

BVS 2733
DATE 28 May 1993
ORIGINATOR *J. Scilipoti*
 J. Scilipoti
REV -

F13

OLS #14 BEARING RETROFIT

ACCEPTANCE TEST REPORT
VOLUME III OF III
ALIGNMENT AND SYNCHRONIZATION CURVES

(CDRL 006A01)

Contract F04701-90-C-0028

Prepared For

UNITED STATES AIR FORCE
Headquarters, Space Division
Los Angeles, California

Prepared By

WESTINGHOUSE ELECTRIC CORPORATION
Defense and Electronics Center
Baltimore, Maryland

F13

विद्यार्थी नाम: _____
वि. सं. _____
वि. सं. _____
वि. सं. _____

विद्यार्थी नाम: _____

वि. सं. _____

विद्यार्थी नाम: _____
वि. सं. _____
वि. सं. _____

वि. सं. _____

विद्यार्थी नाम: _____
वि. सं. _____
वि. सं. _____

TABLE OF CONTENTS

	<u>PAGE</u>
1. REFPLN Plots.....	1
2. Align/Sync Plotted with Respect to REFPLN.....	4
Alignment - HRD.....	5
- T.....	6
- PMT.....	7
Synchronization - HRD.....	8
- T.....	9
- PMT.....	10
3. Alignment/Synchronization For All Modes.....	12
Alignment - HRD.....	13
- T.....	18
- PMT.....	21
Synchronization - HRD SDF & SDS.....	26
- HRD RTDF & RTDS.....	34
- T SDF & RTDS & SDS.....	40
- PMT SDS & RTDS.....	48
4. Synchronization Using Backup Encoder & Encoder Simulator.....	58

ABBREVIATIONS

AS Along Scan (Synchronization)
AT Along Track (Alignment)
SD Surface Distance
SDF Stored Data Fine
SDS Stored Data Smooth
RTD F Real Time Data - Fine
RTD S Real Time Data - Smoothed
H HRD Channel
T T(Thermal) Channel
P PMT Channel

1. REFPLN PLOTS

These are the computer-generated least-squares fits to OLS #14 HRD and T Channel Stored Data Fine (SDF) Alignment and Synchronization data taken from the final +5°C SSS/ -8°C M1 Thermal Vacuum run (Orbit Nominal).

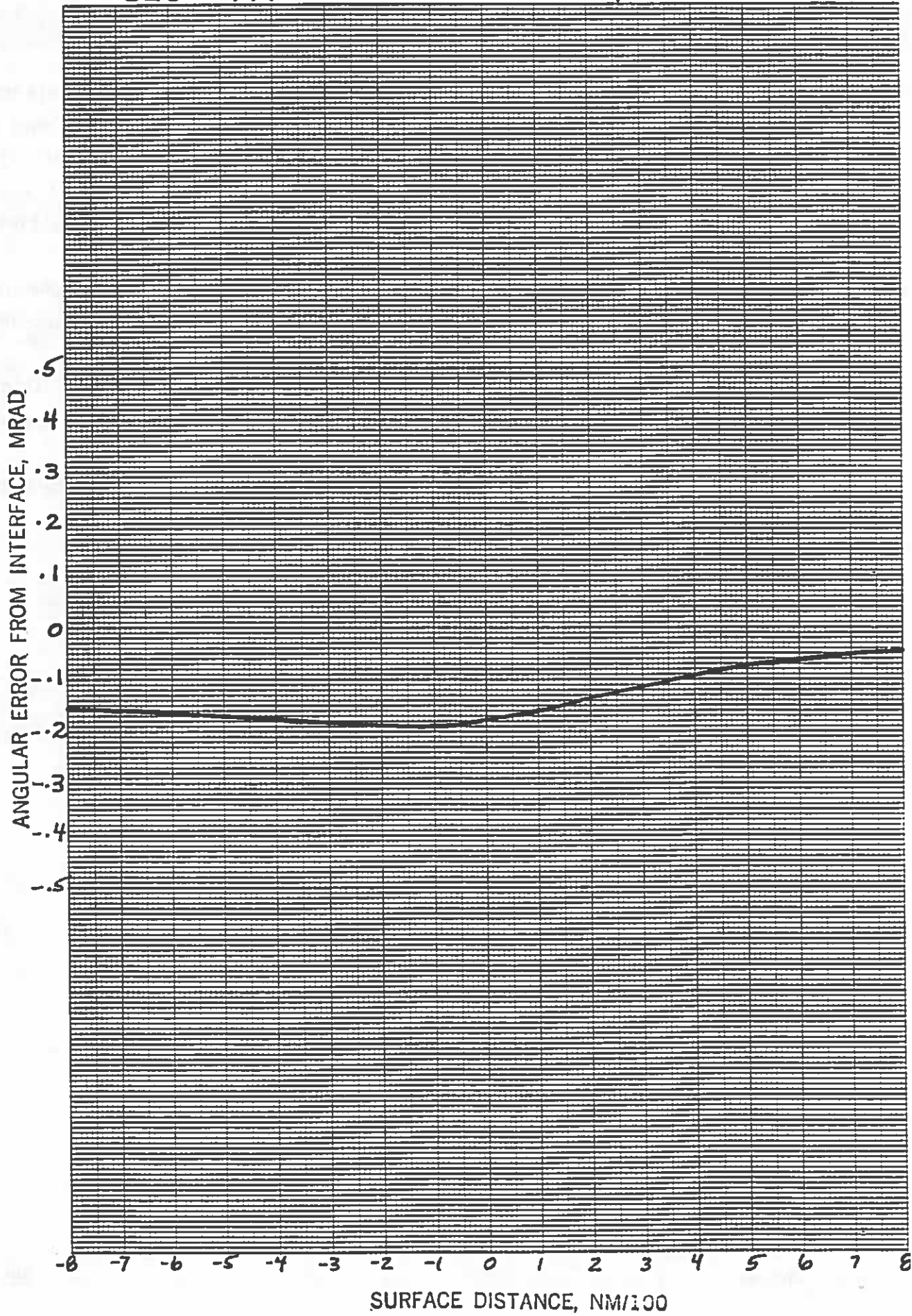
For OLS #14, data from Thermal Vacuum Runs with M1 = +12°C was also used to take into account any Alignment and Synchronization sensitivity to M1 temperature.

REFPLN is a computer program which generates the Alignment and Synchronization which represents the line-of-sight (LOS) or "look-angles" of the SSS with respect to the mounting (Interface) axes.

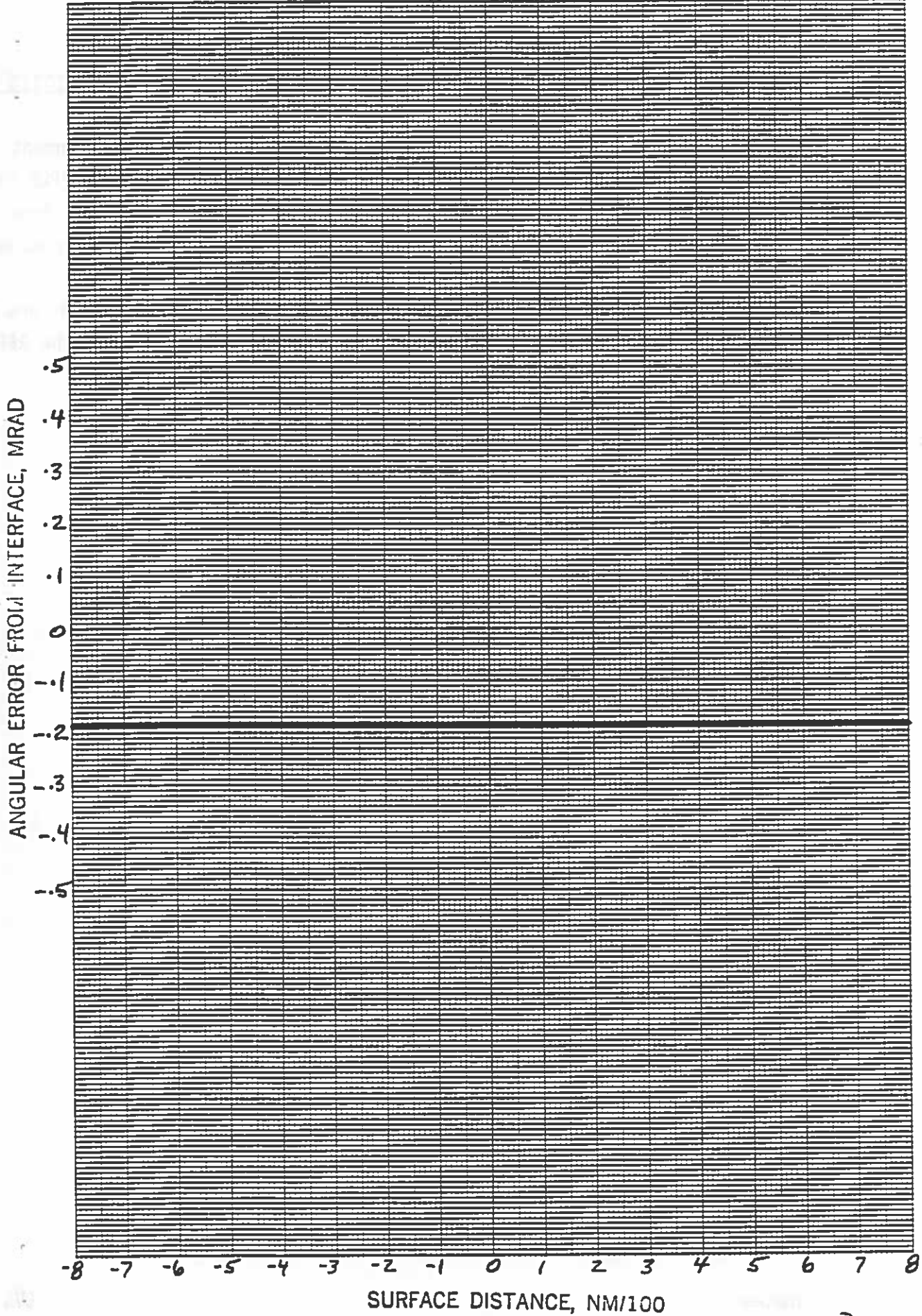
The curves are plotted as Error in milliradians from the OLS Interface Axes (essentially the spacecraft PMP axes), vs. ground surface distance along scan from subtrack (Nadir).

(An error of 0.1 milliradian at 450 naut. mi. altitude represents a ground position error of .045 naut. mi. at nadir.)

OLS #14A REFPLN ALIGNMENT



OLS #14A REFPLY SYNCHRONIZATION



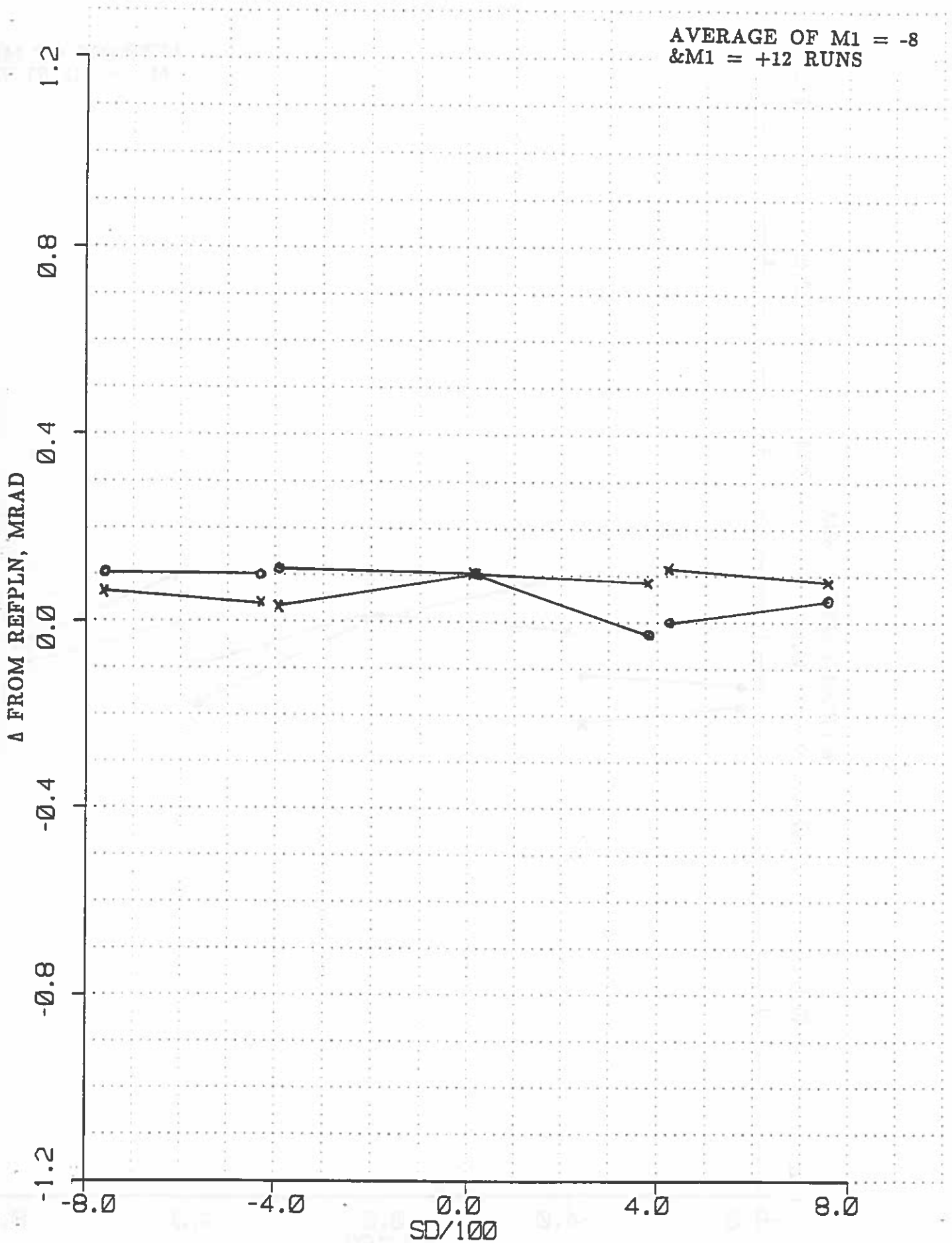
2. ALIGNMENT AND SYNCHRONIZATION FOR ALL MODES AT +5° SSS TEMP PLOTTED WITH RESPECT TO REFPLN

These curves are the difference between the Alignment and Synchronization curves at SSS = +5°C (Orbit Nominal) and the REFPLN Plot

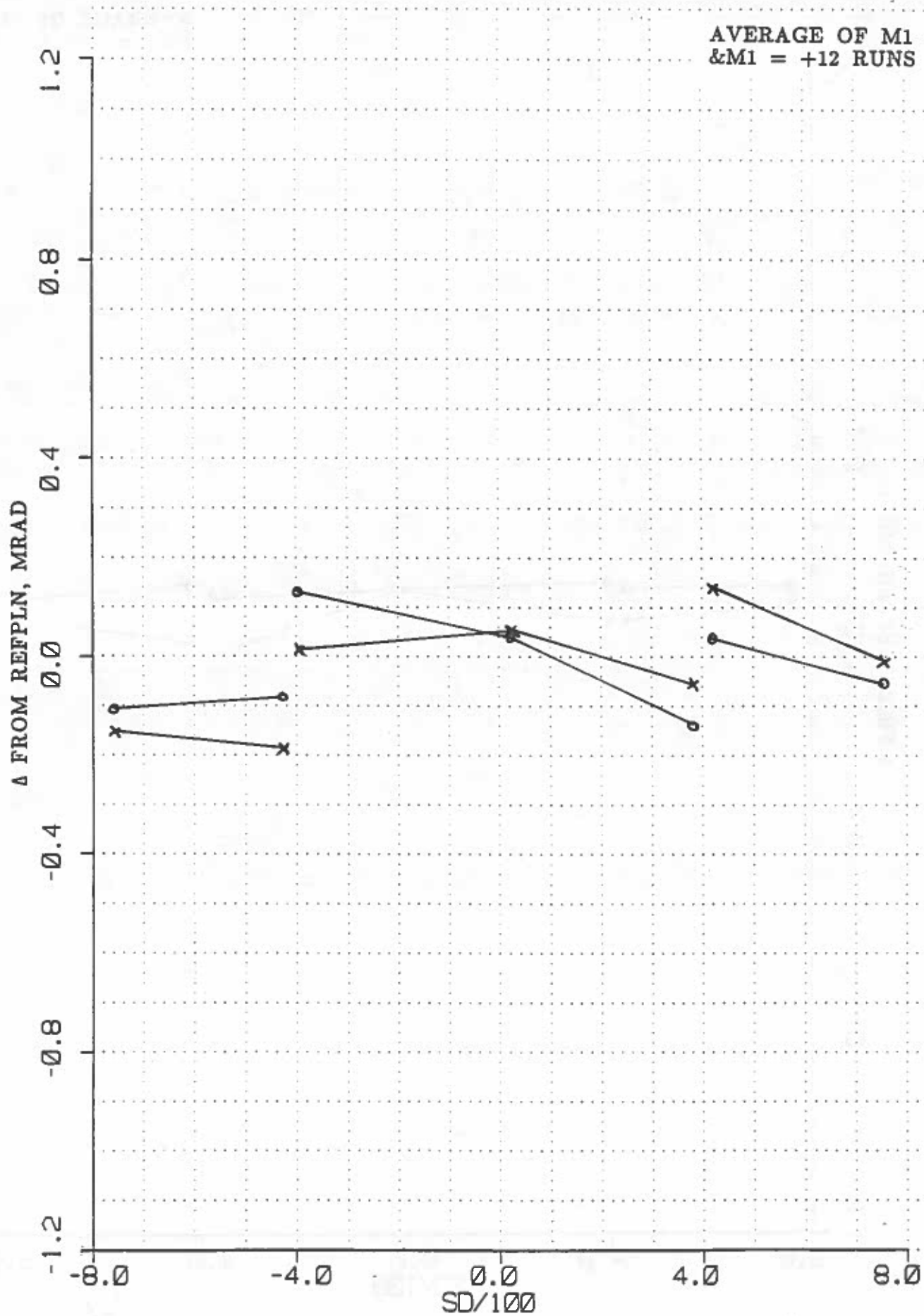
The curves represent the expected angular errors from the line-of-sight (REFPLN) axes for OLS data taken in the various modes of operation at orbit nominal conditions.

The curves were generated by averaging the data at M1 = -8° and M1 = +12° and then finding the difference between the average and the REFPLN

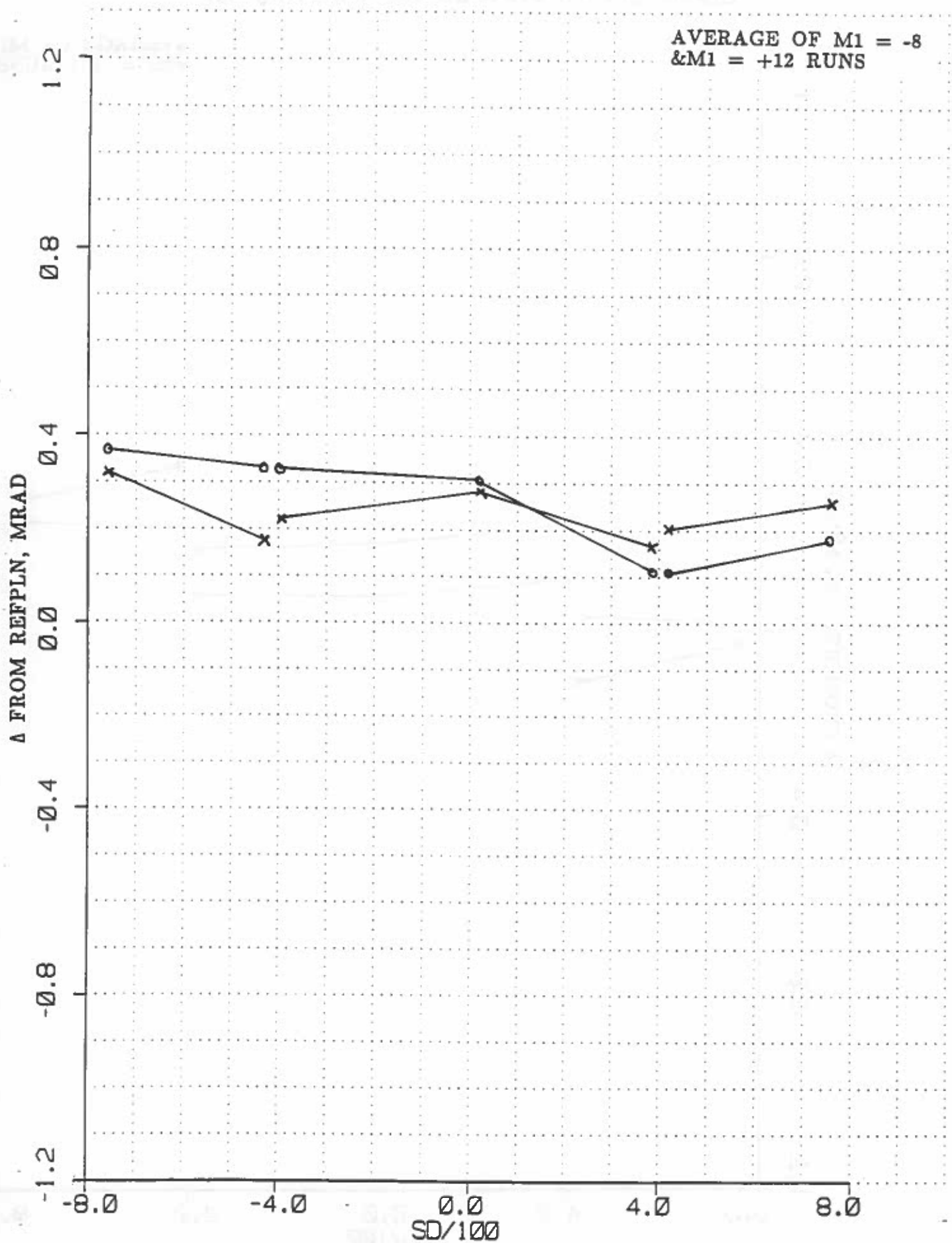
OLS#14 REFPLN ALIGNMENT - HRD SDF



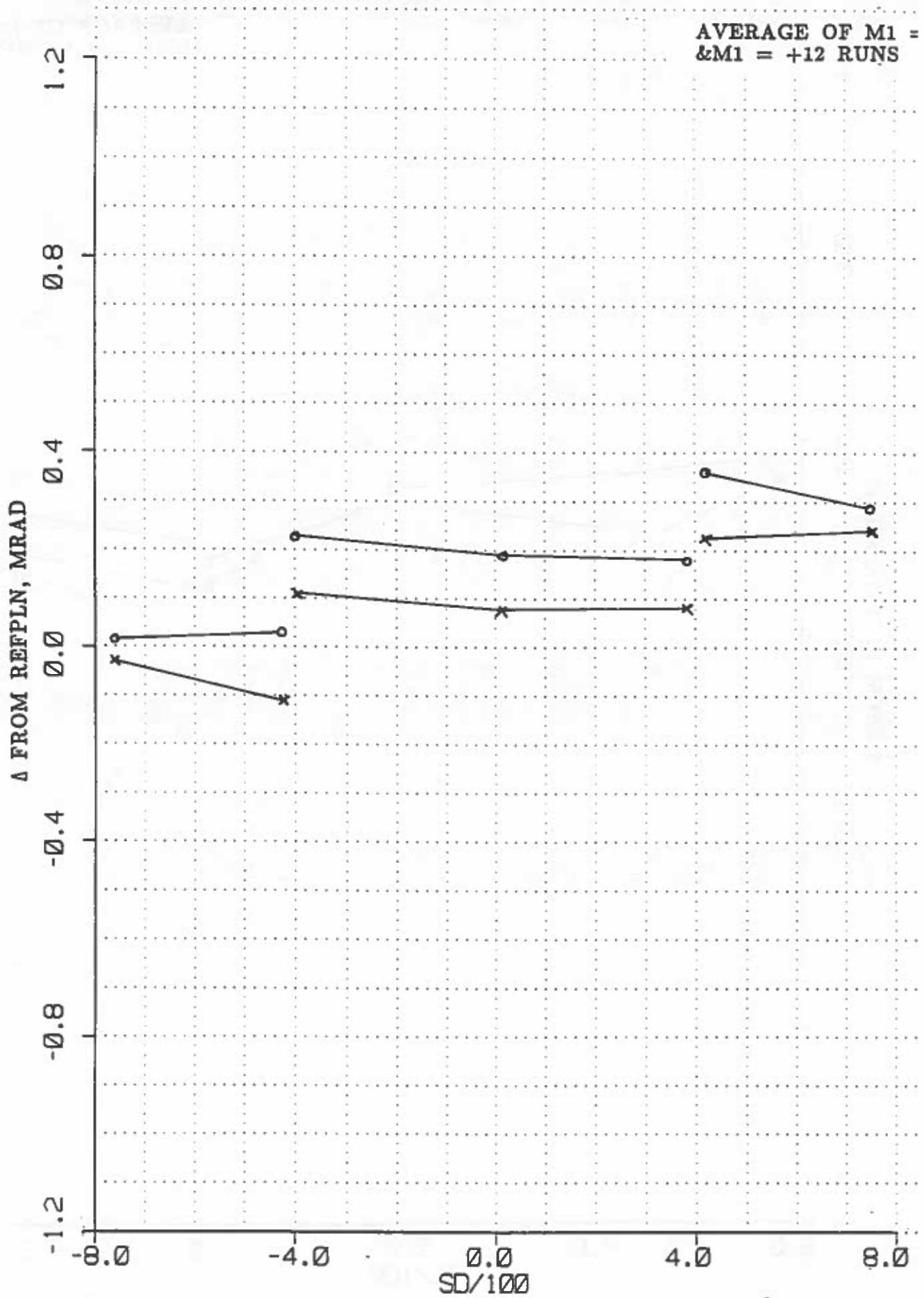
OLS#/4 REFPLN ALIGNMENT - T SDF



OLS#14 REFPLN ALIGNMENT - PMT RTDS

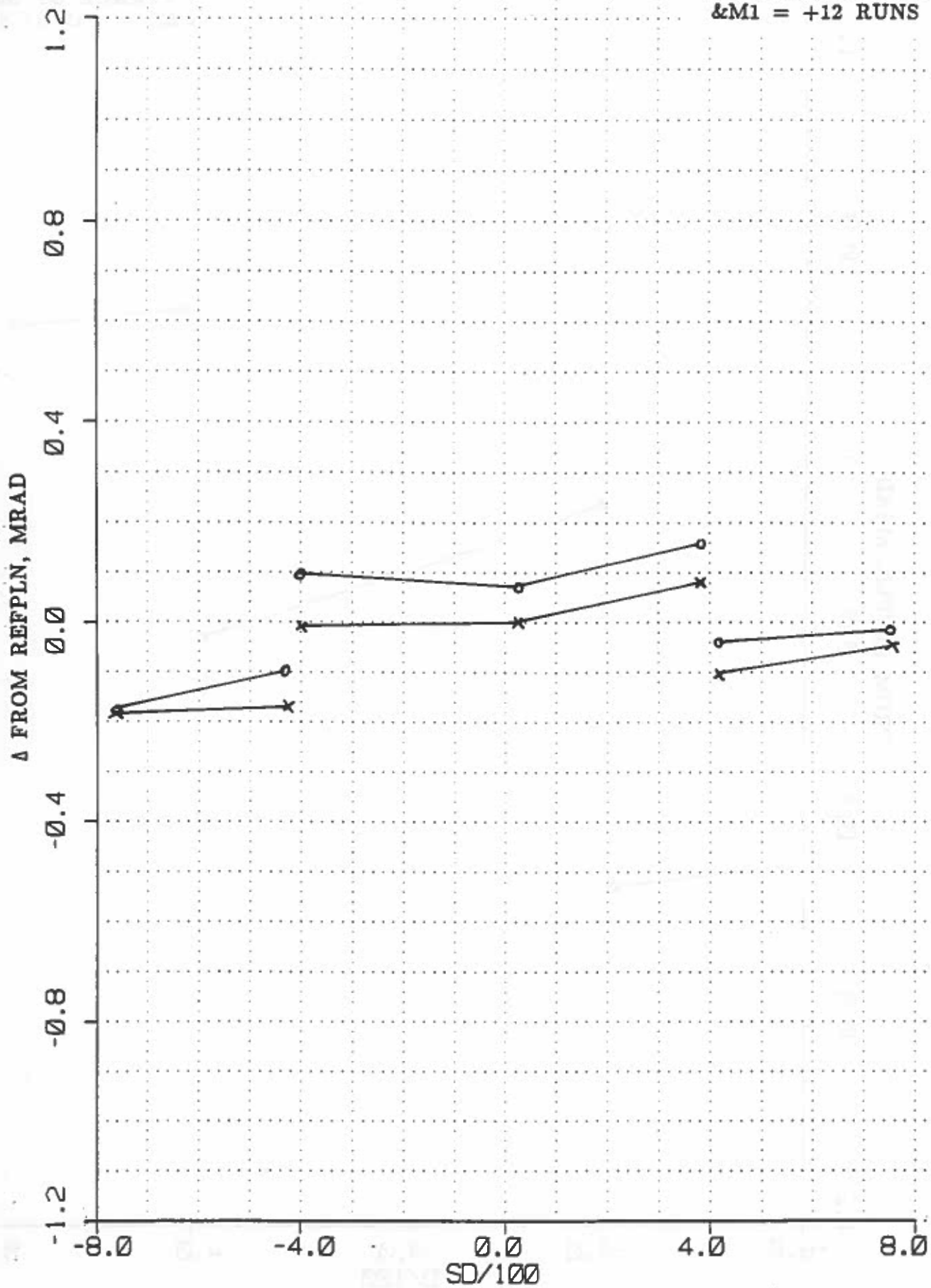


OLS#14 REFPLN SYNCHRONIZATION - HRD SDF



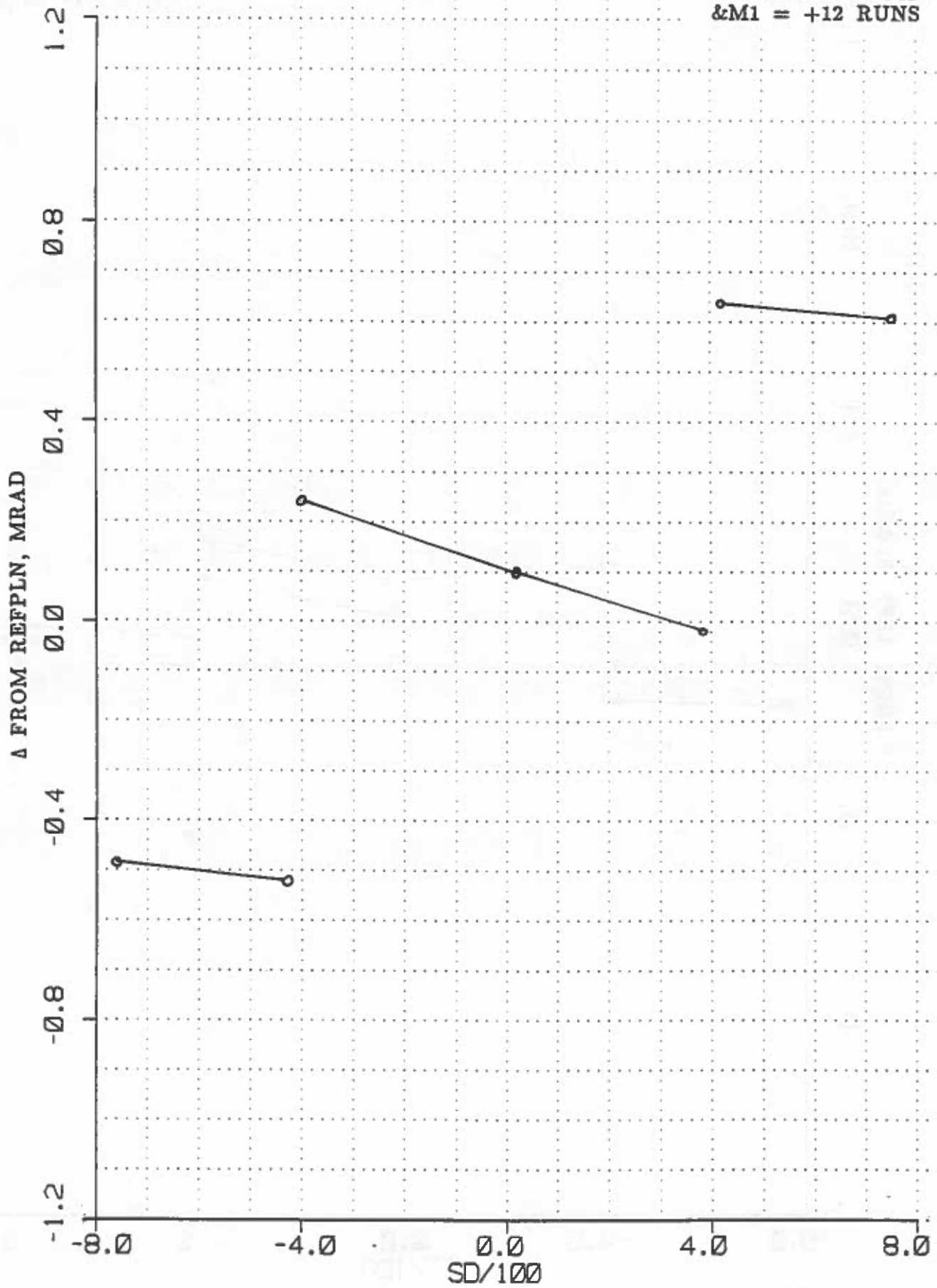
OLS#14 REFPLN SYNCHRONIZATION-T SDF

AVERAGE OF M1 = -8
&M1 = +12 RUNS



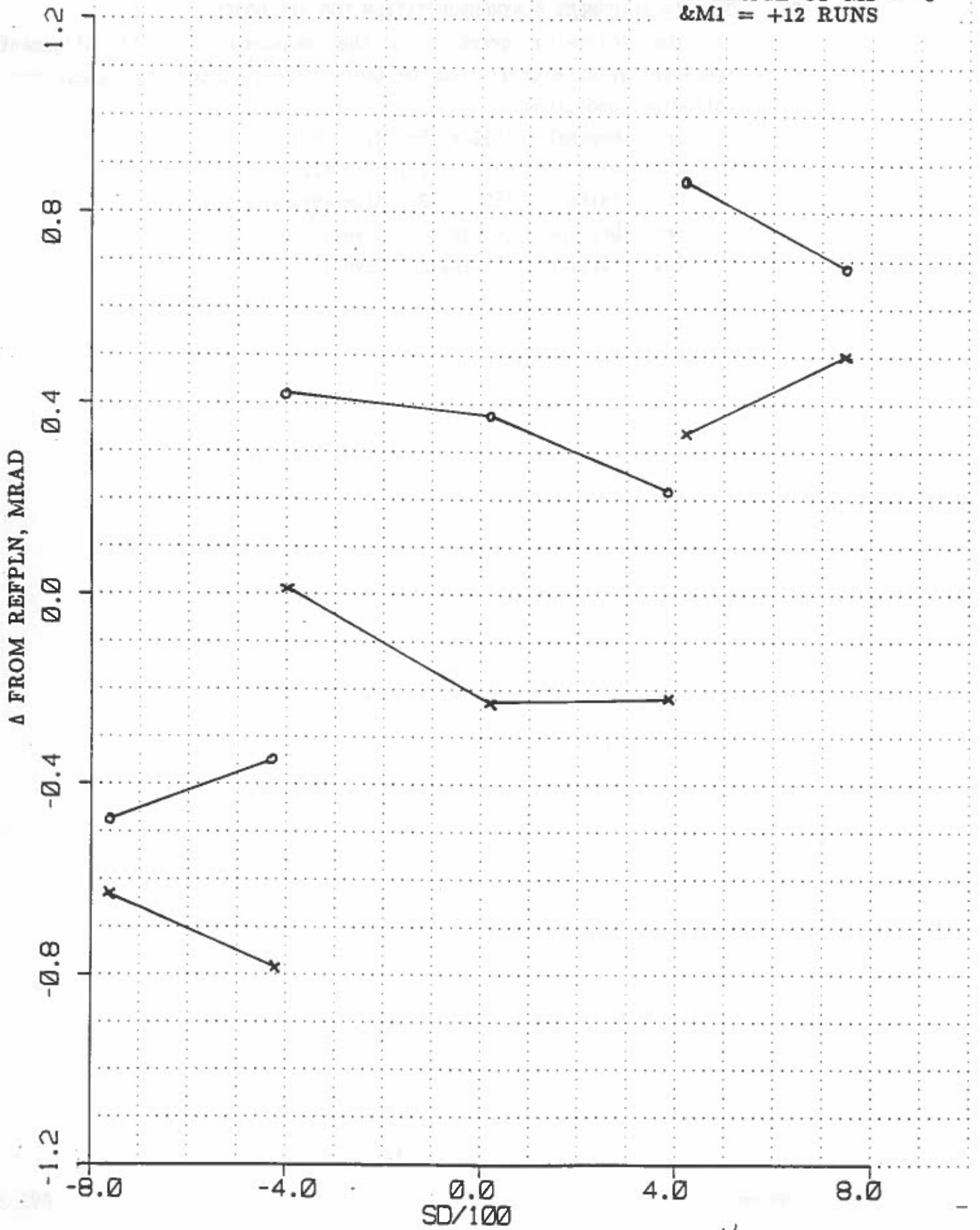
OLS#14 REFPLN SYNCHRONIZATION - PMT SDS

AVERAGE OF M1 =
&M1 = +12 RUNS



OLS#14 REFPLN SYNCHRONIZATION - PMT RTDS

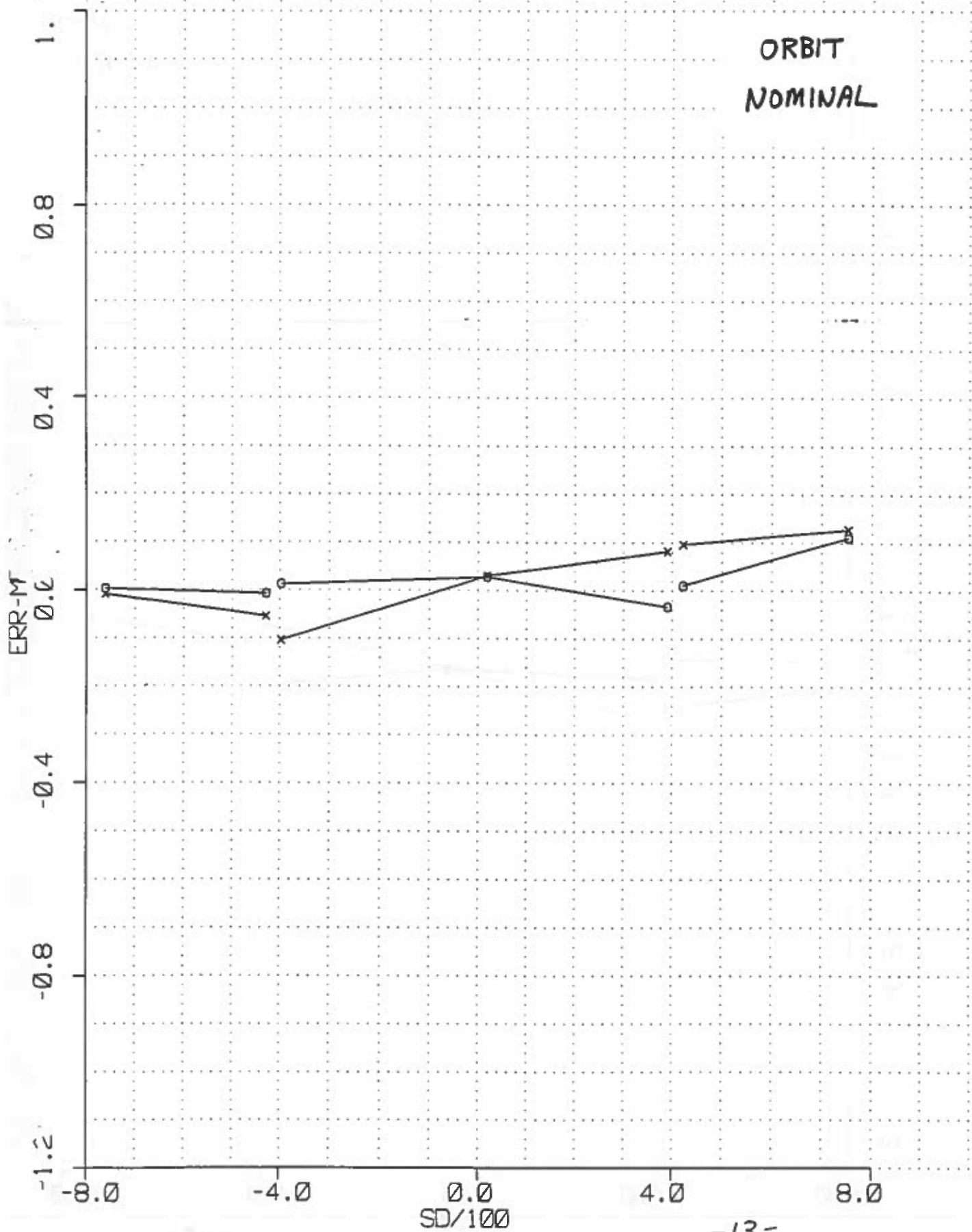
AVERAGE OF M1 = -8
&M1 = +12 RUNS



3. OLS #14 ALIGNMENT & SYNCHRONIZATION FOR ALL MODES

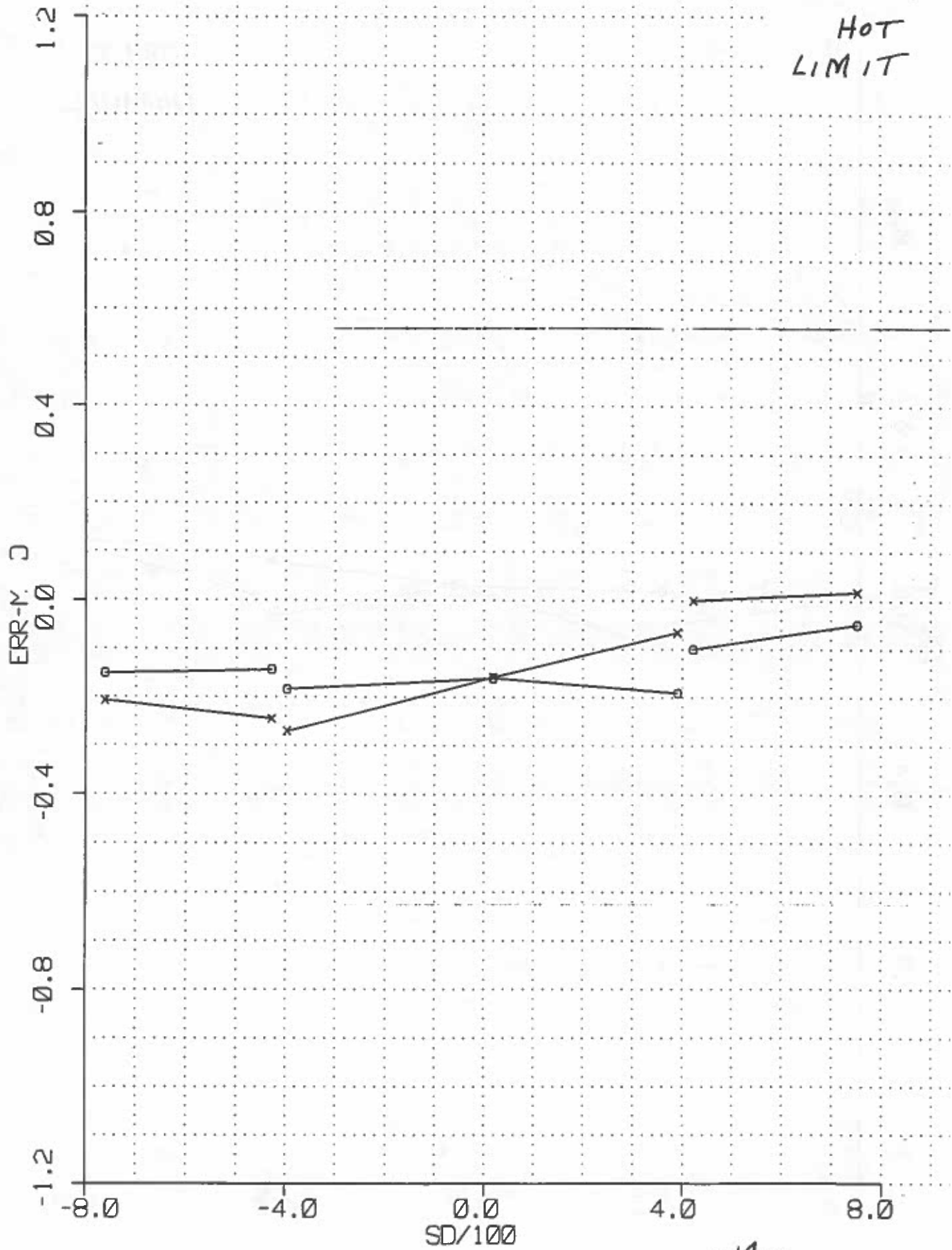
The following graphs are the measured OLS #14 Alignment Synchronization with respect to the mounting (Interface) axes, for the following conditions.

Orbit Nominal (SSS = +5°, M1 = -8°)
Hot Limits (SSS = +7°, M1 = +12°)
Cold Limits (SSS = +3°, M1 = -8°)
Pre Vibration (Acceptance Level)
Post Vibration (Acceptance Level)



ORBIT
NOMINAL

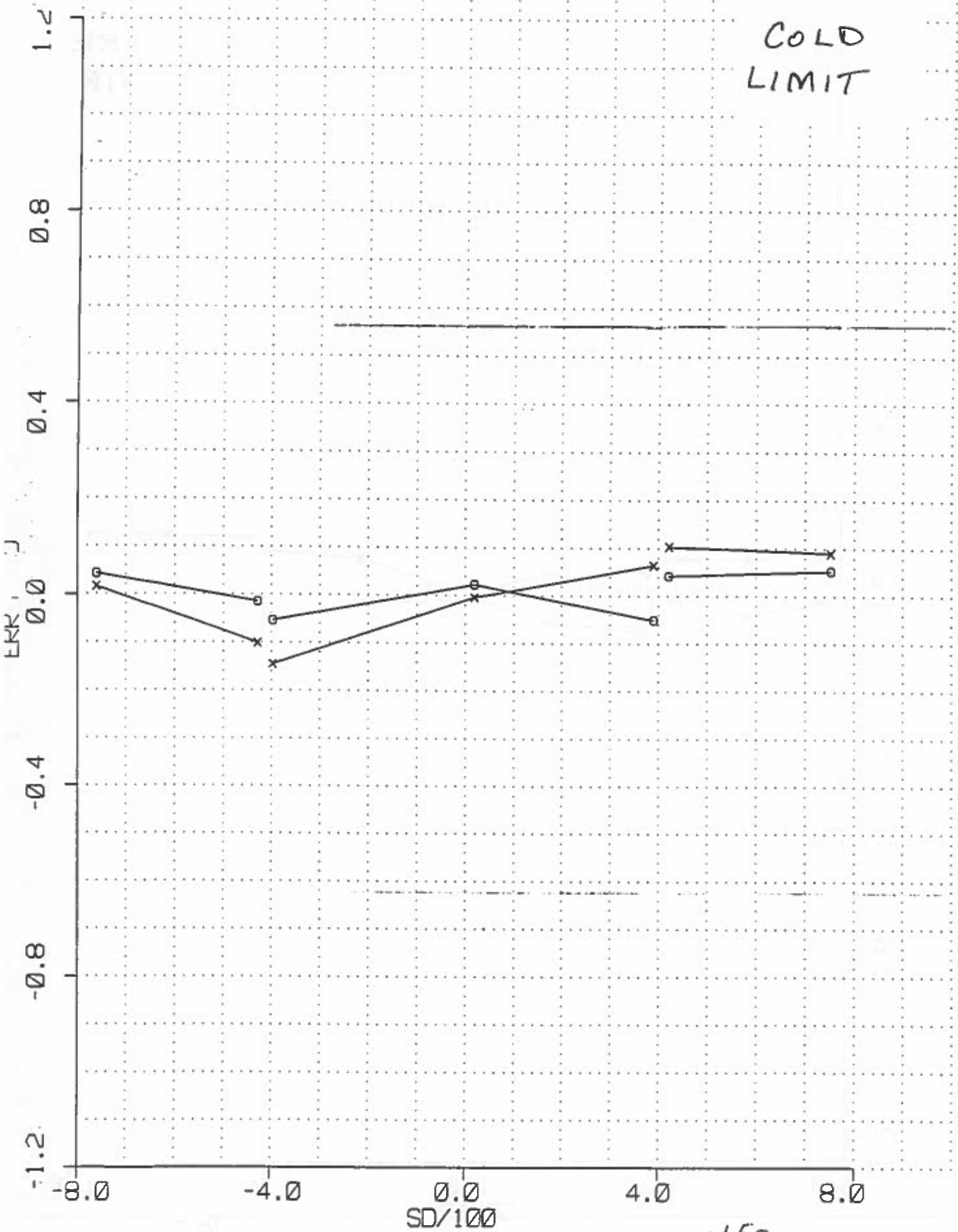
SYSTEM 14
IMC-NORM HRD ALIGN SDF SSS= 7 ,M1= 12 ,DATE: 320



SYSTEM 14

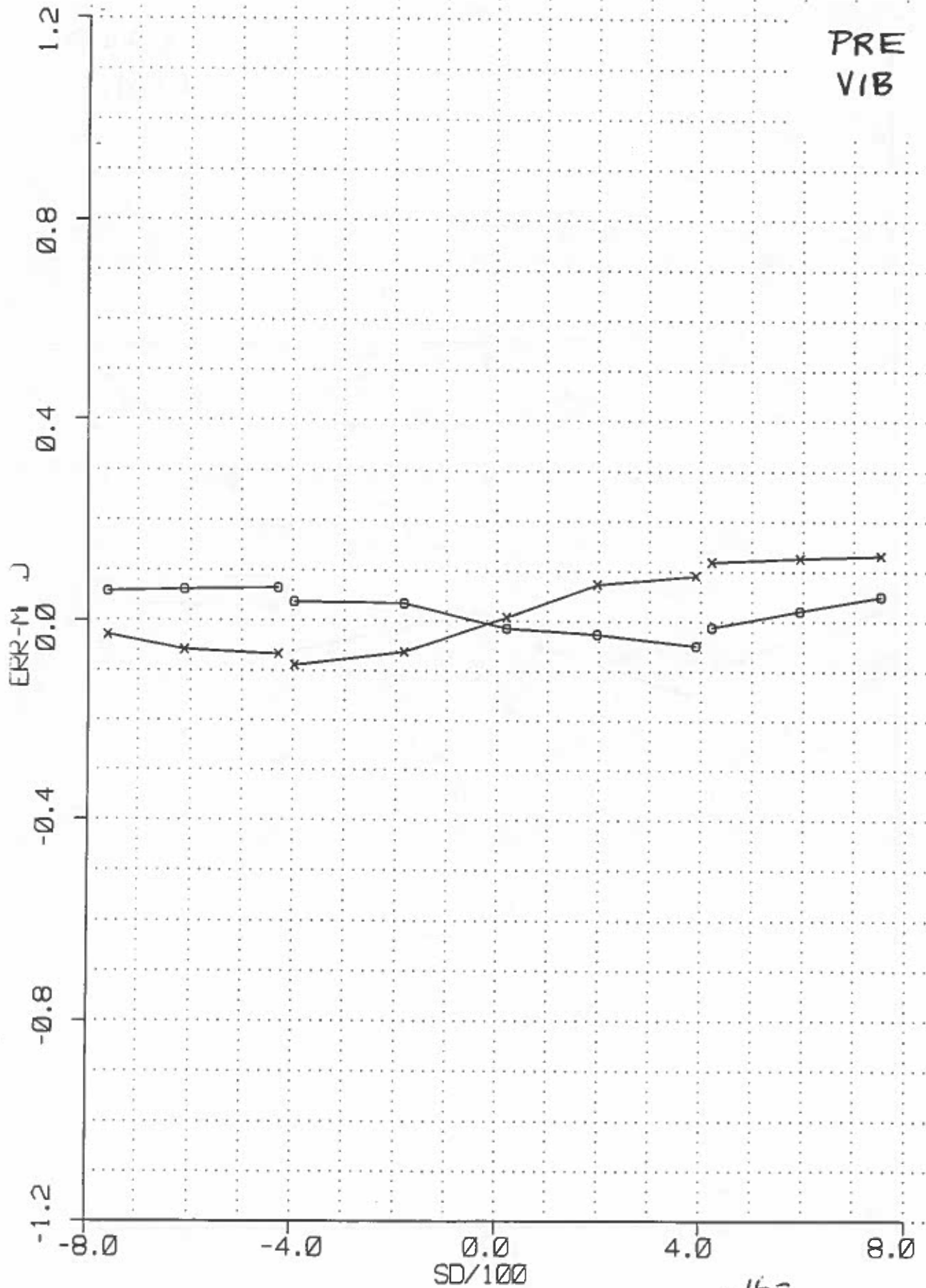
IMC-NORM HRD ALIGN SDF SSS=3 ,M1=-8 ,DATE: 324

COLD
LIMIT

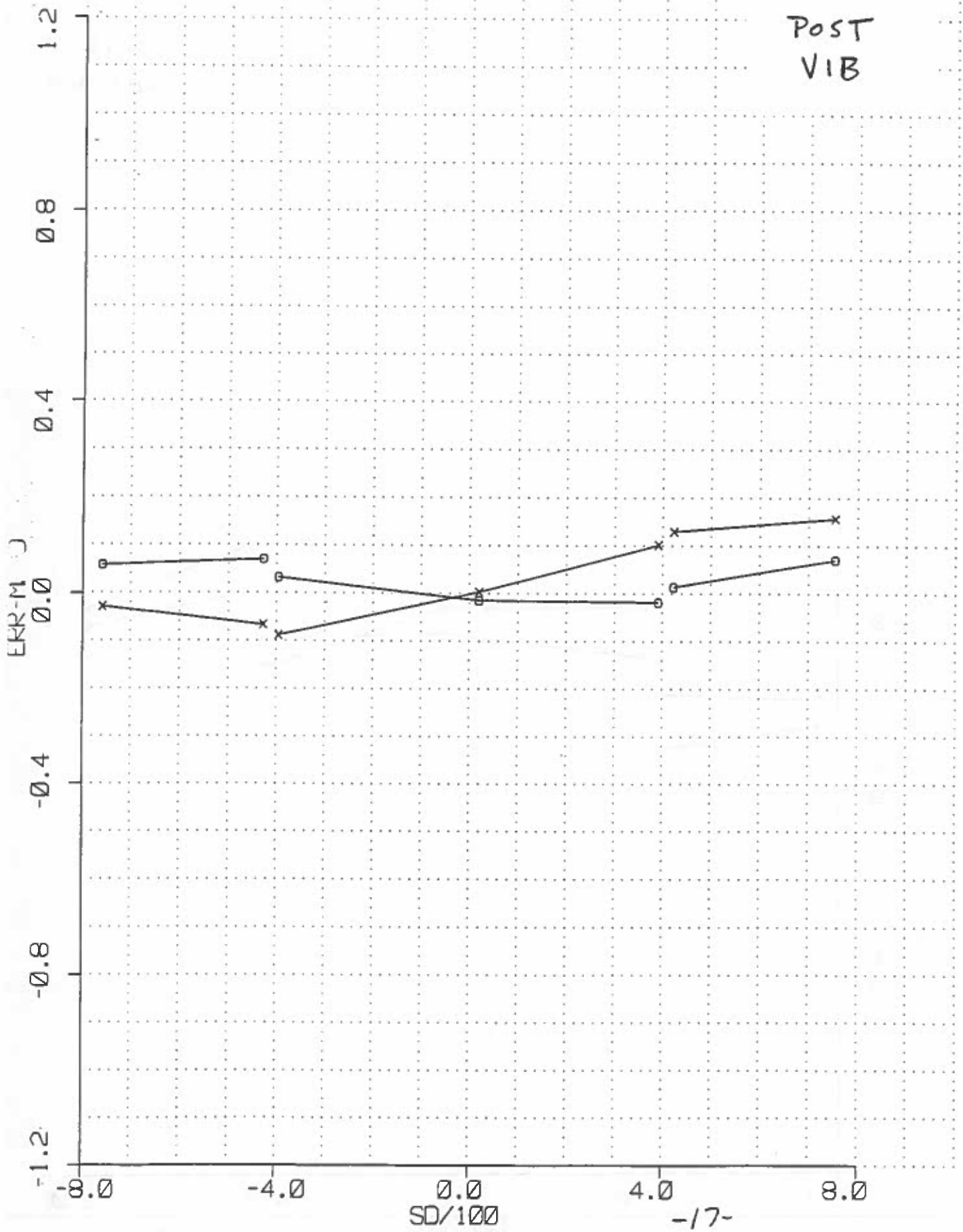


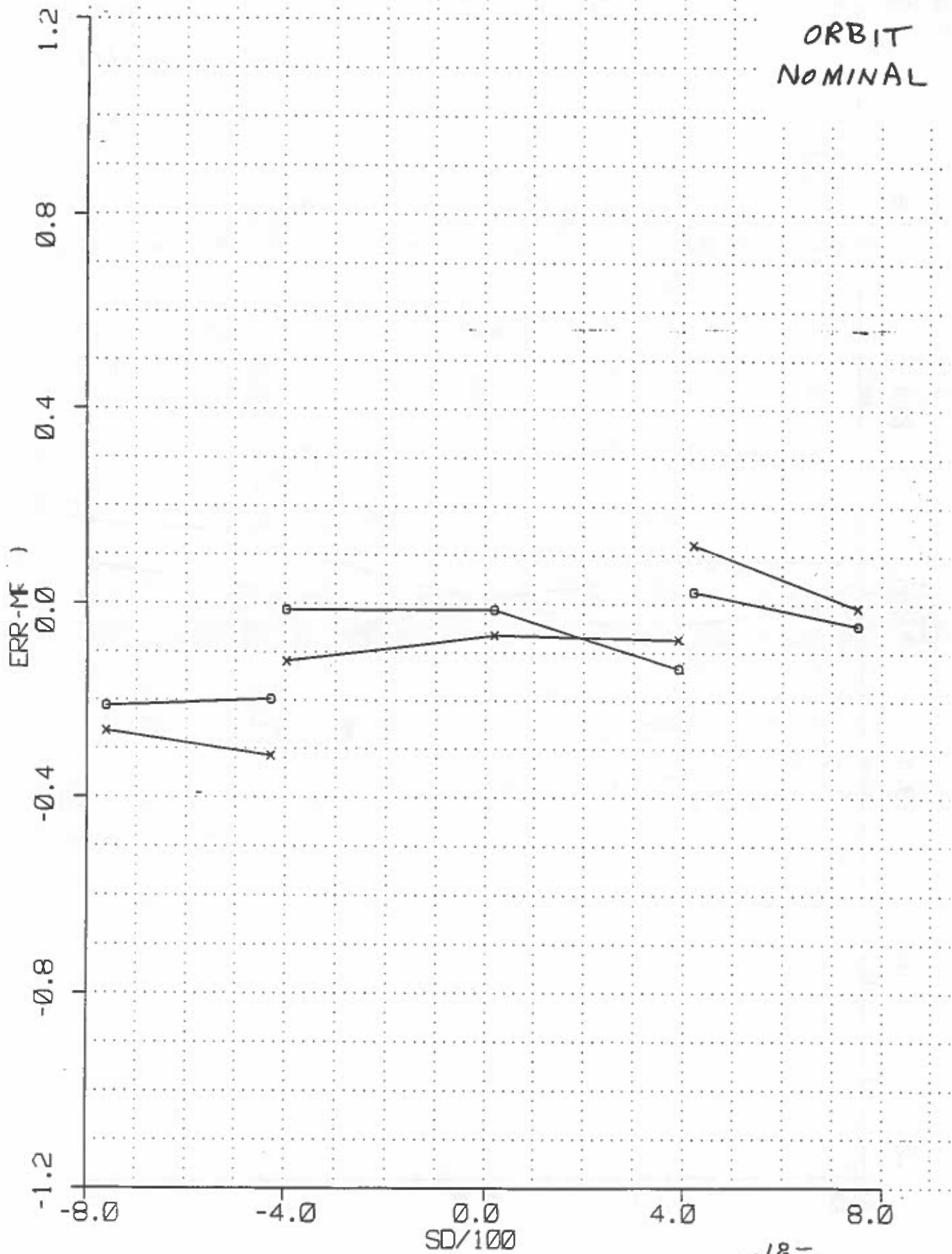
SYSTEM 14

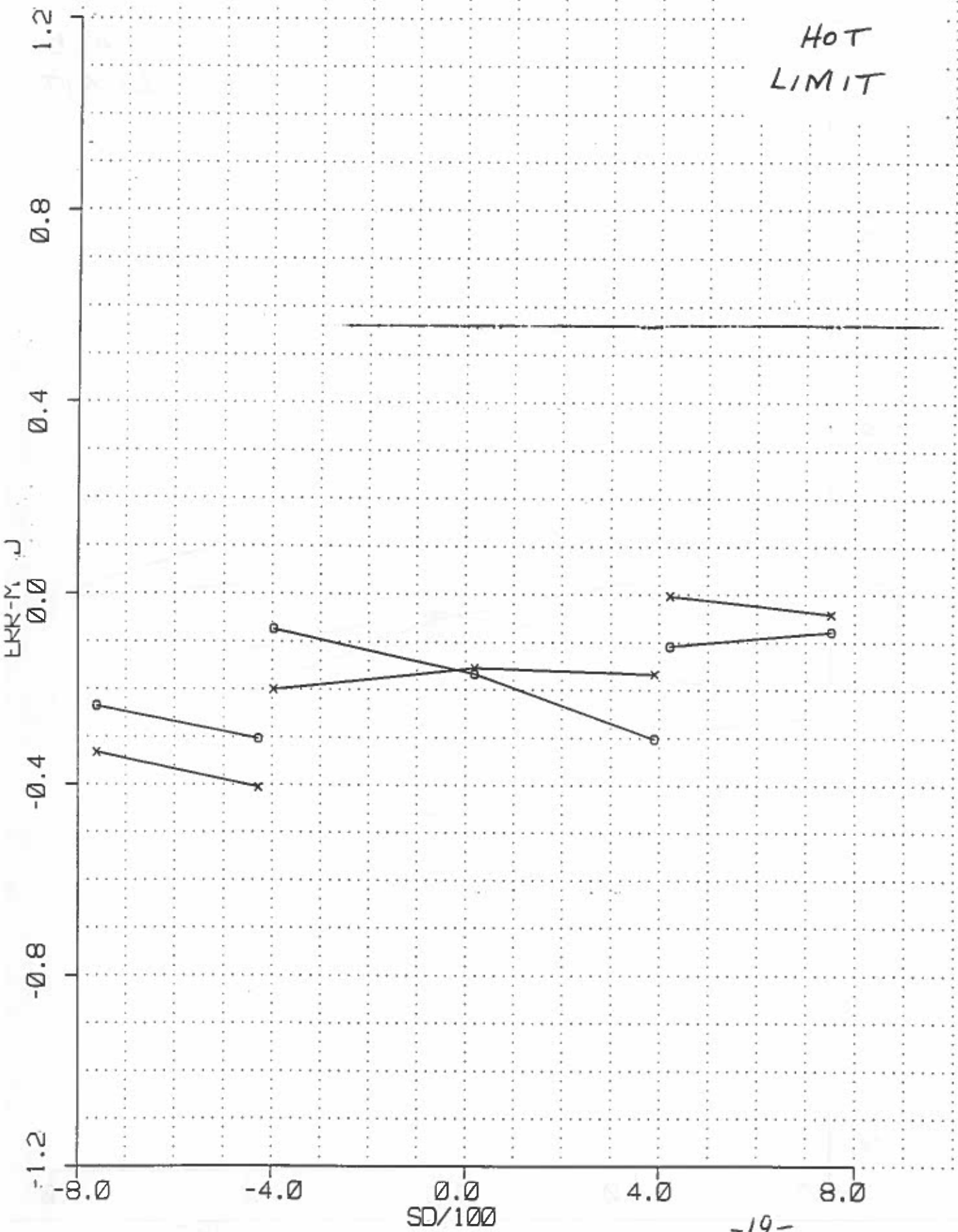
IMC-NORM HRD ALIGN SDF SSS=23, M1=23, DATE: 1123



POST
VIB







SYSTEM 14

IMC-NORM

T

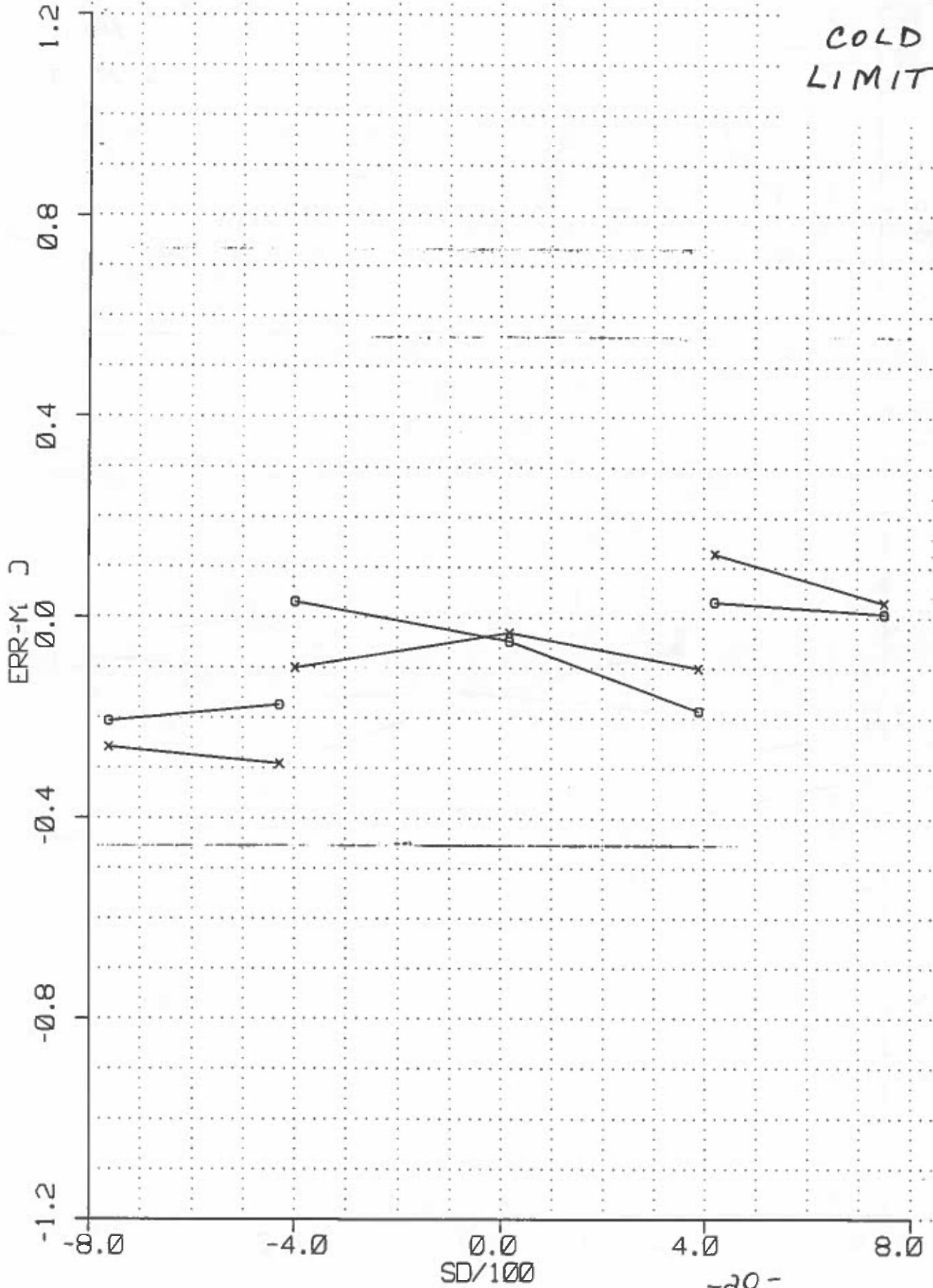
ALIGN

SDF

SSS=3

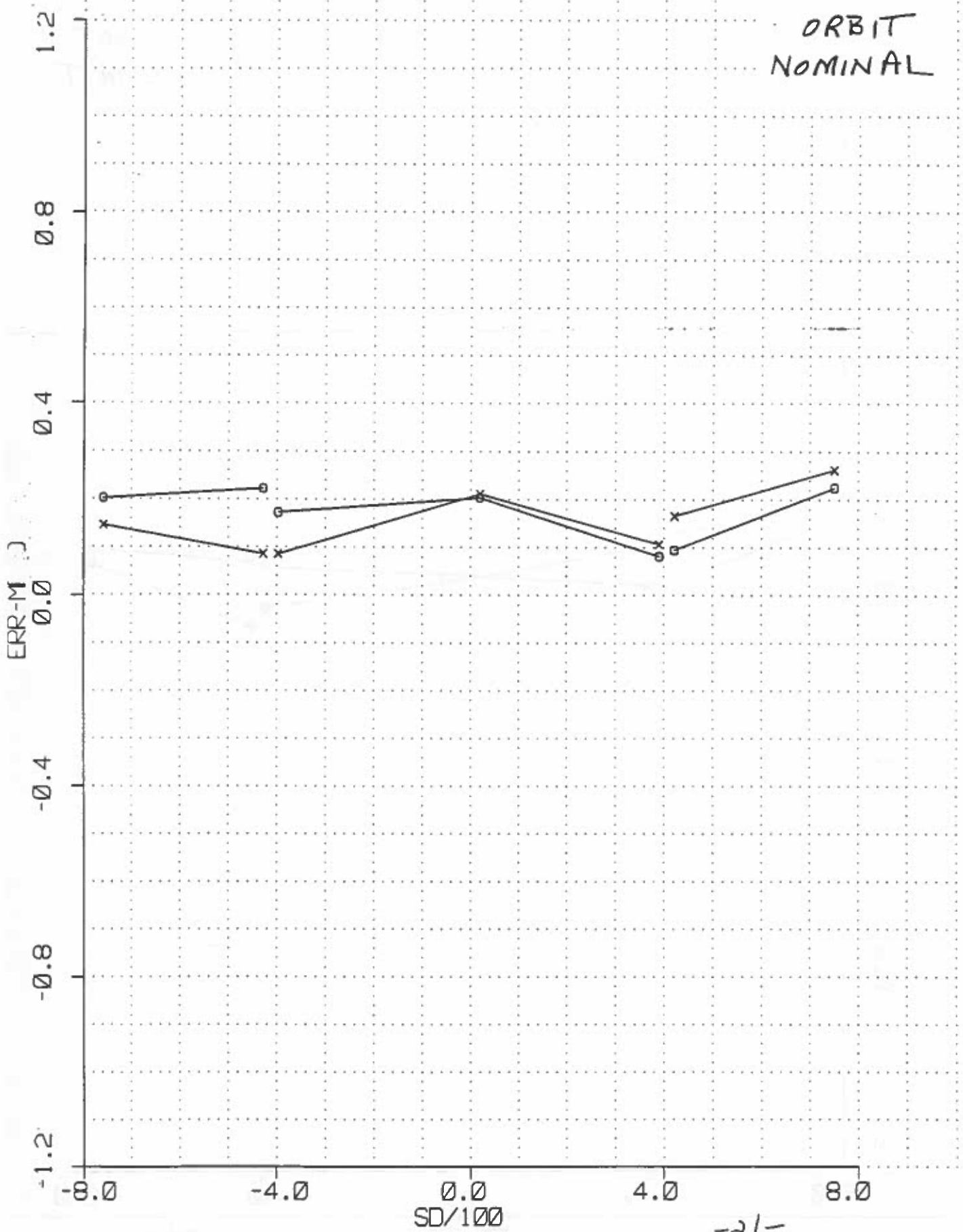
,M1=-8

,DATE: 324

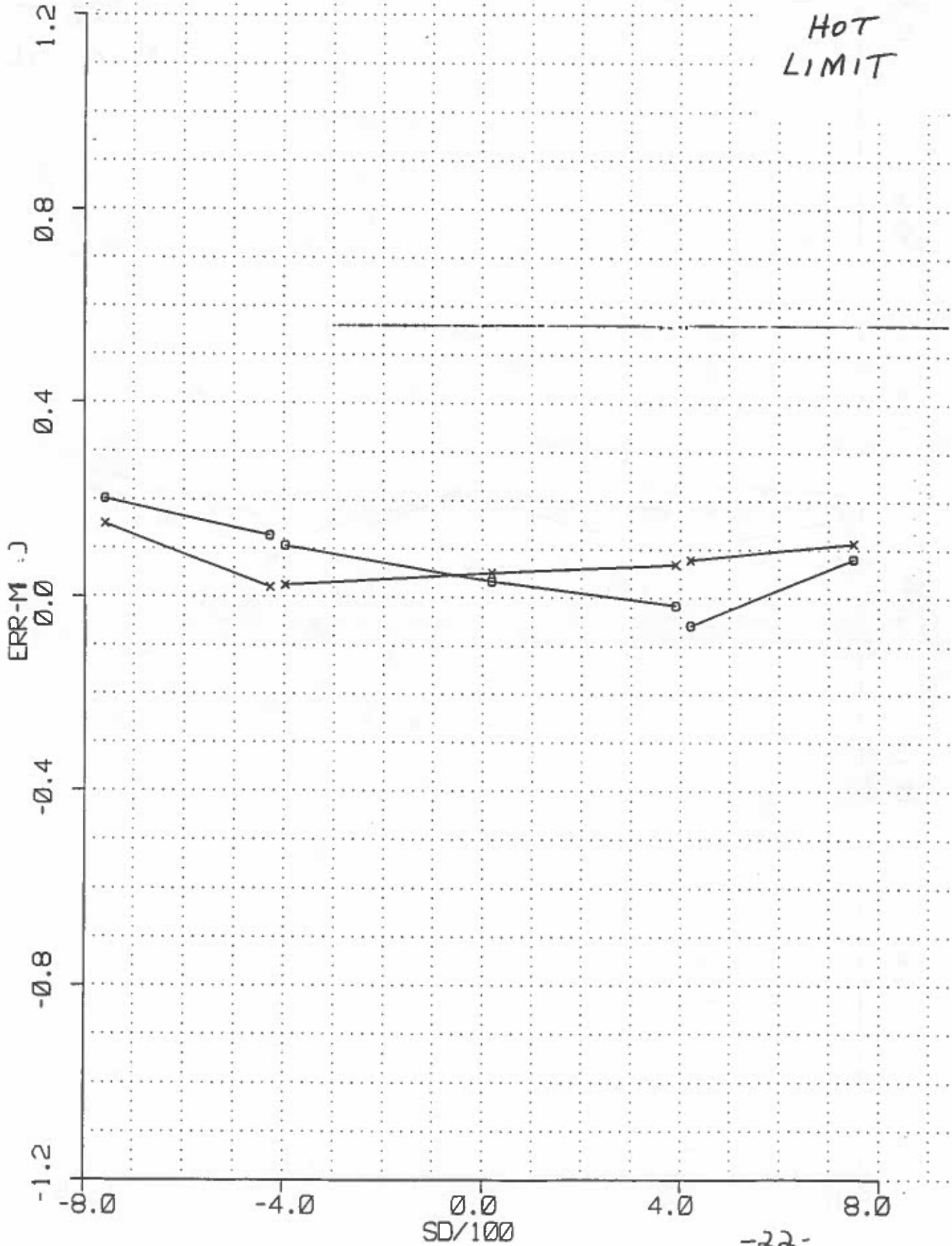


SYSTEM 14

IMC-NORM PMT ALIGN RTD-S SSS= 5 ,M1= -8 ,DATE: 401

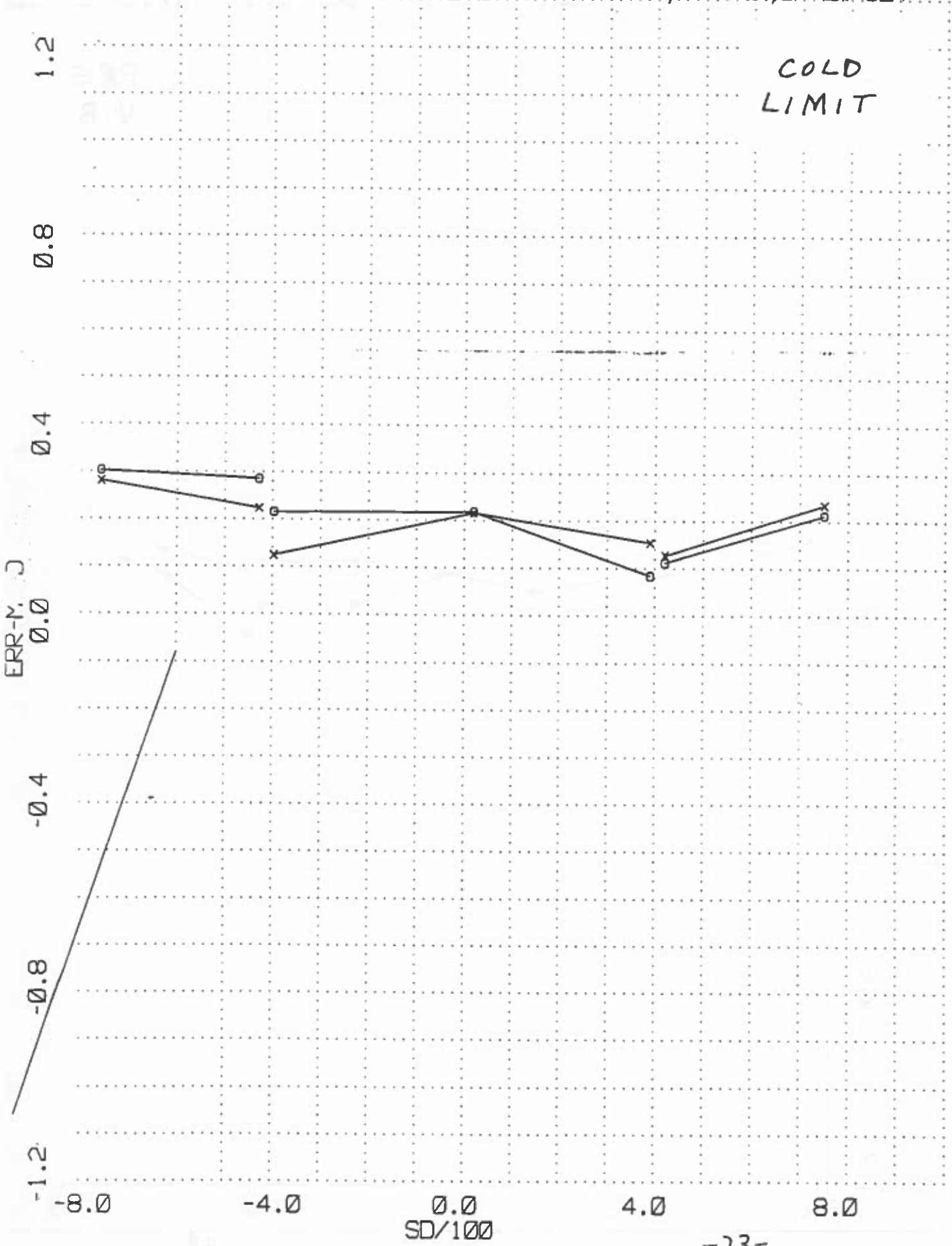


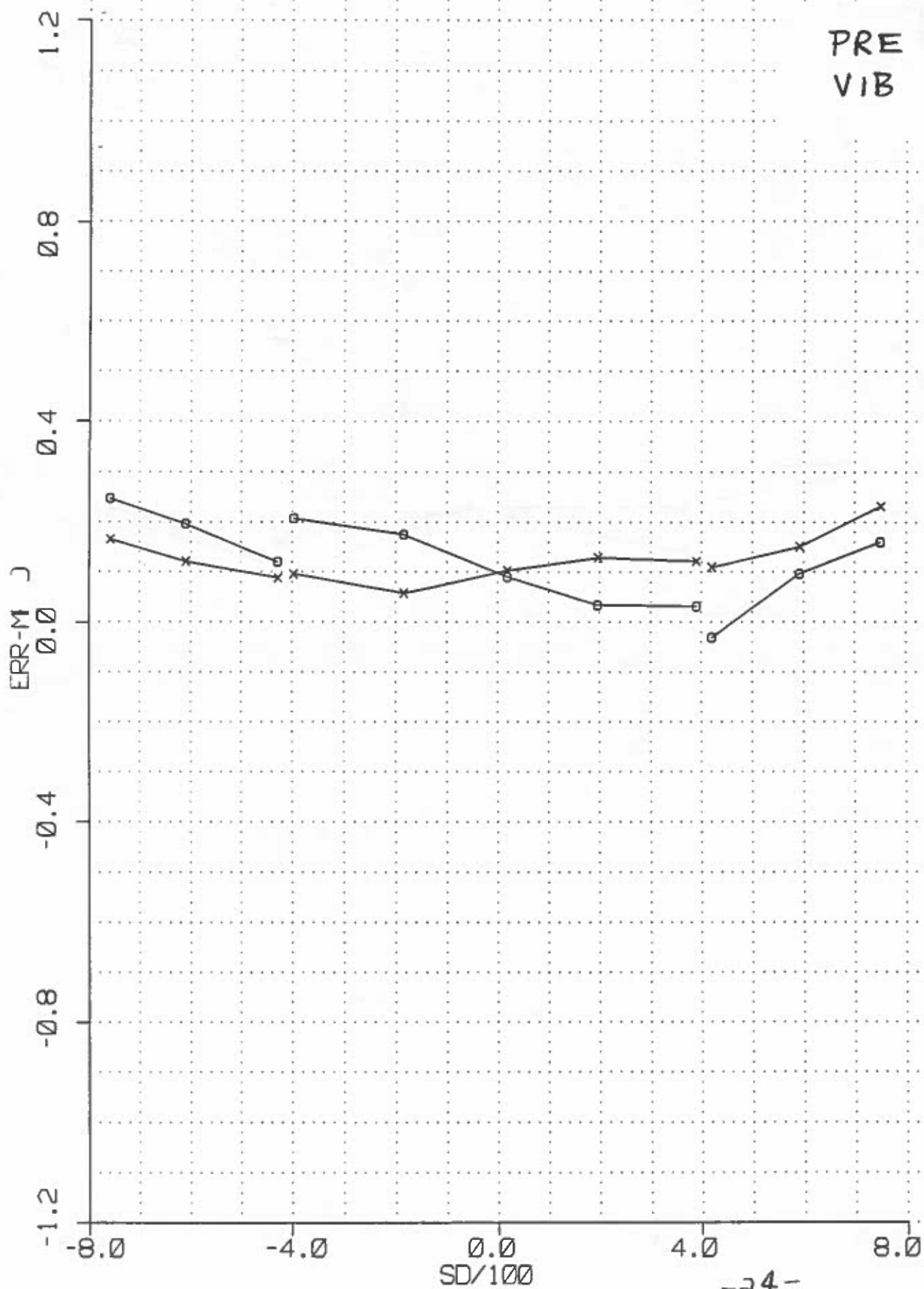
SYSTEM 14
IMC-NORM PMT ALIGN RTD-S SSS=7 ,M1=12 ,DATE: 320



SYSTEM 14

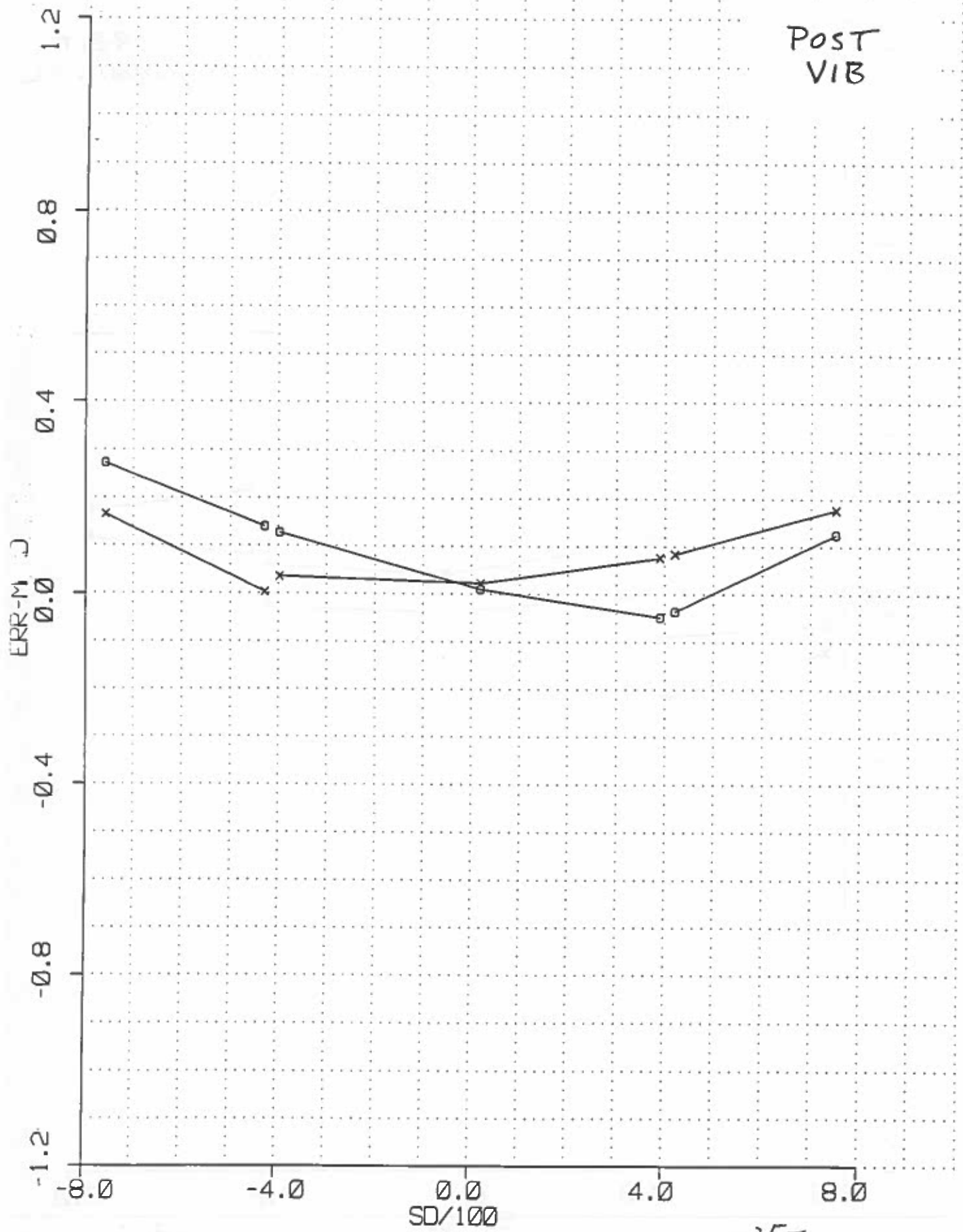
IMC-NORM PMT ALIGN RTD-S SSS=3.,M1=-8.,DATE: 324

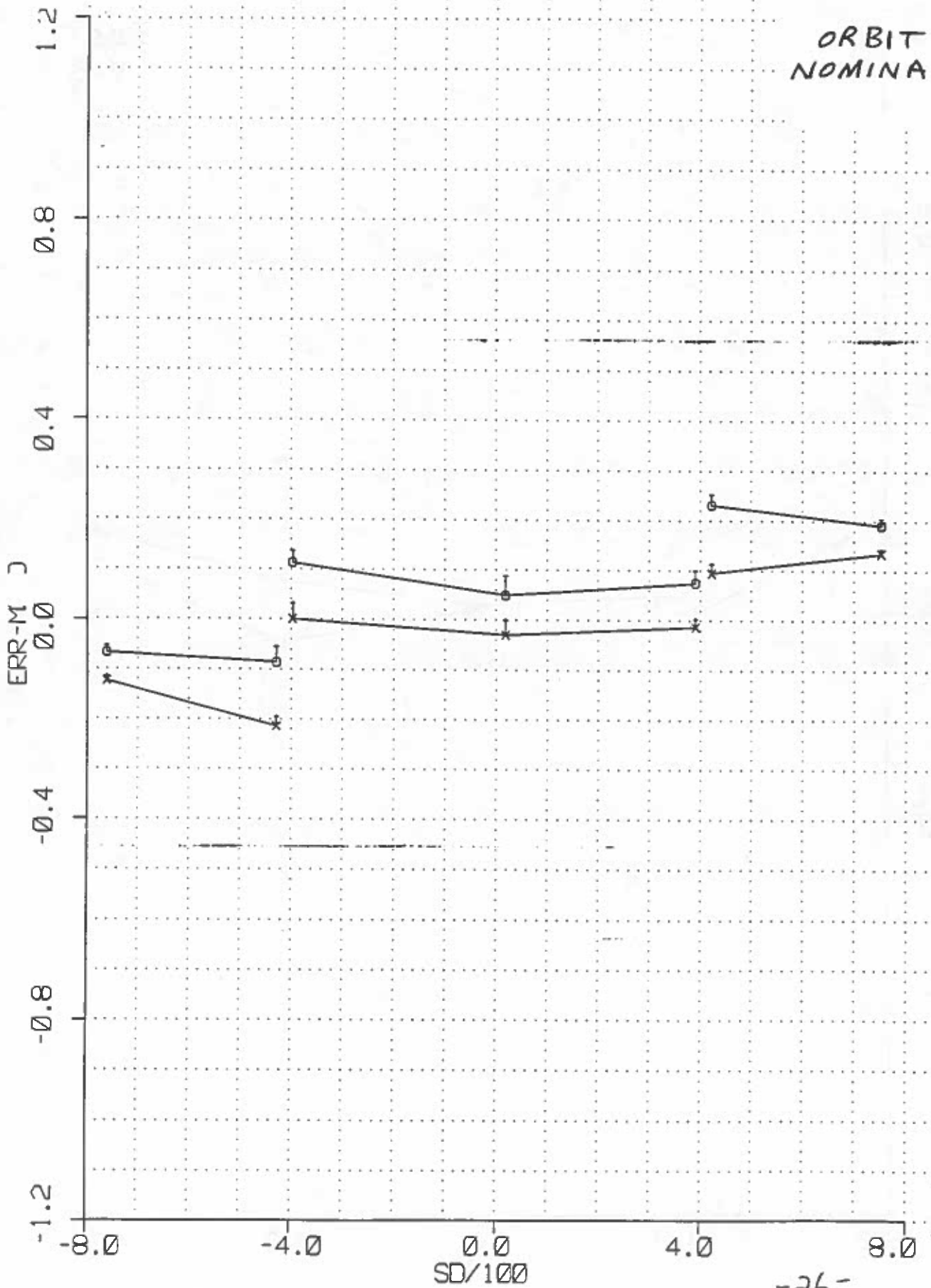




SYSTEM 14

IMC-NORM PMT ALIGN RTD-S SSS=23 ,M1=24 ,DATE: 1219

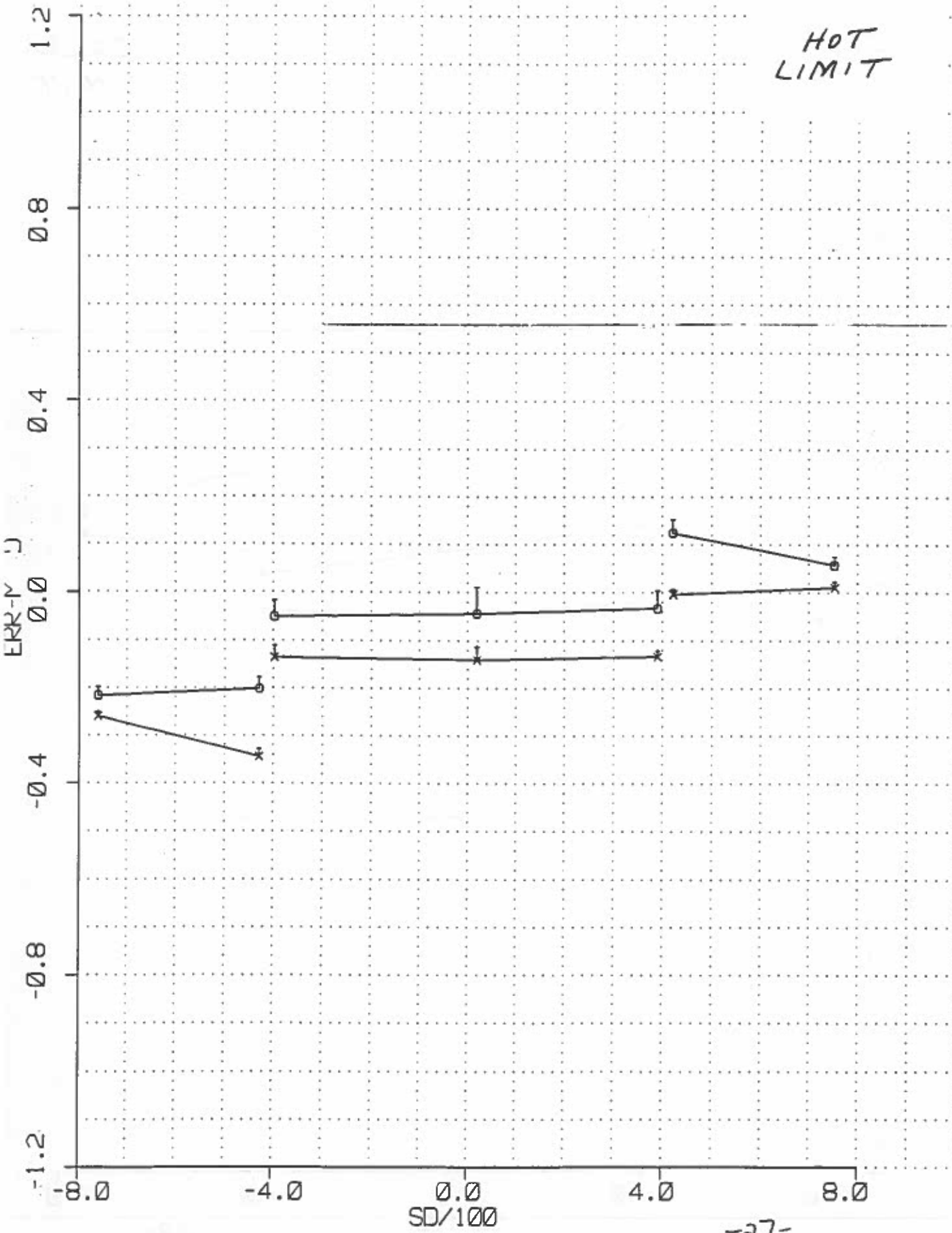


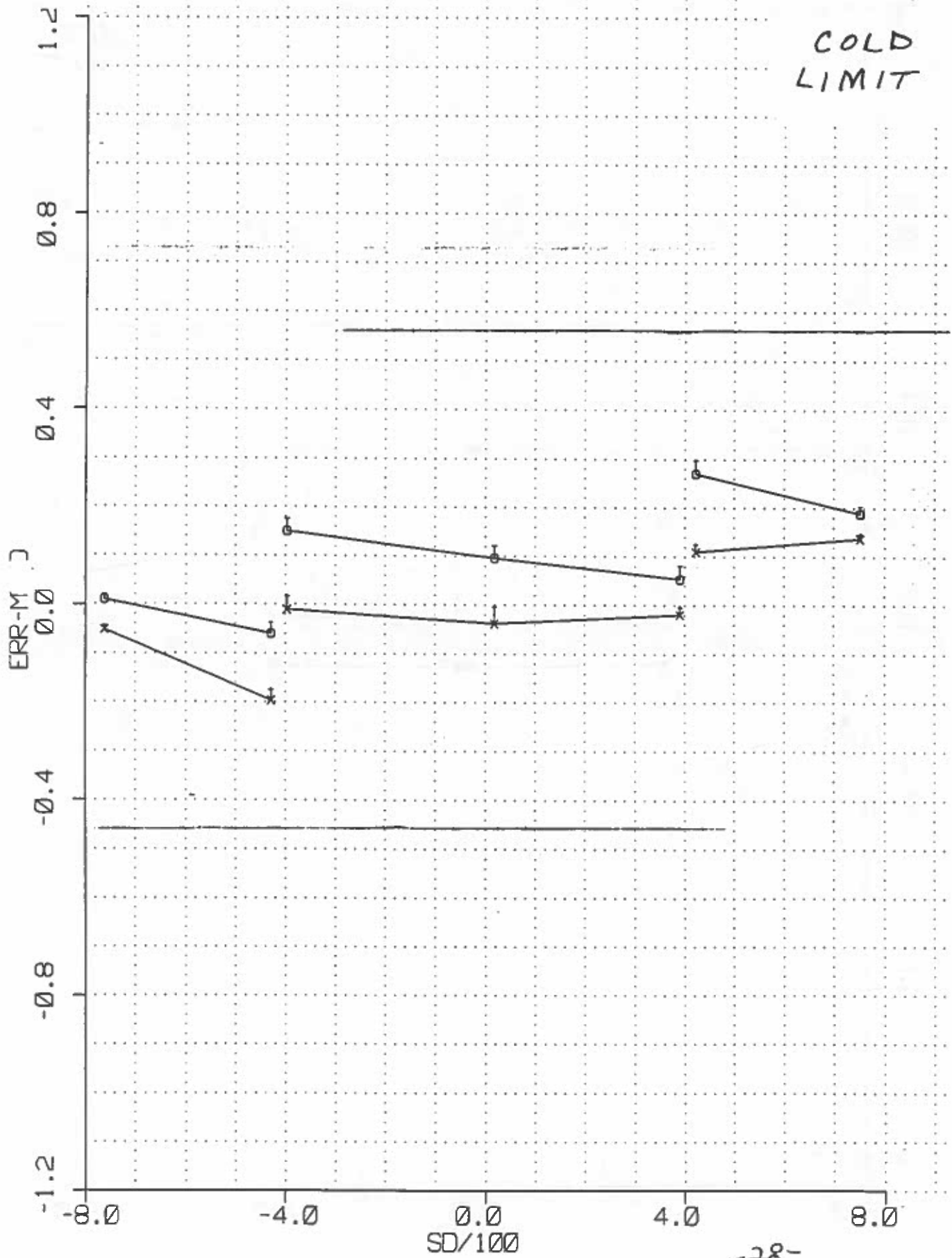


SYSTEM 14

IMC-NORM HRD SYNC SDF SSS=7 ,M1=12 ,DATE: 320

HOT
LIMIT





SYSTEM 14

IMC-NORM

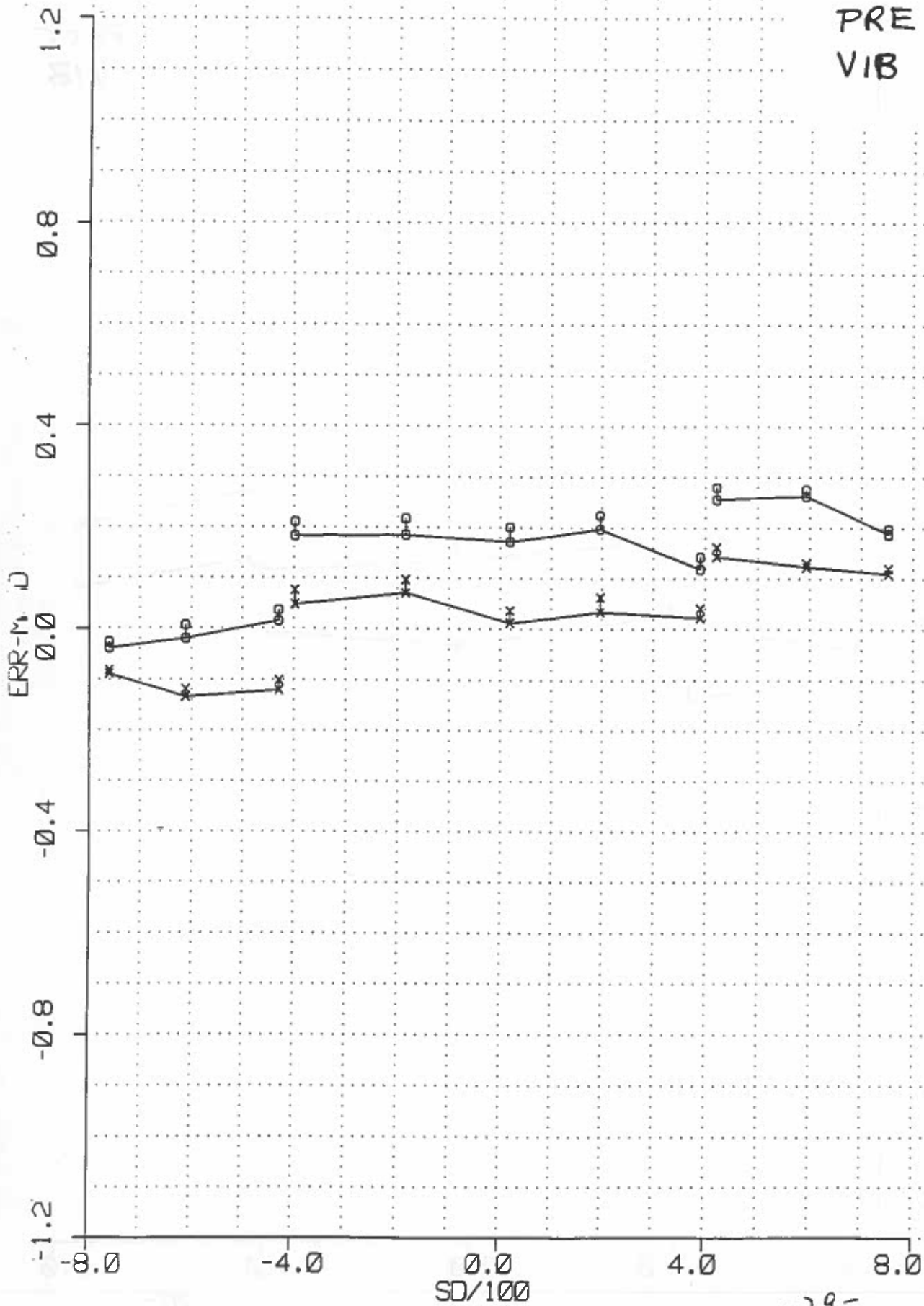
HRD

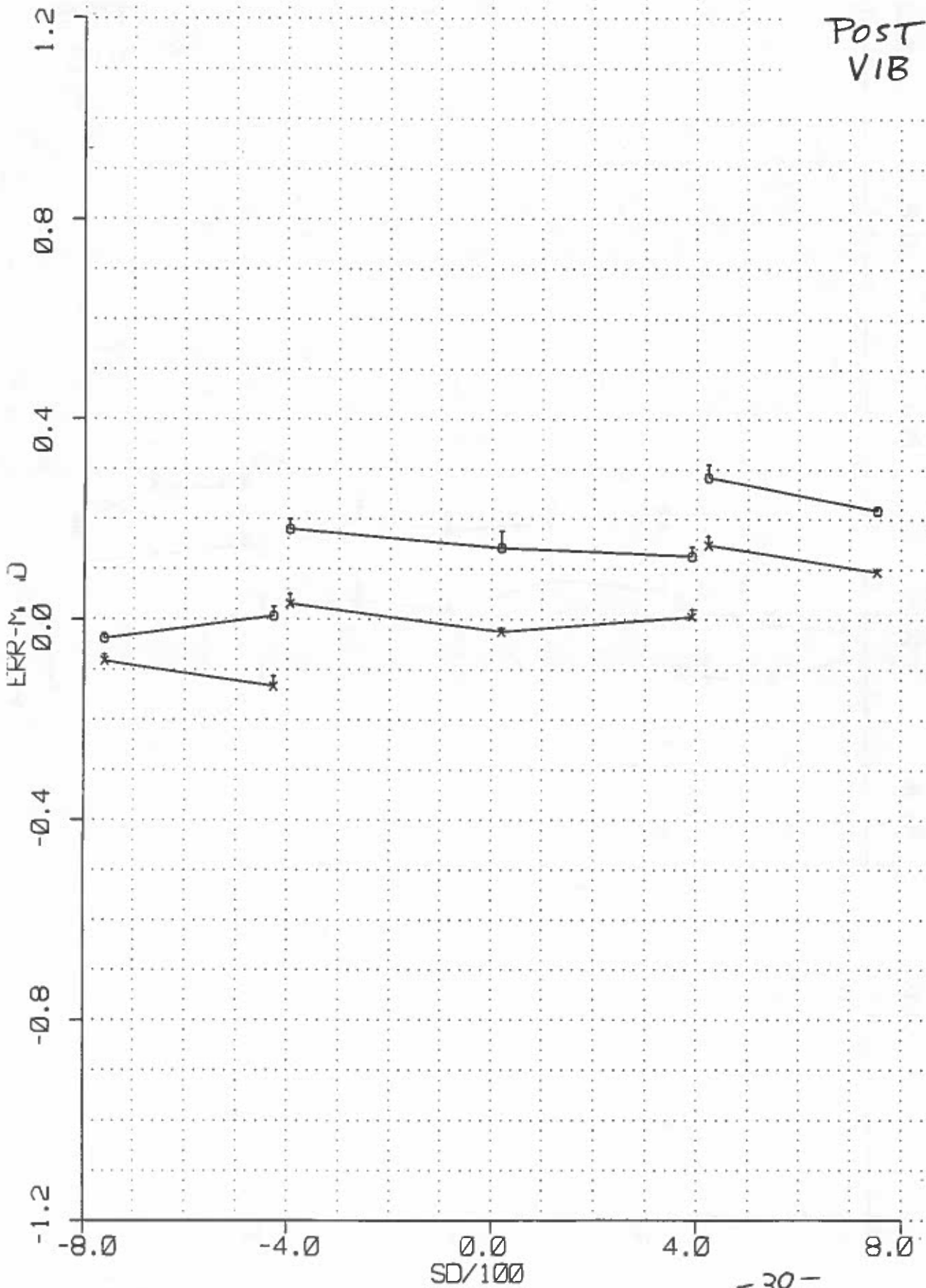
SYNC

SDF

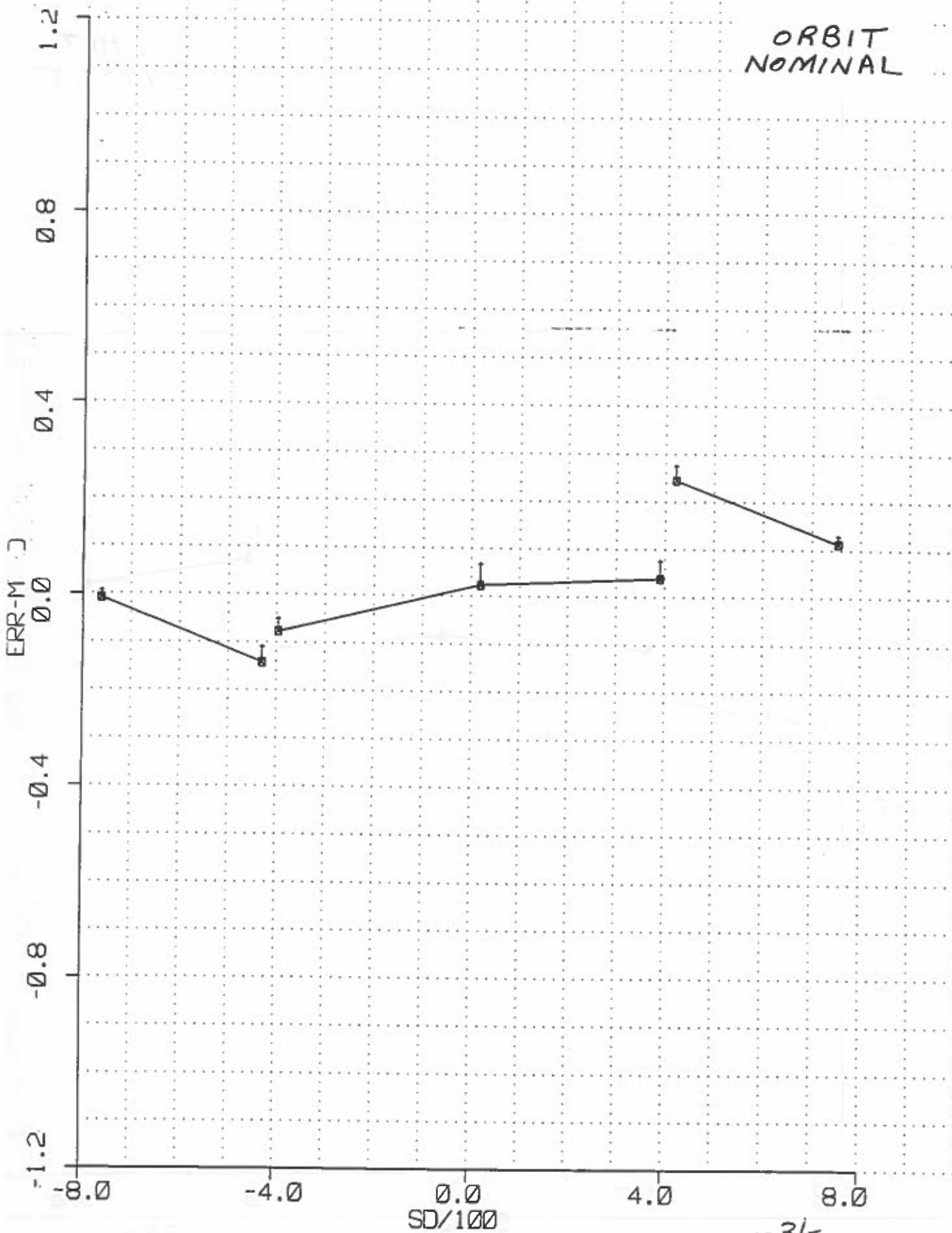
SSS= 23 ,M1= 23 ,DATE: 1123

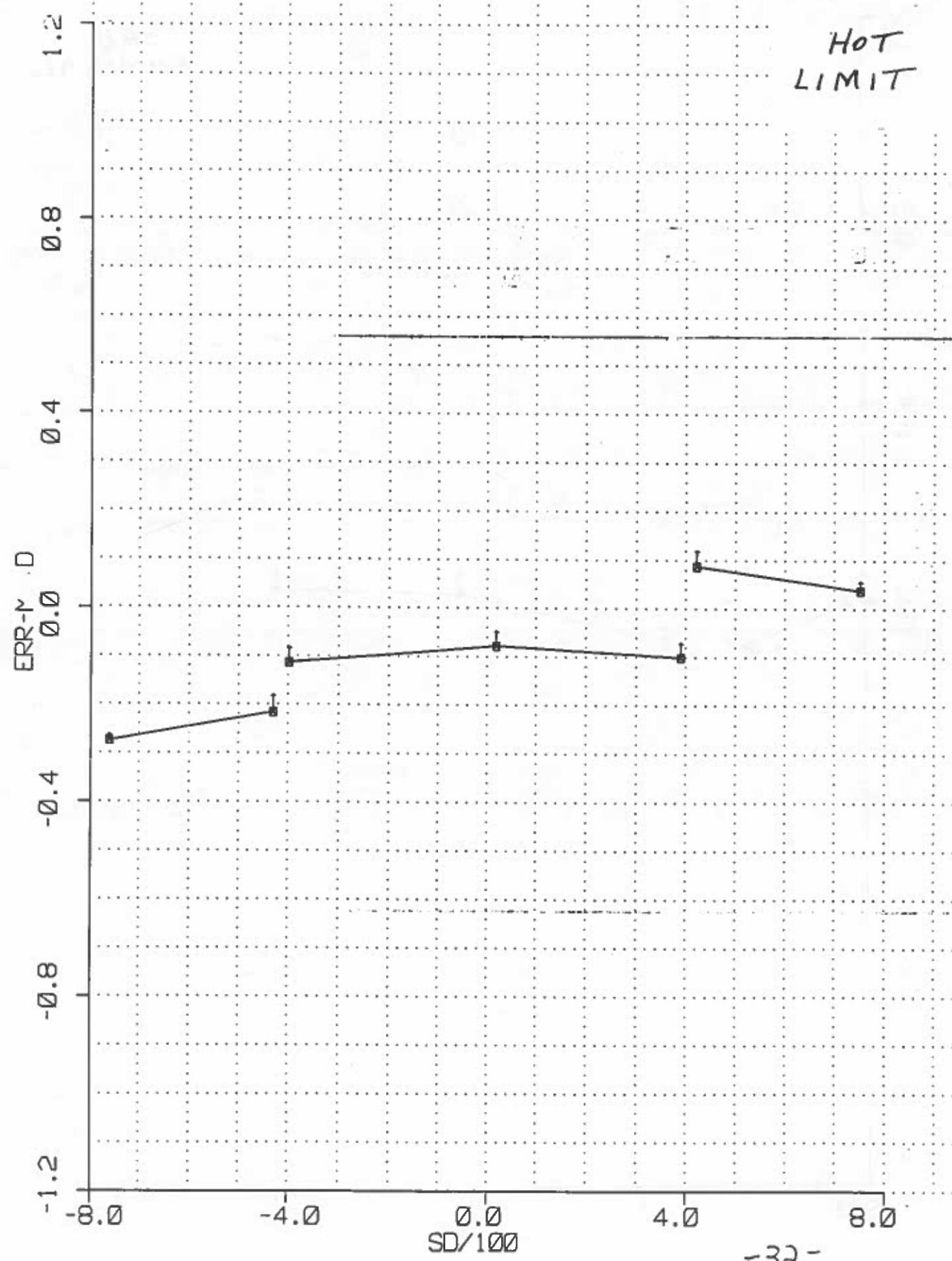
PRE
VIB

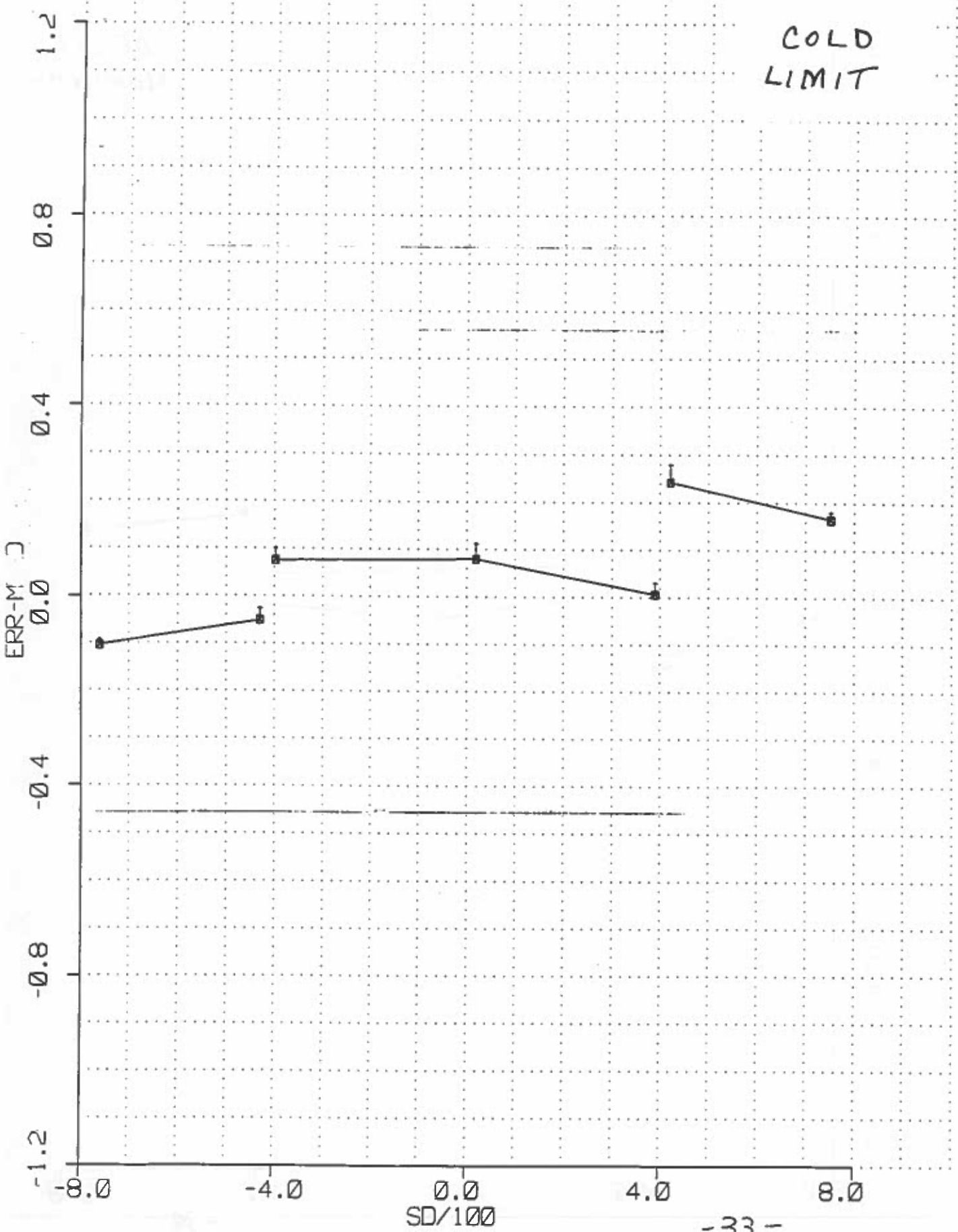


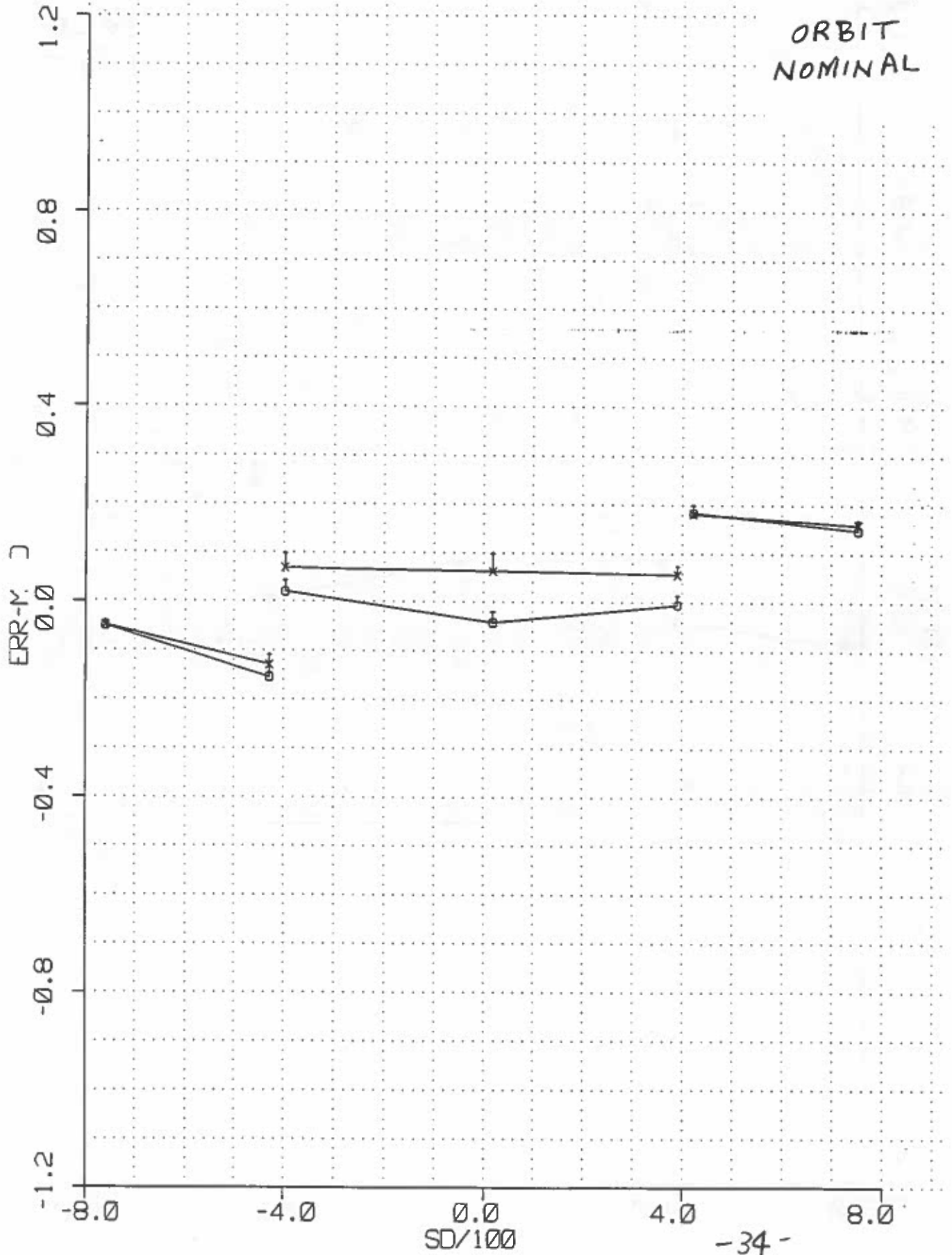


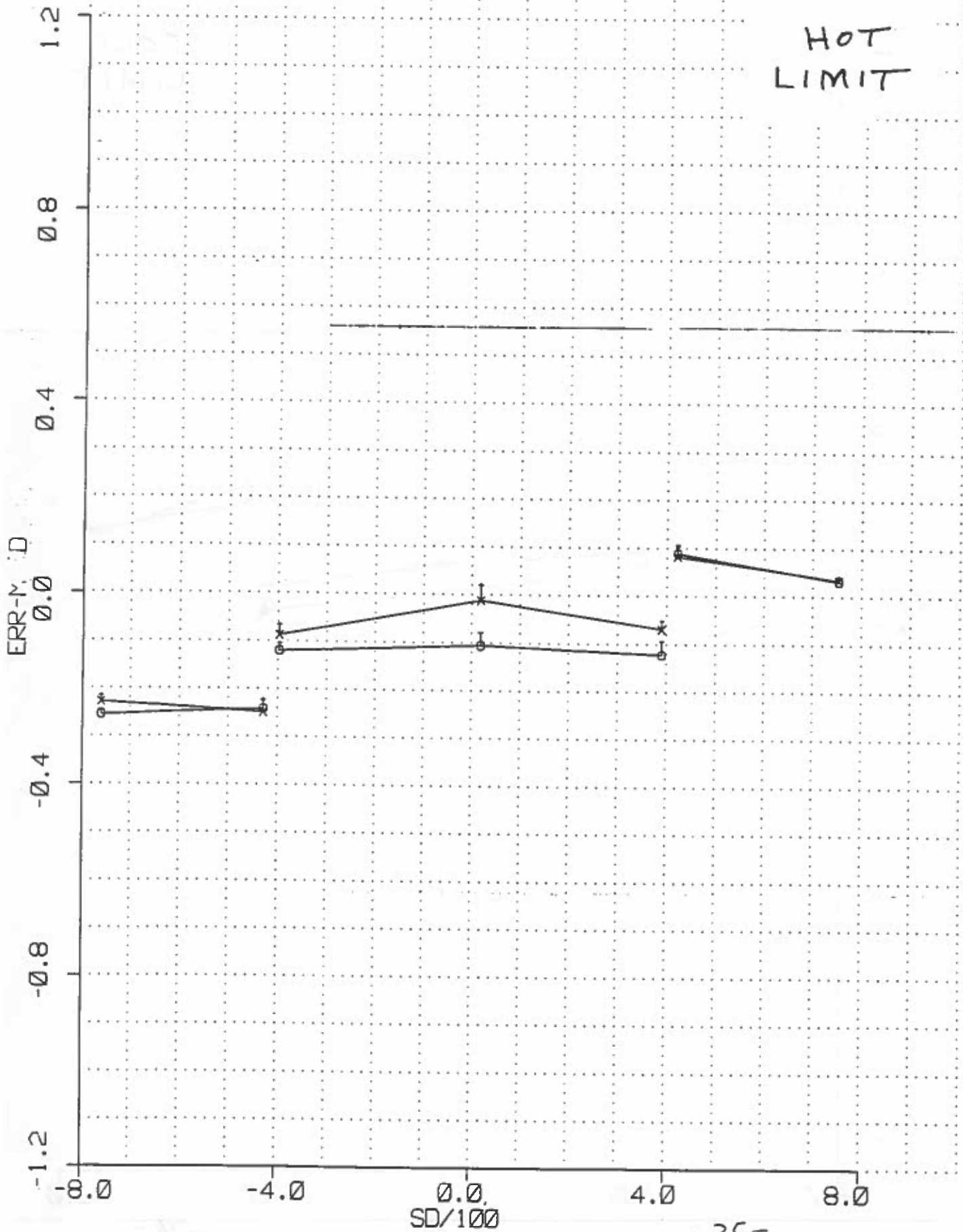
ORBIT
NOMINAL

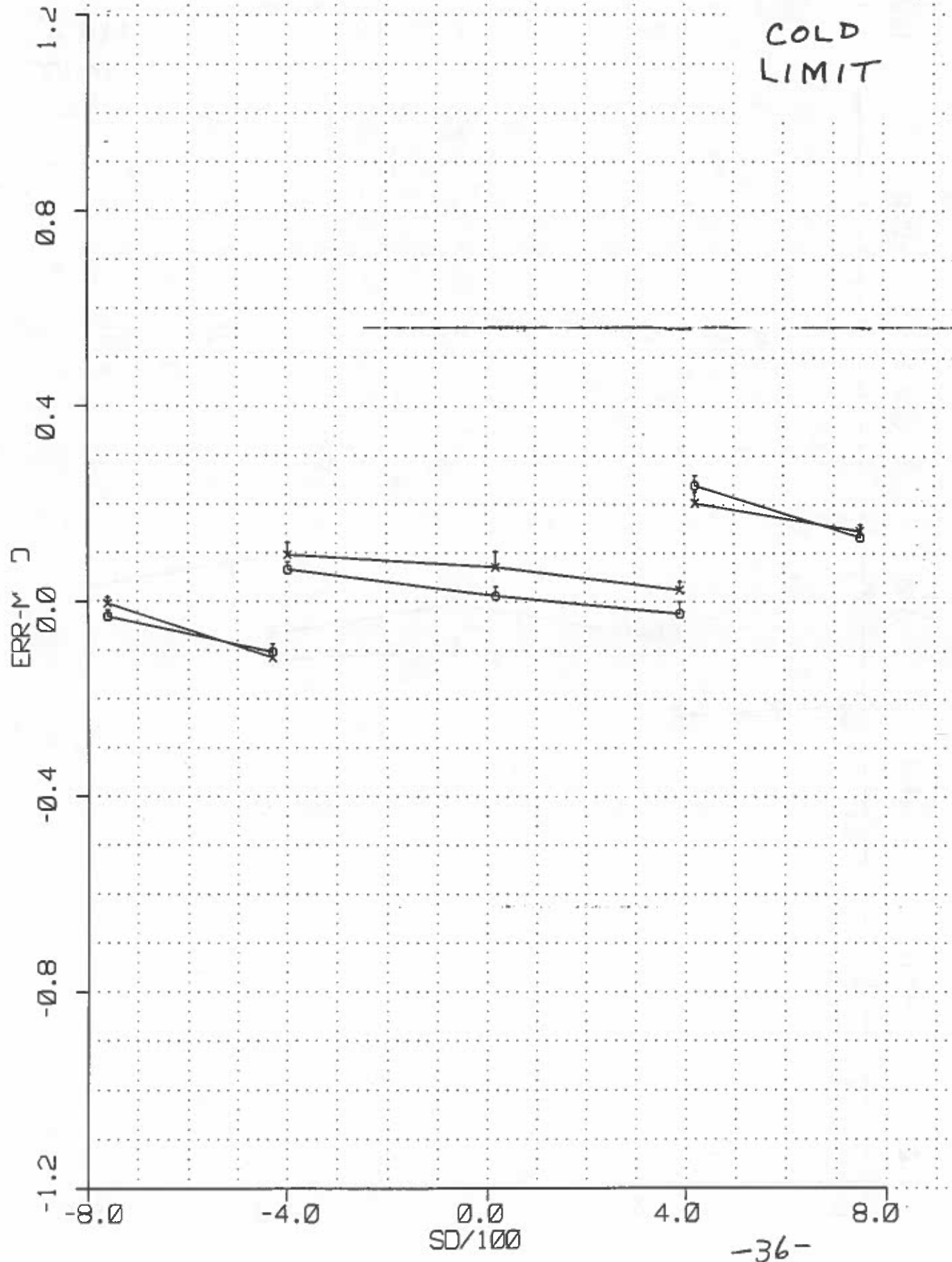








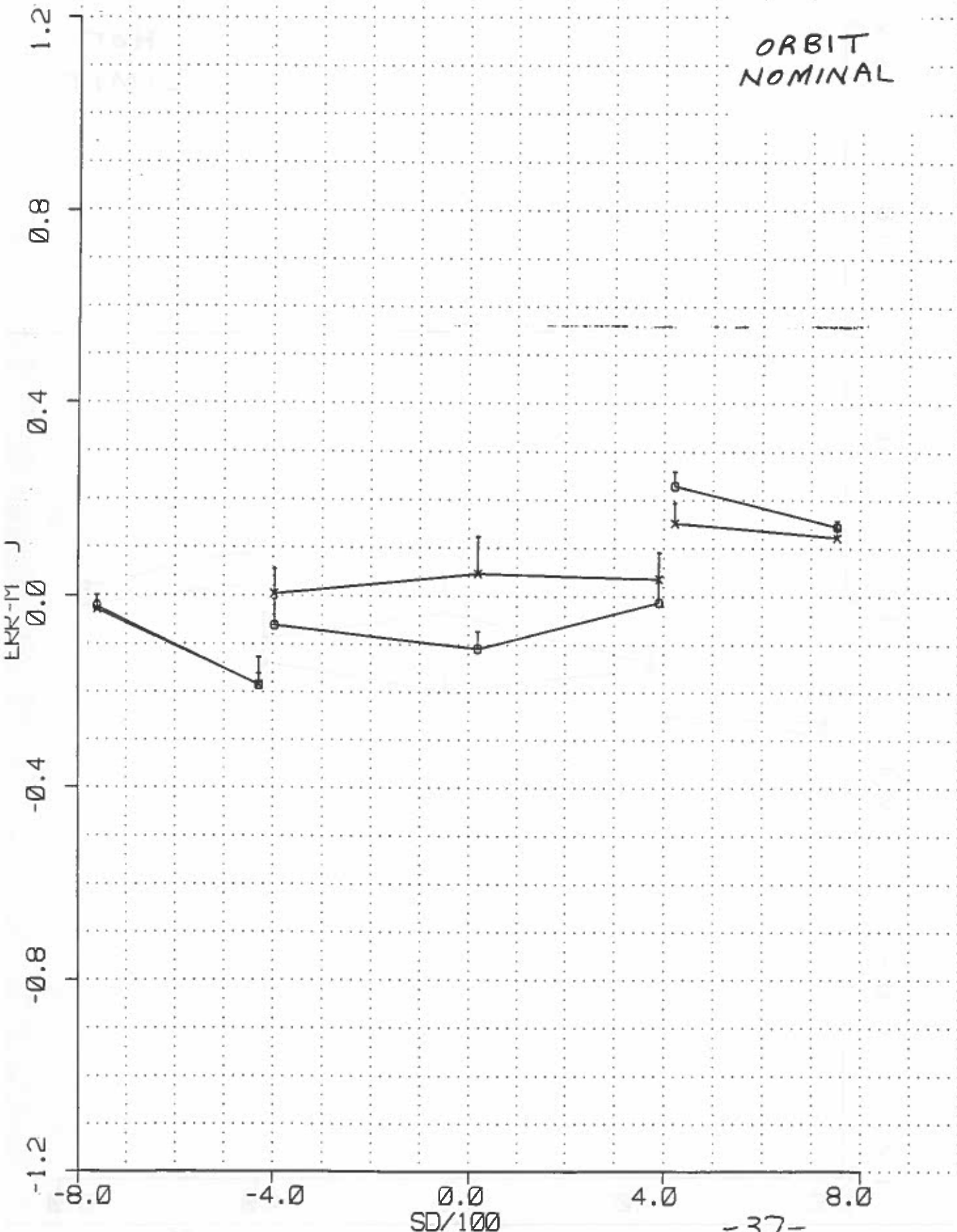




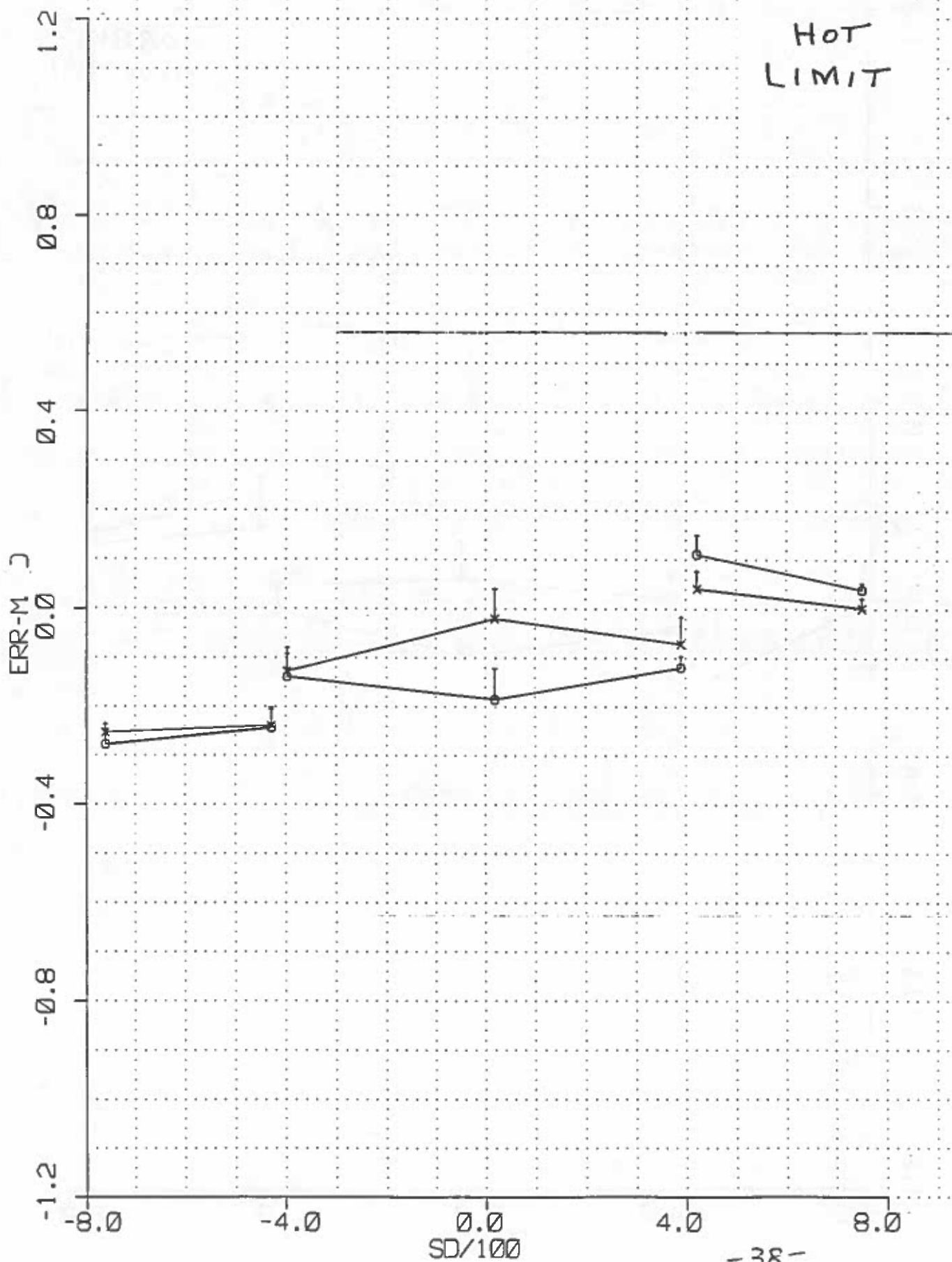
SYSTEM 14

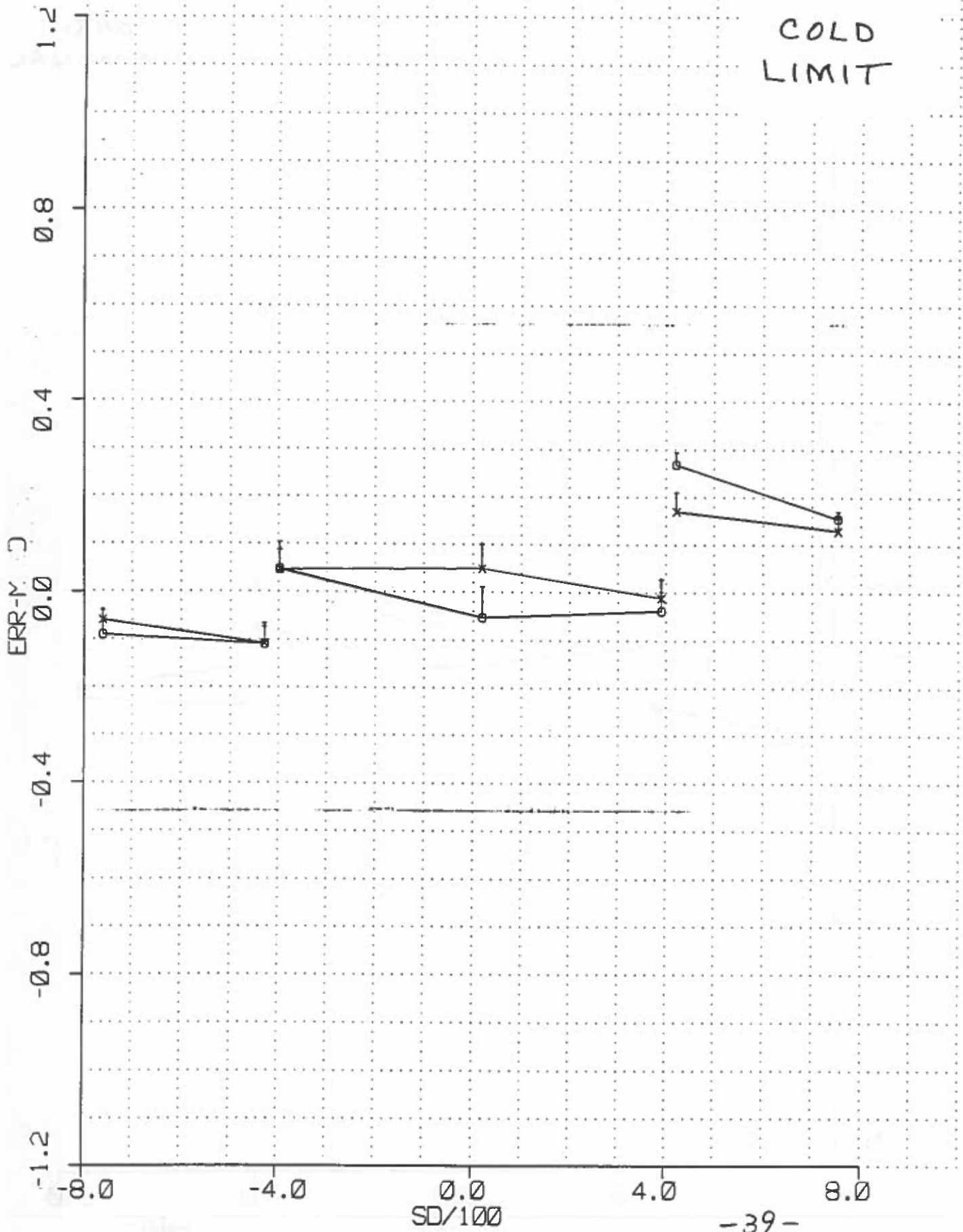
IMC-NORM HRD SYNC RTD-S SSS=5 ,M1=-8,DATE:329

ORBIT
NOMINAL



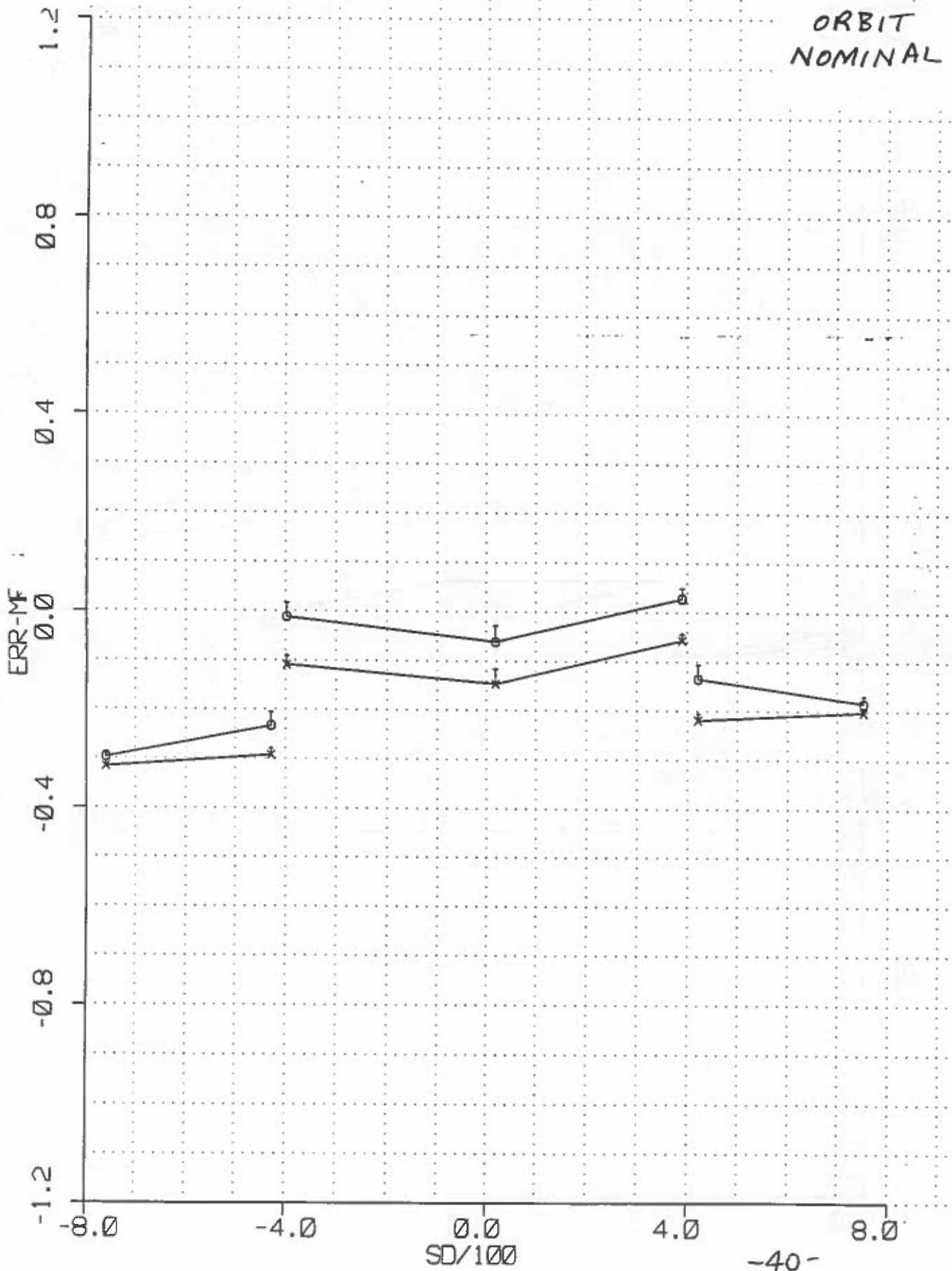
SYSTEM 14
IMC-NORM HRD SYNC RTD-S SSS=7 ,M1=12 ,DATE: 320

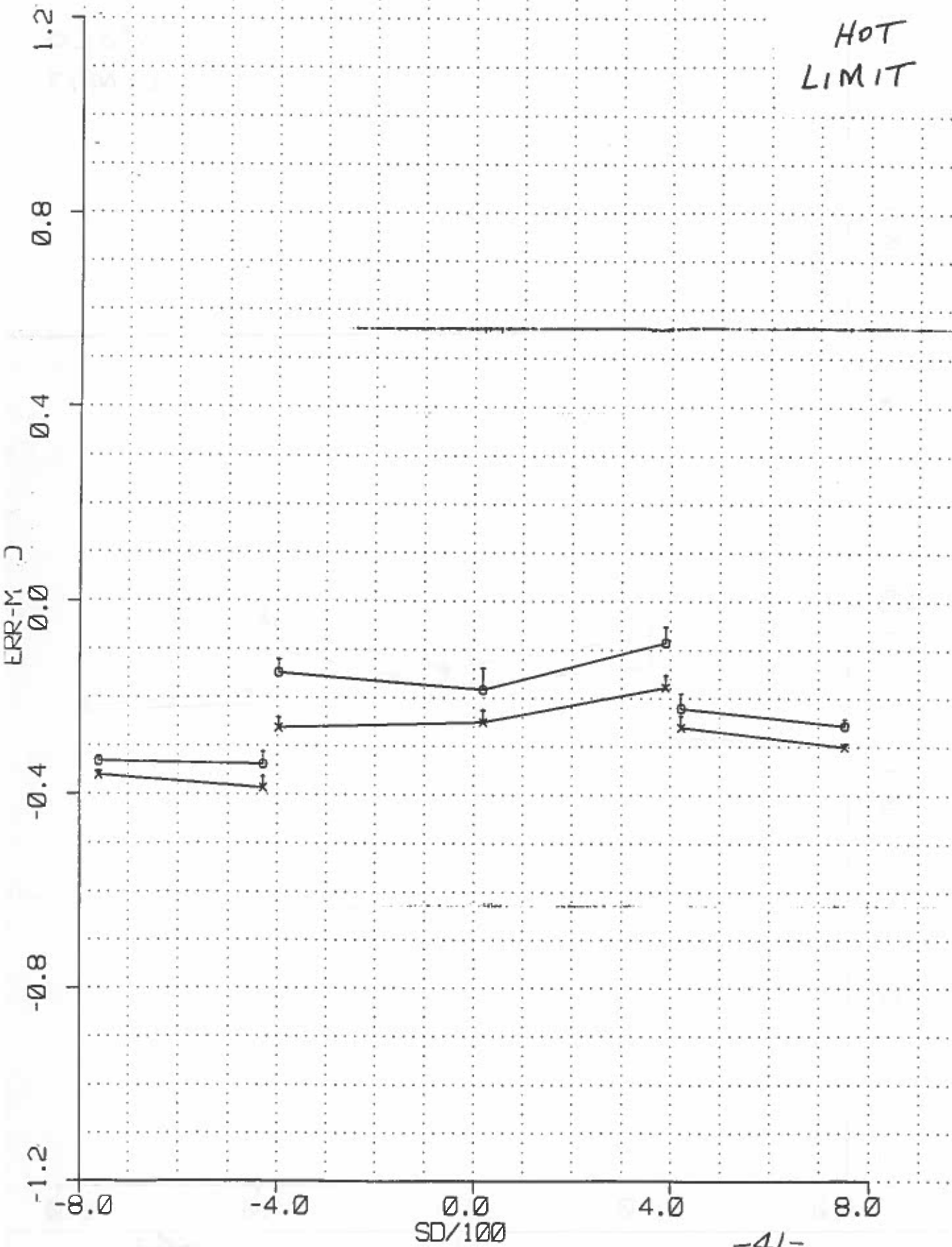




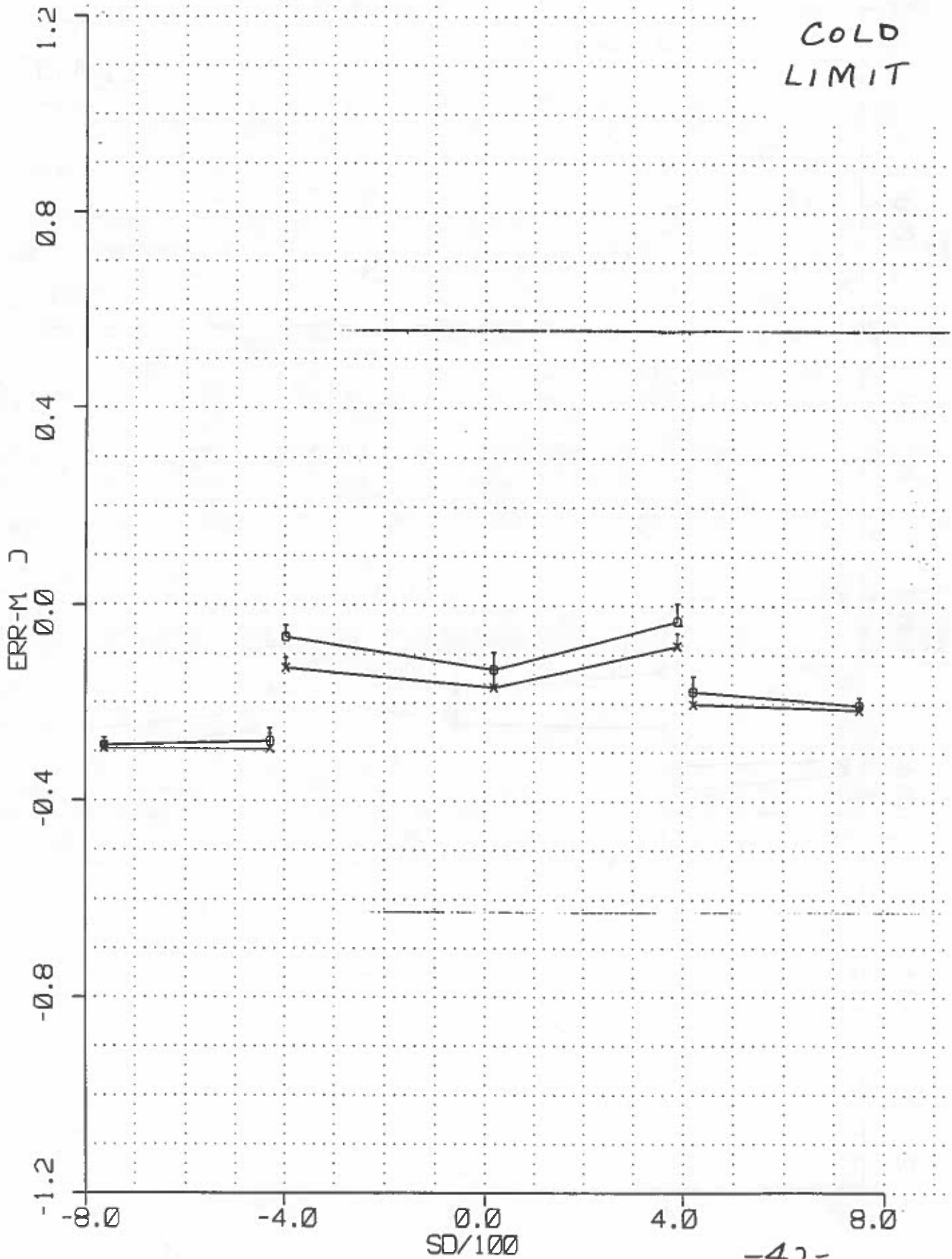
SYSTEM 14
IMC-NORM T SYNC SDF SSS=5, M1=-8, DATE: 330

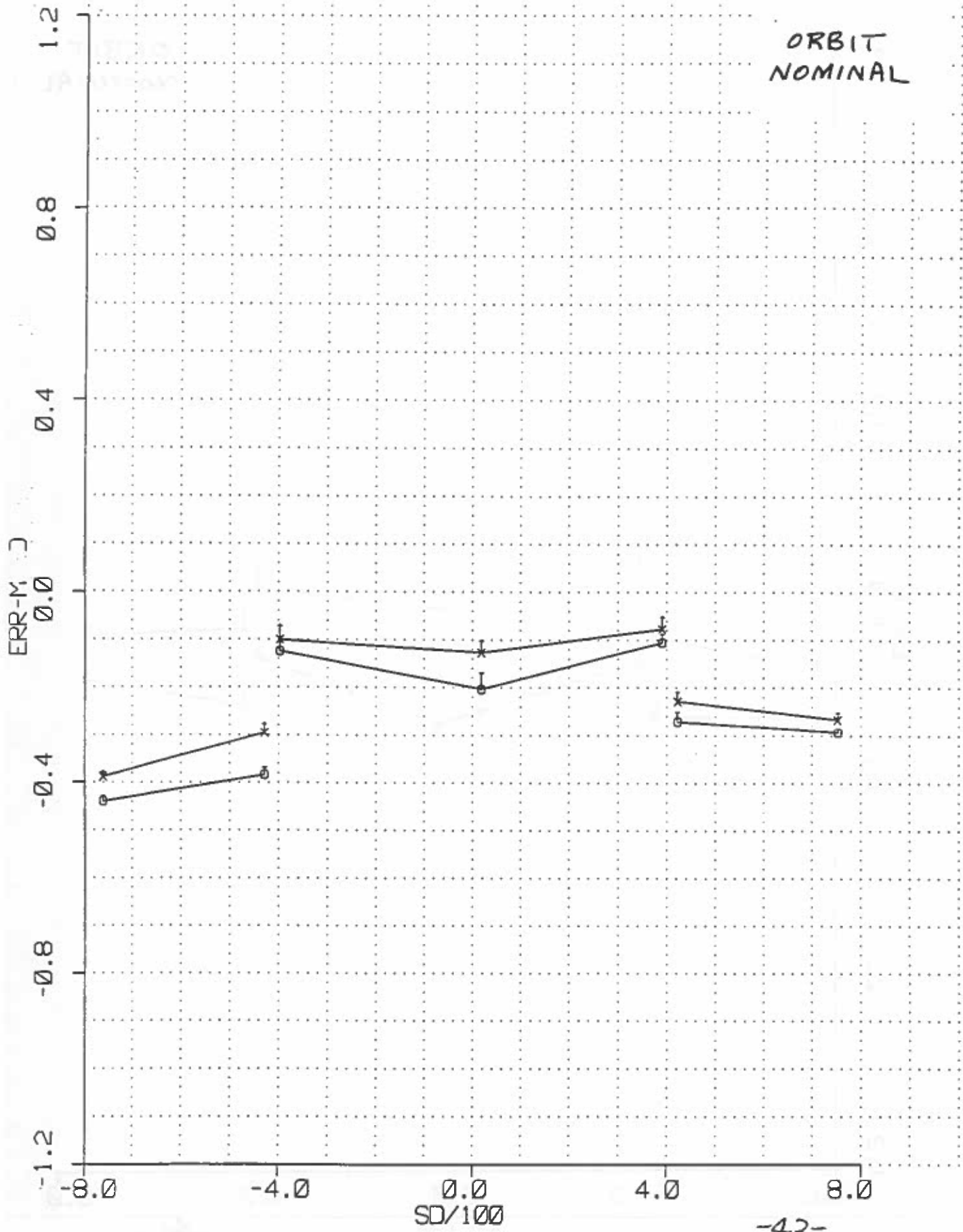
ORBIT
NOMINAL

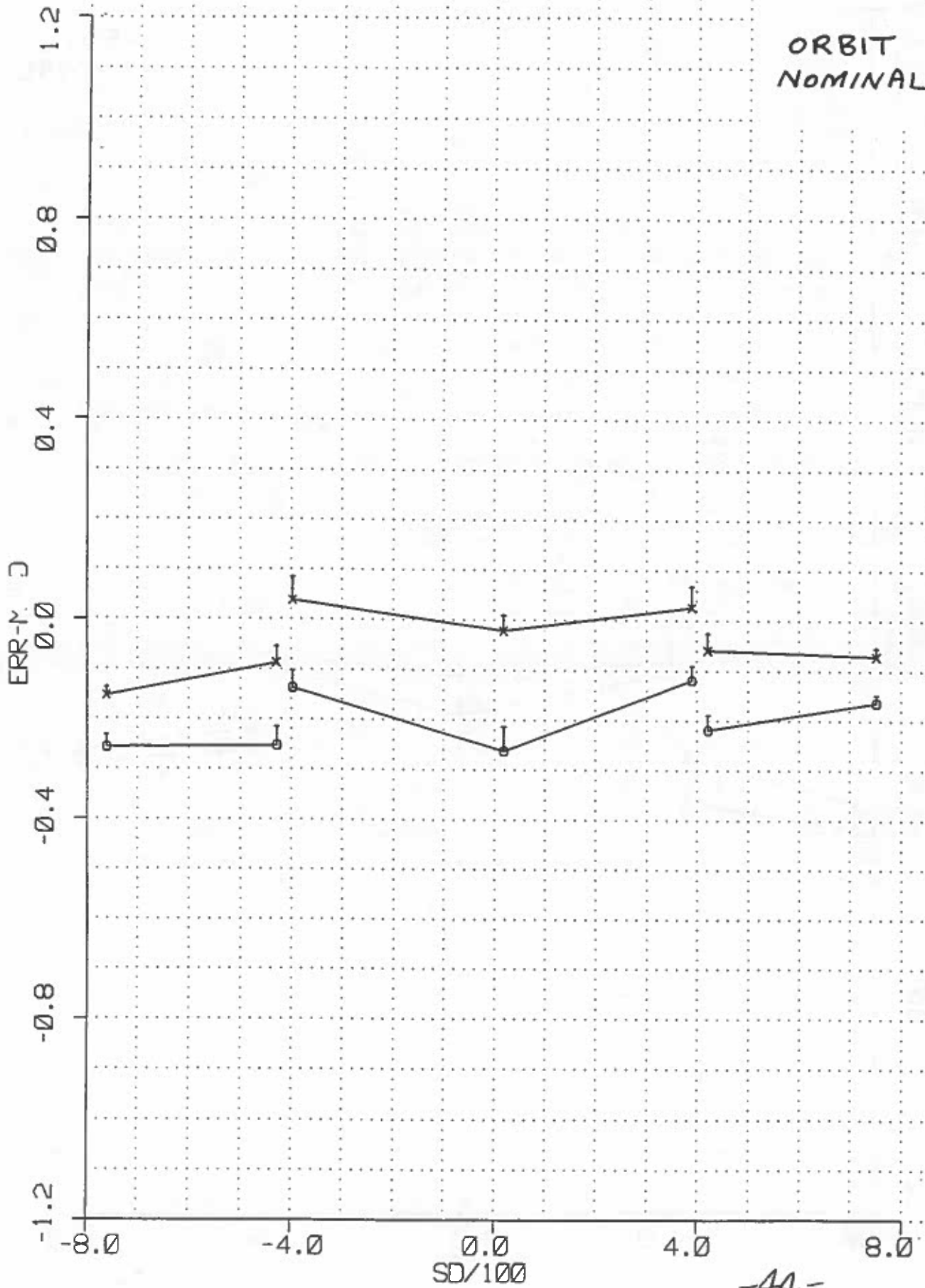




SYSTEM 14
IMC-NORM T SYNC SDF SSS=3 ,M1=-8 ,DATE: 324

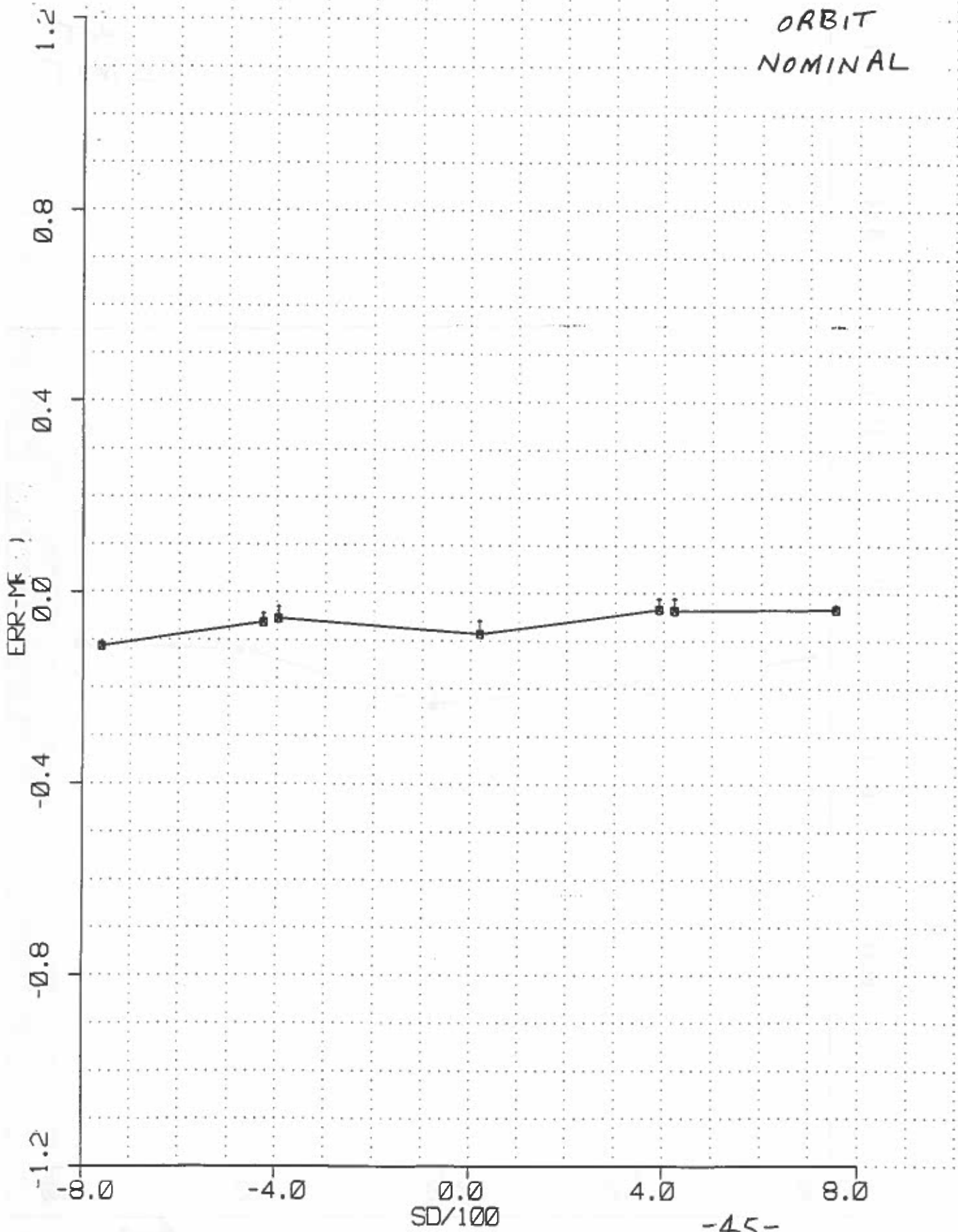


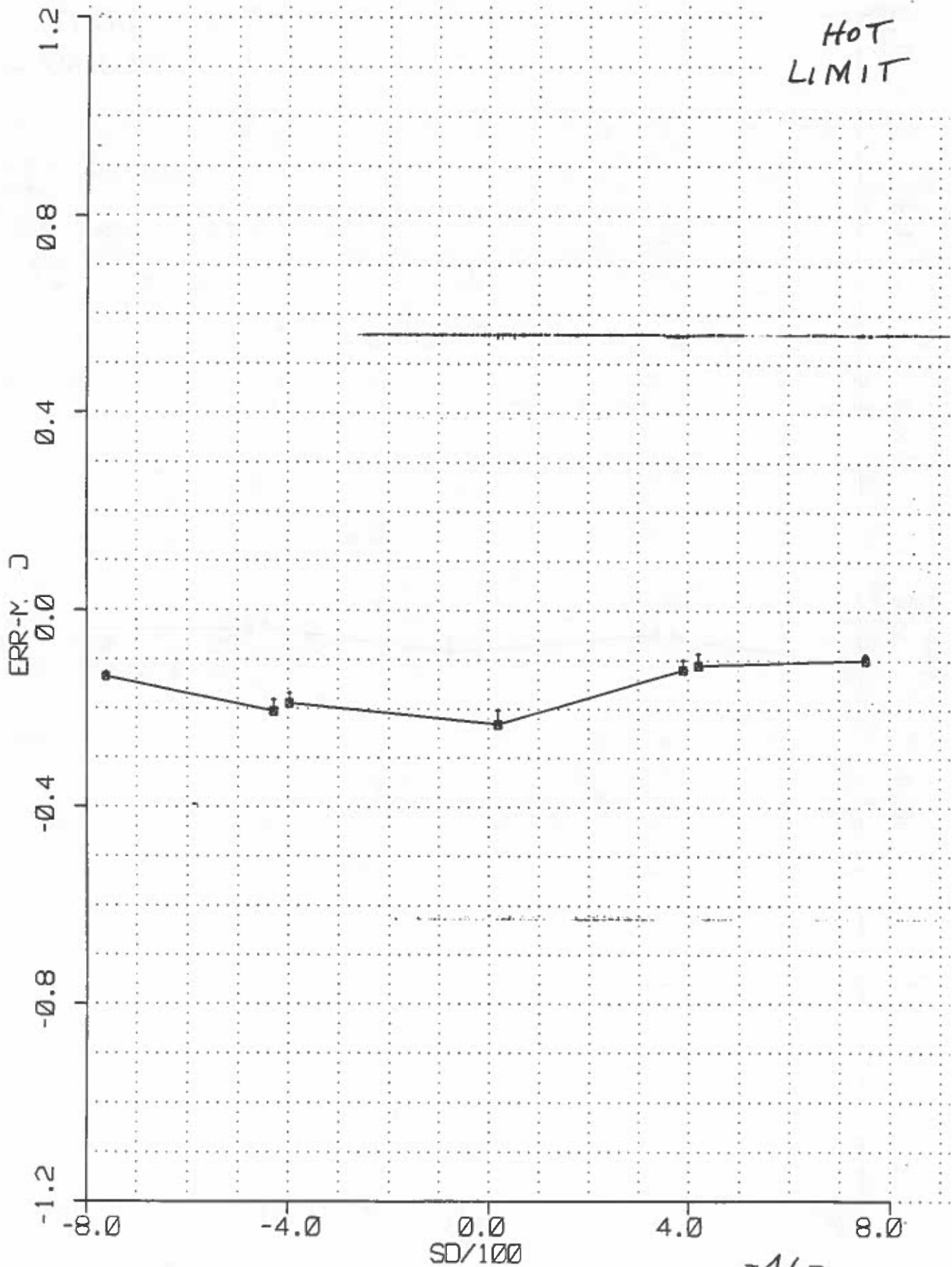


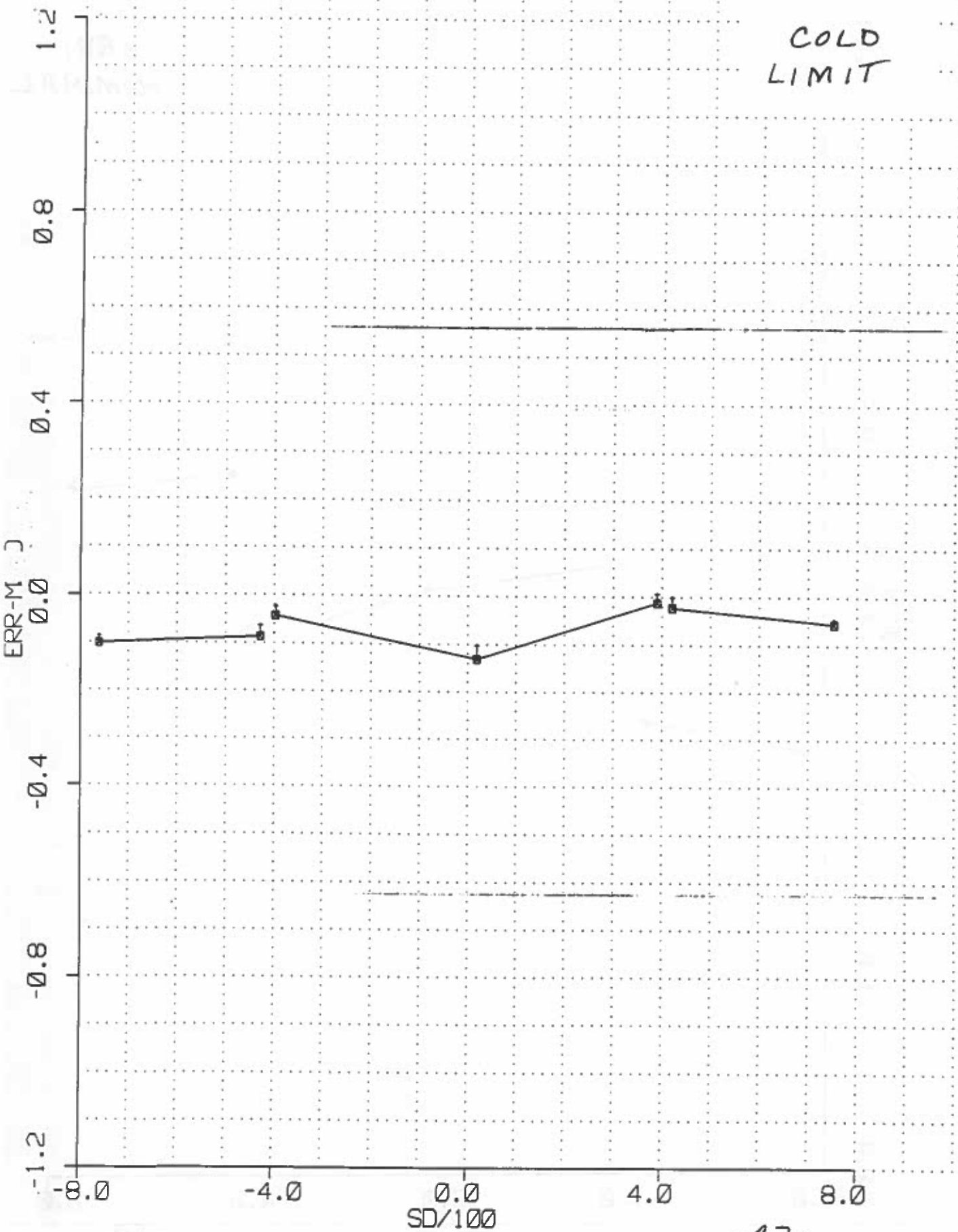


SYSTEM 14
IMC-NORM T SYNC SDS SSS=5 ,M1=-8 ,DATE: 330

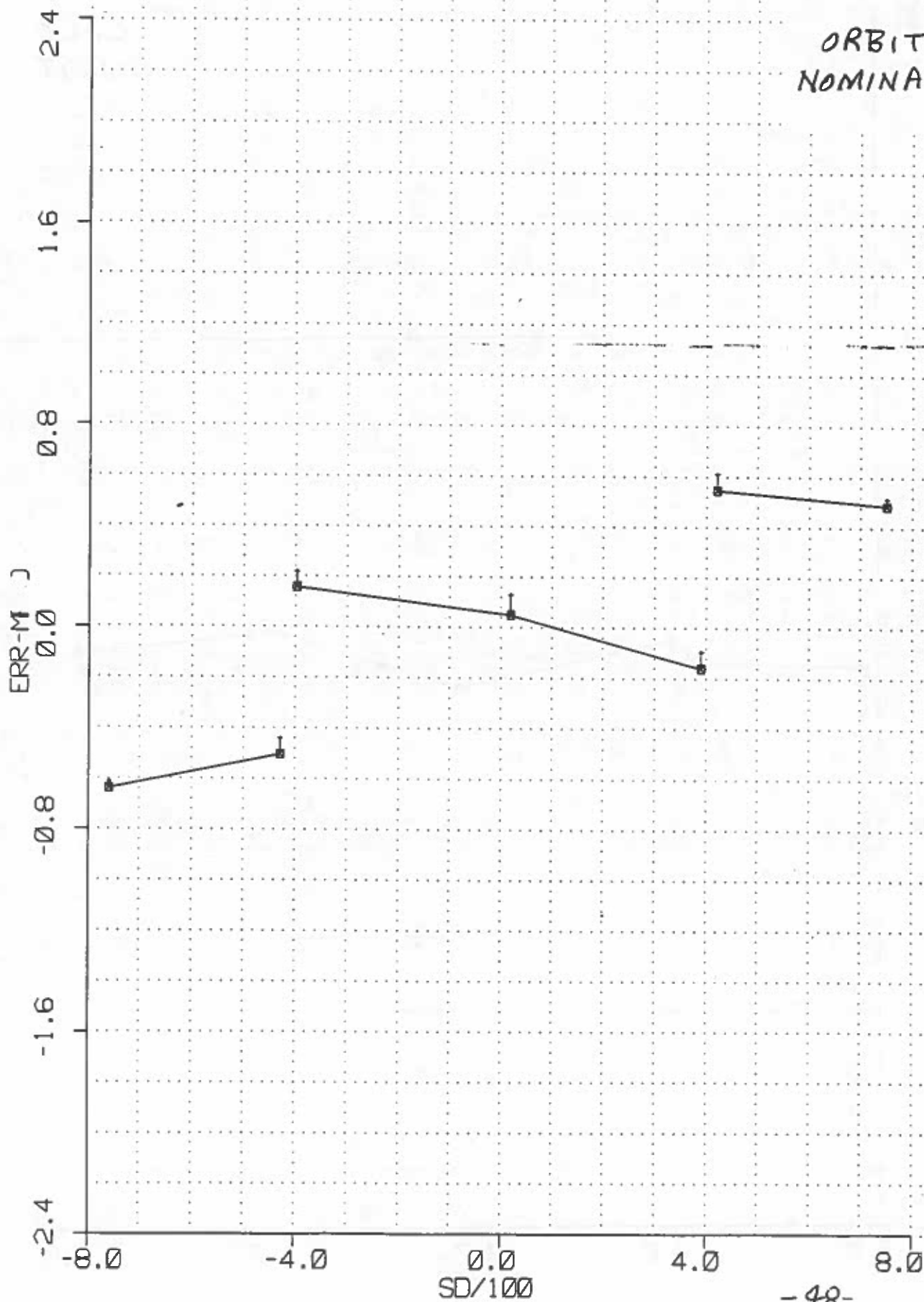
ORBIT
NOMINAL

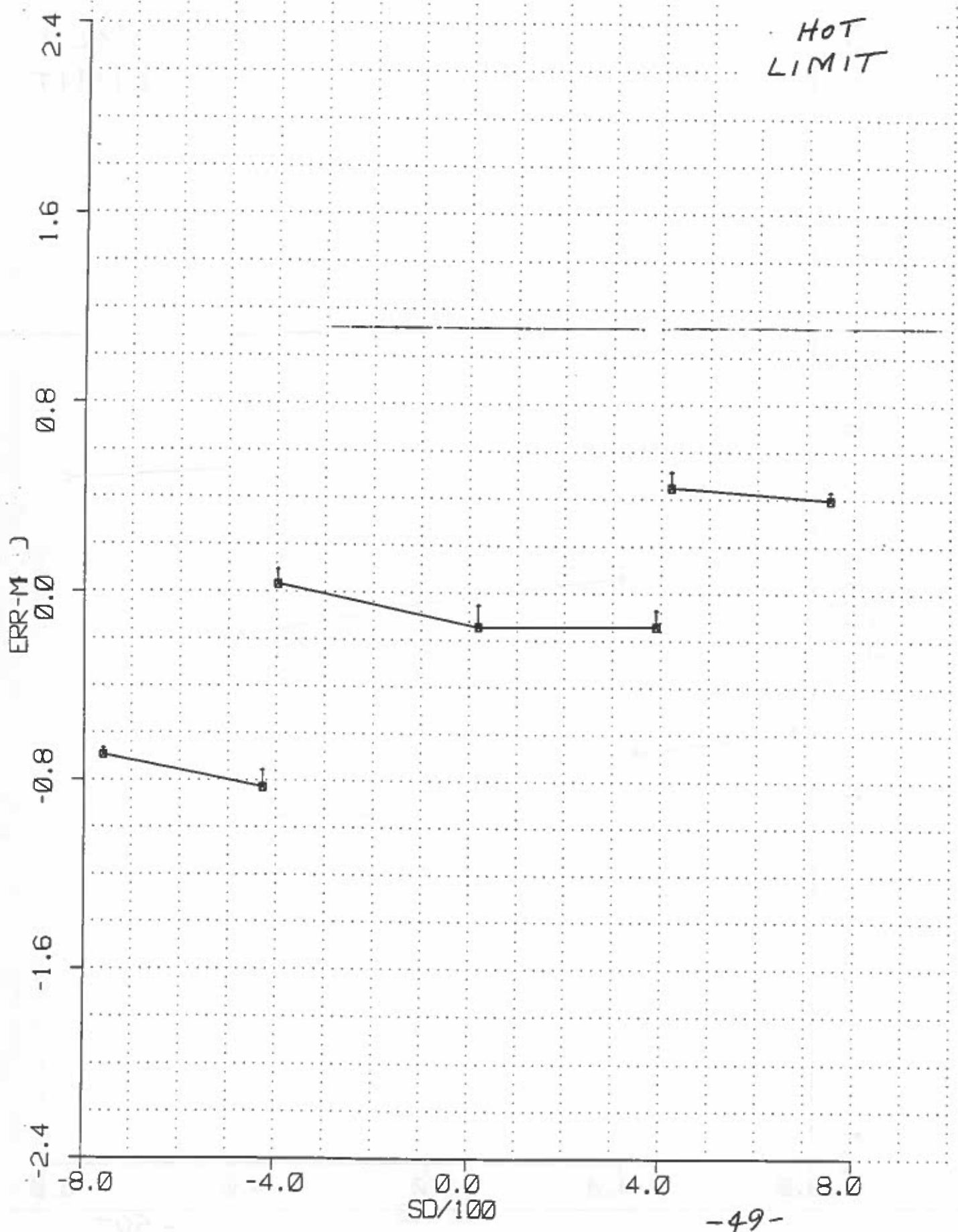


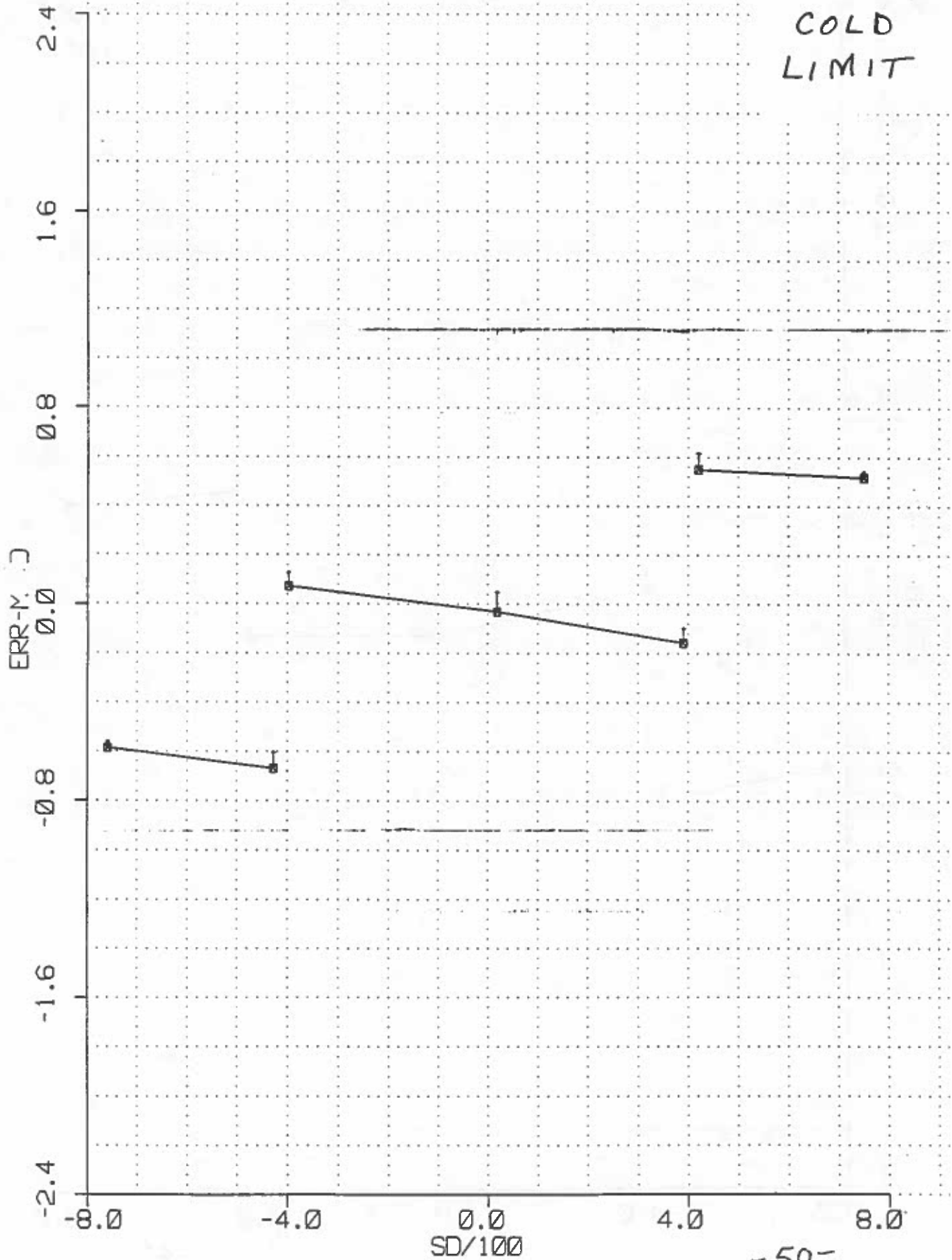




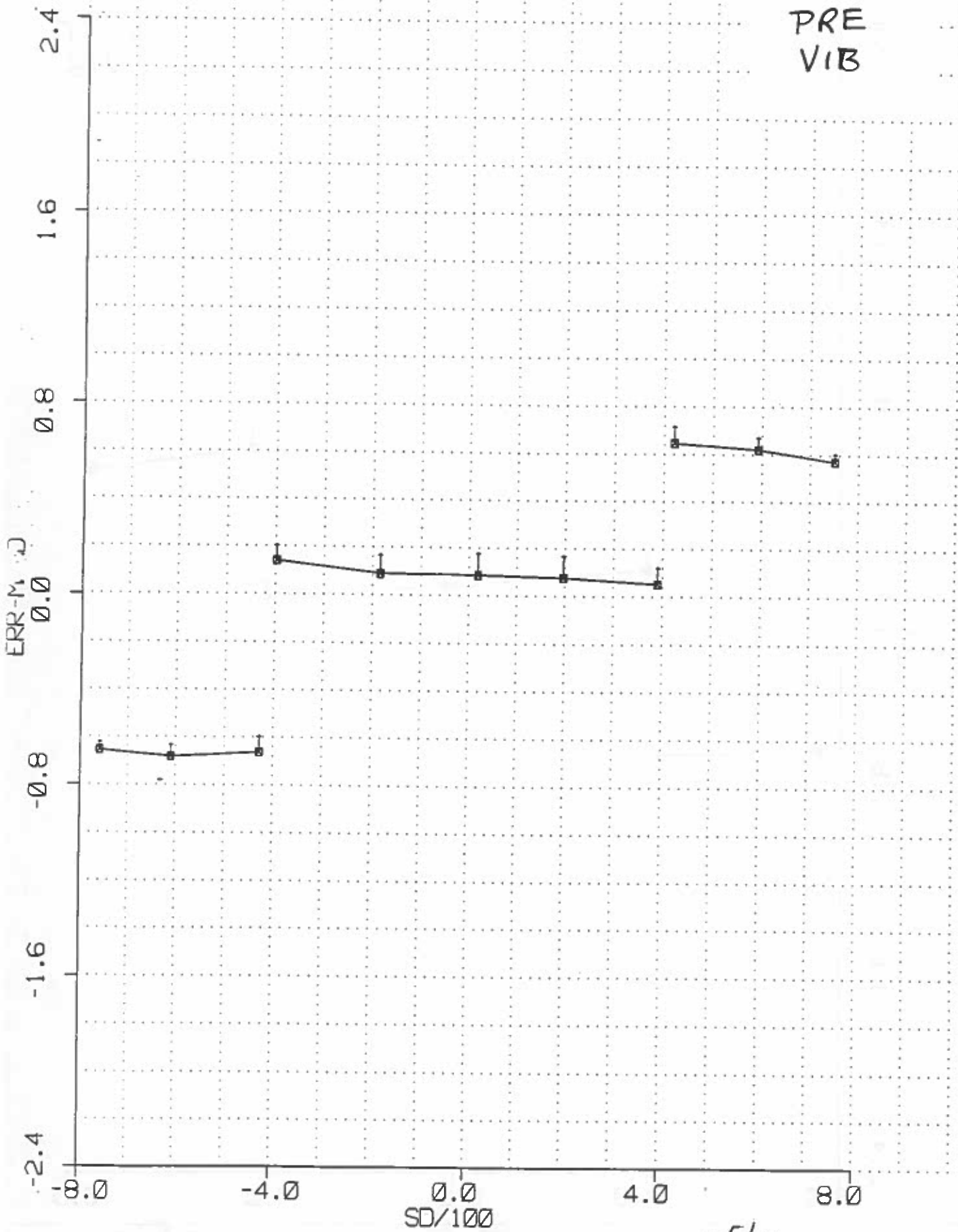
ORBIT
NOMINAL



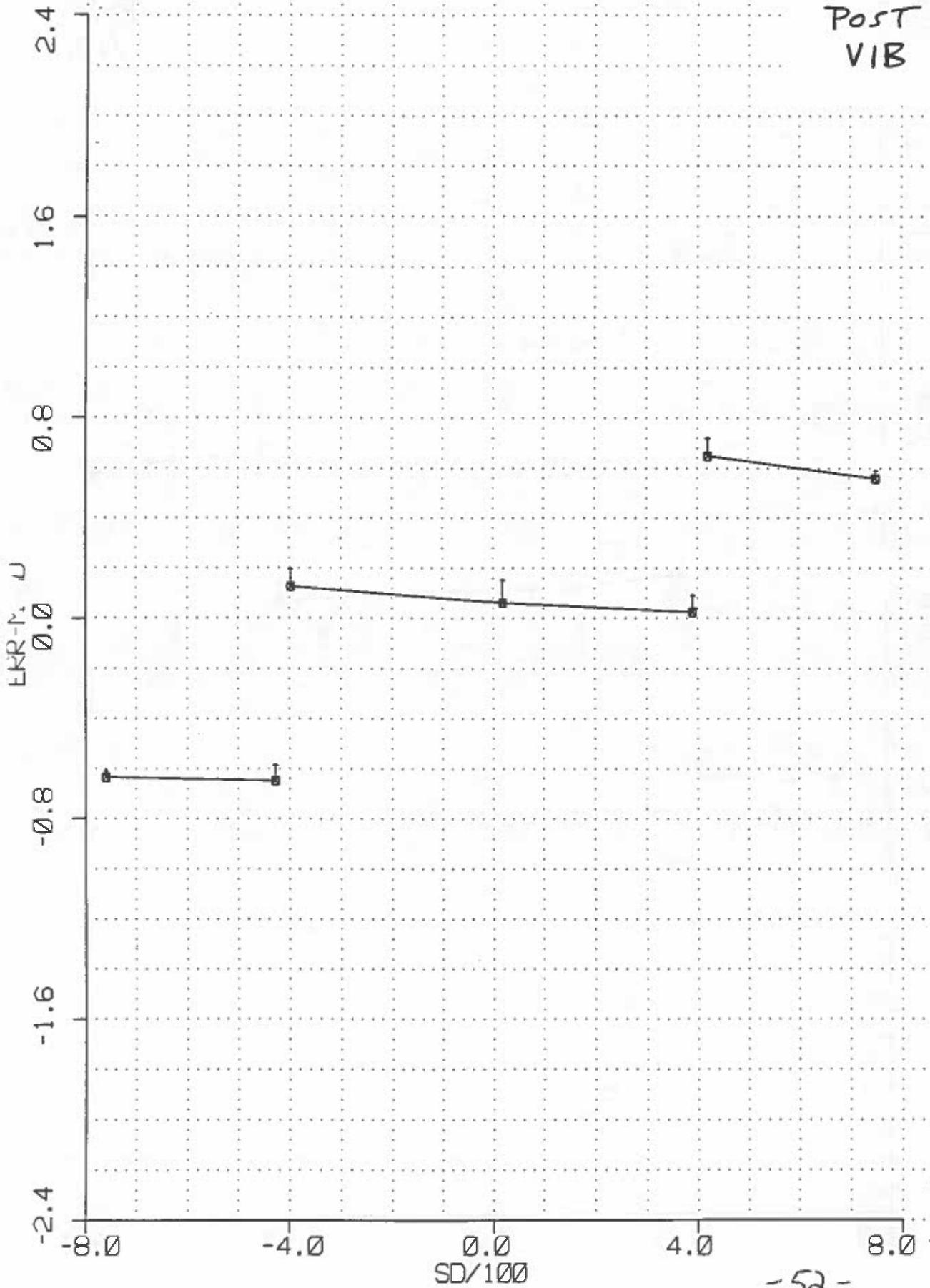




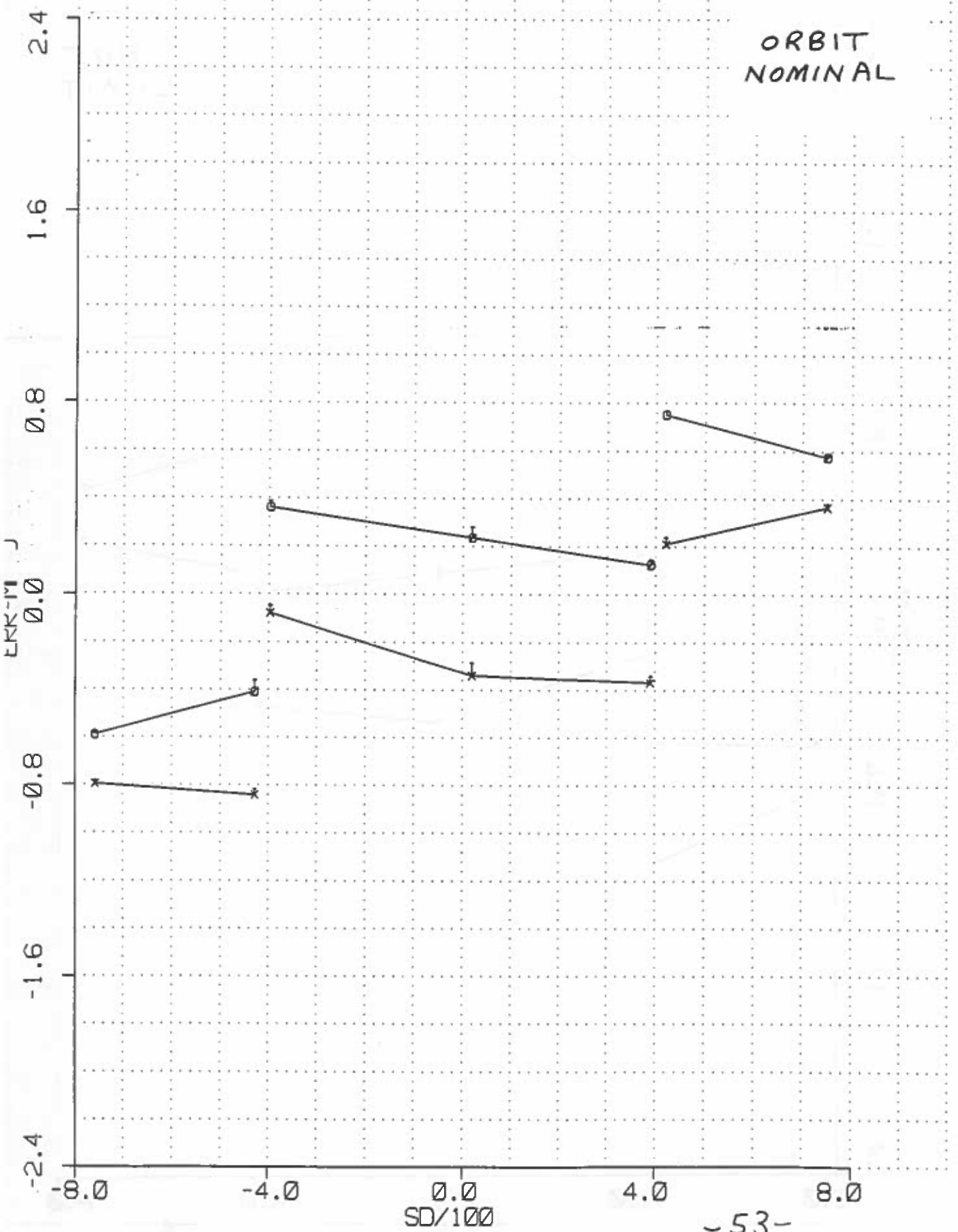
PRE
VIB

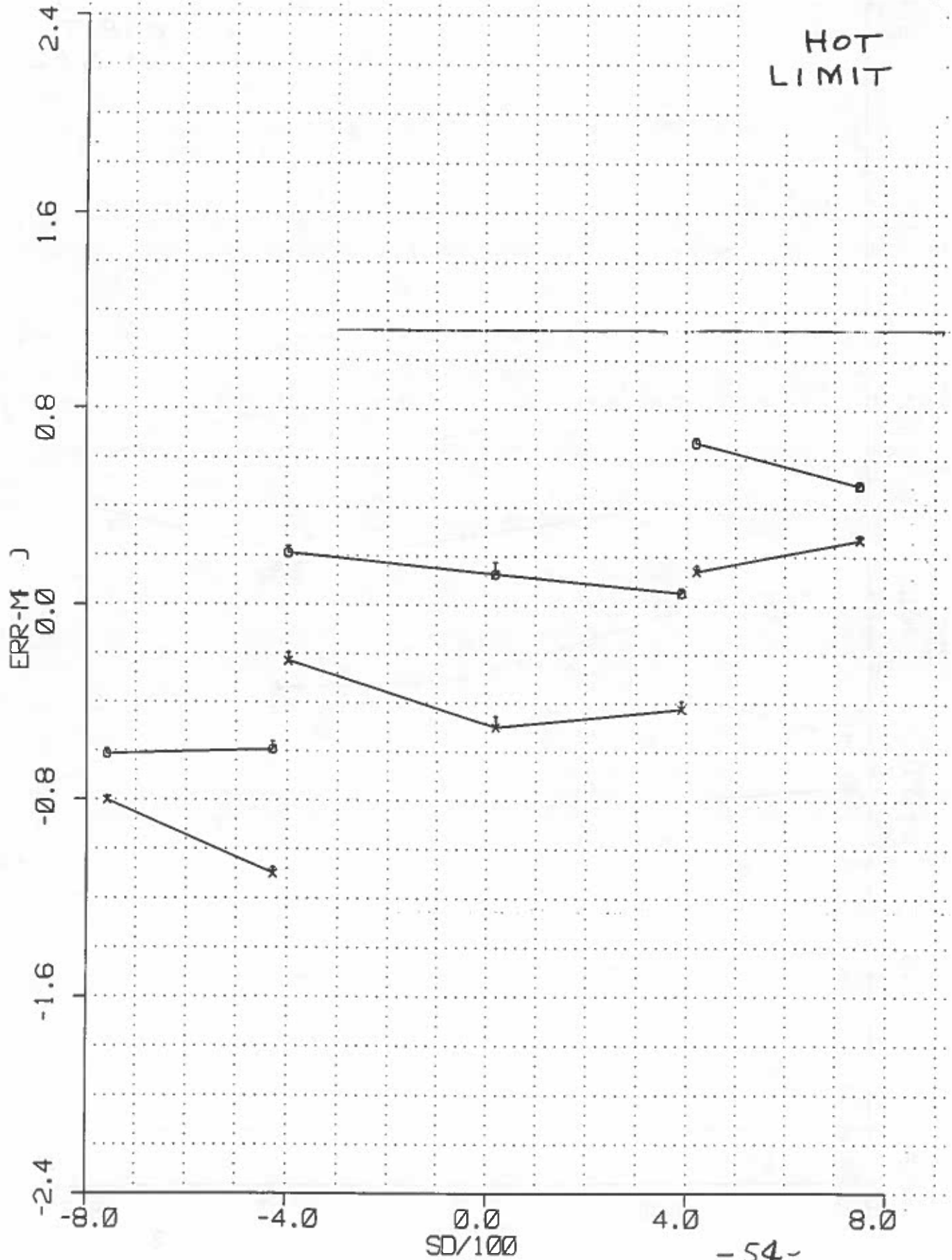


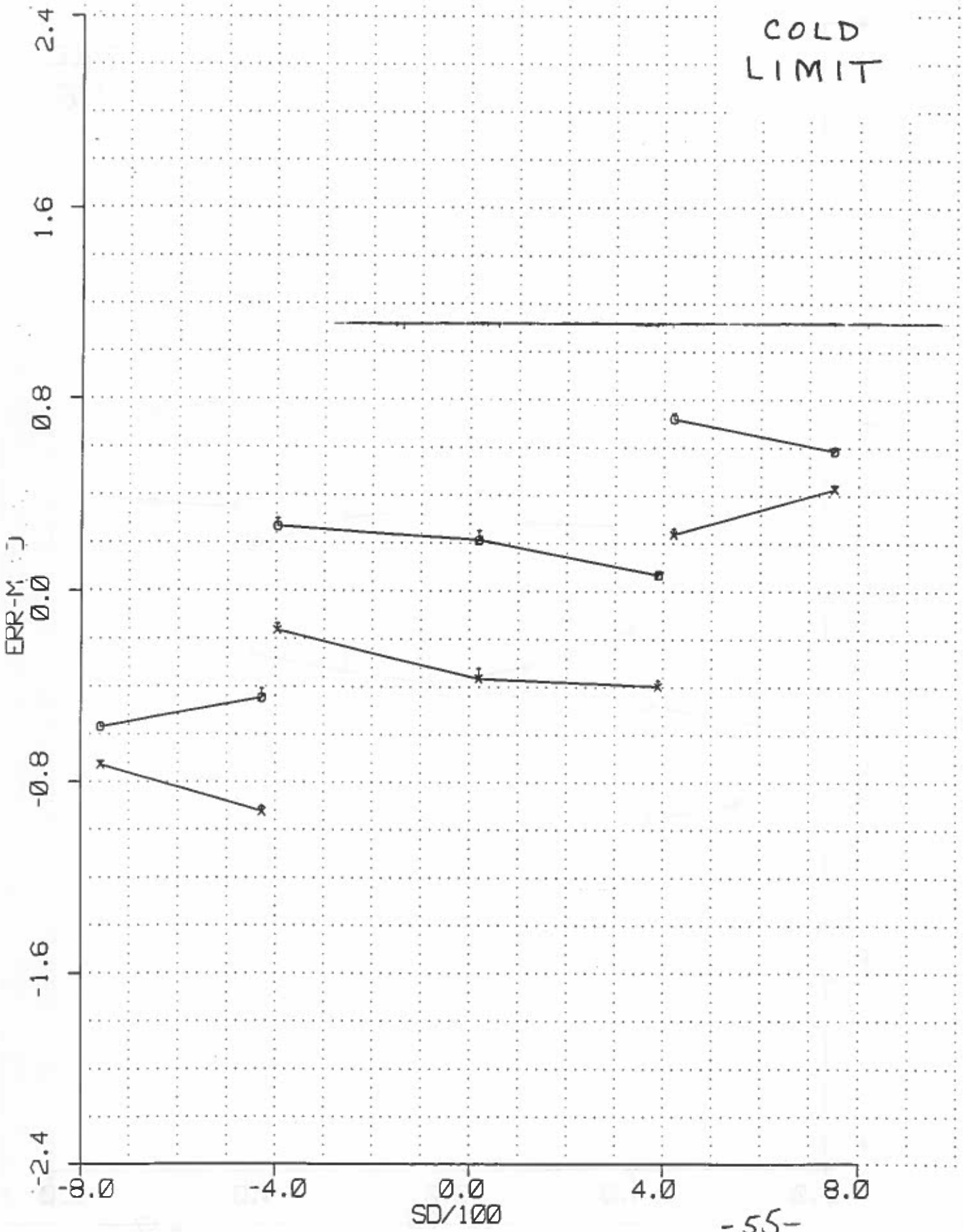
Post
VIB

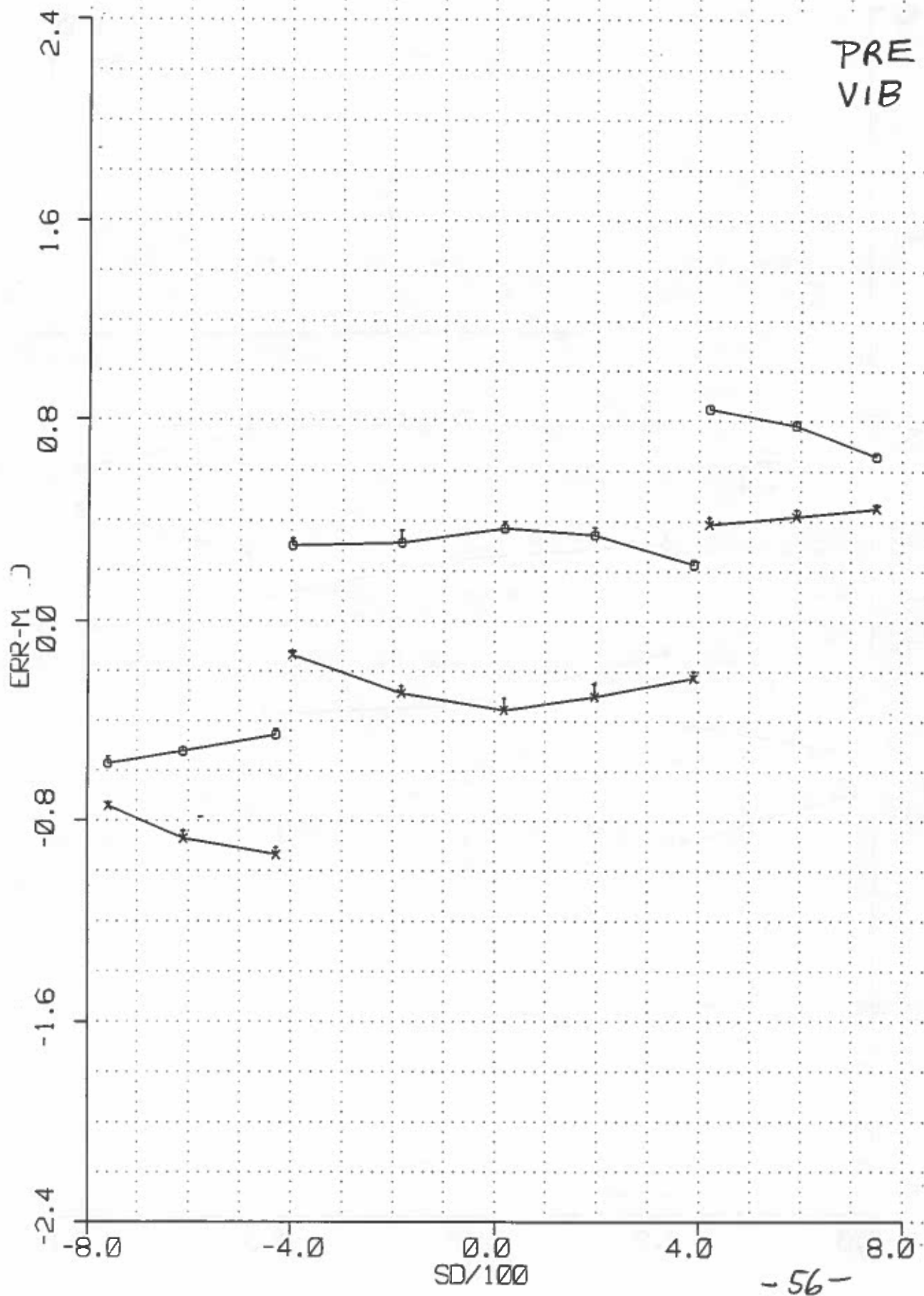


ORBIT
NOMINAL

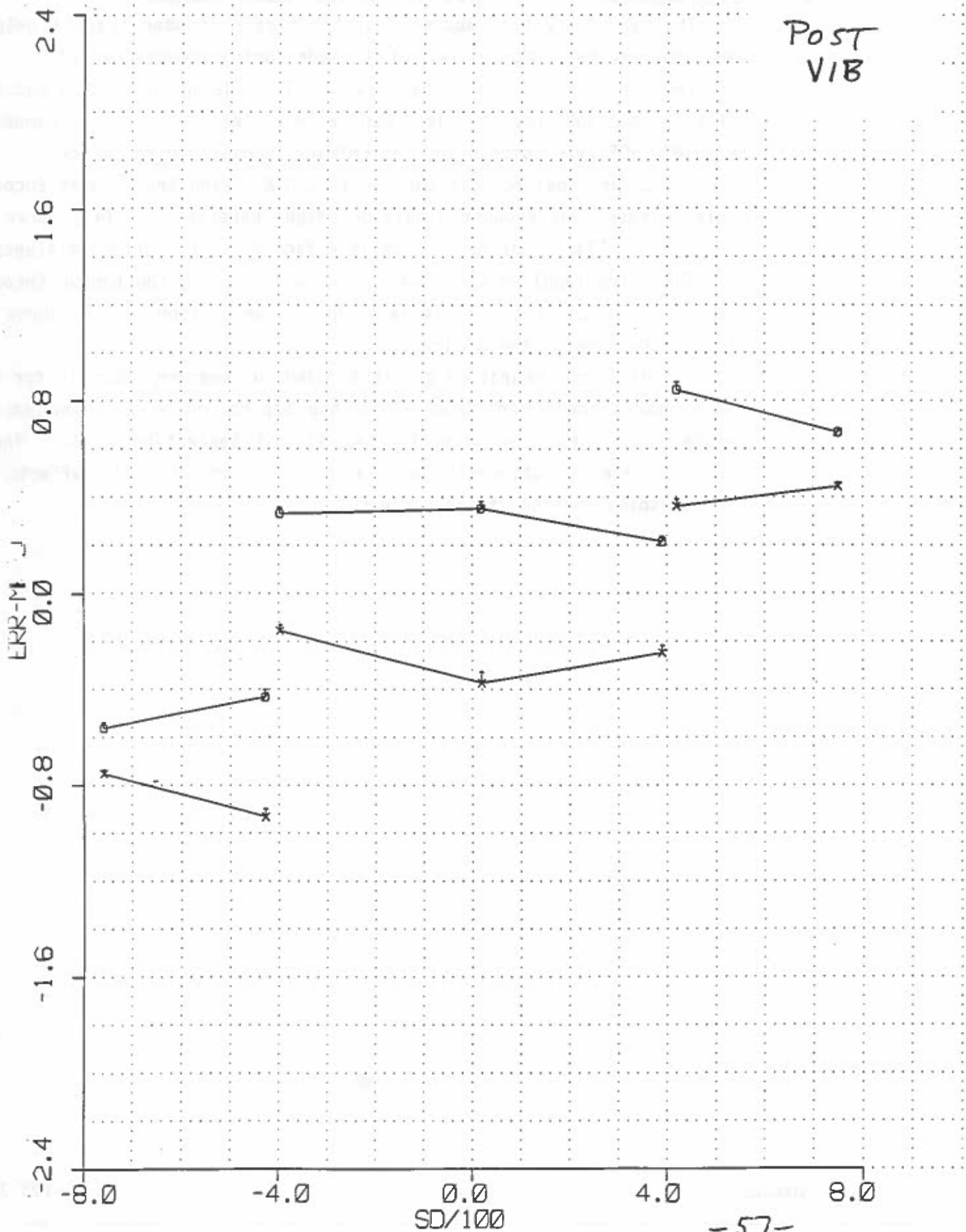








Post
VIB



4. SYNCHRONIZATION USING BACKUP ENCODER & ENCODER SIMULATOR

The synchronization accuracy of the backup encoder track & delp generator are measured in the HRD SDF mode during acceptance test.

The curve labelled A is taken with I/O X, using the Backup Encod Control Track and Encoder Delphi Generation. This curve can be compar to an HRD SDF sync curve using the Primary Encoder Control Track.

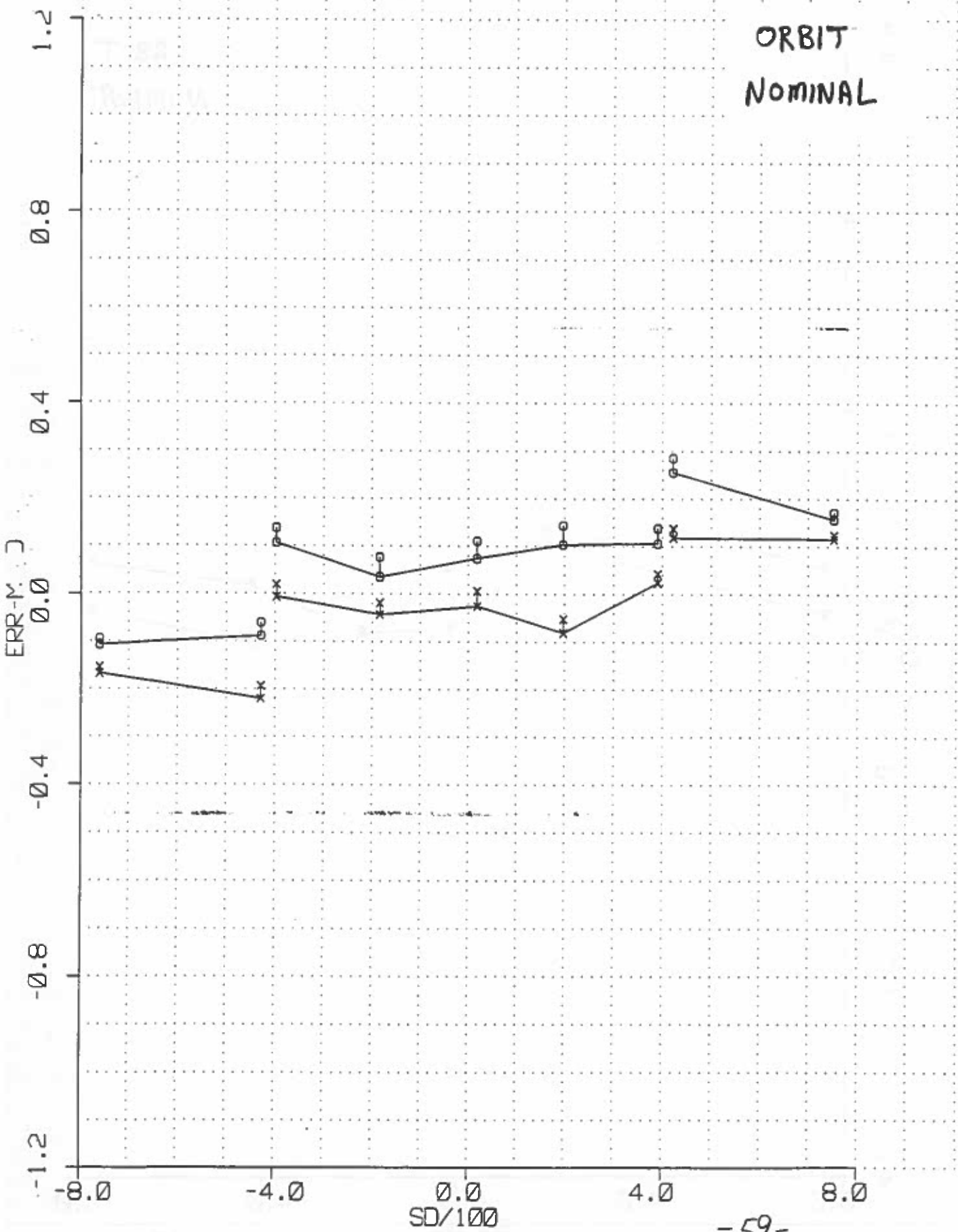
The curve labelled B is taken with I/O X, using the Primary Encod Control Track, and encoder Simulator Delphi Generation. This curve plotted as milliradians error from Interface Axis vs. Surface distance

The curve labelled C is taken with I/O Y, using the Backup Encod Control Track and Encoder Simulator Delphi Generation. This curve plotted the same as the B curve.

The Bias and Separation constants used for bearing retrofit for 0 #14 Primary Encoder are Bias = -18 and Separation = +8. The Back Encoder constants used were Bias = -17 and Separation = +8. The constants are operationally adjustable to account for the effects non-sinusoidal motion of the scanner.

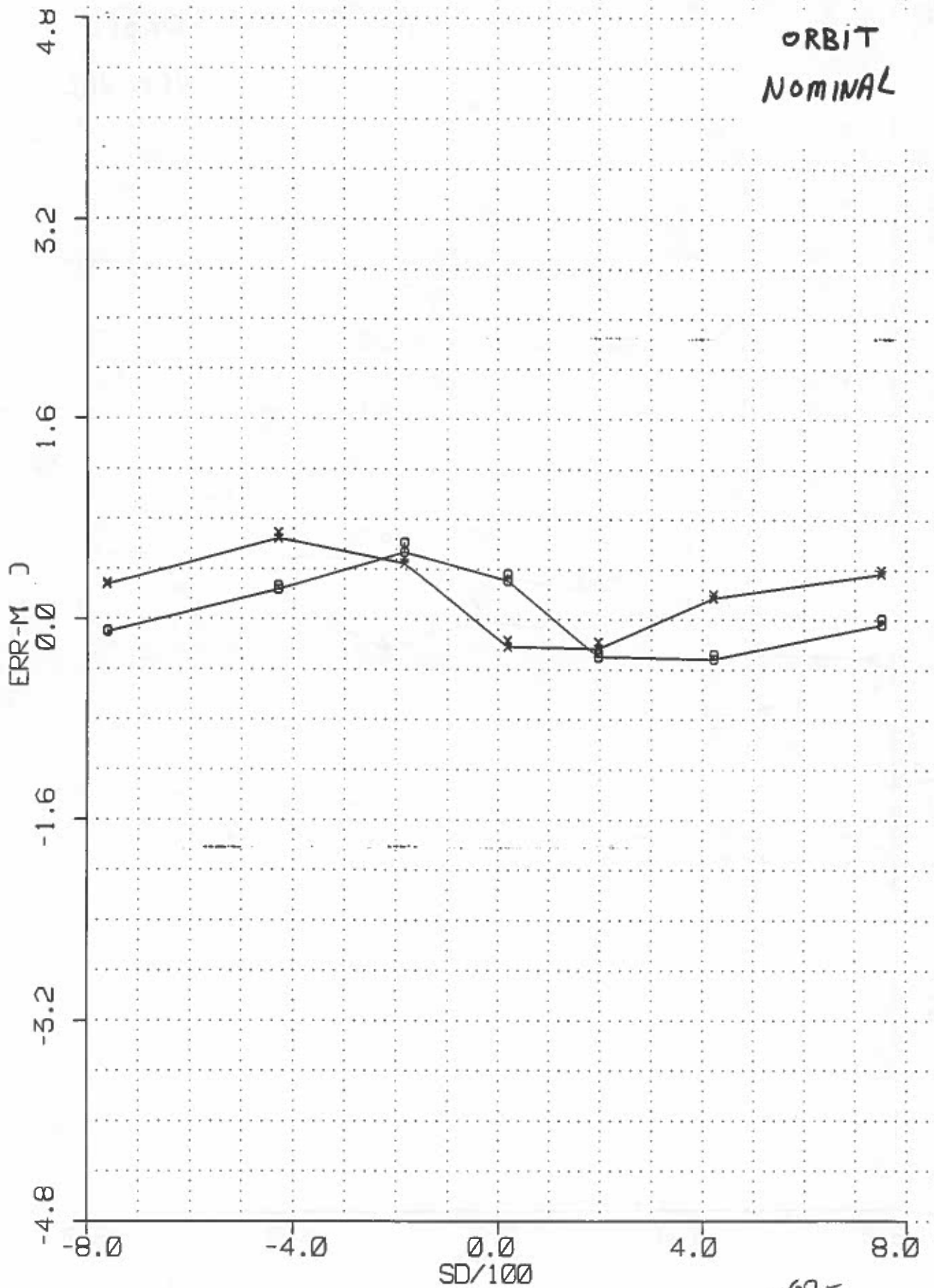
A

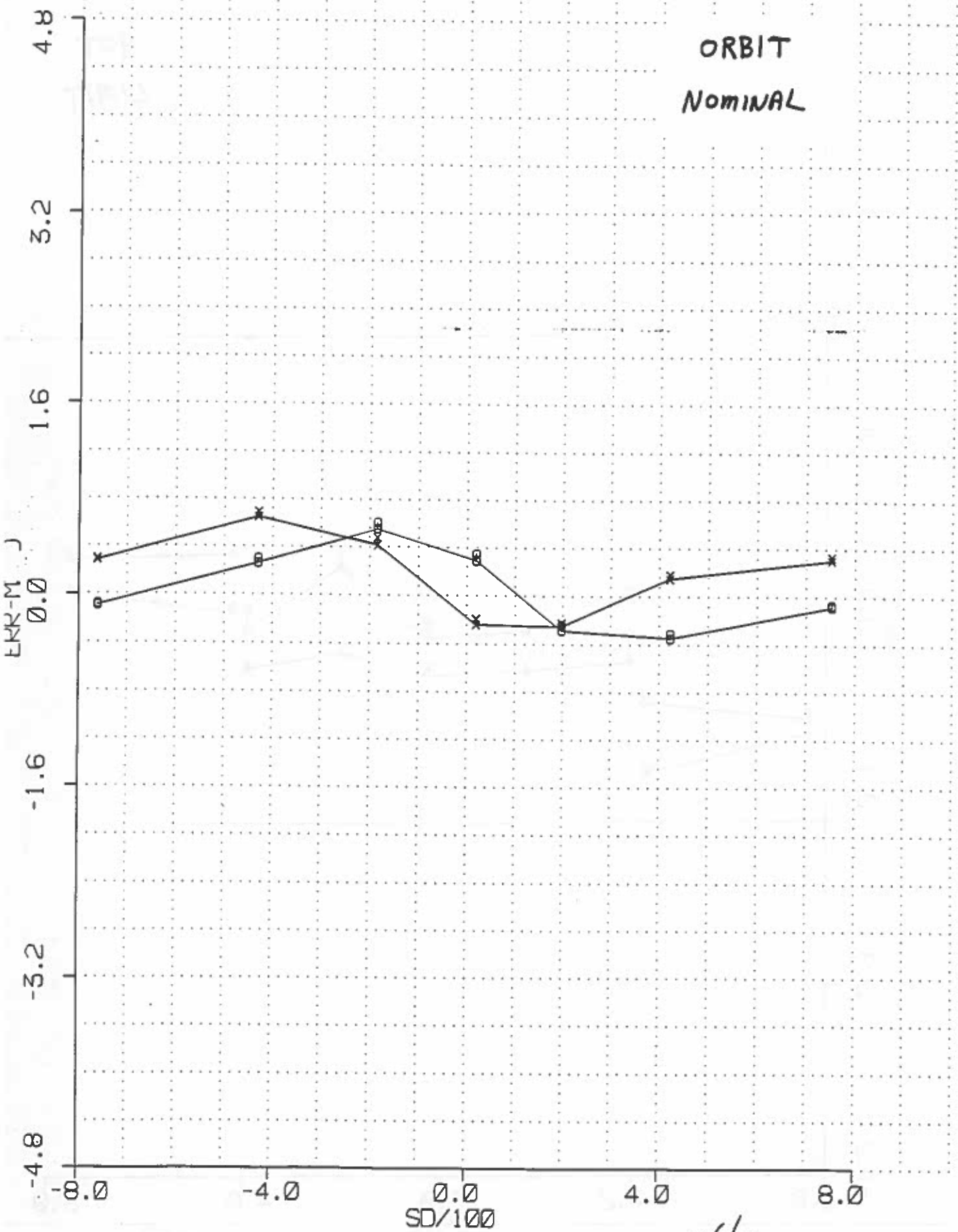
SYSTEM 14 ,AS/AT,H/T/P,MODE= ,SSS=5 ,M1=-8 ,DATE: 330

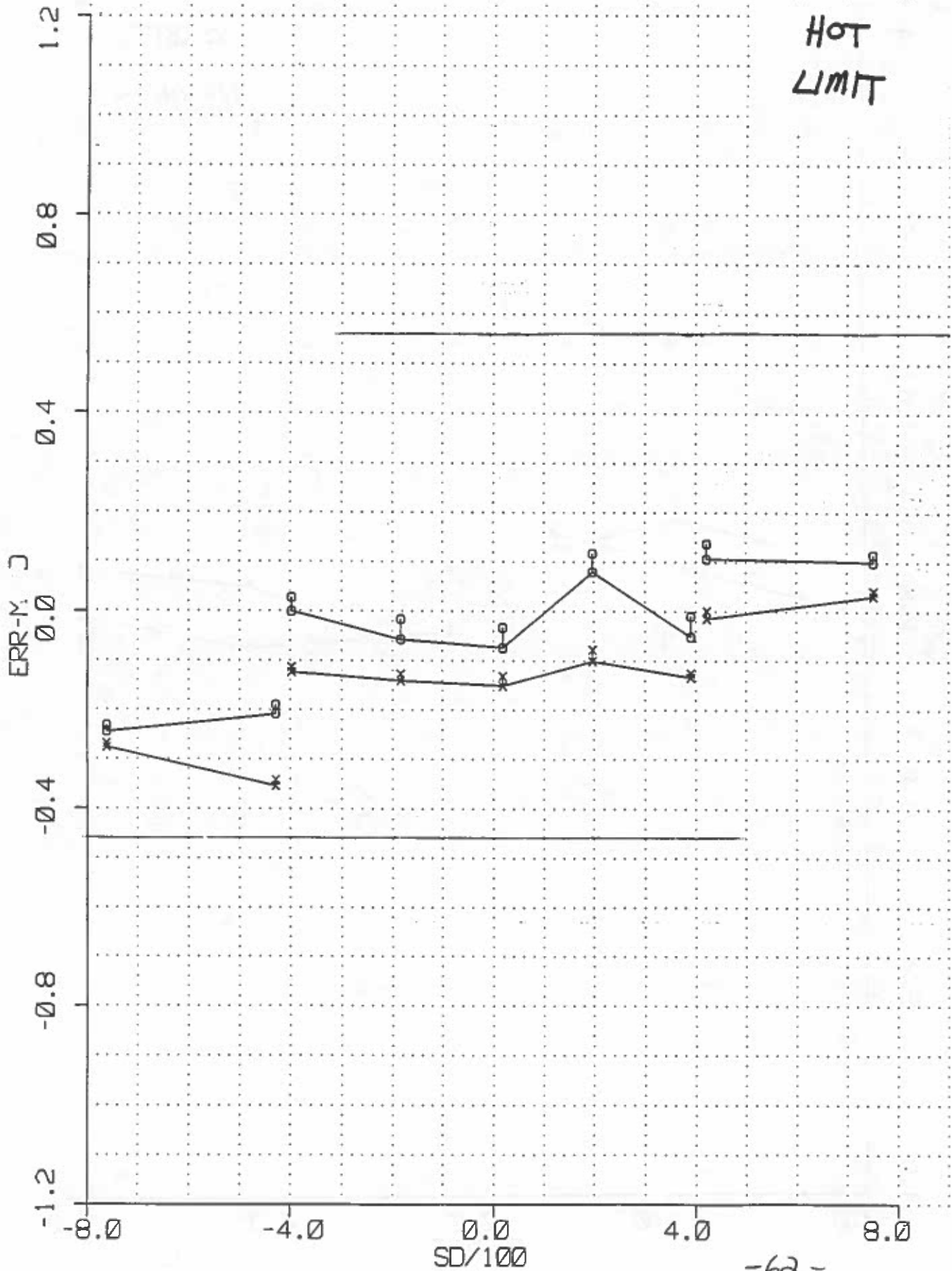


B

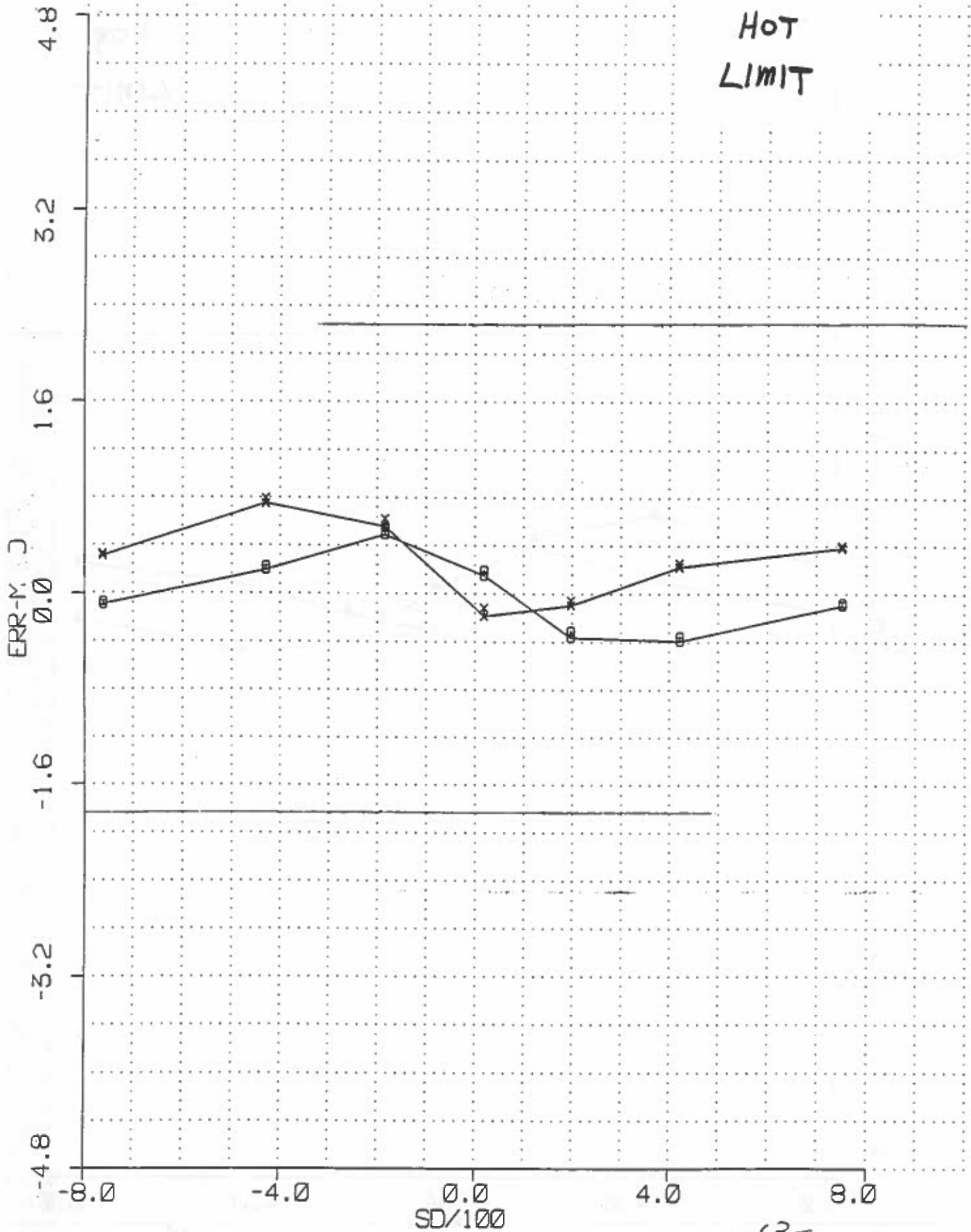
SYSTEM 14, AS/AT, H/T/P, MODE=, SSS=5, M1=-8, DATE: 330

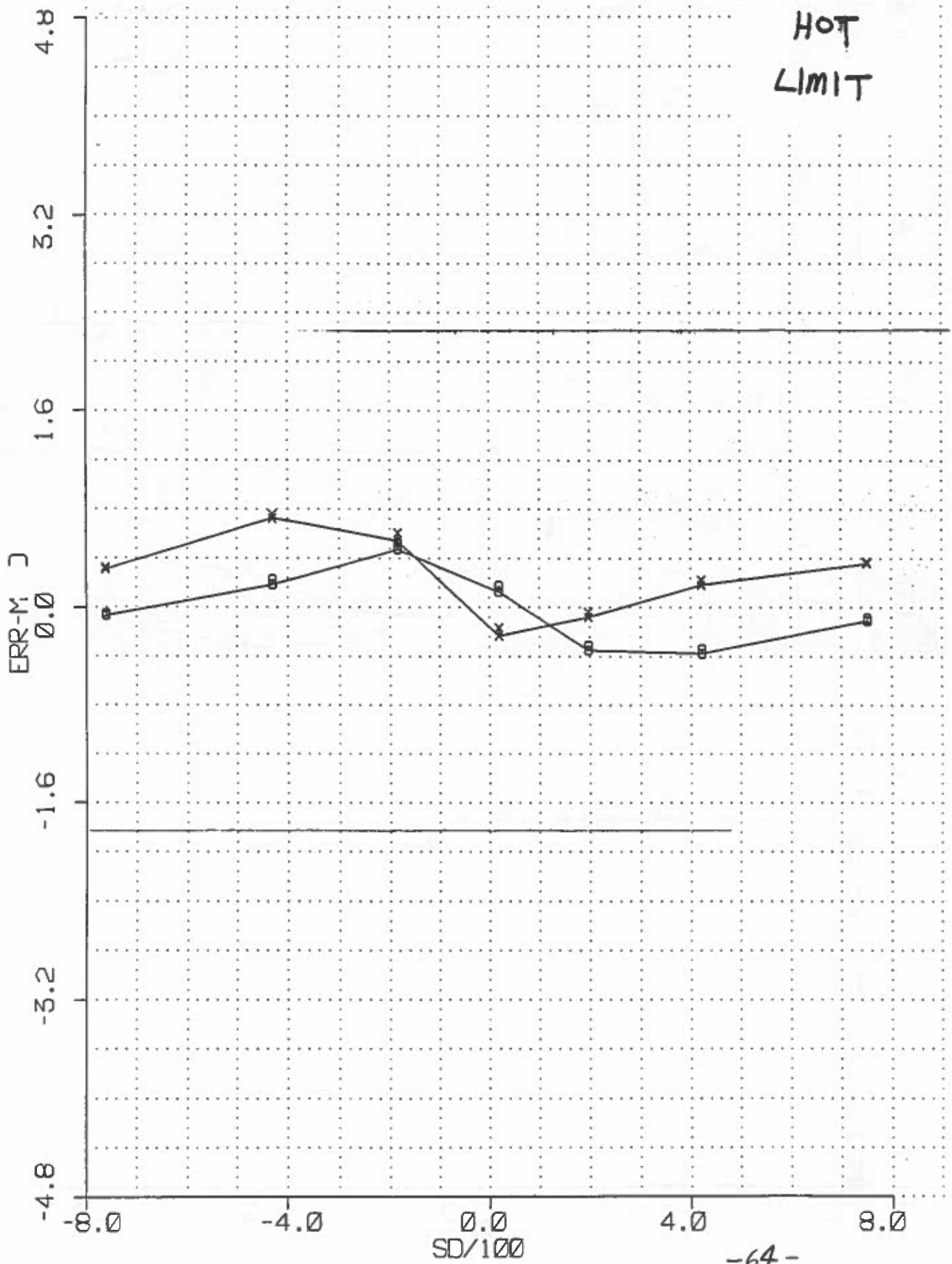






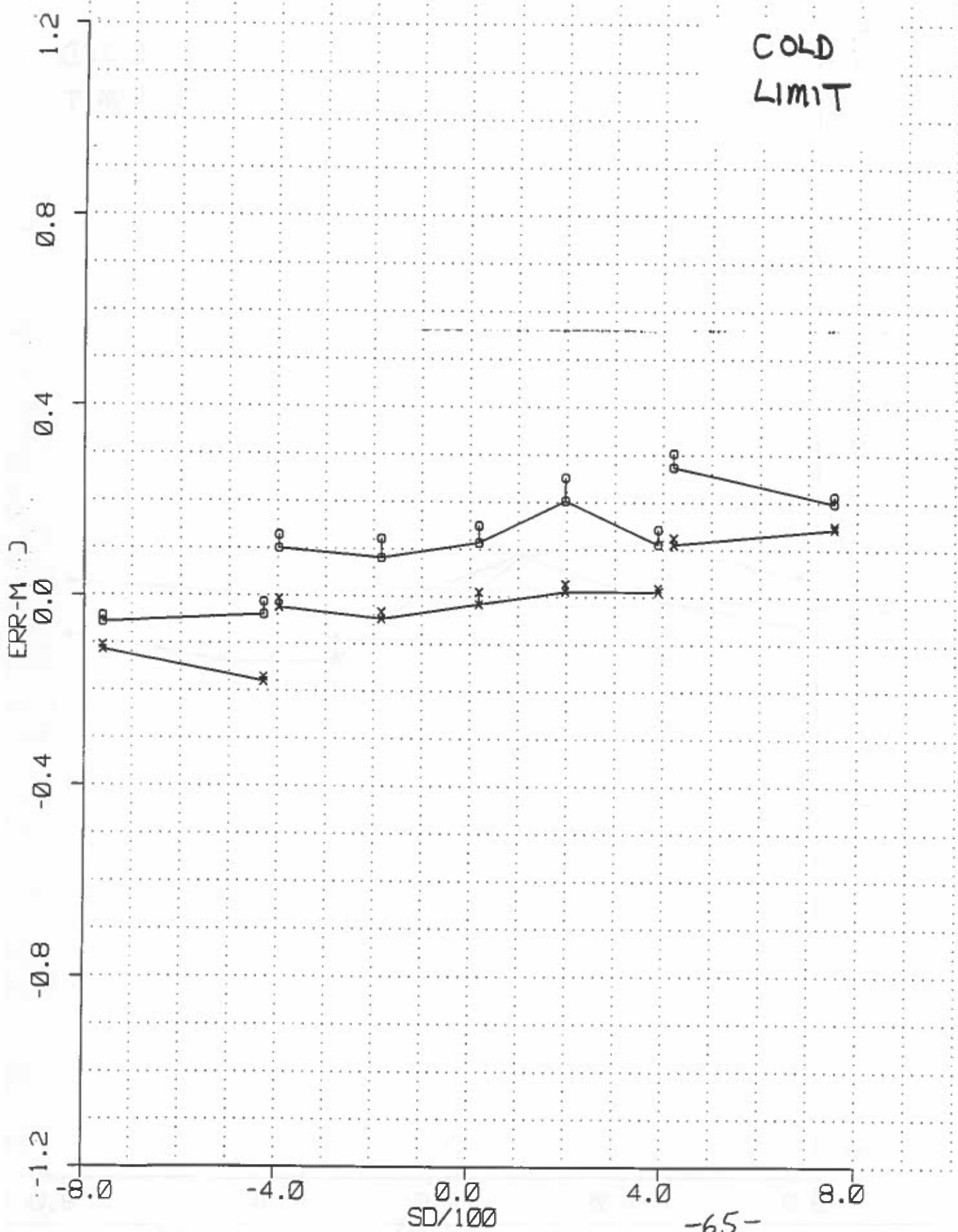
SYSTEM 14 , AS/AT , H/T/P , MODE = , SSS=7 , M1=12 , DATE : 319





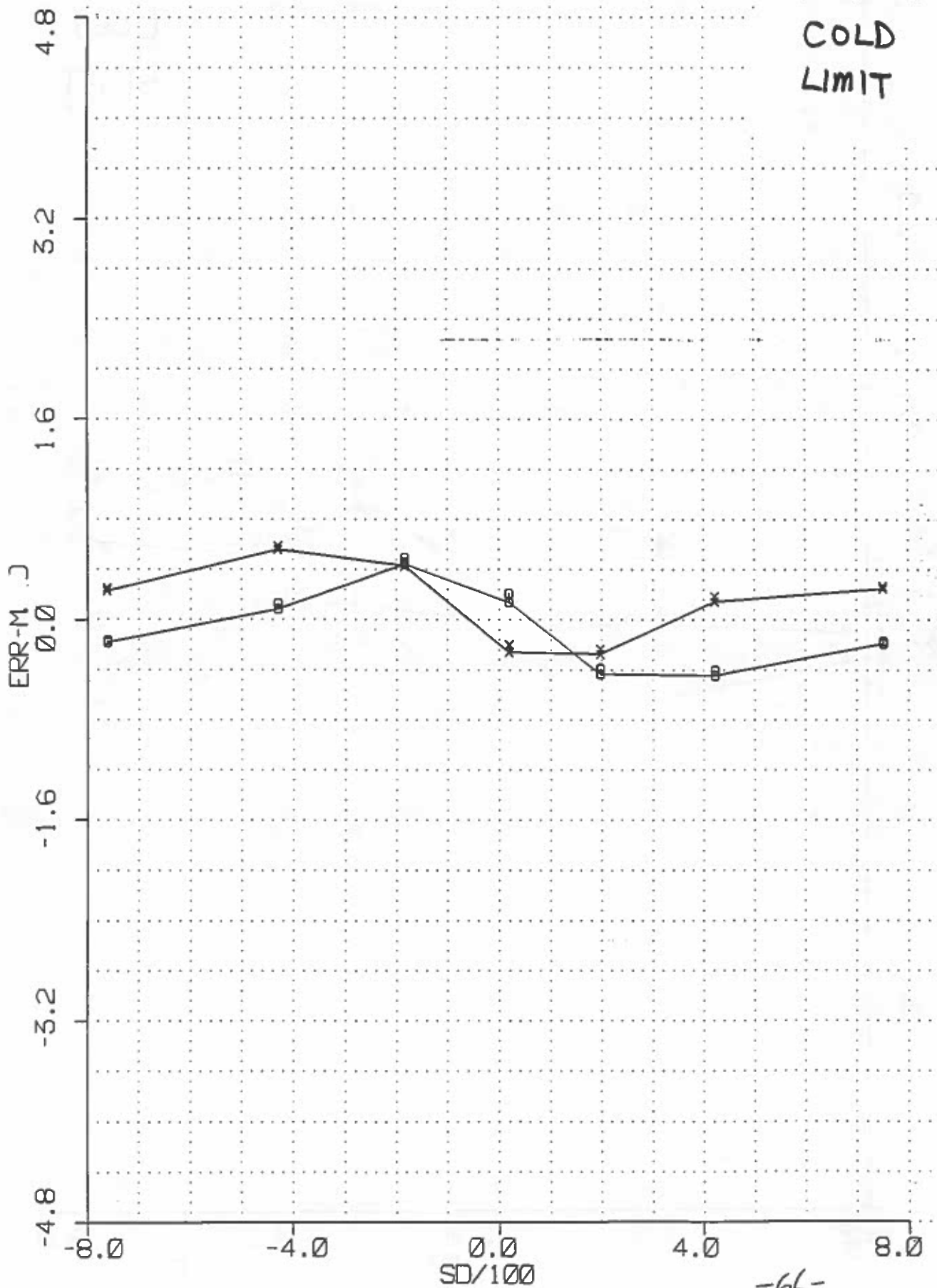
A

SYSTEM 14, AS/AT, H/T/P, MODE=, SSS=3, M1=-8, DATE: 326



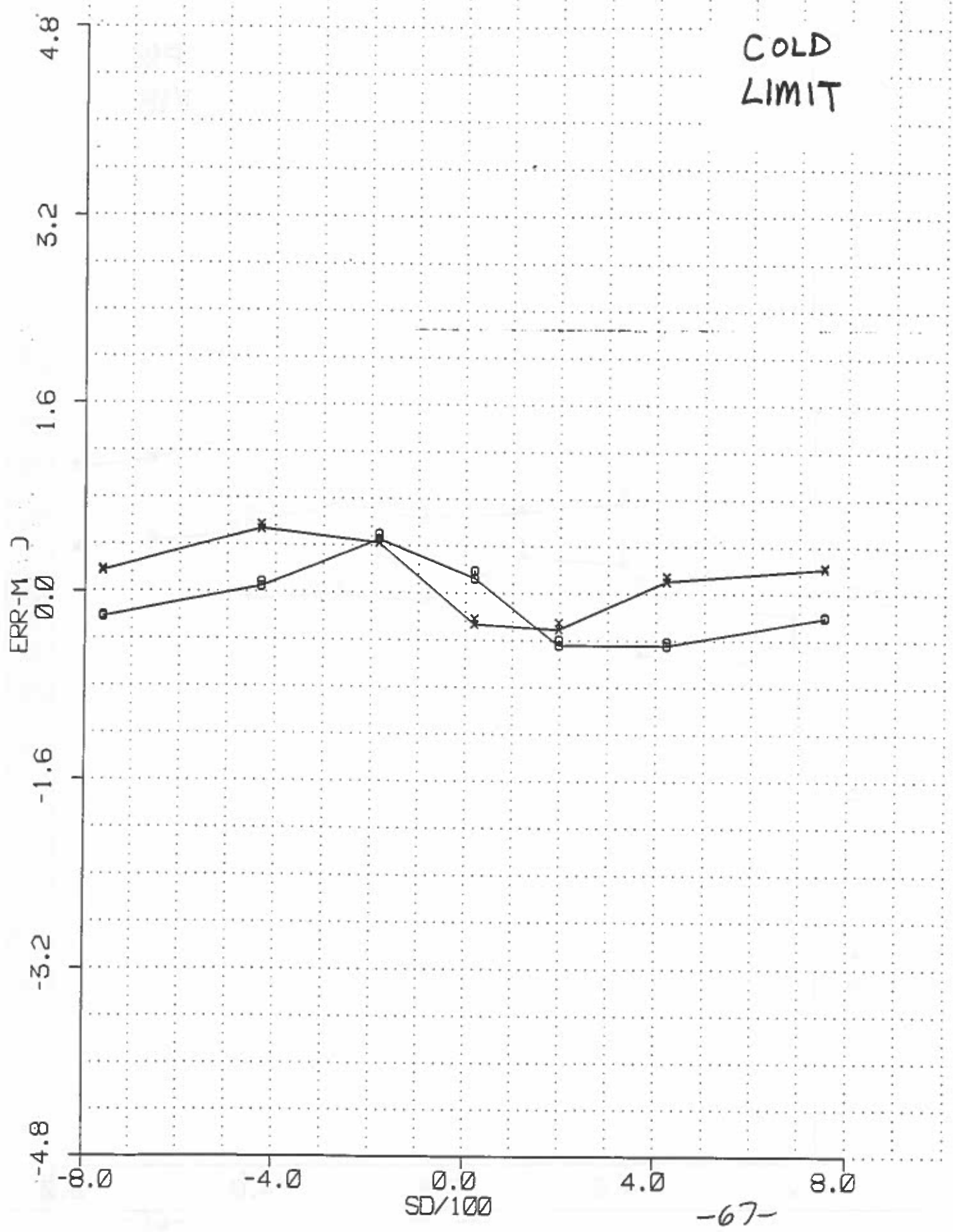
B

SYSTEM 14 , AS/AT, (H) T/P, MODE= , SSS=3 , M1=-8 , DATE: 326



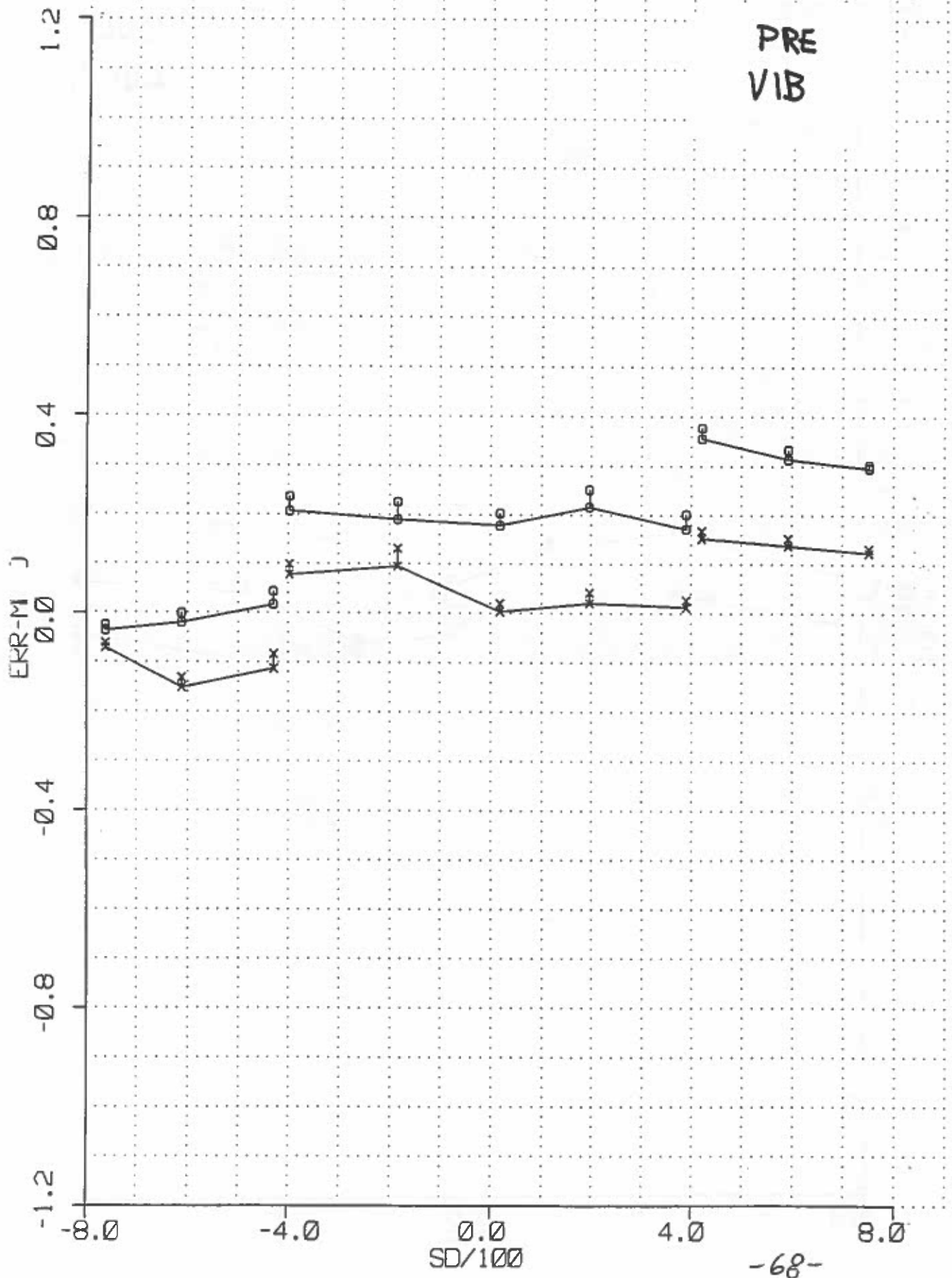
C

SYSTEM 14 , AS/AT (H) T/P, MODE = , SSS=3 , M1 = -8 , DATE: 326



"A"

SYSTEM 14
IMC-NORM HRD SYNC SDF SSS=23, M1=23, DATE: 112

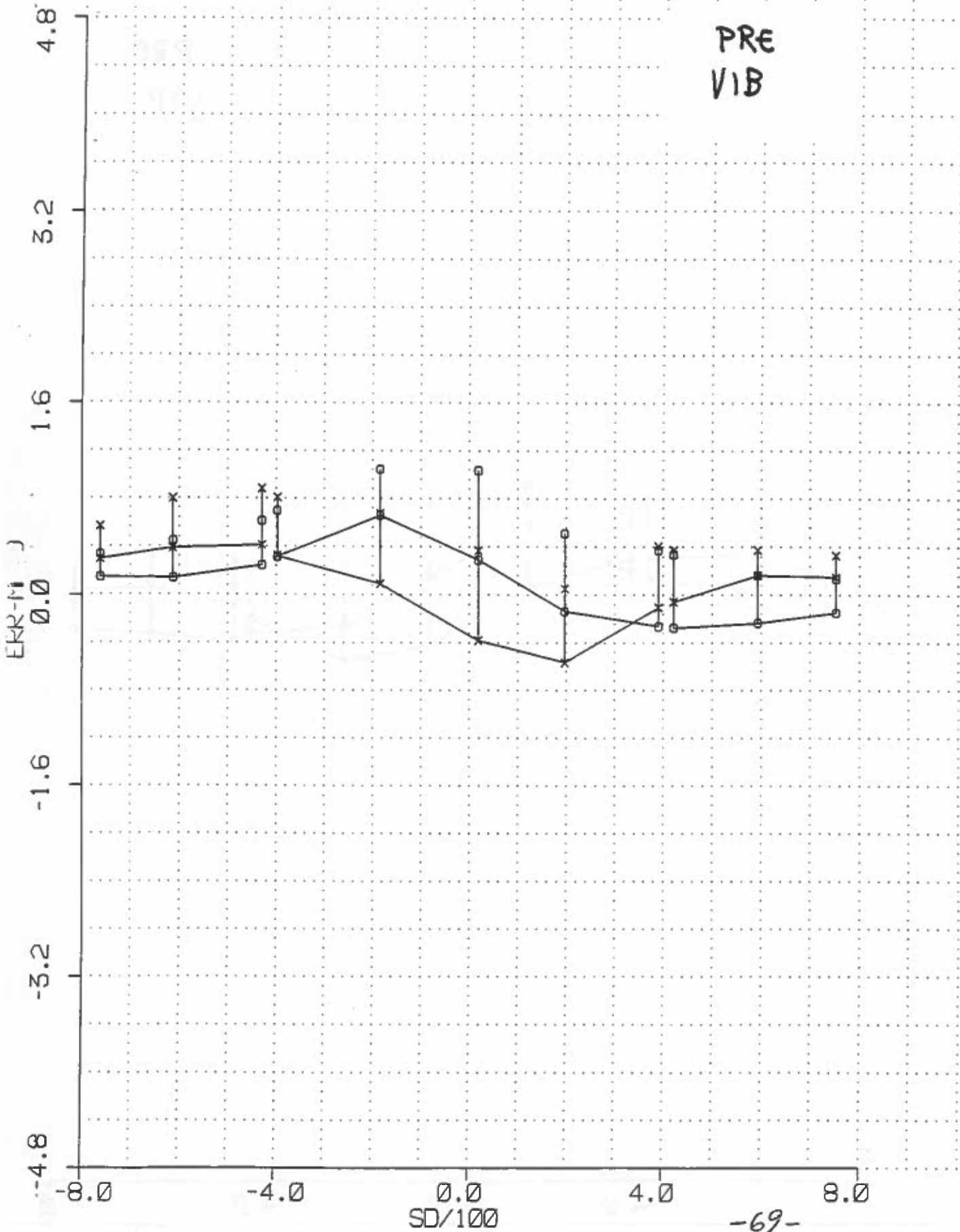


PRE
VIB

"B"

SYSTEM 14, AS/AT, H/T/P, MODE = SDF, SSS = 23, M1 = 23, DATE: 1124

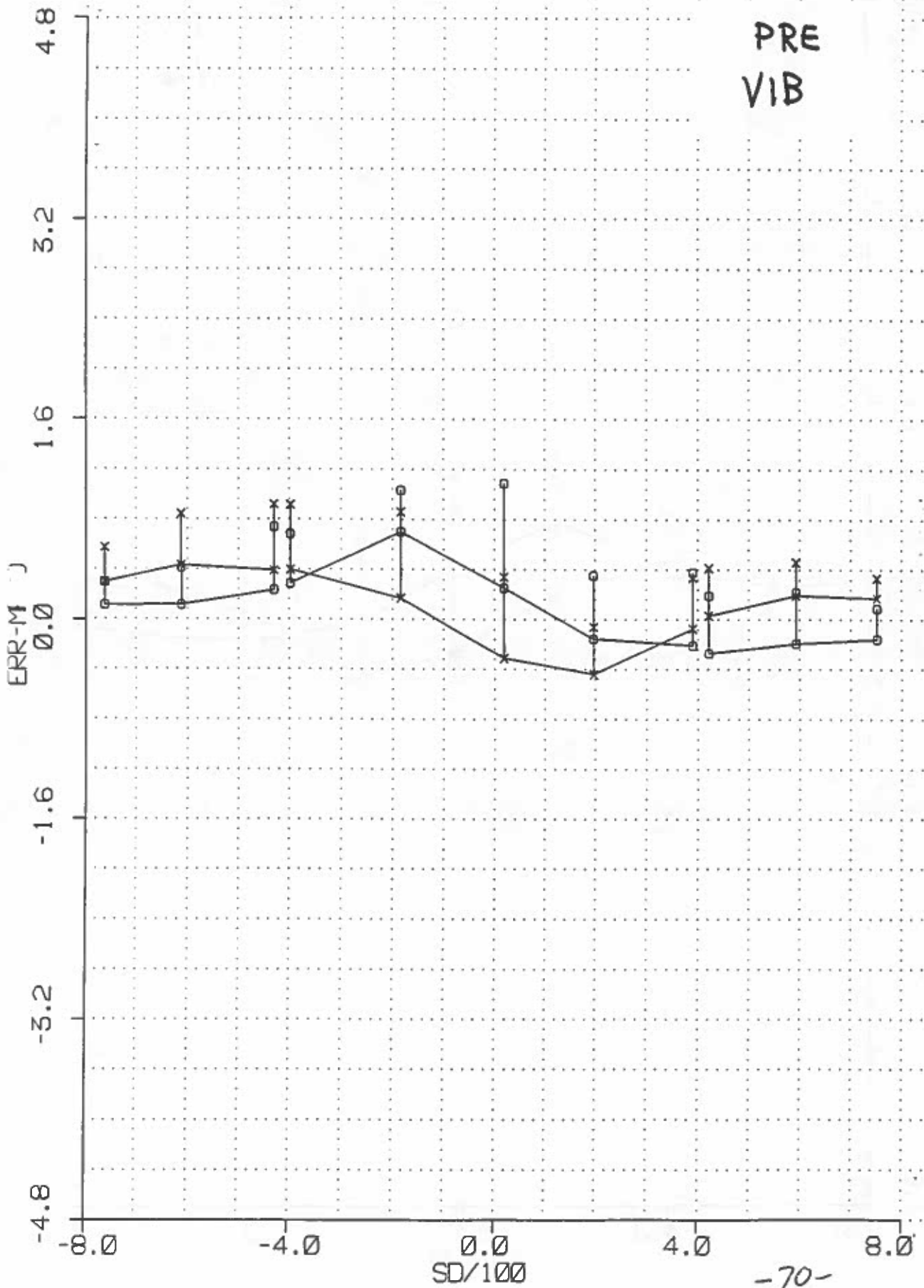
PRE
VIB



"C"

SYSTEM 14, AS/AT, H/T/P, MODE=, SSS=23, M1=23, DATE: 112

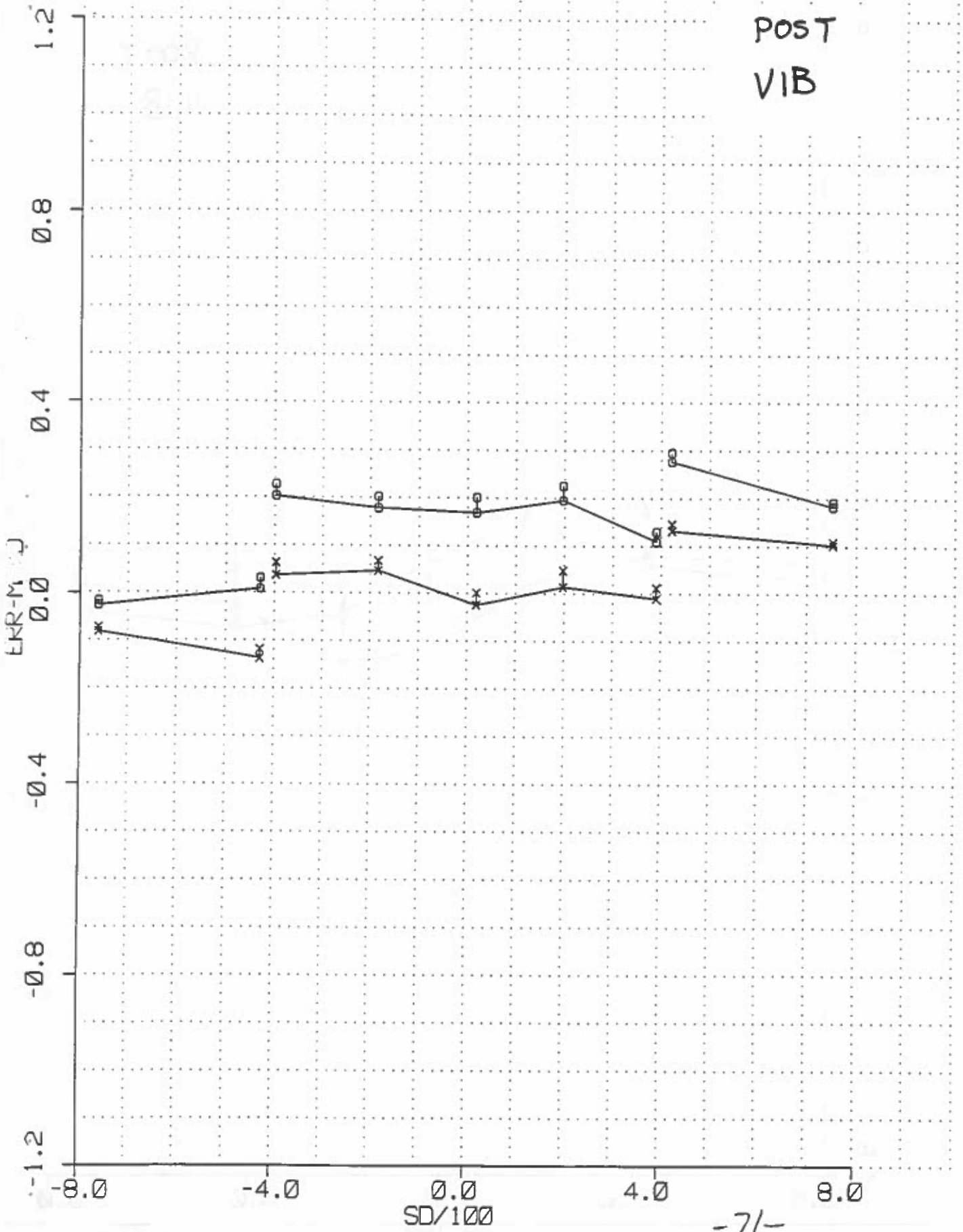
PRE
VIB



A

SYSTEM 14 (ASYM, BYT/P, MODE=SDF, SSS=23, M1=23, DATE: 1220/92

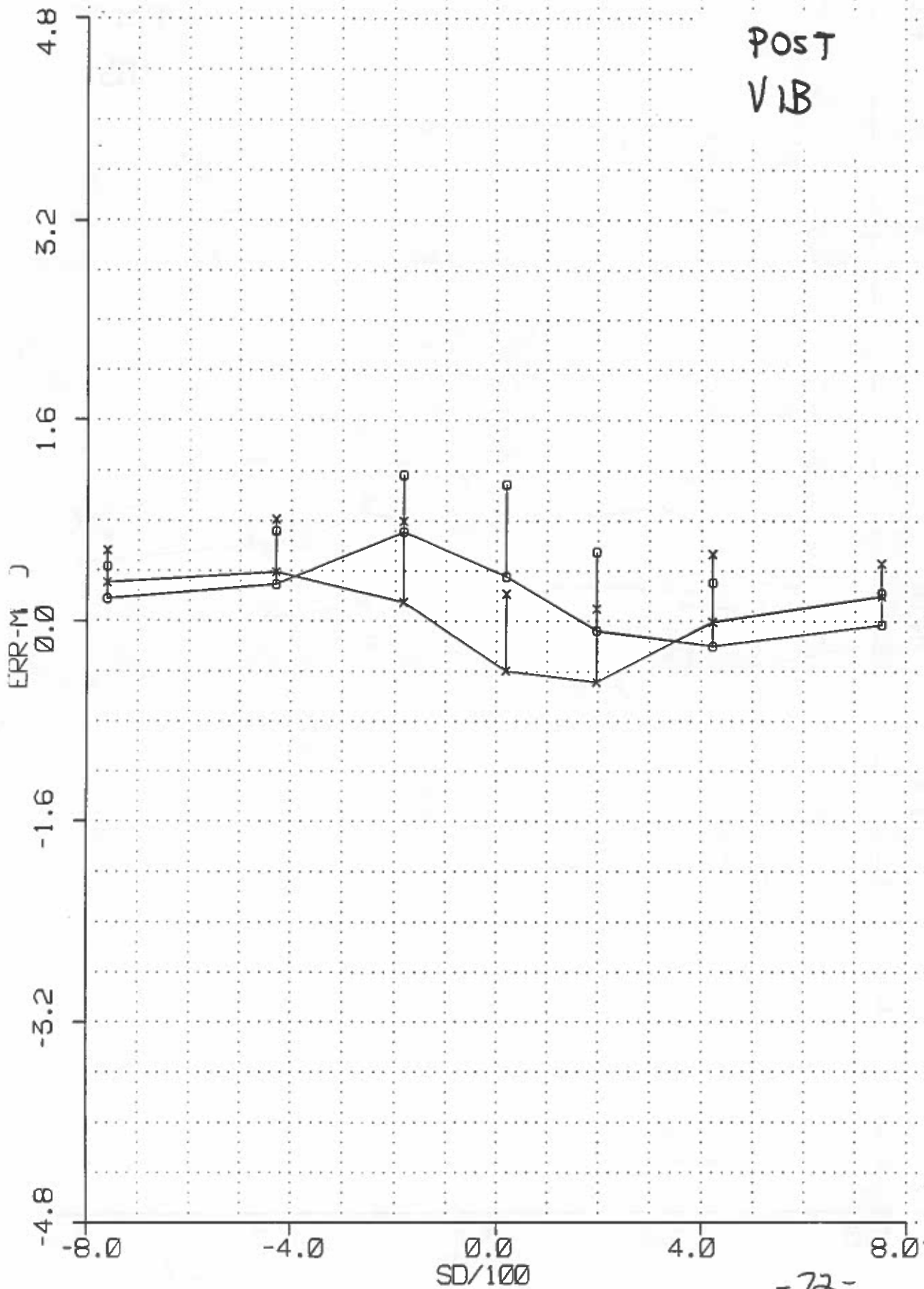
POST
VIB



"B"

SYSTEM 14 , AS/AT, H/T/P, MODE = SDF , SSS = 23 , M1 = 23 , DATE : 122

POST
VIB



2"

SYSTEM 14 , AS/AT, B/T/P, MODE= , SSS=23 , M1=23 , DATE: 1220/92

POST
VIB

