs

04-04-93 Thermal Vac Testing Complete

04-07-93 Adjusted T Gain Pots for Nominal Gain of 4

1.4 Configuration and Serialized Assemblies

The configuration listing on the following pages includes the configuration of OLS #14 as of 03-02-88. It also includes the PMT replacement of October, 1992.

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
Key Drawing	536R500G03	L	5009
<u>SSS Assembly</u>	<u>758R750G02</u>	L	5009
Cable Assy	644R320G03	M	501
<u>OSC Assy</u>	623R765G08	AD	5009
<u>HRD Assy</u>	623R754G05	AB	0008
PWR Bd	623R758G04	R	0008
Pre Amp Bd	623R506G04	U	0008
<u>T-Chan</u>	765R048G02	E	5001
T-Chan Bd	762R539G02	C	5002
Module	623R727G01	B	5015
Module	623R727G01	B	5016
<u>VDGA/Lin Log</u>	644R150G05	G	5009
Lin Log	644R127G05	P	5009
VDGA	644R152G04	P	5009
VDGA	644R153G04	N	5009
<u>Enc. OPT</u>	688R705H01	C	011
<u>PMT</u>	644R909G05	T	6001
PMT Tube	640R920G02	J	0019
EMR Bd	644R905G03	D	0009
LED Assy	536R916G01	D	0009
Switch Bd	644R903G05	M	0009
Doubler Bd	644R907G02	F	0009
Regulator Bd	644R807G04	H	0009
Pre Amp Bd	644R935G04	L	5010
<u>HRD Post Amp</u>	644R220G05	K	5009
Post Amp Bd	644R228G05	AB	5009

<u>DESCRIPTION</u>	<u>ASSEMBLY NO.</u>	<u>REV.</u>	<u>S/N</u>
<u>EST/LMD</u>	644R219G04	D	5007
EST/LMD Bd	758R142G03	E	0007
<u>Heater Cont</u>	633R053G13	J	5023
Elect Assy	633R052G04	Y	5023
<u>Heat Cont</u>	633R053G04	J	5033
Elect Assy	633R052G04	Y	5033
<u>Heat Cont</u>	633R053G15	J	5025
Elect Assy	633R052G04	Y	5025
<u>Heater Cont</u>	633R053G16	J	5026
Elect Assy	633R052G04	Y	5026
Solenoid Mech	758R620G02	D	5001
Optical Relay	701R717H01	A	014
<u>Rel Mech I</u>	640R701G02	F	5009
<u>Rel Mech II</u>	640R753G02	H	5009
<u>Rel Mech III</u>	640R381G02	H	5009
<u>T-Clamp</u>	623R821G01	H	-
<u>T-Cal</u>	623R920G01	B	-
<u>Aux Enccd</u>	640R846G05	L	5009
Bd Assy	640R825G05	F	5010
Bd Assy	640R844G05	J	5010
<u>Motor Assy</u>	623R894G01	B	002
<u>IMC/M3</u>	623R858G02	D	5011
<u>Cover, Cooler</u>	640R320G01	(-)	5009
Cone Cooler	9RA5216H01	K	026

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
<u>ENPA</u>	682R215G06	N	5009
A1 Bd	682R167G04	H	5010
A2 Bd	682R110G06	U	5009
A3 Bd	682R112G04	R	5009
<u>Aux Encd B/U</u>	682R300G04	C	5009
A1 Bd	682R149G04	E	5008
A2 Bd	682R151G04	E	5008
<u>BB1</u>	KG43		026
<u>BB2</u>	KG43		027
<u>BB3</u>	KG43		028
<u>Ther. Blk. Kit</u>	661R564G03	J	5009
<u>GSSA/DOC</u>	640R790G03	M	5009
<u>GSSB</u>	633R906G01	A	5009
PR1	688R461H01	E	044
PR2	688R461H01	E	045
PR3	688R461H01	E	056
PR4	688R461H01	E	057
Cable Assy	9RA5255H09	V	501
Cable Assy	9RA5255H02	V	502
Cable Assy	9RA5255H04	V	501
Cable Assy	9RA5255H10	V	503
Cable Assy	9RA5255H06	V	503
Cable Assy	9RA5255H11	V	501
Cable Assy	9RA5255H12	V	501
Cable Assy	9RA8118G01	G	-
Coax Assy	644R327G01	C	-
Coax Assy	644R327G02	C	-

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
Coax Assy	644R327G03	C	-
Coax Assy	644R328G01	D	-
Coax Assy	644R328G02	D	-
Coax Assy	644R328G03	D	-
Coax Assy	644R328G04	D	-
Coax Assy	644R328G05	D	-
Coax Cable	644R328G06	D	-
Coax Assy	644R329G01	D	-
Coax Assy	644R329G02	D	-
Coax Assy	644R329G03	D	-
Coax Assy	644R329G04	D	-
Coax Assy	644R329G05	D	-
Coax Assy	644R329G06	D	-
Coax Assy	644R329G07	D	-
Coax Assy	644R329G08	D	-
<u>SPS</u>	<u>651R390G03</u>	AF	5009
Matrix	651R342G04	AW	0004
<u>R/B</u>	644R665G05	AF	5016
Matrix	644R081G03	M	0007
A1 Bd	640R618G03	G	5019
A2 Bd	640R518G03	R	5018
A3 Bd	640R520G03	R	5017
Bus Bar	640R714G01	N	5009
Bus Bar	640R714G02	N	5009

<u>DESCRIPTION</u>	<u>ASSEMBLY NO.</u>	<u>REV.</u>	<u>S/N</u>
<u>R/B</u>	644R665G05	AF	5017
Matrix	644R081G03	M	0002
A1 Bd	640R618G03	G	5020
A2 Bd	640R518G03	R	5017
A3 Bd	640R520G03	R	5019
CU 1	640R612G03	L	5017
CU 1	640R612G03	L	5018
CU 2	640R614G03	L	5016
CU 2	640R614G03	L	5017
AU 1	640R608G03	F	5017
AU 1	640R608G03	F	5018
AU 2	640R610G03	F	5016
AU 2	640R610G03	F	5017
MC1X	640R560G03	N	5018
MC1X	640R560G03	N	5019
MC2X	640R662G02	A	5002
MC2X	640R662G02	A	5003
ROM	640R530G03	V	5016
ROM	640R530G03	V	5017
Core	644R910H03	K	020
Core	644R910H03	K	021
SDS2	640R442G03	U	5016
SDS2	640R442G03	U	5017
SDS3	640R444G03	P	5016
SDS3	640R444G03	P	5017

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
SDS4	640R446G03	U	5016
SDS4	640R446G03	U	5017
SDS5	640R498G04	T	5016
SDS5	640R498G04	T	5017
CLSD	640R458G04	AG	5016
CLSD	640R458G04	AG	5017
SDS1X	640R660G03	F	5002
SDS1X	640R660G03	F	5003
FC-1	640R450G03	AC	5016
FC-1	640R450G03	AC	5017
FC-2	640R454G04	W	5016
FC-2	640R454G04	W	5017
FC-3	640R456G03	AA	5015
FC-3	640R456G03	AK	5017
SDF-1	640R474G04	AK	5016
SDF-1	640R474G04	AK	5017
SDF-2	640R476G04	AK	5016
SDF-2	640R476G04	AK	5017
SDF-3X	640R644G03	A	5002
SDF-3X	640R644G03	A	5003
SDF-4X	640R646G03	A	5002
SDF-4X	640R646G03	A	5003
SDF-5X	640R648G03	F	5016
SDF-5X	640R648G03	F	5017
SDS-6	640R650G03	A	5002

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
SDS-6	640R650G03	A	5003
SDS-7	640R546G03	R	5016
SDS-7	640R546G03	R	5017
4B	640R412G03	R	5016
4B	640R412G03	R	5017
7A	640R414G03	AC	5016
7A	640R414G03	AC	5017
7B	640R416G05	AY	5016
7B	640R416G05	AY	5017
1A	640R400G03	AL	5018
1A	640R400G03	AL	5020
1B	640R402G03	AF	5016
1B	640R402G03	AF	5017
FBC	640R448G04	P	5016
FBC	640R448G04	P	5017
RAM	640R626G03	D	5002
RAM	640R626G03	D	5003
2A	640R488G03	AA	5016
2A	640R488G03	AA	5017
2B	640R410G03	Y	5016
2B	640R410G03	Y	5017
3A	640R404G03	AA	5016
3A	640R404G03	AA	5017

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
10X	640R572G03	K	5016
10X	640R572G03	K	5017
CLCL	640R406G05	AH	5016
CLCL	640R406G05	AH	5017
WF-1X	640R664G02	-	5016
WF-1X	640R664G02	-	5017
WF-2	640R432G03	AA	5016
WF-2	640R432G03	AA	5017
WF-3	640R622G03	G	5016
WF-3	640R622G03	G	5018
WF-4	640R436G04	M	5016
WF-4	640R436G04	M	5017
WF-5	640R438G03	AA	5016
WF-5	640R438G03	AA	5017
9A	640R654G02	A	5002
9A	640R654G02	A	5003
9BX	640R656G02	A	5002
9BX	640R656G02	A	5003
9CX	640R658G02	-	5002
9CX	640R658G02	-	5003
WF-6	640R568G03	J	5017
WF-6	640R568G03	J	5018

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
<u>OSU</u>	<u>640R960G04</u>	AE	5009
Matrix	522R783G02	G	0001
A1	640R522G04	V	5010
A2	640R524G03	R	5009
Bottom	644R047G04	V	5009
Top	644R046G03	R	5009
<u>SPU</u>	<u>758R040G03</u>	P	5009
Matrix	640R927G03	W	0003
Bus Bar	640R912G01	L	5011
SSP-8	640R638G03	D	5002
SSP-8	640R638G03	D	5003
RTD-1	640R508G04	AK	5016
RTD-1	640R508G04	A	5017
RTD-2	640R510G04	AU	5016
RTD-2	640R510G04	AU	5018
RTD-3	640R512G03	L	5016
RTD-3	640R512G03	L	5017
RTD-4	640R526G03	P	5016
RTD-4	640R526G03	P	5017
RTD-5	640R514G03	T	5016
RTD-5	640R514G03	T	5017
SSP-1X	640R636G02	C	5002
SSP-1X	640R636G02	C	5003
SSP-2	640R462G04	Y	5016
SSP-2	640R462G04	Y	5017
SSP-3	640R464G04	V	5016

DESCRIPTION	ASSEMBLY NO.	REV.	S/N
SSP-3	640R464G04	V	5017
SSP-4	640R642G02	F	5014
SSP-4	640R642G02	F	5003
SSP-5	640R468G04	U	5016
SSP-5	640R468G04	U	5017
SSP-6	640R470G04	U	5016
SSP-6	640R470G04	U	5017
SSP-7	640R472G04	Y	5016
SSP-7	640R472G04	Y	5017
SSP-9	640R554G03	K	5017
SSP-9	640R554G03	K	5018
<u>PSU</u>	<u>758R050G04</u>	AG	5009
Matrix	758R569G01	C	0004
RFI Plate	690R891G01	B	5007
Reg Assy	682R089G03	N	5006
Misc Bd	756R609G02	D	5000
T-Chan CG	688R483G04	K	5009
T-Left	688R485G04	J	5009
T-Rgt	688R487G04	J	5009
T-Chan BU	688R489G04	H	5009
T-Ana Fil	688R491G04	K	5016
T-Ana Fil	688R491G04	K	5017
L-Ana Fil	688R493G04	H	5016
L-Ana Fil	688R493G04	H	5017

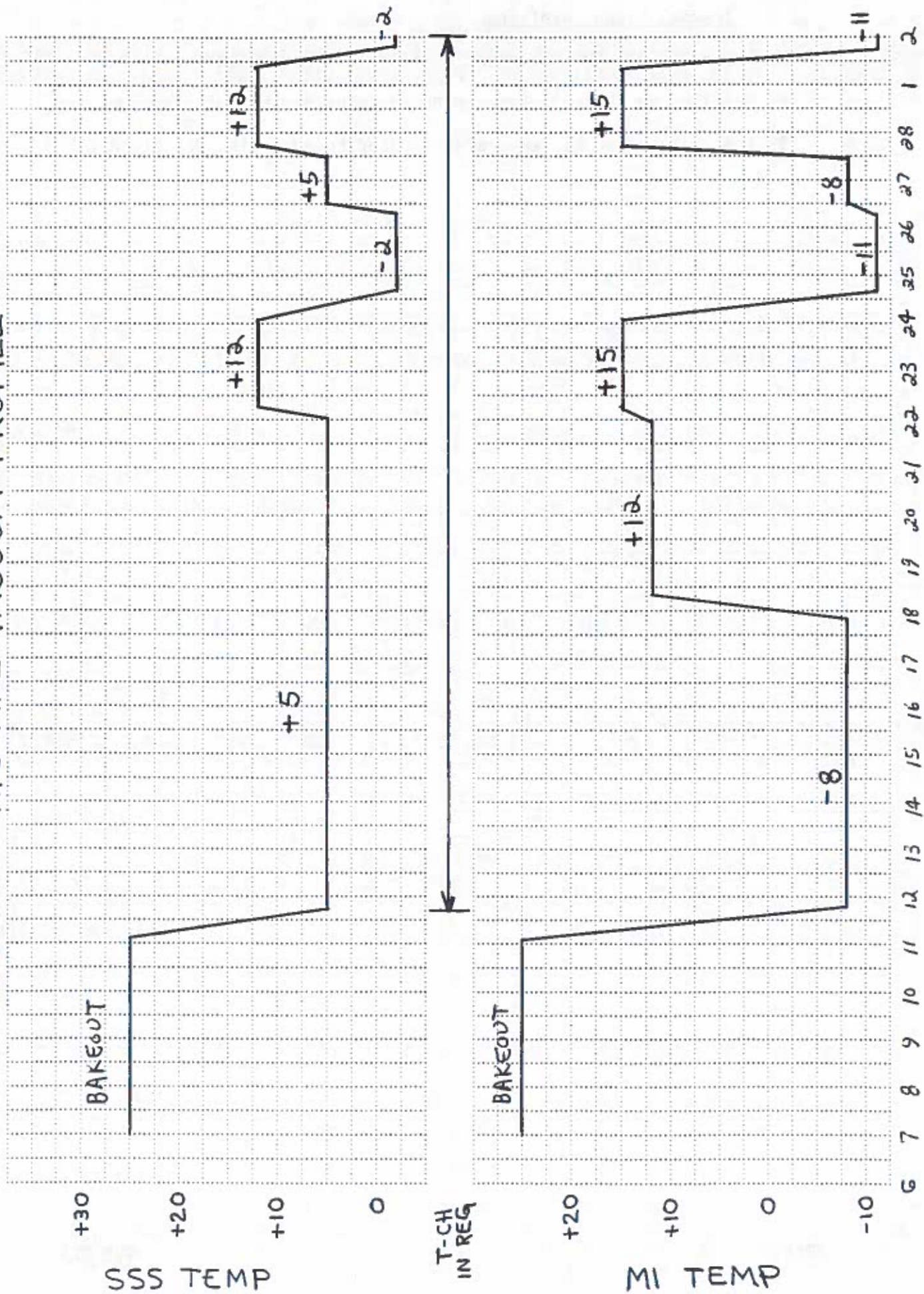
DESCRIPTION	ASSEMBLY NO.	REV.	S/N
PSU TRA BLK	640R998G05	K	5017
PSU TRA BLK	640R998G05	K	5018
DME	688R481G05	H	5019
DME	688R481G04	H	5018
IMC	644R864G04	F	5008
Relay-1	756R589G02	D	5009
+5V	644R078G04	R	5009
Relay-2	688R502G04	E	5009
+12VDA	688R499G04	F	5017
+12VDA	688R499G04	F	5018
Dual ENPA	640R616G02	J	5009
Relay-3	688R503G04	D	5009
-12V	644R069G04	R	5008
Relay-5	688R505G04	D	5009
Relay-4	688R504G04	D	5009
+12V Vm	688R500G04	D	5009
MC	688R495G04	G	5020
MC	688R495G04	G	5018
CPH	688R497G04	E	5009
Enable	682R381G04	E	5009
Driver	756R593G02	E	5000

1.5 Thermal Vacuum Profiles

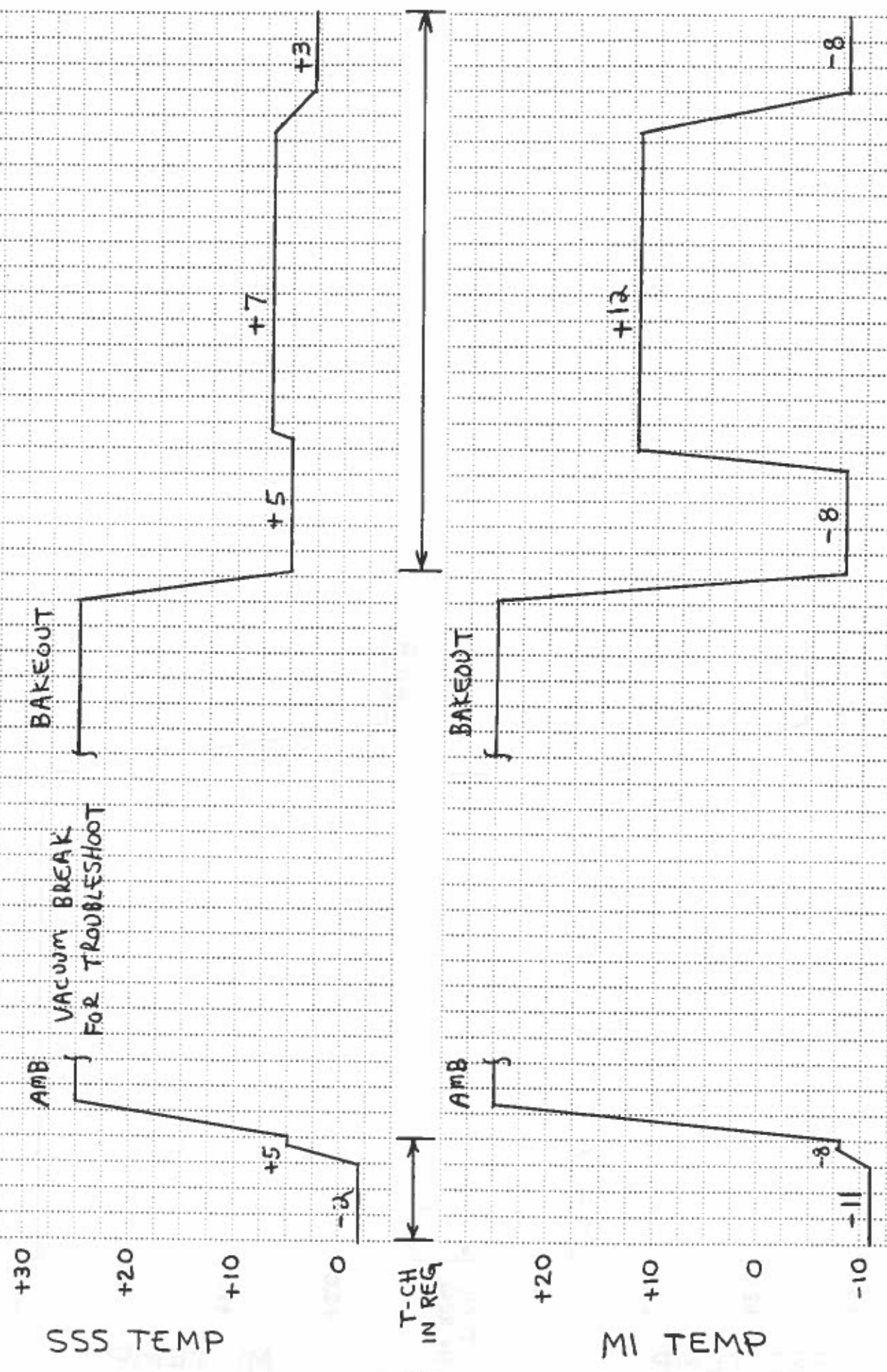
The OLS #14 AVE underwent a series of Thermal Vacuum Tests. The profiles on the next pages represent the history of pumpdowns, SSS temperature and MI temperatures experienced by the OLS #14 AVE.



OLS #14 THERMAL VACUUM PROFILE



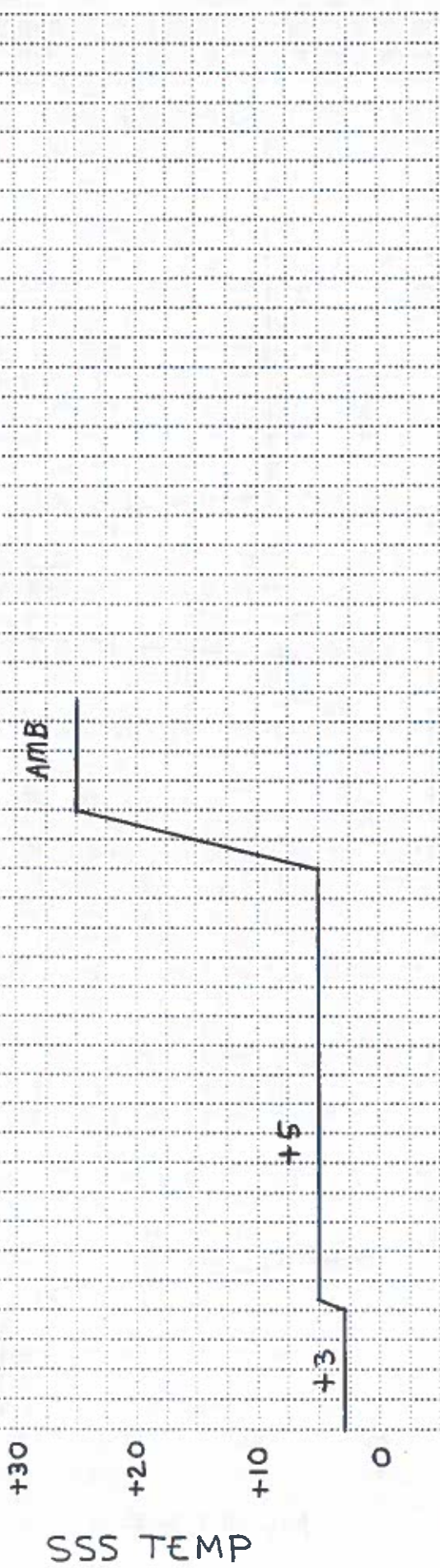
OLS #14 THERMAL VACUUM PROFILE



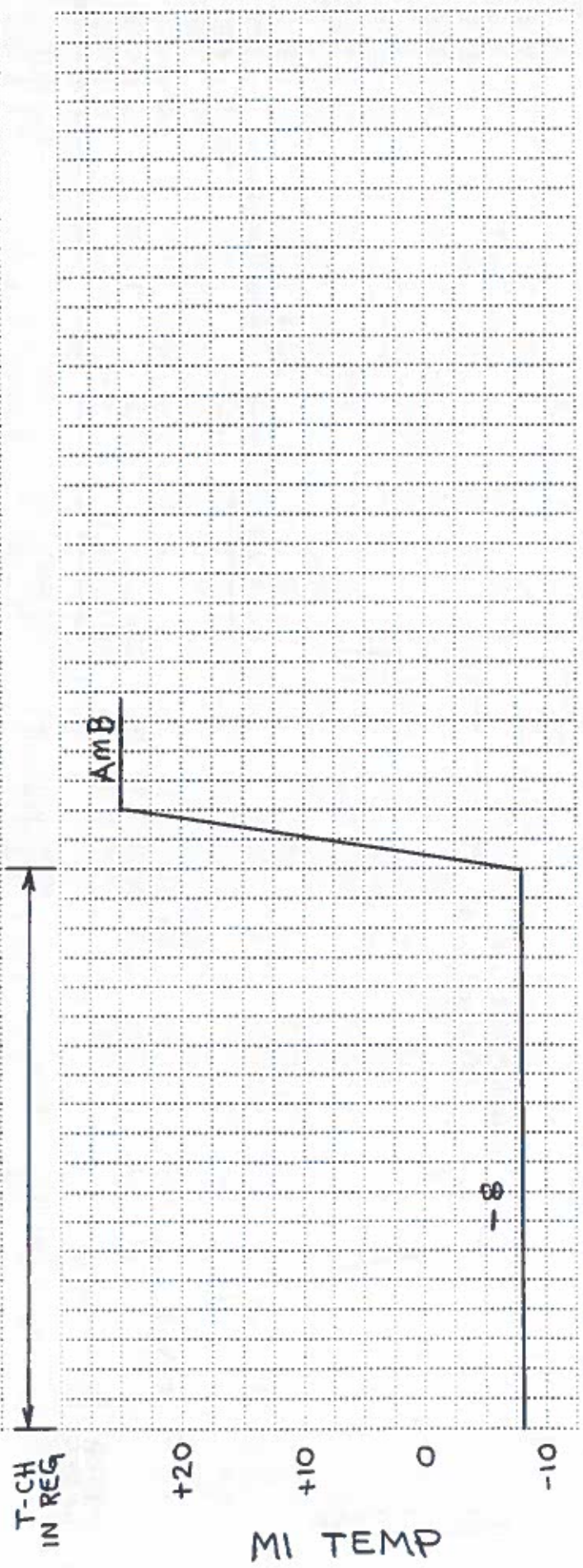
1-23

MARCH 1993

OLS #14 THERMAL VACUUM PROFILE



1-24



26 27 28 29 30 31 2 3 4 5 6 7 8 9 10

APRIL 1993

1.6 Test History Calendar

The test history calendar is a capsule look at the day-to-day progress of the OLS #14 AVE throughout its testing period at WEC.

121

DATE Nov. '92

TEST HISTORY

UNIT OLS #14

11	2	3	4	5	6	7
8	9	10	11	12	13	14
15	16	17	18	19	20	21
22	23	24	25	26	27	28
29	30					

4 Began System Tests BUS 2579

9 Cabled up System Funct Tests

12 Funct Tests

4x9x1

6x2x1

No Tests

15 No Tests

4x3x1

4x3x1

4x5x1

4x7x1
6x2x1

6x3x1
4x13

No Tests

22 No Tests

AHC11PT

AHSFB11PT
6x5x1
6x7x1
6x7x2
6x6x2

25 APC11PT
MPA11PT
Funct Tests
MHC11PT
6x9
6x6x1
6x6x2
7x8
4x7x1
6x3x5

No Tests

No Tests

No Tests

29 No Tests

No Tests

UNIT OLS #14 TEST HISTORY

DATE DEC. '92

1	2	3	4	5
6	7	8	9	10
11	12	13	14	15
16	17	18	19	20
21	22	23	24	25
26	27	28	29	30
31	32	33	34	35

UNIT OLS #14

TEST HISTORY

DATE JAN. '93

31	T/V optical test facility alignment					1	2
3		4	5	6	7	8	9
10	No Tests	11	12	13	14	15	16
17	No Tests	18	19	20	21	22	23
24	No Tests	25	26	27	28	29	30

T/V optical test facility alignment

3

4

5

6

7

8

9

MPA1PT
6x6x2

T/S EST#45
problem

T/S Core
Memory
Test

No
Tests

No
Tests

10

11

12

13

14

15

16

Core Memory
Tests

T/S Core
Memory
Error

T/S Core
Memory
Error

Funcn Tests

Core Memory
Tests

No
Tests

No
Tests

17

18

19

20

21

22

23

No
Tests

No
Tests

No
Tests

No
Tests

No
Tests

No
Tests

24

25

26

27

28

29

30

No
Tests

No
Tests

Uncabled
System

Moved
system
to TV

Waiting
for
TCP

No
Tests

T/V optical
test facility
alignment

UNIT OLS #14

TEST HISTORY

DATE FEB. '93

<p>7 7x9 7x10 Pumpdown @ 1015</p>	<p>8 72 HR Bake - Out</p>	<p>9 72 HR Bake - Out</p>	<p>10 72 HR Bake - Out</p>	<p>11 +5/-8 6x2x2 6x3x1 Funct Tests</p>	<p>12 Funct Tests 6x2x5 6x3x4 6x6x2 6x9 6x7x1 6x2x3A AHSF3PTZ 9x9x1</p>	<p>13 AHC11PT AHSFB9PT Funct Tests APC11PT 6x5x1 MHC11PT 6x3x5</p>	<p>14 MPA11PT T121T231G 6x2x4 6x2x3A ASV 210Q 6x7x3 7x12 T123T229A T125T227A MTC11PT</p>	<p>15 T127T225A ATC11PT ASV 270, ASV 290 T129T223A T131T221A ASV 210, 310 T121T2215</p>	<p>16 ASVCROSS T119T220A TDCRM3A Funct Tests TSTABILITY #1</p>	<p>17 TSTABILITY #2 thru #13 MI to +12 @ 2215 9x1x1</p>	<p>18 9x1x4 6x2x3A AHSF7PT</p>	<p>19 APC7PT MHC7PT 6x5x1 MPA7PT T121T231H</p>	<p>20 SIMFLT Chamber pres- sure Problems ASV 210, 310 T123T229B ATSE7PT T125T227B ASV 270, 290</p>	<p>21 6x3x3 6x5x2 MTC7PT T119T220B TDCRM3B TDCRM3K TDCRM3A To Hot Soak #1</p>	<p>22 9x1x6 Hot Soak #1 @ 0645 6x2x2</p>	<p>23 SIMFLT Funct Tests T121T231B</p>	<p>24 Funct Tests To Cold Soak #1 @ 0225 9x1x3 6x2x2</p>	<p>25 SIMFLT T121T231D Funct Tests</p>	<p>26 TSTABILITY #14 thru #17 To Nom Limits @ 0912 +5/-8 @ 1400 TSTABILITY #14 thru #17</p>	<p>27 TSTABILITY #8 thru #26 9x1x2 To Hot Soak #2</p>	<p>28 T121T231B Funct Tests</p>
---	-------------------------------	-------------------------------	--------------------------------	---	---	--	--	---	--	---	--	--	--	---	--	--	--	--	---	---	-------------------------------------

UNIT OLS #14

TEST HISTORY

DATE MAR. '93

<p>1 SIMFLT To Cold Soak @ 1030 9x1x3 Cold Soak #2 @ 2050</p>	<p>2 T/S Mission Sensor T121T231B</p>	<p>3 SIMFLT Funct Tests To +5/-8 @ 2200 7x4B</p>	<p>4 Shutting down for repairs @ 9130 Door open @ 1600 T/S SPU</p>	<p>5 T/S SPU</p>	<p>6 T/S SPU</p>
<p>7 T/S SPU</p>	<p>8 T/S SPU</p>	<p>9 T/S SPU</p>	<p>10 Pumpdown @ 1008</p>	<p>11 72 HR Bake - Out</p>	<p>12 72 HR Bake - Out</p>
<p>14 72 HR Bake - Out</p>	<p>15 7x7 9x1x6 @ +5/-8</p>	<p>16 Funct Tests 6x2x3A T121T2319</p>	<p>17 ML to +12 @ 0320 +5/-2 @ 1145 6x2x3A T121T231H to To Hot Limits @ 1628 +7/-2 @ 2130</p>	<p>18 Funct Tests 6x2x3A 6x2x5 6x2x2 6x2x4</p>	<p>19 T121T231B ASV 210, 310 AHSF89PT T123T229B Funct Tests T125T227B</p>
<p>21 APC7PT AT57PT MTC7PT 6x11x1 6x6x1 6x11x2 6x11x3 6x11x4</p>	<p>22 6x10 T125T227B TDCRM3B AT57PT SIMFLT</p>	<p>23 9x1x1 9x1x4 To Cold Limit @ 1618 7x4A 9x1x5 +3/-8 @ 2350</p>	<p>24 Funct Tests 6x2x3A AT57PT 6x2x5 MHC7PT 6x3x1 T121T231B AHC7PT 6x3x3</p>	<p>25 APC7PT ASV 210, 310 6x3x4 6x2x4 T123T229B T125T227B T119T231B 6x11x1 6x3x5 TDCRM3B 6x6x1 6x6x2 6x6x3 MPA7PT</p>	<p>26 6x11x2 6x11x3 6x11x4 MTC7PT Funct Tests 6x10 AHSF3PTI</p>
<p>28 To +5/-8 @ 0026 +5/-8 @ 0300 9x1x6 T-Stability #27 thru #32</p>	<p>29 T-Stability #33 thru #39 6x2x3A Funct Tests AHC7PT T121T231C ASV 210</p>	<p>30 Funct Tests ASV 310 6x2x4 AHSF89PT 6x11x1 T122T230 T123T226C T124T228C T125T227C T126T226C</p>	<p>31 AT57PT T128T224C 6x11x1 6x2x5 MHC7PT 6x2x2 T130T222C Funct Tests T119T220C TDCRM3C 6x3x1 ASV 310 6x3x3 6x3x5</p>	<p>27 AHSF89PT SIMFLT 9x1x1 9x1x4</p>	<p>20 T119T231B TDCRM3B Funct Tests ASV 210 6x3x5 6x3x1 6x3x4 6x3x3 MHC7PT AHSF3PTI AHC7PT 6x6x3</p>

2.0 DEVELOPMENT SPECIFICATION REQUIREMENTS

2.1 Spectra (3.2.1.1.1.1, 3.1.1.1.2, 3.1.1.1.3)

The OLS #14 Channel spectral responses are calculated by computer programs (GAINSET for L Channel and TGAIN for T channel) utilizing vendor spectral response data for the detectors, mirror and lens transmissivities/ reflectivities, and solar & lunar spectral radiance.

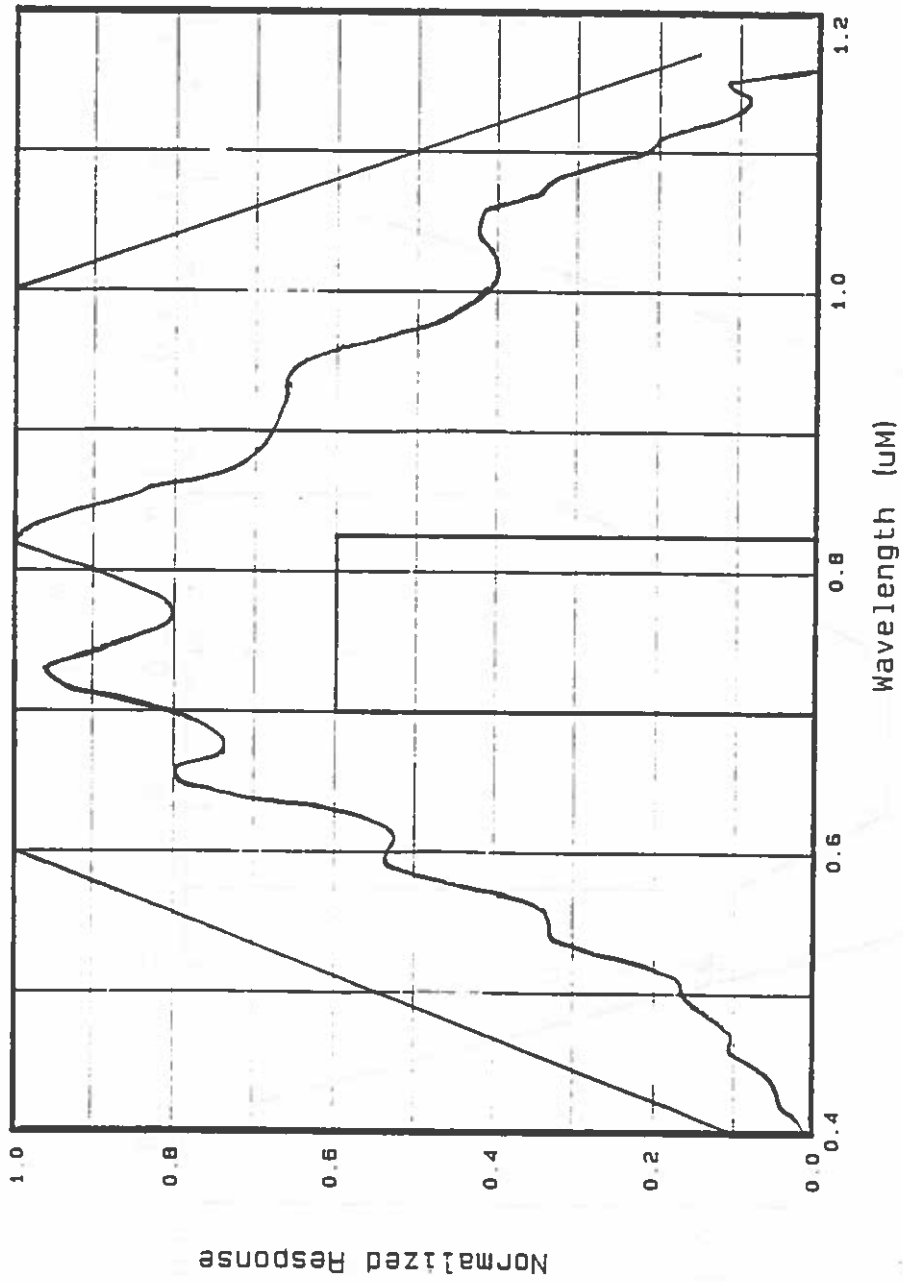
The HRD & T spectral responses are unchanged and those of the original ATR (BVS 2367) are included here. Due to the replacement of the PMT assembly, the PMT spectral response was recalculated using current data and replaces the spectral response of the original ATR.

All OLS #14 spectral responses are within specification.

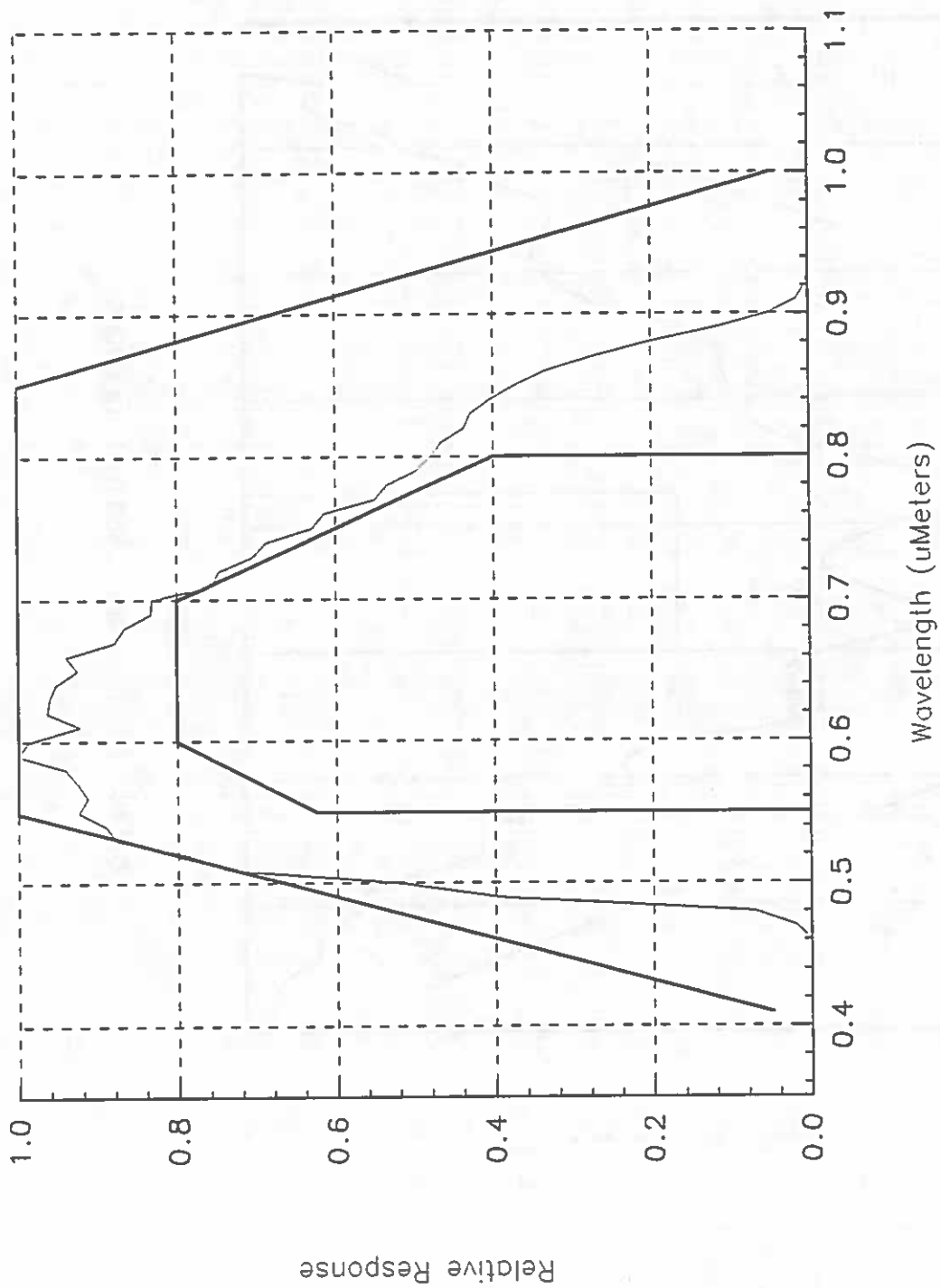
ATTACHMENTS: OLS #14 HRD Channel Spectral Response.

OLS #14 PMT Channel Spectral Response using assumed IRA/ORR.

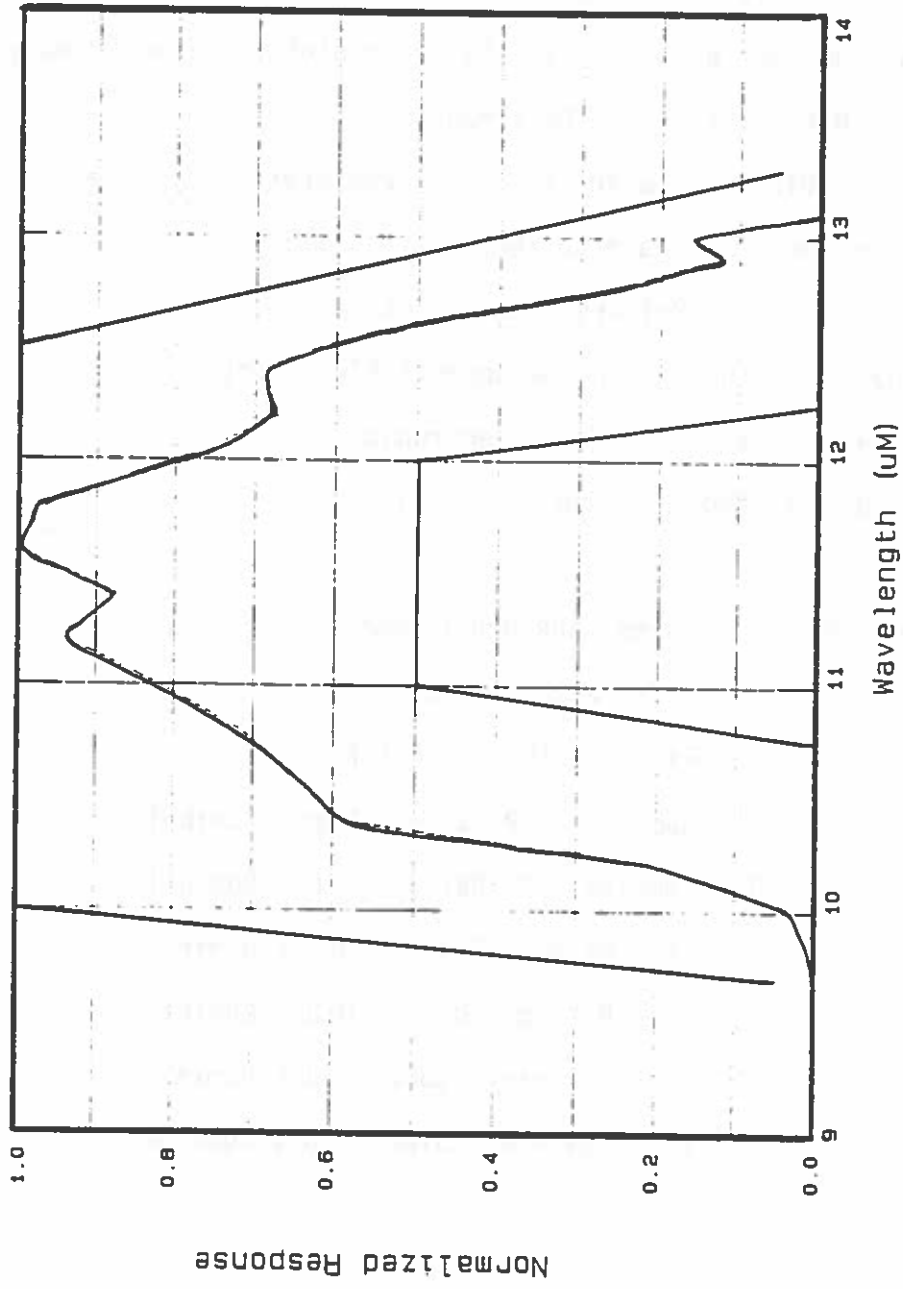
OLS #14 T Channel Spectral Response.



OLS14 HRD CHANNEL SPECTRAL RESPONSE



OLS14 VISIBLE NIGHTTIME SPECTRUM



OLS14 T CHANNEL SPECTRAL RESPONSE

2.2 GEOMETRIC RESOLUTION

2.2.1 Fine Geometric Resolution - Infrared (3.2.1.1.2.1)

2.2.1.1 Baseline (Orbit Nominal)

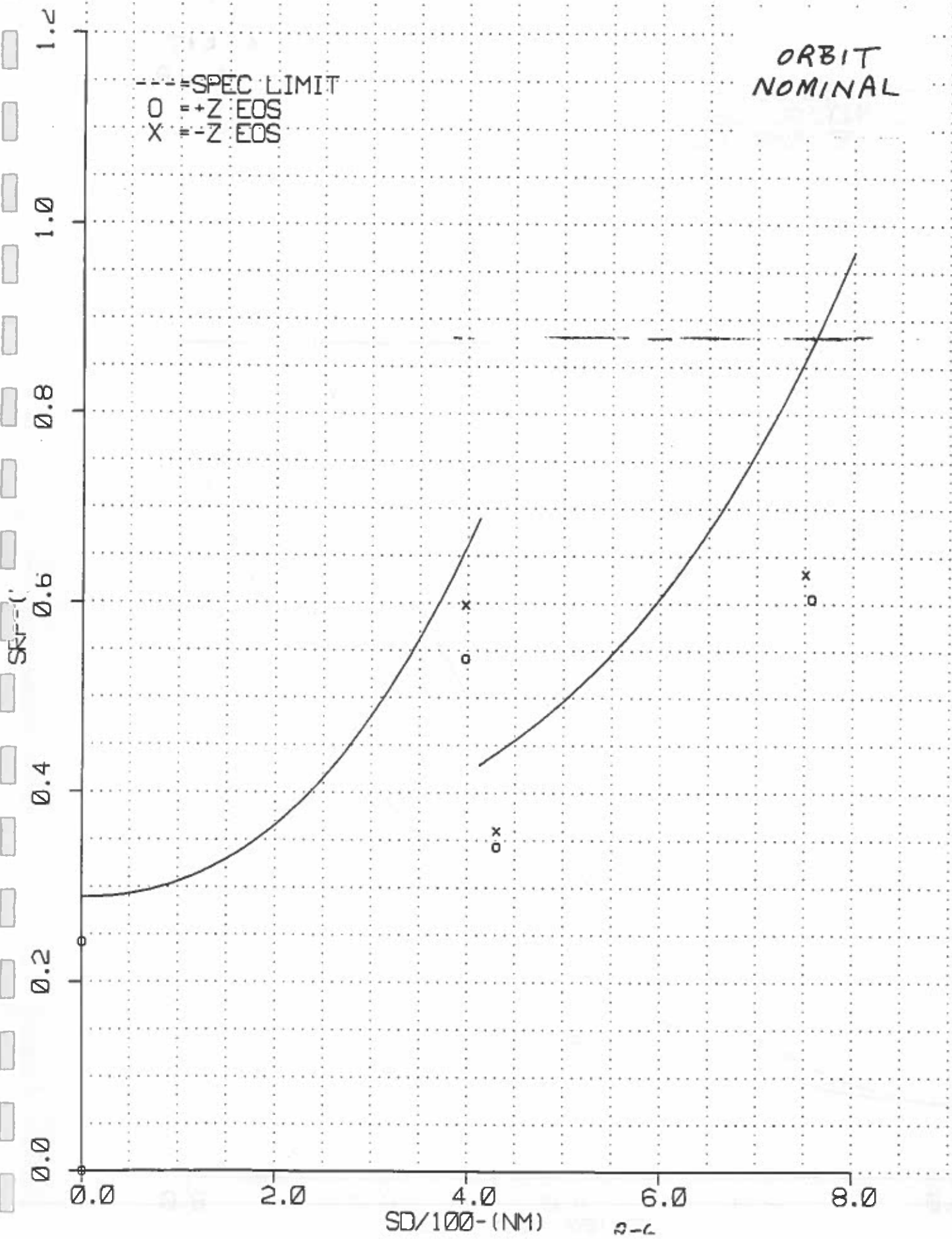
The TF Surface Resolution Parameter (SRP) is within the development specification limits.

The VAX Computer programs calculate and plot the Fine Primary SRP, and the T Right & Left Fallback modes. In addition, all Specification required modes are tabulated and presented. The designations on the graphs are defined as follows:

TFP T Fine Primary Electronics
TFB T Fine Backup (Redundant) Electronics
TSP T Smooth Primary Electronics
TSB T Smooth Backup Electronics

ATTACHMENTS: TF Normal SRP Orbit Nominal
TF Left Fallback Orbit Nominal
TF Right Fallback Orbit Nominal
T Complete SRP Tables Orbit Nominal
T Complete SRP Ratios Orbit Nominal
T Left Primary Tables Orbit Nominal
T Left Backup Tables Orbit Nominal
T Right Primary Tables Orbit Nominal
T Right Backup Tables Orbit Nominal

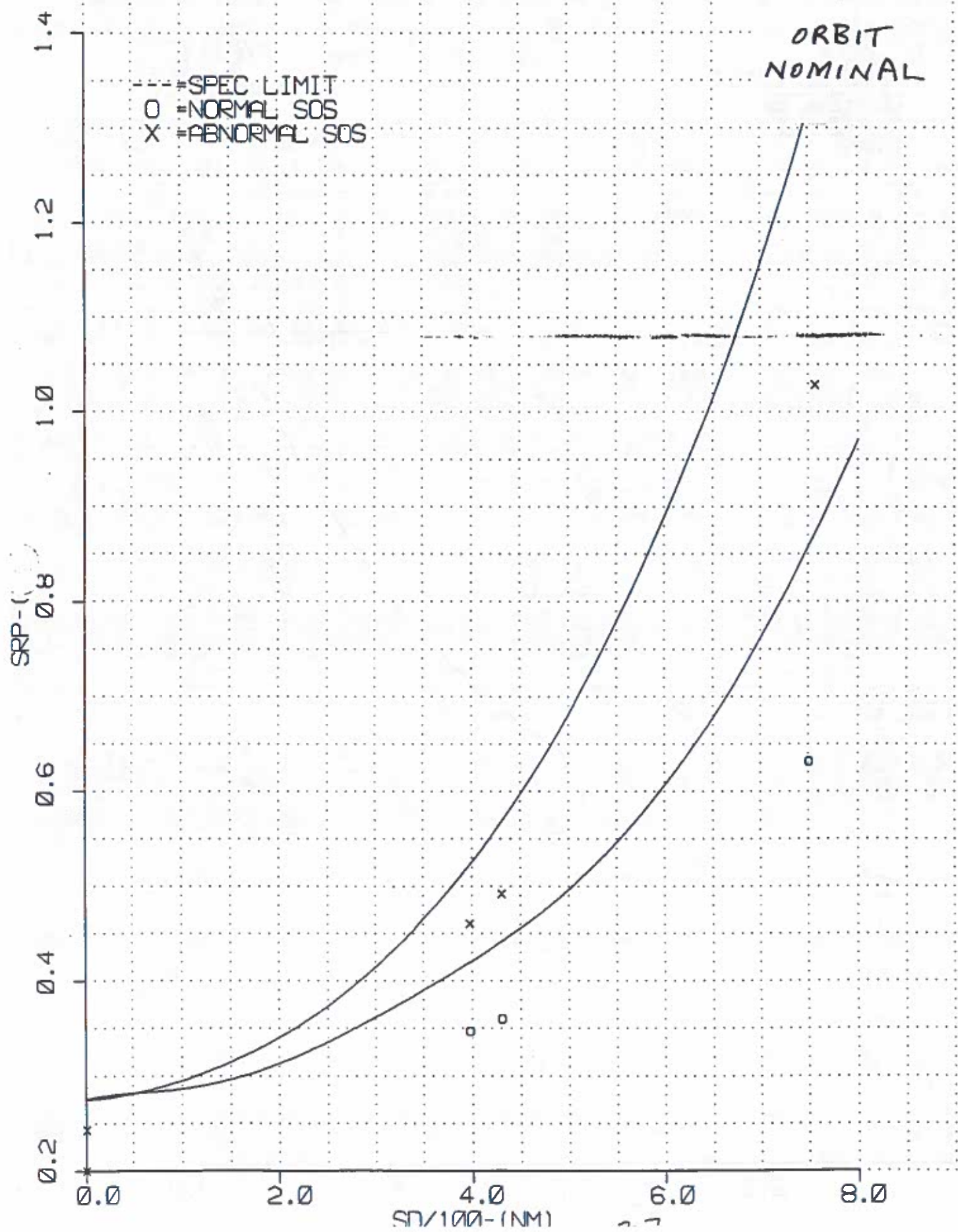
SYSTEM 14 ,SRP TF NORMAL,SSS=5 ,M1=-8 ,DATE:331



SYSTEM 14 , SRP TF L FBAK , SSS=5 , M1= -8 , DATE:331

ORBIT
NOMINAL

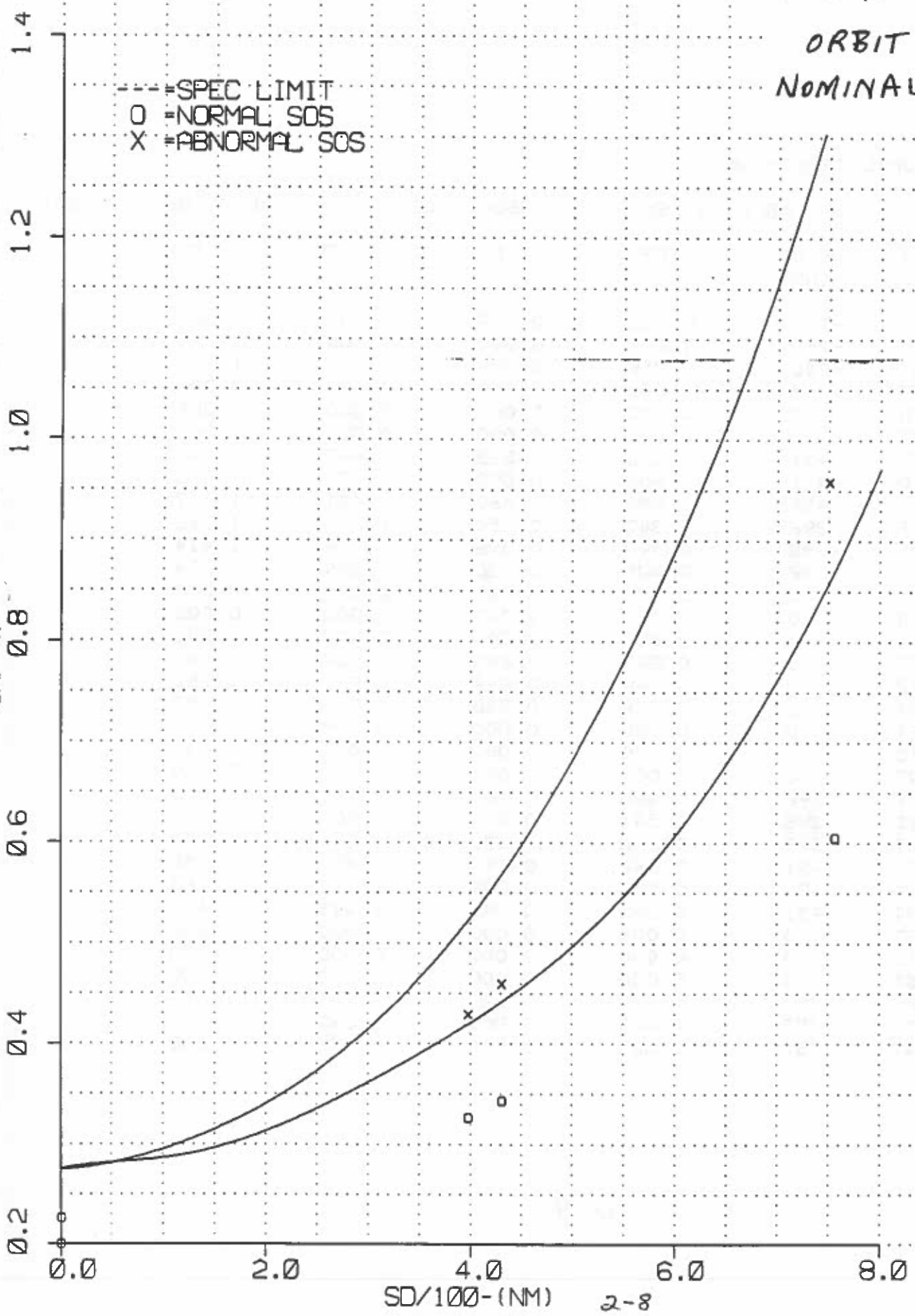
- - - = SPEC LIMIT
O = NORMAL SOS
X = ABNORMAL SOS



SYSTEM 14 , SRP TF R FBAK , SSS=5 , M1= -8 , DATE:331

ORBIT
NOMINAL

- - - = SPEC LIMIT
O = NORMAL SOS
X = ABNORMAL SOS



T, COMPLETE, SRP (NM)

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -SDEGC DATE: 331

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	-750.	0. 632	0. 632	1. 685	1. 680
MID	-750.	1. 209	0. 000	1. 802	1. 791
RGT	-750.	0. 958	0. 958	1. 731	1. 724
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	-431.	0. 360	0. 363	1. 447	1. 443
MID	-431.	0. 602	0. 000	1. 478	1. 473
RGT	-431.	0. 459	0. 460	1. 451	1. 447
LFT	-398.	0. 347	0. 350	1. 396	1. 392
MID	-398.	0. 598	0. 598	1. 419	1. 414
RGT	-398.	0. 429	0. 430	1. 398	1. 394
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	0.	0. 243	0. 245	0. 964	0. 961
MID	0.	0. 242	0. 244	0. 964	0. 962
RGT	0.	0. 226	0. 228	0. 959	0. 957
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	398.	0. 460	0. 461	1. 396	1. 392
MID	398.	0. 541	0. 541	1. 406	1. 402
RGT	398.	0. 326	0. 330	1. 390	1. 386
LFT	431.	0. 492	0. 492	1. 457	1. 452
MID	431.	0. 550	0. 000	1. 470	1. 465
RGT	431.	0. 343	0. 347	1. 445	1. 441
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	757.	1. 028	1. 027	1. 763	1. 754
MID	757.	1. 428	0. 000	1. 882	1. 866
RGT	757.	0. 605	0. 606	1. 673	1. 668

T. COMPLETE, SRP RATIO

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	-750.	0.736	0.736	0.749	0.747
MID	-750.	0.000	0.000	0.801	0.796
RGT	-750.	0.725	0.725	0.770	0.767
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	-431.	0.815	0.822	0.904	0.901
MID	-431.	0.000	0.000	0.923	0.920
RGT	-431.	0.809	0.810	0.906	0.904
LFT	-398.	0.826	0.834	0.910	0.907
MID	-398.	0.910	0.910	0.925	0.922
RGT	-398.	0.820	0.823	0.912	0.909
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	0.	0.883	0.891	0.918	0.915
MID	0.	0.836	0.843	0.919	0.916
RGT	0.	0.822	0.830	0.913	0.911
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	398.	0.881	0.882	0.910	0.908
MID	398.	0.824	0.824	0.917	0.914
RGT	398.	0.778	0.786	0.906	0.904
LFT	431.	0.865	0.866	0.910	0.907
MID	431.	0.000	0.000	0.918	0.915
RGT	431.	0.777	0.785	0.902	0.900
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	757.	0.766	0.765	0.780	0.776
MID	757.	0.000	0.000	0.833	0.826
RGT	757.	0.694	0.695	0.740	0.738

TF, LEFT, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (MM)	SRP ACTUAL (MM)	SRP RATIO
-750.	0.632	0.736
G.	0.000	0.000
-431.	0.360	0.815
-398.	0.347	0.826
G.	0.000	0.000
G.	0.243	0.883
G.	0.000	0.000
398.	0.460	0.881
431.	0.492	0.865
G.	0.000	0.000
757.	1.028	0.766

TF, LEFT, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (MM)	SRP ACTUAL (MM)	SRP RATIO
-750.	0.632	0.736
G.	0.000	0.000
-431.	0.363	0.822
-398.	0.350	0.834
G.	0.000	0.000
G.	0.245	0.891
G.	0.000	0.000
398.	0.461	0.882
431.	0.492	0.866
G.	0.000	0.000
757.	1.027	0.765

TF, RIGHT, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 5DEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-750.	0.958	0.725
0.	0.000	0.000
-431.	0.459	0.809
-398.	0.429	0.820
0.	0.000	0.000
0.	0.226	0.822
0.	0.000	0.000
398.	0.326	0.778
431.	0.343	0.777
0.	0.000	0.000
757.	0.605	0.694

TF RIGHT, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 5DEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-750.	0.958	0.725
0.	0.000	0.000
-431.	0.460	0.810
-398.	0.430	0.823
0.	0.000	0.000
0.	0.228	0.830
0.	0.000	0.000
398.	0.330	0.786
431.	0.347	0.785
0.	0.000	0.000
757.	0.606	0.695

2.2 Geometric Resolution (Cont'd)

2.2.1 Fine Geometric Resolution, Infrared (Cont'd.) (3.2.1.1.2.1)

2.2.1.2 Acceptance - Vibration

Consistant with the test philosophy established in BVS 2579, no vibration data was taken for the T-channel as part of Bearing Retrofit since stability of the T alignment for the common optics can be inferred from H alignment testing. Non-common optics were verified to be stable during the original OLS 14 testing.

ATTACHMENTS: N/A

2.2 Geometric Resolution (Cont'd)

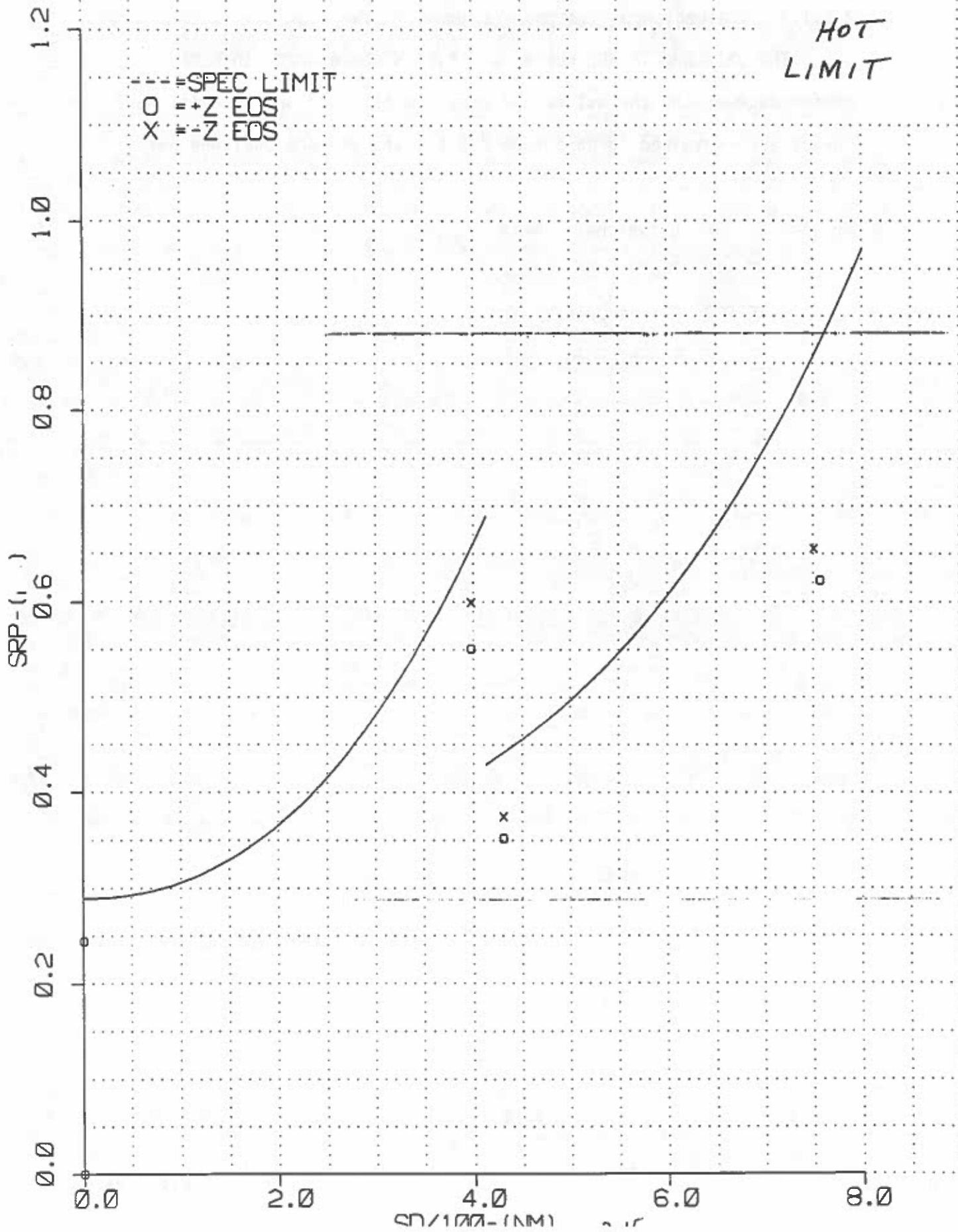
2.2.1 Fine Geometric Resolution, Infrared (Cont'd) (3.2.1.1.2.1)

2.2.1.3 Acceptance - Thermal Vacuum

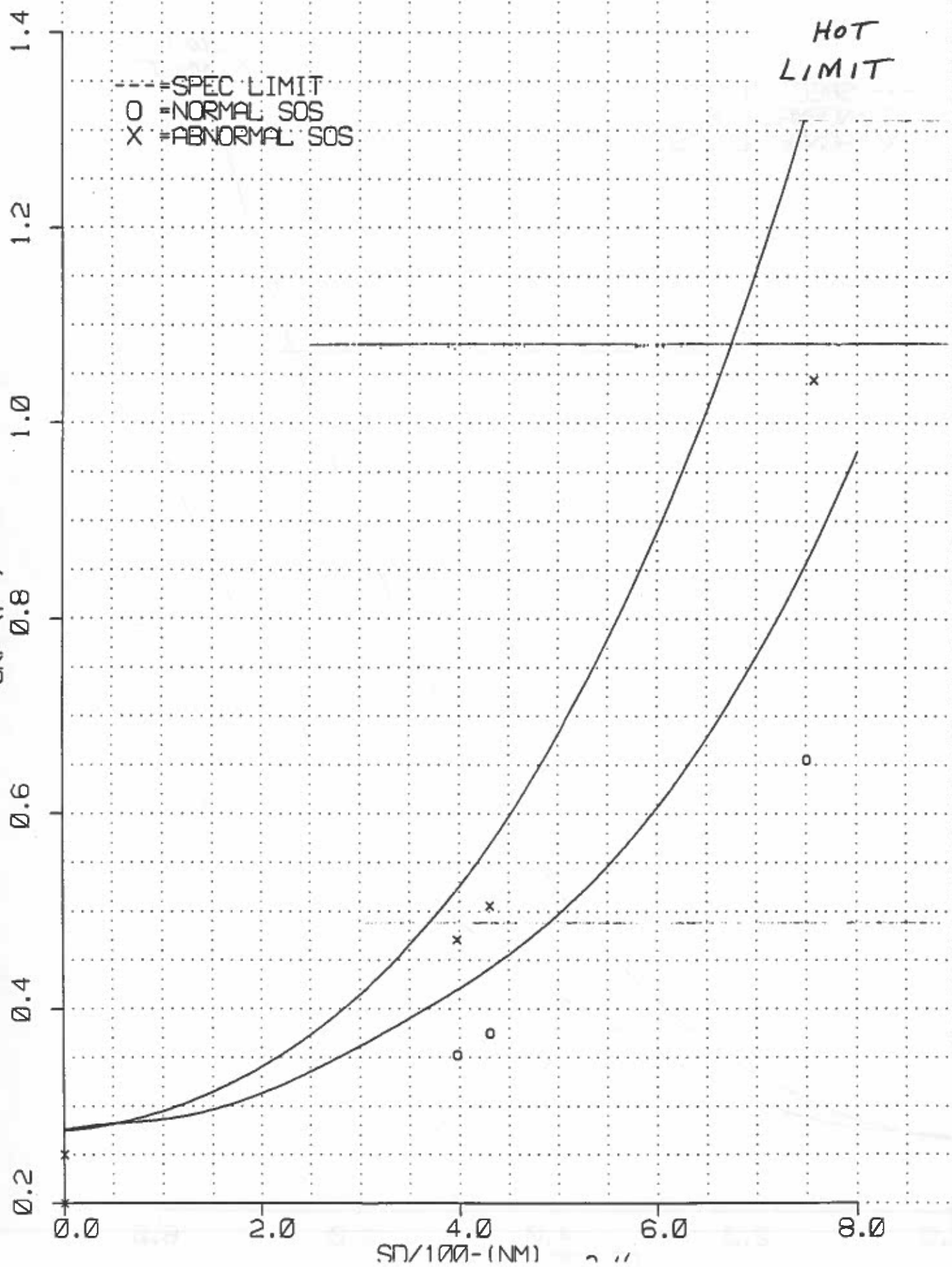
The attached TF SRP curves and tables demonstrate in-spec performance at the thermal vacuum test limits. The Orbit Nominal SRP curves are contained in paragraph 2.2.1.1 and are not included here.

ATTACHMENTS: TF SRP Curves Hot Limits
TF SRP Tables Hot Limits
TF SRP Curves Cold Limits
TF SRP Tables Cold Limits

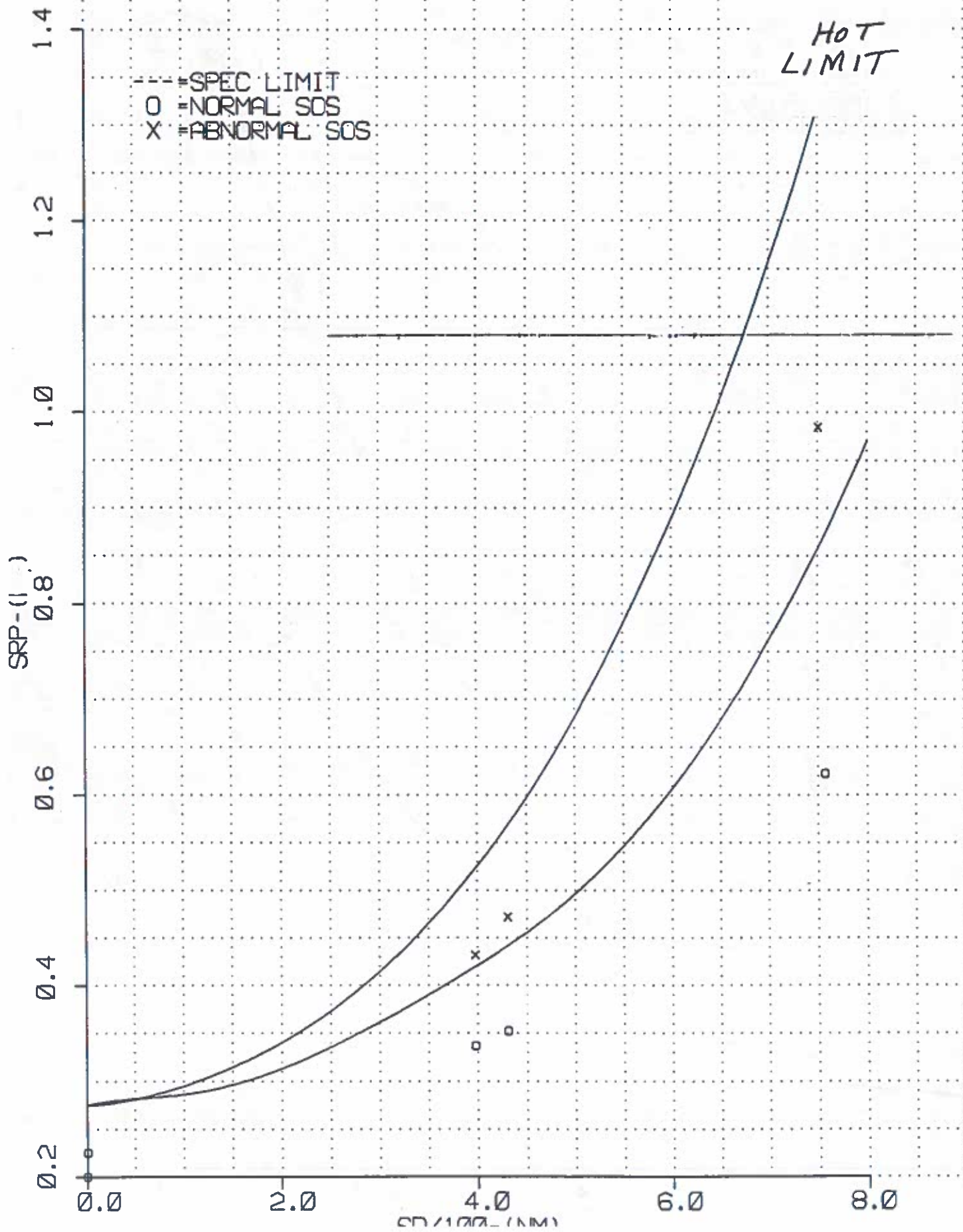
SYSTEM 14 , SRP TF NORMAL , SSS=7 , M1=12 , DATE:321



SYSTEM 14, SRP TF L FBAK, SSS=7, M1=12, DATE:321



SYSTEM 14, SRP TF R FBAK, SSS=7, M1=12, DATE:321



T, COMPLETE, SRP (NM)

FLT. NO. = 14 ENV. = 14 SSS= 7DEGC M1= 12DEGC DATE: 321

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	-750.	0.655	0.653	1.698	1.699
MID	-750.	1.201	0.000	1.798	1.795
RGT	-750.	0.985	0.983	1.746	1.745
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	-431.	0.375	0.376	1.450	1.451
MID	-431.	0.610	0.000	1.478	1.478
RGT	-431.	0.472	0.472	1.450	1.452
LFT	-398.	0.352	0.353	1.392	1.393
MID	-398.	0.600	0.596	1.420	1.421
RGT	-398.	0.432	0.432	1.394	1.395
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	0.	0.250	0.250	0.958	0.959
MID	0.	0.244	0.245	0.960	0.961
RGT	0.	0.225	0.227	0.953	0.955
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	398.	0.470	0.470	1.395	1.397
MID	398.	0.551	0.548	1.405	1.407
RGT	398.	0.337	0.339	1.387	1.389
LFT	431.	0.506	0.505	1.453	1.454
MID	431.	0.566	0.000	1.472	1.472
RGT	431.	0.352	0.354	1.447	1.449
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	757.	1.044	1.043	1.767	1.765
MID	757.	1.450	0.000	1.889	1.882
RGT	757.	0.622	0.621	1.683	1.685

T, COMPLETE, SRP RATIO

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	-750.	0. 763	0. 760	0. 755	0. 756
MID	-750.	0. 000	0. 000	0. 799	0. 798
RGT	-750.	0. 745	0. 744	0. 776	0. 776
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	-431.	0. 849	0. 851	0. 905	0. 906
MID	-431.	0. 000	0. 000	0. 923	0. 923
RGT	-431.	0. 831	0. 831	0. 906	0. 907
LFT	-398.	0. 839	0. 842	0. 907	0. 908
MID	-398.	0. 912	0. 908	0. 926	0. 926
RGT	-398.	0. 826	0. 827	0. 909	0. 910
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	0.	0. 908	0. 910	0. 913	0. 914
MID	0.	0. 844	0. 846	0. 914	0. 916
RGT	0.	0. 820	0. 825	0. 908	0. 909
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	398.	0. 900	0. 900	0. 910	0. 911
MID	398.	0. 839	0. 835	0. 916	0. 917
RGT	398.	0. 803	0. 807	0. 905	0. 906
LFT	431.	0. 890	0. 889	0. 907	0. 908
MID	431.	0. 000	0. 000	0. 919	0. 919
RGT	431.	0. 797	0. 801	0. 904	0. 905
LFT	0.	0. 000	0. 000	0. 000	0. 000
MID	0.	0. 000	0. 000	0. 000	0. 000
RGT	0.	0. 000	0. 000	0. 000	0. 000
LFT	757.	0. 777	0. 777	0. 782	0. 781
MID	757.	0. 000	0. 000	0. 836	0. 833
RGT	757.	0. 714	0. 712	0. 745	0. 745

TF, LEFT, PRIMARY

FLT. NO. = 14 ENV. = 14 SSS= 7DEGC M1= 12DEGC DATE: 321

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.655	0.763
0.	0.000	0.000
-431.	0.375	0.849
-398.	0.352	0.839
0.	0.000	0.000
0.	0.250	0.908
0.	0.000	0.000
398.	0.470	0.900
431.	0.506	0.890
0.	0.000	0.000
757.	1.044	0.777

TF, LEFT, BACKUP

FLT. NO. = 14 ENV. = 14 SSS= 7DEGC M1= 12DEGC DATE: 321

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.653	0.760
0.	0.000	0.000
-431.	0.376	0.851
-398.	0.353	0.842
0.	0.000	0.000
0.	0.250	0.910
0.	0.000	0.000
398.	0.470	0.900
431.	0.505	0.889
0.	0.000	0.000
757.	1.043	0.777

TF, RIGHT, PRIMARY

FLT. NO. = 14 ENV. = 14 SSS= 7DEGC M1= 12DEGC DATE: 321

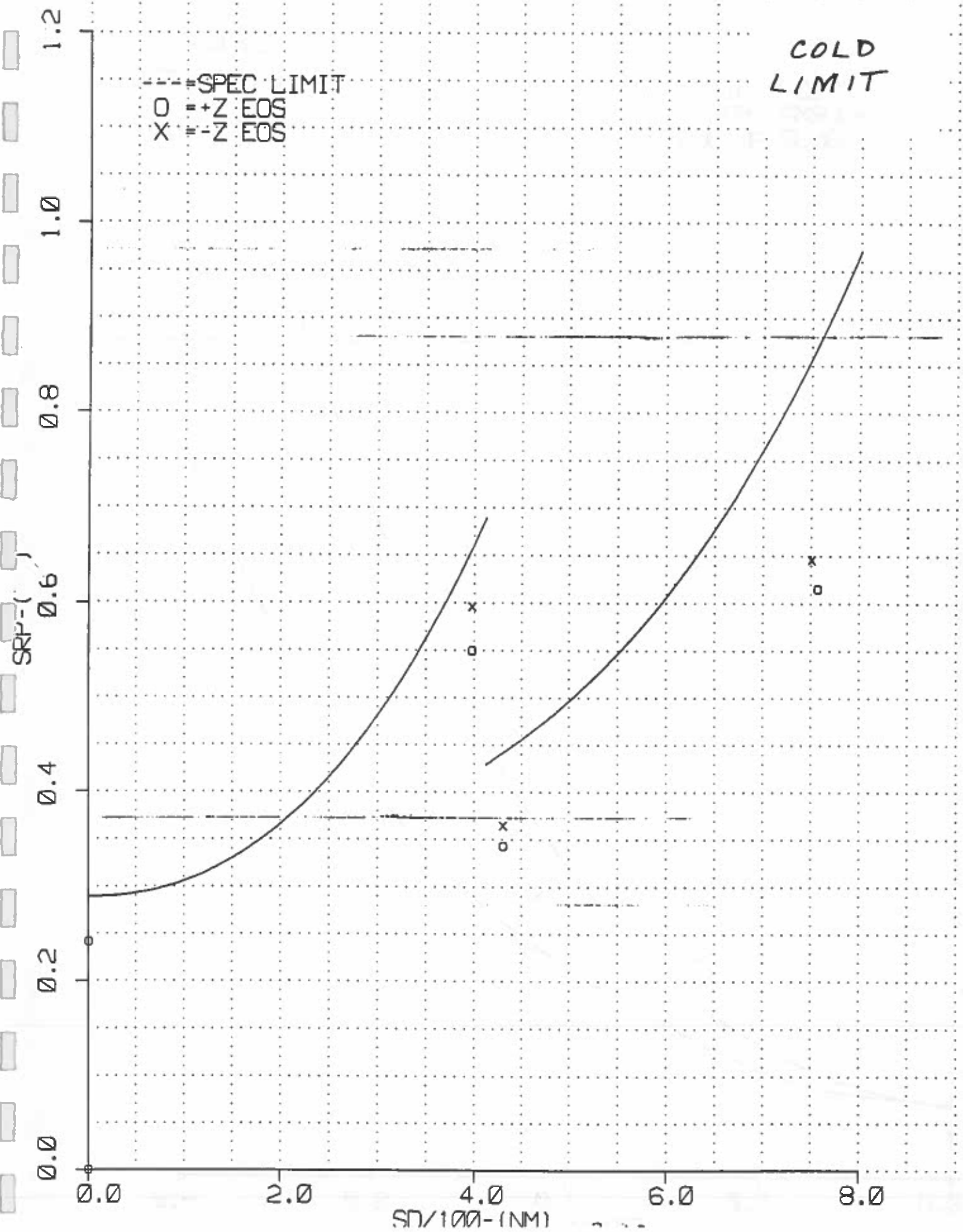
SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.985	0.745
0.	0.000	0.000
-431.	0.472	0.831
-398.	0.432	0.826
0.	0.000	0.000
0.	0.225	0.820
0.	0.000	0.000
398.	0.337	0.803
431.	0.352	0.797
0.	0.000	0.000
757.	0.622	0.714

TF RIGHT, BACKUP

FLT. NO. = 14 ENV. = 14 SSS= 7DEGC M1= 12DEGC DATE: 321

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.983	0.744
0.	0.000	0.000
-431.	0.472	0.831
-398.	0.432	0.827
0.	0.000	0.000
0.	0.227	0.825
0.	0.000	0.000
398.	0.339	0.807
431.	0.354	0.801
0.	0.000	0.000
757.	0.621	0.712

SYSTEM 14, SRP TF NORMAL, SSS=3, M1=-8, DATE:326



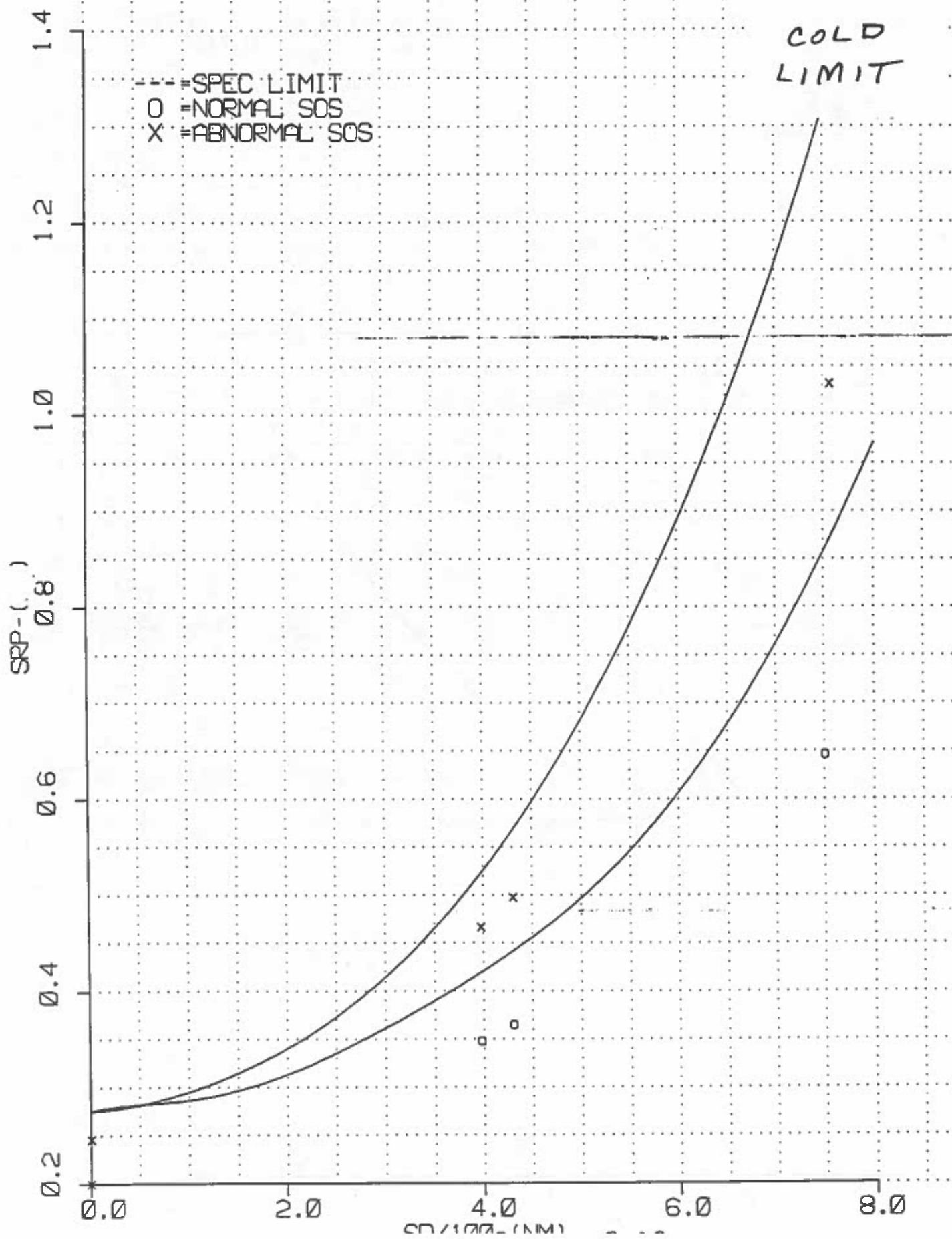
--- = SPEC LIMIT
O = +Z EOS
X = -Z EOS

COLD
LIMIT

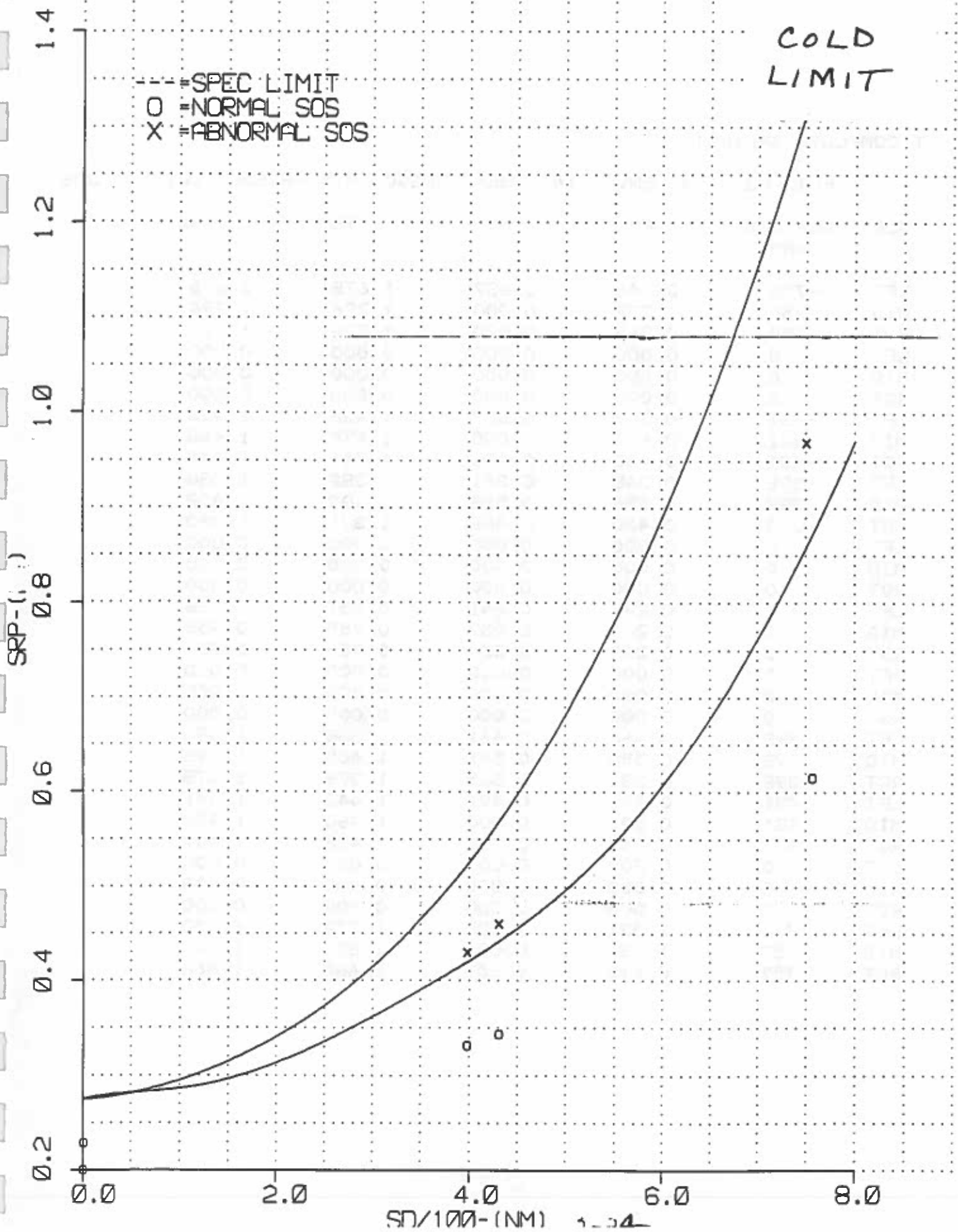
SRP-()

SN/100-(NM)

SYSTEM 14 , SRP TF L FBAK , SSS=3 , M1=-8 , DATE:326 /



SYSTEM 14, SRP TF R FBAK, SSS=3, M1=-8, DATE:326



T, COMPLETE, SRP (NM)

FLT. NO. = 14 ENV. = 14 SSS= 3DEGC M1= -8DEGC DATE: 326

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	-750.	0.646	0.637	1.678	1.676
MID	-750.	1.237	0.000	1.796	1.794
RGT	-750.	0.969	0.970	1.723	1.721
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	-431.	0.365	0.357	1.436	1.435
MID	-431.	0.604	0.000	1.470	1.468
RGT	-431.	0.460	0.456	1.441	1.440
LFT	-398.	0.348	0.341	1.382	1.380
MID	-398.	0.596	0.588	1.409	1.408
RGT	-398.	0.430	0.426	1.391	1.390
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	0.	0.246	0.241	0.957	0.956
MID	0.	0.241	0.237	0.955	0.955
RGT	0.	0.228	0.224	0.955	0.954
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	398.	0.466	0.461	1.388	1.386
MID	398.	0.550	0.541	1.400	1.398
RGT	398.	0.331	0.325	1.379	1.378
LFT	431.	0.497	0.491	1.443	1.441
MID	431.	0.554	0.000	1.460	1.458
RGT	431.	0.343	0.336	1.438	1.437
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	757.	1.031	1.035	1.752	1.750
MID	757.	1.434	0.000	1.870	1.867
RGT	757.	0.615	0.607	1.668	1.666

T, COMPLETE, SRP RATIO

SEG	SUR. DIST. (NM)	TFP	TFB	TSP	TSB
LFT	-750.	0.752	0.742	0.746	0.745
MID	-750.	0.000	0.000	0.799	0.798
RGT	-750.	0.734	0.735	0.766	0.766
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	-431.	0.826	0.809	0.897	0.896
MID	-431.	0.000	0.000	0.918	0.917
RGT	-431.	0.810	0.802	0.900	0.899
LFT	-398.	0.829	0.812	0.901	0.900
MID	-398.	0.907	0.894	0.919	0.918
RGT	-398.	0.822	0.814	0.907	0.906
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	0.	0.893	0.876	0.912	0.911
MID	0.	0.834	0.818	0.910	0.909
RGT	0.	0.830	0.815	0.909	0.908
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	398.	0.892	0.882	0.905	0.904
MID	398.	0.837	0.824	0.913	0.912
RGT	398.	0.790	0.774	0.899	0.898
LFT	431.	0.875	0.863	0.901	0.900
MID	431.	0.000	0.000	0.912	0.911
RGT	431.	0.777	0.761	0.898	0.897
LFT	0.	0.000	0.000	0.000	0.000
MID	0.	0.000	0.000	0.000	0.000
RGT	0.	0.000	0.000	0.000	0.000
LFT	757.	0.768	0.771	0.775	0.774
MID	757.	0.000	0.000	0.827	0.826
RGT	757.	0.705	0.696	0.738	0.737

TF, LEFT, PRIMARY

FLT. NO. = 14 ENV. = 14 SSS= 3DEGC M1= -8DEGC DATE: 326

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.646	0.752
0.	0.000	0.000
-431.	0.365	0.826
-398.	0.348	0.829
0.	0.000	0.000
0.	0.246	0.893
0.	0.000	0.000
398.	0.466	0.892
431.	0.497	0.875
0.	0.000	0.000
757.	1.031	0.768

TF, LEFT, BACKUP

FLT. NO. = 14 ENV. = 14 SSS= 3DEGC M1= -8DEGC DATE: 326

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.637	0.742
0.	0.000	0.000
-431.	0.357	0.809
-398.	0.341	0.812
0.	0.000	0.000
0.	0.241	0.876
0.	0.000	0.000
398.	0.461	0.882
431.	0.491	0.863
0.	0.000	0.000
757.	1.035	0.771

TF, RIGHT, PRIMARY

FLT. NO. = 14 ENV. = 14 SSS= 3DEGC M1= -8DEGC DATE: 326

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.969	0.734
0.	0.000	0.000
-431.	0.460	0.810
-398.	0.430	0.822
0.	0.000	0.000
0.	0.228	0.830
0.	0.000	0.000
398.	0.331	0.790
431.	0.343	0.777
0.	0.000	0.000
757.	0.615	0.705

TF RIGHT, BACKUP

FLT. NO. = 14 ENV. = 14 SSS= 3DEGC M1= -8DEGC DATE: 326

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	0.970	0.735
0.	0.000	0.000
-431.	0.456	0.802
-398.	0.426	0.814
0.	0.000	0.000
0.	0.224	0.815
0.	0.000	0.000
398.	0.325	0.774
431.	0.336	0.761
0.	0.000	0.000
757.	0.607	0.696

2.2 Geometric Resolution (Cont'd)

2.2.2 Fine Geometric Resolution - Daytime Visual (3.2.1.1.2.1)

2.2.2.1 Baseline (Orbit Nominal)

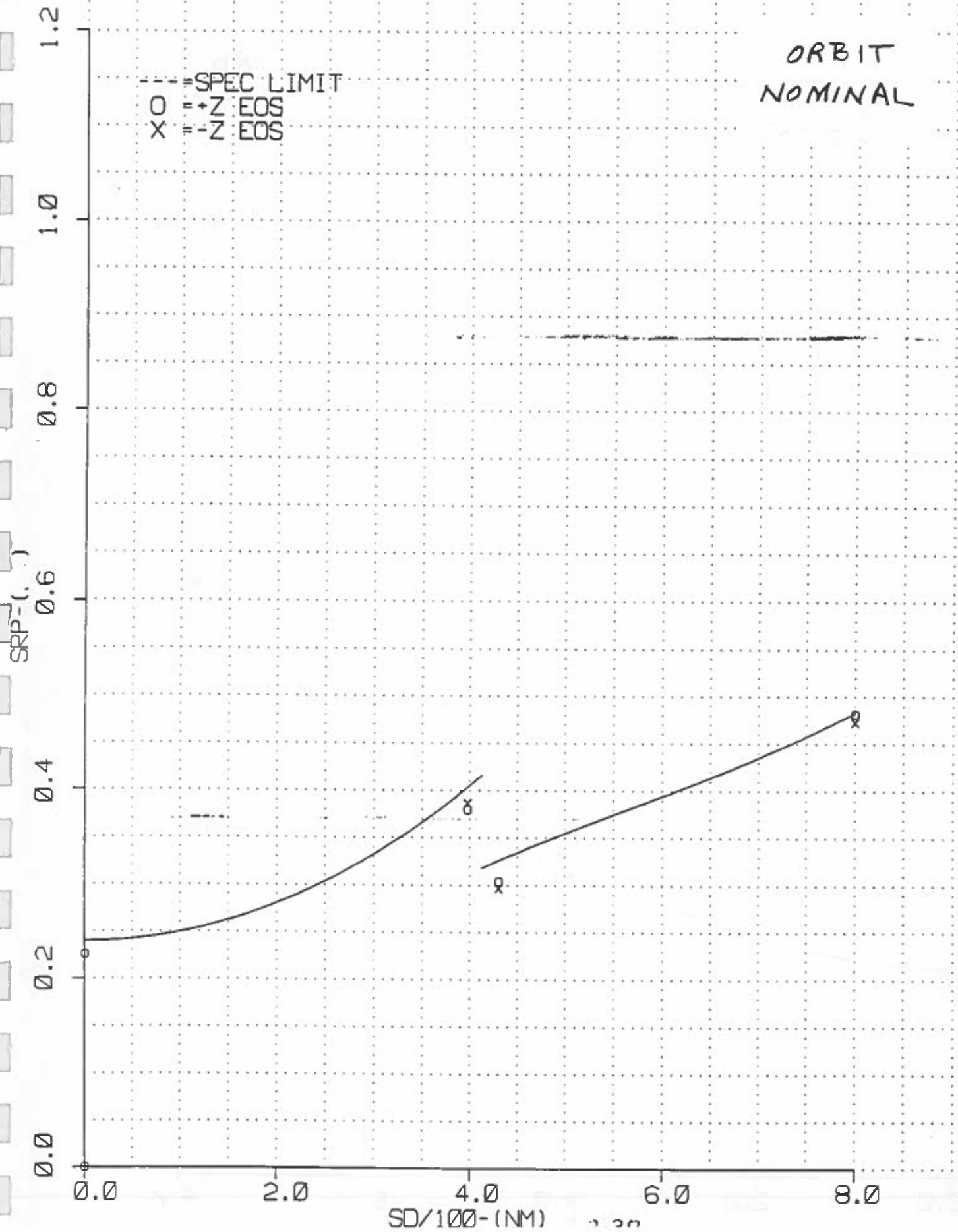
The LF SRP is within the specification limits in both Primary and Redundant configurations.

- ATTACHMENTS: LF SRP Curves Orbit Nominal
- LF SRP Tables Orbit Nominal

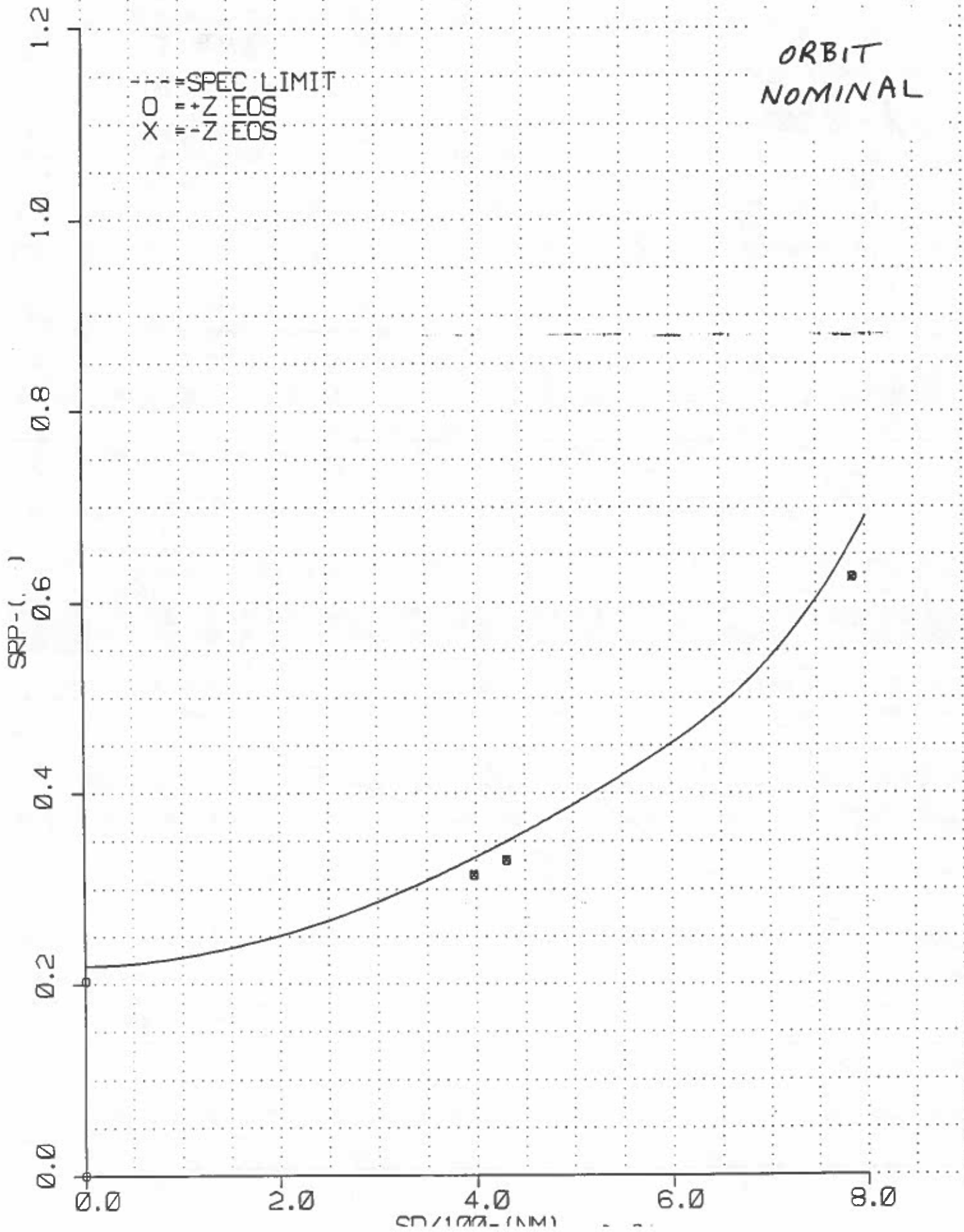
SYSTEM 14 ,SRP LF NORMAL,SSS=5... ,M1=-8 ,DATE:331

ORBIT
NOMINAL

--- = SPEC LIMIT
O = +Z EOS
X = -Z EOS



SYSTEM 14 ,SRP LF FBACK ,SSS=5 ,M1=-8 ,DATE:331



LF, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-800.	0.472	0.978
0.	0.000	0.000
-431.	0.297	0.909
-398.	0.387	0.959
0.	0.000	0.000
0.	0.225	0.940
0.	0.000	0.000
398.	0.380	0.941
431.	0.303	0.930
0.	0.000	0.000
800.	0.481	0.996

LF, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-800.	0.473	0.981
0.	0.000	0.000
-431.	0.296	0.909
-398.	0.388	0.961
0.	0.000	0.000
0.	0.225	0.939
0.	0.000	0.000
398.	0.381	0.944
431.	0.303	0.929
0.	0.000	0.000
800.	0.482	0.998

LF, DAY, FALLBACK, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-787.	0.626	0.942
0.	0.000	0.000
-431.	0.330	0.944
-398.	0.315	0.947
0.	0.000	0.000
0.	0.203	0.926
0.	0.000	0.000
398.	0.314	0.945
431.	0.330	0.943
0.	0.000	0.000
788.	0.626	0.941

LF, DAY, FALLBACK, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-787.	0.625	0.942
0.	0.000	0.000
-431.	0.330	0.944
-398.	0.315	0.946
0.	0.000	0.000
0.	0.203	0.925
0.	0.000	0.000
398.	0.314	0.945
431.	0.330	0.943
0.	0.000	0.000
788.	0.626	0.941

2.2 Geometric Resolution (Cont'd)

2.2.2 Fine Geometric Resolution - Daytime Visual (Cont'd)

(3.2.1.1.2.1)

2.2.2.2 Acceptance - Vibration

OLS #14 underwent Acceptance-level SSS vibration on December 11, 1992. The Pre-to-Post Vibration SRP performance is within Specification requirements and is shown on the attached curves and tables. No changes in SRP performance occurred as a result of vibration.

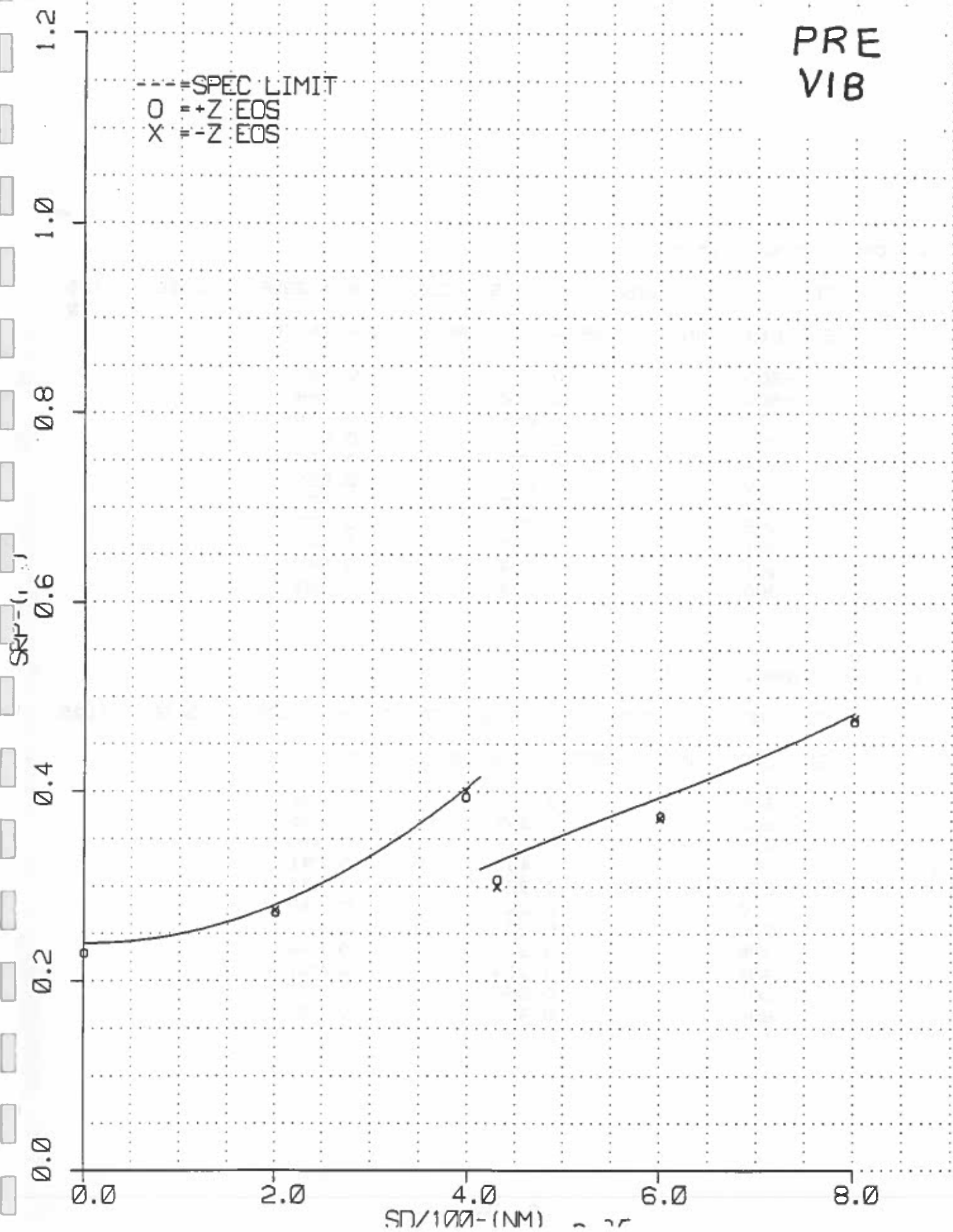
ATTACHMENTS: LF SRP Curves Pre-Vibration
LF SRP Tables Pre-Vibration
LF SRP Curves Post-Vibration
LF SRP Tables Post-Vibration

Faint, illegible text covering the majority of the page, likely bleed-through from the reverse side. The text is too light to transcribe accurately.



PRE
VIB

--- =SPEC LIMIT
O =+Z EOS
X =-Z EOS



BLRM

LF, DAY, NORMAL, PRIMARY

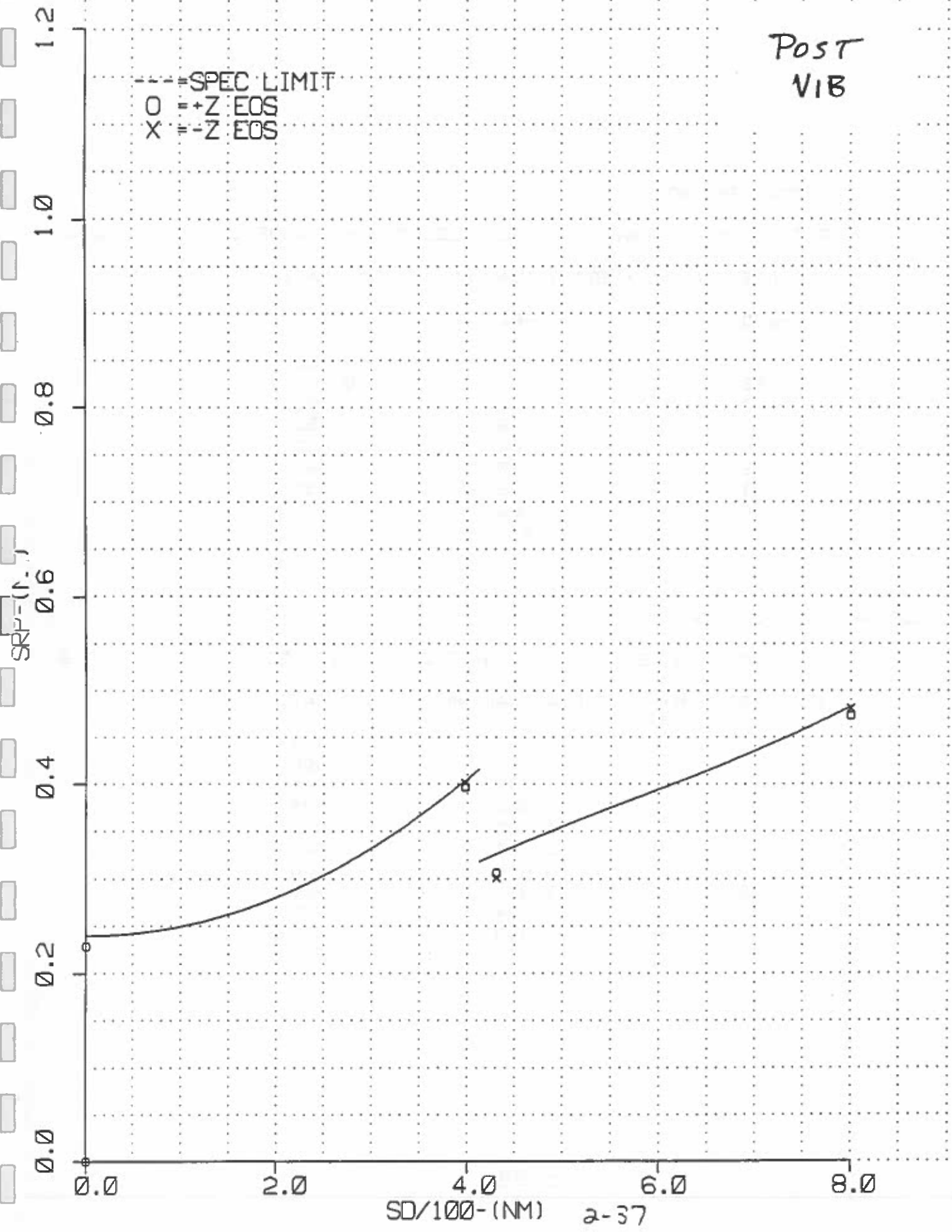
FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1125
1992

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.477	0.987
-600.	0.372	0.944
-431.	0.299	0.918
-398.	0.400	0.992
-200.	0.275	0.980
0.	0.229	0.955
200.	0.272	0.971
398.	0.394	0.977
431.	0.307	0.940
601.	0.374	0.948
800.	0.474	0.981

LF, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1125

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.476	0.986
-600.	0.369	0.935
-431.	0.297	0.910
-398.	0.400	0.991
-200.	0.273	0.975
0.	0.227	0.948
200.	0.271	0.965
398.	0.394	0.976
431.	0.304	0.931
601.	0.370	0.938
800.	0.473	0.980



LF, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1218

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.481	0.996
0.	0.000	0.000
-431.	0.301	0.923
-398.	0.401	0.993
0.	0.000	0.000
0.	0.228	0.952
0.	0.000	0.000
398.	0.397	0.984
431.	0.306	0.938
0.	0.000	0.000
800.	0.473	0.979

LF, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1218

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.482	0.998
0.	0.000	0.000
-431.	0.299	0.916
-398.	0.402	0.995
0.	0.000	0.000
0.	0.227	0.948
0.	0.000	0.000
398.	0.397	0.985
431.	0.304	0.931
0.	0.000	0.000
800.	0.474	0.982

2.2 Geometric Resolution (Cont'd)

2.2.2 Fine Geometric Resolution - Daytime Visual (Cont'd)

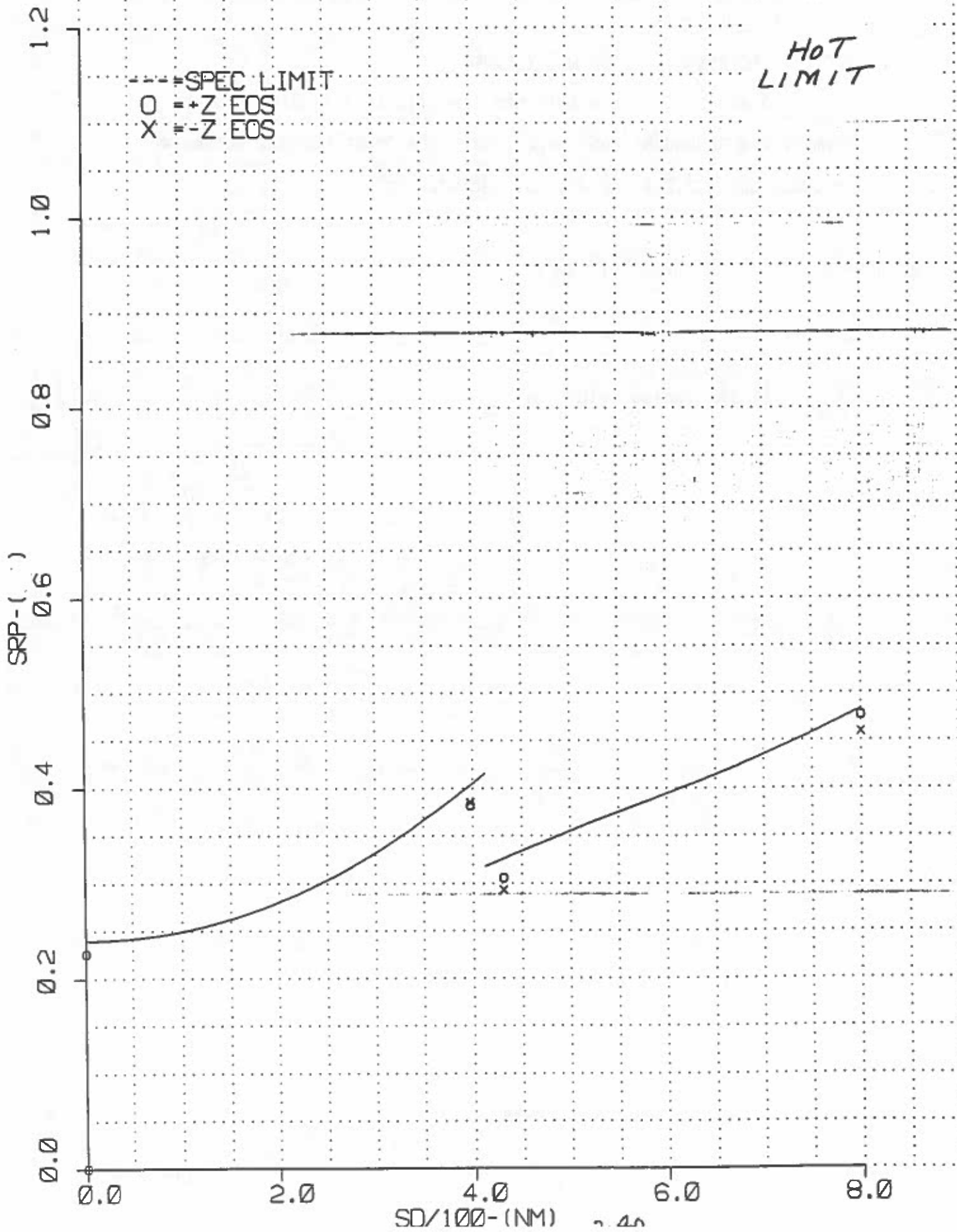
(3.2.1.1.2.1)

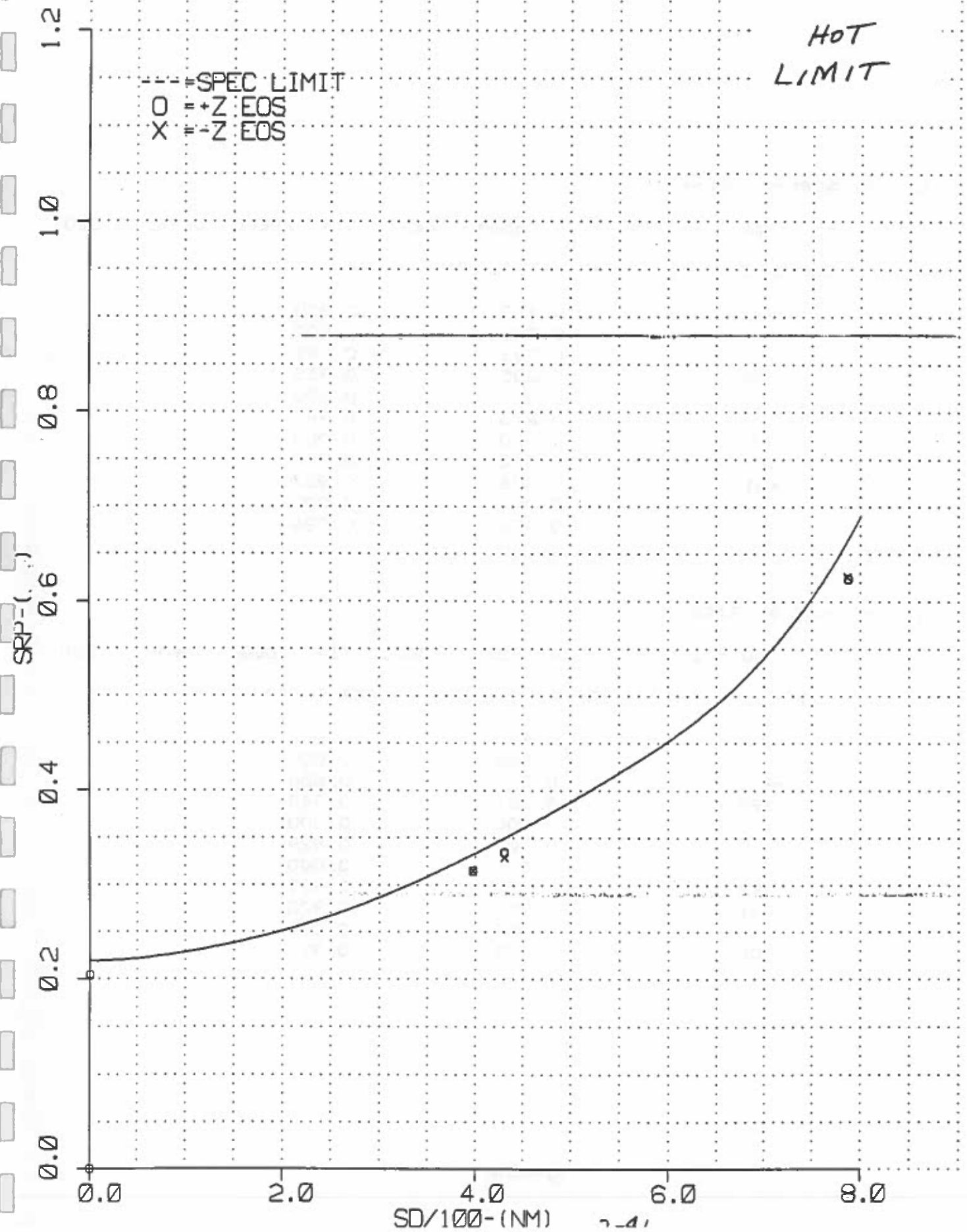
2.2.2.3 Acceptance - Thermal Vacuum

OLS #14 LF SRP is within the specification limits in both Primary and Redundant configurations. The Orbit Nominal curves are in paragraph 2.2.2.1 and are not included here.

ATTACHMENTS: LF SRP Curve Hot Limit
LF SRP Tables Hot Limit
LF SRP Curves Cold Limit
LF SRP Tables Cold Limit

SYSTEM 14 , SRP LF NORMAL , SSS=7 , M1=12 , DATE:320





LF, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 320

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.459	0.950
0.	0.000	0.000
-431.	0.293	0.899
-398.	0.385	0.955
0.	0.000	0.000
0.	0.226	0.942
0.	0.000	0.000
398.	0.382	0.947
431.	0.306	0.937
0.	0.000	0.000
800.	0.476	0.986

LF, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 320

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.454	0.940
0.	0.000	0.000
-431.	0.289	0.886
-398.	0.381	0.945
0.	0.000	0.000
0.	0.223	0.929
0.	0.000	0.000
398.	0.378	0.937
431.	0.301	0.923
0.	0.000	0.000
800.	0.471	0.975

LF, DAY, FALLBACK, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 320

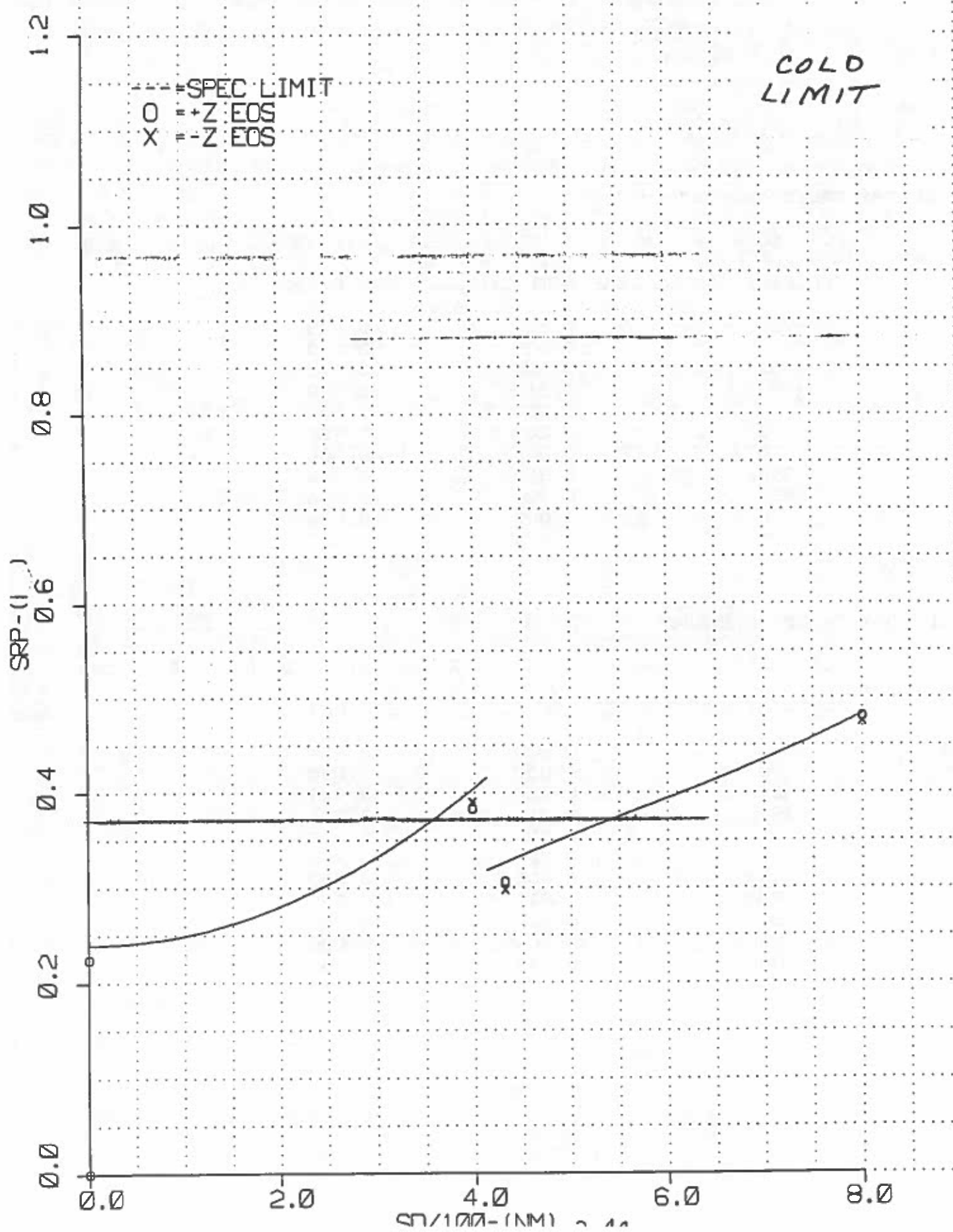
SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-787.	0.626	0.943
0.	0.000	0.000
-431.	0.330	0.943
-398.	0.315	0.948
0.	0.000	0.000
0.	0.204	0.930
0.	0.000	0.000
398.	0.315	0.948
431.	0.334	0.956
0.	0.000	0.000
788.	0.624	0.937

LF, DAY, FALLBACK, BACKUP

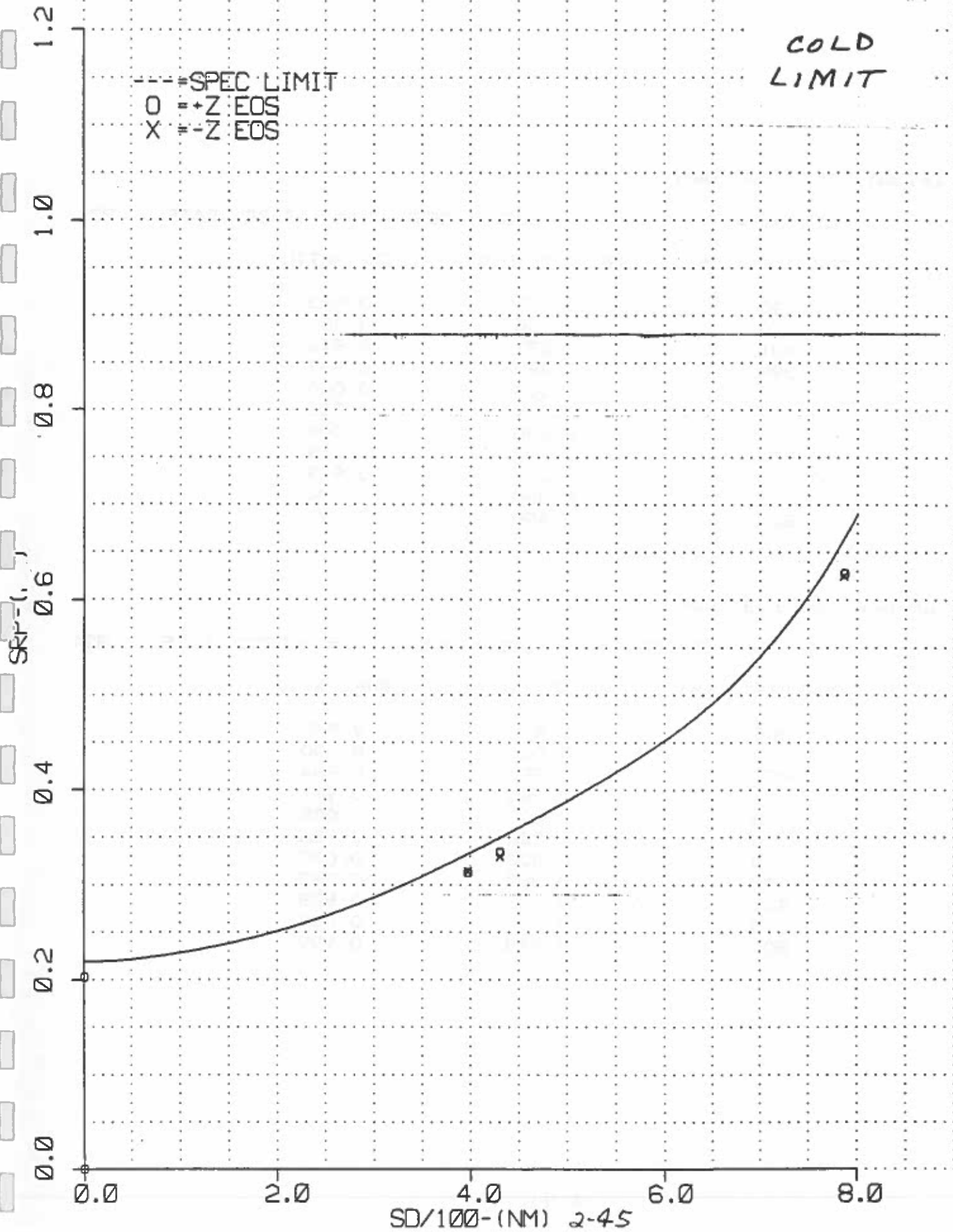
FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 320

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-787.	0.620	0.934
0.	0.000	0.000
-431.	0.325	0.929
-398.	0.311	0.934
0.	0.000	0.000
0.	0.201	0.916
0.	0.000	0.000
398.	0.311	0.935
431.	0.330	0.942
0.	0.000	0.000
788.	0.619	0.930

SYSTEM 14 , SRP LF NORMAL , SSS=3 , M1=-8 , DATE:324



SYSTEM 14 ,SRP LF FBACK ,SSS=3... ,M1=-8 ,DATE:324



LF, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 324

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.475	0.983
0.	0.000	0.000
-431.	0.297	0.912
-398.	0.390	0.967
0.	0.000	0.000
0.	0.224	0.935
0.	0.000	0.000
398.	0.383	0.949
431.	0.306	0.937
0.	0.000	0.000
800.	0.480	0.994

LF, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 324

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	0.472	0.978
0.	0.000	0.000
-431.	0.295	0.904
-398.	0.388	0.961
0.	0.000	0.000
0.	0.222	0.926
0.	0.000	0.000
398.	0.380	0.943
431.	0.303	0.928
0.	0.000	0.000
800.	0.478	0.989

LF, DAY, FALLBACK, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 324

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-787.	0.625	0.941
0.	0.000	0.000
-431.	0.330	0.944
-398.	0.314	0.944
0.	0.000	0.000
0.	0.202	0.925
0.	0.000	0.000
398.	0.313	0.941
431.	0.334	0.955
0.	0.000	0.000
788.	0.628	0.943

LF, DAY, FALLBACK, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 324

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-787.	0.621	0.936
0.	0.000	0.000
-431.	0.327	0.934
-398.	0.311	0.935
0.	0.000	0.000
0.	0.201	0.916
0.	0.000	0.000
398.	0.310	0.932
431.	0.331	0.946
0.	0.000	0.000
788.	0.624	0.937

2.2 Geometric Resolution (Cont'd)

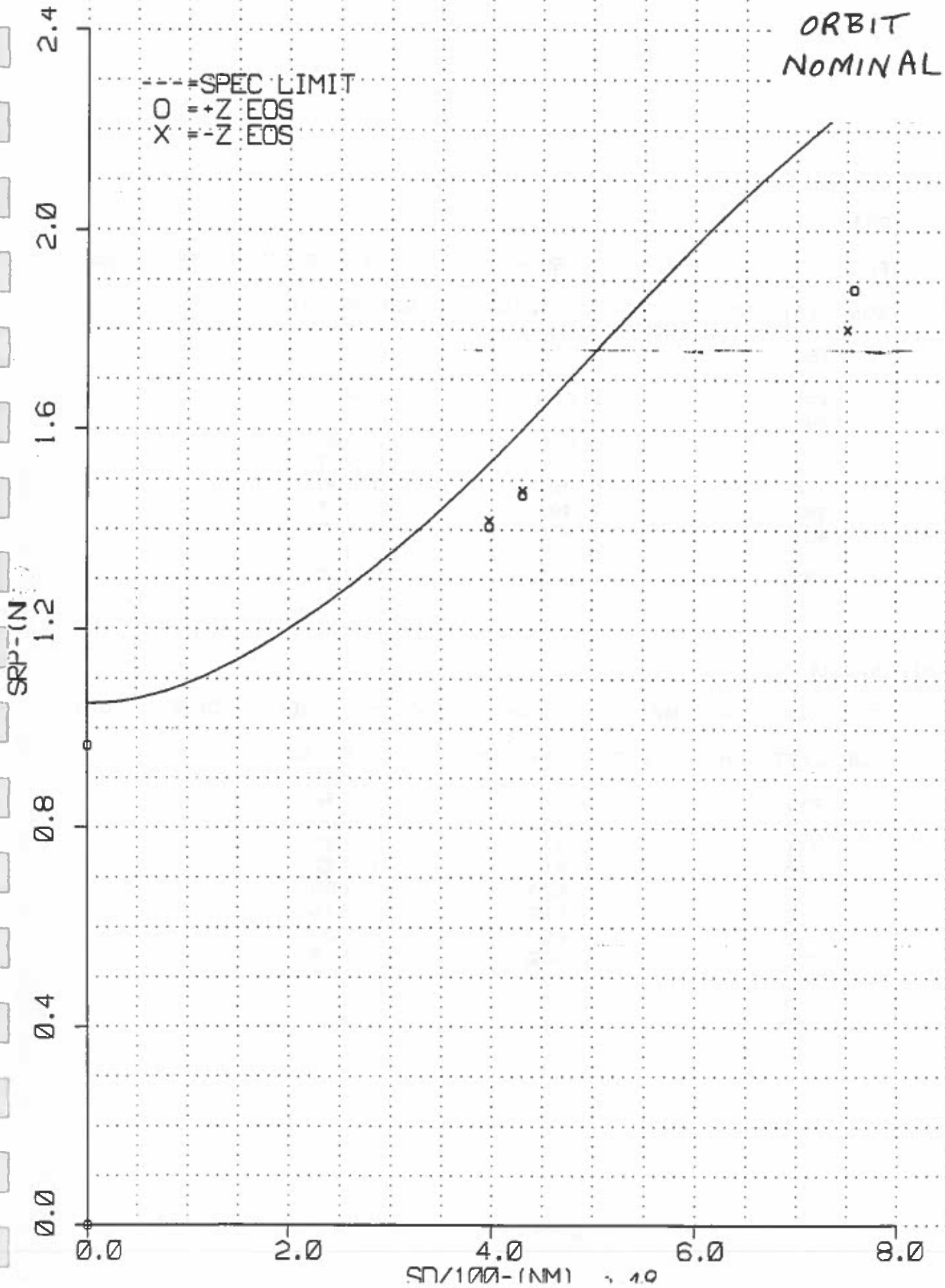
2.2.3 Smoothed Geometric Resolution - Infrared (3.2.1.1.2.2)

2.2.3.1 Baseline (Orbit Nominal)

The TS SRP is within spec for all measured scan angles.

ATTACHMENTS: TS SRP Curve Orbit Nominal

TS SRP Tables Orbit Nominal



TS, MID, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	1.802	0.801
0.	0.000	0.000
-431.	1.478	0.923
-398.	1.419	0.925
0.	0.000	0.000
0.	0.964	0.919
0.	0.000	0.000
398.	1.406	0.917
431.	1.470	0.918
0.	0.000	0.000
757.	1.882	0.833

TS, MID, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	1.791	0.796
0.	0.000	0.000
-431.	1.473	0.920
-398.	1.414	0.922
0.	0.000	0.000
0.	0.962	0.916
0.	0.000	0.000
398.	1.402	0.914
431.	1.465	0.915
0.	0.000	0.000
757.	1.866	0.826

2.2 Geometric Resolution (Cont'd)

2.2.3 Smoothed Geometric Resolution - Infrared (Cont'd)

(3.2.1.1.2.2)

2.2.3.2 Acceptance - Vibration

Consistant with the test philosophy established in BVS 2579, no vibration data was taken for the T channel as part of Bearing Retrofit since stability of the T alignment for the common optics can be inferred from H alignment testing. Non-common optics were verified to be stable during the original OLS 14 testing.

ATTACHMENTS: N/A

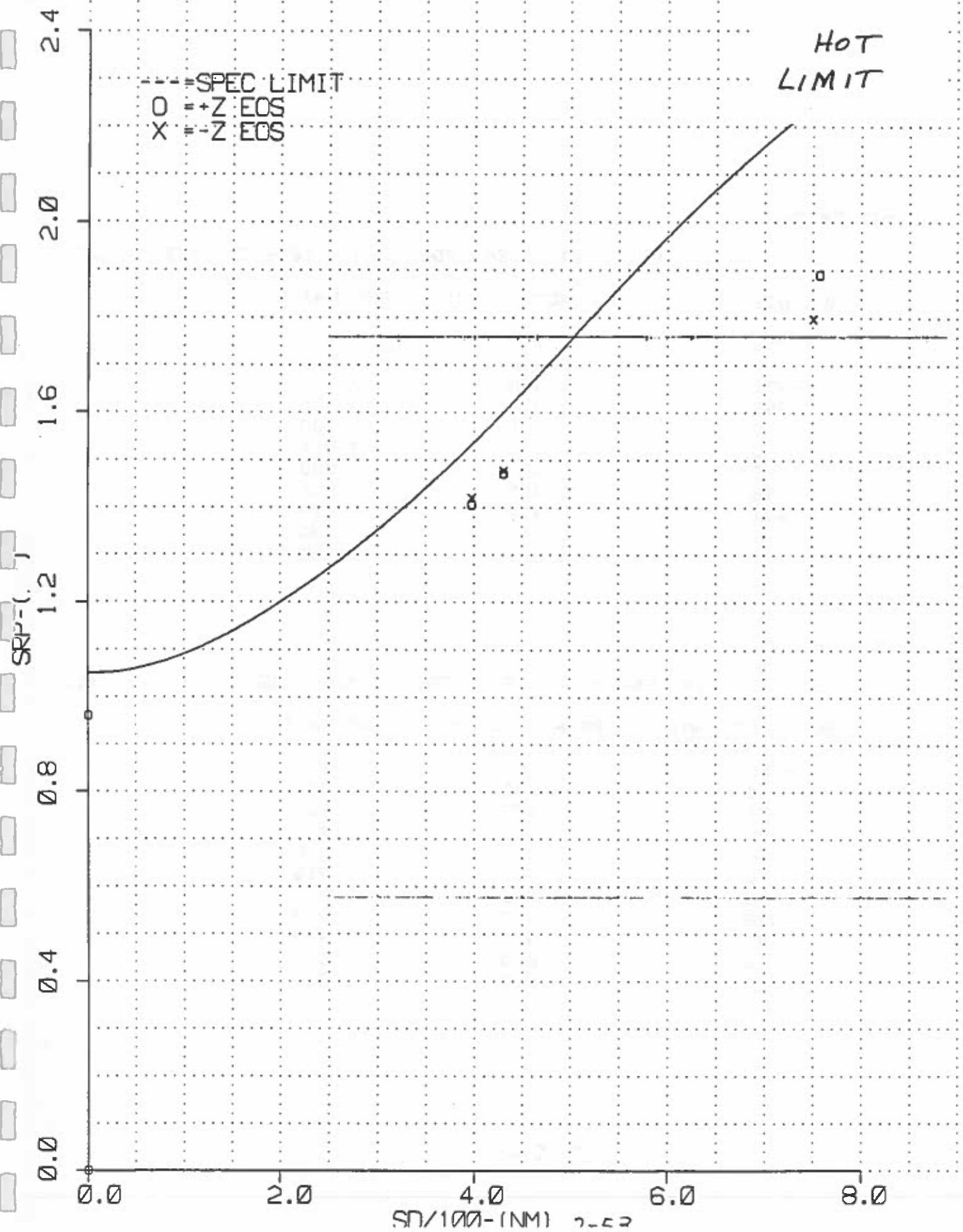
2.2 Geometric Resolution (Cont'd)

2.2.3 Smoothed Geometric Resolution - Infrared (3.2.1.1.2.2)

2.2.3.3 Acceptance - Thermal Vacuum

The TS SRP is within spec for the extremes of hot and cold Thermal Vacuum limit testing.

ATTACHMENTS: TS SRP Curve Hot Limits
TS SRP Tables Hot Limits
TS SRP Curve Cold Limits
TS SRP Tables Cold Limits



--- = SPEC LIMIT
o = +Z EOS
x = -Z EOS

HOT
LIMIT

TS, MID, PRIMARY

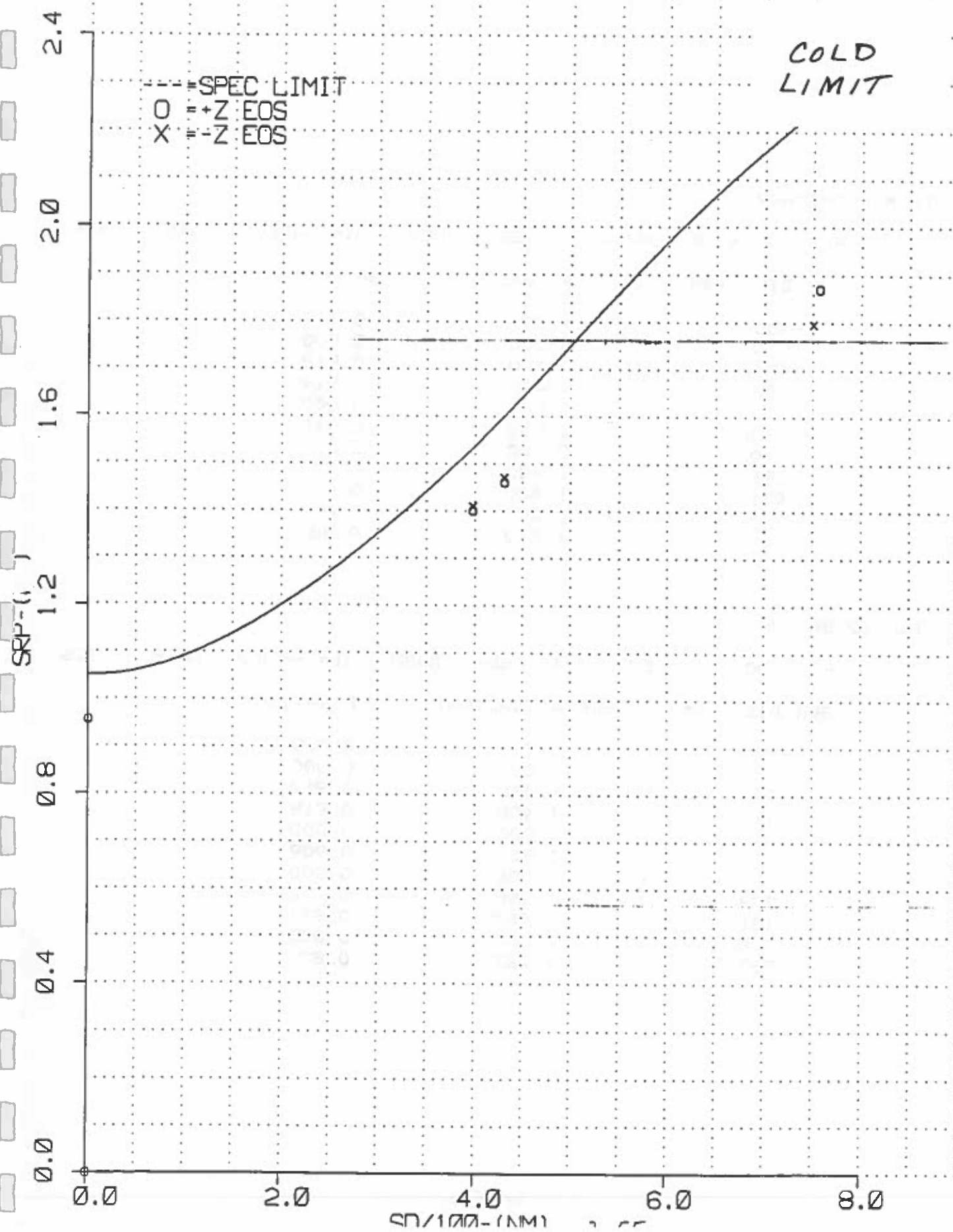
FLT. NO. = 14 ENV. = 14 SSS= 7DEGC M1= 12DEGC DATE: 321

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	1.798	0.799
0.	0.000	0.000
-431.	1.478	0.923
-398.	1.420	0.926
0.	0.000	0.000
0.	0.960	0.914
0.	0.000	0.000
398.	1.405	0.916
431.	1.472	0.919
0.	0.000	0.000
757.	1.889	0.836

TS, MID, BACKUP

FLT. NO. = 14 ENV. = 14 SSS= 7DEGC M1= 12DEGC DATE: 321

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	1.795	0.798
0.	0.000	0.000
-431.	1.478	0.923
-398.	1.421	0.926
0.	0.000	0.000
0.	0.961	0.916
0.	0.000	0.000
398.	1.407	0.917
431.	1.472	0.919
0.	0.000	0.000
757.	1.882	0.833



TS, MID, PRIMARY

FLT. NO. = 14 ENV. = 14 SSS= 3DEGC M1= -8DEGC DATE: 326

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	1.796	0.799
0.	0.000	0.000
-431.	1.470	0.918
-398.	1.409	0.919
0.	0.000	0.000
0.	0.955	0.910
0.	0.000	0.000
398.	1.400	0.913
431.	1.460	0.912
0.	0.000	0.000
757.	1.870	0.827

TS, MID, BACKUP

FLT. NO. = 14 ENV. = 14 SSS= 3DEGC M1= -8DEGC DATE: 326

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-750.	1.794	0.798
0.	0.000	0.000
-431.	1.468	0.917
-398.	1.408	0.918
0.	0.000	0.000
0.	0.955	0.909
0.	0.000	0.000
398.	1.398	0.912
431.	1.458	0.911
0.	0.000	0.000
757.	1.867	0.826

2.2 Geometric Resolution (Cont'd)

2.2.4 Smoothed Geometric Resolution - Daytime Visual (3.2.1.1.2.2)

2.2.4.1 Baseline (Orbit Nominal)

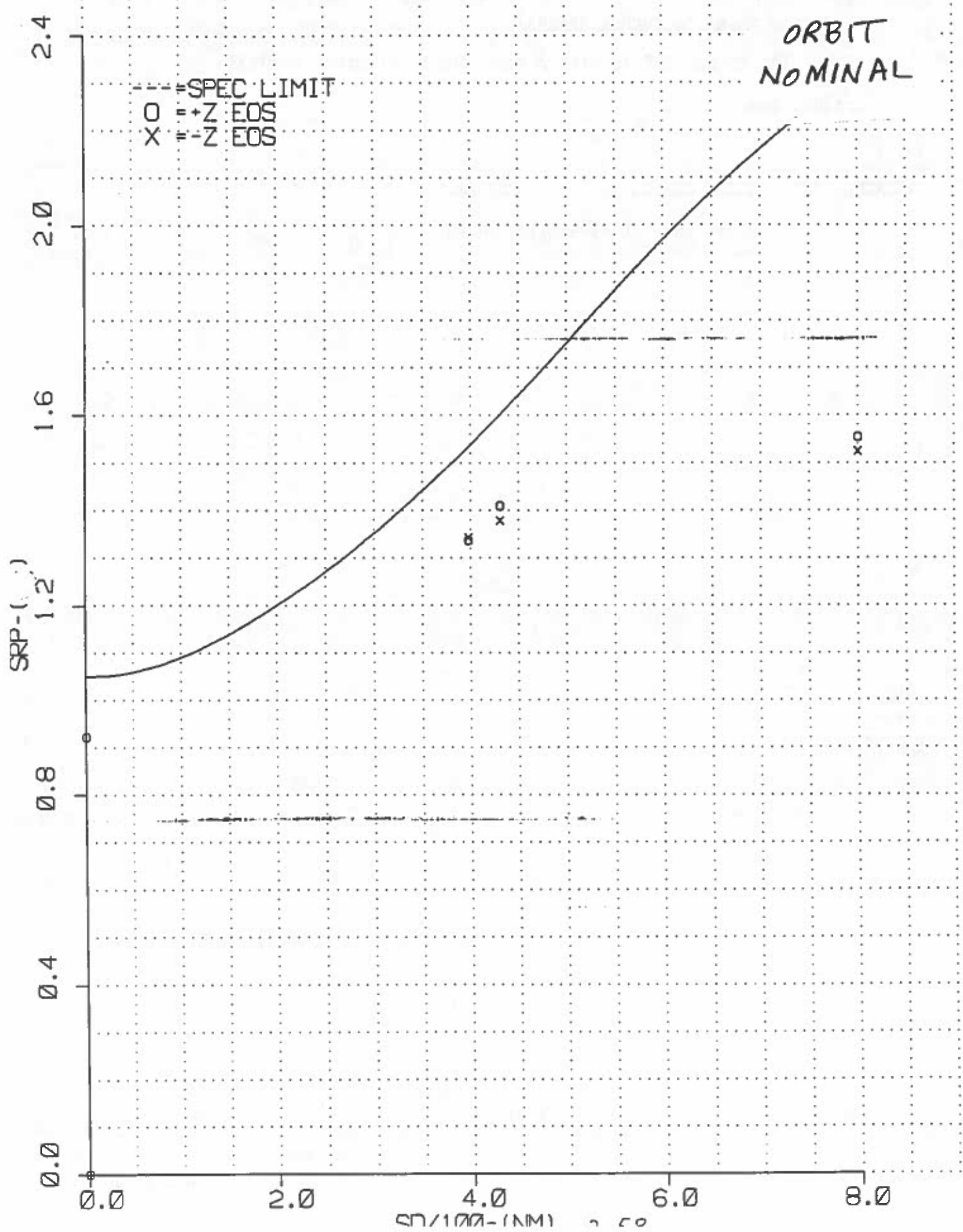
The LS Day SRP is within spec limits at Orbit Nominal conditions.

ATTACHMENTS: LS Day SRP Curve - Orbit Nominal

LS Day SRP Tables - Orbit Nominal

SYSTEM 14 , SRP LS. DAY/ ~~1968~~

SSS=5 , M1 = -8 , DATE: 331



LS, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-800.	1.524	0.649
0.	0.000	0.000
-431.	1.379	0.861
-398.	1.342	0.875
0.	0.000	0.000
0.	0.922	0.878
0.	0.000	0.000
398.	1.337	0.872
431.	1.409	0.880
0.	0.000	0.000
800.	1.554	0.661

LS, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= SDEGC M1= -8DEGC DATE: 331

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-800.	1.514	0.644
0.	0.000	0.000
-431.	1.370	0.856
-398.	1.334	0.870
0.	0.000	0.000
0.	0.916	0.872
0.	0.000	0.000
398.	1.328	0.866
431.	1.400	0.874
0.	0.000	0.000
800.	1.542	0.656

2.2 Geometric Resolution (Cont'd)

2.2.4 Smoothed Geometric Resolution - Daytime Visual (Cont'd)

(3.2.1.1.2.2)

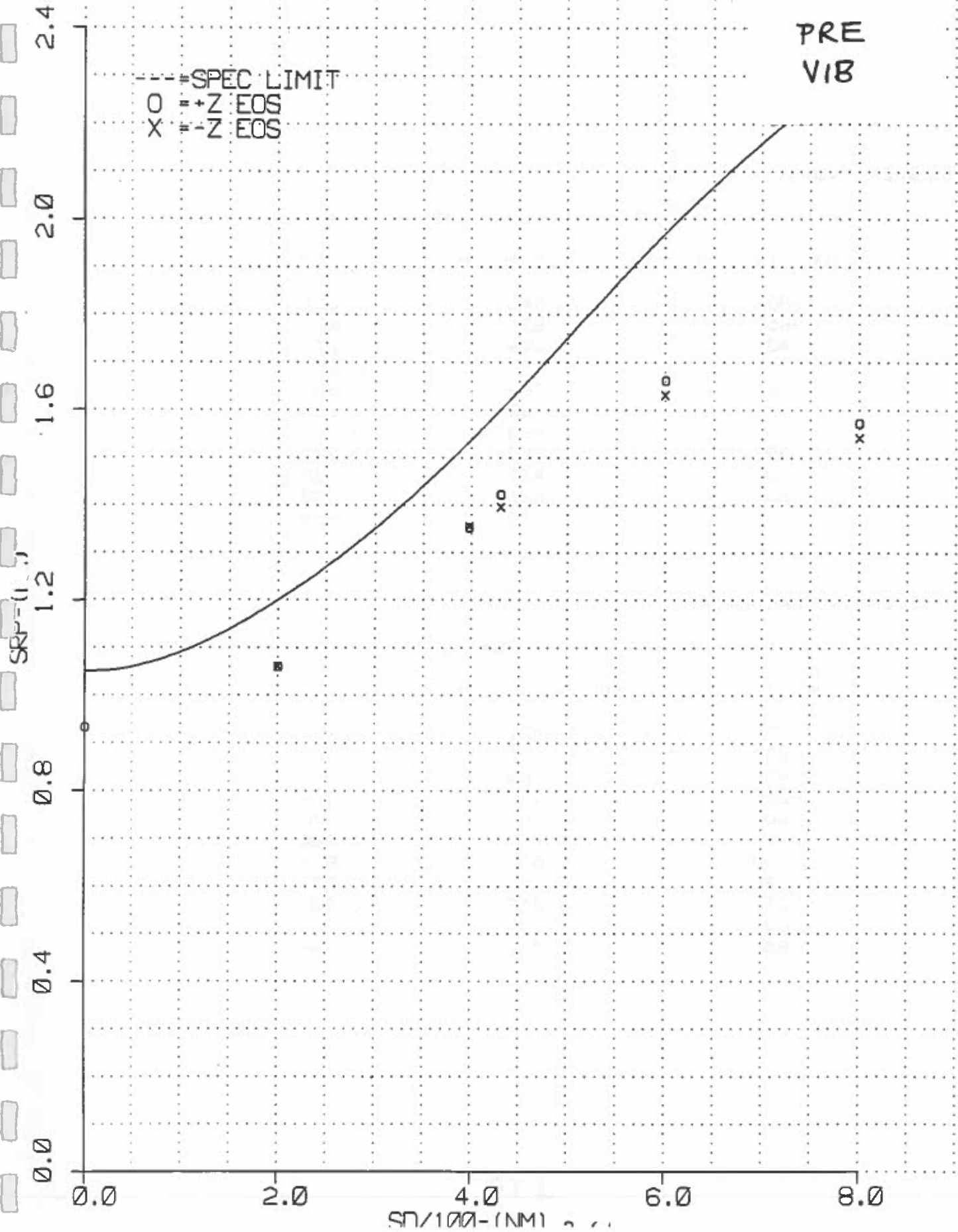
2.2.4.2 Acceptance - Vibration

The OLS #14 SSS underwent vibrations on December 11, 1992. The LS Day SRP is within specification both before and after SSS vibration. No vibration-related changes in SRP were observed.

ATTACHMENTS: LS Day SRP Curve Pre-Vibration
LS Day SRP Tables Pre-Vibration
LS Day SRP Curve Post-Vibration
LS Day SRP Table Post-Vibration

SYSTEM 14 ,SRP .LS. DAY/IS ~~SRM~~

SSS=23 ,M1=23 ,DATE:1125



LS, DAY, NORMAL, PRIMARY

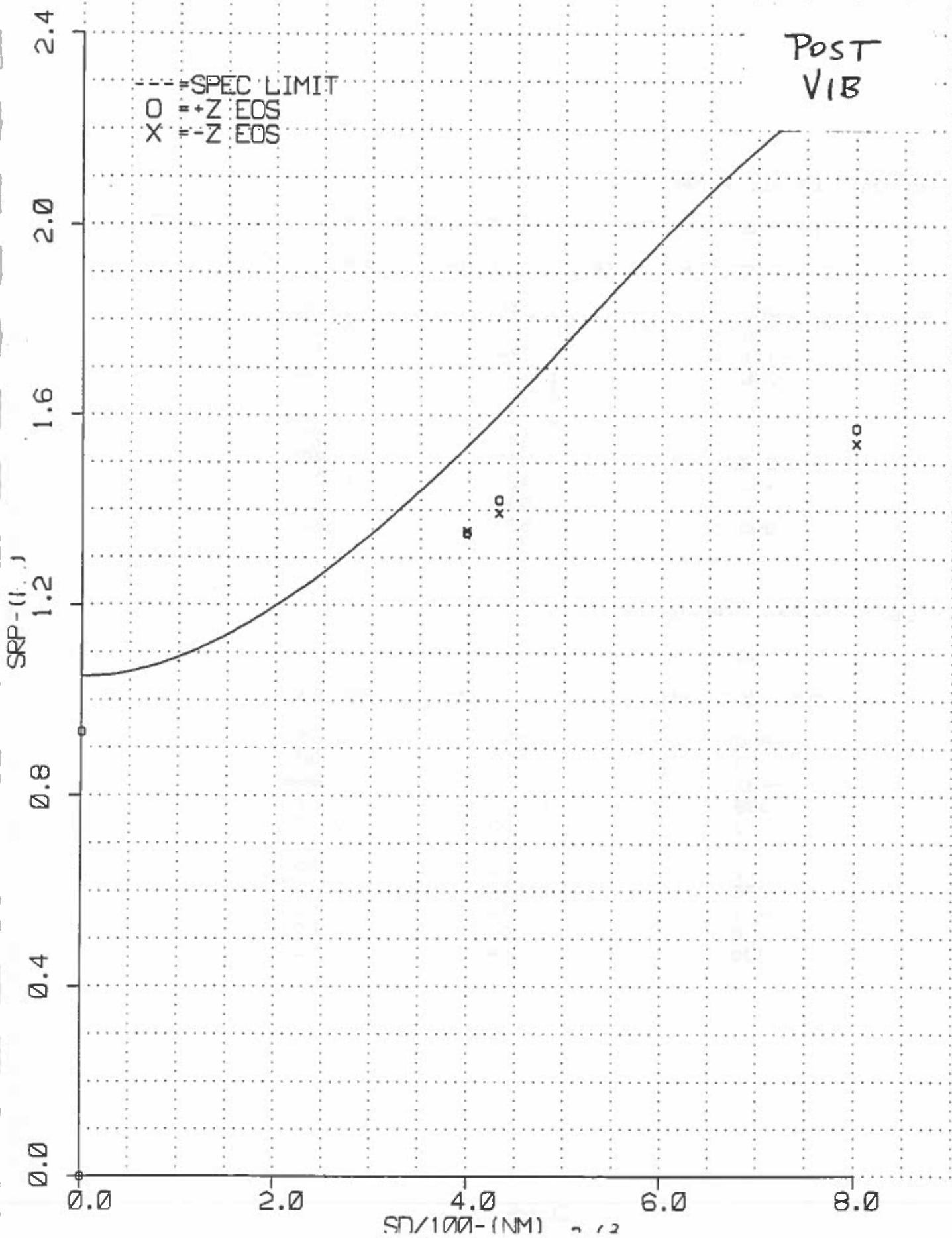
FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1125

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	1.542	0.656
-600.	1.631	0.828
-431.	1.396	0.872
-398.	1.356	0.884
-200.	1.060	0.884
0.	0.932	0.888
200.	1.060	0.883
398.	1.353	0.882
431.	1.422	0.888
601.	1.661	0.843
800.	1.572	0.669

LS, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1125

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	1.532	0.652
-600.	1.621	0.823
-431.	1.388	0.867
-398.	1.348	0.879
-200.	1.053	0.878
0.	0.926	0.882
200.	1.053	0.878
398.	1.344	0.877
431.	1.412	0.882
601.	1.650	0.837
800.	1.561	0.664



LS, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1218

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	1.541	0.656
0.	0.000	0.000
-431.	1.394	0.870
-398.	1.356	0.884
0.	0.000	0.000
0.	0.933	0.888
0.	0.000	0.000
398.	1.352	0.882
431.	1.420	0.887
0.	0.000	0.000
800.	1.573	0.669

LS, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1218

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	1.533	0.652
0.	0.000	0.000
-431.	1.387	0.866
-398.	1.349	0.879
0.	0.000	0.000
0.	0.928	0.884
0.	0.000	0.000
398.	1.345	0.877
431.	1.413	0.882
0.	0.000	0.000
800.	1.564	0.665

2.2 Geometric Resolution (Cont'd)

2.2.4 Smoothed Geometric Resolution - Daytime Visual (Cont'd)

(3.2.1.1.2.2)

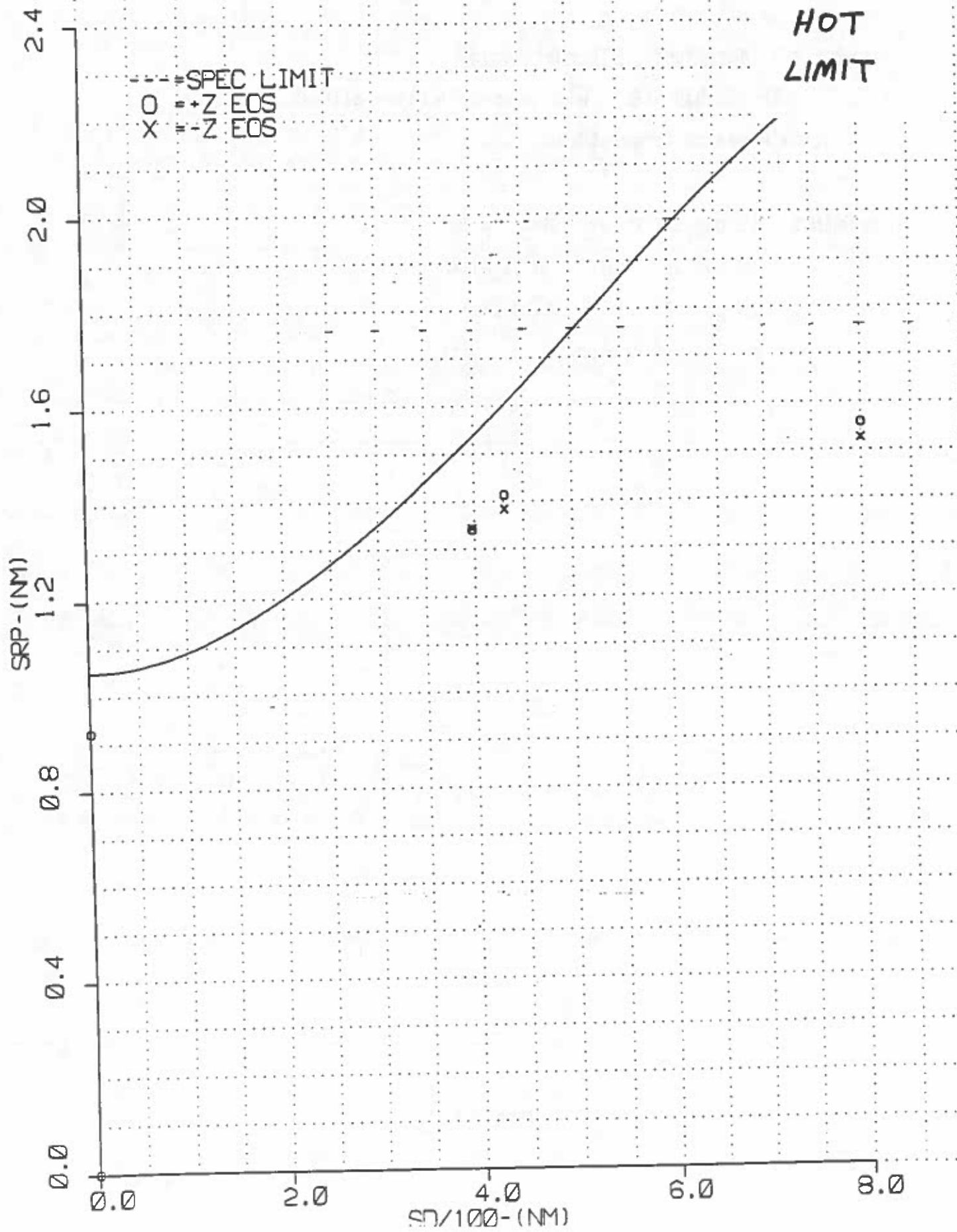
2.2.4.3 Acceptance - Thermal Vacuum

The LS Day SRP is within specification allowance over the entire range of temperatures.

ATTACHMENTS: LS Day SRP Curve Hot Limits
LS Day SRP Tables Hot Limits
LS Day SRP Curve Cold Limits
LS Day SRP Tables Cold Limits

SYSTEM 14, SRP LS. DAY/

SSS=7, M1=12, DATE:320



LS, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 320

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-800.	1.525	0.649
0.	0.000	0.000
-431.	1.383	0.864
-398.	1.344	0.876
0.	0.000	0.000
0.	0.923	0.879
0.	0.000	0.000
398.	1.340	0.874
431.	1.413	0.882
0.	0.000	0.000
800.	1.557	0.663

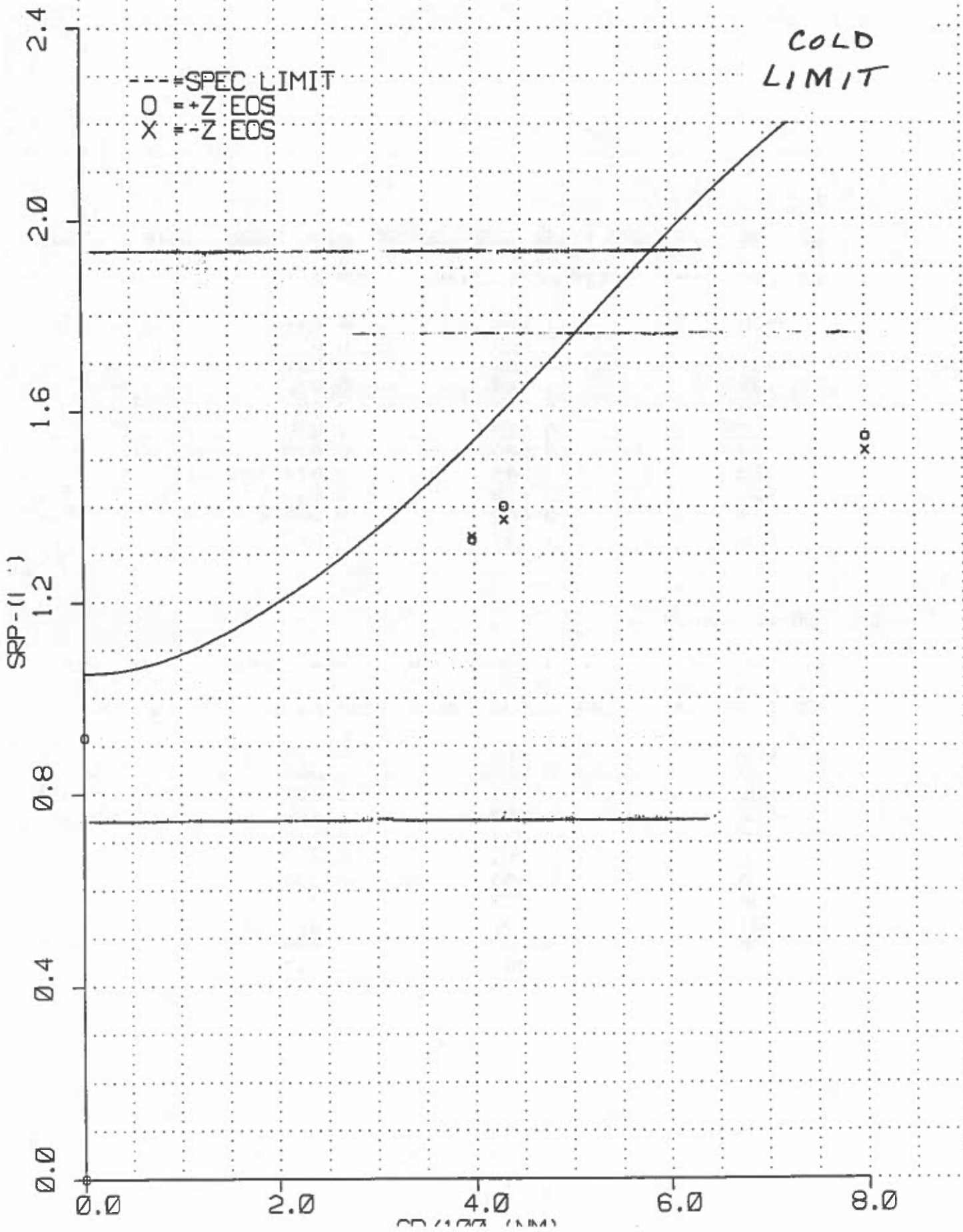
LS, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 320

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-800.	1.513	0.644
0.	0.000	0.000
-431.	1.373	0.857
-398.	1.333	0.869
0.	0.000	0.000
0.	0.916	0.873
0.	0.000	0.000
398.	1.330	0.867
431.	1.401	0.875
0.	0.000	0.000
800.	1.544	0.657

SYSTEM 14., SRP. LS. DAY/FS ~~1963~~

SSS=3., M1=-8., DATE:324



LS, DAY, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 324

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	1.518	0.646
0.	0.000	0.000
-431.	1.373	0.858
-398.	1.337	0.871
0.	0.000	0.000
0.	0.916	0.873
0.	0.000	0.000
398.	1.331	0.868
431.	1.401	0.875
0.	0.000	0.000
800.	1.547	0.658

LS, DAY, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 324

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-800.	1.510	0.643
0.	0.000	0.000
-431.	1.366	0.853
-398.	1.330	0.867
0.	0.000	0.000
0.	0.912	0.869
0.	0.000	0.000
398.	1.324	0.863
431.	1.393	0.870
0.	0.000	0.000
800.	1.538	0.654

2.2 Geometric Resolution (Cont'd)

2.2.5 Smoothed Geometric Resolution - Nighttime Visual (3.2.1.1.2.2)

LS Night SRP routinely is deliberately adjusted to be close to spec limit during system integration in order to optimize PMT signal-to-noise ratio, at the expense of SRP margin.

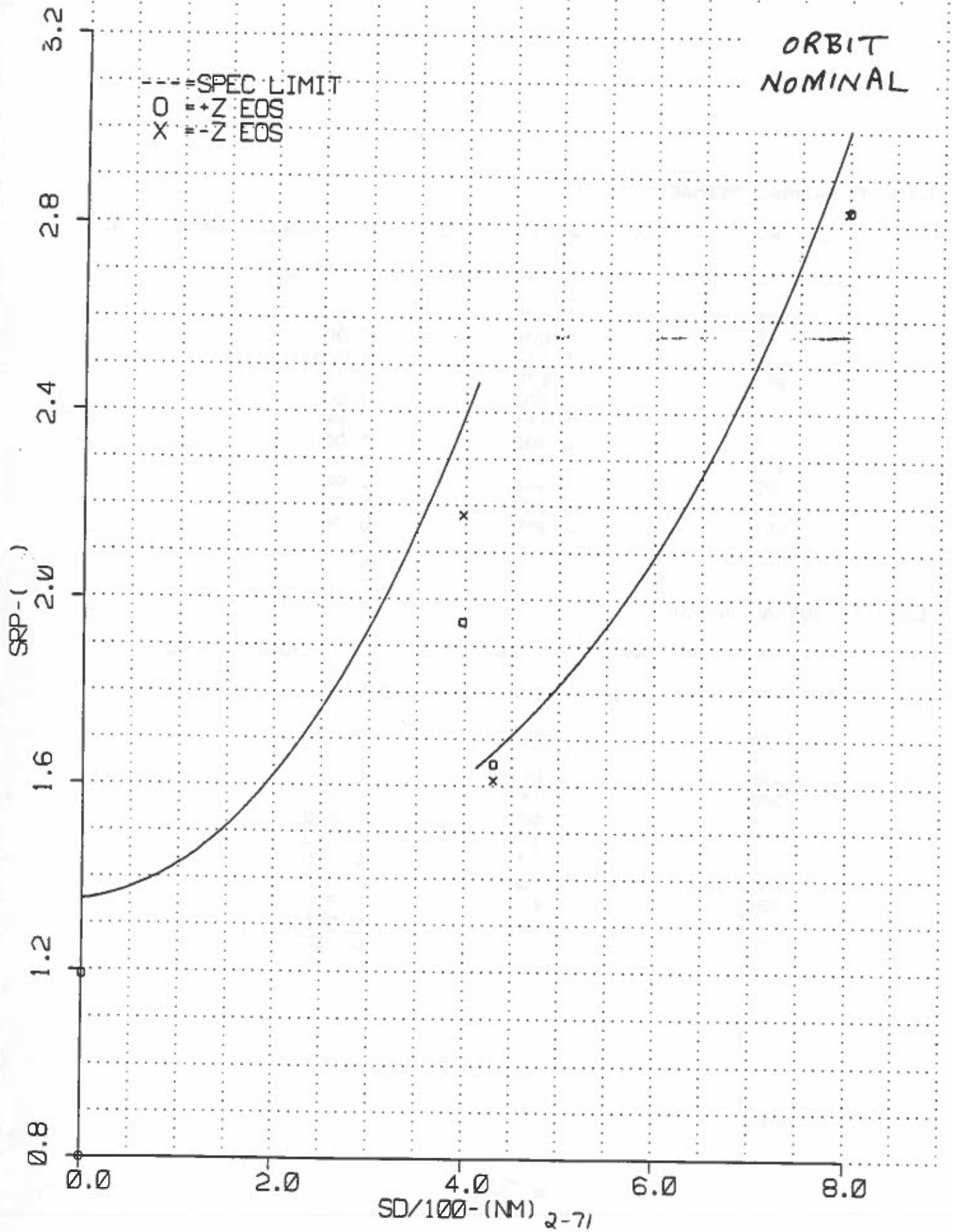
2.2.5.1 Baseline (Orbit Nominal)

The LS Night SRP is within spec for all measured scan angles

ATTACHMENTS: LS Night SRP Curve - Orbit Nominal

LS Night SRP Table - Orbit Nominal

SYSTEM 14, SRP LS NITE, SSS=5, M1=-8, DATE:401



LS, NITE, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 5DEGC M1= -8DEGC DATE: 401

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-799.	2.827	0.944
0.	0.000	0.000
-430.	1.612	0.965
-397.	2.179	0.915
0.	0.000	0.000
0.	1.191	0.882
0.	0.000	0.000
397.	1.951	0.819
430.	1.646	0.985
0.	0.000	0.000
801.	2.830	0.941

LS, NITE, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 5DEGC M1= -8DEGC DATE: 401

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-799.	2.800	0.935
0.	0.000	0.000
-430.	1.599	0.957
-397.	2.156	0.905
0.	0.000	0.000
0.	1.179	0.873
0.	0.000	0.000
397.	1.930	0.810
430.	1.631	0.976
0.	0.000	0.000
801.	2.804	0.932

2.2 Geometric Resolution (Cont'd)

2.2.5 Smoothed Geometric Resolution - Nighttime

Visual (Cont'd) (3.1.2.2)

2.2.5.2 Acceptance - Vibration

The LS Night SRP is within specification before and after SSS vibration. No vibration-induced changes in SRP were observed.

ATTACHMENTS: LS Night SRP Curve Pre-Vibration
 LS Night SRP Tables Pre-Vibration
 LS Night SRP Curve Post-Vibration
 LS Night SRP Tables Post-Vibration

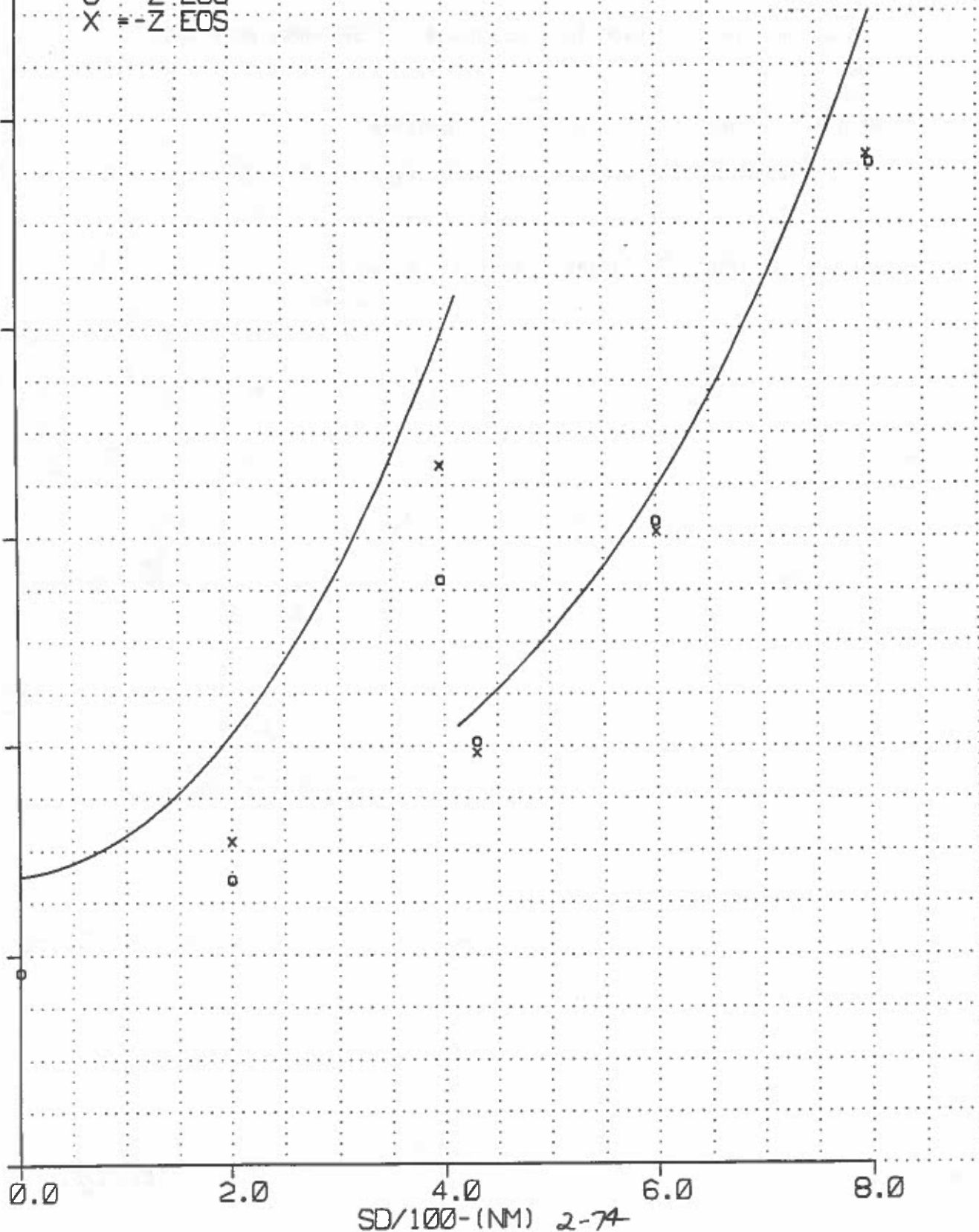
SYSTEM 14, SRP LS NITE, SSS=23, M1=23, DATE: 1125

PRE
VIB

--- = SPEC LIMIT

O = +Z EOS

X = -Z EOS



SD/100-(NM) 2-74

LS, NITE, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1125

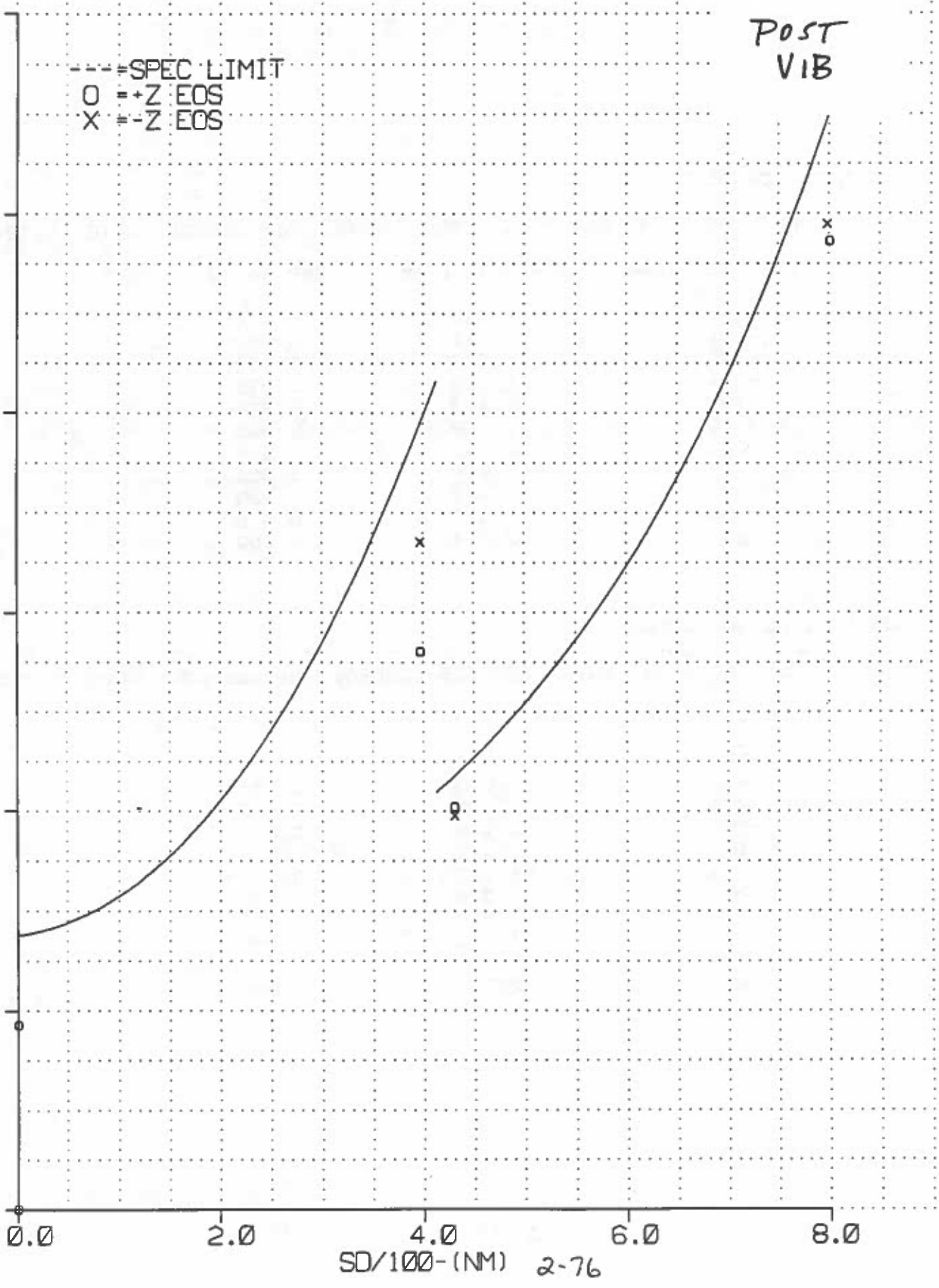
SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-799.	2.730	0.912
-601.	2.011	0.959
-430.	1.588	0.950
-397.	2.138	0.898
-200.	1.417	0.874
0.	1.168	0.865
200.	1.343	0.829
397.	1.919	0.806
430.	1.609	0.963
600.	2.030	0.970
801.	2.714	0.902

LS, NITE, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 1125

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-799.	2.720	0.908
-601.	2.003	0.955
-430.	1.582	0.947
-397.	2.129	0.894
-200.	1.411	0.870
0.	1.163	0.861
200.	1.338	0.825
397.	1.910	0.802
430.	1.603	0.959
600.	2.022	0.966
801.	2.704	0.899

SYSTEM 14, SRP LS NITE, SSS=23, M1=23, DATE: 105



LS, NITE, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 105

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-799.	2.779	0.928
0.	0.000	0.000
-430.	1.591	0.952
-397.	2.142	0.899
0.	0.000	0.000
0.	1.171	0.867
0.	0.000	0.000
397.	1.921	0.807
430.	1.608	0.962
0.	0.000	0.000
801.	2.745	0.913

LS, NITE, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 2 SSS= 23DEGC M1= 23DEGC DATE: 105

SUR. DIST. (NM)	SRP ACTUAL (NM)	SRP RATIO
-799.	2.774	0.927
0.	0.000	0.000
-430.	1.586	0.949
-397.	2.135	0.896
0.	0.000	0.000
0.	1.167	0.864
0.	0.000	0.000
397.	1.914	0.803
430.	1.603	0.959
0.	0.000	0.000
801.	2.739	0.911

2.2 Geometric Resolution (Cont'd)

2.2.5 Smoothed Geometric Resolution - Nighttime

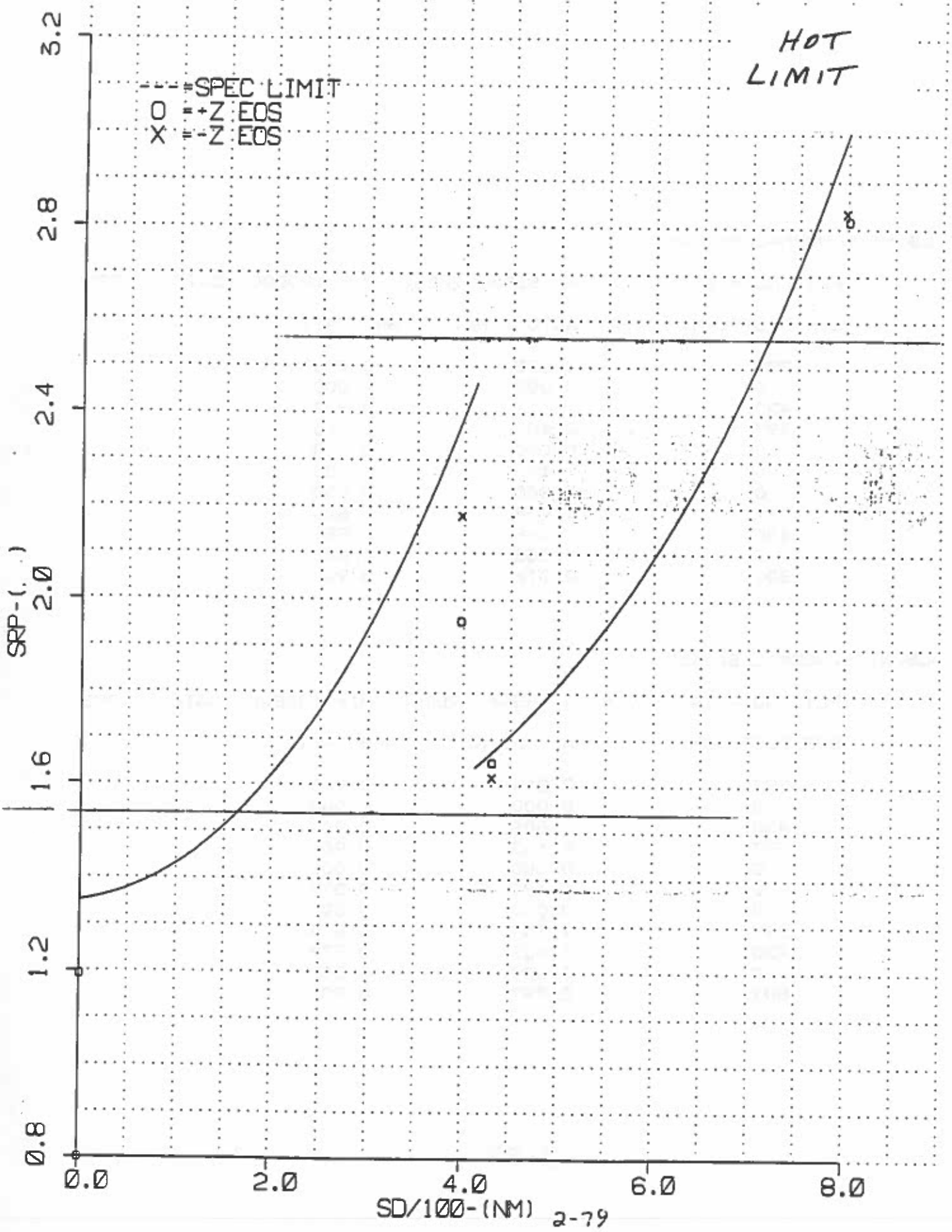
Visual (Cont'd) (3.1.2.2)

2.2.5.3 Acceptance - Thermal Vacuum

The LS Night SRP is within specification limits over the Acceptance temperature range. No temperature-related changes in SRP over the Acceptance temperature range were observed.

ATTACHMENTS: LS Night SRP Curve Hot Limits
LS Night SRP Tables Hot Limits
LS Night SRP Curve Cold Limits
LS Night SRP Tables Cold Limits

SYSTEM 14, SRP LS NITE, SSS=7, M1=12, DATE:393



LS, NITE, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 393

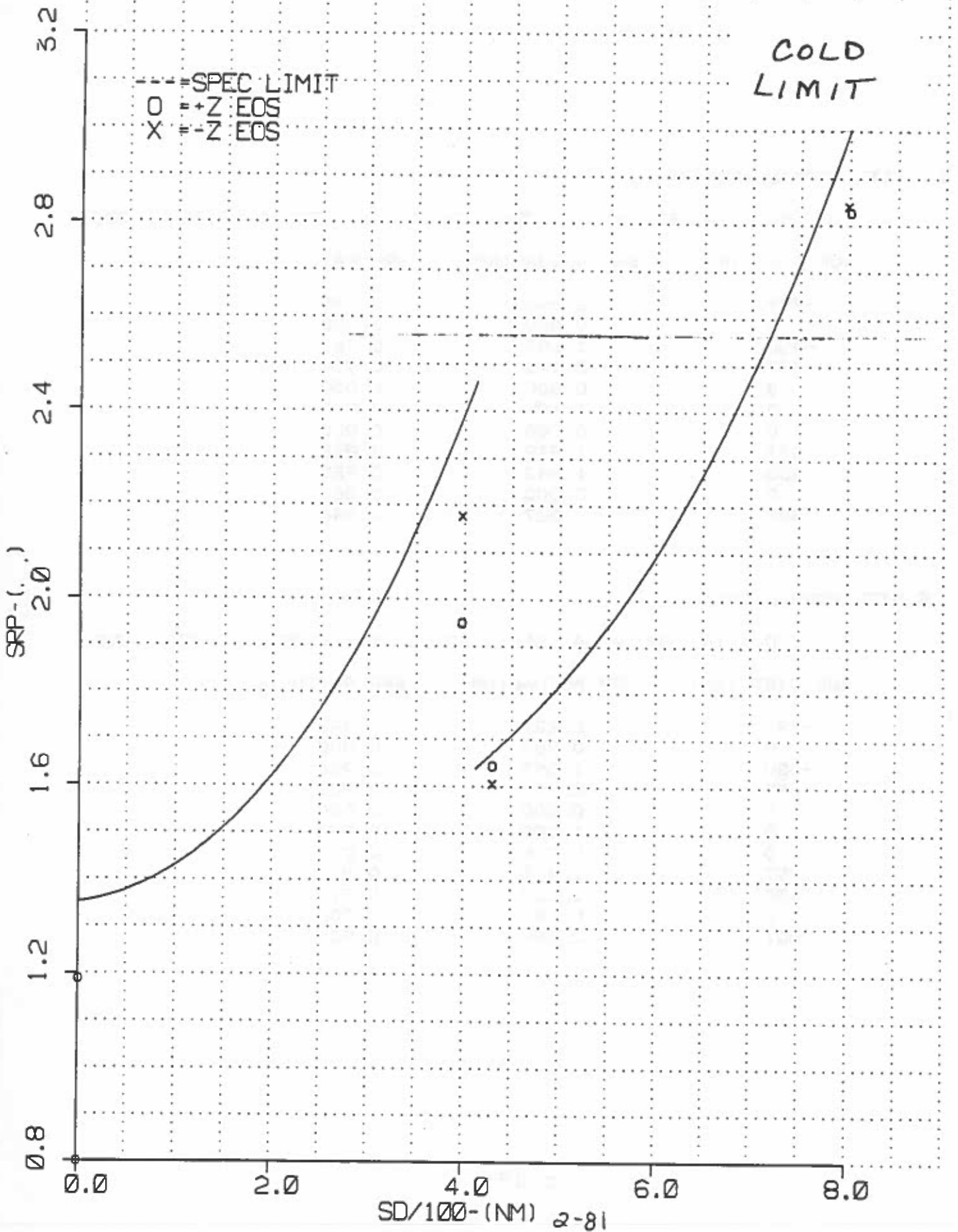
SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-799.	2.833	0.946
0.	0.000	0.000
-430.	1.616	0.967
-397.	2.180	0.915
0.	0.000	0.000
0.	1.193	0.883
0.	0.000	0.000
397.	1.954	0.820
430.	1.648	0.986
0.	0.000	0.000
801.	2.816	0.936

LS, NITE, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 7DEGC M1= 12DEGC DATE: 393

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-799.	2.814	0.940
0.	0.000	0.000
-430.	1.606	0.961
-397.	2.163	0.908
0.	0.000	0.000
0.	1.184	0.877
0.	0.000	0.000
397.	1.939	0.814
430.	1.637	0.979
0.	0.000	0.000
801.	2.797	0.930

SYSTEM 14 , SRP LS NITE , SSS=3 , M1 = -8 , DATE: 325



LS, NITE, NORMAL, PRIMARY

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 325

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-799.	2.840	0.949
0.	0.000	0.000
-430.	1.605	0.961
-397.	2.176	0.914
0.	0.000	0.000
0.	1.188	0.880
0.	0.000	0.000
397.	1.949	0.818
430.	1.643	0.983
0.	0.000	0.000
801.	2.827	0.940

LS, NITE, NORMAL, BACKUP

FLT. NO. = 14 ENV. = 4 SSS= 3DEGC M1= -8DEGC DATE: 325

SUR. DIST. (NM)	SRP ACTUAL(NM)	SRP RATIO
-799.	2.821	0.942
0.	0.000	0.000
-430.	1.597	0.956
-397.	2.161	0.907
0.	0.000	0.000
0.	1.180	0.874
0.	0.000	0.000
397.	1.935	0.812
430.	1.633	0.977
0.	0.000	0.000
801.	2.808	0.934

2.2 Geometric Resolution (Cont'd)

2.2.6 Data Sampling (3.2.1.1.2.3)

The sampling frequency ratios for all modes of the 5D-3 OLS satisfy the specification requirements. The calculations are contained in the 5D-3 OLS System Summary Report. The results are summarized below. The worst-case sampling frequency ratio for each mode is given.

<u>MODE</u>	<u>SAMPLING FREQ. RATIO (Spec: > 2.4)</u>
LF Day - Normal	2.58
LF Day - Fallback	2.61
LS & TS Day - Normal	2.50
LS Night - Normal	2.66
TF - Normal	3.44
TF Fallback - Normal Side of scan	3.28
TF Fallback - Abnormal Side of Scan	3.28

2.3 Geometric Accuracy (3.2.1.1.3.1, 3.2.1.1.3.2, 3.2.1.1.3.3)

The alignment and synchronization of the SSS determine the Geometric Accuracy. Geometric accuracy is specified in 3 categories (Repeatability, Stability & Fixed Errors). Within these categories, accuracy is further specified for Along Track, Along Scan (Stored Data), Along Scan (Direct Data), and Along Scan (with digital delphi generation, i.e., the Encoder Simulator locked mode).

There was an observed shift in alignment between the extremes of M1 temperature of -8°C and $+12^{\circ}\text{C}$ on the order of 0.2 milliradians or less for all channels. There was also an observed shift in synchronization in all modes in OLS #14 of approximately 0.1 milliradians or less between M1 temperature extremes of -8°C and $+12^{\circ}$.

The Repeatability error is calculated using the rms difference of the measured alignment/synchronization (as a function of the variation of M1 temperature on orbit) from the hot-to-cold average values of alignment/ synchronization. The total Repeatability contribution for synchronization is determined by rssi'ng the wow-flutter error (as measured in test 6x11.ST) and the jitter error (which was negligible on OLS #14) with the repeatability shift between the Hot and Cold Limits.

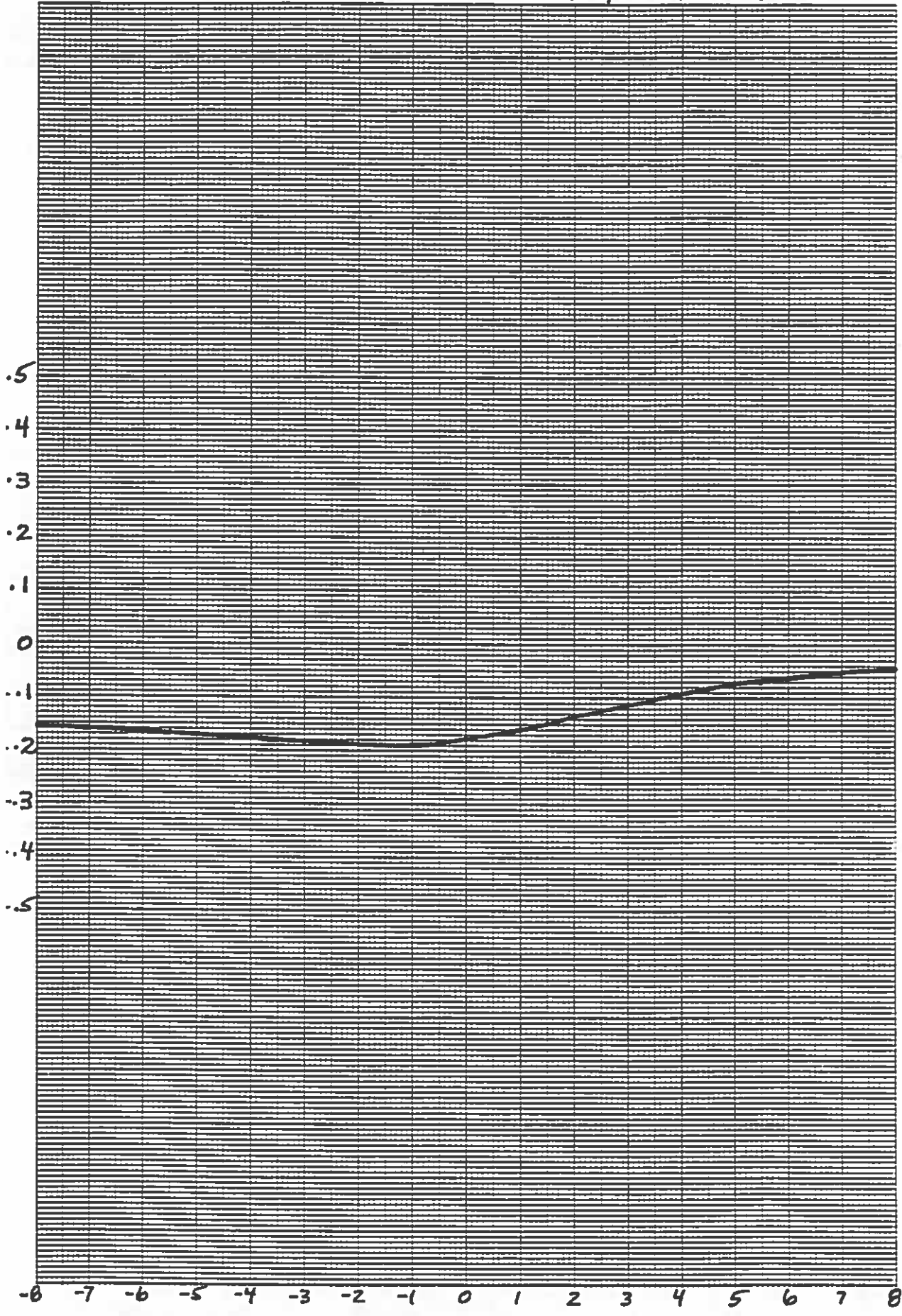
The Primary Alignment Reference Axes (REFPLN) are calculated in a computer program using HRD & T channel Alignment and Sync data from several Orbit Nominal tests. The data determine a best-fit alignment with respect to the Interface Mounting Axes. The REFPLN Alignment and REFPLN Synchronization curves are included here. The remainder of the Alignment and Synchronization data for the OLS #14 SSS are in BVS 2733, Vol. III of this Acceptance Test Report.

The 5D-3 System has a Fallback mode utilizing an encoder control track and Encoder Simulator. The spec limits and measured results are shown in Table 2.3-2.

Note that stability errors for the T-channel are not presented. This is because in bearing retrofit testing no thermal vacuum adjustment testing is performed. Therefore, no pre-vibration data for the T-channel exists. However, note that all T-channel data taken is within spec.

ATTACHMENTS: OLS #14 REFPLN ALIGNMENT
OLS #14 REFPLN SYNCHRONIZATION
OLS #14 ALIGN/SYNC vs SPEC, Table 2.3-1
OLS #14 Encoder Simulator Sync, Table 2.3-2

OLS #14A REFPLN ALIGNMENT



SURFACE DISTANCE, NM/100

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OLS #14A REFPLN SYNCHRONIZATION

ANGULAR ERROR FROM INTERFACE, MRAD

.5
.4
.3
.2
.1
0
-.1
-.2
-.3
-.4
-.5

-8 -7 -6 -5 -4 -3 -2 -1 0 1 2 3 4 5 6 7 8

SURFACE DISTANCE, NM/100

2-87

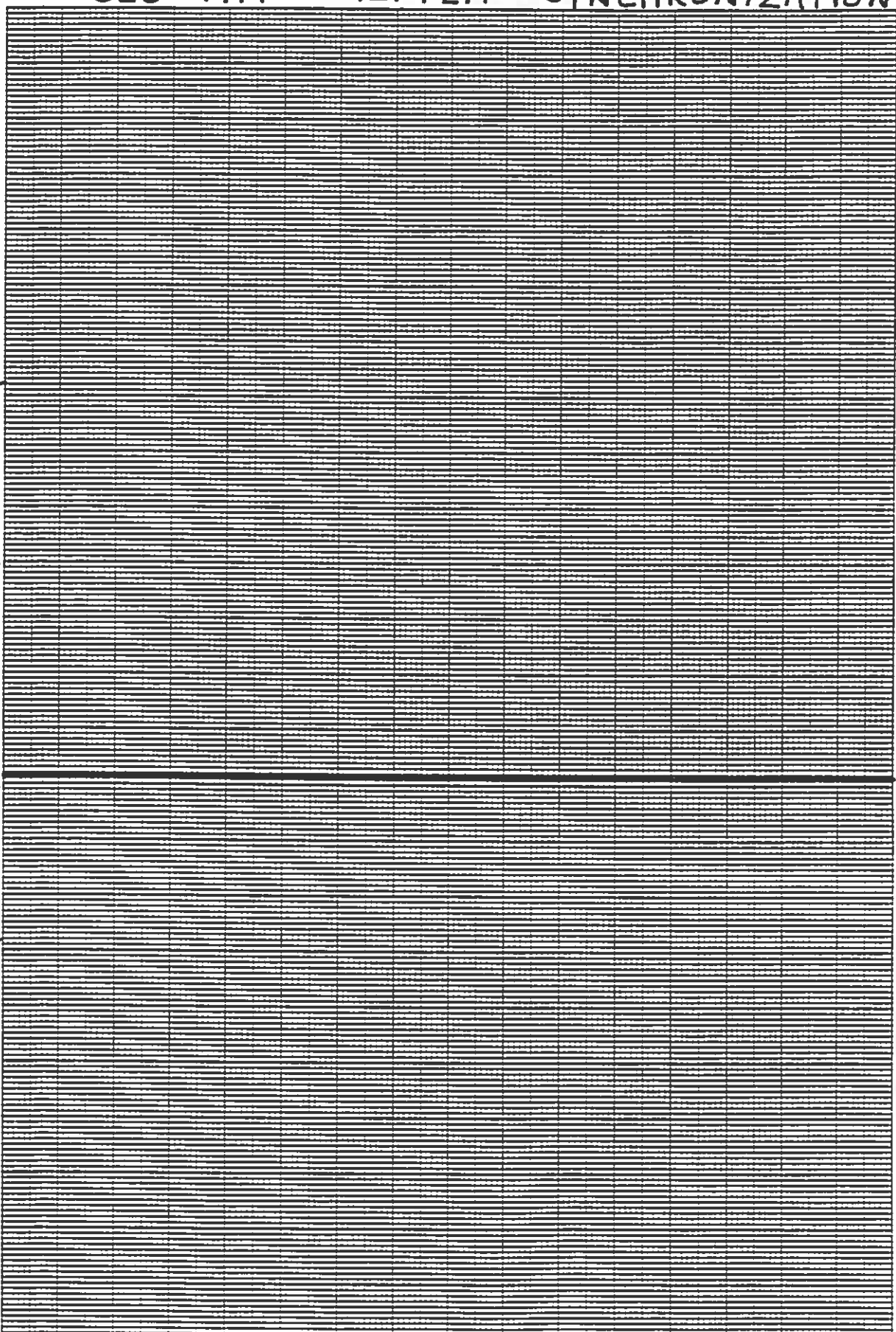


Table 2.3-1

OLS #14 ALIGN/SYNC vs. SPEC
all numbers in milliradians

	<u>HRD</u>	<u>T</u>	<u>PMT</u>
<u>FIXED - Delta between "REFPLN"</u>			
<u>& Optic Hot - Cold Average</u>			
AT SPEC	0.45	0.70	0.60
Measured (worst-case)	0.13	0.19	0.43
AS STORED SPEC	0.80	0.80	1.90
Measured (worst-case)	0.33	0.23	0.64
AS DIRECT FINE SPEC	0.80	0.80	1.90
Measured (worst-case)	0.31	0.25	N/A
AS DIRECT SMOOTH SPEC	0.80	0.80	1.90
Measured (worst-case)	0.37	0.23	0.88
<u>STABILITY - Delta Between Pre & Post - Vibration</u>			
AT SPEC	0.50	0.55	0.55
Measured (worst-case)	0.04	N/A	0.08
AS STORED SPEC	0.20	0.25	0.25
Measured (worst-case)	0.04	N/A	0.04
AS DIRECT FINE SPEC	0.20	0.25	0.25
Measured (worst-case)	0.04*	N/A	N/A
AS DIRECT SMOOTH SPEC	0.20	0.25	0.25
Measured (worst-case)	0.04*	N/A	0.04
<u>REPEATABILITY - Delta between TV Hot & Cold Limits</u>			
AT SPEC	0.20	0.22	0.20
Measured (rms)	0.07	0.05	0.07
AS STORED SPEC	0.30	0.30	0.30
Measured (rms)	0.10	0.07	0.11
AS DIRECT FINE SPEC	0.50	0.50	0.50
Measured (rms)	0.09	0.07*	N/A
AS DIRECT SMOOTH SPEC	2.00	2.00	2.00
Measured (rms)	0.10	0.07*	0.09
<u>TOTAL -</u>			
AT SPEC	1.00	1.30	1.20
Calculated	0.21	N/A	0.54
AS STORED SPEC	1.16	1.19	2.29
Calculated	0.44	N/A	0.76
AS DIRECT FINE SPEC	1.34	1.36	2.46
Calculated	0.41	N/A	N/A
AS DIRECT SMOOTH SPEC	2.81	2.82	3.92
Calculated	0.48	N/A	0.98

N/A = not applicable

* = Inferred from AS Stored number.

Table 2.3-2
 ALONG-SCAN GEOMETRIC ACCURACY WITH ENCODER SIMULATOR

	<u>Stored</u>	<u>Direct Fine</u>	<u>Direct Smooth</u>
Repeatability-Spec, mrad	1.0	1.1	2.2
Measured	0.10	0.08	0.10
Stability - Spec, mrad	0.50	0.50	0.50
Measured	0.08	0.08*	0.08*
Fixed - Spec, mrad	10.0	10.0	10.0
Measured	0.94	0.94*	0.94*
Total - Spec, mrad	11.1	11.2	12.3
Calculated	1.07	1.05	1.07

*Inferred from stored number

2.4 RADIOMETRIC ACCURACY

2.4.1 T Channel Radiometric Accuracy (3.2.1.1.4.1 a,b,c)

The overall one sigma accuracy of the OLS #14 T Channel DC response is 0.71°K compared to a 1.1°K spec and therefore OLS #14 does meet this specification requirement.

Table 2.4.1-1 presents the overall summary of performance, which is discussed more fully in sections 2.4.1.1, 2.4.1.2 and 2.4.1.3.

ATTACHMENTS:

Table 2.4.1-1	Overall Contributors
Table 2.4.1-2	210° to 310°K Best Straight Line Calibrations
Table 2.4.1-2A	210° to 310°K Best Straight Line Calibrations with Gain Compensation
Table 2.4.1-3	T DC Response Compilation of Test Runs
Table 2.4.1-3A	T DC Response Compilation of Test Runs with Gain Compensation
Table 2.4.1-4	BSL Equation T Right, Run #11
Table 2.4.1-5	BSL Equation T Mid, Run #11
Table 2.4.1-6	BSL Equation T Left, Run #11
Figure 2.4.1-1	T DC Response Plots, Run #1 - Primary
Figure 2.4.1-2	T DC Response Plots, Run #1 - Redundant
Figure 2.4.1-3	T DC Response Plots, Run #2 - Primary
Figure 2.4.1-4	T DC Response Plots, Run #2 - Redundant
Figure 2.4.1-5	T DC Response Plots, Run #9 - Primary
Figure 2.4.1-6	T DC Response Plots, Run #9 - Redundant
Figure 2.4.1-7	T DC Response Plots, Run #10- Primary
Figure 2.4.1-8	T DC Response Plots, Run #10- Redundant
Figure 2.4.1-9	T DC Response Plots, Run #11- Primary
Figure 2.4.1-10	T DC Response Plots, Run #11- Redundant

TABLE 2.4.1-1

OLS #14

OVERALL CONTRIBUTORS TO T-CHANNEL RADIOMETRIC ACCURACY

<u>SPECIFICATION PARA. 3.1.4.1</u>	<u>RMS DEVIATION (\cdotK)</u>	<u>SPECIFICATION MAX ONE SIGMA ERROR (\cdotK)</u>
a) Repeatability (<1 day)	0.24	0.42
b) Stability (>1 day)	0.61	0.80
c) Fixed Deviations	0.27	0.60
	<hr/>	<hr/>
TOTAL (RSS) ACCURACY	0.71	1.10

Discussion of T DC Response Test and Overview

The measurement of T DC Response is accomplished in the thermal vacuum chamber because the T detector must be cooled to its operating temperature of near 110°K. Two controlled blackbody targets variable in temperature over the dynamic range of 190°K to 310°K provide the absolute infrared radiance reference. The temperature of each target is measured by five thermocouples which have been calibrated against a precision platinum resistance temperature standard. The five thermocouples provide target temperature gradient information to indicate target stability as well as the capability to average the five for the reference target temperature. T Channel DC Response consists of comparing the average target temperature to the channel output voltage at the input to the A/D converters for several target temperatures.

Eleven vacuum runs were made on OLS #14. The T DC Response data from vacuum runs (1 through 11) is compiled in Table 2.4.1-2 and -3, which show the equipment temperature environments and characteristics of each run. The column headed "Data Points" indicates how many target temperatures were in that run. The characteristics of the response itself are indicated in three columns each for T LFT, T MID and T RGT. The compared channel response to target temperature results in a difference for each data point. This difference is corrected for MI Temperature so that all data for a given run reflect the same MI temperature and the expected shaper circuit difference is subtracted. In this form the difference data for a given run should ideally be a linear function to target temperature. A linear least- squares fit to the corrected data is used to determine the equation of the best straight line (BSL). In Table 2.4.1-2 the slope error, the 210°K ordinate and the RMS data fit values for these different BSL's are listed in the columns headed Slope, Ordinate at 210°K, and RMS Dev.

In order to distinguish between gain and bias type effects caused by environment, the 310°K value that results when the 210°K BSL value is forced to zero difference is also calculated. Table 2.4.1-3 shows the pre-shaper % gain Difference from Nominal, the Bias Diff. from Nominal (at 190°K) where the pre-shaper Gain is forced to nominal, and the M1 temperature coefficient (K factor); in three columns each for TRGT, TMID and TLEFT.

Tables 2.4.1-4, -5 and -6 show the STS computer processed T DC Response Data of the final "Orbit Nominal" Run (Run #11) for TRGT, TMID, and TLEFT respectively. The fourth line down in the body of the Best Straight Line Equation, "RMS Deviation", is the RMS error (for 210°K to 310°K) of the data points fitted to the best straight line. "FP" is T Fine Primary; "FB" is T Fine, Backup (Redundant); "SP" is T Smoothed, Primary; "SB" is T Smoothed, Backup (redundant). The SP and SB are not used for T Right Only or for T Left Only in the Primary or Redundant normal modes; these are utilized only in the Fallback (slightly degraded) modes of operation. SP and SB are applicable to T Mid in normal Primary or Redundant modes.

Runs 1 and 2 together indicate the changes which accompany operation over the foreoptics cold-to-warm temperature range as indicated by M1 temperature.

Runs 3 and 4 or runs 5 and 6 together indicate the magnitude of the variation over the extremes of SSS temperature, (+11° to -3°C); when compared to the +4°C and +5°C SSS run pairs with the corresponding M1 temperatures, (Runs 2 and 1, respectively). However, changes between these runs are not only due to SSS temperature differences. The PSU, which contains the shaper networks was varied in temperature along with the SSS, from a low of -0.4°C to a high of +38.2°C.

Figures 2.4.1-1 through 2.4.1-10 inclusive show, for Runs No. 1 and 2, and runs No. 9 through 11, (respectively), the test data points for Targets 1 and 2 and the BSL plots for Right, Mid, and Left. (No BSL data plot was obtained for Runs 3 through 8).

The OLS #14 average M1 coefficient (coupling factor) measured for the final run (#11) was 0.185°K at 210°K scene per 1°C temperature change of M1. The lower the M1 coefficient value, the better the performance. The T Clamp leakage was +0.150% T LEFT and +0.083% Tu RIGHT.

The Orbit Nominal BSL differences (from Table 2.4.1-4,5 and 6) between Fine Primary and Fine Backup are small, the largest being 0.43°K for T Left, at the 310°K end. In the Smooth Primary and Backup modes, T Left differs by 0.39°K (at 310°K).

The difference between T LEFT and T RIGHT segments calibration (from Tables 2.4.1.3-4 and 5) is 0.63°K worst-case, vs. a spec limit of 1.0°K.

At the completion of testing, the T channel gain pots were readjusted to set $T_{RGT} = T_{LFT} = 4$ and to eliminate the channel gain imbalance.

TABLE 2.4.1-2
 OLS #14
 210° TO 310°K BEST STRAIGHT LINE CALCULATIONS

DATE	RUN#	R/L TG	# OF DATA POINTS	TEMPERATURE °C			T RIGHT			T MID			T LEFT			COMMENTS
				SSS	HI	PSU	SLOPE	ORD. @ 210°	RMS DEV	SLOPE	ORD. @ 210°	RMS DEV	SLOPE	ORD. @ 210°	RMS DEV	
2/14/93	1	3/3	14	5.1	-8.0	26.2	.0006	0.00	0.05	.0039	-0.05	0.06	.0104	-0.63	0.08	Cold Optic Limit
2/20/93	2	3/3	11	4.1	12.1	25.8	-.0059	0.50	0.05	-.0052	0.59	0.05	.0011	0.12	0.03	Hot Optic Limit
2/23/93	3	3/3	11	11.2	15.1	37.7	-.0080	0.59	0.00	-.0067	0.52	0.00	-.0005	-0.14	0.00	Hot Soak #1
2/25/93	4	3/3	14	-2.4	-10.5	-0.4	.0021	0.52	0.00	.0043	0.58	0.00	.0112	-0.01	0.00	Cold Soak #1
2/28/93	5	3/3	11	11.0	15.6	38.2	-.0098	0.63	0.00	-.0079	0.55	0.00	-.0028	0.03	0.00	Hot Soak #2
3/02/93	6	3/3	14	-3.3	-11.1	0.2	-.0002	0.64	0.00	.0030	0.65	0.00	.0094	0.07	0.00	Cold Soak #2
VACUUM BREAK																
3/16/93	7	3/3	13	5.0	-6.9	23.8	.0010	0.05	0.00	.0033	0.03	0.00	.0103	-0.64	0.00	Cold Optic Limit
3/17/93	8	3/3	10	3.7	12.4	23.6	-.0088	0.81	0.00	-.0094	0.99	0.00	-.0029	0.43	0.00	Hot Optic Limit
3/18/93	9	3/3	10	5.9	12.2	33.3	-.0079	0.64	0.11	-.0072	0.73	0.09	-.0018	0.27	0.04	Hot Limit
3/24/93	10	3/3	13	2.2	-8.0	4.5	.0017	0.37	0.04	.0034	0.40	0.02	.0094	-0.12	0.08	Cold Limit
3/31/93	11	3/3	13	4.8	-7.8	23.7	-.0020	0.24	0.03	-.0001	0.24	0.04	.0063	-0.35	0.05	Nominal

T DC RESPONSE COMPILATION OF TEST RUNS W/O GAIN COMPENSATION

DATE	RUN #	R/L TG	TL	# OF DATA POINTS	TEMPERATURE °C			T RIGHT			T MID			T LEFT			COMMENTS
					SSS	M1	PSU	% GAIN DIFF. FROM NOM.	BIAS DIFF. FROM NOM.	K FACTOR	% GAIN DIFF. FROM NOM.	BIAS DIFF. FROM NOM.	K FACTOR	% GAIN DIFF. FROM NOM.	BIAS DIFF. FROM NOM.	K FACTOR	
2/14/93	1	3/3	14	14	5.1	-8.0	26.2	0.08	0.02	.176	0.52	0.08	.183	0.93	-0.65	.190	Cold Optic Limit
2/20/93	2	3/3	11	6	4.1	12.1	25.8	-0.40	0.61	.176	-0.20	0.82	.183	0.28	0.27	.190	Hot Optic Limit
2/23/93	3	3/3	11	2	11.2	15.1	37.7	-0.62	0.64	.176	-0.49	0.59	.183	-0.21	-0.29	.190	Hot Soak #1
2/25/93	4	3/3	14	2	-2.4	-10.5	-0.4	0.82	1.00	.176	1.21	1.18	.183	1.66	0.43	.190	Cold Soak #1
2/28/93	5	3/3	11	2	11.0	15.6	38.2	-0.86	0.62	.176	-0.65	0.58	.183	-0.39	-0.10	.190	Hot Soak #2
3/02/93	6	3/3	14	2	-3.3	-11.1	0.2	0.59	1.11	.176	1.08	1.25	.183	1.47	0.49	.190	Cold Soak #2
VACUUM BREAK																	
3/16/93	7	3/3	13	2	5.0	-6.9	23.8	0.19	0.13	.177	0.51	0.20	.187	0.91	-0.66	.192	Cold Optic Limit
3/17/93	8	3/3	10	2	3.7	12.4	23.6	-0.52	1.05	.177	-0.44	1.34	.187	-0.01	0.64	.192	Hot Optic Limit
3/18/93	9	3/3	10	7	5.9	12.2	33.3	-0.55	0.77	.177	-0.35	0.97	.187	-0.01	0.40	.192	Hot Limit
3/24/93	10	3/3	13	7	2.2	-8.0	4.5	0.61	0.73	.177	0.90	0.87	.187	1.28	0.20	.192	Cold Limit
3/31/93	11	3/3	13	18	4.8	-7.8	23.7	-0.06	0.34	.177	0.21	0.42	.187	0.59	-0.35	.192	Nominal

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JTS6.rc0

BVS 2731

TABLE 2.4.1-4

OLS NUMBER 14

T RGT DATA OF 03/29/93

SSS AT 4.8°C

M1 AT -7.8°C

PSU TEMP = 23.7°C

M1 Coefficient = 0.177 K/C

T GAIN = 3

T LEVEL = 13

V2 <T Clamp> = 1.64497

K9 <TL Step Size> = .9237

BEST STRAIGHT LINE EQUATION

	FP	(Δ)	FB	SP	(Δ)	SB
BSL SLOPE	-0.0020	-	-0.0040	-0.0019	-	-0.0046
BSL AT 190K<K>	0.28	(.14)	0.14	0.29	(.09)	0.20
BSL AT 210K<K>	0.24	(.18)	0.06	0.25	(.14)	0.11
BSL AT 310K<K>	0.04	(.38)	-0.34	0.06	(.41)	-0.35
RMS DEVIATION<K>	0.03	-	0.03	0.04	-	0.05
BSL AT 310K; 190 AT OV<K>	-0.05	-	-0.39	-0.03	-	-0.42
% CHANGE FROM NOM GAIN	-0.06	-	-0.54	-0.04	-	-0.58
BIAS DIFF FROM NORMAL 190K<K>	0.34	-	-0.08	0.36	-	-0.02

TABLE 2.4.1-5

; NUMBER 14
 T MID DATA OF 03/29/93
 SSS AT 4.8°C
 M1 AT -7.9°C
 PSU TEMP = 23.7°C
 M1 Coefficient = 0.187 K/C
 T GAIN = 0
 T LEVEL = 13
 V2 <T Clamp> = 1.62435
 K9 <TL Step Size> = 0.9237

BEST STRAIGHT LINE EQUATION

	FP	(Δ)	FB	SP	(Δ)	SB
. SLOPE	-0.0001	-	-0.0030	-0.0005	-	-0.0032
. AT 190K<K>	0.24	(.07)	0.17	0.30	(.08)	0.22
. AT 210K<K>	0.24	(.13)	0.11	0.30	(.14)	0.16
. AT 310K<K>	0.22	(.41)	-0.19	0.25	(.41)	-0.16
; DEVIATION<K>	0.04	-	0.05	0.05	-	0.05
. AT 310K;						
190 AT OV<K>	0.15	-	-0.24	0.16	-	-0.23
CHANGE FROM						
NOM GAIN	0.21	-	-0.33	0.22	-	-0.32
AS DIFF FROM						
NORMAL 190K<K>	0.42	-	0.07	0.51	-	0.14

TABLE 2.4.1-6

OLS NUMBER 14

T LFT DATA OF 03/29/93

SSS AT 4.8°C

M1 AT -8.0°C

PSU TEMP = 23.7°C

M1 Coefficient = 0.192 K/C

T GAIN = 3

T LEVEL = 13

V2 <T Clamp> = 1.61241

K9 <TL Step Size> = .9237

BEST STRAIGHT LINE EQUATION

	FP	(Δ)	FB	SP	(Δ)	SB
BSL SLOPE	0.0063	-	0.0025	0.0055	-	0.0027
BSL AT 190K<K>	-0.48	(.02)	-0.46	-0.41	(.04)	-0.45
BSL AT 210K<K>	-0.35	(.06)	-0.41	-0.30	(.10)	-0.40
BSL AT 310K<K>	0.27	(.43)	-0.16	0.26	(.39)	-0.13
RMS DEVIATION<K>	0.05	-	0.04	0.04	-	0.02
BSL AT 310K; 190 AT OV<K>	0.43	-	-0.02	0.39	-	0.01
% CHANGE FROM NOM GAIN	0.59	-	-0.02	0.54	-	0.02
BIAS DIFF FROM NORMAL 190K<K>	-0.35	-	-0.62	-0.28	-	-0.58

NEW K FACTORS
CORRECTED TO T_{ENV} = 14

OLS# 14 T DC RESPONSE: DEV. FROM NOM PSU=26 SSS=5 MI=-8 IG=5 IL=20 LHIL:2 /14/45

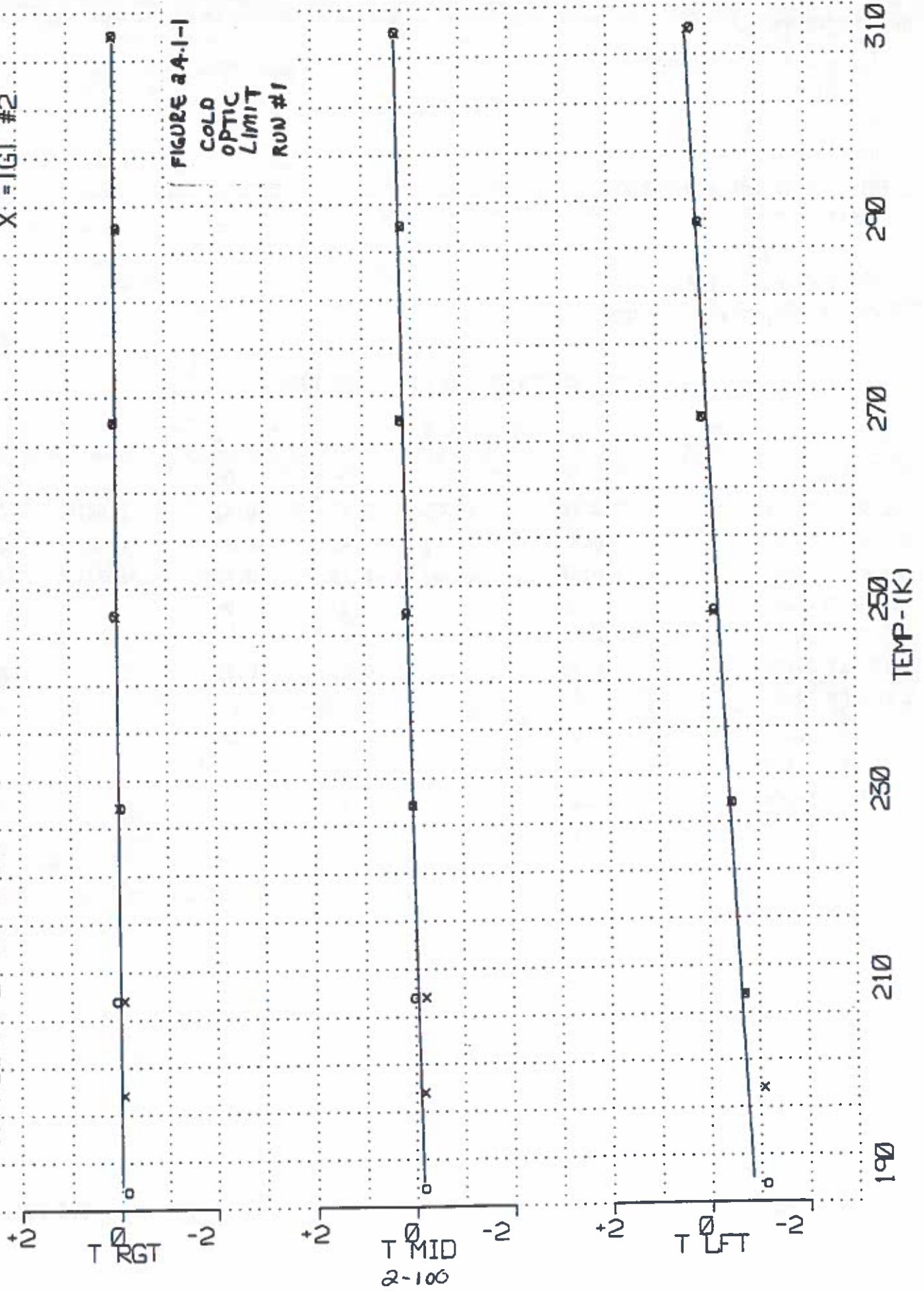
O = TGT #1
X = TGT #2

PS1 - PRIMARY PATH

MAX DELTA LT-RT: .6277

FIGURE 2A-1-1

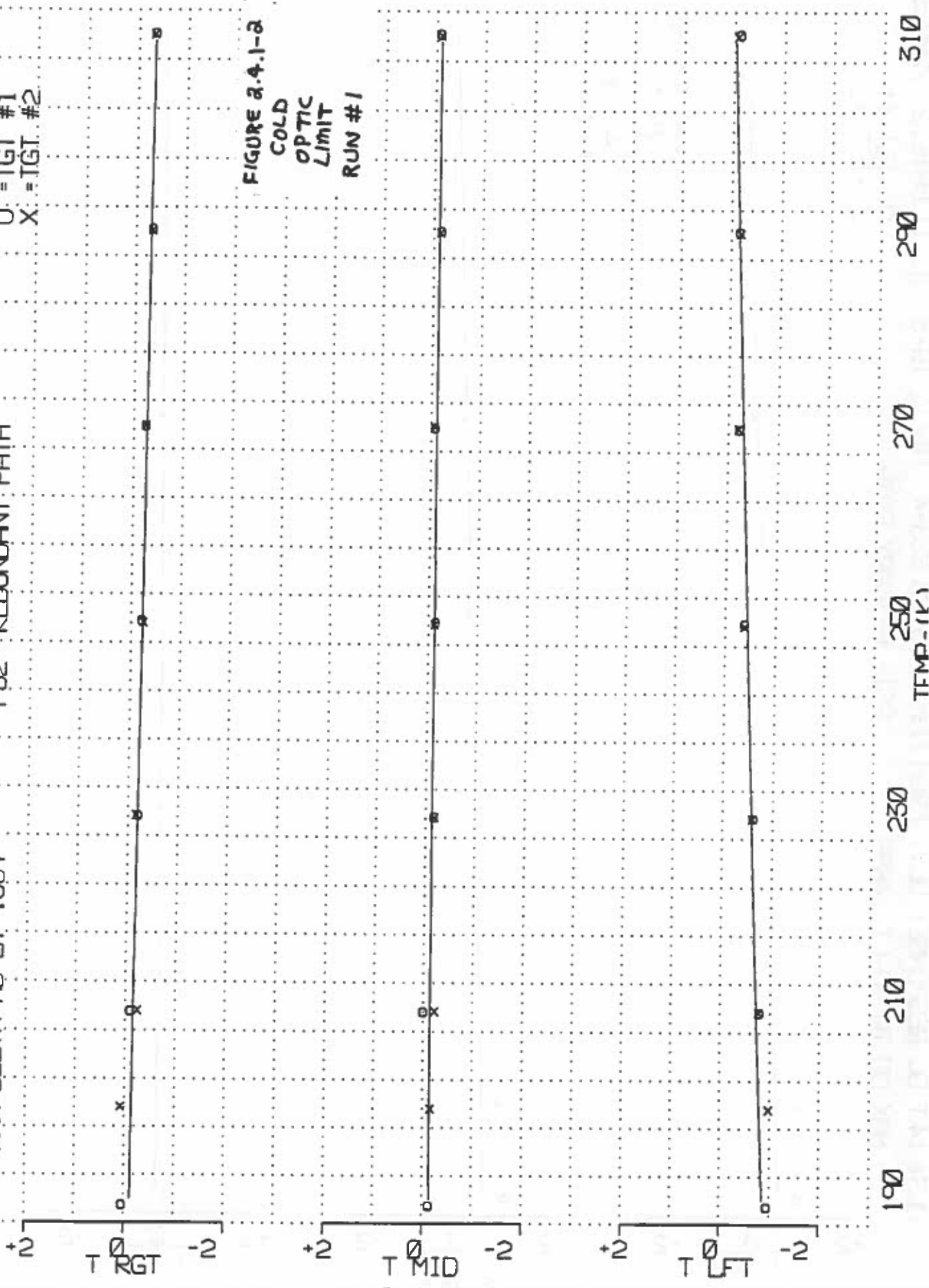
COLD
OPTIC
LIMIT
RUN #1



2-100

NEW K FACTORS
CORRECTED TO TLV=14

14
OLS# 14 T DC RESPONSE: DEV. FROM NOM PSU=26 SSS=5 M1=-8 TG=3 TL=~~14~~ DATE: 2 /14/93
MAX DELTA MD-LT .584 PS2- REDUNDANT PATH
O = TGT #1
X = TGT #2



ULOH# 141 DU NLS UNBL. DLY. I NUI I M I I J U T E U J U U T T I I I - 1 6 1 0 T U I U - 1 1 U T I L I C / C W / N

PS1 - PRIMARY PATH

MAX DELTA MD-LT .4695

O = IGT #1

X = IGT #2

FIGURE 2.4.1-3
HOT
OPTIC
LIMIT
RUN #2

+2
T RGT
-2

+2
T MID
-2
2-102

+2
T LFT
-2

190 210 230 250 270 290 310
TEMP-(K)



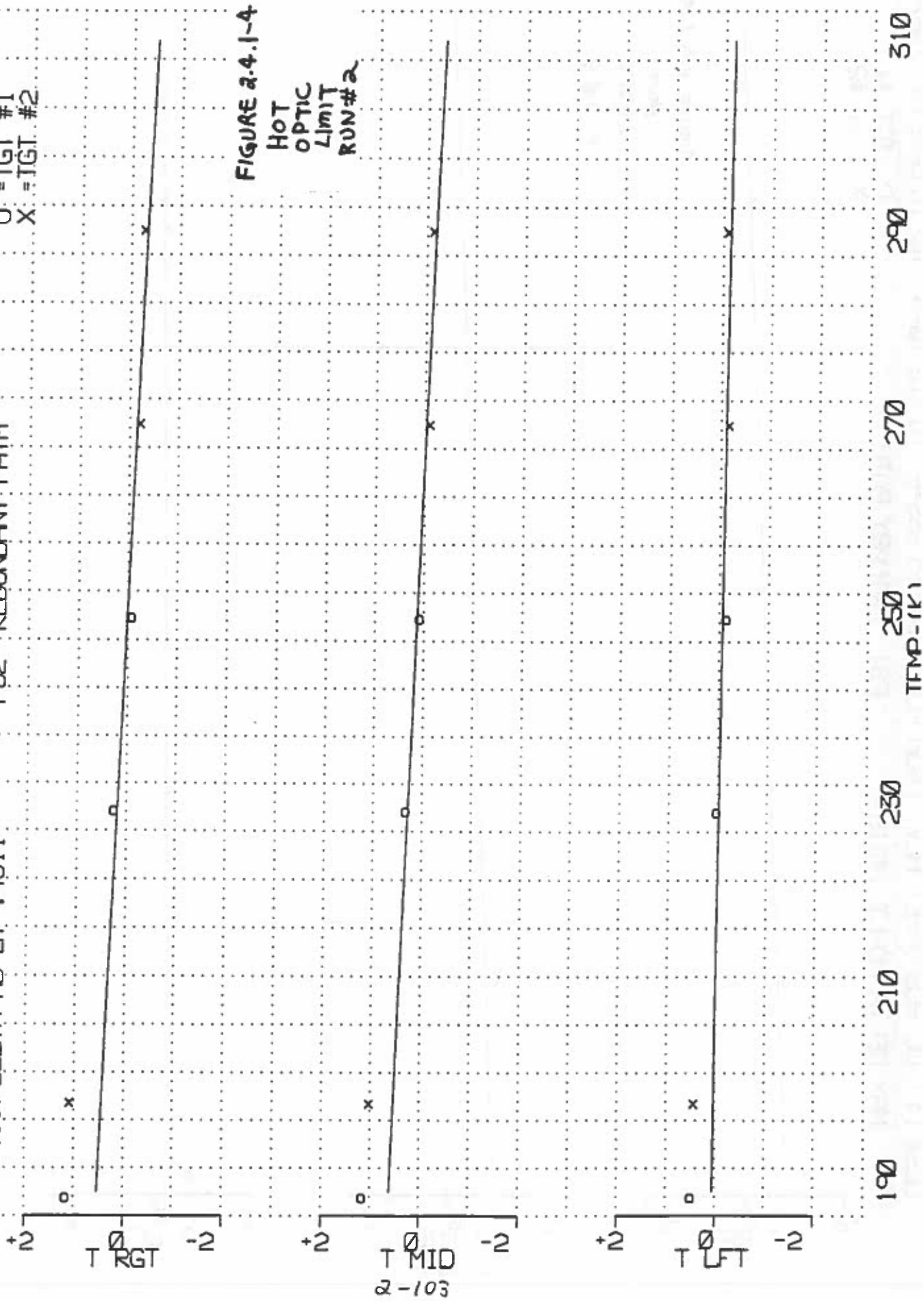
OLS# 14 T DC RESPONSE: DEV. FROM NOM PSU=26 SSS=4 MI=12 TG=3 TL=11 DATE:2 /20/93

MAX DELTA MD-LT: .4311

PS2- REDUNDANT PATH

O = TGT #1

X = TGT #2



PSI - PRIMARY PATH

MAX DELTA MD-LT .4613

O = TGT #1
X = TGT #2

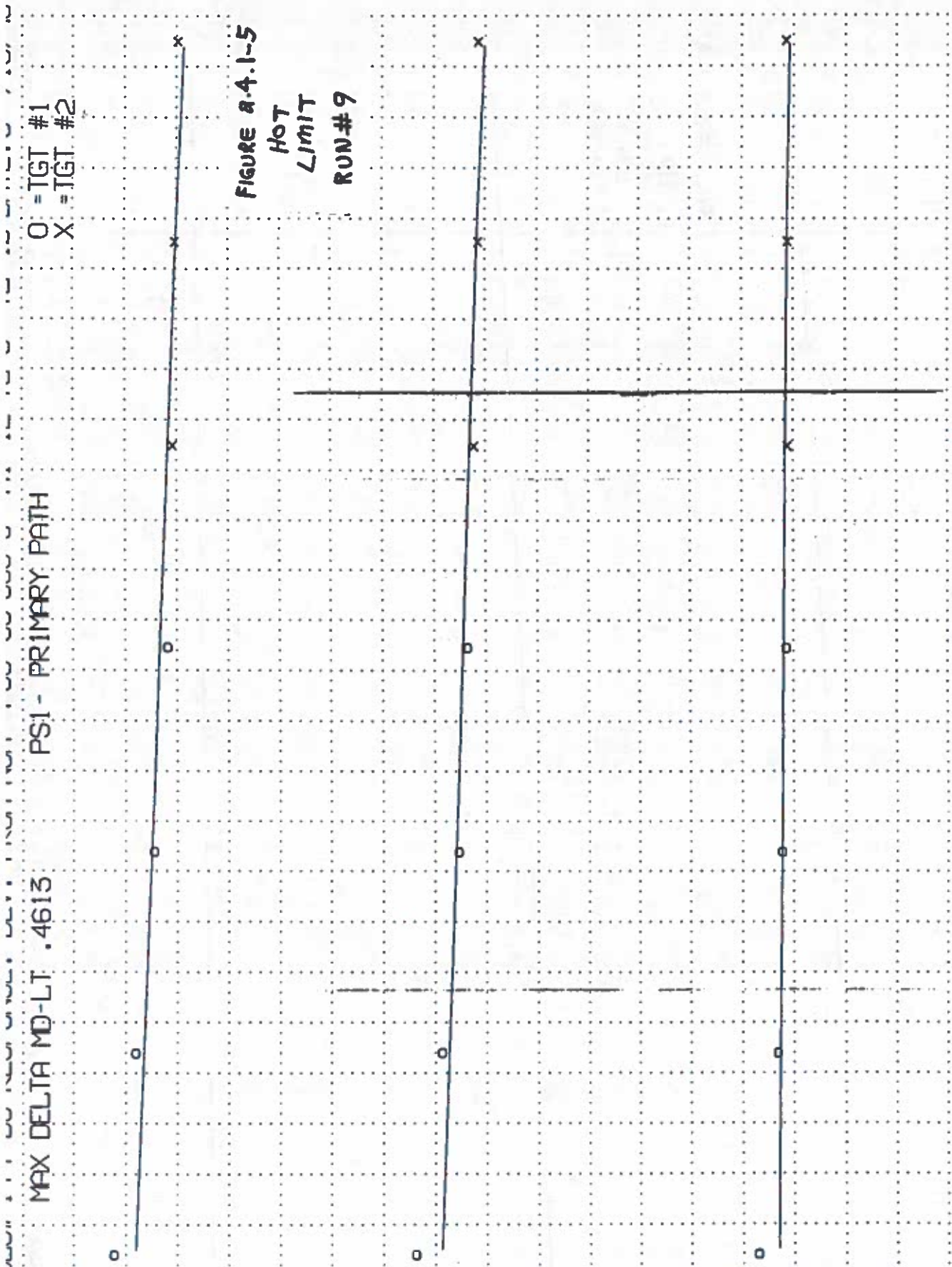
FIGURE A-4.1-5
HOT
LIMIT
RUN#9

+2 0 -2
T RGT

+2 0 -2
T MID

+2 0 -2
T LFT

190 210 230 250 270 290 310
TEMP - (K)



OLS# 14 T DC RESPONSE: DEV. FROM NOM PSU=33 SSS=6 M1=12 TG=3 TL=10 DATE:3 /18/93

MAX DELTA MD-LT .4807

PS2- REDUNDANT PATH

O = TGT #1

X = TGT #2

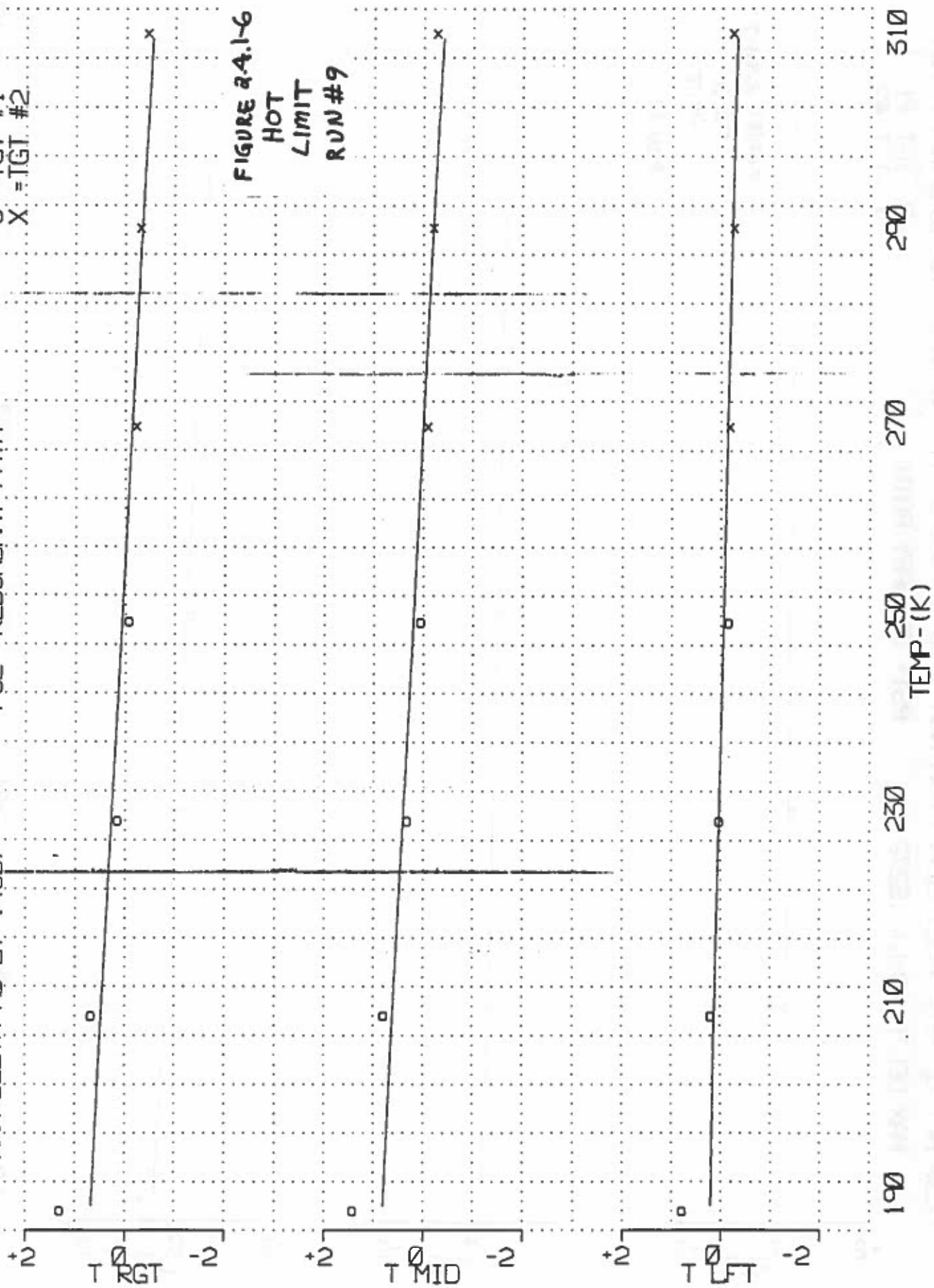


FIGURE 2.4.1-6
HOT
LIMIT
RUN #9

PS1 - PRIMARY PATH

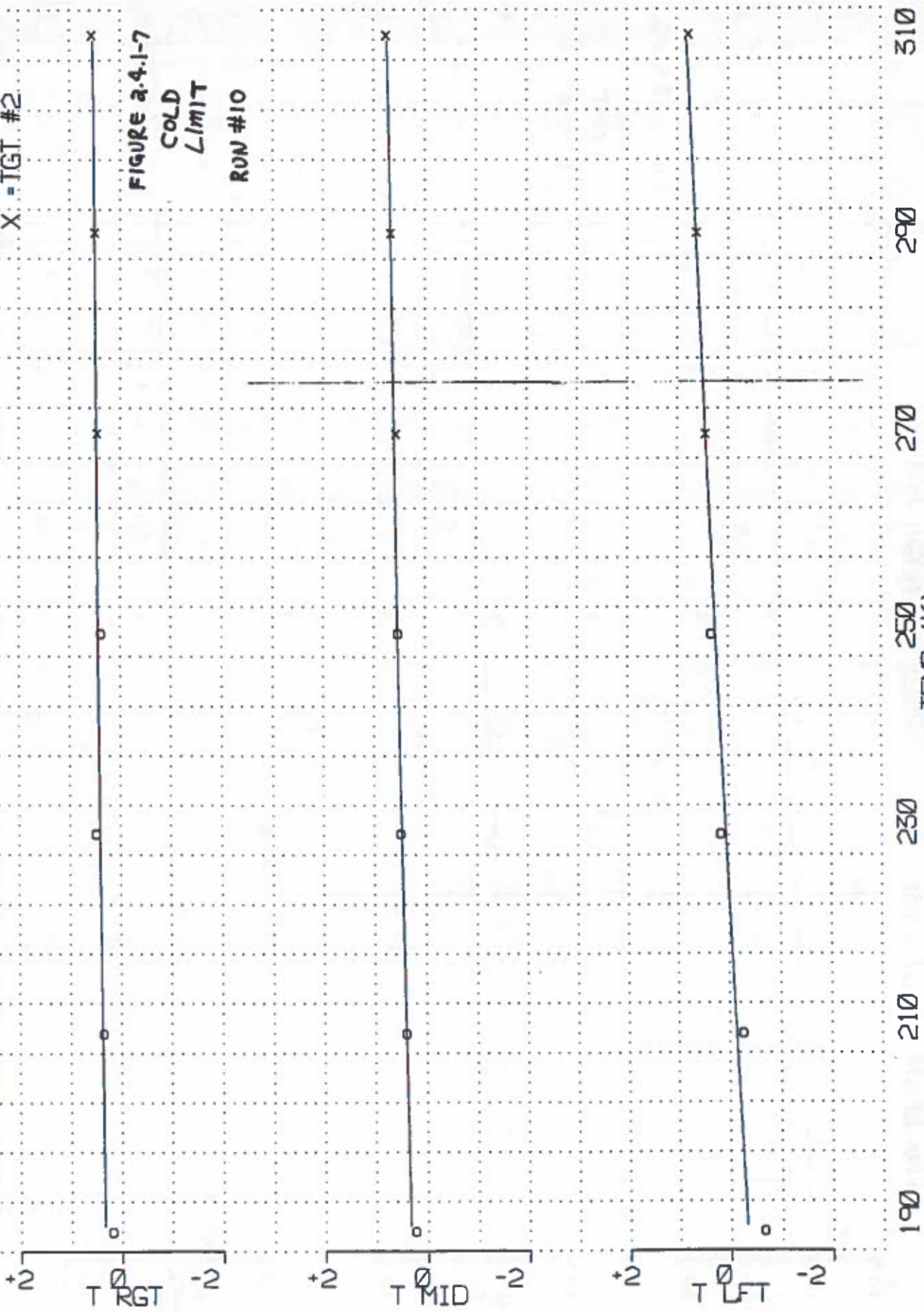
MAX DELTA MD-LT .5272

O = IGT #1
X = IGT #2

FIGURE 2.4.1-7

COLD
LIMIT

RUN #10

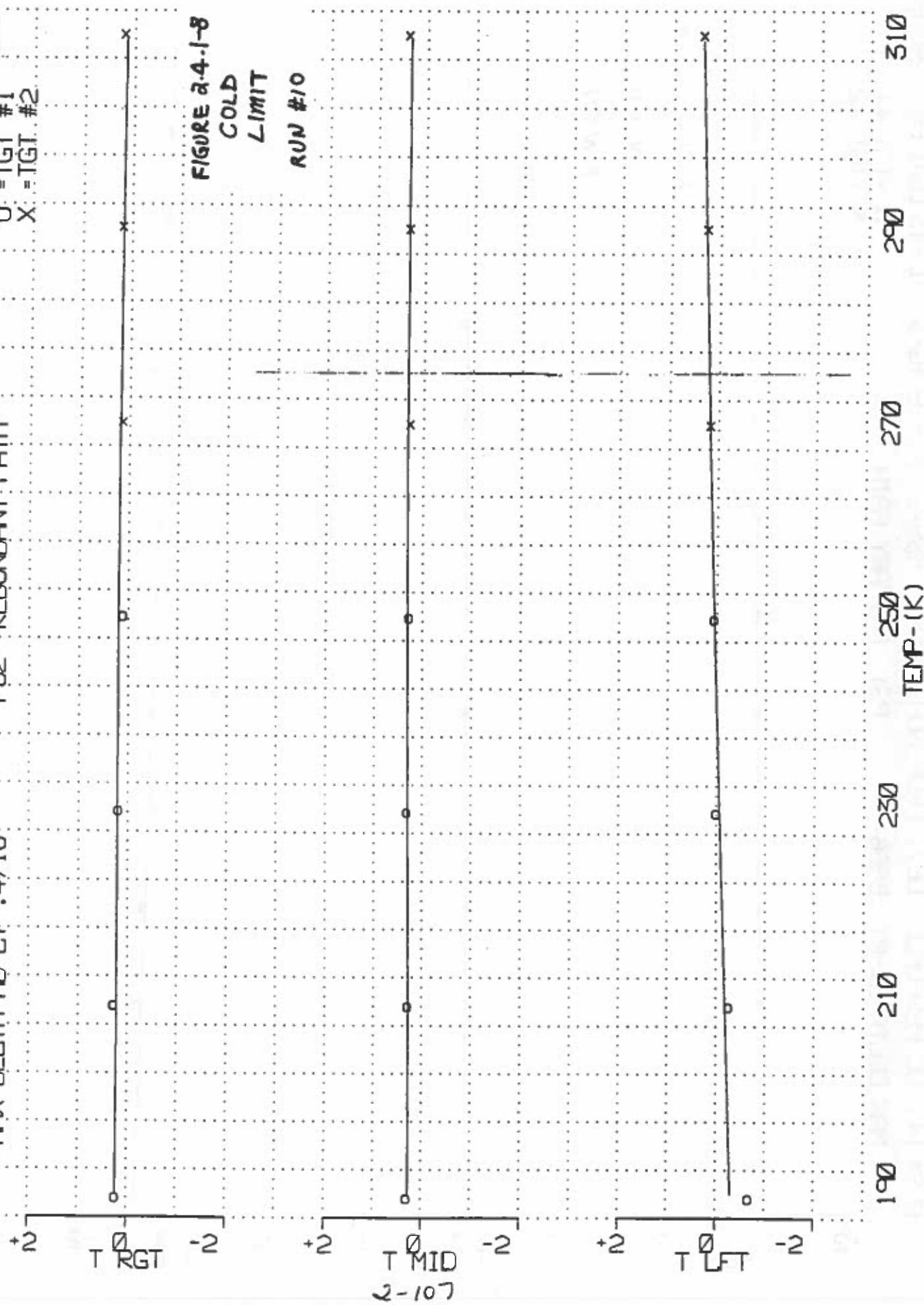


OLS# 14 T DC RESPONSE: DEV. FROM NOM PSU=5 SSS=2 M1=-8 TG=3 TL=13 DATE:3 /24/93

MAX DELTA MD-LT .4716

PS2- REDUNDANT PATH

O = TGT #1
X = TGT #2



UNIT 141 LN NL31 UNAL, LLY, 1 NUT1 NUT1 1 30-24 30300 111- 0 1970 147 10 UNIL.00 / 2 14 12

MAX DELTA LT-RT .5966
PS1 - PRIMARY PATH

O = TGT #1
X = TGT #2

FIGURE 2.A.1-9

NOMINAL

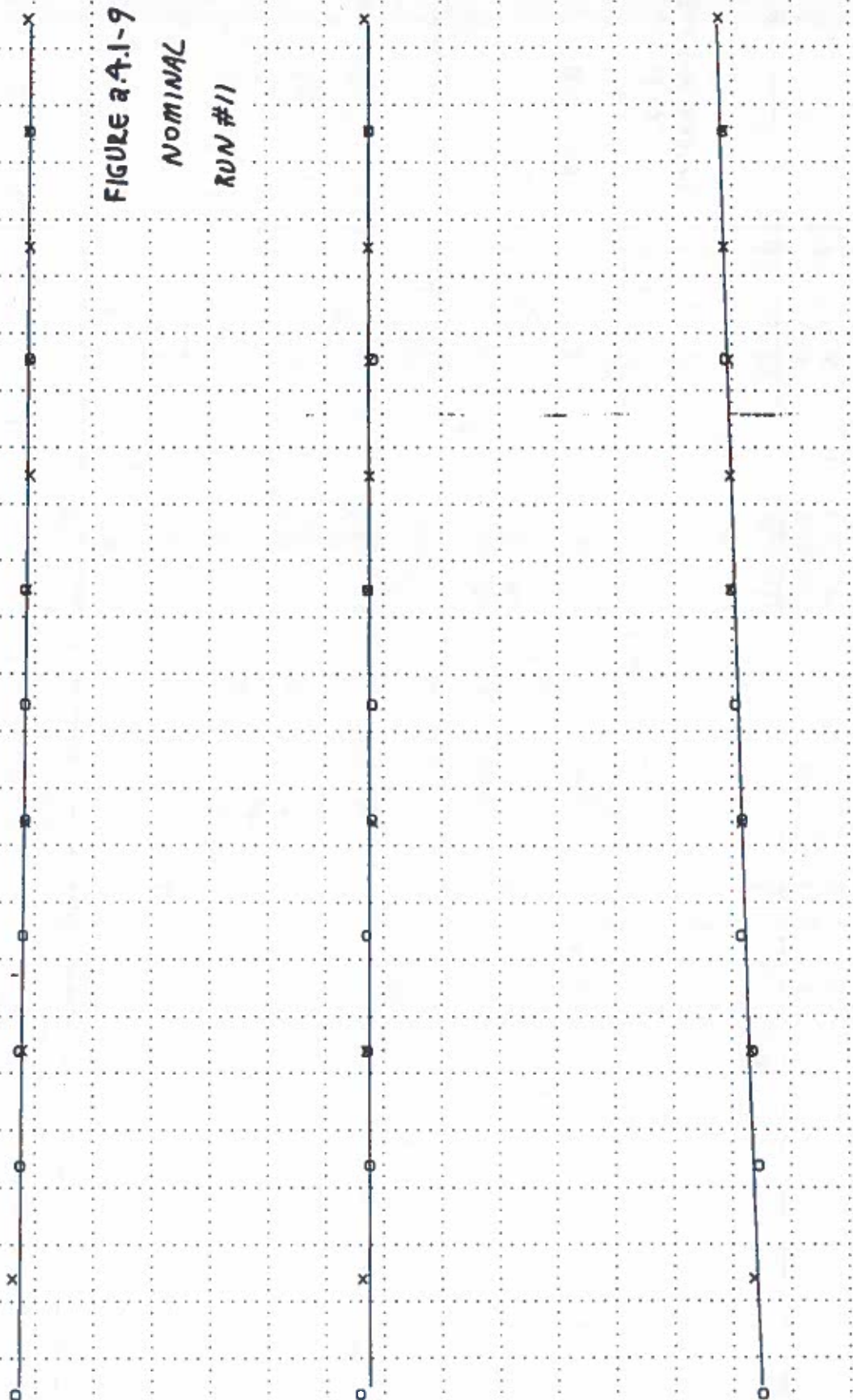
RUN #11

+2
T RGT
-2

+2
T MID
-2
2-108

+2
T LFT
-2

190 210 230 250 270 290 310
TEMP-(K)



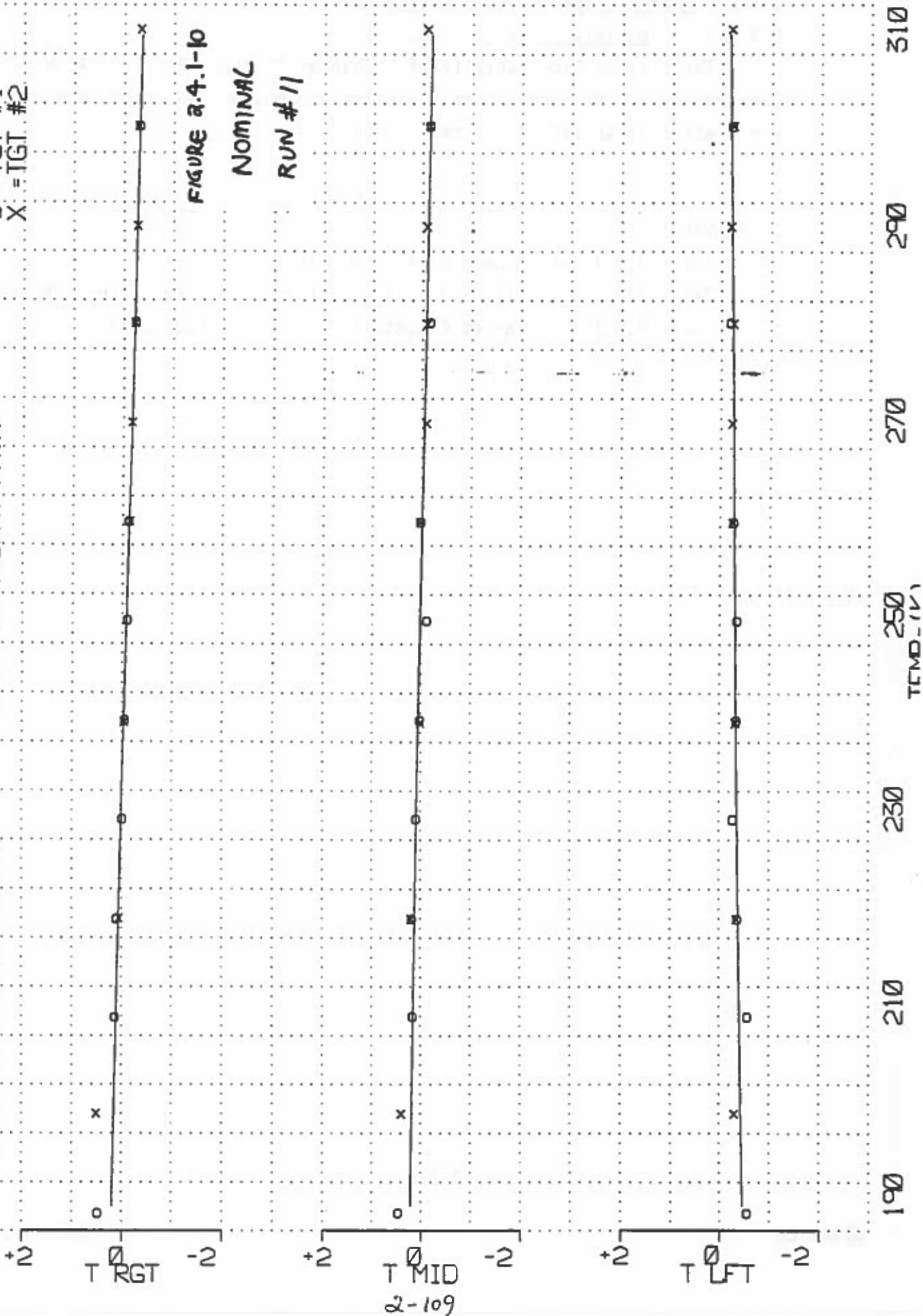
OLS# 14 T DC RESPONSE: DEV. FROM NOM PSU=24 SSS=5 M1=-8 TG=3 TL=13 DATE:3 /29/93

MAX DELTA MD-LT .5568

PS2- REDUNDANT PATH

O = TGT #1

X = TGT #2



2.4 RADIOMETRIC ACCURACY

2.4.1 T Channel Radiometric Accuracy (Cont'd)

2.4.1.1 Repeatability (3.2.1.1.4.1a)

The 1 sigma Repeatability of T Channel DC Response is 0.241°K compared to a 0.42°K one sigma specification maximum and therefore OLS #14 does meet this specification requirements.

ATTACHMENTS

Table 2.4.1.1-1 Repeatability Contributors

Table 2.4.1.1-2 Gain and Bias Variations with Temperature Change

Table 2.4.1.1-3 Target Crosstalk, T Clamp Leakage Data

TABLE 2.4.1.1-1
OLS #14
REPEATABILITY CONTRIBUTORS SUMMARY

<u>ERROR SOURCE</u>	<u>ONE SIGMA ERROR (K·)</u>
1. Diurnal M1 Temperature Change (4·C)	
A. Quantization of T Level Command	0.17
B. Inability to Compensate Actual Effect Exactly	0.060
2. Temperature Change PSU \pm 4.5·C, SSS \pm 1·C	
A. Effect due to Gain Change	0.033*
B. Effect due to Bias Change	0.043*
3. T Clamp Shaper Compensation	0.08
4. T Clamp Leakage	<u>0.128*</u>
TOTAL RSS REPEATABILITY ERROR (·K)	0.241
SPECIFICATION LIMIT, ·K, ONE SIGMA	0.42 MAXIMUM

*FROM TEST DATA (REDUCED)

Discussion of Repeatability Calculations

1. Diurnal M1 Temperature Change

A. The effects of M1 temperature (more properly the foreoptics temperature) are a Repeatability error source. The foreoptics thermal time constant is short enough to permit significant diurnal temperature variations. The ability to compensate for foreoptics temperature using the T Level command greatly reduces this error but does not eliminate it. Although calculations enabling ground compensation smaller than the quantization of the T Level command are possible, it is herein assumed that they will not generally be made. Therefore an error is ascribed due to the T Level quantization as follows:

$$\begin{aligned} & 0.294\text{°K RMS T Level Cmd. Quantization Error at } 210\text{°K } (1.02\text{°} \times 1/\sqrt{12}) \\ \times & 0.564 \text{ RMS Temperature Linearity Effects over } 210\text{-}310\text{°K dynamic range} \\ = & 0.17\text{°K RMS error} \end{aligned}$$

B. The fact that foreoptics temperature effect cannot be accurately predicted by the single monitor of M1 temperature means that in times of sharp transition the ability to compensate is impaired. It has been assumed that this error may be represented by a 1°C lag in M1 temperature during the 1/3 of the orbit that sharp transitions occur. Therefore the inability to compensate the actual effect is ascribed the following error:

$$\begin{aligned} & 1\text{°C} \quad \text{Lag in M1 Temperature} \\ \times & 1/\sqrt{3} \quad \text{RMS Over total orbit} \\ \times & 0.185 \quad \text{T Left T Mid T Right average sensitivity coefficient of} \\ & \quad \text{video at } 210\text{K to M1 temperature change for OLS \#14 (K} \\ & \quad \text{factor)} \\ \times & 0.564 \quad \text{Temperature Linearity Effects over dynamic range.} \\ = & 0.060\text{°K RMS error} \end{aligned}$$

Discussion of Repeatability Calculations

Total T channel gain change with temperature may be broken down into two components as follows:

$$\text{Total Gain } \Delta = [(\text{PSU } \Delta T) \times P_G] + [(\text{SSS } \Delta T) \times S_G]$$

where: P_G = PSU coefficient of gain, % per °C.
 S_G = SSS coefficient of gain, % per °C.

Similarly for bias changes with temperature:

$$\text{Total Bias } \Delta = [(\text{PSU } \Delta T) \times P_B] + [(\text{SSS } \Delta T) \times S_B]$$

where: P_B = PSU coefficient of bias, °K per °C.
 S_B = SSS coefficient of bias, °K per °C.

Data from two pairs of runs designated A and B, were used to solve these equations simultaneously:

$$\begin{aligned} \text{Gain } \Delta, \text{ Run A} &= G_A = (T_{PA})(P_G) + (T_{SA})(S_G) \\ \text{Gain } \Delta, \text{ Run B} &= G_B = (T_{PB})(P_G) + (T_{SB})(S_G) \\ \text{Bias } \Delta, \text{ Run A} &= B_A = (T_{PA})(P_B) + (T_{SA})(S_B) \\ \text{Bias } \Delta, \text{ Run B} &= B_B = (T_{PB})(P_B) + (T_{SB})(S_B) \end{aligned}$$

where: G = Total Gain change over temperature
 B = Total Bias change over temperature
 T_P = PSU Temperature change
 T_S = SSS Temperature change

Solved simultaneously for the temperature sensitivity factors, these equations can be reduced to the following:

$$S_G = \frac{(T_{PA})(G_B) - (T_{PB})(G_A)}{(T_{PA})(T_{SB}) - (T_{PB})(T_{SA})}$$

$$P_G = \frac{G_A - (T_{SA})(S_G)}{T_{PA}}$$

$$S_B = \frac{(T_{PA})(B_B) - (T_{PB})(B_A)}{(T_{PA})(T_{SB}) - (T_{PB})(T_{SA})}$$

$$P_B = \frac{B_A - (T_{SA})(S_B)}{T_{PA}}$$

2. SSS and PSU Temperature Change: Effect On Gain Change

The effects of SSS and PSU temperature change on gains were determined from parametric analysis of the four runs in Table 2.4.1.1-2. Solving simultaneous equations yielded sensitivity coefficients of gain change for both SSS and PSU temperature change. Temperature data from 5D-2 systems currently on-orbit indicates that the worst case SSS temperature variations are 1°C while worst case PSU temperature variations are 4.5°C. Using the worst-case factors yields:

$$\begin{aligned} S_G &= .035\% \text{ Gain change per degree SSS change} \\ &\quad \times 1^\circ\text{C temperature change} \\ &\quad \times .31^\circ\text{K RMS over 210K to 310K range} \\ &\quad \times 1/\sqrt{3} \text{ for uniform temperature distribution} \\ &= .006 \text{ deg} \end{aligned}$$

$$\begin{aligned} P_G &= -.040\% \text{ Gain change per degree PSU change} \\ &\quad \times 4.5^\circ \text{ temperature change} \\ &\quad \times .31^\circ\text{K RMS over 210K to 310K range} \\ &\quad \times 1/\sqrt{3} \text{ for uniform temperature distribution} \\ &= -.032 \text{ deg} \end{aligned}$$

RSS'ing these two contributors yields 0.033 degree total.

3. SSS and PSU Temperature Change, Effect On Bias Change

From Table 2.4.1.1-2:

$$\begin{aligned} S_B &= .049 \text{ deg Bias change per degree SSS change} \\ &\quad \times 1^\circ \text{ temperature change} \\ &\quad \times 0.564 \text{ RMS Temperature Linearization Effects, 210K to 310K} \\ &\quad \times 1/\sqrt{3} \text{ for uniform temperature distribution} \\ &= -.016 \text{ deg} \end{aligned}$$

$$\begin{aligned} P_B &= -.027 \text{ deg Bias change per degree PSU change} \\ &\quad \times 4.5^\circ \text{ temperature change} \\ &\quad \times 0.564 \text{ RMS Temperature Linearization Effects, 210K to 310K} \\ &\quad \times 1/\sqrt{3} \text{ for uniform temperature distribution} \\ &= -.040 \text{ deg} \end{aligned}$$

RSS'ing these two contributors yields 0.043 degree total.

4. T Clamp Shaper Compensation

The SSS temperature changes throughout each orbit are expected to cause a one sigma error of 0.146•K at 210K due to the compensation for T clamp temperature variation from 228K to 253K. This times the 0.564 RMS Temperature Linearization Effect over the dynamic range equals 0.08•K RMS error.

5. T Clamp Leakage

An along scan variation (ASV) effect may be caused by some of the scene radiance being viewed at the time of T clamp during the overscan period getting into the reference T Clamp value. This can happen if the T detector sensitivity extends slightly beyond the boundaries of M4' during the clamp time. The test performed for leakage is to view one target at 210•K during active scan while the other target (which is at the T Clamp angle) is varied over the 210• to 310•K dynamic range. This data is presented in Table 2.4.1.1-3. The effect on the response to the active scan target is attributed to the T clamp optical leakage.

Using the OLS #14 T data from Table 2.4.1.1-3, the T clamp leakage contribution can be calculated as follows:

$$\begin{aligned} \text{T clamp leakage ratio (LR)} &= 100\% \times \Delta T \times \frac{\frac{\partial P}{\Delta T} \cdot 214}{\Delta P} \\ &= \Delta T \times .50552\% \end{aligned}$$

This calculation is performed in the MODE 4 data reduction of T121T221S. The ratio calculated is:

0.150% T LEFT
0.083% T RIGHT

The peak error from T clamp leakage (due to the 310° background) can be calculated as follows:

$$\text{peak error} = LR \times \Delta N \times \left(\frac{\partial P^{-1}}{\Delta T_{210}} \right) - 1$$

$$\begin{aligned} &= \Delta T \times \frac{\Delta N}{\Delta P} \times \frac{\frac{\partial P}{\Delta T} 214}{\frac{\partial P}{\Delta T} 210} \\ &= \Delta T \times 0.8156 \times 1.0788 \\ &= \Delta T \times 0.880 \end{aligned}$$

where:

ΔP = Difference in radiance between 210° and 310°K

$$\begin{aligned} &= 16.742 \text{ E-4 w cm}^{-2} \text{ sr}^{-1} \text{ @ } 310\text{°K} \\ &\quad - \underline{2.3468\text{E-4 w cm}^{-2} \text{ sr}^{-1} \text{ @ } 210\text{°K}} \end{aligned}$$

$$= 14.395\text{E-4 w cm}^{-2} \text{ sr}^{-1}$$

and: ΔN = Difference in radiance between 240° and 310°K

$$\begin{aligned} &= 16.742 \text{ E-4 w cm}^{-2} \text{ sr}^{-1} \text{ @ } 310\text{°K} \\ &\quad - \underline{5.001 \text{ E-4 w cm}^{-2} \text{ sr}^{-1} \text{ @ } 240\text{°K}} \end{aligned}$$

$$11.741 \text{ E-4 w cm}^{-2} \text{ sr}^{-1}$$

$$\frac{\partial P}{\Delta T_{210}} = \text{slope of radiance curve at } 210\text{°K} = 6.7452 \text{ E-6}$$

$$\frac{\partial P}{\Delta T_{214}} = \text{slope of radiance curve at } 214\text{°K} = 7.277 \text{ E-6}$$

ΔT = measured change in response to 210° target as the background is varied from 210° to 310°K.

RMS ERROR = PEAK ERROR

 x 0.7605 for RMS distribution of leakage radiance over dynamic

range.

 x 0.564 RMS Temperature Linearization Effect

FROM MODE 4 Data reduction:

$$\begin{aligned} \text{Calculated RMS leakage error} &= 0.128\text{°K T LEFT} \\ &= 0.071\text{°K T RIGHT} \end{aligned}$$

The worst-case contribution to repeatability error by T-clamp leakage is therefore

0.128°K RMS.

TABLE 2.4.1.1-2

OLS #14

GAIN AND BIAS VARIATIONS WITH TEMPERATURE CHANGE
(M1 TEMP CORRECTED TO +12°C OR -8°C)

	SSS TEMP	PSU TEMP	T RGT		T MID		T LFT	
			% GAIN DELTA (%)	BIAS CHG. @ 190°K (-K)	% GAIN DELTA (%)	BIAS CHG. @ 190°K (-K)	% GAIN DELTA (%)	BIAS CHG. @ 190°K (-K)
M1 = -8°C (Run A)	4.8	23.7	-0.06	0.34	0.21	0.42	0.59	-0.35
	2.2	4.5	0.61	0.73	0.90	0.87	1.28	0.20
	2.6 T _{SA}	19.2 T _{PA}	-0.67 G _A	-0.39 B _A	-0.69 G _A	-0.45 B _A	-0.69 G _A	-0.55 B _A
M1 = +12°C (Run B)	11.2	37.6	-0.62	0.64	-0.49	0.59	-0.21	-0.29
	4.1	25.8	-0.40	0.61	-0.20	0.82	0.28	0.27
	7.1 T _{SB}	11.8 T _{PB}	-0.22 G _B	0.03 B _B	-0.29 G _B	-0.23 B _B	-0.49 G _B	-0.56 B _B
Calculated Sensitivity Factors	SSS: S _G PSU: P _G	S _B P _B	0.0349* -0.0396*	0.0490* -0.0269*	0.0244 -0.0392	0.0085 -0.0246	-0.0120 -0.0343	-0.0403 -0.0232

*WORST CASE VALUES

TABLE 2.4.1.1-3
 OLS #14
 TARGET CROSSTALK, T CLAMP LEAKAGE DATA*

SSS = +5°
 M1 = -8°

	T RIGHT	T MID	T CPL	T CPR	T LEFT	
T1 210° [T2 @ 310°] (T121T231G)	-1.10	-1.16	-1.77	-0.78	-1.90	02-14-93
Difference, ΔT	0.07	0.11	0.01	0.00	0.18	
T1 210° [T2 @ 210°] T121T221S	-1.03	-1.05	-1.77	-0.78	-1.72	02-15-93
Worst Case Data From T121T221S.ST Mode 4 Data Reduction: 2/15/93						
T clamp leakage ratio is			0.150%	0.083%		
Peak leakage error at 210°K is			0.261 K	0.145 K		
RMS leakage error at 210°K is			0.128 K	0.071 K		

*Data is FP Deviation in °K

2.4 Radiometric Accuracy

2.4.1 T Channel Radiometric Accuracy (Cont'd)

2.4.1.2 Stability (3.2.1.1.4.1b)

The T Channel Radiometric Accuracy (Stability) analysis in the OLS 5D-3 System Summary Report Paragraph 3.5.1.2, predicts 0.61°K 1 sigma error in stability compared to the 0.8°K maximum specification requirement. This calculation is applicable to all 5D-3 systems with 190-310°K total range.

ATTACHMENTS

Table 2.4.1.2-1 Stability Contributors Summary

Table 2.4.1.2-2 Change in BSL 210°, 310°K Points Between Runs

Table 2.4.1.2-3 Change in 210°, 310° Output Deviation From Nominal (°K) between Power Supply 1 and Power Supply 2

TABLE 2.4.1.2-1

STABILITY CONTRIBUTORS SUMMARY

		<u>ONE SIGMA ERROR (°K)</u>
<u>Shaped Bias</u>		
a) Open Loop Mirror Emissivity	0.1	
b) T Clamp Shaper Compensation - Temperature	0.23	
- Age	<u>0.17</u>	
RSS Total	0.30•K	
X RMS Temperature Linearization Effect = RMS Shaped Bias Errors	0.564	0.17•K
<u>Bias</u>		
a) Preshaper Gain	0.28	
- Inner Stage Temperature	0.28	
- Bias Current	0.24	
- Amplifiers	0.22	
b) Post Shaper DC Drift	<u>0.12</u>	
RSS Total = RMS Bias Error		0.45•K
<u>Gain</u>		
Postshaper Gain Changes - Amplifier		
over the 210-310K range, °K RMS Error		<u>0.38</u>
<u>TOTAL RSS Stability Error (Total Dynamic Range)</u>		0.61
Stability Error Specification (°K, 1 Sigma)		0.80 Maximum

Discussion of Stability Errors

The experimentally derived RMS change of the BSL(s) between runs was calculated to be 0.11°K, 0.07°K and 0.13°K for TRGT, TMID and TLEFT respectively. The two runs used were Run #3 and run #5. The results verify the analytical estimate of the stability over time intervals greater than one day. This data is tabulated in Table 2.4.1.2-2.

As an additional check of stability, the Fine-Primary outputs of the three segments as a deviation from nominal (°K) at 210° and 310° were compared using power supply 1 data of TDCRM3C.ST and power supply 2 data of 6X2X3A.ST. (Both from Run #11). This data is tabulated in Table 2.4.1.2-3. The deltas were calculated and RMS'd over the temperature range. The results are comparable to the "Change Between Runs" data.

TABLE 2.4.1.2-2

OLS #14

CHANGE IN BSL 210, 310K POINTS BETWEEN RUNS

SSS = 11•C, M1 = 15•C

	TG R/L	TL	T RGT		T MID		T LFT	
			210	310	210	310	210	310
T121T231B 02-23-93	3/3	11	0.59	-0.21	0.53	-0.15	-0.14	-0.19
T121T231B 02-28-93	3/3	11	0.63	-0.36	0.55	-0.24	0.03	-0.25
Change Between Runs			0.04	-0.15	0.02	-0.09	0.17	-0.06
RMS Change			0.11•K		0.07•K		0.13•K	

TABLE 2.4.1.2-3

OLS #14

T CHANNEL DC RESPONSE

DIFFERENCE BETWEEN POWER SUPPLIES 1 and 2

From Orbit Nominal (Run #11), SSS = +5°C, M1 = -8°C

	RIGHT		MID		LEFT	
	TGT-1 210°K	TGT-2 310°K	TGT-1 210°	TGT-2 310°	TGT-1 210°	TGT-2 310°
FP DEV [K] Power Supply 1 TDCRM3C.ST 03/31/93	0.25	0.09	0.23	0.30	-0.45	0.24
FP DEV [K] Power Supply 2 6X2X3A.ST 03/29/93	0.18	0.09	0.25	0.24	-0.41	0.23
Change °K	0.07	0.00	-0.02	0.06	-0.04	0.01
RMS °K	0.05		0.04		0.03	

2.4 RADIOMETRIC ACCURACY

2.4.1 T Channel Radiometric Accuracy (Cont'd)

2.4.1.3 Fixed Deviations (3.2.1.1.4.1c)

The Fixed deviations for OLS #14 are 0.27•K, 1 sigma, compared to the 0.6•K specification maximum. The portion of the fixed deviations which is calibrateable is 0.14•K RMS compared to the 0.4•K RMS specification maximum. The Fixed deviation calibration for separate detector segments is 0.63•K (worst case) compared to the 1•K spec. maximum. The maximum along scan variation was 0.12•K RMS for TF (Right) and 0.10•K RMS for TS compared to the 0.2•K RMS specification maximum.

ATTACHMENTS

Table 2.4.1.3-1	Fixed Deviations Contributors
Table 2.4.1.3-2	T Shaper Error Tabulation
Table 2.4.1.3-3	Target Deviation from Mean of Both Targets
Table 2.4.1.3-4	BSL Calibration Equations
Table 2.4.1.3-5	Fixed Deviation Calibration Differences for Separate Segments
Table 2.4.1.3-6	Along Scan Variation (265° to 310•K) within a Separate Segments
Table 2.4.1.3-7	Cone (Inner Stage) Patch Temp EST
Table 2.4.1.3-8	Cone Cooler Outer Stage Temp EST
Figure 2.4.1.3-1	5D3 Nominal Shaper Curve
Figure 2.4.1.3-2	Along Scan Variation, T Right, M1 = 12•C
Figure 2.4.1.3-3	Along Scan Variation, T Mid, M1 = 12•C
Figure 2.4.1.3-4	Along Scan Variation, T Left, M1 = 12•C
Figure 2.4.1.3-5	Along Scan Variation, T Right, M1 = -8•C
Figure 2.4.1.3-6	Along Scan Variation, T MID, M1 = -8•C
Figure 2.4.1.3-7	Along Scan Variation, T Left, M1 = -8•C
Figure 2.4.1.3-8	Along Scan Variation, T Auto M1 = 12•C
Figure 2.4.1.3-9	Along Scan Variation, T Auto M1 = -8•C

TABLE 2.4.1.3-1

OLS #14

FIXED DEVIATION CONTRIBUTORS

<u>DEVIATION SOURCE</u>	<u>ONE SIGMA ERROR (°K)</u>
1. Foreoptics Mirror Emissivity	0.11*
2. T Clamp Shaper Compensation	0.09
3. Transfer Function	
A. Non-Linearity	0.14* 0.4°K Spec Max
B. Shaper Components Variation	0.10
C. Detector Spectrum Variation (included in 3A)	-
4. Test Targets	
A. Temperature	0.10
B. Emissivity	0.10
C. Repeatability	<u>0.04*</u>
TOTAL (RSS) FIXED DEVIATION	0.27
FIXED DEV. SPECIFICATION LIMIT, °K ONE SIGMA	0.60 Maximum

* FROM TEST DATA ANALYSIS

	<u>DATA</u>	<u>SPEC MAX</u>
5. Fixed Deviation BSL Calibrations Match for Separate Segments (Worst Case)	0.63°K	1.°K
6. Along Scan Variations within a segment (265° to 310°K) Worst Case	0.12°K RMS	0.2°K RMS

Discussion of Fixed Deviation Tests and Calculations

1. Foreoptics Mirror Emissivity

The foreoptics mirror emissivity is a source of Fixed deviations as well as of Repeatability and Stability deviations. The correction in operation is made on M1 temperature only; whereas the entire foreoptics causes the offset phenomena. The ground calibration is made in the thermal vacuum chamber, where M1 is cooled radiatively via a cold tunnel, which fills the M1 view beyond the optical field of view of the system. Thus the foreoptics temperature distribution in the chamber differs somewhat from operation in space, especially in that M2 is colder in the chamber. It is not possible to determine accurately this effect based upon present knowledge. It is therefore assumed that it is equivalent to a 1°C difference in M1 temperature, or about 0.185°K at 210° using the actual OLS #14 M1 coefficient (K factor). The RMS Temperature Linearization Effect, 0.564, transforms this to a 0.11°K RMS contribution to fixed deviation.

2. T Clamp Shaper Compensation

The T Clamp shaper compensation contribution arises as follows. If the T Clamp emissivity were initially only 0.98 rather than the 0.995 used to calculate the compensation for T Clamp temperature, the error at 242°K would be 0.70°K. Although this error would be compensated for when the T channel adjustments were made, a change in T Clamp temperature to 256°K or 230°K would result in an error of 0.78°K, producing an uncompensated error of 0.08°K. Over the dynamic range this is equivalent to 0.09°K RMS.

3. Transfer Function

- A. The departure of the T channel radiometric transfer function from a linear relationship is not an error as such because it is known and compensation can be made for it. However, this type of deviation is included within the constraints of the Fixed deviation portion of the T channel radiometric accuracy spec. The nominal T Channel non-linear transfer function (shaper) error is tabulated in Table 2.4.1.3-2 and plotted in Figure 2.4.1.3-1. The nominal shaper error is 0.13•K RMS. This calculation is made with the 5D-3 shaper, which is also used on OLS #14. The worst-case reduced test data (from Tables 2.4.1-4,5 & 6) RMS Deviations of the points from the BSL for OLS #14, are 0.05•K for T Right (Smooth Backup), 0.05•K for T MID (Fine Backup and Smooth Primary and Backup) and 0.05•K for T Left (Fine Primary). The analytic value, (0.13•K RMS) and the worst-case test value of 0.05•K are RSS'ed to become 0.14•K RMS for this Fixed deviation source versus the 0.4•K maximum spec allowance for the calibrateable portion of the Fixed deviations.
- B. Departure of shaper components from design nominal values may cause additional peak errors of 0.25•K and are assigned a one sigma error of 0.1•K. These deviations would not be included in the measured deviation, because the data points are not close enough together to adequately detect them.
- C. The T detector spectrum difference from unit to unit is included in the calibration data and is therefore included in 3A, the non-linearity of transfer function.

4. Test Targets

The deviation of the reference test target indicated blackbody temperature from absolute is the result of contributions from three Fixed deviation sources.

A. The measurement of the averaged target temperature using the 5 thermocouples immersed in the target baseplate is subject to the accuracy of the PQL/Block V Thermocouple Acquisition and Control System calibrated per procedure MCS0116801B. A review of the calibration procedure and the equipment used has led to ascribing 0.1•K RMS to these sources.

B. The actual effective blackbody temperature of the target also deviates from that measured because of differing emissivity from that assumed and because the "true" radiant temperature differs from the measured temperature using the thermocouples. The effective emissivity is a combination of true emissivity (which is better than 0.996 according to Eppley) and reflectance of up to 0.004. An average emissivity of 0.998 is corrected for along with the thermocouple calibration. This source of deviation has been assigned 0.1•K RMS.

C. Actual measurements with OLS #14 system of the two Eppley IR Reference test targets over the 210• to 310•K range reveal small differences between the targets. These differences, as tabulated in Table 2.4.1.3-3, represent the target differences from the mean of both targets for T LFT and T RGT averaged from file TDCRM3C.ST (Run #11). The eleven differences when RMS'ed yield 0.04•K RMS deviation for this source.

Fixed Deviation Calibrations for Separate Segments

The calibrations are represented by the best straight line (BSL) equations for the separate segments. The BSL deviation expressions (from ideal), in $y = mx + b$ form for the segments are tabulated in Table 2.4.1.3-4. In T Smooth the Right and Left detector segments are averaged and used across the entire scan line.

In order to determine the differences in calibrations for separate segments, (for comparison to the specification) the T Fine BSL deviation equations of Table 2.4.1.3-4 were evaluated at 210°K and 310°K. The 3 possible segment differences were then calculated. Also, the worst-case deviations between segments were taken from the Best Straight Line Plots (Figures 2.4.1-1 thru 2.4.1-12). These results are tabulated in Table 2.4.1.3-5. The calibration differences for separate segments are within the 1°K maximum spec throughout the dynamic range of 210 to 310K for OLS #14.

Along Scan Variation (265°K to 310°K) Within a Segment

The Along Scan Variation (ASV) in T DC Response is considered to be a Fixed deviation, since as a function of scan angle it does not vary as a function of time per se. Figures 2.4.1.3-2, 3, 4, 5, 6, 7, 8 and 9 show the deviations along-scan vs. surface distance (nmi/100) for T RGT, T MID, TLFT, and T AUTO. The vertical scale factor is 1.0°K per cm. Four figures are for $M1 = -8^{\circ}\text{C}$ and four are for $M1 = +12^{\circ}\text{C}$. The computer printed number to the right of each curve is the computed RMS deviation in millidegrees K for the associated ASV plot. the RMS ASV values are only printed for the target temperatures above 265°K, i.e., the 270°, 290°, and 310°K plots.

The T DC response BSL calibrations are taken using test scan angles of -50° (-600nmi) for T LFT, +50° (+600nmi) for T RGT, and +0° (0 nmi) for T MID.

The OLS #14 has some ASV, but is within spec. The worst case (max) ASV RMS value within a segment for OLS #14 was 0.12•K and is entered in Table 2.4.1.3-6 to compare with the specification limit.

TABLE 2.4.1.3-2

T SHAPER ERROR LIST

The 190 to 310•K 5D3 T Shaper used for OLS #13 and up has 6 straight line segments of decreasing slope and 5 (inflections) diode break points. The significant features in the shaper curve are tabulated below:

<u>T</u> (•K)	<u>ERROR</u> (•K)	<u>REMARKS</u>
190	0	End point adjusted to be an Ideal Curve
195	-0.38	1st slope is parallel to Radiance (Smooth) Curve
201.5	0	
205	+0.26	1st diode cut-in
210	0	
215	-0.18	2nd slope is parallel to Radiance (Smooth) Curve
220.5	0	
224	+0.20	2nd diode cut-in
228.5	0	
235	-0.20	3rd slope is parallel to Radiance Curve
240.5	0	
245	+0.19	3rd diode cut-in
251.5	0	
257	-0.21	4th slope is parallel to Radiance Curve
263	0	
267	+0.165	4th diode cut-in
272	0	
279	-0.195	5th slope is parallel to Radiance Curve
284.5	0	
291	+0.16	5th diode cut-in
296	0	
303	-0.13	6th slope is parallel to Radiance Curve
310	+0.023	

The largest plus and minus errors in the 210K-310•K range are +0.20• and -0.21•K respectively.

The standard deviation = 0.13•K RMS over the 210• to 310•K dynamic range.

TABLE 2.4.1.3-3

OLS #14

TARGET DEVIATION FROM MEAN OF BOTH TARGETS

<u>TARGET_TEMP (°K)</u>	<u>DEVIATION (°K)</u>
210	-.06
220	-.02
230	-.04
240	+.00
250	+.02
260	+.00
270	-.01
280	-.04
290	+.04
300	+.02
310	+.06

$$RMS = \sqrt{\frac{\sum (Dev. \text{ } ^\circ K)^2}{11}} = 0.035^\circ K$$

TABLE 2.4.1.3-4

OLS #14

BSL CALIBRATION EQUATIONS

(From Tables 2.4.1-4,5,6)

			EVALUATED	
			@ 210°	@ 310°
T FINE (Primary)				
T-Right:	Error = -0.0020 (T-190) + 0.28	(°K)	.240	.040
T-Mid:	Error = -0.0001 (T-190) + 0.24	(°K)	.238	.228
T-Left:	Error = +0.0063 (T-190) - 0.48	(°K)	-.354	.276
T FINE (Redundant)				
T-Right:	Error = -0.0040 (T-190) + 0.14	(°K)	.060	-.340
T-Mid:	Error = -0.0030 (T-190) + 0.17	(°K)	.110	-.190
T-Left:	Error = +0.0025 (T-190) - 0.46	(°K)	-.410	-.160
T SMOOTH (Primary - SP MID)				
	Error = -0.0005 (T-190) + 0.30	(°K)		
T SMOOTH (Redundant - SB MID)				
	Error = -0.0032 (T-190) + 0.22	(°K)		

TABLE 2.4.1.3-5

OLS #14

FIXED DEVIATION CALIBRATION DIFFERENCES FOR SEPARATE SEGMENTS

Calculated from Run #11 BSL's in Table 2.4.1.3-4:

	<u>DIFFERENCE AT 210°K (°K)</u>	<u>DIFFERENCE AT 310K (°K)</u>	<u>SPECIFICATION (MAX)</u>
<u>PRIMARY</u>			
T Mid to T Right	0.002	0.188	1°K
T Mid to T Left	0.592	0.048	1°K
T Right to T Left	0.594	0.236	1°K
<u>REDUNDANT</u>			
T Mid to T Right	0.050	0.150	1°K
T Mid to T Left	0.520	0.030	1°K
T Right to T Left	0.470	0.180	1°K

Worst Case Differences from Best Straight Line
Plots (Figures 2.4.1-1 thru 2.4.1-12):

<u>RUN #</u>	<u>PRIMARY PATH</u>	<u>REDUNDANT PATH</u>	<u>SPEC. MAX.</u>
1	* 0.63° LFT-RGT	* 0.58° MID-LFT	1°K
2	0.47° MID-LFT	0.43° MID-LFT	1°K
9	0.46° MID-LFT	0.48° MID-LFT	1°K
10	0.53° MID-LFT	0.47° MID-LFT	1°K
11	0.60° LFT-RGT	0.56° MID-LFT	1°K

*WORST-CASE DATA

TABLE 2.4.1.3-6

OLS #14

ALONG SCAN VARIATION (265-K to 310-K) WITHIN A SEGMENT

(From ASV Graphs)

	ONE SIGMA ERROR (•K RMS)	SPEC LIMIT (•K RMS)
<u>T-FINE</u>		
T-Left Segment	0.092	0.2
T-Mid (Sum) Segment	0.102	0.2
T-Right Segment	0.116	0.2
<u>T-SMOOTH</u>		
T-Sum	0.102	0.2

TABLE 2.4.1.3-7
CONE COOLER S/N 026
THERMISTOR S/N KC-50
OLS-14

CONE (INNER STAGE) PATCH TEMP. EST

<u>TEMPERATURE °K</u>	<u>PATCH EST. VOLTS</u>
95	5.935
96	5.513
97	5.124
98	4.766
99	4.435
100	4.129
101	3.847
102	3.587
103	3.346
104	3.124
105	2.919
106	2.729
107	2.553
108	2.390
109	2.239
110	2.100
111	1.970
112	1.850
113	1.738
114	1.635
115	1.538
116	1.448
117	1.365
118	1.287
119	1.215
120	1.147
121	1.084
122	1.025
123	0.970
124	0.919
125	0.870

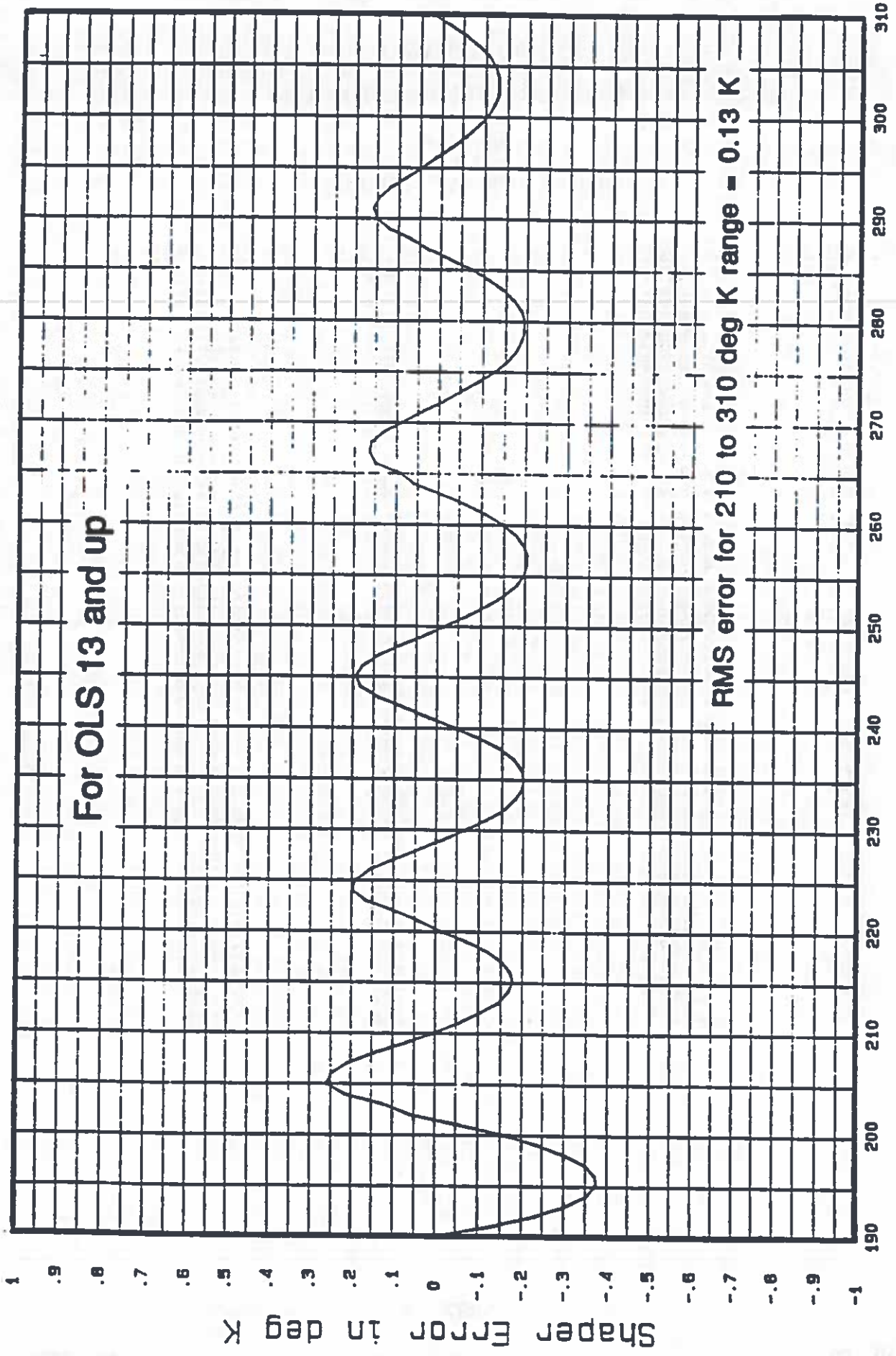
TABLE 2.4.1.3-8
 CONE COOLER OUTER STAGE TEMP EST

OLS #14

T CONE TEMP EST (EST #33)

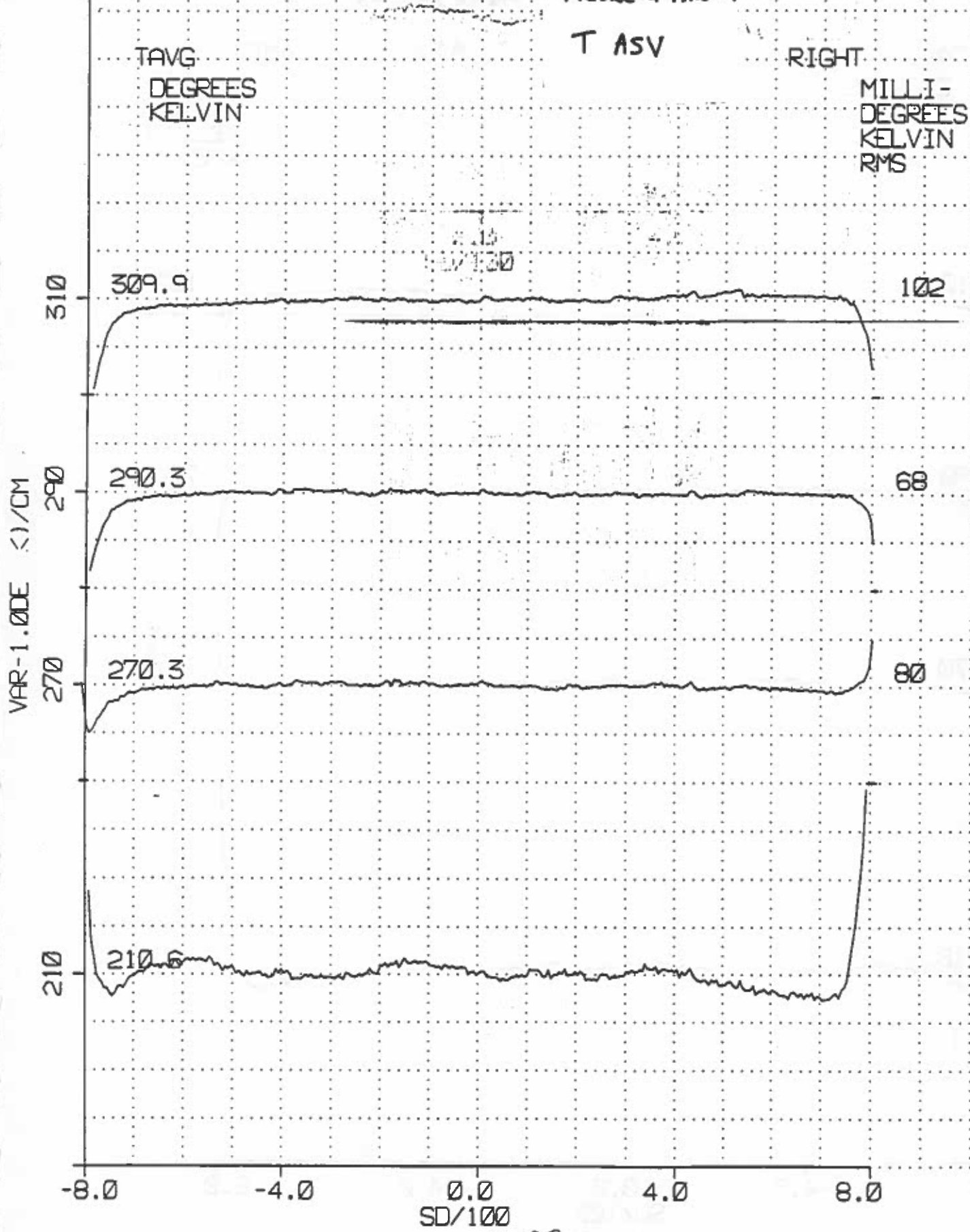
<u>T (DEG K)</u>	<u>EST VOLTS</u>	<u>T (DEG K)</u>	<u>EST VOLTS</u>
158	4.8221	194	4.1282
159	4.8181	196	4.0328
160	4.8136	198	3.93
161	4.8088	200	3.8195
162	4.8035	202	3.7016
163	4.7978	204	3.5769
164	4.7915	206	3.4468
165	4.7848	208	3.3115
166	4.7775	210	3.1719
167	4.7695	212	3.0292
168	4.7609	214	2.8844
169	4.7515	216	2.7386
170	4.7414	218	2.5924
171	4.7306	220	2.4475
172	4.7188	222	2.305
173	4.7063	224	2.1659
174	4.6926	226	2.0302
175	4.678	228	1.8995
176	4.6622	230	1.7735
177	4.6454	235	1.4832
178	4.6273	240	1.2308
179	4.608	245	1.0159
180	4.5874	250	0.8359
181	4.5654	255	0.6873
182	4.5418	260	0.5650
183	4.517	265	0.4653
184	4.4904	270	0.3842
185	4.4622	275	0.3182
186	4.4323	280	0.2646
187	4.4008	285	0.2207
188	4.3673	290	0.1852
189	4.3322	295	0.1560
190	4.2951	300	0.1320
192	4.2156	305	0.1123

5D--3 Nominal Shaper Error Function



Temperature in deg K
FIGURE 2.4.1.3-1

FIGURE 4.13-2



SYSTEM 14, DATE: 220 TIME 1000 SSS= 5., M1= 12, TG= 3., TL= 11

FIGURE 2.4.1.3-3

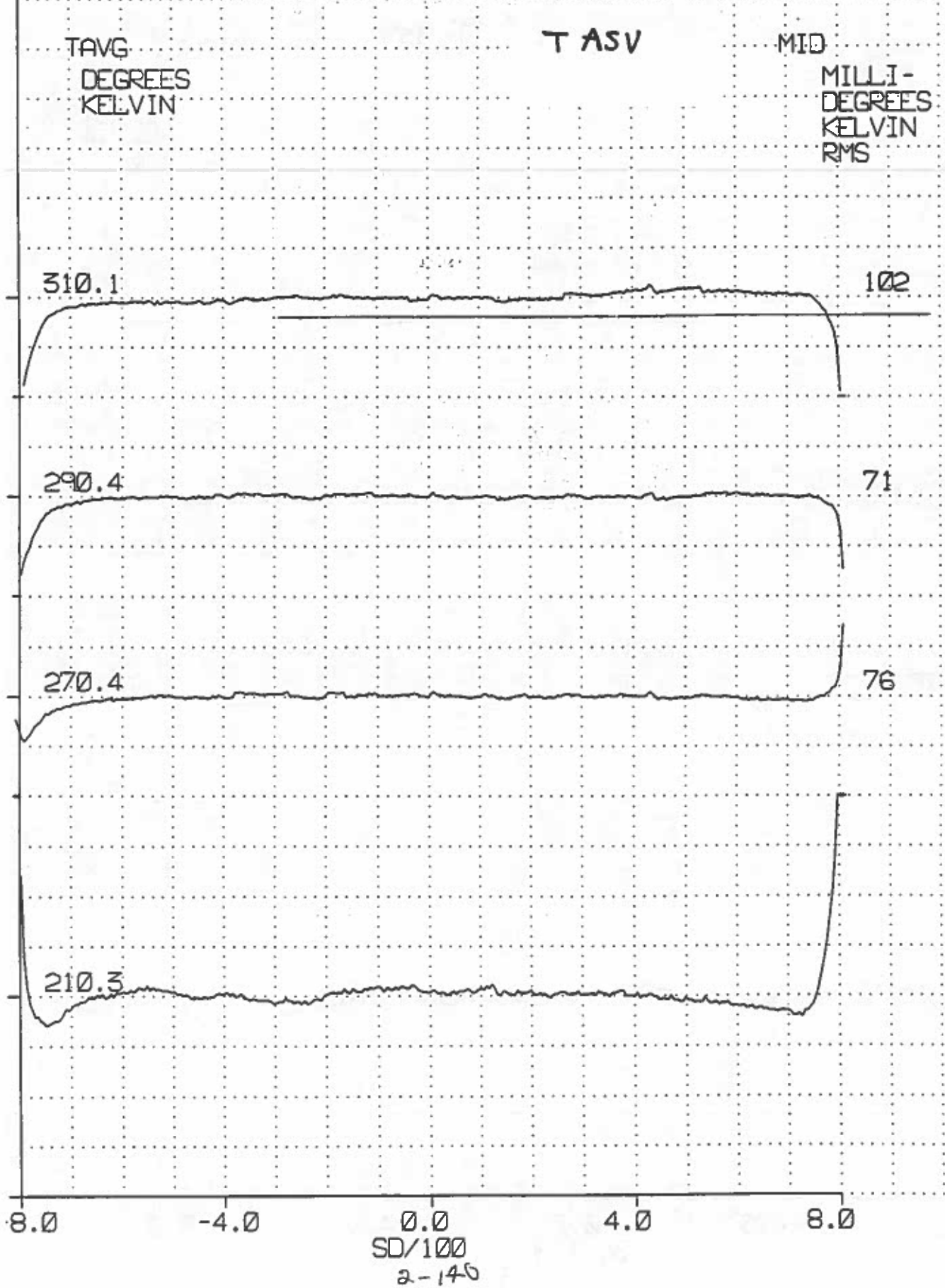
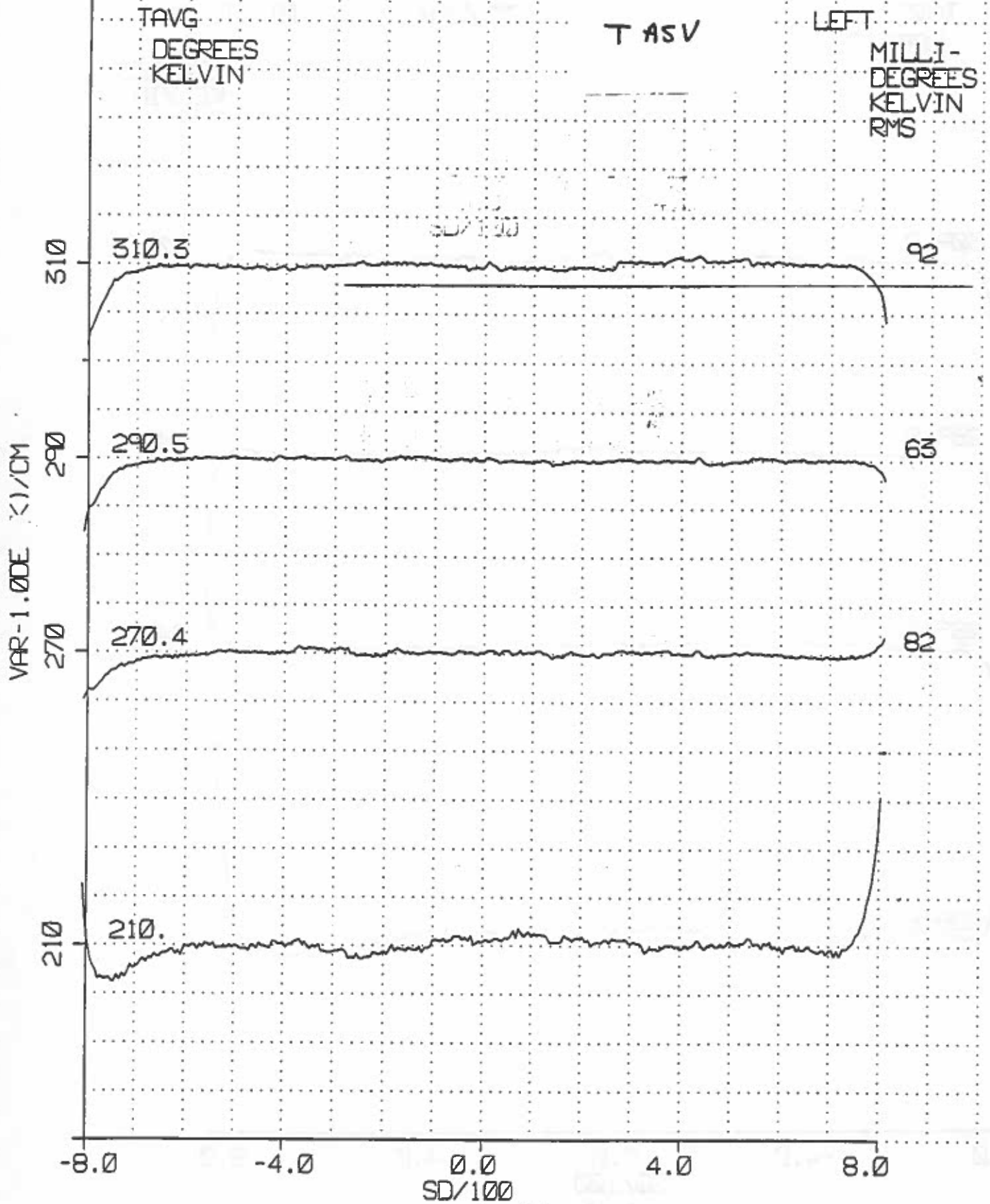
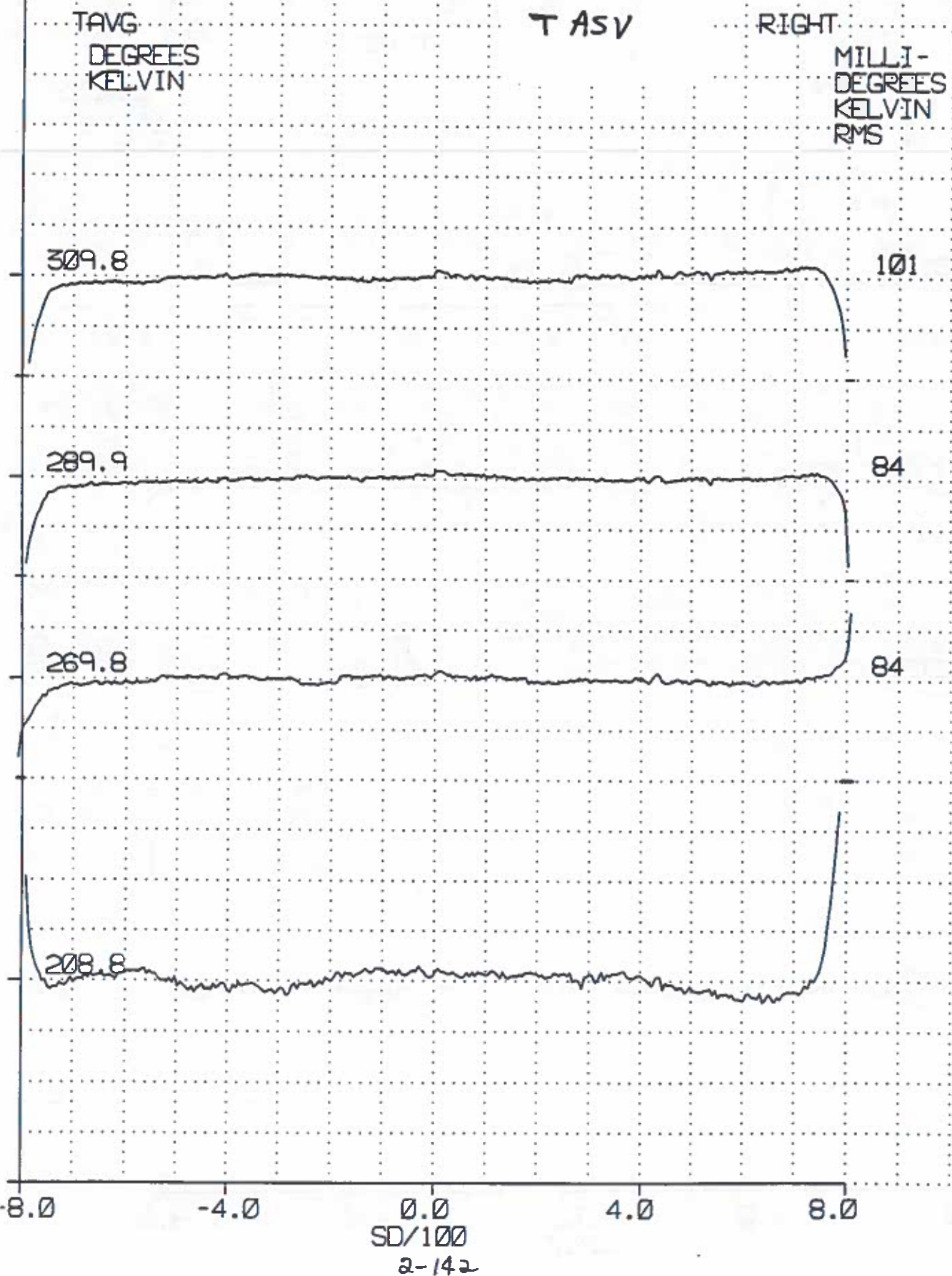


FIGURE 2.4.1.3-4



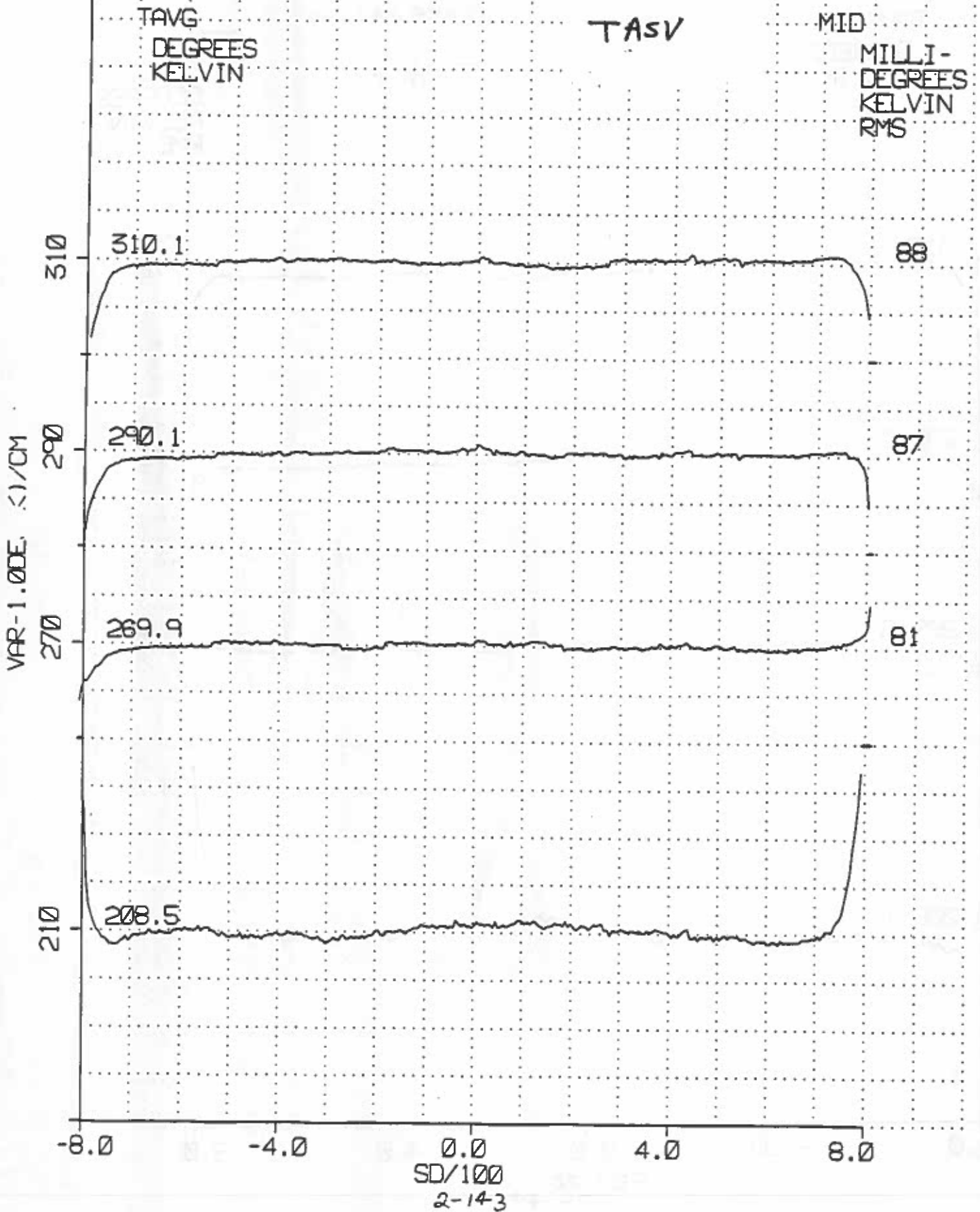
SYSTEM 14, DATE: 215... TIME 1700 SSS= 5., M1= -8, TG= 3., TL= 13

FIGURE 24.13-5



SYSTEM 14, DATE: 215 TIME 1700 SSS= 5., M1= -8, TG= 3., TL= 13

FIGURE 2-4.1.3-6



SYSTEM 14, DATE: 215 TIME 1700 SSS= 5, M1= -8, TG= 3, TL= 13

FIGURE 24.1-3-7

LEFT

TAVG
DEGREES
KELVIN

T ASV

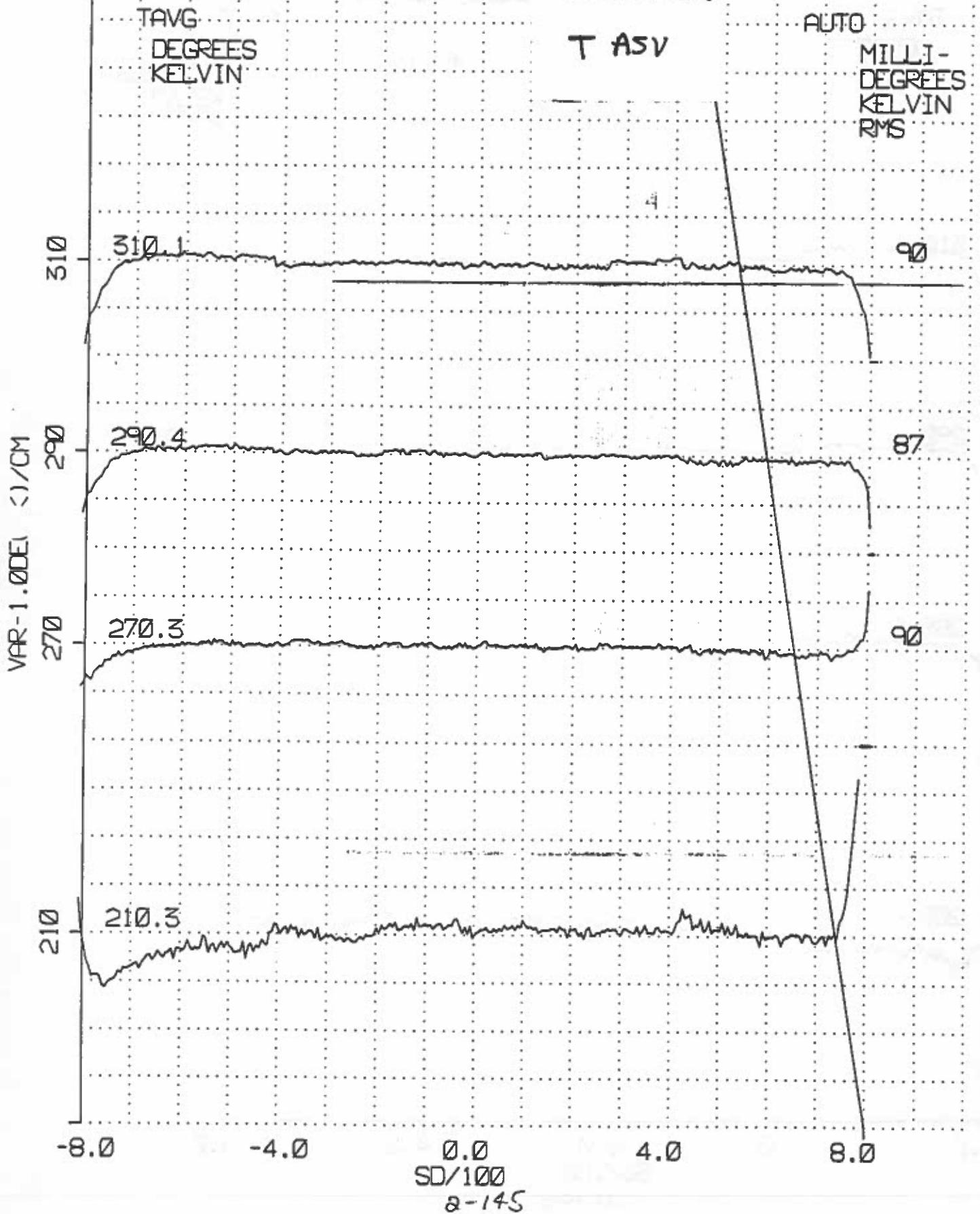
MILLI-
DEGREES
KELVIN
RMS



SD/100
2-144

SYSTEM 14, DATE: 220 TIME 1000 SSS= 5, M1= 12, TG= 3, TL= 11

FIGURE 2-4.3-8



SYSTEM 14 , DATE: 215 TIME 1700 SSS= 5 , M1= -8, TG= 3 , TL= 13

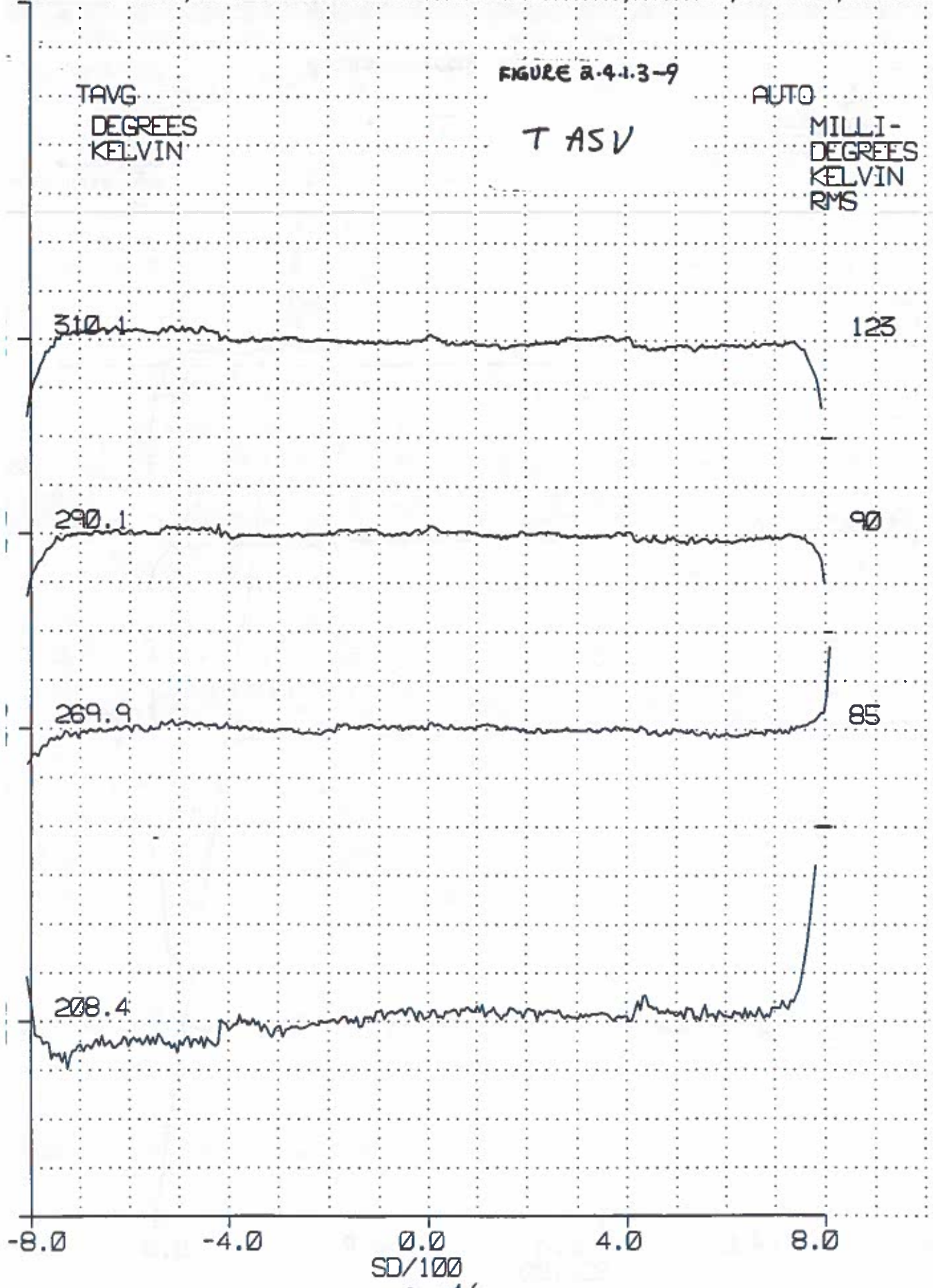
FIGURE 2-4-1.3-9

TAVG
DEGREES
KELVIN

AUTO

TASV

MILLI-
DEGREES
KELVIN
RMS



SD/100
2-146

2.4 Radiometric Accuracy (Cont'd)

2.4.2 Daytime Radiometric Accuracy (3.2.1.1.4.2)

OLS #14 achieved the 7% absolute radiance requirement by setting the HRD channel gain as shown on the L channel DC Response plot, using the calibrated light source (VULS).

The analysis of the calibration accuracy and the L-Day channel stability show within-specification performance. The gain ratios (PMH/PML, PML/HRD, and PMH/HRD) were measured during bearing retrofit retest using the VULS during Acceptance Test in test 6x2x1.ST, and vary less than 0.9% from the average of the ratios. The gain ratios measured in test 6x2x2.ST using a less accurate test method show greater variation.

The plot of L DC Response contains the calculated sensor switch points, $S(x)$, and relative gains $P(X)$, which are stored in the OLS Constants Memory page zero, locations 071 through 077. $P(2)$ and $S(2)$ represent the bypass of the PMT 1/9 mode, which is not usually implemented on orbit.

OLS #14 exhibited 1.33dB drop in transmission when varying the SSS from room temperature to +5°C. Additionally, OLS 14 exhibited a lower optics transmission than typical OLS units by approximately 2.5dB in the HRD channel. Thus $P(0)$ must be reset to $6.0 \text{ (nom)} + 1.33 + 2.5 = 9.83\text{dB}$. Rounding off to the nearest 1/8 dB gives 9.875dB as the new setting for $P(0)$.

P1 is derived using the PMT LO/HRD average gain value of 48.44dB with a compensation for the HRD Loss and PMT Gain Ratios with temperature from the bearing retrofit retest data as plotted in figure 2.4.2-2 and converted to dB of 1.33 dB and 0.18 dB, respectively. The P1 value is $48.44 + 1.33 + 0.18 = 49.95$ (rounded to nearest 1/8th dB = 50.00).

ATTACHMENT: OLS #14 L Channel DC Response Plot

Table 2.4.2-1 OLS #14 DC Response Stability

Table 2.4.2-2 OLS #14 PMT/HRD DC Response vs. SSS Temp.

OLS14 L CHANNEL DC RESPONSE

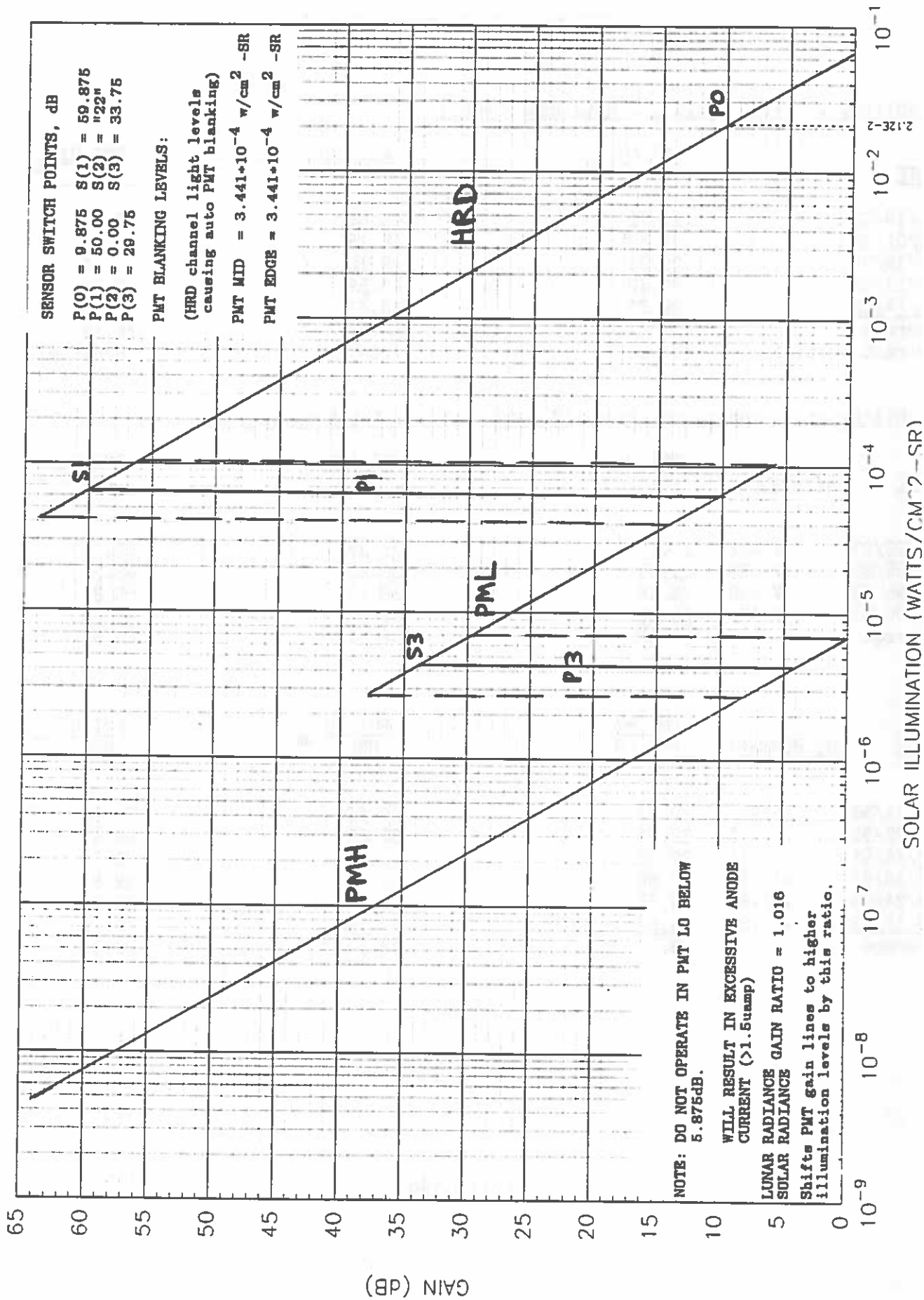


Table 2.4.2-1

OLS #14 L DC Response Stability

ability vs. Time (6x2x1.ST data using VULS)

TE	<u>PMT HI</u> <u>PMT LO</u> dB	<u>PMT LO</u> <u>HRD</u> dB	<u>PMT HI</u> <u>HRD</u> dB
/19/92	29.72	48.83	78.55
/01/92	29.69	48.15	77.85
/19/92	29.71	48.18	77.84
/13/93	29.70	48.59	78.30
<u>/14/93</u>	<u>29.71</u>	<u>48.43</u>	<u>78.13</u>
average	29.71db	48.44	78.13
irect Multiple)	(30.58)	(264.24)	(8063.06)

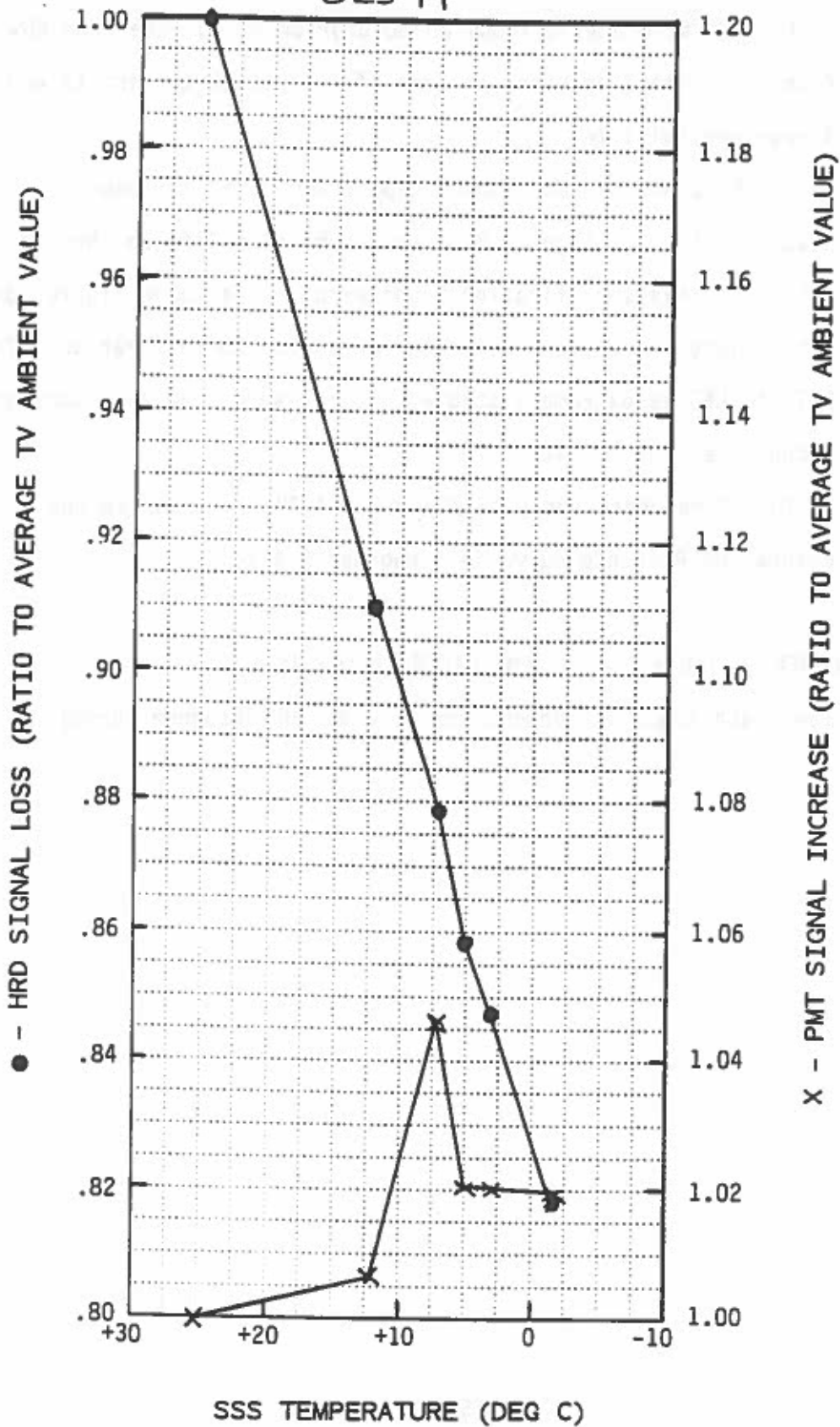
ability vs. Temperature (6x2x2.ST data using half sphere source)

TE	ENVIRONMENT	<u>PMT HI</u> <u>PMT LO</u> dB	<u>PMT LO</u> <u>HRD</u> dB	<u>PMT HI</u> <u>HRD</u> dB
/03/93	TV AMB	29.71	38.37	68.08
/05/93	TV AMB	29.63	38.63	68.26
/06/93	TV AMB	29.69	38.63	68.33
/06/93	TV AMB	<u>29.68</u>	<u>38.51</u>	<u>68.19</u>
average		29.68	38.54	68.22

TE	ENVIRONMENT	<u>PMT HI</u> <u>PMT LO</u> dB	<u>PMT LO</u> <u>HRD</u> dB	<u>PMT HI</u> <u>HRD</u> dB
/11/93	+5/-8	29.64	40.62	70.26
/22/93	+12/+15	29.64	39.83	69.46
/24/93	-2/-11	29.62	40.93	70.54
/18/93	+7/+12	29.66	40.27	69.93
/24/93	+3/-8	29.63	40.00	69.63
<u>/31/93</u>	<u>+5/-8</u>	<u>29.57</u>	<u>40.58</u>	<u>70.15</u>
average		29.63	40.37	70.00

TABLE 2.4.2-2. PMT/HRD DC RESPONSE vs. SSS TEMPERATURE

OLS 14



2.4 Radiometric Accuracy, (Cont'd)

2.4.3 Nighttime Radiometric Accuracy (3.2.1.1.4.3)

The PMT accuracy is required to degrade by no more than 60% from its initial accuracy at time of Acceptance testing to end of 3 year orbital life.

The 5D-3 OLS System Summary Report indicates PMT channel stability to be within 25.7% over the mission life so that the PMT meets this specification requirement. The above figures do not include corrections utilizing on-board LED calibration. The PMT CAL LED is extremely stable, and has an essentially constant output over the mission life.

The DC response curve of the OLS #14 PMT is shown in the L Channel DC Response curve in paragraph 2.4.2.

ATTACHMENT: Table 2.4.3-1 PMT CAL Baseline data

(See para 2.4.2 attachment for L. Chan. DC Response curve)

TABLE 2.4.3-1

PMT CAL BASELINE DATA

The PMT Cal voltage EST is a monitor of PMT sensitivity and stability characteristics. In order to relate on-orbit measurement of PMT CAL to measurement during Acceptance Test, the following data are provided. PMT CAL Voltage (EST #40) and PMT BU (Back-up) CAL V (EST #41) are sampled 500 times in test 6x6x2.ST. The PMT Cal voltage EST output is tabulated below.

TABLE 2.4.3-1

PMT CAL BASELINE DATA

<u>DATE</u>	<u>SSS TEMP</u>	<u>OUTPUT VOLTAGE (mV)</u>	
		<u>PMT CAL V (EST #40)</u>	<u>PMT BU CAL V (EST #41)</u>
11-25-92	+25	2534	2542
01-05-93	+25	2577	2589
02-03-93	+25	2509	2520
02-12-93	+5	2548	2557
03-20-93	+7	2524	2527
03-25-93	+3	2523	2525
04-01-93	+5	<u>2524</u>	<u>2533</u>
	AVERAGE	2534	2542
	Max change from AVERAGE	1.30%	1.85%

2.4 RADIOMETRIC ACCURACY, (Cont'd)

2.4.4 Gain Control Accuracy (3.1.4.4)

Along-Scan Gain Control (ASGC) accuracy is within the specification limit of 4dB of the smooth monotonic curve drawn through the nominal values of Gain Value versus Scense Solar Elevation (GVVSSE), tabulated in para 3.2.1.1.1.4 of DMSS-OLS-300.

System Tests 5x6x3.ST and 5x6x6.ST exercise the ASGC function through various combinations of extremes of slope and bias adjustment; and measure channel output for a wide range of values. These are automatically compared against stored test limits which ensure spec compliance.

An analysis using the ASGC software algorithm showing 3.25dB maximum error is summarized in the System Summary Report paragraph 3.5.4.

ATTACHMENTS: None.

2.4 RADIOMETRIC ACCURACY, (Cont'd)

2.4.5 Gain Control Adjustability (3.2.1.1.4.4 et al)

The OLS #14 Gain Control Adjustability is the same as for 5D-1 systems. The OLS 5D-3 System Summary Report paragraph 3.5.5 contains the analysis required to demonstrate conformance with the specifications, with additional information below.

TERMINATOR LOCATION (3.1.4.5.1)

The GNC Command (an uplink command) has a sub mode (GNC 10 0 X) which allows the GVSSE bias (terminator location) to be adjusted by X degrees, where X is a 6-bit 2's complement word with an LSB of 0.5°. This results in the required range of + 15.5 degrees.

GAIN CHANGE RATE (3.1.4.5.2)

The GNC command submode (GNC 10 1 X) allowed the GVSSE slope to be varied by a factor of 1+X over $\pm 48\%$ in 1.6% increments where X is a 6-bit 2's complement word with an LSB of 2^{-6} . This results in the required range of $\pm 48\%$. However, the BRDF change in the L channel gain calculation required the deletion of this capability to meet timing limitations in the OLSP. The 886 spec must be revised to reflect this change.

MAXIMUM GAIN SETTING (3.2.1.1.4.5.3)

The maximum ASGC gain is commandable. An operational value is determined in Early Orbit Calibrations. The value is stored in the Operational Constants Memory location page 0 Address 104 (BCMAX). The ASGC function & performance are exercised in tests 5x6x3.ST & 5x6x6.ST

COMMANDABLE T-CHANNEL GAIN (3.2.1.1.4.5.4)

The T-Channel Commandable Gain is exercised in test 6x8x2.ST. The channel output is measured for the entire range of commandable gains. The T Channel Gain is required to have the capability of being varied by command to be between 146% and 149% of the established minimum gain value (0 dB).

The actual percentage of TGAIN change was not measured as part of bearing retrofit. However, previously this was measured for OLS #14 as 49.3% for T Right and 48.6% for T Left.

Each step of TGAIN is required to be between 1.7% and 3.7% above the preceeding lower gain value. Measured gain steps on OLS #14 ranged from 1.85% to 3.47%, within specification.

COMMANDABLE T CHANNEL LEVEL (3.2.1.1.4.5.5)

The T Channel Commandable Level is also exercised in test 6x8x2.ST. The specification requires that TLEVEL be variable over at least a 14° range in steps of 1.1°K or less. Measured results during the original OLS 14 testing were 14.96° range and step sizes between 0.942°K to 1.049°K worst-case; all within specification.

2.4 RADIOMETRIC ACCURACY, Cont'd

2.4.6 A/D Conversions & Algorithms (3.2.1.1.4.6.2 & 3.2.1.1.4.6.3)

DMSS-OLS-300 specifies that the Stored Smooth Algorithm accuracy with an ideal A/D shall be verified by analysis. OLS 5D-3 System Summary Report contains the analysis which shows that the LS % Full Scale Deviation does meet the specification. The results are summarized below:

Mode	% Full Scale Deviation	Analysis
LS	$\leq \pm 2.2\%$	-1.70%, +2.09%
TS	$\leq \pm 0.4\%$	$\pm 0.39\%$

The Actual A/D Conversion Radiometric Accuracy was measured in system test 6x10.ST. The worst-case results from the OLS #14 Thermal Vacuum runs (Cold Limit & Orbit Nominal) are summarized below:

A/D	BSL SLOPE (% DEV FROM IDEAL)	BSL OFFSET (% OF FULL SCALE)	RMS DEV FROM BSL (% OF FULL SCALE)
SDF-L PRIM	-0.24	+0.04	0.05
RED	-0.10	+0.04	0.05
SDF-T PRIM	-0.29	-0.24	0.10
RED	+0.21	+0.14	0.12
RTD-F PRIM	-0.25	-0.04	0.04
RED	+0.13	-0.04	0.04
SPEC	± 1.0	± 1.0	0.50
RTD-S PRIM	-0.18	-0.06	0.03
RED	-0.20	-0.04	0.05
SDS-L PRIM	-0.14	-0.16	0.04
RED	-0.29	-0.08	0.03
SDS-T PRIM	-0.21	-0.38	0.05
RED	-0.20	-0.24	0.04
SPEC	± 0.5	± 0.5	0.25

2.5 RADIOMETRIC RESOLUTION (3.2.1.1.5 et al.)

DMSS-OLS-300 apportions the Radiometric Resolution verification between Test and Analysis.

The Fine and Direct Smoothed Radiometric Resolution (para. 3.2.1.1.5.1), as well as the Stored Smoothed A/D Converter Error (para. 3.2.1.1.5.2), were measured in System Test 6x10.ST during the OLS #14 vacuum runs and are tabulated below:

A/D		PEAK DEV FROM BSL (% OF FULL SCALE)	SPEC
SDF-L	PRIM	-0.07	±0.8%
	RED	+0.09	
SDF-T	PRIM	+0.22	±0.8%
	RED	+0.24	
RTD-F	PRIM	-0.07	±0.8%
	RED	+0.07	
RTD-S	PRIM	+0.06	±0.25%
	RED	-0.07	
SDS-L	PRIM	+0.07	±0.5%
	RED	+0.08	
SDS-T	PRIM	+0.10	±0.5%
	RED	-0.08	

The Stored Smoothed Algorithms Resolution with Ideal A/D (para. 3.2.1.1.5.2.1) are verified by Analysis in OLS 5D-3 System Summary Report, and are summarized in Table 2.5.1.

TABLE 2.5-1
 Stored Smoothed Algorithms
 Resolution With Ideal A/D

<u>ITEM</u>	<u>SPEC</u>	<u>ACTUAL</u>
Accuracy		
LS Algorithm	<±2.2%	-1.70% +2.09%
TS Algorithm	<±0.4%	±0.39%
Resolution		
LS Algorithm	<1.6%	1.57%
TS Algorithm		
Population 1 Density Quantization	25% <0.8%	25% 0.78%
Population 2 Density Quantization	75% <0.4%	75% 0.39%
Population Distribtution	Uniform	Uniform
Quantization Capability	0.4%	0.4%

2.6 NOISE

2.6.1 T Channel Noise (3.2.1.1.6.1)

The T Channel noise equivalent temperature difference (NETD) is specified between 210K and 310K, although the T channel responsivity extends down to 190K.

The NETD is measured during T Channel DC response tests in the Thermal Vacuum Chamber.

The Channel views a stable blackbody target at 210K (worst-case noise). The channel output is sampled & the noise (std. deviation) of 2000 samples is converted to NETD using the following formula:

$$\text{NETD} = [\text{Avg Noise in Volts} * 24 \text{ } \circ/\text{Volt}] * 1.074 \text{ (the shaper slope correction)}$$

The OLS #14 NETD is in-spec. The noise in the T Right segment is 2.1% larger than in the T Left segment.

	<u>TF</u>	<u>TS</u>	<u>TS Fallback</u>
SPEC	2.2•K	0.90•K	1.3•K
Worst-Case Measured NETD	1.22•K	0.47•K	0.67•K
Worst-Case Average NETD	1.03•K	0.40•K	0.55•K

ATTACHMENT: Table 2.6.1-1 OLS #14 NETD

Table 2.6.1-1

OLS #14 PRIMARY SIDE NETD										
DATE	TG				Noise mV					
	SSS	M1	R/L	TL	FINE RGT	SMOOTH RGT	FINE MID	SMOOTH MID	FINE LFT	SMOOTH LFT
2/14/93	5	-8	3/3	13	39.15	20.19	26.83	15.06	38.24	19.79
2/19/93	5	12	3/3	11	37.40	19.56	25.51	13.99	37.55	19.49
2/23/93	12	15	3/3	11	38.20	20.61	25.23	14.26	36.25	18.89
2/25/93	-3	-11	3/3	14	41.09	22.45	27.47	16.50	39.15	19.29
2/28/93	12	15	3/3	11	37.26	21.10	25.11	14.60	36.15	19.12
3/2/93	-3	-11	3/3	14	40.10	23.06	28.08	16.18	38.35	19.74
3/16/93	5	-8	3/3	13	37.45	20.10	26.73	14.94	38.12	20.02
3/17/93	5	12	3/3	10	37.02	20.06	25.24	14.14	36.23	19.22
3/18/93	7	12	3/3	10	36.98	19.29	25.61	13.92	36.61	19.59
3/24/93	3	-8	3/3	13	38.62	20.60	26.67	14.79	37.80	20.32
3/29/93	5	-8	3/3	13	37.58	20.91	26.61	15.80	37.70	20.26
AVERAGE NETD					38.26	20.72	26.28	14.93	37.47	19.61
					0.918	0.497	0.631	0.358	0.899	0.471
NETD Correction for Shaper Slope**					0.986	0.534	0.677	0.385	0.966	0.506

* Worst Case Measured

** Shaper Slope Correction Factor = 1.074

Table 2.6.1-1

OLS #14 REDUNDANT SIDE NETD										
DATE	TG				Noise mV					
	SSS	M1	R/L	TL	FINE RGT	SMOOTH RGT	FINE MID	SMOOTH MID	FINE LFT	SMOOTH LFT
2/12/93	5	-8	5/5	11	46.13	25.10	33.19	17.80	46.90*	25.18*
2/18/93	5	+12	3/3	5	47.26*	25.98*	33.66*	18.21*	45.06	23.38
3/16/93	5	-8	3/3	14	37.07	19.98	25.39	14.22	38.24	19.11
3/17/93	5	+12	3/3	11	36.21	19.02	24.54	13.55	34.97	17.93
3/18/93	7	12	3/3	10	37.20	18.93	25.07	13.93	36.53	18.89
3/24/93	3	-8	3/3	13	38.18	20.09	26.19	15.19	37.74	20.46
3/29/93	5	-8	3/3	13	38.49	20.72	27.18	15.02	36.50	20.49
AVERAGE NETD					40.08	21.40	27.89	15.42	39.42	20.78
					0.961	0.514	0.669	0.370	0.946	0.499
NETD Correction for Shaper Slope**					1.033	0.552	0.719	0.397	1.016	0.536
Worst Case Corrected NETD					1.218	0.670	0.868	0.469	1.209	0.649

*Worst Case Measured

**Shaper Slope Correction Factor = 1.074

2.6 NOISE (Cont'd)

2.6.2 L-Channel Noise (Day) (3.2.1.1.6.2)

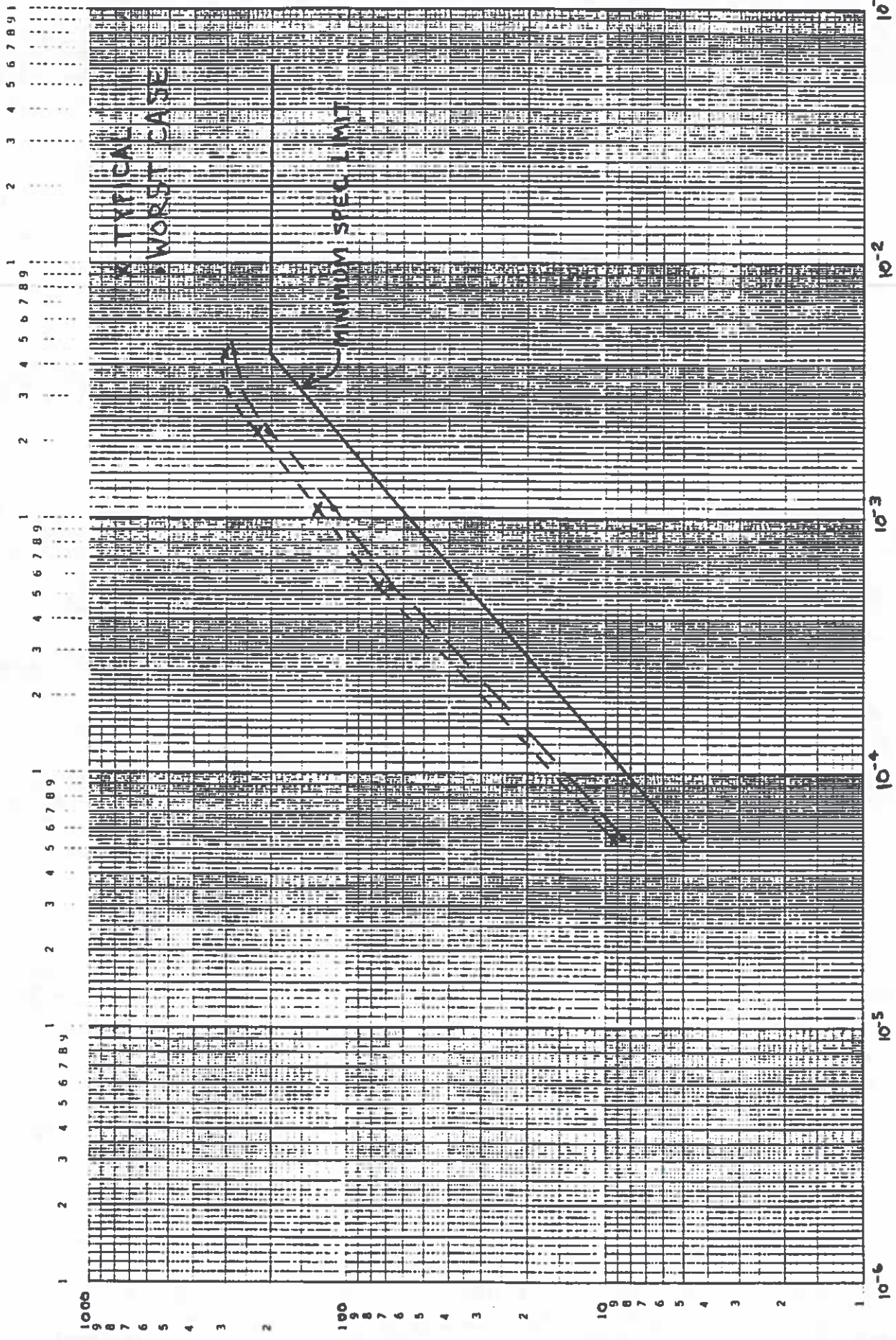
The L Channel Noise is measured using the calibrated Variable Uniform Light Source (VULS). Dark noise is measured in test 6x3x1.ST and shot noise is measured in 6x3x5.ST.

The OLS #14 HRD is in-spec for the entire range of illumination. Worst-case HRD SNR exceeds the specification.

In summary:

	<u>SNR</u>		
<u>LIGHT LEVEL</u>	<u>SPEC</u>	<u>WORST CASE MEASURED</u>	<u>AVERAGE</u>
5.5×10^{-5}	5	8.6	9.2
5.5×10^{-4}	34.8	68.6	74.9
1.1×10^{-3}	62.3	122.9	133.2
2.2×10^{-3}	112	203.5	213.6
4.4×10^{-3}	200	272.6	300.5

ATTACHMENT: OLS #14 HRD Channel SNR Graph



SOLAR ILLUMINATION
WATTS/CM²-SR

SNR
2-164

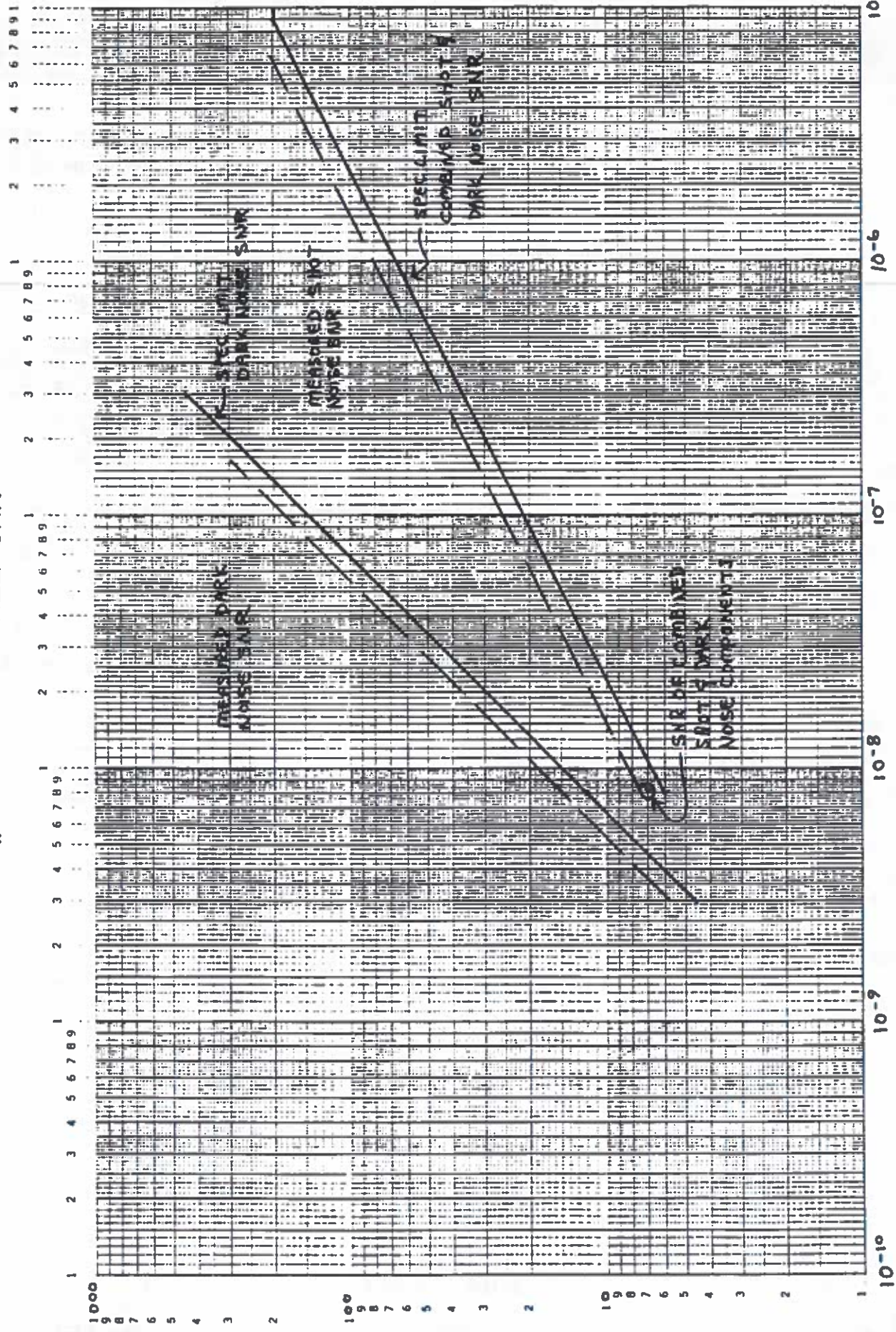
2.6 NOISE (Cont'd)

2.6.3 L Channel Noise (Night) 3.2.1.1.6.3)

The PMT dark noise is measured in all environments in Tests 6x3x1.ST, 6x3x2.ST and 6x3x4.ST. The SNR is calculated from the measured noise (std. deviation of multiple voltage samples) vs. light level and compared against spec values.

The minimum (worst case) OLS #14 PMT Shot Noise SNR was 7.6 at 8.0×10^{-9} watts/cm²-sr. The worst case measured Dark Noise SNR was also better than the specification requirement (15.6 vs. spec of 12). The worst case combined PMT shot noise and dark noise SNR from retest is 6.98 calculated as $SNR = 1 / \sqrt{1/(SNR \text{ dark})^2 + 1/(SNR \text{ shot})^2}$ vs. a specification limit of 6.0.

ATTACHMENT: OLS #14 PMT channel SNR graph.



SOLAR ILLUMINATION
WATTS/CM²-SR

2.6 NOISE (Cont'd)

2.6.4 Dark Current (3.2.1.1.6.4)

The Dark Current (the PMT noise with no signal input) is determined from the graph of PMT SNR in paragraph 2.6.3. The Dark Noise SNR is calculated from data gathered during PMT Smoothed Noise measurements. These measurements are made in Test 6x3x1.ST during Thermal Vacuum testing. For the OLS #14 bearing retrofit retest, the average Dark Noise SNR of 6 measurements at 8×10^{-9} watts/cm²-SR is 16.1, or 37.3% of the noise corresponding to an SNR of 6. The MINIMUM Dark Noise SNR measured at 8×10^{-9} watts/cm²-SR was 15.6, or 38.5% of the noise corresponding to a SNR of 6. This is well within the spec requirement for the dark current to be 50% or less of the noise corresponding to an SNR of 6.00.

2.6 NOISE, (Cont'd)

2.6.5 Stability (3.2.1.1.6.5) (L - Channel (night))

The OLS 5D-3 System Summary Report contains the analysis for this spec requirement.

The loss in sensitivity after 3 years on orbit of the PMT channel will be < 23%. This would require 2.27dB change in VDGA gain to compensate and over 6dB of VDGA gain is available.

ATTACHMENT: None.

2.6 NOISE (Cont'd)

2.6.6 Along-Track Noise Integration (3.2..1.1.6.6)

OLS 5D-3 System Summary Report contains the analysis which concludes that the OLS 5D-3 algorithm is consistently above 0.6 times the SNR resulting from perfect integration with 8 bit A/D for T Channel; and above $1/\sqrt{2}$ times the SNR resulting from perfect integration with 6-bit A/D for L channel. Therefore, the Along-Track Noise Integration is in-spec.

ATTACHMENT: None.

2.6 NOISE. (Cont'd)

2.6.7 Glare Suppression (3.2.1.1.6.7)

OLS 5D-3 System Summary Report contains the analysis which verifies that the OLS does provide effective protection against solar glare for sun angles between 75° and 95°.

ATTACHMENTS: None.

2.7 SURVIVABILITY (3.2.7)

The OLS 5D-3 System Summary Report contains calculations of survivability. See BVS 2353 (Verification of Survivability Requirements) for further details.

ATTACHMENTS: None.

2.8 SCAN ANGLE (3.2.1.1.8)

Tests 6x7x1.ST and 6x7x3.ST (End of Scan Vignette for HRD & T channels respectively) measure the delphi number at which 1% vignetting of scene begins to occur. The measured delphis enable calculation of the altitude needed to obtain contiguous coverage at the equator. The contiguous coverage requirement is based on the average of +Z and -Z scan angles.

For OLS #14, the following results were obtained:

<u>CHANNEL</u>	<u>DELPHI</u>	<u>SCAN ANGLE</u>	<u>CONTIGUOUS COVERAGE ABOVE:</u>	
+Z HRD	+988.0	+55.79°	429.50 n. mi.	430.59 avg.
-Z HRD	-987.0	-55.73°	431.68 n. mi.	
+Z T	+978.0	+55.22°	441.58 n. mi.	435.54 avg.
-Z T	-989.0	-55.85°	429.50 n. mi.	

Thus, both the HRD channel and the T channel meet the requirements for contiguous coverage above 440 naut. mi, since both channels will provide contiguous coverage for all altitudes above 435.54 naut. mi.

ATTACHMENTS: None.

2.9 DATA COLLECTION RATE (3.2.1.1.9)

OLS #14 does scan the field of view at the prescribed 11.88 +/- .12 Hz rate. This parameter is measured in Scanner Functional tests 5x12x1.ST (Primary Side) and 5x12x2.ST (Redundant Side).

The test results are summarized below for all TV tests:

<u>Date</u>	<u>Frequency, Hz</u>	
	<u>Primary</u>	<u>Redundant</u>
02-13-93 Optic Limit	11.91	11.91
03-20-93 Hot Limit	11.89	11.89
03-24-93 Cold Limit	11.91	11.91
03-29-93 Orbit Nom.	11.90	11.90

ATTACHMENTS: None.

2.10 POWER (3.3.1 and 3.3.2)

Both +28V and +5V power is measured and monitored continuously throughout all of the test sequence.

The power required in the 8 Development-Spec-defined modes is tabulated below. The Development Spec Power Profile is measured in test 5x2x11.ST for modes 1 through 8 and 5x2x2.ST for mode 0.

10V power consumption was not tested on OLS #14. The current system test equipment is not capable of monitoring 10V power. Analysis of the components using S/C supplied 10V power indicates that 5D-3 10V power consumption is in spec.

DMSS-OLS-300 limits 28V power consumed for SSS thermal control to 23 watts maximum. SSS heater power consumption was not measured on OLS #14. Analysis of the heater resistances and tolerances indicates that 5D-3 SSS heater power consumption is in spec.

OLS #14 28V power consumption is in spec for all modes in the primary and redundant configurations. Fallback (dual power) configuration power consumption is also in spec. See the attached table for further details.

ATTACHMENTS: OLS #14 Power Profile

OLS #14 POWER PROFILE

SINGLE POWER				DUAL POWER	
28V MODE/LIMIT	TV +5/-8 03-30-93	TV HOT LIMIT 03-17-93	TV COLD LIMIT 03-24-93	28V LIMIT	**WORST CASE (CALCULATED)
0 88W	55	55	55	131W	89
1 105W	83	85	83	148W	119
2 116W	90	92	90	159W	126
3 125W	96	98	96	168W	132
4 157W	131	134	131	200W	168
5 167W	138	142	138	210W	176
6 198W	166	170	165	241W	204
7 207W	178	182	177	250W	216
8 218W	186	189	184	261W	223
5V MODE/LIMIT					
0 4.8W	3	3	3		
1 4.8W	3	4	3		
2 4.8W	3	4	3		
3 4.8W	3	4	3		
4 4.8W	3	4	3		
5 4.8W	3	4	3		
6 4.8W	3	4	3		

2.11 MASS

2.11.1 Total Mass (3.4.1)

The weight and center of gravity of OLS #14 components were not measured as part of bearing retrofit. Weight data taken on 3-03-88 during the original OLS 14 selloff is provided for reference. The tape recorder and encrypter serial numbers are those belonging to the system at OLS #14 sell-off and may change.

All Westinghouse furnished parts meet their center of gravity specification limits and their maximum specified weight allocation. All Typical encrypters exceed the spec limit in center-of-gravity. The encrypters are GFE to WEC and their weight and C.G. are not controlled by WEC.

The total weight of the OLS #14 AVE is 292.52 pounds, (less BBX's, but including GSSA/DOC & Test Cable), vs. a spec limit of 300 pounds.

ATTACHMENT: OLS #14 Weight and Center-of-Gravity Tables

TABLE 1

WESTINGHOUSE FURNISHED PARTS SUPPLIED WITH OLS 14 SYSTEM
SUMMARY OF WEIGHT AND CENTER GRAVITY

UNIT	UNIT SER. NO.	X			Y			Z			WEIGHT			
		SPEC	MPR	ACT	SPEC	MPR	ACT	SPEC	MPR	ACT	MAX* SPEC	MPR** W/O CONT	MPR** W CONT	ACT
SSS	5009	1.8±.5	1.8±.5	1.74	6.2±.5	6.2±.5	6.19	0.7±.5	0.7±.6	0.69	59.0	54.92	56.02	55.08
SPS	5009	3.0±.5	3.0±.5	2.90	13.8±1.0	13.8±1.0	13.60	8.6±.8	8.6±.8	8.50	70.0	67.60	68.95	69.06
SPU	5009	3.0±.5	3.0±.5	2.92	6.6±.5	6.6±.5	6.72	6.0±.5	6.0±.5	5.81	18.0	17.00	17.34	17.13
PSU	5009	2.3±.5	2.3±.5	2.81	7.0±.6	7.0±.6	6.69	7.2±.5	7.2±.5	7.12	27.0	25.93	26.45	26.63
DSU	5009	1.2±.25	1.2±.25	1.27	4.0±.5	4.0±.5	4.27	3.0±.5	3.0±.5	2.73	4.0	3.47	3.53	3.38
GSSA/DOC	5009	4.2±.5	4.2±.5	4.33	+0.1±.3	+0.1±.3	0.36	2.4±.5	2.4±.5	2.77	9.0	7.83	7.99	7.83
PR1	044	3.45±.25	3.45±.25	3.43	6.36±.25	6.36±.25	6.30	4.23±.25	4.23±.25	4.34	22.75	21.14	21.57	21.20
PR2	045	3.45±.25	3.45±.25	3.37	6.36±.25	6.36±.25	6.27	4.23±.25	4.23±.25	4.28	22.75	21.14	21.56	21.50
PR3	056	3.45±.25	3.45±.25	3.29	6.36±.25	6.36±.25	6.33	4.23±.25	4.23±.25	4.31	22.75	21.14	21.56	21.15
PR4	057	3.45±.25	3.45±.25	3.21	6.36±.25	6.36±.25	6.19	4.23±.25	4.23±.25	4.28	22.75	21.14	21.56	21.15
CABLES	(1)	-	-	-	-	-	-	-	-	-	32.0	24.31	24.79	22.41
TEST														
CABLE	(2)	-	-	-	-	-	-	-	-	-	6.0	6.0	6.0	6.0
TOTAL WEIGHT											298	286.78	292.36	292.52

*DMSS-OLS-300, SCH 011, 20 Nov. 1987
**503 Mass Properties Report, 18 Nov. 1988

(1) SERIAL NUMBERS ARE AS RECORDED ON DATA SHEET
(2) A MASS ALLOCATION OF 6 LBS. HAS BEEN ASSIGNED FOR TEST CABLE FROM THE TOTAL OLS MASS ALLOCATION.
TEST CABLE IS PROVIDED AND CONTROLLED BY THE SPACECRAFT INTEGRATOR.

GOVERNMENT FURNISHED PARTS SUPPLIED WITH OLS 14 SYSTEM
SUMMARY OF WEIGHT AND CENTER GRAVITY

UNIT	UNIT SER. NO.	X			Y			Z			WEIGHT			
		SPEC	MPR	ACT	SPEC	MPR	ACT	SPEC	MPR	ACT	MAX* SPEC	MPR** W/O CONT	MPR** W CONT	ACT
BB1	026	1.8±.1	1.8±.1	1.80	2.7±.1	2.7±.1	2.79	2.2±.1	2.2±.1	2.17	3.67	3.34	3.59	3.44
BB2	027	1.8±.1	1.8±.1	1.86	2.7±.1	2.7±.1	2.79	2.2±.1	2.2±.1	2.12	3.67	3.33	3.59	3.50
BB3	028	1.8±.1	1.8±.1	1.86	2.7±.1	2.7±.1	2.79	2.2±.1	2.2±.1	2.12	3.66	3.33	3.59	3.50
TOTAL WEIGHT											11.00	10.00	10.77	10.44

*DHSS-OLS-300, SCN 011, 20 Nov. 1987
 **503 Mass Properties Report, 18 Nov. 1988

2.11 MASS (Cont'd)

2.11.2 Component Mass (3.4.2, 3.4.3)

The mass of the individual components of the OLS #14 AVE are tabulated below.

<u>Component</u>	<u>Spec</u>	<u>Measured</u>
SSS	59.0	55.08
SPS	70.0	69.06
SPU	18.0	17.13
PSU	27.0	26.63
OSU	4.0	3.38
GSSA/DOC	9.00	7.83
PR1	22.75	21.20
PR2	22.75	21.50
PR3	22.75	21.15
PR4	22.75	21.15
BB1	3.67	3.44
BB2	3.67	3.50
BB3	3.66	3.50
Cables	32.00	22.41

The cable figure does not include Special Sensor cables which are not supplied by WEC.

2.12 COOLER TRANSIENT MARGIN (3.2.2.5)

The T channel cone cooler transient cooldown cooling capacity margin is tested by the file 7x7.ST. An external power supply provides 1/2 Watt of heating to the outer stage of the cooler during a normal cooldown. On OLS #14 bearing retrofit, cone cooler S/N 026 successfully reached its operating set-point with 1/2 watt of external power applied, demonstrating the required margin.

ATTACHMENTS: None

2.13 DESIGN FEATURES

The following design features of the 5D-3 OLS are addressed in the analyses. The analyses are contained in the OLS 5D-3 System Summary Report. The requirements of the Design Features are met or exceeded in each category.

<u>SUBJECT</u>	<u>SPEC. PARA.</u>
DESIGN INTERFACES	(3.1.2)
RELIABILITY	(3.2.3)
WEAROUT/CONSUMPTION	(3.2.3.3)
STORAGE	(3.2.3.4)
CONTAMINATION CONTROL	(3.2.3.5)
CORROSION OF METAL PARTS	(3.2.3.6)
MAINTAINABILITY	(3.2.4)
INTERCHANGEABILITY	(3.3.5)

ATTACHMENTS: None.

2.14 REDUNDANT AND FALLBACK SUBSYSTEMS (3.2.3.1 & 3.2.3.2)

Paragraph 3.2.3.1 of the development spec requires the OLS to incorporate the following redundant subsystems:

- (1) Along-Scan Gain Control
- (2) Main Bus Power Supplies
- (3) Data Processors
- (4) Memories
- (5) I/O Interfaces
 - a. Bus Controls
 - b. I/O Controls
 - c. S/C Interfaces
 - d. WOW/Flutter Signals/Clock Drivers
 - e. Drive Motor Controls
 - f. Sensor Controls
 - g. Gain Controls
 - h. Encoder Processors
- (6) Data Channels
 - a. Smooth Video Filters
 - b. Fine Video Filters
 - c. T-Channel Post Amplifiers
 - d. SDS Channel
 - e. SDF Channel
 - f. RTD Channel
 - g. Special Sensor Processors
 - h. Output Data Multiplexers
- (7) Output Switching Unit Oscillator and Clock Circuits
- (8) Digital Tape Recorders - (Three of Four Required)

(9) Output Data Channels - (Three of Four Required)

Paragraph 3.2.3.2 requires the OLS to incorporate the following fallback subsystems:

- (1) IMC Shut-Off Mode.
- (2) HRD - Detector Single Segment Select, with associated electronics.
- (3) T-Detector Single Segment Select, with associated electronics.
- (4) Digital Generation of Delphi Scanner Clock.
- (5) PMT Shut-Off Mode

Redundant and fallback subsystems are verified by test during the normal test flow either by repetition of the relevant test on the alternate subsystem (in the case of a redundant subsystem) or by tests designed to verify the specific subsystem (in the case of a fallback subsystem). In each case, any out-of-specs or anomalies are reported as part of the relevant Test Report paragraph.

2.15 ENVIRONMENT

The following environmental requirements are addressed in analyses and are contained in the OLS 5D-3 System Summary Report. The Thermal Vacuum (except CHA), Random Vibration, and shock requirements are verified by the successful completion of the approved Test Procedure.

The 5D-3 Environmental requirements of the Development Specification are met or exceeded by the OLS #14 AVE.

<u>SUBJECT</u>	<u>SPEC. PARA</u>
GROUND ENVIRONMENT	20.2.1
LAUNCH ENVIRONMENT	20.2.2
THERMAL VACUUM (CHA)	20.2.2.1
ACCELERATION	20.2.2.5
CHARGED PARTICLE ENVIRONMENT	20.2.2.6
LAUNCH PRESSURE PROFILE	20.2.2.8
ACCOUSTIC FIELD	20.2.2.9
TRANSPORATION & HANDLING ENVIRONMENT	20.2.3

ATTACHMENTS: None.

3.0 INTERFACE SPECIFICATION REQUIREMENTS

Electrical Interface parameters are measured in the OLS Detailed Electrical Test (T927989) and the OLS Special Sensor Detailed Electrical Test (T927992). These tests demonstrated conformance with all applicable Interface Specification requirements. The only Interface related system measurements that vary significantly from system to system are the SSS Alignment axes which are included here.

3.1 SSS ALIGNMENT AXES

The OLS #14 SSS Reference Plane axes are within the specification allowances. The results of system test and calculation are given below. The designations are those in Interface Spec IS-YD-810, para. 3.2.7ff.

SECONDARY REFERENCE AXES TO PRIMARY AXES

$$X_{R-P} = 0.708 \text{ mrad} = 146 \text{ arc sec}$$

$$Y_{R-P} = 0.621 \text{ mrad} = 128 \text{ arc sec}$$

$$Z_{R-P} = 0.698 \text{ mrad} = 144 \text{ arc sec}$$

SECONDARY REFERENCES AXES TO MOUNTING (INTERFACE) AXES

$$X_{R-M} = 0.727 \text{ mrad} = 150 \text{ arc sec}$$

$$Y_{R-M} = 0.732 \text{ mrad} = 151 \text{ arc sec}$$

$$Z_{R-M} = 0.509 \text{ mrad} = 105 \text{ arc sec}$$

These are within the specification limits of 600 arc seconds. The Mounting (Interface) Axes to Primary axes are also calculated, using the computer program REFPLN and are given below:

$$X_{M-P} = 0.257 \text{ mrad} = 53 \text{ arc sec}$$

$$Y_{M-P} = 0.189 \text{ mrad} = 39 \text{ arc sec}$$

$$Z_{M-P} = 0.199 \text{ mrad} = 41 \text{ arc sec}$$

These are within the specification limits of 120 arc seconds.

ORIGINAL

OLS - 14

2/8/93 TRH

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BVS 2579

DATE 08 February 1993

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REVISION 0

BEARING RETROFIT
AND RETEST PLAN
FOR OLS 12 THRU 16

Support and Services Contract No. F04701-90-C-0028

Prepared for

UNITED STATES AIR FORCE
Headquarters, Space Systems Division
Los Angeles, California

Prepared by

WESTINGHOUSE ELECTRIC CORPORATION
Defense and Electronics Center
Baltimore, Maryland

BVS No.: 2579

REVISION SHEET & NOTES PAGE

NOTICE: Unless otherwise instructed, the marked-up pages showing actual changes incorporated in a new Rev. will be maintained in the BVS Master File for future reference and the remainder of the document will be discarded when the new Rev. is filed.

REVISION LETTER	REVISION DATE	AFFECTED PAGES	REVISION MADE BY
A	4/6/90	1,2, *3-6, 7-11, *12, 13-16, *17-20, 21, 22	J. SMU
B	8/7/90	1, 2, 16, 16a	D. Q'EE
C	8/26/90	1, 2, 16b, 17	J. S. J
D	9/21/90	1, 2, 5, 6, 18, 21	G. POLI
E	2/20/91	1, 2, 17-23	R. BARI
F	2/28/91	1, 2, 17	R. BARI
G	5/20/91	22, 23	M. BARI
H	5/22/91	22	M. BARI
J	6/28/91	1-3, 22-23	SCILIP
K	8/29/91	1-2, 18, 22-23	SCILIP
L	11/05/91	1, 2, 23, 24, 25	G. POLI
M	7/20/92	1, 2, 12-14	P. KIE
N	9/2/92	1, 2, 16b, 16c, 17, 18, 23	J. S. J
P	01/27/93	1, 2, 16d	J. SMU
Q	02/05/93	1, 2, 22, 22a	SCILIP

(* INDICATES PAGE # IS ONLY CHANGE TO THAT PAGE)

Notes:

WP51\JSm01.d1

3.2 The modification is accomplished by working revision notice G9.

This consists of replacing the two bearing pairs in oscillator Assembly 623R765.

Special instructions have been written to supplement the RNs and describe the mechanical operations necessary to retrofit scale bearings after an SSS has been fully assembled, see 9TA9354.

3.3 For those systems in the field, return the system to WEC, Baltimore for rework and retest. Follow all applicable handling procedures including Program Directives 024, 026 and 027. As an option, the PSU and SSS only may be returned if another system is available to support the retest effort.

3.4 INCOMING INSPECTION AND SYSTEM TESTS.

<u>OPERATION</u>	<u>VERIFICATION</u>	<u>DATE</u>
Unpack	<u>GJS</u>	<u>8-5-91</u>
Record serial nos. of rcvd. units:		
SSS 640R800G08 S/N <u>N/A</u>		
or 758R750G0 <u>2</u> S/N <u>5009</u>		
PSU 758R050G0 <u>4</u> S/N <u>5009</u>		
SPS 651R390G0 <u>2</u> S/N <u>5008</u>		
SPU 758R040G0 <u>3</u> S/N <u>5009</u>		
OSU 640R960G0 <u>4</u> S/N <u>5009</u>		
<u>9RA5255H02 S/N 502; 9RA5255H04 S/N 501; 9RA5255H06</u> <u>S/N 503; 9RA5255H04 S/N 501; 9RA5255H10 S/N 503</u> <u>9RA5255H11 S/N 501; 9RA5255H02 S/N 501; 644R327G01</u> <u>THRU 603, 644R328G01-606, 644R329G01-605</u>		
	<u>GJS</u>	<u>8-5-91</u>
Attach copy of incoming DD1149 to this BVS for control purposes	<u>GJS</u>	<u>8-5-91</u>
WEC Receiving Inspection	<u>[Stamp]</u>	<u>8-6-91</u>
AFPRO Inspection	<u>[Signature]</u> <u>[Stamp]</u>	<u>6/11/92</u>

Baseline Electrical Tests - Deleted

3.5 OPTICAL ALIGNMENT BASELINE MEASUREMENTS

Prior to bearing retrofit certain optical tests must be performed in order to accumulate baseline data with which to compare readings taken after the work is completed. This is necessary so that alignment integrity can be verified after the SSS has been partially disassembled and reassembled. These tests will be the same as some of the tests performed in 9RA3681 "Assembly and alignment procedure SSS assembly". However, there may be slight differences in technique because the assembly status of the SSS will not be exactly the same as in the normal building sequence. The steps herein, then, will be excerpts from that procedure and all step numbers referred to will be taken from 9RA3681.

Data should be noted in the applicable flight log book and used for post retrofit alignment comparisons.

In order to perform the necessary tests, the PMT and HRD detector must be removed. The spring assemblies must also be tied in order to permit positioning of the telescope.

STEPS FROM 9RA3681 TO BE PERFORMED

<u>STEPS</u>		<u>VERIFICATION</u>	<u>DATE</u>
8.0	"Adjustment of optical alignment, test and integration facility" - prepares the facility for required tests.	<u>with</u>	<u>15 Jul</u>
9.0	"SSS assembly mounting procedure" - Mounts the SSS to the test facility reference interface for testing.	<u>with</u>	<u>15 Jul</u>
11	"Alignment of the oscillating assembly rotating axis with the Moore table axis." - positions the SSS for optical measurements.	<u>with</u>	<u>15 Jul</u>
12	"Mounting interface alignment measurements" - Determines SSS reference axis position in relation to the OATIF mirrors/SSS mounting interface.	<u>with</u>	<u>15 Jul</u>
—	A test will be performed to determine the position of the HRD detector prior to removal as follows: Clamp a Gaertner bench microscope to the T/T table aligning the microscope reticle with the T/T reticle. Observing the HRD detector through the microscope, center the reticle on at least 2 corners of each segment of the detector. Note the T/T Y and Z axis positions for each point observed.	<u>with</u>	<u>15 Jul</u>
18	"Oscillating assembly transmission test" - Determines % transmission of telescope prior to retrofit.	<u>with</u>	<u>16 Jul</u>
—	Inspect mirror M1 to determine if a scatter test should be performed. If, on inspection, M1 appears very dirty perform a scatter test per step 19 of 9RA3681.	<u>no scatter test reqd. 16 Jul</u>	<u>16 Jul</u>

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STEPS FROM 9RA3681 TO BE PERFORMED

<u>STEPS</u>		<u>VERIFICATION</u>	<u>DATE</u>
20.4	"HRD detector alignment check" - checks alignment of the HRD in relation to the ORA field splitter. Illumination via the PMT light as described in step 20.4.5 should not be necessary. If the light from the T/T point source is insufficient to view the HRD the PMT must be removed prior to performing step 20.4.	<u>Wdh</u>	<u>16 Jun 92</u>
—	Before proceeding remove the HRD and PMT if still installed. Mount the PMT and HRD reticles.	<u>Wdh</u>	<u>16 Jun 92</u>
14.2 to 14.3	"M1 centering test" - verifies that the optical beam is centered on M1 prior to retrofit.	<u>Wdh</u>	<u>16 Jun 92</u>
16.1.1 to 16.1.4, 16.1.10, & 16.1.19 to 16.1.26	"T-Cal alignment..." Determines pre-retrofit T-Cal end of scan position. Make no adjustments.	<u>Wdh</u>	<u>16 Jun 92</u>
16.2.1 to 16.2.4, 16.2.11, 16.2.20 to 16.2.27, & 16.2.29	"T-Clamp alignment..." - Determines pre-retrofit T-Clamp end of scan position. Make no adjustments.	<u>Wdh</u>	<u>17 Jun 92</u>
15.2.2 to 15.2.7 & 15.2.16 to 15.2.18	"Encoder optics alignment" - Determines max. and min. clock voltages. The facet from 15.1.15 referred to in steps 15.2.6 and 15.2.7 will be assumed to be facet 8.	<u>Wdh</u>	<u>17 Jun 92</u>
15.2.19.6 to 15.2.19.8	Defines minimum allowable voltages and angular displacement on the faceted ring.	<u>Wdh</u>	<u>17 Jun 92</u>

STEPS FROM 9RA3681 TO BE PERFORMED

<u>STEPS</u>		<u>VERIFICATION</u>	<u>DATE</u>
15.3.1 to 15.3.6 Note: Make <u>NO</u> adjust- ment in 15.3.6	"Encoder nadir adjustment" - Determines pre-retrofit position of the encoder at the Nadir position W/R to target translator position.	<u>Wdh</u>	<u>17 June</u>
15.3.12	"Encoder nadir alignment error" - Provides a formula for determining nadir alignment error.	<u>Wdh</u>	<u>17 June</u>
15.4.1 to 15.4.26 Omit steps 15.4.18, 15.4.19, 15.4.20 & 15.4.23	"Encoder linearity and signal amplitude measurements." - Determine pre-retrofit position for facets of the encoder W/R to the target translator. In steps 15.4.17, 15.4.22 and 15.4.25 only a sampling of the numbered pulses shall be taken. Sample pulses 15, 60, 97 and 142.	<u>Wdh</u>	<u>17 June</u>
15.5 Note: Make <u>NO</u> adjust- ments in 15.5.12, 15.5.13 or 15.5.16	"Back-up auxiliary encoder alignment." - Determines pre-retrofit electro-optical position of the back-up aux. encoder.	<u>Wdh</u>	<u>18 June</u>
15.6.1 to 15.6.3	"SSS Preparation for primary aux. encoder alignment." - Prepares SSS for testing of primary aux. encoder.	<u>Wdh</u>	<u>18 June</u>
15.9 Note: Make <u>NO</u> adjust- ment in 15.9.13 or 15.9.16	"Primary aux. encoder alignment" - Performs tests to determine electro-optical position of the primary aux. encoder.	<u>Wdh</u>	<u>18 June</u>

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STEPS FROM 9RA3681 TO BE PERFORMED

STEPS

VERIFICATION

DATE

15.1.1 to
15.1.15
Omit step
15.1.14

"Faceted ring angular measurements"
- Determines optical positions of the facets of the polygon ring. In steps 15.1.4, 15.1.8, 15.1.10 and 15.1.13 where the step refers to specific facets, it shall be required to perform the procedure only on a sample of the facets. A facet shall be chosen at the beginning, the middle and as near to the end as can be seen. Perform the procedure on facets 1, 8 and 14 if these are accessible. Step 15.1.14 will be omitted and in step 15.1.15 the facet closest to the mean facet Y axis reading will be assumed to be facet 8.

Weth

18 Dec 92

Cover the HRD and PMT reticles and ORA parts with lens tissue. Data should be noted in the applicable flight log book and used for post retrofit alignment comparisons.

Weth

15 Dec 92

3.6 MECHANICAL OPERATIONS

Perform the operations outlined in 9TA9354. This procedure describes all the mechanical operations necessary to retrofit scanner bearings after an SSS has been fully assembled. This step-by-step procedure includes check-off lines for each operation and inspection point.

For OLS-14 only: Reference paragraph 5.5 of 9TA9354. Replace Rod-413R524H01. Original rod found to have a chipped thread prior to reassembly.

Following completion of the procedure, attach the working copy of 9TA9354 to this BVS.

Verification of completion

GJS 10/27/92

Inspection

10/29/92

3.7 OPTICAL RE-ALIGNMENT

Optical Tests after Bearing Retrofit

After the bearing retrofit certain optical tests from 9RA3681 must be performed both for comparison to baseline tests as well as to ensure the unit is ready for integration tests. The bulk of these post bearing retrofit tests are the same as the optical baseline tests discussed in section 3.5. Record data in the applicable system SSS log book.

In order to perform the necessary tests, the PMT and HRD detector must be removed. The spring assemblies must also be tied in order to permit positioning of the telescope.

STEPS FROM 9RA3681 TO BE PERFORMED

<u>STEPS</u>		<u>VERIFICATION</u>	<u>DATE</u>
8.0	"Adjustment of optical alignment, test and integration facility" - prepares the facility for required tests.	<u>WTH</u>	<u>8/3/92</u>
9.0	"SSS assembly mounting procedure" - Mounts the SSS to the test facility reference interface for testing.	<u>WTH</u>	<u>8/3/92</u>
11	"Alignment of the oscillating assembly rotating axis with the Moore table axis." - positions the SSS for optical measurements.	<u>WTH</u>	<u>8/2/92</u>
12	"Mounting interface alignment measurements" - Determines SSS reference axis position in relation to the OATIF mirrors/SSS mounting interface.	<u>WTH</u>	<u>8/3/92</u>
13.0 to 13.3, 13.5	"M3, M5 and M5 Mask Alignment" - Determines whether adjustments are needed in M3 and M5. (Hopefully M5 will not need alignment. If it does, consult an optical engineer because the ORA must be removed to align M5. Then it will be necessary to realign the ORA when it is reinstalled).	<u>WTH</u>	<u>8/5/92</u>
	For OLS 14 only: Remove RTV potting from M4, adjust M4 to correct orientation of M2 spider image (found displaced at incoming inspection) and repot M4 with RTV.	<u>WTH</u>	<u>8/11/92</u>
18	"Oscillating assembly transmission test" - Determines % transmission of telescope.	<u>WTH</u>	<u>8/7/92</u>

(JSm01.d1)

STEPS FROM 9RA3681 TO BE PERFORMED

<u>STEPS</u>		<u>VERIFICATION</u>	<u>DATE</u>
—	Inspect mirror M1 to determine if a scatter test should be performed. If, on inspection, M1 appears very dirty perform a scatter test per step 19 of 9RA3681.	<u>WCH</u>	<u>8/1/94</u>
20.4	"HRD detector alignment check" - checks alignment of the HRD in relation to the ORA field splitter. Illumination via the PMT light as described in step 20.4.5 should not be necessary. If the light from the T/T point source is insufficient to view the HRD the PMT must be removed prior to performing step 20.4.	<u>WCH</u>	<u>8/18/94</u>
—	Before proceeding remove the HRD and PMT if still installed. Mount the PMT and HRD reticles.	<u>WCH</u>	<u>8/18/94</u>
14.2 to 14.3	"M1 centering test" - verifies that the optical beam is centered on M1.	<u>WCH</u>	<u>8/19/94</u>
16.1.1 to 16.1.4, 16.1.10, & 16.1.19 to 16.1.26	"T-Cal alignment..." Determines T-Cal end of scan position. Make no adjustments.	<u>WCH</u>	<u>8/1/94</u>
	For OLS 14 only: Readjust T-Cal mirror per 9RA3681 except align the T-Cal (-)Z edge with the HRD at encoder pulse (1018-26)(Ref. paragraphs 16.1.14 and 16.1.18).		
16.2.1 to 16.2.4, 16.2.11, 16.2.20 to 16.2.27, & 16.2.29	"T-Clamp alignment..." - Determines T-Clamp end of scan position. Make no adjustments.	<u>WCH</u>	<u>8/1/94</u>

STEPS FROM 9RA3681 TO BE PERFORMED

<u>STEPS</u>		<u>VERIFICATION</u>	<u>DATE</u>
15.2.2 to 15.2.7 & 15.2.16 to 15.2.18	"Encoder optics alignment" - Determines max. and min. clock voltages. The facet from 15.1.15 referred to in steps 15.2.6 and 15.2.7 will be assumed to be facet 8.	<u>W/A</u>	<u>8/12/92</u>
15.2.19.6 to 15.2.19.8	Defines minimum allowable voltages and angular displacement on the faceted ring.	<u>W/A</u>	<u>8/12/92</u>
15.3.1 to 15.3.6 Note: Make <u>NO</u> adjustment in 15.3.6	"Encoder nadir adjustment" - Determines position of the encoder at the Nadir position W/R to target translator position.	<u>W/A</u>	<u>8/12/92</u>
15.3.12	"Encoder nadir alignment error" - Provides a formula for determining nadir alignment error.	<u>W/A</u>	<u>8/13/92</u>
15.4.1 to 15.4.26 Omit steps 15.4.18, 15.4.19, 15.4.20 & 15.4.23	"Encoder linearity and signal amplitude measurements." - Determine position for facets of the encoder W/R to the target translator. In steps 15.4.17, 15.4.22 and 15.4.25 only a sampling of the numbered pulses shall be taken. Sample pulses 15, 60, 97 and 142.	<u>W/A</u>	<u>8/14/92</u>
15.5 Note: Make <u>NO</u> adjustments in 15.5.12, 15.5.13 or 15.5.16	"Back-up auxiliary encoder alignment." - Determines electro-optical position of the back-up aux. encoder.	<u>W/A</u>	<u>8/12/92</u>

STEPS FROM 9RA3681 TO BE PERFORMED

<u>STEPS</u>		<u>VERIFICATION</u>	<u>DATE</u>
15.6.1 to 15.6.3	"SSS Preparation for primary aux. encoder alignment." - Prepares SSS for testing of primary aux. encoder.	<u>W.H.</u>	8/12
15.9 Note: Make <u>NO</u> adjust- ment in 15.9.13 or 15.9.16	"Primary aux. encoder alignment" - Performs tests to determine electro-optical position of the primary aux. encoder.	<u>W.H.</u>	8/13
15.1.1 to 15.1.15 Omit step 15.1.14	"Faceted ring angular measurements" - Determines optical positions of the facets of the polygon ring. In steps 15.1.4, 15.1.8, 15.1.10 and 15.1.13 where the step refers to specific facets, it shall be required to perform the procedure only on a sample of the facets. A facet shall be chosen at the beginning, the middle and as near to the end as can be seen. Perform the procedure on facets 1, 8 and 14 if these are accessible. Step 15.1.14 will be omitted and in step 15.1.15 the facet closest to the mean facet Y axis reading will be assumed to be facet 8.	<u>W.H.</u>	8/13
Reinstall HRD & PMT (OLS-16 see next page, 16a)		<u>W.H.</u>	8/13
—	A test will be performed to determine the position of the HRD detector after reinstallation as follows: Clamp a Gaertner bench microscope to the T/T table aligning the microscope reticle with the T/T reticle. Observing the HRD detector through the microscope, center the reticle on at least 2 corners of each segment of the detector. Note the T/T Y and Z axis positions for each point observed.	<u>W.H.</u>	8/12
Inspection		<u>W.H.</u>	8/12
(JSm01.d1)	-16-		BVS 257

3.7.1 PMT ASSEMBLY SPECTRAL RESPONSE STABILITY CHECK - OLS-16 Only

With the PMT assembly removed during the optical re-alignment, a check of the PMT spectral response will be performed to check spectral stability for any evidence of a shift since the last PMT spectral response made on 07/30/88.

The test will be performed in accordance with T-361A88, test paragraph 9.12 - Spectral Response and Effective Sensitivity.

STEPS TO BE PERFORMED

<u>STEPS</u>	<u>VERIFICATION</u>	<u>DATE</u>
Inspection of PMT (Damage Verification)	4/24/93 BVS N/A	
Install in Transport Case	↑	
Spectral Response Test from T-361A88, Para. 9.12		
Inspect PMT prior to SSS Installation for Damage		
Reinstall PMT on SSS	↓	
Inspection (<u>W</u> & DPRO)	N/A	

3.7.2 THERMAL BLANKET UPGRADE PER ECP-25 - OLS-16 only

In place of the thermal blanket hardware originally installed on OLS-16, install the following oscillating assembly insulation covers and insulation:

	<u>VERIFICATION</u>	<u>DATE</u>
540R561G01	4/24/93 BVS N/A	
540R561G02	↑	
540R562H01		
540R562H02		
540R563H01	↓	
Inspection	N/A	

3.7.2 (Cont'd.)

Install ECP-25 upgraded bracket on the 1A8 HRD/PMT Postamplifier Assembly as follows:

<u>STEPS</u>	<u>VERIFICATION</u>	<u>DATE</u>
Remove the cover assembly, 644R288, from the HRD/PMT Postamplifier, 644R220 located on the SSS. Be careful not to disturb the potentiometer adjustments. It may be necessary to cut the RTV used to stake the potentiometers if it has adhered to the cover.	<i>y/h/s</i> <i>DWS</i> N/A	<i>COLS-16 oh</i>
On the cover assembly, 644R288G01, replace mount 522R838G01 and 432R269G01 with items 16 and 17 respectively on the Thermal Blanket Retrofit drawing, 765R630. (540R564G01 and 540R584G01) Re-mark the cover assembly to 644R288G02.		
Inspect modified cover		
Inspect 644R220, OK to re-install cover, WEC & DPRO		
Install modified cover on 1A8 postamplifier		
Inspection (Note - retest of 644R220 postamplifier not required, will be tested at system level).	N/A	

3.7.3 REPLACEMENT OF PMT AND EST/LMD - OLS-14 only

The PMT and EST/LMD assemblies originally delivered on OLS-14 were transferred to OLS-12 in August of 1991 (with customer approval) to replace a failed PMT assembly. The flight-quality spare PMT assembly 644R909G05, Serial No. 6001, will be installed on OLS-14 as replacement. The EST/LMD from OLS-12 will be re-configured to match the characteristics of the replacement PMT. The PMT will be vibrated in the SSS assembly per paragraph 3.11.

<u>STEPS</u>	<u>VERIFICATION</u>	<u>DATE</u>
PMT S/N 6001 confidence test; 30 day vacuum test per the procedures of T927096, para. 4.13	<i>J.S.</i>	<i>7/26/91</i>
PMT post-vacuum test per T927096, para. 4.14	<i>J.S.</i>	<i>8/14</i>

(JSm01.d1)

-16b-

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3.7.3 (Cont'd.) (OLS-14 only)

STEPS

VERIFICATION

DATE

PMT final electro-optical test at Arpt. Square III per T361A89.

JS for R.L.

10/2/92

Data Review

JS

10/22/92

Inspect PMT

ES

01 23 '92

DPRO

ES

01 23 '92

Modify EST/LMD, 644R219G03, Serial No. 5007 as follows:

Add system-selected resistor
R16 = 825 ohms (9TA8250H89/RNR55C8250FR)

JS

9/14/92

Delete jumper wire from terminal 19 to 21.
Add a jumper wire from terminal 18 to 21.

JS

9/14/92

Circuit-check

ES

9/25/92

Inspect EST/LMD

ES

9/21/92

Test per T927060, 4.10

ES

9/21/92

Data Review

L.S.

9/25/92

Inspect EST/LMD

ES

10/5/92

DPRO

ES

10/23/92

Install EST/LMD, 644R219G03, Serial No. 5007
on OLS-14 SSS.

G.S.







10/27/92

Install PMT assembly, 644R909G05, Serial No. 6001
on OLS-14 SSS.

G.S.

10/27/92

3.7.4 INSPECTION OF TERMINALS on OLS-14 VDGA - OLS-14 ONLY

STEPS	VERIFICATION	DATE
Remove the 1A6 VDGA assembly from the SSS. The 1A23 Solenoid mechanism will also have to be removed for access, and its cable may have to be unbonded from the HRD and/or SSS structure.	<u>BJ.</u>	<u>1/26/93</u>
Remove the cover from the VDGA assembly and inspect the terminals for solder cracks similar to those observed on OLS-18.	<u>BJ.</u> 	<u>1/27/93</u>
Terminal inspection complete	<u>BJ.</u> 	<u>1/27/93</u>
WEC inspection - OK to cover		<u>1/28/93</u>
DPRO		<u>1/28/93</u>
Install cover	<u>BJ.</u>	<u>1/28/93</u>
Final inspection - OK to install		<u>1/28/93</u>
No retest of the VDGA assembly is required. (test will be performed as part of the thermal vacuum ambient testing per paragraph 3.13 of this BVS.)		
Install the VDGA assembly on the SSS. Re-mount the solenoid mechanism and re-bond cables as required.	<u>GJS.</u>	<u>1/29/93</u>
Inspection		<u>1/29/93</u>

STEPS

VERIFICATION

DATE

3.8 SSS TEST PER T927002

Disconnect SSS main cable connector 1A9P2

(N/A) JS

8/28/92

** Perform the test procedures of T927002 including 50 hour bearing confidence test, scanner centering, scanner frequency, and limit switch adjustments if required. It is not necessary to repeat paragraph 4.16, T detector bias current measurement.

JS

9/9/92

Reconnect 1A9P2

MA GJS

10/27/92

Data Review

[Stamp]

9-10-92

Inspection (SSS ready for system test)

10/29/92

*3.8.1 Perform encoder optics ambient functional test per T927002, paragraph 4.12.4 JS TECH 9/9/92 J.S. LESS OPEN DRA 173 Blank PROBL

*3.8.2 Apply additional adhesive to encoder optics assembly per RNs GL54D and GL62D.

DATE 9/12/92 GJS MANUF [Stamp] INSP [Stamp] DPRO

NOTE: Note after 24 hours the SSS may be removed from the handling fixture and installed on the base plate/test block.

*3.8.3 Seven day cure at room temperature.

DATE COMPLETE 9/16/92

*3.8.4 Reinstall cover and torque screws to 4 to 6 in/lbs.

[Signature] MANUF [Stamp] INSP [Stamp]
DATE 9-23-92

*3.8.5 Repeat step 3.8.1

TECH J.S. DATE 9/23/92

*NOTE: For OLS 16 perform this action after completion of paragraph 3.13 of this BVS.

** CORRECT SCANNER FREQUENCY BY REMOVING 0.095" FROM EACH OF THE FOUR FREQUENCY WEIGHTS ON THE INERTIA WHEEL.

(JSm01.d1)

-17-

8-31-92 J. SMUTKO
(GJS 9/10/92)

BVS 2579

ADD
JRH

4X13.ST Scanner Limit Definition

JRH

11/20/9

STEPS

VERIFICATION

DATE

3.9 AMBIENT SYSTEM TESTS

QUICKTESTN.ST

QUICKTESTR.ST

4X9X1.ST

6X2X1.ST

AHC11PT.ST - R

APC11PT.ST - P

AHSFB11PT.ST - R

6X3X1.ST - P

6X3X2.ST - R

6X3X5.ST - R

MHC11PT.ST - R

6X5X1.ST - P

6X7X1.ST

6X7X2.ST

6X9.ST

7X3.ST

229
AIA

AM14A-4

✓	0825	11/10/9
✓	058	11/11/9
✓	016	11/11/9
✓	024	12/1/9
✓	026	11/23/9
✓	RMP	11/24/9
✓	089	11/24/9
✓	089	12/2/9
✓	089	12/1/9
✓	RMP	11/24/9
✓	089	11/24/9
✓	089	11/24/9
✓	089	11/24/9
✓	RMP	11/25/9
✓	089	11/25/9
✓	089	12/1/9

Data Review

For OLS 14 only due to the PMT assembly replacement, add the following tests

4X3X1.ST

4X5X1.ST

4X7X1.ST

MPA11PT.ST

6X6X1.ST

6X6X2.ST

10X1.ST

STDVALTST.ST

5X5X2.ST

SIMFLT1.ST

SIMFLT2.ST

SIMFLT3.ST

SFPDR.ST



SIMFLT4.ST

JRH

✓	JRH	11/17/9
✓	0730	11/19/9
✓	0825	11/23/9
✓	RMP	11/24/9
✓	JRH	11/24/9
✓	089	11/25/9
✓	089	11/25/9
✓	089	11/25/9
✓	RMP	11/25/9
✓	JRH	12/2/9
✓	JRH	12/2/9
✓	JRH	12/3/9
✓	JRH	12/3/9
✓	JRH	12/3/9

(JSm01.d1)

-18-

4X10X1.ST M4' + M4" Mirror Adjustment

JRH

BVS 25-9

12/3/9

STEPS

VERIFICATION

DATE

3.10 THERMAL VACUUM ADJUST

Deleted

3.11 VIBRATION - SSS Only

Inspection per PD045
checkpoint 3a
checkpoint 3b

(4/5/8)

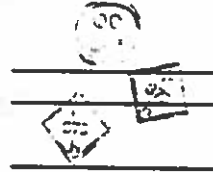
Notify AFPRO ~~DEPRO~~ **DEPRO** PRIOR TO VIBRATION

Vibrate SSS, 3 Axis,
acceptance level
per T927686 para. 3.5

WEC Inspection per PD 055

AFPRO ~~DEPRO~~ **DEPRO** POST VIBRATION

PD 045 Checkpoint #4



12/9/92
12/11/92

12/11/92

POL 243

12/11/92

5/5/92

12/14/92

30

12-15-92

30 345

12/21/92

3.12 POST-VIBRATION, AMBIENT

Perform the test procedures of T927686 paragraph 3.6 except delete paragraph 3.6.5 and in paragraph 3.6.3, only the following test files need to be run:

NOMINAL CONFIGURATION TESTS

	VERIFICATION	DATE
-5X1X1.ST Primary Configuration Power	JAR	12/17/9
5X1X4.ST Load Operational Prog. - Processor C	JAR	12/17/9
5X1X6.ST Initialize Primary Configuration	JAR	12/17/9
-5X2X1.ST Quiescent Mode Power	JAR	12/17/9
-5X2X2.ST Primary Configuration	JAR	12/17/9
-5X2X3.ST Primary Configuration Dual I/O	JAR	12/17/9
-5X2X4.ST Primary Config. Dual Formatter	JAR	12/17/9
-5X2X5.ST Primary Config. Dual Formatters & I/O	JAR	12/17/9
-5X3X1.ST Primary Configuration EST Check	RM	12/17/9
-5X5X1.ST HRD Analog Test	RM	12/18/9
-5X5X2.ST PMT Analog Test	RM	12/18/9
-5X5X3.ST T-Channel Analog Test	RM	12/18/9
-5X6X1.ST PGC	RM	12/18/9
-5X6X2.ST ATGC	RM	12/18/9
-5X6X3.ST ASGC	RM	12/18/9
-5X12X1.ST Scanner Functional	RM	12/18/9

REDUNDANT CONFIGURATION TESTS

		<u>VERIFICATION</u>	<u>DATE</u>
✓	5X1X2.ST Redundant Configuration Power	<u>RM</u>	<u>12/18/92</u>
✓	5X1X5.ST Load Operational Prog. - Processor D	<u>RM</u>	<u>12/18/92</u>
	5X1X7.ST Initialize Redundant Configuration	<u>RM</u>	<u>12/18/92</u>
✓	5X3X2.ST Redundant Configuration EST Check	<u>RM</u>	<u>12/18/92</u>
✓	5X6X4.ST PGC	<u>RM</u>	<u>12/18/92</u>
-	5X6X5.ST ATGC	<u>RM</u>	<u>12/18/92</u>
✓	5X6X6.ST ASGC	<u>RM</u>	<u>12/18/92</u>
✓	5X12X2.ST Scanner Functional	<u>RM</u>	<u>12/18/92</u>
 (3.6.4)			
✓	6.1 IMC HRD A/S - Redundant - AHSF3PTI.ST	<u>JRH</u>	<u>12/18/92</u>
✓	6.1.1 HRD A/S - Redundant - AHSF7PT.ST	<u>JRH</u>	<u>12/18/92</u>
✓	6.1.2 PMT A/S - Primary - APC7PT.ST	<u>RM</u>	<u>12/20/92</u>
✓	6.1.4 Backup Encoder HRD Sync - Redundant - AHSFB9PT.ST	<u>RM</u>	<u>12/20/92</u>
✓	6X2X1.ST L DC Response (must precede 6X3X1.ST)	<u>JRH</u>	<u>12/19/92</u>
✓	6X2X4.ST T Chan Elec DC Response	<u>RM</u>	<u>12/20/92</u>
✓	6X3X1.ST L Chan Dark Noise - Primary	<u>JRH</u>	<u>12/19/92</u>
✓	6.4.1 HRD MTF - Primary - MHA7PT.ST	<u>RM</u>	<u>12/18/92</u>
✓	6.4.2 PMT MTF - Redundant - MPA7PT.ST	<u>JRH</u>	<u>1/5/93</u>
✓	6X4X3A.ST Ambient T MTF - Redundant	<u>RM</u>	<u>12/20/92</u>
✓	6X6X2.ST PMT CAL	<u>JRH</u>	<u>1/5/93</u>
✓	6X7X2.ST 990 Test	<u>JRH</u>	<u>12/19/92</u>
✓	7X5.ST Actuator EST Test	<u>JRH</u>	<u>12/21/92</u>

3.13 Thermal Vacuum Acceptance Test

Perform the Thermal Vacuum Test procedures per T927686, paragraph 3.7. Do not install encrypters, use the BBT simulator box. Attach the system test log sheets to this BVS. *ADD 4X11X1.ST to paragraph*

For OLS14 only:

Add a 0.002" shim at the HRD preamp - relay optics interface.

VERIFICATION	DATE
<i>GJS</i>	<i>2/5/91</i>
<i>OCW</i>	<i>2/2/91</i>

Perform 6x5x1.ST

Have Systems Engineering review the 6x5x1.ST data

<i>JTS</i>	<i>2/1/91</i>
------------	---------------

Based on data analysis determine if:

- a. Shim is left in
- b. Shim must be removed

<i>JTS</i>	<i>2/6/91</i>
------------	---------------

The following tests may be deleted from T927686 for this retest:

In paragraph 3.7.3.1, delete the following test files -

5X4X2	Core Test
5X4X3	Core Test
5X16X2	A/D Test
5X16X4	A/D Test

In paragraph 3.7.3.2 the run of 6X2X2.ST shall be considered a reference run.

Do a T channel electrical adjust by adding test "TSET.ST" to paragraph ~~3.7.3.2~~. *3.7.13.4 JPH*

Delete paragraphs 3.7.3.4 and 3.7.3.5.

Add one day of T stabilization testing at the completion of Optics Limit with ml=-8° by performing the following at approximately 2 hour intervals: *T1 at -63°C T2 at +31°C*

Execute 6X2X3.ST by entering "DSK 6X2X3.ST". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.S".

In paragraphs 3.7.5.1.1, 3.7.10.3, 3.7.11.3, and 3.7.12.2, delete the following test files -

5X2X16	BB Signature
5X4X1,2,3,4	Core Tests
5X8X1,2	DMDM
5X10X1,2,3,4	Output Data Switching
5X14X1,2,3,4	SSP Formatter Tests
5X16X1,2,3,4	A/D Tests

Add 4X9X1.ST to paragraph 3.7.5.1.2.

Add a one day nominal temperature T channel stability test between the two soak cycles by performing the following at approximately 2 hour intervals: $T1$ at -63°C $T2$ at $+37^{\circ}\text{C}$

Execute 6X2X3.ST by entering "DSK 6X2X3.ST". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

Delete paragraph 3.7.12.8.

Add a day at the beginning and a day at the end of the nominal temperature plateau for additional T channel stability by performing the following at approximately 2 hour intervals:

Execute 6X2X3.ST by entering "DSK 6X2X3.ST". When the operator is prompted for the P2S job to be executed, enter "DSK TSTABILITY.ST".

	<u>VERIFICATION</u>	<u>DATE</u>
Verify a minimum of 500 hours of scanner operation with new bearings has been performed in vacuum. Any deficiency should be made up at this time.	JRH	4/26/93
Verify completion of Thermal Vacuum	JRH	4/27/93
3.15 <u>OLS-12 Only:</u> SPS Coax Connector repair on J10 due to defective female contact per NR 20250959.		
- Remove top cover of SPS	N/A	(MFG)
- Remove 2 P.C. boards - A241 & A242	N/A	(MFG)
- Remove 640 R913G01 cable	N/A	(MFG)
- Remove coax connector J10 and replace with new connector	N/A	(MFG)
- Circuit check	N/A	(TES)
- WEC Insp.	N/A	(INSP)
- DPRO Insp.	N/A	(DPRO INS)
- OK to reinstall 913 cable in SPS		
- Reinstall boards A241 & A242		
- OK to cover Insp.		
- WEC Insp.		
- DPRO Insp.		
- Install cover		
- WEC Insp.		
- Reinstall buffer connector		

3.16 Due to male contact pin damage on cables 644R329G02 and G03, replace OLS-12 cables with OLS-14 cables.

3.17 Final Ambient

For OLS-16 perform additional adhesive operation pr RN GL54D prior to final ambient. See paragraph 3.8.1 thru 3.8.5.

4/24/93
DWS

For OLS 12 only perform the following post coax connector repair tests:

- NEWON1.ST
- QKTESTN.ST
- 5X10X1SS.ST
- NEWON2.ST
- QKTESTR.ST
- 5X10X2SS.ST

n/A	
n/A	
n/A	
n/A	
n/A	
n/A	
n/A	
IRH	4/15/93

Perform T927686 paragraph 3.8, Final Scan Plane Definition.

Perform T927686 paragraph 3.9, Inspection, Data Review, and Preparation for Shipment except Delete paragraph 3.9.2, Weight and Center of Gravity.

- Pin Retention
- Inspection
- Data Review
- AFPRO
- Pack
- Ship

	4/29/93
	4/29/93
	4/29/93
	4/29/93
	4/29/93
	4/29/93
	4/29/93

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CRIGGIN HL

07/27/92

BVS 2600

REVISION E

DATE 24 July 92

ORIGINATOR M. Epperly

QUALITY ASSURANCE RW Bole

MANUFACTURING DW Siffes

5-3659

RDS REWORK AND RETEST PROCEDURE

For OLS 12, 13, 14, 15 and 16

OLS 14

Contract F04701-90-C-0028

Prepared For

UNITED STATES AIR FORCE
Headquarters, Space Division
Los Angeles, California

Prepared By

WESTINGHOUSE ELECTRIC CORPORATION
Electronics Systems Group
Baltimore, Maryland

REVISION SHEET

Revision Letter	Revision Date	Affected Pages	Revision Made by
-	25 Jul 90	Released	
A	14 Aug 90	Pages 2 and 13-27	
B	5 Oct 90	Pages 2-5	Rennenkampf
C	19 Feb 91	Pages 1, 2 and 13-27	Epperly
D	20 Jul 92	Pages 1-4 and 13	Barrett
E	23 Jul 92	All Pages	Epperly

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B-2

1.0 Introduction

This document describes the detailed rework and retest plan for SPS and OSU units returned from the field for incorporation of Real-time Data Smooth. A copy of this document will serve as a checklist for accomplishing the rework and retest procedures. Because there is a possibility of units being interchanged between systems, it is necessary to work to unit serial numbers rather than system serial numbers.

Incorporation of RD into an OLS requires the modification of the following assemblies:

- 9C and SDF-5 boards in the SPS
- SPS Matrix Plate
- OSU-1 and OSU-2 boards in the OSU
- OSU Matrix Plate
- OSU Top Cover Assembly

For OLS-14 (System S/N 5009) only, the following SPS boards must also be reworked:

FC-2, CLCK and 9BX

Record Serialization of Units to be reworked here:

OLS 536R500G <u>03</u>	<u>5009</u>
SPS 651R390G <u>02</u>	<u>5008</u>
OSU 640R960G <u>04</u>	<u>5009</u>

Record Assembly Serial Numbers here:

SPS Mother Plate Assy	(651R342) <u>604</u>	SN	<u>8619-0001</u>
Prime 9C Board	(640R570 or 640R658) <u>602</u>	SN	<u>5006</u>
Redundant 9C Board	(640R570 or 640R658) <u>602</u>	SN	<u>5001</u>
Prime SDF-5 Board	(640R544 or 640R648) <u>602</u>	SN	<u>5014</u>
Redundant SDF-5 Board	(640R544 or 640R648) <u>602</u>	SN	<u>5015</u>
OSU Mother Plate Assy	(522R783) <u>602</u>	SN	<u>0001</u>
OSU Top Cover Assy	(644R046) <u>603</u>	SN	<u>5009</u>
OSU-1 Board	(640R522) <u>604</u>	SN	<u>5010</u>
OSU-2 Board	(640R524) <u>603</u>	SN	<u>5009</u>

For OLS-14 (System S/N 5009) only, record the serial numbers of the FC2, CLCK and 9BX Boards are:

Prime FC2	(640R454)	SN	<u>5015</u>
Redundant FC2	(640R454)	SN	<u>5014</u>
Prime CLCK	(640R406)	SN	<u>5014</u>
Red CLCK	(640R406)	SN	<u>5015</u>
Prime 9BX	(640R656) <u>602</u>	SN	<u>5001</u>
Redundant 9BX	(640R656) <u>602</u>	SN	<u>5000</u>

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NOTE: For OLS 14 and 15, BVS 2711 describes REU interface modifications which are to be made at the same time as these changes.

Labor for this task is to be charged as follows:

- OLS 14 - A 53864 CACA
- OLS 15 - A 53864 CADA

2.0 Rework and Assembly Retest Plan

2.1 Pre-Rework

Incoming Inspection of Returned Units	
SPS (651R390)	WEC
SPS (651R390)	DPRO
OSU (640R960)	WEC
OSU (640R960)	DPRO

Verification	
Mfg/Date	Insp/Date
	7/10/92
	7/10/92
	7/10/92
	7/10/92

2.2 Rework and Inspection

The RN numbers listed below are for reference only - All assemblies should be configured to their latest revision. Serial numbers are unit serial numbers.

System Rework (536R500)	GG42D		
SPS Chassis Rework (651R390)		DWS 8/21/92	GG50
S/N 5007, 5009-5011	GG36D		
S/N 5008	GG71D		
SPS Mother Plate Assy Rework (651R342)		S&IS 8/4/92	GG70
S/N 5007, 5009-5011	GG35D		
S/N 5008	GG70D (ref. GG16D)		
SPS Matrix Plate Wiring Rework (wiretabs 322R959 or 322R960)		S&IS 8/4/92	GG70
S/N 5007	GG17D		
S/N 5008-5011	GG16D		
9C board assy rework (775R076 or 775R077) ⁵⁰⁰⁰ ₅₀₀₁		DWS 8/14/92	GG30
S/N 5007	GG10D, GG15D & GG20D		
S/N 5008	GG69D (ref. GG11D, GG15D, GG21D)		
S/N 5009-5011	GG11D, GG15D & GG21D		

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ORIGINAL
07/22/92
NY

SDF-5 board assy rework (~~775R078~~ & 775R079) ⁵⁰¹⁴ 5015 DWS 8/14/92 60
38
 S/N 5007 GG08D, GG14D & GG18D
 S/N 5008 GG68D (ref. GG09D, GG14D, GG19D)
 S/N 5009-5011 GG09D, GG14D & GG19D

NOTE: For OLS 14 and 15, BVS 2711 changes must also be worked. Record work on that BVS.

			<u>Verification</u>	
			<u>Mfg/Date</u>	<u>Insp/Date</u>
OSU chassis rework (640R960)			Bf 8/21/92	60 38
S/N 5007	5009-5011	GG33D		
S/N 5008		GG67D		
OSU Top Cover Rework (644R046)			Bf 8/21/92	60 38
S/N 5007	5009-5011	GG32D		
S/N 5008		GG65D		
OSU Mother Plate Rework (522R783)			Bf 8/21/92	60 38
S/N 5007	5009-5011	GG34D		
S/N 5008		GG66D		
OSU Matrix Plate Wiring Rework (wiretab 322R958)			Bf 8/21/92	60 38
S/N 5007	5011	GG22D		
OSU-1 board assy rework (775R080)			Bf 8/21/92	60 38
S/N 5007	5009-5011	GG12D, GG15D & GG23D		
S/N 5008		GG54A, GG55A & GG56A GG63D		
OSU-2 board assy rework (775R081)			NY 8/21/92	60 38
S/N 5007	5009-5011	GG13D, GG15D & GG24D		
S/N 5008		GG64D		

For OLS-14 System S/N (5009) only, record rework and inspection:

FC-2 board assy rework (640R454)			DWS 8/13/92	60 38
GU351				
GU355				
CLCK board assy rework (640R406)			DWS 8/13/92	60 38
GR99B				
9BX board assy rework (640R656)			DWS 8/13/92	60 38
GU452				
GU456				

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ORIGINAL
07/27/92
AY

2.3 Assembly Level Retest

2.3.1 Prime Side 9C Retest (~~775R075~~/775R077)

SN 5000

	<u>Date</u>	<u>Verification</u> (SP-1)
Rework Complete - No open items on ICT	<u>8/6/92</u>	
Room Temperature Retest per paragraph 4.3 of T814A76	<u>8/7/92</u>	
Pre Coat Data Review	<u>8/8/92</u>	
WEC Inspection - OK to Coat	<u>8/8/92</u>	
DPRO Inspection - OK to Coat	<u>8/8/92</u>	
Conformal Coat	<u>8/8/92</u>	
Eight Non-powered Temperature Cycles	<u>8/11/92</u>	
Hi/Low Temperature Test per paragraph 4.7 of T814A76	<u>8/12/92</u>	
Data Review Complete	<u>8-13-92</u>	
WEC Inspection - Assembly Complete	<u>8/13/92</u>	
DPRO Inspection - Assembly Complete	<u>8/14/92</u>	

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ORIGINAL
07/27/92
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2.3.2	<u>Redundant Side 9C Retest (775R076/775R077)</u>	SN	<u>5001</u>
	Rework Complete - No open items on ICT		<u>8/9/92</u>
	Room Temperature Retest per paragraph 4.3 of T814A76		<u>8/9/92</u>
	Pre Coat Data Review	<u>L.O.</u>	<u>8/10/92</u>
	WEC Inspection - OK to Coat		<u>AS 10'92</u>
	DPRO Inspection - OK to Coat		<u>8/10/92</u>
	Conformal Coat		<u>8/10/92</u>
	Eight Non-powered Temperature Cycles		<u>8/11/92</u>
	Hi/Low Temperature Test per paragraph 4.7 of T814A76		<u>8/13/92</u>
	Data Review Complete		<u>8/3/92</u>
	WEC Inspection - Assembly Complete		<u>8/13/92</u>
	DPRO Inspection - Assembly Complete		<u>8/14/92</u>

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07/21

2.3.3 Prime Side SDF-5 Retest (~~775R078~~/775R079) 603 SN

5014

	Date	Verification
Rework Complete - No open items on ICT	8/6/92	109
Room Temperature Retest per paragraph 4.3 of T814A78	8/7/92	VIA AIA
Pre Coat Data Review	8/7/92	9E
WEC Inspection - OK to Coat	8/8/92	OC 109
DPRO Inspection - OK to Coat	8/8/92	601 109 5 N
Conformal Coat	8/10/92	109 107
Eight Non-powered Temperature Cycles	8/11/92	322A AIA
Hi/Low Temperature Test per paragraph 4.7 of T814A78	8/13/92	322A AIA
Data Review Complete	8/13/92	109
WEC Inspection - Assembly Complete	8/13/92	109
DPRO Inspection - Assembly Complete	8/14/92	109 107 109 N

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2.3.4 Redundant Side SDF-5 Retest (775R078/775R079) SN 5012

Rework Complete - No open items on
ICT

Room Temperature Retest per paragraph
4.3 of T814A78

Pre Coat Data Review

WEC Inspection - OK to Coat

DPRO Inspection - OK to Coat

Conformal Coat

Eight Non-powered Temperature Cycles

Hi/Low Temperature Test per paragraph
4.7 of T814A78

Data Review Complete

WEC Inspection - Assembly Complete

DPRO Inspection - Assembly Complete



8/2/92



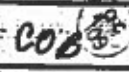
8/3/92



8/7/92



8/11/92



8/10/92



8/13/92



8/13/92



8/14/92



8/14/92



8/14/92

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ORIGINAL TEST 5/7/92

2.3.5 OSU-1 Retest (775R080)G03

SN 5010

	<u>Date</u>	<u>Verific</u>
Rework Complete - No open items on ICT	<u>8/7/92</u>	
Room Temperature Retest per paragraph 4.3 of T814A80	<u>8/2/92</u>	
Pre Coat Data Review	<u>8/3/92</u>	
WEC Inspection - OK to Coat	<u>8/7/92</u>	
DPRO Inspection - OK to Coat	<u>8/7/92</u>	
Conformal Coat	<u>8/9/92</u>	
Eight Non-powered Temperature Cycles	<u>8/11/92</u>	
Hi/Low Temperature Test per paragraph 4.7 of T814A80	<u>8/13/92</u>	
Data Review Complete	<u>8/13/92</u>	
WEC Inspection - Assembly Complete	<u>8/14/92</u>	
DPRO Inspection - Assembly Complete	<u>8/14/92</u>	

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ORIGINAL
07/27/92
AT

2.3.6 OSU-2 Retest (775R081)G02 SN 5009

	<u>Date</u>	<u>Verification</u>
Rework Complete - No open items on ICT	<u>8/3/92</u>	<u>[Signature]</u>
Room Temperature Retest per paragraph 4.3 of T814A81	<u>8/5/92</u>	<u>[Stamp]</u>
Pre Coat Data Review	<u>8/5/92</u>	<u>L.O.</u>
WEC Inspection - OK to Coat	<u>8/6/92</u>	<u>[Stamp]</u>
DPRO Inspection - OK to Coat	<u>8/6/92</u>	<u>[Stamp]</u>
Conformal Coat	<u>8/9/92</u>	<u>COB</u>
Eight Non-powered Temperature Cycles	<u>8/11/92</u>	<u>[Stamp]</u>
Hi/Low Temperature Test per paragraph 4.7 of T814A81	<u>8/13/92</u>	<u>[Stamp]</u>
Data Review Complete	<u>8/14/92</u>	<u>L.O.</u>
WEC Inspection - Assembly Complete	<u>8/14/92</u>	<u>[Stamp]</u>
DPRO Inspection - Assembly Complete	<u>8/14/92</u>	<u>[Stamp]</u>

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ORIGINAL
07/27/92
/

2.3.7 OSU Assembly Retest (640R960)

SN _____

Rework Complete - No open items on
ICT

Date
DWS 8/25/92

Verificati

Room Temperature Retest per paragraph
4.1 and 4.2 of T814A56

8/26/92

Hi/Low Temperature Test per paragraph
4.7 of T814A56

8/26/92

Data Review Complete

8/24/92

S.O.

WEC Inspection - Assembly Complete

MS 28 '92

DPRO Inspection - Assembly Complete












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ORIGINAL
9/23/92
11

OLS-14 (SYSTEM S/N 5009) ONLY

2.3.8	<u>Prime Side FC-2 Retest (640R454)</u>	SN	<u>5015</u>
Rework Complete - No open items on ICT	<u>8/6/92</u>		<u> </u>
Room Temperature Retest per paragraph 4.3 of T927025	<u>8/7/92</u>		<u> </u>
Pre Coat Data Review	<u>8/7/92</u>		<u> </u>
WEC Inspection - OK to Coat	<u>8/8/92</u>		<u> </u>
DPRO Inspection - OK to Coat	<u>8/8/92</u>		<u> </u>
Conformal Coat	<u>8/9/92</u>		<u> </u>
Eight Non-powered Temperature Cycles	<u>8/10/92</u>		<u> </u>
Hi/Low Temperature Test per paragraph 4.7 of T927025	<u>8/11/92</u>		<u> </u>
Data Review Complete	<u>8/12/92</u>		<u> </u>
WEC Inspection - Assembly Complete	<u> </u>		<u> </u>
DPRO Inspection - Assembly Complete	<u>9/20/93</u>		<u> </u>

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JERWIN HL
07/27/92
15

OLS-14 (SYSTEM S/N 5009) ONLY

2.3.9 Redundant Side FC-2 Retest (640R454)

SN 5014

Rework Complete - No open items on
ICT



8/6/92

Room Temperature Retest per paragraph
4.3 of T927025



8/7/92

Pre Coat Data Review

DOPE

8/7/92

WEC Inspection - OK to Coat



8/7/92

DPRO Inspection - OK to Coat



8/7/92

Conformal Coat

8/9/92

8/9/92

Eight Non-powered Temperature Cycles



8/10/92

Hi/Low Temperature Test per paragraph
4.7 of T927025

8/11/92

8/11/92

Data Review Complete



8/12/92

WEC Inspection - Assembly Complete



8/12/92

DPRO Inspection - Assembly Complete



9-08-92

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OLS-14 (SYSTEM S/N:5009) ONLY

2.3.10 Prime Side CLCK Retest (640R406)

SN 7

5014

Rework Complete - No open items on ICT

8/1/92



Room Temperature Retest per paragraph 4.3 of T927019

8/1/92



Pre Coat Data Review

8/1/92



WEC Inspection - OK to Coat

8/1/92



DPRO Inspection - OK to Coat

8/1/92



Conformal Coat

8/5/92

C.O.B

Eight Non-powered Temperature Cycles

8/11/92



Hi/Low Temperature Test per paragraph 4.7 of T927019

8/12/92



Data Review Complete

8/12/92

J.O.

WEC Inspection - Assembly Complete

8/12/92



DPRO Inspection - Assembly Complete

920813



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ORIGINAL
07/87
M-

OLS-14 (SYSTEM S/N 5009) ONLY

2.3.11 Redundant Side CLCK Retest (640R406)

SN 5015

Rework Complete - No open items on
ICT



8-6-92

Room Temperature Retest per paragraph
4.3 of T927019



8/2/92

Pre Coat Data Review



8-3-92

WEC Inspection - OK to Coat



8-6-92

DPRO Inspection - OK to Coat



8-7-92

Conformal Coat



8/9/92

Eight Non-powered Temperature Cycles



8/11/92

Hi/Low Temperature Test per paragraph
4.7 of T927019



8-11-92

Data Review Complete



8-12-92

WEC Inspection - Assembly Complete



8-12-92

DPRO Inspection - Assembly Complete














8-13-92

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ORIGINAL
07/21/92
11

OLS-14 (SYSTEM S/N 5009) ONLY

2.3.12 <u>Prime Side 9BX Retest (640R656)</u>	SN <u>5001</u>
Rework Complete - No open items on ICT	AS 8 '92 
Room Temperature Retest per paragraph 4.3 of T928339	<u>8/5/92</u> 
Pre Coat Data Review	<u>8-6-92</u> 
WEC Inspection - OK to Coat	AS 7 '92 
DPRO Inspection - OK to Coat	<u>8/7/92</u> 
Conformal Coat	<u>8/8/92</u> 
Eight Non-powered Temperature Cycles	<u>8/11/92</u> 
Hi/Low Temperature Test per paragraph 4.7 of T928339	<u>8/12/92</u> 
Data Review Complete	<u>8/12/92</u> 
WEC Inspection - Assembly Complete	<u>9/12/92</u> 
DPRO Inspection - Assembly Complete	<u>9/20/92</u> 

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ORIGINAL
07/11/92

OLS-14 (SYSTEM S/N 5009) ONLY

		SN	5000
2.3.13	<u>Redundant Side 9BX Retest (640R656)</u>		
Rework Complete - No open items on ICT	<u>8/6/92</u>		911 200
Room Temperature Retest per paragraph 4.3 of T928339	<u>8/7/92</u>		911 200
Pre Coat Data Review	<u>8/7/92</u>		907 RE
WEC Inspection - OK to Coat	<u>AS 7 '92</u>		OC 21
DPRO Inspection - OK to Coat	<u>8/7/92</u>		OC 21 OC 21
Conformal Coat	<u>8/9/92</u>		OC 21 COB
Eight Non-powered Temperature Cycles	<u>8/11/92</u>		3038 ATA
Hi/Low Temperature Test per paragraph 4.7 of T928339	<u>8/12/92</u>		8867 ATA
Data Review Complete	<u>8/12/92</u>		J.O.
WEC Inspection - Assembly Complete	<u>AS 12 '92</u>		121 200
DPRO Inspection - Assembly Complete	<u>920813</u>		920813

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OLS PROGRAM DIRECTIVE

-CONTINUATION SHEET-











DIRECTIVE NO. 045

DATE 12/12/88

*0LS14 BUS2600
9-15-92*

CHECKPOINT #A
(Before Subsystem Test, in Block 5 Clean-Room, per T927000)

THERMAL CHAMBER 

- | | | |
|---|---|------------|
| A. Area is clean and contains no miscellaneous parts or extraneous hardware. |  | RESP
QE |
| B. The anti-static mat and wrist straps are in place and ready for use. |  | QE |
| C. Unit and System cable connector pins are checked and none are bent or pushed in. (NOTE: This can be verified at presystem unit inspection and the buffer connectors are inspected and installed at this point per PD034) |  | INSP |
| D. Verify correct color code on buffer connectors to certify inspected, tested and approved connectors per PD 034 (Appendix B). |  | INSP |
| E. Test equipment configuration checked per Program Directive #022 ⁰³³ less paragraphs IID and IIE. |  | QE |
| F. Perform Test Equipment Operational Check per PD022 ^{PD033} paragraphs IID and IIE. |  | TD |
| G. Review open ICT items on the SPS ^{N/A} , SPU ^{N/A} , PSU ^{OSU} and OSU ^{OSU} and evaluate closure prior to moving to subsystem test. |  | QE |
| <i>SPS - TEST RN'S AK to AD to be done @ system test.</i> |  | TD |
| H. TCP, SPS, SPU, PSU and OSU are connected to system cable connectors and each is grounded to the ground bus per 9R07845. |  | MFG |
| I. Item annotated on ^{SPS and OSU} ICT that units are ready for subsystem test. - <i>SPS and PSU ICTS</i> |  | TD |

NOTE 1:
WHEN APPLICABLE,
RECORD ACTIONS
PERFORMED IN SYSTEM
LOG BOOK.

NOTE 2:
"CAUTION"
BUMP, SHOCK,
ABRUPT MOVEMENT
OF TAPE RECORDERS
CAN CAUSE SEVERE
DAMAGE.

OLS PROGRAM DIRECTIVE

-CONTINUATION SHEET-

DIRECTIVE NO. 045

DATE 12/12/88

CHECKPOINT #8

(Before Sub-system Thermal Test, in PQL, per T927000)

0LS14 9-
3452600

CONNECTORS
CHK ON
SPS AND OSU
OK

- A. PQL thermal chamber is clean and area contains no miscellaneous parts or extraneous hardware.
- B. Assembly and cable connectors (buffered and unbuffered) were checked before cabling up, and no damage was found. (Note: Where buffer connectors are in-place, check the viewable side; do not remove from assembly.) VERIFY COLOR CODE ON BUFFER CONNECTORS PER PD 034, APPENDIX B.
- C. SPS, SPU, BSU and OSU transported per 9RA4220.
- D. Assemblies SPS, SPU and TCP are connected to system cable connectors and each is grounded to the ground bus per 9R07845.
- E. Test equipment is in calibration as required in Program Directive #022 less paragraphs IID and IIE.
- F. Perform Test Equipment operational check per PD 022 paragraphs IID and IIE.

QF
IN
MF
92
92/12/88

NOTE 1:
WHEN APPLICABLE,
RECORD ACTIONS
PERFORMED IN SYSTEM
LOG BOOK.

NOTE 2:
"CAUTION"
BUMP, SHOCK,
ABRUPT MOVEMENT
OF TAPE RECORDERS
CAN CAUSE SEVERE
DAMAGE.

3.0 Subsystem Level Retest Procedure

3.1 Ambient Subsystem Verification

Date

Verification

AT THERMAL CHAMBER 9-15-92

Rework Complete - No unexplained Open Items on ICT

10 92



Rework of BVS 2711 complete (OLS 14 and 15 only)

9/16/92



Checkpoint A of PD 045 (attach copy)

9/15/92



Run the following Test Files (Room Temperature):

NEWON1SS.ST	9/17/92	MCE 7/1/92
QKTESTN.ST	9/17/92	MCE 7/1/92
5X1X1SS.ST		
RDSTSTSS.ST	9/17/92	MCE 7/1/92
NEWON2SS.ST	9/17/92	MCE 7/1/92
QKTESTR.ST	9/17/92	MCE 7/1/92
5X1X2SS.ST		
RDSTSTSS.ST	9/18/92	AIA 7/1/92
5X18X1SS.ST	9/18/92	AIA 7/1/92
5X18X2SS.ST	9/18/92	AIA 7/1/92
5X18X3SS.ST	9/18/92	AIA 7/1/92
5X18X4SS.ST	9/19/92	AIA 7/1/92

3.1.1 Ambient Encrypter Verification

Note: To allow for scheduling and security constraints the ambient encrypter verification may be performed out of sequence, however, paragraph 3.1.1 must be complete prior to starting paragraph 3.3.9, Thermal Cycle #8.

KG-46 is GFE.
No KG-46 is available for test. Therefore tests in para 3.1.1 cannot be run and are not applicable

Install the KG-46 data encrypter and KG-28 decrypter. Check out the KG-28 set-up using the ST-19 verification procedure

para. 3.1.1 is Not Applicable
MCE 9/14/92

Run the following test files:

NEWON1SS.ST

RDSTST.ST

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NEWON2SS.ST

RDSTST.ST

NEWOND1.ST

RDSTSTSS.ST

NEWOND2.ST

RDSTST.ST

3.2 SPS and OSU Vibration

Note: To allow for scheduling, OSU and SPS vibration may occur before ambient tests per paragraph 3.1. Vibration must be completed prior to starting paragraph 3.3.6. Later vibrations due to rework or RN incorporation shall be recorded on the units ICT.

Pre-Vib Data Review (TEST DIRECTOR)

NEE

9/21/92

WEC Inspection - OK to Vibrate

[Stamp]

9/21/92

DPRO Inspection - OK to Vibrate

[Stamp]

9/21/92

Vibrate SPS - x-axis, random only
5D3 acceptance level per PQL737, nonpowered

AJS

9/21/92

Vibrate OSU - x-axis, random only
5D3 acceptance level per PQL737, nonpowered

AJS

9/21/92

WEC Inspection - Post Vib

[Stamp]

9-22-92

DPRO Inspection - Post Vib

[Stamp]

9/22/92

3.3 Thermal Test

Checkpoint B of PD-045
(attach copy)

[Stamp]

9-23-92

Install Thermocouples (PQL operation)

AJS

9-23-92

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3.3.1 Ambient Verification

Run the following Test Files (Room Temperature):
 May be postponed until after Vibration *AEZ*

	Date	Verification
NEWON1.ST	9/22/92	<i>mg</i>
QKTESTN.ST	9/22/92	<i>mg</i>
RDSTSTSS.ST	9/22/92	<i>mg</i>
NEWON2.ST	9/22/92	<i>mg</i>
QKTESTR.ST	9/22/92	<i>mg</i>
RDSTSTSS.ST	9/22/92	<i>mg</i>
<i>Am 14-8</i> 5X18X1SS.ST	9/22/92	<i>mg</i>
5X18X2SS.ST	9/23/92	<i>mg</i>
5X18X3SS.ST	9/23/92	<i>mg</i>
5X18X4SS.ST	9/23/92	<i>mg</i>

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3.3.2 Thermal Cycle #1

3.3.2.1 Hot Temperature

Date

Verificatio

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

9-19-92

3146
A/A

Run 5x18x1SS.ST

9-19-92

3146
A/A

~~RDSSTSTSS.ST~~ ~~RDSSTSTSS.ST~~

9-19-92

3146
A/A

Enter CON 0 42

9-19-92

3146
A/A

Enter OLS OFF

9-19-92

3146
A/A

3.3.2.2 Cold Temperature

Date

Verificatio

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

9-19-92

3146
A/A

Run 5x18x1SS.ST

9-19-92

3146
A/A

~~RDSSTSTSS,ST~~ ~~RDSSTSTSS.ST~~

9-19-92

3146
A/A

Enter CON 0 42

9-19-92

3146
A/A

Enter OLS OFF

9-19-92

3146
A/A

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3.3.3 Thermal Cycle #2

3.3.3.1 Hot Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x2SS.ST

9/19/92

2899
AIA

RDS†TSTSS.ST

9/19/92

2899
AIA

Enter CON 0 42

9/19/92

2899
AIA

Enter OLS OFF

9/19/92

2899
AIA

9/19/92

2899
AIA

3.3.3.2 Cold Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x2SS.ST

9/20/92

0301
AIA

RDS†TSTSS.ST

9/20/92

0301
AIA

Enter CON 0 42

9/20/92

0301
AIA

Enter OLS OFF

9/20/92

0301
AIA

9/20/92

0301
AIA

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3.3.4 Thermal Cycle #3

3.3.4.1 Hot Temperature

Date

Verificatio

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x3SS.ST

9-20-92

3146
A/A

RDS/TSTSS.ST

9-20-92

3146
A/A

Enter CON 0 42

9-20-92

3146
A/A

Enter OLS OFF

9-20-92

3146
A/A

3.3.4.2 Cold Temperature

Date

Verificatio

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x3SS.ST

9-20-92

3146
A/A

RDS/TSTSS.ST

9/20/92

3146
A/A

Enter CON 0 42

9/20/92

3146
A/A

Enter OLS OFF

9/20/92

3146
A/A

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3.3.5 Thermal Cycle #4

3.3.5.1 Hot Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x4SS.ST

9/20/92

U234
ATA

RDSSTSTSS.ST

9/20/92

U200
ATA

Enter CON 0 42

9/20/92

U209
ATA

Enter OLS OFF

9/20/92

U230
ATA

9/20/92

U295
ATA

3.3.5.2 Cold Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x4SS.ST

9/21/92

U301
ATA

RDSSTSTSS.ST

9/21/92

U301
ATA

Enter CON 0 42

9/21/92

U381
ATA

Enter OLS OFF

9/21/92

U381
ATA

9/21/92

U381
ATA

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3.3.6 Thermal Cycle #5

3.3.6.1 Hot Temperature






Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

- Run 5x18x1SS.ST
- RDS/TSTSS.ST
- Enter CON 0 42
- Enter OLS OFF

9/23/92
9/23/92
9/23/92
9/23/92

3.3.6.2 Cold Temperature






Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

- Run 5x18x1SS.ST
- RDS/TSTSS.ST
- Enter CON 0 42
- Enter OLS OFF

9/23/92
9/24/92
9/24/92
9/24/92
9/24/92

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3.3.7 Thermal Cycle #6

3.3.7.1 Hot Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x2SS.ST

9/24/92

3146 A/A

RDS#TSTSS.ST

9/24/92

3146 A/A

Enter CON 0 42

9/24/92

3146 A/A

Enter OLS OFF

9/24/92

3146 A/A

9/24/92

3146 A/A

3.3.7.2 Cold Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x2SS.ST

9/24/92

3146 A/A

RDS#TSTSS.ST

9/24/92

3146 A/A

Enter CON 0 42

9/24/92

3146 A/A

Enter OLS OFF

9/24/92

3146 A/A

9/24/92

3146 A/A

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3.3.8 Thermal Cycle #7

3.3.8.1 Hot Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x3SS.ST

9/24/92

RDS/TSTSS.ST

9/24/92

Enter CON 0 42

9/24/92

Enter OLS OFF

9/24/92

9/24/92

3.3.8.2 Cold Temperature

Date

Verification

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

Run 5x18x3SS.ST

9/25/92

RDS/TSTSS.ST

9/25/92

Enter CON 0 42

9/25/92

Enter OLS OFF

9/25/92

9/25/92

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3.3.9 Thermal Cycle #8

3.3.9.1 Hot Temperature

Allow chamber air in the vicinity of the SPS to stabilize at +50°C +4°/-0° for 2 hours. During this time the OLS shall not have power on.

9/25/92

U301
A/A

Run the following dual prime Test Files:

NEWOND1.ST ✓	<u>9/25/92</u>	<u>U301 A/A</u>	<i>[Signature]</i>
5x3x1SS.ST ✓	<u>9/25/92</u>	<u>U301 A/A</u>	<i>[Signature]</i>
5x3x5SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5x5x1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5x6x1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5x8x1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5X9X1SS.ST	<u>9/25/92</u>	<u>U301 A/A</u>	<i>[Signature]</i>
5X10X1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5X11X1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5X12X1SS.ST	<u>9/20/92</u>	<u>U301 A/A</u>	<i>[Signature]</i>
5X13X1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5x13x3SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5X14X1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5X16X1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5X17X1SS.ST ✓	<u>9/25/92</u>	<u>3145 A/A</u>	<i>[Signature]</i>
5X19X3SS.ST <i>ACE</i>	<u>—</u>	<u>—</u>	<u>—</u>
5X2X1SS.ST	<u>9/26/92</u>	<u>U301 A/A</u>	<i>[Signature]</i>
5X2X2SS.ST	<u>9/27/92</u>	<u>U301 A/A</u>	<i>[Signature]</i>

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Run the following dual prime Test Files:

NEWOND2.ST ✓	<u>9/25/92</u>		<i>MM</i>
5x3x2SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5x3x6SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5x5x2SS.ST ✓	<u>9/25/92</u>		<i>MM</i>
5x6x2SS.ST ✓	<u>9/25/92</u>		<i>MM</i>
5x8x2SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5X9X2SS.ST ✓	<u>9/26/92</u>		<i>MM</i>
5X10X2SS.ST ✓	<u>9/26/92</u>		<i>MM</i>
5X11X2SS.ST ✓	<u>9/25/92</u>		<i>MM</i>
5X12X2SS.ST	<u>9/30/92</u>	<i>MSA</i>	<i>MM</i>
5x13x2SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5x13x4SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5X14X2SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5X16X2SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5X17X2SS.ST ✓	<u>9/27/92</u>		<i>MM</i>
5X19X4SS.ST Deleted <i>MM</i>	<u> </u>	<u> </u>	<u> </u>

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B-32

Paragraph is not applicable *MCE*

Install the KG-46 data encrypter and KG-28 decrypter. Check out the KG-28 set-up using the ST-19 verification procedure.

KG-28 is GFE and is not available for this test 9/29/92 *MCE*

Run the following test files:

NEWOND1.ST

RDS~~ST~~STSSE.ST

NEWOND2.ST

RDS~~ST~~STSSE.ST

Enter CON 0 42

Enter OLS OFF

Remove the KG-46 encrypter and KG-28 decrypter.

3.3.9.2 Cold Temperature

Allow chamber air in the vicinity of the SPS to stabilize at -10°C +0°/-4° for 2 hours. During this time the OLS shall not have power on.

MCE

9/27/92

Run the following dual prime Test Files:

NEWOND1.ST

5x3x1SS.ST

5x3x5SS.ST

5x5x1SS.ST

5x6x1SS.ST

5x8x1SS.ST

5X9X1SS.ST

5X10X1SS.ST

5X11X1SS.ST

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

MCE

9/28/92 *MCE*

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5X12X1SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x13x1SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x13x3SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X14X1SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X16X1SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X17X1SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X19X3SS.ST		
DELETED <u>MSE 7-30-92</u>		
5X2X1SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X2X2SS.ST	<u>MSE</u>	<u>9/28/92</u>

Run the following dual redundant Test Files:

NEWOND2.ST	<u>MSE</u>	<u>9/28/92</u>
5x3x2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x3x6SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x5x2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x6x2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x8x2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X9X2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X10X2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X11X2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X12X2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x13x2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5x13x4SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X14X2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X16X2SS.ST	<u>MSE</u>	<u>9/28/92</u>
5X17X2SS.ST	<u>MSE</u>	<u>9/28/92</u>

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